Linear differential equations
Laplace transform
Solving LDEs with the Laplace transform
Properties of state-space representation
Transfer functions
ransient response analysis of first order and second order systen

Chapter 5: Continuous time systems

July 17, 2015

- Linear differential equations
- 2 Laplace transform
- Solving LDEs with the Laplace transform
- 4 Properties of state-space representation
- Transfer functions
 - Impulse response and time constant
 - Relationship between state space and transfer functions
- Transient response analysis of first order and second order systems
 - First order systems
 - Second order systems



Outline

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$$L[y(t)] = f(t),$$

where L is some linear operator.

Properties of state-space representation

Linear differential equations: definitions 1/2

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The **order of a LDE** is the index of the highest derivative of y.



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- if $A_{0:n}(t)$ are constants (ie. not functions of time), the LDE is said to have constant coefficients

Example: radioactive decay 1/2

Let N(t) be the number of radioactive atoms at time t, then:

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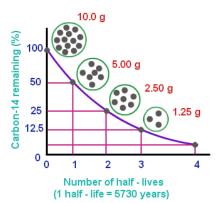
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for some constant k > 0.

This is a first order homogeneous LDE with constant coefficients.

Example: radioactive decay 2/2

Decay of Carbon - 14



Transient response analysis of first order and second order systems

Solving homogeneous LDEs with constant coefficients 1/3

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Dividing by e^{zt} yields the *n*th order **characteristic polynomial**:

$$F(z) = \sum_{i=0}^{n} A_i z^{n-i} = 0.$$

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The specific linear combination depends on initial conditions.

Example:

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These zeros correspond to the following basis functions t:

$$e^{jt}$$
, e^{-jt} , e^t , te^t .

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The final value theorem states $f(\infty) = \lim_{s\to 0} sF(s)$, if all poles of sF(s) are in the left half plane (ie. real part < 0).

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with $u(t) = \int_0^t \delta(t) dt$ (Heaviside) and $\delta(t)$ the Dirac delta $\frac{1}{2}$

Inverse Laplace transform

The inverse Laplace transform converts *s*-domain to time domain:

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Practically, the inverse Laplace transform takes two steps:

- \bullet write F(s) in terms of partial fractions
- 2 transform each term in the partial fraction based on tables of s/t-domain pairs (course notes p 4.32-4.33)

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Solving LDEs with the Laplace transform 1/3

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Via induction, the Laplace transform of the *n*th order derivative:

$$\mathcal{L}\{f^{(n)}\} = s^n F(s) - \sum_{i=1}^n s^{n-i} f^{(n-i)}(0)$$

Solving LDEs with the Laplace transform 2/3

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Via the linearity of the Laplace transform:

$$\sum_{i=0}^{n} A_i \mathcal{L}\{y^{(n-i)}(t)\} = \mathcal{L}\{f(t)\}$$

Solving LDEs with the Laplace transform 3/3

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Expanding Eq. (2) into (1) yields:

$$Y(s)\sum_{i=0}^{n}A_{i}s^{i}-\sum_{i=1}^{n}\sum_{j=1}^{i}A_{i}s^{i-j}y^{j-1}(0)=F(s)$$

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The solution in the time domain is obtained via the inverse Laplace transform: $y(t) = \mathcal{L}^{-1}\{Y(s)\}.$

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This holds for linear, time-invariant systems with n states if:

$$rank(\mathcal{O}) = n, \quad \mathcal{O} = \begin{bmatrix} \mathbf{C} \\ \mathbf{CA} \\ \vdots \\ \mathbf{CA}^{n-1} \end{bmatrix}, \quad \mathcal{O} : \mathbf{observability \ matrix}$$

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A linear, time-invariant system with n states is controllable if:

$$rank(C) = n, \quad C = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \dots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix},$$

where C is called the **controllability matrix**.



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Transfer function

The transfer function of input i to output j is defined as:

$$H_{i,j}(s) = \frac{Y_j(s)}{U_i(s)}, \quad \mathbf{U}(s) = \mathcal{L}\{u(t)\}, \quad \mathbf{Y}(s) = \mathcal{L}\{y(t)\}.$$

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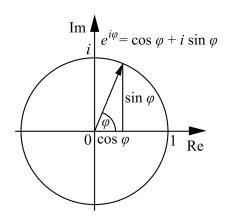
The complex Laplace variable can be rewritten: $s = \sigma + j\omega$.

