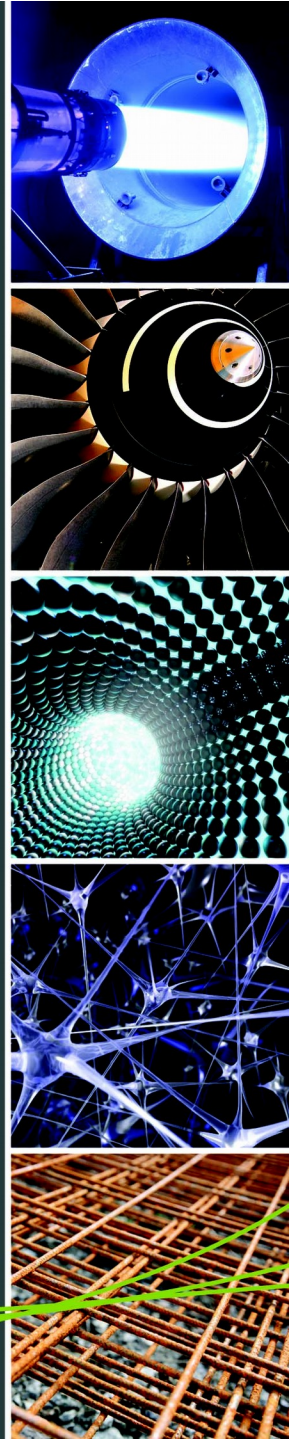




Swansea University  
Prifysgol Abertawe

# Advanced Structural Analysis EGF316

## Stress and Strain relationships



# Lecture Content



Swansea University  
