

Simplified 2D Problem: Dynamic Programming

Vehicle is well controlled by arbitrary control surface (CS) inputs
 No limit on control surface or vehicle α
 Constant thrust, don't consider aero forces

State variables (discretized)

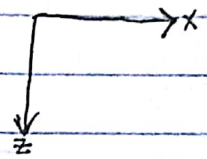
$$x = x_1^k = x \text{ coord (abs) @ pseudo time } k, \text{ meters}$$

$$z = x_2^k = z \text{ coord (abs) @ pseudo time } k, \text{ meters}$$

$$\dot{x} = x_3^k = x \text{ vel (abs) @ pseudo time } k, \text{ m/s}$$

$$\dot{z} = x_4^k = z \text{ vel (abs) @ pseudo time } k, \text{ m/s}$$

$$t = x_5^k = \text{true time corresponding to pseudo time } k$$



Controls

$$u^k = \theta^k = \text{missile pitch Euler angle @ pseudo time } k, \text{ deg}$$

$$-90 \leq u \leq 90$$

Dynamics

$$x(k+1) = \begin{bmatrix} x_1(k) + x_3(k) \frac{x_5(N)}{N-1} \\ x_2(k) + x_4(k) \frac{x_5(N)}{N-1} \\ x_3(k) + \frac{m}{T} \cos \theta(k) \frac{x_5(N)}{N-1} \\ x_4(k) + \left(g - \frac{m}{T} \sin \theta(k) \right) \frac{x_5(N)}{N-1} \\ x_5(k) + \frac{x_5(N)}{N-1} \end{bmatrix} \quad k=0, 1, 2, \dots, N$$

$$\frac{x_5(N)}{N-1} = \frac{t_f - t_0}{N-1} = \Delta t$$

Discretize

$$x_{1,\min} \leq x_1(k) \leq x_{1,\max}, \quad k=0, 1, \dots, N \quad \text{on } m_1 \text{ points}$$

$$x_{5,\min} \leq x_5(k) \leq x_{5,\max}, \quad k=0, 1, \dots, N \quad \text{on } m_5 \text{ points}$$

$$\theta_{\min} \leq \theta(k) \leq \theta_{\max}, \quad k=0, 1, \dots, N-1 \quad \text{on } m_\theta \text{ points}$$