

# Journey to Spaceport

Charlie Nitschelm

# My Background

- Senior mechanical engineering major and physics minor at the University of New Hampshire
- Founded UNH SEDS at the end of my freshman year and am the current President and Lead Engineer
- Elected to the board of SEDS USA, the largest studentrun space organization, as the Member at Large

### **Job Experiences**

Summer 2018

Researcher at NIST on the mechanical behavior of Inconel in extreme conditions

2018-2019 School Year

Manufacturing engineering intern at TURBOCAM international focused on the affect tool coatings have on the lifetime of a tool

Summer 2019

Propulsion manufacturing intern at Rocket Lab USA focused on process improvement within printing, machining and welding







### The 3-Year Plan





# Year 1 – Sophomore Year

### **Organization Statistics**

Committed Members: 14

Majors: 3

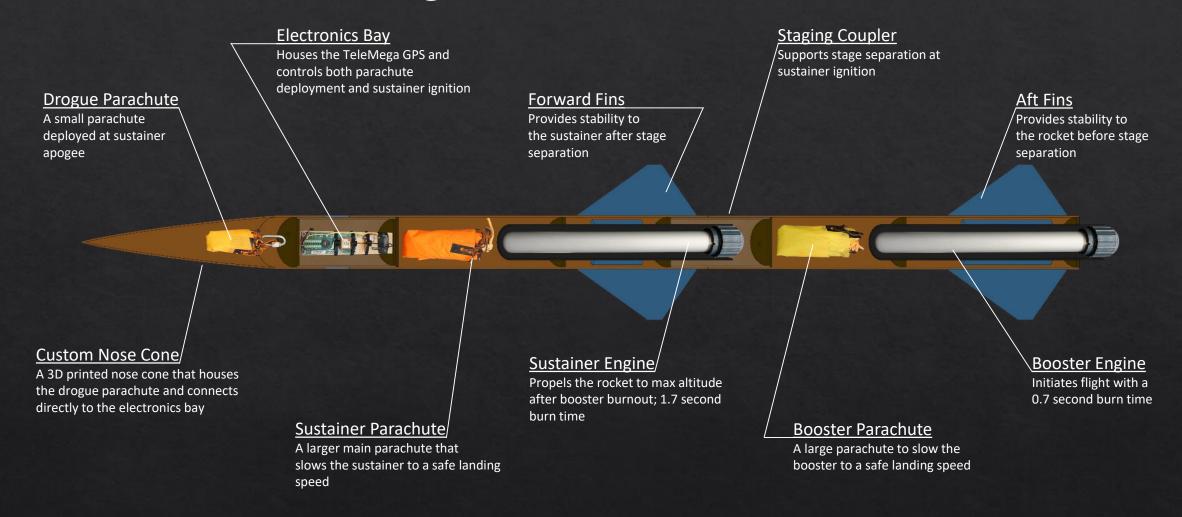
Senior Projects: 4

Funding: \$2,250



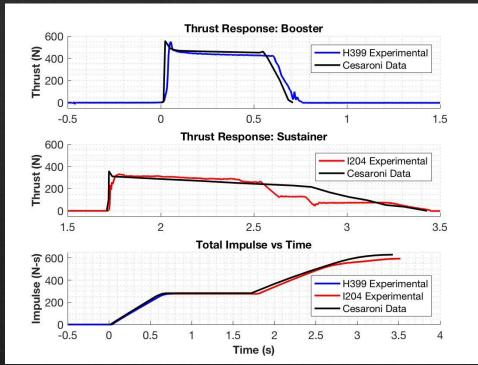
SEDS Rocketry Competition, May 2018, Maine

# Year 1 – Rocket Building



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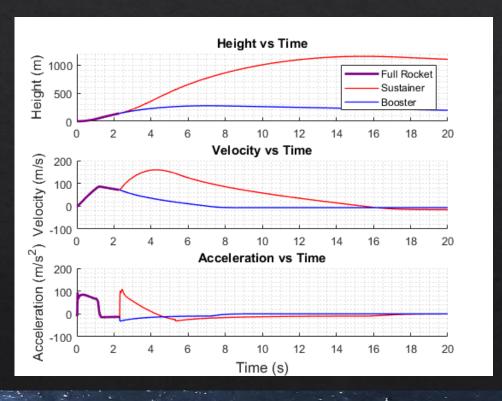
# Year 1 – Simulating





	Reported Max Thrust	Measured Max Thrust	Reported Total Impulse	Measured Total Impulse
Booster Engine: Cesaroni H399	545.8 N	549.6 N	282.2 N-s	277.1 N-s
Sustainer Engine: Cesaroni I204	356.8 N	329.7 N	347.7 N-s	322.7 N-s

