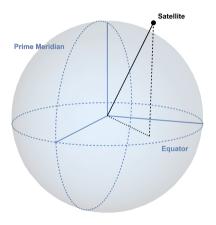
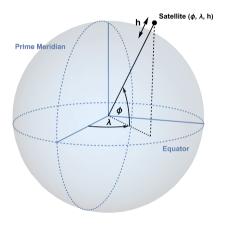
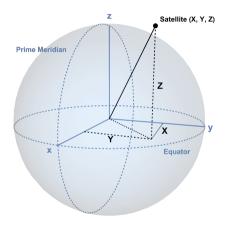
Part 8: Solving Building a GPS receiver from scratch

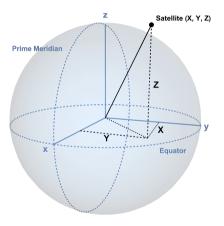
Chris Doble

- The pseudorange equation
- 2 Satellite location and transit time
- System of equations
 - Definition
 - Solving

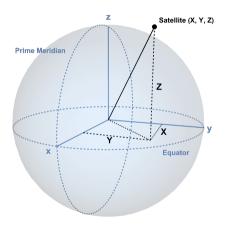




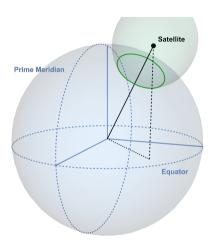




Transit time T



Transit time $T \Rightarrow \text{distance } cT$



Transit time $T \Rightarrow \text{distance } cT$

$$\sqrt{(X-x)^2 + (Y-y)^2 + (Z-z)^2} = cT$$

where (x, y, z) is our unknown location.

$$\sqrt{(X-x)^2 + (Y-y)^2 + (Z-z)^2} - cT = 0$$

where (x, y, z) is our unknown location.

$$\sqrt{(X-x)^2+(Y-y)^2+(Z-z)^2}-c(T-t)=0$$

where (x, y, z) is our unknown location and t is our clock bias.

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The user shall correct the time received from the SV with the equation (in seconds)

$$t = t_{sv} - \Delta t_{sv} \qquad (1)$$

where

= GPS system time (seconds),

t_{sv} = effective SV PRN code phase time at message transmission time (seconds),

 Δt_{sv} = SV PRN code phase time offset (seconds).

The SV PRN code phase offset is given by

$$\Delta t_{sv} \quad = \; a_{f0} + a_{f1}(t \text{ - } t_{oc}) + a_{f2}(t \text{ - } t_{oc})^2 + \Delta t_r \quad (2)$$

$$\vdots$$

Thus, the user who utilizes the L1 $\rm C/A$ signal only shall modify the code phase offset in accordance with paragraph 20.3.3.3.3.1 with the equation

$$(\Delta t_{SV})_{L1C/A} \ = \ \Delta t_{SV} \ \ \text{-} \ \ T_{GD}$$

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 - T_{GD}

$$t_{sv} = \mathsf{TOW} \times 6\,\mathsf{s}$$

$$t_{sv} = \mathsf{TOW} \times \mathsf{6s} + \mathsf{PRN} \; \mathsf{count} \times \mathsf{1ms}$$

• Record the time at which we finish receiving the PRN code (our clock)

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- Add 18 leap seconds
- Calculate the number of seconds since GPS started operating
- Calculate the remainder when divided by the number of seconds in a GPS week

Location

$$\mu = 3.986005 \times 10^{14} \text{ meters}^3/\text{sec}^2$$

$$\dot{\Omega}_{e} = 7.2921151467 \text{ x } 10^{-5} \text{ rad/sec}$$

$$A = (\sqrt{A})^2$$

$$\mathbf{n}_0 = \sqrt{\frac{\mu}{A^3}}$$

$$t_k = t - t_{\rm oc} *$$

$$\mathbf{n}=\mathbf{n}_0+\Delta\mathbf{n}$$

$$\mathbf{M}_k = \mathbf{M}_0 + \mathbf{n} \mathbf{t}_k$$

$$E_0 = M_k$$

$$E_{j} = E_{j-1} + \frac{M_{k} - E_{j-1} + e \sin E_{j-1}}{1 - e \cos E_{j-1}}$$

$$E_k = E_i$$

$$v_k = 2 \tan^{-1} \left(\sqrt{\frac{1+e}{1-e}} \tan \frac{E_k}{2} \right)$$

$$\Phi_{\nu} = \nu_{\nu} + \omega$$

$$\begin{split} \delta u_k &= c_{us} sin2\Phi_k + c_{uc} cos2\Phi_k \\ \delta r_k &= c_{rs} sin2\Phi_k + c_{rc} cos2\Phi_k \\ \delta i_k &= c_{is} sin2\Phi_k + c_{ic} cos2\Phi_k \end{split}$$