Prifysgol Abertawe

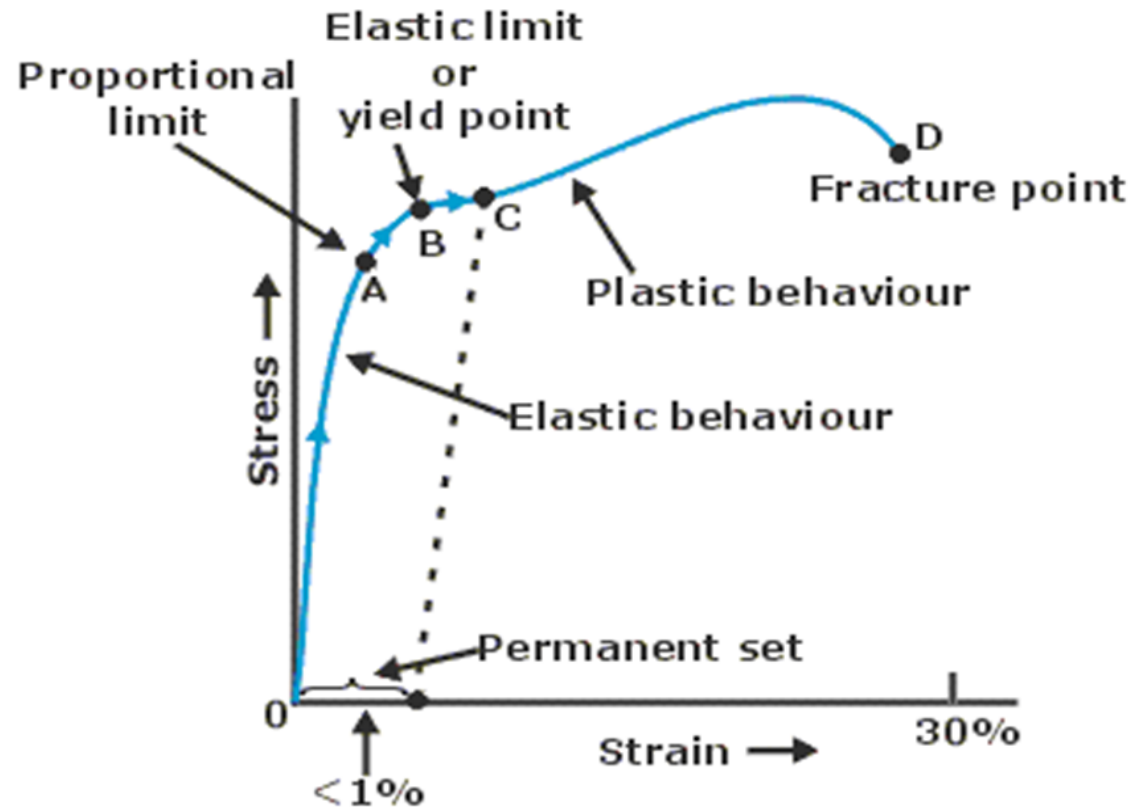
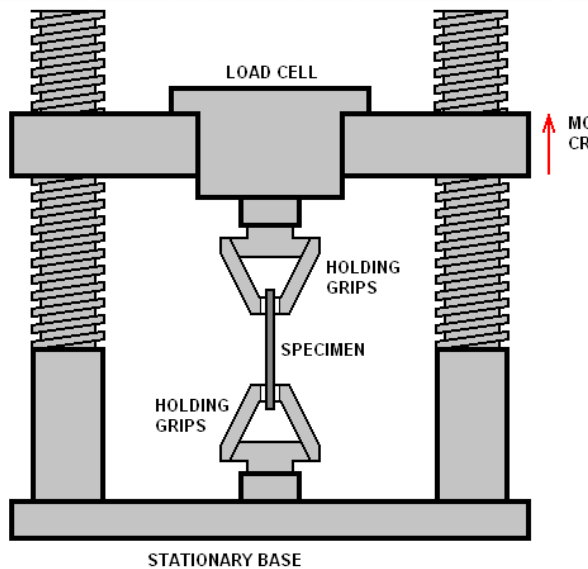
- Stress-strain diagram
- Hooke's law
- Young's modulus ( $E$ )
- Poisson's ratio ( $\nu$ )
- Plane stress
- Plane strain

