#### **SDM366 Optimal Control and Estimation**

# Lecture Note 2 State Space Model and Stability

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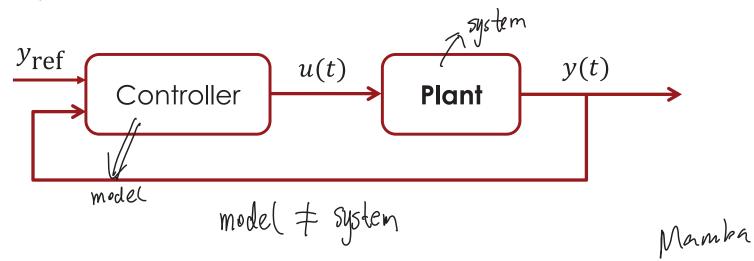
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### **Outline**

- State space model: definition and examples
- From continuous-time to discrete time model
- From nonlinear to linear model
- System solution and stability

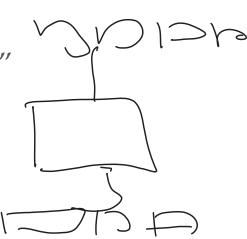
# State-space model based feedback control system:

Goal: determine control input to achieve desired output



- Controller design is based on plant model
  - Model is different from the actual plant
  - "all models are wrong, but some are useful"
- Modeling approach:
   First principle

  - Data driven (System ID) ←



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# • Static vs. Dynamic Systems

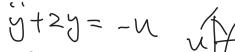
Static system

$$y = \phi(u)$$

$$y(t)$$

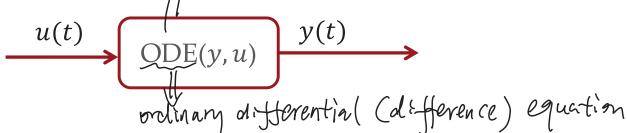
$$y = \psi(u)$$

- u(t) completely and immediately determines y(t)
- Desired output  $y_{ref}$  can be perfectly tracked (in absence of disturbance) by open-loop plant inversion



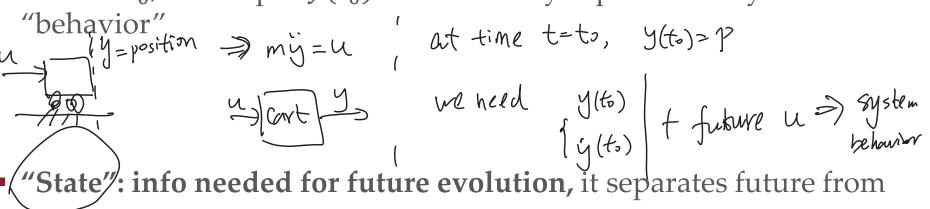


Dynamic system: differential/or difference equation



• u(t) does not fully determines y(t)

• At time  $t_0$ , the output  $y(t_0)$  does not fully captures the system



past

• State  $x(t_0)$  at time  $t_0$  and input u(t) over time  $[t_o, t_f]$ , completely **determines** the system behaviors

$$X \in \mathbb{R}^{N}$$
,  $X = \begin{bmatrix} x^{N} \\ \vdots \\ x^{N} \end{bmatrix}$ 

• General continuous-time state space model

All 'finite - din' dynamic system 
$$(\dot{x}) = f(x, u)$$
 ) Ist order differential equ in  $IR^n$  can be written in this "state-syace" form  $y = h(x, u)$ 

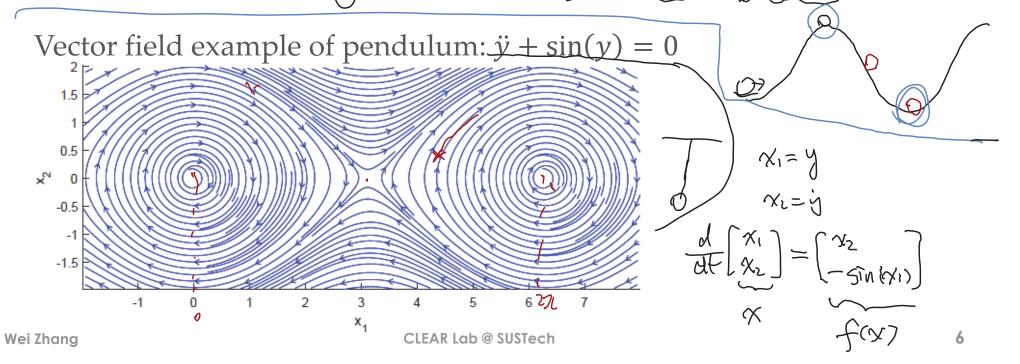
•  $x \in \mathbb{R}^n$  state vector,  $u \in \mathbb{R}^m$  control input,  $y \in \mathbb{R}^p$  output,