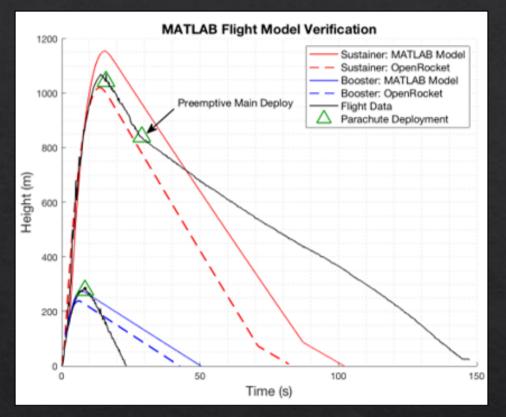




# Year 1 – Launching

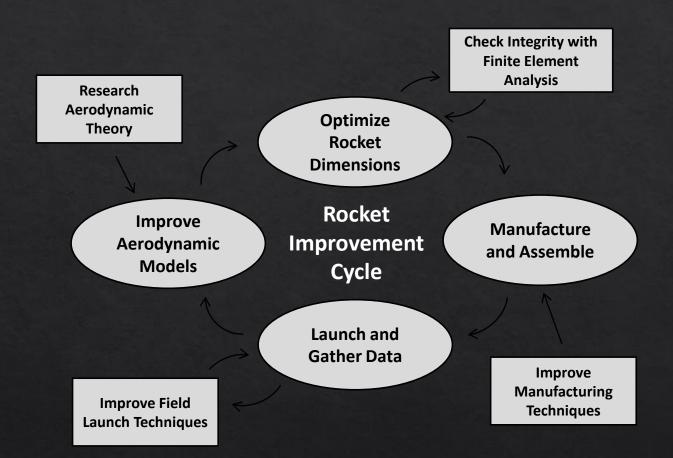


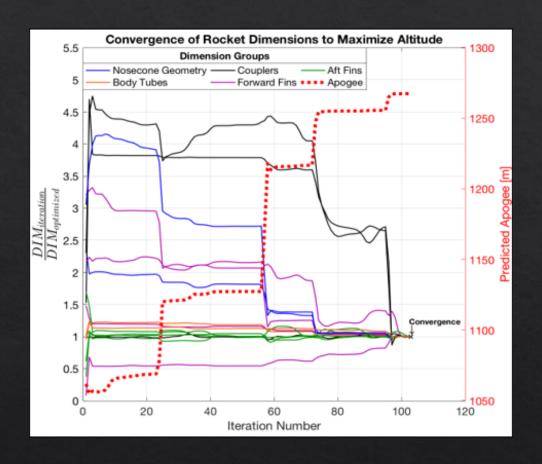




	Flight Data	OpenRocket Model	MATLAB Model
Sustainer Apogee	1071.1 m	1020.6 m	1154.1 m
Booster Apogee	290.0 m	238.6 m	276.3 m

# Year 1 – Optimizing and Repeat







### Year 2 – Junior Year

### **Organization Statistics**

Committed Members: 20

Majors: 5

Senior Projects: 9

Funding: \$5,500



Spacevision 2018, November 2018, San Diego

# Year 2 – Hybrid Engine Development

#### Oxidizer Tank

A highly pressurized vessel that contains liquid Nitrous Oxide, acting as our oxidizer within the combustion chamber

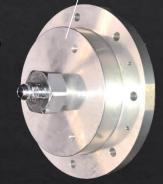
#### Injection Plate

Responsible for providing desired oxidizer flow into the combustion chamber with the assistance of the impinging plate

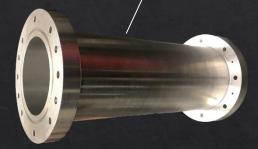
#### Combustion Chamber

An enclosed volume where the solid reducer and liquid oxidizer react to produce a superheated, highly pressurized chamber of gas











### \Flow Regulator

A motorized valve that monitors flow regulation of the oxidizer into the injection plate controlled by an electric motor and an Arduino.



An interchangeable cylindrical plate responsible for the impingement and atomization of the oxidizer flow streams

### \<u>Nozzle</u>

Graphite was machined into a de Laval curve responsible for directing the flow of hot gases outside of the combustion chamber into the environment providing thrust

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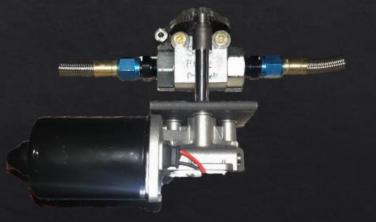
# Year 2 – Engine Design

### Oxidizer Selection and Flow Regulation

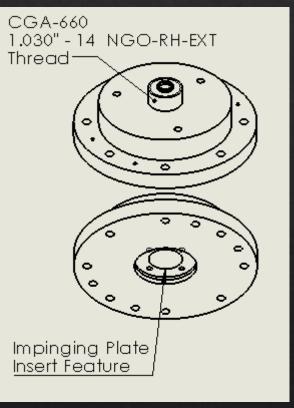


- Needs to be self pressurizing
- Needs to be safe to handle
- Can be refilled within a day
- Nitrous Oxide meets these requirements