# Stress-strain relations

- Stress =  $f_1(\text{strain})$
- Strain =  $f_2(\text{stress})$
- Varies from simple linear mapping to quite complicated models depending upon the material under consideration and also the strain regime (small or large/finite strains).
- Also, the relationships vary depending upon the Stress and Strain measures/definitions used.
- Tensile testing using Universal Testing Machine



# Universal Testing Machine (UTM)



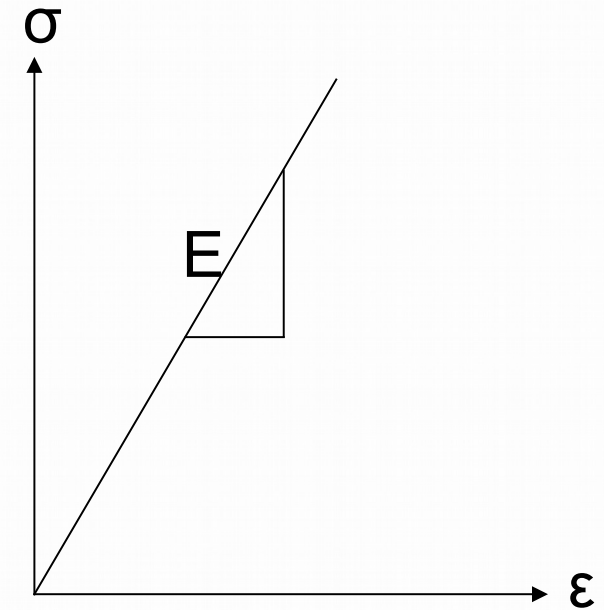
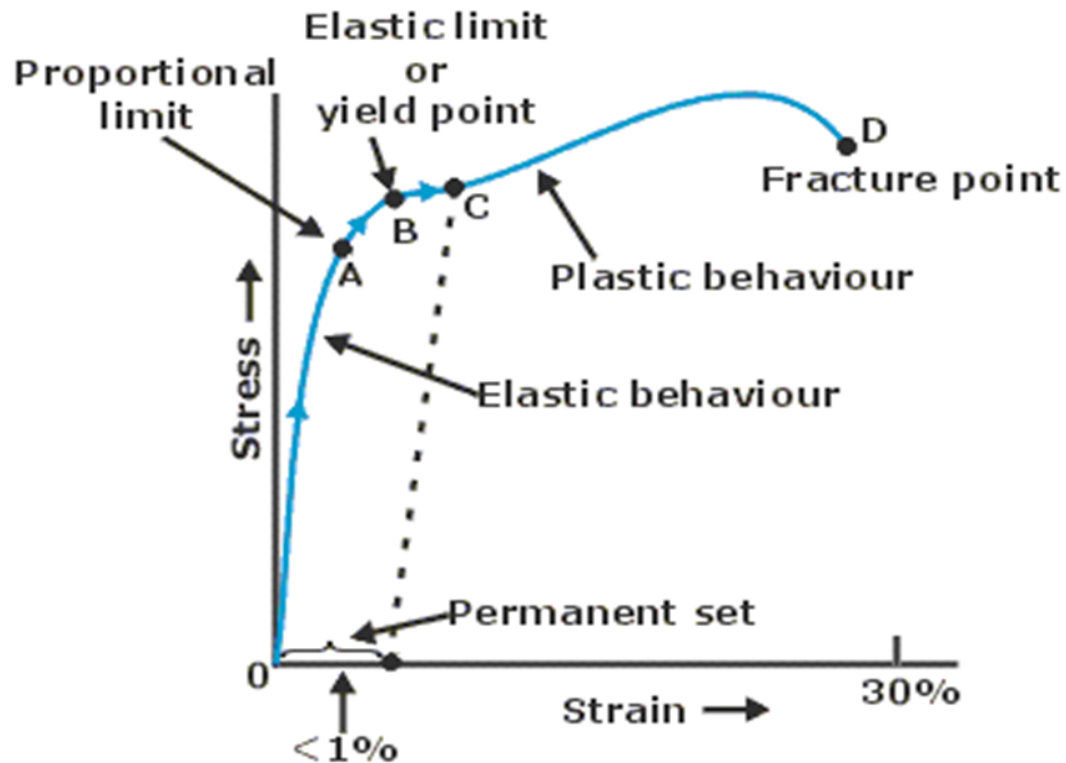
Stress-strain graph for ductile material