•  $f: \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^n$ : called vector field specify velocity of state

•  $h: \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^p$ : output function

• Called autonomous system if there is no control f(x, u) = f(x)

• For autonomous sys,  $\hat{x} \in \mathbb{R}^n$  is called **equilibrium** if  $f(\hat{x}) = 0$ 



General discrete-time state space model

me state space model
$$\chi(k+1) = f(\chi(k), \mu(k))$$

$$\chi(k) = h(\chi(k), \mu(k))$$

$$\chi(k) = h(\chi(k), \mu(k))$$

$$\chi(k) = h(\chi(k), \mu(k))$$

- $x \in \mathbb{R}^n$  state vector,  $u \in \mathbb{R}^m$  control input,  $y \in \mathbb{R}^p$  output
- $f: \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^n$ : state update equation
- $h: \mathbb{R}^n \times \mathbb{R}^m \to \mathbb{R}^p$ : output function
- Called autonomous system if there is no control f(x, u) = f(x)
- For autonomous sys,  $\hat{x} \in \mathbb{R}^n$  is called **equilibrium** if  $\hat{x} = f(\hat{x})$
- Discrete-time system:
  - Some discrete-time system is obtained from continuous time model by sampling
  - Some systems naturally evolve in discrete time.

# • Linear Systems: system is called linear if:

Continuous time 
$$\dot{x} = f(x, u) = Ax + Bu$$
,  $y = h(x, u) = Cx + Du$ ,

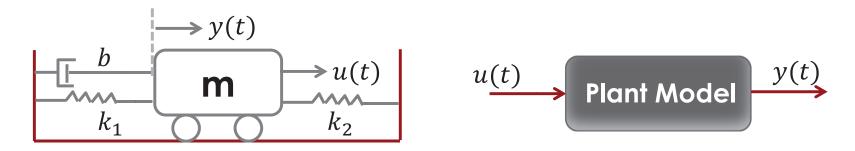
Discrete time 
$$x(k+1) = f(x(k), u(k)) = Ax(k) + Bu(k),$$
 
$$y(k) = h(x(k), u(k)) = Cx(k) + Du(k),$$

for some matrices A, B, C, D

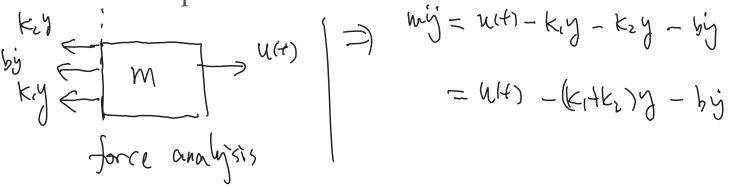
# State-space modeling:

- Find the functions  $f(\cdot,\cdot)$ ,  $h(\cdot,\cdot)$
- Or find *A*, *B*, *C*, *D* matrices if the system is linear

**Example 1**: Consider spring-damper cart system with zero initial conditions (initially at y = 0 and not moving). No friction



Differential equation model



State space model of Example 1 (infinitely many)

Let 's define 
$$x_1 = y$$
,  $x_2 = \dot{y} \Rightarrow x = \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \in \mathbb{R}^2 \iff \dot{x} = f(x, u)$ 