$$u_k = \Phi_k + \delta u_k$$

$$r_k = A(1 - e \cos E_k) + \delta r_k$$

$$i_k = i_0 + \delta i_k + (IDOT) t_k$$

$$x_k' = r_k cosu_k$$

 $y_k' = r_k sinu_k$

$$\Omega_k \; = \; \Omega_0 \; + \; (\stackrel{\bullet}{\Omega} \; - \stackrel{\bullet}{\Omega}_e) \; t_k \; - \; \stackrel{\bullet}{\Omega}_e \; t_{oe} \;$$

$$\begin{split} x_k &= x_k' cos \Omega_k - y_k' cosi_k sin \Omega_k \\ y_k &= x_k' sin \Omega_k + y_k' cosi_k cos \Omega_k \\ z_k &= y_k' sini_k \end{split}$$

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$$\sqrt{(X_1 - x)^2 + (Y_1 - y)^2 + (Z_1 - z)^2} - c(T_1 - t) = 0$$

$$\sqrt{(X_2 - x)^2 + (Y_2 - y)^2 + (Z_2 - z)^2} - c(T_2 - t) = 0$$

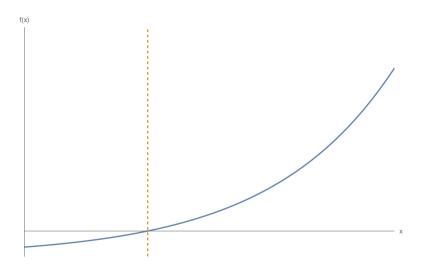
$$\sqrt{(X_3 - x)^2 + (Y_3 - y)^2 + (Z_3 - z)^2} - c(T_3 - t) = 0$$

$$\sqrt{(X_4 - x)^2 + (Y_4 - y)^2 + (Z_4 - z)^2} - c(T_4 - t) = 0$$

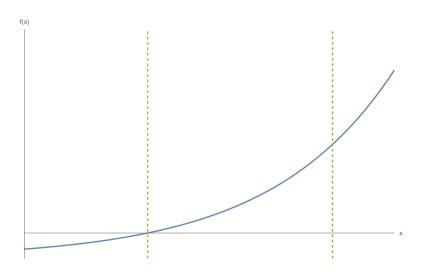
- The pseudorange equation
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The Newton-Raphson method

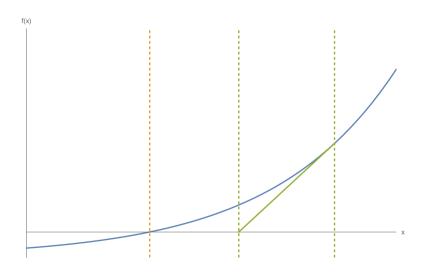
The Newton-Raphson method



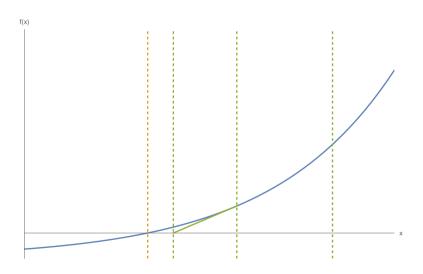
Chris Doble Part 8: Solving 15/

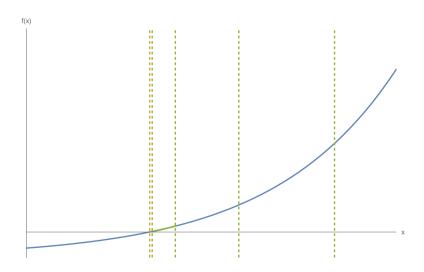


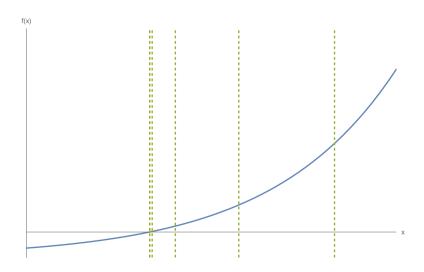
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Chris Doble Part 8: Solving 15/1







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- Convert from ECEF to geodetic coordinates using Bowring's method

Chris Doble Part 8: Solving 16/19

Hooray!



• We express locations using ECEF coordinates

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Chris Doble Part 8: Solving 18 / 19

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Chris Doble Part 8: Solving 18 / 19

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- The satellite's location can be calculated using equations in the GPS spec
- ullet We need a system of at least four pseudorange equations to solve for x, y, z, and t
- The Gauss-Newton algorithm is used to estimate x, y, z, and t

Thank you!

https://github.com/chrisdoble/gps-receiver