# Tensile Testing – ductile Vs brittle materials



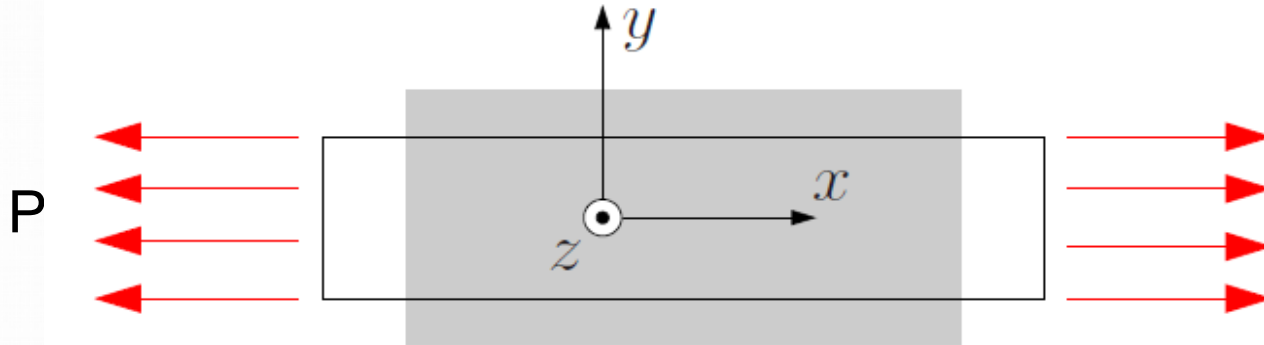
# Linear Elasticity - Hooke's law



$$\sigma = E \epsilon \rightarrow \text{Hooke's law}$$

$E$  = Young's modulus

# Linear Elasticity – uniform bar

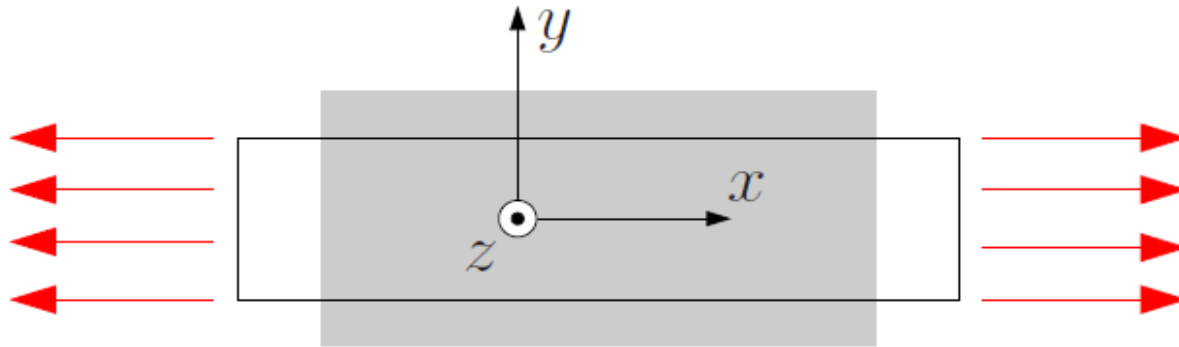


$$\sigma = \frac{P}{A}$$

$$\epsilon = \frac{\delta L}{L}$$

$$E = \text{constant} = \frac{\text{stress}}{\text{strain}} = \frac{\sigma}{\epsilon} = \frac{P}{A} / \frac{\delta L}{L} = \frac{PL}{A\delta L}$$

# Poisson's Ratio



$$\nu = - \frac{\text{lateral strain}}{\text{longitudinal strain}}$$

The minus sign compensates for the fact that the lateral and axial strains, in general, have opposite signs.

$$\nu = - \frac{\epsilon_{yy}}{\epsilon_{xx}} = - \frac{\epsilon_{zz}}{\epsilon_{xx}}$$



# Hooke's law in shear

$$\tau = G\gamma$$

G is called the **shear modulus** or the **modulus of rigidity**

$$G = \frac{E}{2(1 + \nu)}$$

# Types of materials

- ◆ Homogeneous
  - ◆ Material properties are same everywhere
- ◆ Nonhomogeneous
  - ◆ Material properties depend upon the location
- ◆ Isotropic
  - ◆ Material properties are same in every direction
- ◆ Anisotropic
  - ◆ Material properties are different in different directions

**We focus on Linear, Homogeneous and Isotropic materials.**

# Direct-Stress-Strain Relationship

$$\begin{aligned}\varepsilon_{xx} &= \frac{1}{E} [\sigma_{xx} - \nu\sigma_{yy} - \nu\sigma_{zz}] \\ \varepsilon_{yy} &= \frac{1}{E} [\sigma_{yy} - \nu\sigma_{xx} - \nu\sigma_{zz}] \\ \varepsilon_{zz} &= \frac{1}{E} [\sigma_{zz} - \nu\sigma_{xx} - \nu\sigma_{yy}]\end{aligned}$$

$$\begin{aligned}\sigma_{xx} &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\varepsilon_{xx} + \nu\varepsilon_{yy} + \nu\varepsilon_{zz}] \\ \sigma_{yy} &= \frac{E}{(1+\nu)(1-2\nu)} [\nu\varepsilon_{xx} + (1-\nu)\varepsilon_{yy} + \nu\varepsilon_{zz}] \\ \sigma_{zz} &= \frac{E}{(1+\nu)(1-2\nu)} [\nu\varepsilon_{xx} + \nu\varepsilon_{yy} + (1-\nu)\varepsilon_{zz}]\end{aligned}$$