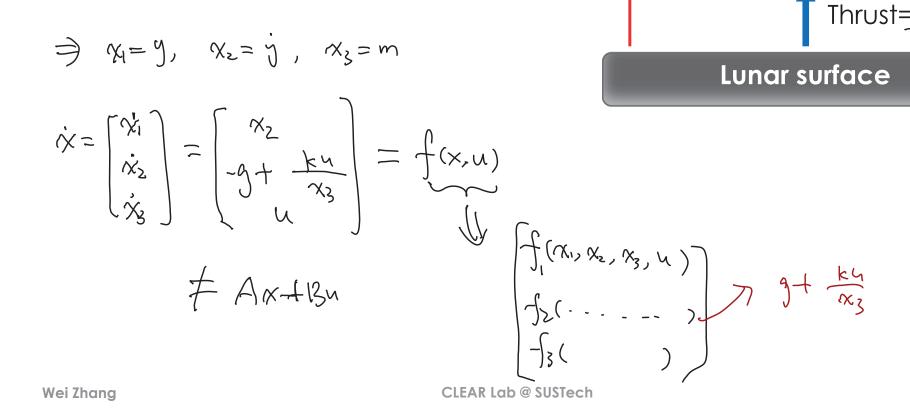
$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} \dot{x}_2 \\ \frac{1}{m}(u - bx_2 - (k_1 + k_2)x_1) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{k_1 + k_2}{m} & -\frac{b_2}{m} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u$$

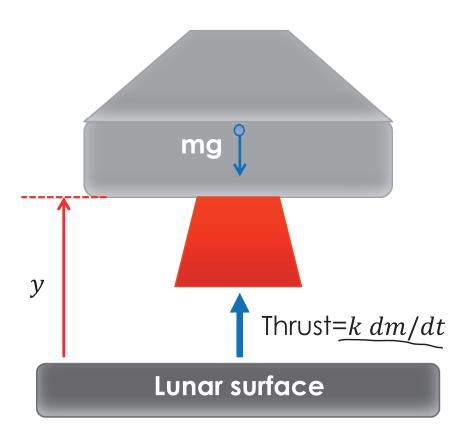
$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} \dot{x}_2 \\ \frac{1}{m} \end{bmatrix} \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{1}{m} \end{bmatrix} u$$

$$\int_{C} = [x_1] + x_2 + x_3$$

**Example 2**: soft landing of a lunar

$$\dot{x} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} x_2 \\ -3 + \frac{ku}{x_3} \end{bmatrix} = f(x, u)$$