The frequency response of a system can be analyzed via $\mathbf{H}(j\omega)$:

$$e^{\sigma+j\omega}=e^{\sigma}(\cos\omega+j\sin\omega).$$



Illustration of Euler's formula



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Poles and zeros may cancel, ie. if D(s) = N(s) = 0 for some s.

Steady state response

The output of a linear time-invariant system yields consists of:

- a steady-state output $y_{ss}(t)$, which similar periodicity to u(t)
 - $\rightarrow v_{ss}$ comprises the same frequencies as u(t)
- a transient output $y_{tr}(t)$
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The steady-state output $y_{ss}(t)$ of a linear time invariant system:

- ullet consists of signals of same frequencies as the input signal u(t)
- which may have been magnified and/or phase changed

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, because $\mathbf{D} = 0$ and $\lim_{t\to\infty} \mathbf{x}(t) = 0$.

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The speed of convergence depends on the position of the poles.

Time constant

The transfer function of first order systems can be written as:

$$H(s) = \frac{K}{\tau s + 1} \quad \leftrightarrow \quad h(t) = \frac{K}{\tau} e^{-t/\tau},$$

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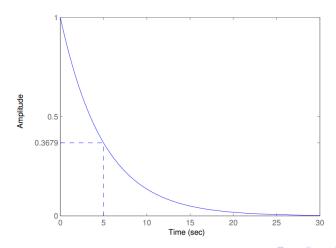
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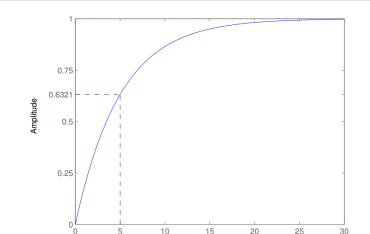
where τ is called the system's **time constant**.

The time constant summarizes the speed of a system's dynamics:

- after τ seconds, the impulse response reaches h(0)/e.
- after au seconds, the step response has reached $1-e^{-1} \approx 63\%$ of its regime value.

Impulse response $H(s) = 5/(5s+1) \leftrightarrow h(t) = exp(-t/5)$





Time (sec)

Outline

- 1 Linear differential equations
- 2 Laplace transform
- 3 Solving LDEs with the Laplace transform
- Properties of state-space representation
- **5** Transfer functions
 - Impulse response and time constant
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- 6 Transient response analysis of first order and second order systems
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From state-space to transfer functions

We start from the linear state-space representation:

time domain

$$\begin{cases} \dot{\mathbf{x}}(t) = \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) \\ \mathbf{y}(t) = \mathbf{C}\mathbf{x}(t) + \mathbf{D}\mathbf{u}(t) \end{cases} \leftrightarrow \begin{cases} s\mathbf{X}(s) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s) \\ \mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) + \mathbf{D}\mathbf{U}(s) \end{cases}$$

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$$(sI - A)X(s) = BU(s)$$

 $X(s) = (sI - A)^{-1}BU(s)$
 $\Rightarrow Y(s) = C(sI - A)^{-1}BU(s) + DU(s)$
 $\Rightarrow H(s) = C(sI - A)^{-1}B + D$

Relationship between poles and eigenvalues of A 1/2

Poles are zeros of the denominator of $\mathbf{H}(s)$, e.g. those values of s for which $\mathbf{H}(s)$ is singular.

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Consider the following SISO system with 2 states:

$$\begin{bmatrix}
sX_1(s) \\
sX_2(s)
\end{bmatrix} = \begin{bmatrix}
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X_2(s)
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2
\end{bmatrix} U(s)$$

$$Y(s) = \begin{bmatrix}
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The transfer function $H(s) = \frac{\beta}{s-\alpha}$ has only one pole $(s_1 = \alpha)$. \rightarrow not all eigenvalues of **A** are poles in transfer functions $\mathbf{H}(s)$.

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Transient Response

The time response of a control system may be written as:

$$y(t) = y_{tr}(t) + y_{ss}(t)$$

Where $y_{tr}(t)$ is the transient response and $y_{ss}(t)$ is the steady state response. Most important characteristic of dynamic system is absolute stability.