# Shear Stress-Strain Relationship

3D state of stress:

$$\gamma_{xy} = \frac{\tau_{xy}}{G}, \quad \gamma_{yz} = \frac{\tau_{yz}}{G}, \quad \gamma_{zx} = \frac{\tau_{zx}}{G}$$

$$\gamma_{xy} = \frac{2(1 + \nu)\tau_{xy}}{E}, \quad \gamma_{yz} = \frac{2(1 + \nu)\tau_{yz}}{E}, \quad \gamma_{zx} = \frac{2(1 + \nu)\tau_{zx}}{E}$$



# 2D stress state - Plane Stress

Stresses in the plane normal to the plane of deformation are assumed to be zero.

$$\sigma_{zz} = \tau_{xz} = \tau_{yz} = 0$$

Also  $\gamma_{xz} = \gamma_{yz} = 0$

$$\begin{aligned}\epsilon_{xx} &= \frac{1}{E} [\sigma_{xx} - \nu\sigma_{yy}] \\ \epsilon_{yy} &= \frac{1}{E} [\sigma_{yy} - \nu\sigma_{xx}] \\ \epsilon_{zz} &= \frac{1}{E} [-\nu\sigma_{xx} - \nu\sigma_{yy}]\end{aligned}$$

$$\begin{aligned}\sigma_{xx} &= \frac{E}{(1 - \nu^2)} [\epsilon_{xx} + \nu\epsilon_{yy}] \\ \sigma_{yy} &= \frac{E}{(1 - \nu^2)} [\nu\epsilon_{xx} + \epsilon_{yy}]\end{aligned}$$

# 2D stress state - Plane Strain

Strains in the plane normal to the plane of deformation are assumed to be zero.

$$\epsilon_{zz} = \gamma_{xz} = \gamma_{yz} = 0$$

Also  $\tau_{xz} = \tau_{yz} = 0$

$$\begin{aligned}\sigma_{xx} &= \frac{E}{(1+\nu)(1-2\nu)} [(1-\nu)\epsilon_{xx} + \nu\epsilon_{yy}] \\ \sigma_{yy} &= \frac{E}{(1+\nu)(1-2\nu)} [\nu\epsilon_{xx} + (1-\nu)\epsilon_{yy}] \\ \sigma_{zz} &= \frac{E}{(1+\nu)(1-2\nu)} [\nu\epsilon_{xx} + \nu\epsilon_{yy}]\end{aligned}$$

Find expressions for the strains in terms of stresses.

# Plane Stress Example:

The principal strains at a point on a loaded thin plate are found to be  $320 \times 10^{-6}$  and  $200 \times 10^{-6}$ . What are the principal stresses if the modulus of elasticity is 200GPa and the Poisson's ratio is 0.3?

# Plane Strain Example:

A strain gauge rosette is attached to a thin flat steel plate. When loaded in-plane, the following strain values are measured:

$$\varepsilon_A = 500 \times 10^{-6}$$

$$\varepsilon_B = -400 \times 10^{-6}$$

$$\varepsilon_C = -200 \times 10^{-6}$$

Gauge *A* lies along the  $x$  axis of the structure and Gauge *B* and *C* are at  $+45^\circ$  and  $-45^\circ$  to Gauge *A*, respectively. Determine:

- i) The in-plane strains and stresses
- ii) The in-plane principal strains
- iii) The principal stresses and the maximum shear stress

For steel, assume  $E = 200\text{GPa}$ ,  $\nu = 0.3$  and  $\sigma_{yield} = 300\text{MPa}$ .