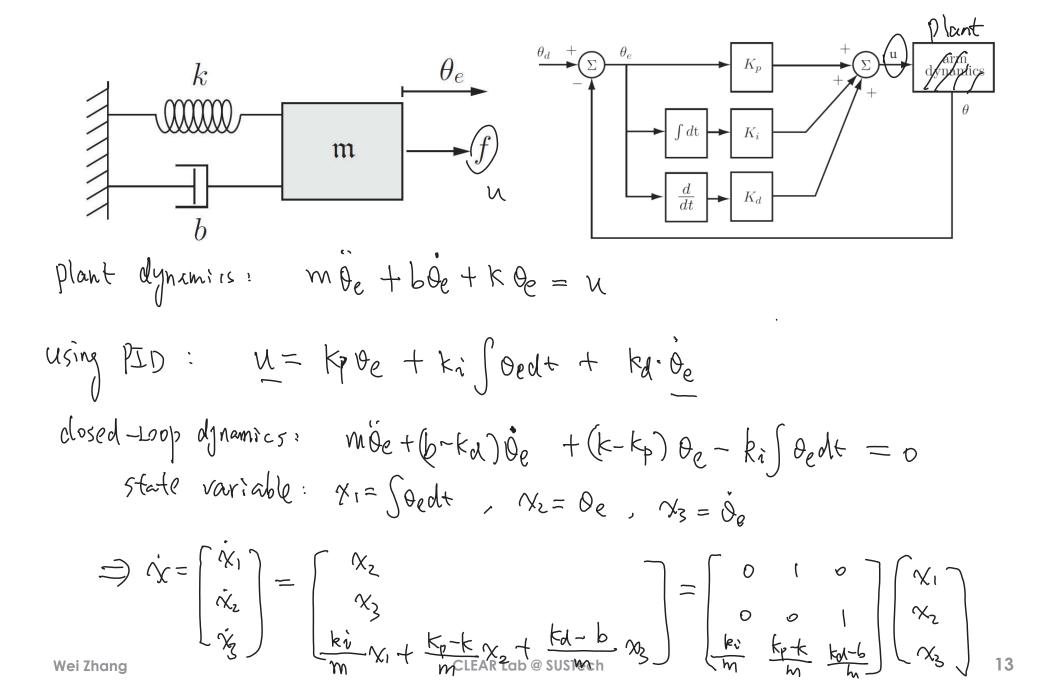
# **Example 3**: Sensor Network

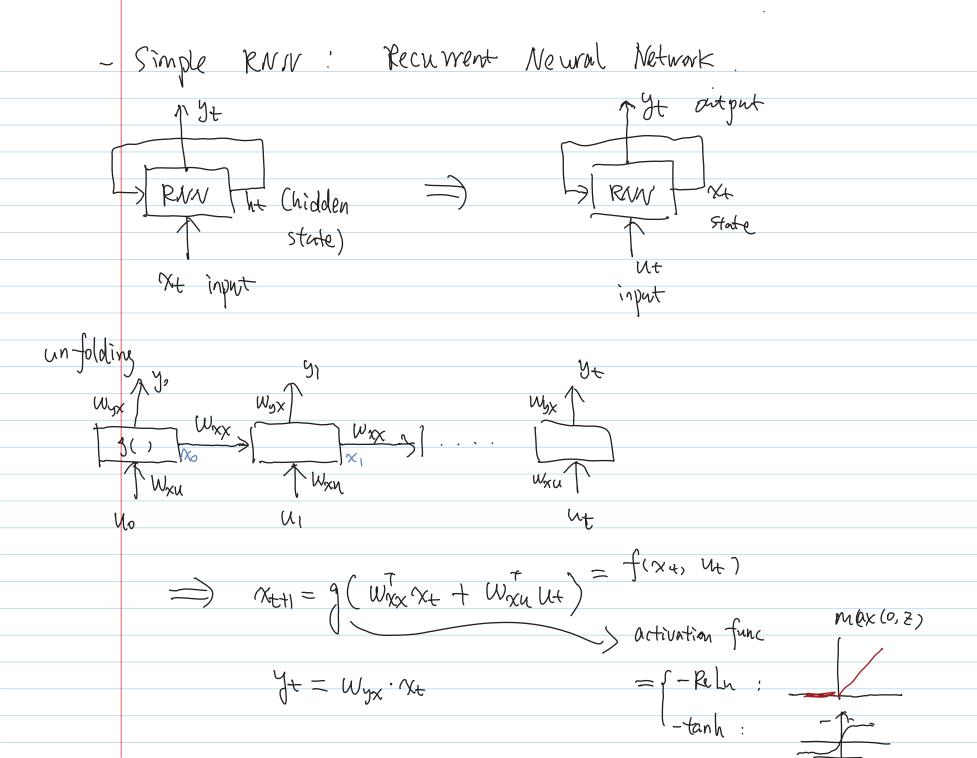
- Each iteration, exchange measurements with neighbors
- The updated value is the average of its own value with the neighbors

$$(x_{1}(k)) \qquad (x_{2}(k)) \qquad (x_{3}(k)) \qquad (x_{3}(k)) \qquad (x_{1}(k)) \qquad (x_{1}(k)) \qquad (x_{1}(k)) \qquad (x_{1}(k)) \qquad (x_{1}(k)) \qquad (x_{1}(k)) \qquad (x_{2}(k)) \qquad (x_{2}(k)) \qquad (x_{2}(k)) \qquad (x_{3}(k)) \qquad (x_{3}(k)) \qquad (x_{4}(k)) \qquad (x_$$

# closed-Loop

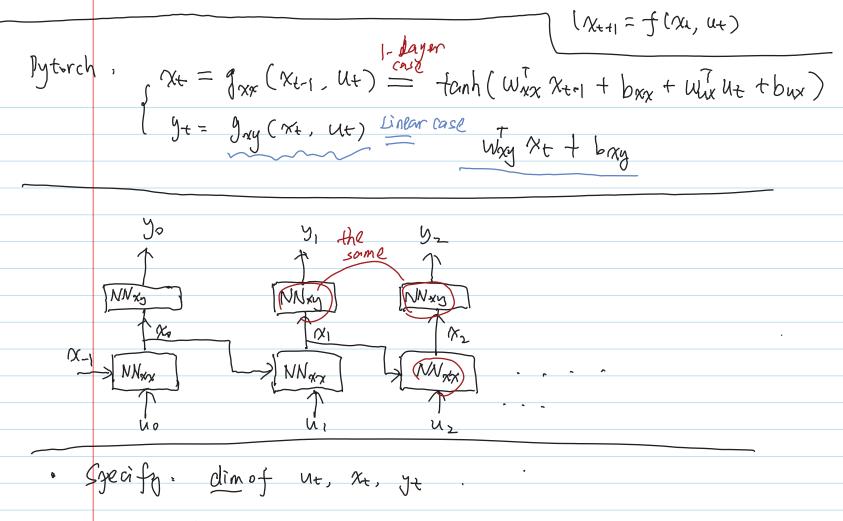
# **Example 4**: PID for spring-damper system