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Transient response analysis of first order and second order systems

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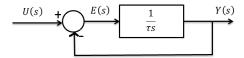
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Transient response: when input of system changes, output does not change immediately but takes time to go to steady state.



First order systems

E.g.RC circuit, thermal system, \dots



First order systems

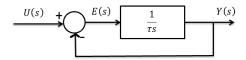
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First order systems

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Unit step response

- Laplace of unit-step is $\frac{1}{s} \to \text{substituting } U(s) = \frac{1}{s}$ into equation $Y(s) = \frac{1}{s} \frac{1}{\tau s + 1}$
- Expanding into partial fractions gives

$$Y(s) = \frac{1}{s} - \frac{\tau}{\tau s + 1} = \frac{1}{s} - \frac{1}{s + \frac{1}{s}}$$



Unit step transient response

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$$y(t) = 1 - e^{-\frac{t}{\tau}}$$
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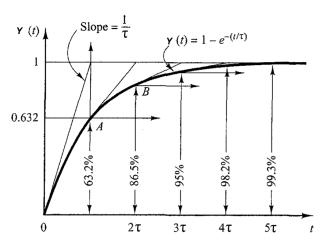
Slope at time t = 0 is $\frac{1}{\tau}$

$$\frac{dy}{dt}|_{t=0} = \frac{1}{\tau}e^{-\frac{t}{\tau}}|_{t=0} = \frac{1}{\tau}$$

Where τ is called the system time constant



Unit step transient response



Unit ramp transient response

① Laplace transform of unit ramp is $\frac{1}{s^2}$

$$Y(s) = \frac{1}{\tau s + 1} \frac{1}{s^2}$$

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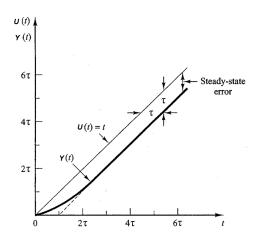
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5 For t approaching infinity, e(t) approaches au

$$e(\infty) = \tau$$



Unit ramp transient response



Unit-Impulse Response

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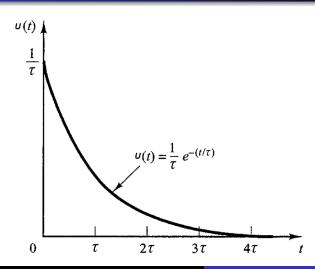
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For
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Unit-Impulse Response



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Second order systems

A second order system can generally be written as:

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A system where the closed-loop transfer function possesses two poles is called a second-order system

If the transfer function has two real poles, the frequency response can be found by combining the effects of both poles



Second order systems

Sometimes the transfer function has two complex conjugate poles. In that case we have to find a different solution for finding the frequency response.

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In order to study the transient behaviour, let us first consider the following simplified example of a second order system

$$H(s) = \frac{c}{ds^2 + es + c}$$

Step response second order system

2 The transfer function can be rewritten as:

$$H(s) = \frac{\frac{c}{d}}{s^2 + \frac{e}{d}s + \frac{c}{d}}$$

$$= \frac{\frac{c}{d}}{\left[s + \frac{e}{2d} + \sqrt{\left(\frac{e}{2d}\right)^2 - \frac{c}{d}}\right]\left[s + \frac{e}{2d} - \sqrt{\left(\frac{e}{2d}\right)^2 - \frac{c}{d}}\right]}$$

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3 The poles are complex conjugates if

$$e^2 - 4dc < 0$$

The poles are real if

$$e^2-4dc\geq 0$$

To simplify the transient analysis, it is convenient to write

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Where

 σ is the attenuation

 ω_n is the natural frequency

 ζ is the damping ratio



Step response second order system

The transfer function can now be rewritten as

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The dynamic behavior of the second-order system can then be described in terms of only two parameters ζ and ω_n

If $0<\zeta<1$, the poles are complex conjugates and lie in the left-half s-plane

- The system is then called underdamped
- The transient response is oscillatory

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If $\zeta > 1$, the system is called **overdamped**

We will now look at the unit step response for each of these cases



Underdamped system

For the underdamped case (0 < ζ < 1), the transfer function can be written as

$$H(s) = \frac{\omega_n^2}{(s + \zeta \omega_n + j\omega_d)(s + \zeta \omega_n - j\omega_d)}$$