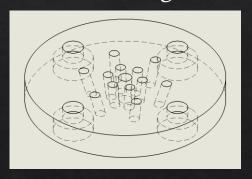
- Suitable for ground testing
- Ability to throttle
- Cheap



### **Injector and Impinging Plate**

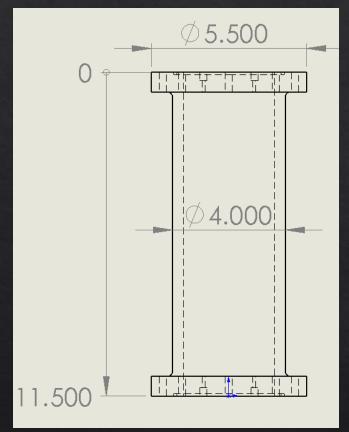


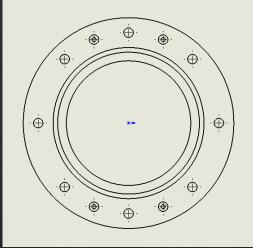
- Interface with COTS piping from flow regulator
- Quick testing of different impinging geometries
- Transitions the liquid oxidizer to a gas



# Year 2 – Engine Design

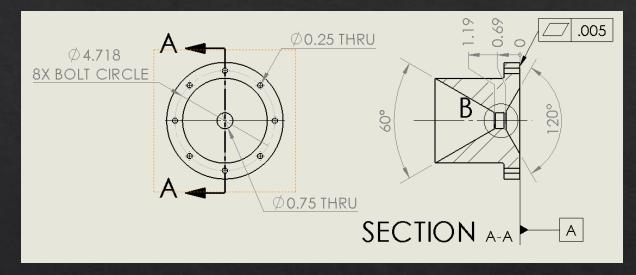
### **Combustion Chamber**





- Withstand 400 psi internal pressure
- Hold a volume of fuel to support the defined mixture ratio
- Seals with the nozzle and injector

#### Nozzle



- Accelerate the hot gases within the combustion chamber using converging/diverging design
- Withstand a corrosive, high temperature environment



## Year 2 – Manufacture



Combustion chamber chucked on the lathe



Machining and assembly of impinging plate



Assembly of the flow regulator



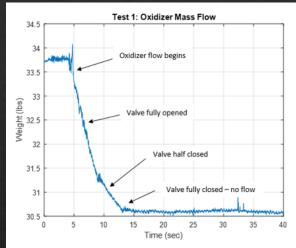
Molding the rubber into the combustion chamber



Runaway fully assembled with the aluminum chamber, injector and graphite nozzle

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## Year 2 – Test

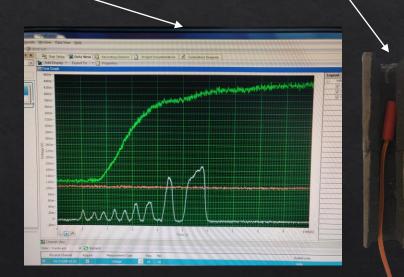




Rapid impinging plate designs tested for impingement and desired flow rate

Obtain thrust and temperature data during the hot-fire test

Electric spark and igniter assembly





Control bunker for hot-fire tests



Hot-fire test 1 – Ignition

## Year 3 – Senior Year

### **Organization Statistics**

Committed Members: 42

Majors: 7

Senior Projects: 14

Funding: \$9,000 and counting



# Year 3 – Spaceport Competition

- Rocket competition hosted in New Mexico from June 16<sup>th</sup> to June 20<sup>th</sup>
- Requires teams to launch a rocket using COTS or experimental rocket engines to as close to 10,000/30,000 feet with full recovery
- UNH SEDS will be competing in the 10,000-foot experimental hybrid engine category





### Year 3 – Rocket Vehicle

#### **Avionics**

An electronics module responsible for GPS tracking, engine ignition and flow control, real-time data collection, and recovery triggers

#### Research Payload

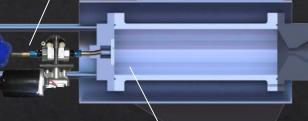
A payload built in-house to collect data throughout flight for use in the future

#### Throttle Control

A flow regulation system to insert oxidizer flow into the combustion chamber to provide thrust to the rocket







#### Recovery System

A recovery module that is deployed at apogee to land the rocket safely back on Earth

#### Oxidizer Storage

A pressurized vessel holding the liquid oxidizer to be pumped into the combustion chamber

#### \Runaway

New Hampshire's first hybrid rocket engine that utilizes a liquid oxidizer and solid fuel to propel its rocket to 10,000 feet

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# Year 3 – Engine Optimization

Currently the propulsion lead working on optimizing Runaway for integration into the rocket

• Team of 9 students in propulsion



# Year 3 – Project Management

Chicken suit