. Xi RNN is a special case of state space model This lecture · Mytorch implementation of a simple RNN & Simulation of RNN · State space model implementation of RNN in Python) NN(w) function mapping from UEIR to y FIR output-dim u1,..., u | RWN (W) | 1,..., 7~ memps from sequence of ight vectors to ... output ---· Assumption: () (ausality: Yk = g(u,, ..., uk)

Finite-dim representation: I state  $x_{t} \in \mathbb{R}^{n_{x}}$ such that  $y_{t} = y(x_{t}, u_{t})$ 



NNXX, NNXY

K= (X0 , X1 , X2 , ...)

### **Outline**

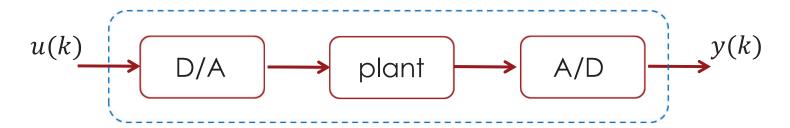
State space model: definition and examples

From continuous-time to discrete time model

From nonlinear to linear model

System solution and stability

#### From continuous time to discrete time model

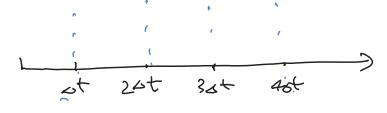


- Approximate differential equation with difference equation
  - Euler forward rule:

### From continuous-time to discrete-time model

• General nonlinear case:

$$\dot{x} = f(x, u)$$
$$y = h(x, u)$$



$$X(k+1) = X((k+1)st) = X(kst) + (X(kst)) at$$

$$= (x(k) + f(x(k), u(k)).st)$$

$$= f_d(x(k), u(k))$$

$$y(k) = h(x(k), u(k))$$

$$= h_{d}(...)$$

$$y(k) = f_{d}(x(k), u(k))$$

$$y(k) = h_{d}(x(k), u(k))$$

$$\int X[k+1] = \int d(X[k], u[k])$$

# From continuous-time to discrete-time model

Linear case:

$$\dot{x} = \underbrace{A_c x + B_c u}_{C_c x + D_c u},$$

discretization

 $[\frac{1}{2}] + [\frac{1}{2}]$ 

NE[R"

UCIRM

using previous result:

ohscrete

time

$$\frac{\chi(k+1) = \chi(k) + (A_c \chi Ck) + B_c u k)}{\chi(k) + (B_c \chi Ck)} \cdot \chi(k) + (B_c \chi Ck) \cdot \chi(k) + (B_c \chi Ck) \cdot \chi(k)$$

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=) DT sys with at sampling interval

$$f(k) = Adx(k) + Bdu(k)$$

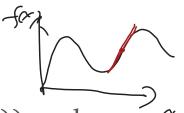
$$f(k) = Cdx(k) + Ddu(k)$$

$$Cd = Cc, Pd = Pc$$

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- State space model: definition and examples
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- Given model: x(k+1) = f(x(k), u(k)), y(k) = h(x(k), u(k)) and operating point:  $(\hat{x}, \hat{u})$  nonlinear
- Goal: find a linearized model around  $(\hat{x}, \hat{u})$

Define: 
$$\Delta x = x - \hat{x}$$
,  $\Delta u = u - \hat{u}$ ,  $\partial y = y - h(\hat{x}, \hat{x})$ 