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For a unit-step input we can write

$$Y(s) = \frac{\omega_n^2}{(s^2 + 2\zeta\omega_n s + \omega_n^2)s}$$

Which can be rewritten as partial fractions

$$Y(s) = \frac{1}{s} - \frac{s + 2\zeta\omega_n}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$
$$= \frac{1}{s} - \frac{s + \zeta\omega_n}{s^2 + 2\zeta\omega_n s + \omega_n^2} - \frac{\zeta\omega_n}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

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It can be shown that

$$\mathcal{L}^{-1} \Big[rac{s + \zeta \omega_n}{(s + \zeta \omega_n)^2 + \omega_d^2} \Big] = e^{-\zeta \omega_n t} cos(\omega_d t)$$
 $\mathcal{L}^{-1} \Big[rac{\omega_d}{(s + \zeta \omega_n)^2 + \omega_d^2} \Big] = e^{-\zeta \omega_n t} sin(\omega_d t)$

Underdamped system

Therefore:

$$\mathcal{L}^{-1}\Big[Y(s)\Big]=y(t)$$

Therefore:

$$\begin{split} &\mathcal{L}^{-1}\Big[Y(s)\Big] = y(t) \\ &= 1 - e^{-\zeta\omega_n t} (\cos(\omega_d t) + \frac{\zeta}{\sqrt{1-\zeta^2}} \sin(\omega_d t)) \\ &= 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \sin(\omega_d t + \tan^{-1}(\frac{\sqrt{1-\zeta^2}}{\zeta})) \end{split}$$

Therefore:

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It can be seen that the frequency of the transient oscillation is the damped natural frequency ω_d and thus varies with the damping ratio ζ

The error signal is the difference between input and output

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The error signal exhibits a damped sinusoidal oscillation

At steady state, or at $t = \infty$, the error goes to zero

Underdamped system

Underdamped system

If damping $\zeta = 0$, the response becomes undamped

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- ω_d is always lower than ω_n



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The inverse Laplace transform gives us

$$y(t) = 1 - e^{-\omega_n t} (1 + \omega_n t)$$
 for $t \ge 0$



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Thus y(t) includes two decaying exponential terms

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- In that case, H(s) can be approximated by



With the approximate transfer function, the unit-step response becomes

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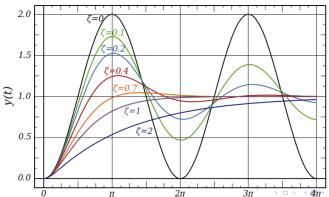
The time response for the approximate transfer function is then given as

$$y(t) = 1 - e^{-(\zeta - \sqrt{\zeta^2 - 1})\omega_n t}$$
, for $t \le 0$



Second order systems unit step response curves

Response on a step function



Second order systems - characteristics

Overshoot: Highest amplitude above steady state.

$$M_p = e^{\frac{-\pi\zeta}{\sqrt{1-\zeta^2}}}$$

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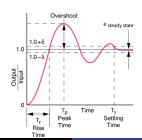
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time. $t_r = \frac{1.8}{\omega_n}$

Peak Time: Time to reach overshoot.

$$t_p = \frac{\pi}{\omega_d}$$



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Settling Time: Time needed to approximate the steady state.

$$t_s = \frac{4.6}{\zeta \omega_n}$$

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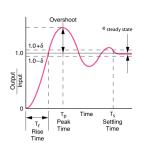
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We find:



$$e^{-\zeta\omega_n T_s} < 0.02$$

$$T_s = \frac{4}{\omega_n \zeta}$$



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A different view on the Tacoma bridge disaster: https://www.youtube.com/watch?v=6ai2QFxStxo

In fact the collapse was a result of a number of effects like Aerodynamic flutter and vortices. Read the full article here: http://www.ketchum.org/billah/Billah-Scanlan.pdf

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Another phenomenon with bridges and resonance is that many people marching with the same rhythm can cause a bridge to start resonating like the Angers bridge in 1850. A more recent example is the Millennium bridge in London who started resonating.

Second order systems - damping

When we want a system with no resonance, we choose one with damping < 0.707. This means a pole between 135° and 225° :

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We mostly want a short settling time (< 4s). This results in another restriction on the poles of the system:

$$au_{\it n}=rac{4}{\omega\zeta}<4$$
s $\omega_{\it n}\zeta>1$

Second order systems - damping

