

# Flying Carpet Testing and ROTCFD Analysis

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# Chapter 1

## Abstract

The author was introduced to the EACG, its projects, and standard practices. He acquired the proper safety training and was assigned to work on Glitter Belt (particularly the Flying Carpet concept) and ROTCFD projects. He was made familiar with the nature of CFD while awaiting of licence renewal for the necessary software. He was informed of the prior work on the Flying Carpet and the need for demonstration and testing models. A Flying Carpet wind tunnel model was designed and constructed, and is ready for testing. Concepts for a more advanced model with a deployable sheet were considered.

# Chapter 2

## Introduction

The author's work in the EACG is divided among two projects: Glitter Belt and ROTCFD. The former is concerned with using high altitude aircraft, carrying reflectors, to stop or reverse global warming, and the latter is concerned with using computational techniques to model the flow around the EACG's various test models to supplant experimental results. The two projects and the author's work on them are introduced separately below.

### 2.0.1 Glitter Belt

The Glitter belt project aims to reverse climate change by reflecting solar radiation out to space. The reflection will be accomplished by solar-powered aircraft carrying ultralight mylar sheets at approximately 100,000 feet of altitude. Cost analysis shows that this is feasible to do using government funding. The name "Glitter Belt" refers to the appearance the reflectors may have when viewed from space.

There are three different concepts for implementing the Glitter Belt: The Flying Carpet, the Quadrotor, and the Balloon Beanie. The author's work concerns primarily the first, which includes more challenging aerodynamic questions, appropriate of this lab group's title and purpose. In the Flying Carpet, the first, the mylar sheet is supported by aerodynamic lift. During the day, it is towed through the air by propellers, driven by electric motors. The propellers are mounted on a flying wing, and the motors are powered by solar cells on the wing. During the night, the aircraft maintains forward flight by gliding downward, using gravitational potential energy, staying above the upper limit of controlled airspace, 60,000 feet. Incidentally, this concept may also be useful for transportation on Mars, since the martian atmosphere at lower altitude is similar to that of earth at 100,000 feet.

The second concept is the Quadrotor. This involves supporting the sheets using four rotary wings. Thus far no feasible way has been

found to keep such an aircraft above 60,000 feet at night, so the author's work is not concerned with it.

The third and final concept is the Balloon Beanie. In this concept, a flat reflector sheet is supported by hydrogen balloons. Some solar powered rotors are included to provide trim and propulsion. (It will be necessary under certain conditions to move the aircraft, although most of the time it will drift in the wind.) This may be particularly useful near the poles, where due to the low angle of elevation of the sun, a horizontal reflector such as the Flying Carpet will be less effective. and the later nearer to the equator.

### 2.0.2 ROTCFD

CFD (Computational Fluid Dynamics) is a branch of aerodynamic research where computational methods are used to solve the Navier-Stokes equations in order to model the flow around a body. It is necessary to model the flow around our test models in addition to obtaining experimental results, in order to better understand the flow around them.

# Chapter 3

## Define Objectives

### Objectives:

- Create CFD simulations of wind tunnel experiments to the desired degree of accuracy
- Build a scale model of the Flying carpet which can:
  - carry a reflector sheet internally, as in the climb phase of the mission
  - deploy the sheet
  - hold the sheet steady (without significant flutter) during a wind tunnel test

# Chapter 4

## Prior Work

Cost analysis shows that the Glitter Belt project can be implemented using government expenditures. It also shows that the flying carpet will probably be cheaper to produce per unit reflector area than the Balloon Beanie. However, the Balloon Beanie has the unique property of being capable of orienting to be normal to the Sun's rays no matter the angle. This eliminates the need to place them on the part of the Earth directly beneath the Sun, near the poles in particular (for the purpose of stopping ice melting). The author suggests that both concepts may be manufactured, and the Flying Carpet may be deployed beneath the Sun and the Balloon Beanie may be deployed near the poles.

The primary challenge of the Flying Carpet concept is keeping the reflector sheet smooth and flat. Present design calls for a sheet of reflective mylar trailing behind the wing. However, wind tunnel testing has shown that the sheet oscillates in a self-excited manner. When the aspect ratio is high, the oscillations propagate spanwise. When the aspect ratio is low (i.e. less than 1) the oscillations are longitudinal. Moving the sheet away from the wing, or making spanwise slits in it does not help. Limited success has been achieved by introducing rigid structures made of drinking straws into the sheet. The oscillations are detrimental in that they increase drag and reduce the effective area of the sheet.

Another design calls for stretching the sheet by its four corners which will be connected to a rigid frame. This causes a problem because the sheet bends upward like a parachute, which has inferior aerodynamic and reflective characteristics.



# Chapter 5

## Project Schedule

- Monday morning 2-5-17: Show existing model to Prof. Komerath and obtain directives for further development/testing
- Monday afternoon 2-5-17:
  - Make any necessary changes to the model
  - Finalize deployable sheet concept and design model
  - Check availability of CFD software
- Friday 2-9-17:
  - Begin construction of deployable sheet model
  - Begin CFD work if software is available

# Chapter 6

## Experimental Setup

An experiment has yet to be formally designed. It will involve attempting to float the model using the wind generated by the wind tunnel's ventilation fan, and running wind tunnel tests on the model to see which configurations minimize sheet oscillations.

### 6.0.1 Model Details

The model will be made primarily of styrofoam and balsa wood. The reflective sheet will be 60cm span by 30cm chord. It will be attached to the top of the winglets using a dowel 92cm long. The wing will be of rectangular planform, 71cm span by 10cm chord. The winglets will be 10.5 cm high by 10cm chord. They will connect to the wing by means of two pins each. By using pins bent at different angles and inserting them into different holes in the winglets, the winglets may be installed at any dihedral and sweep angle. The wings will be constructed primarily of styrofoam. A balsa wood spar may be inserted for additional stiffness and robustness to bending moments.

Before designing this model, the author and his colleagues were given rather various mission requirements. The need to make a model that could support its own weight using lift was posited, in addition to the need for a modular test model where the different winglet configurations could be tested. This model is expected to meet both requirements by being both modular and light.

### 6.0.2 Load Cell Details

A load cell has not been selected.

### 6.0.3 Calculations of expected forces and moments

The priority thus far has been to produce a model, not to conduct analysis.

### 6.0.4 Static testing of the model

Testing procedures are TBD.

## Chapter 7

### Flow and Test Conditions

The model will be tethered and exposed to gentle wind in the wind tunnel doorway to test its ability to support itself. Then the model will be held at fixed low angle of attack to observe sheet oscillations.

## Chapter 8

### Expected results and plots

Not yet determined

## Chapter 9

### Conclusions

Now that information concerning objectives and previous work has been obtained, model designs are being finalized. The Flying Carpet model can be expected to be complete soon (next week at the latest). CFD work will commence as soon as software is available.

## Bibliography