

Thermo HW 5

3. a) $\eta_{\text{Heat Engine}} = \frac{W}{Q_H} = 1 - \frac{T_L}{T_H} = 1 - \frac{27+273}{177+273} = .33 = 33\%$

b) $\beta = \frac{Q_L}{W} = \frac{Q_L}{Q_H - Q_L} = \frac{1}{\frac{T_H}{T_L} - 1} = \frac{1}{\frac{20+273}{-10+273} - 1} = 8.76$

c) $\beta = \frac{Q_H}{W} = \frac{Q_H}{Q_H - Q_L} = \frac{1}{1 - \frac{T_L}{T_H}} = \frac{1}{1 - \frac{273}{293}} = 14.65$

4. a) $\eta = \frac{W}{Q_H} = \frac{1 \text{ mW}}{5 \text{ mW}} = .2 = 20\%$

b) $\eta = \frac{W}{Q_H} = \frac{W}{W + Q_L} = \frac{1 \text{ mW}}{1 \text{ mW} + 3 \text{ mW}} = .25 = 25\%$

c) $\eta = \frac{W}{Q_H} = \frac{Q_H - Q_L}{Q_H} = \frac{6 \text{ mW} - 4 \text{ mW}}{6 \text{ mW}} = .33 = 33\%$

d) $\beta = \frac{Q_L}{W} = \frac{4 \text{ kW}}{1 \text{ kW}} = 4$

e) $\beta = \frac{Q_L}{W} = \frac{Q_H - W}{W} = \frac{6 \text{ kW} - 1 \text{ kW}}{1 \text{ kW}} = 5$

f) $\beta = \frac{Q_L}{W} = \frac{Q_L}{Q_H - Q_L} = \frac{4.5}{6 - 4.5} = 3$

g) $\beta = \frac{Q_H}{W} = \frac{W + Q_L}{W} = \frac{1 + 4.5}{1} = 5.5$

h) $\beta = \frac{Q_H}{W} = \frac{4.5}{1} = 4.5$ | i) $\beta = \frac{Q_H}{W} = \frac{Q_H}{Q_H - Q_L} = \frac{8}{8 - 4} = 4$

$$C_v = \frac{dU}{dT}$$

$$5) T ds = PdV + dU$$

$$ds = \frac{P}{T} dV + \frac{dU}{T}$$

$$P_v = RT \rightarrow \frac{P}{T} = \frac{R}{V}$$

$$ds = R \frac{dV}{V} + C_v \frac{dT}{T}$$

$$\int_{s_1}^{s_2} ds = R \int_{V_1}^{V_2} \frac{dV}{V} + C_v \int_{T_1}^{T_2} \frac{dT}{T}$$

$$S_2 - S_1 = R \ln \frac{V_2}{V_1} + C_v \ln \frac{T_2}{T_1}$$

$$6) dU = \delta Q + \delta W$$

$\downarrow \quad \downarrow \quad \downarrow$
 $0 \quad \quad -pdV$

$$\text{and } H = U + pV \rightarrow dH = dU + p dV + V dp$$

$$\text{so, } dU = -p dV \quad \text{and} \quad dH = -p dV + p dV + V dp = V dp$$

$$\text{Always, } dU = n C_v dT \quad \text{and} \quad dH = n C_p dT$$

$$dU = n C_v dT = -p dV \quad \text{and} \quad dH = n C_p dT = V dp$$

$$C_p = \frac{V dp}{n dT} \quad \text{and} \quad C_v = \frac{-p dV}{n dT}$$

constant \downarrow

$$\gamma = \frac{C_p}{C_v} = \frac{dp/p}{dV/V} \rightarrow \gamma \int \frac{dV}{V} = - \int \frac{dp}{p} \rightarrow \gamma \ln V = - \ln p \rightarrow \ln V^\gamma = \ln \frac{1}{p}$$

$\ln V^\gamma = \ln \frac{1}{p}$

$$\rightarrow \text{so } pV^\gamma = \text{constant}$$

$$7) a) C_{v, \text{air}} = .718 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

$$S_2 - S_1 = C_v \ln \frac{T_2}{T_1} + R \ln \frac{v_2}{v_1}$$

$$S_2 - S_1 = .718 \cdot \ln \frac{350}{300} = \boxed{.11068 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}}$$

$$b) S_2 - S_1 = .718 \ln \frac{T_2}{T_1} = .718 \ln \frac{350}{300} = .02706 \frac{\text{kJ}}{\text{kg} \cdot \text{K}}$$