Goal: 
$$\Delta X(k+1) \approx \hat{A} \Delta X(k) + \hat{B} \omega k + C$$

$$\hat{L}_{R}^{n} \qquad \hat{R}^{n} \qquad \hat{R}^{m}$$

■ Jacobian matrix of multivariable function  $\underline{f}: \mathbb{R}^n \to \mathbb{R}^m$ 

$$\begin{bmatrix}
f_1(z_1, z_2, z_3) \\
f_2(z_1, z_2, z_3)
\end{bmatrix}$$

$$\frac{\partial f}{\partial z} \stackrel{\triangle}{=} \left[ \frac{\partial f_i}{\partial z_j} \right] = \left[ \frac{\partial f_i}{\partial z_i} \right], \quad \int \frac{\partial f}{\partial z_i} \frac{\partial f$$

$$df = (\frac{32}{34}) dz$$

Example of Jacobian matrix: 
$$f(z) = \begin{bmatrix} 2z_1 + e^{z_2} \\ \log(z_3) + \frac{1}{z_2} \end{bmatrix}, \ \hat{z} = \begin{bmatrix} 1 \\ 2 \\ 3 \end{bmatrix}$$

$$\frac{2f(z)}{8z}(z) = \begin{bmatrix} 2 & e^{2z} & 0 \\ 0 & -\frac{1}{2} & \frac{1}{2} \end{bmatrix}$$

$$= \begin{bmatrix} 2 & e^{2z} & 0 \\ 0 & -\frac{1}{2} & \frac{1}{2} \end{bmatrix}$$

$$= \begin{bmatrix} 2 & e^{2z} & 0 \\ 0 & -\frac{1}{2} & \frac{1}{2} \end{bmatrix}$$

Taylor expansion of multivariate function

• General expression: 
$$f(z) = f(\hat{z}) + \left(\frac{\partial f}{\partial z}(z)\Big|_{z=\hat{z}}\right) \Delta z + \text{H.O.T}$$

$$f(z) = f(\hat{z}) + \left(\frac{\partial f}{\partial z}(z)\Big|_{z=\hat{z}}\right) \Delta z + \text{H.O.T}$$

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• Linearization around  $(\hat{x}, \hat{u})$  using Taylor expansion:

$$f(z) = f(\hat{x}) + \frac{\partial f}{\partial \hat{x}} = \hat{x} + \forall x + \hat{x}$$

$$f(x, u) \approx f(\hat{x}, \hat{u}) + \left(\frac{\partial f(x, u)}{\partial x}\Big|_{x = \hat{x}, u = \hat{u}}\right) \cdot (x - \hat{x}) + \left(\frac{\partial f(x, u)}{\partial u}\Big|_{x = \hat{x}, u = \hat{u}}\right) \cdot (u - \hat{u})$$

$$= \hat{A} \cdot \Delta x + \hat{B} \cdot \Delta u + f(\hat{x}, \hat{u})$$

$$= \hat{A} \cdot \Delta x + \hat{B} \cdot \Delta u + f(\hat{x}, \hat{u})$$

$$\Rightarrow \hat{A} = \hat{A} \cdot \hat{A} + \hat{B} \cdot \hat{A} + \hat{B} \cdot \hat{A} + \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A} \cdot \hat{A} \cdot \hat{A} \cdot \hat{A} \cdot \hat{A} + \hat{A} \cdot \hat{A}$$

$$h(x,u) \approx h(\hat{x},\hat{u}) + \left(\frac{\partial h(x,u)}{\partial x}\Big|_{x=\hat{x},u=\hat{u}}\right) \cdot (x-\hat{x}) + \left(\frac{\partial h(x,u)}{\partial u}\Big|_{x=\hat{x},u=\hat{u}}\right) \cdot (u-\hat{u})$$

$$\Rightarrow (x_{k+1}) = \int_{\hat{x}_{k+1}} (x_{k+1}) \cdot \hat{x} = \int_{\hat{x}_{k+1}} (x_{k+1}) \cdot \hat{x}$$

$$\begin{bmatrix} x_1(k+1) \\ x_2(k+1) \end{bmatrix} = \begin{bmatrix} \sin(x_2(k)) + \cos(u_2(k)) \\ x_1(k)x_2(k) + u_1u_2(k) \end{bmatrix}$$

fxe ue)

$$\underline{y(k)} = \cos(x_2(k)) + 2x_1(k) \qquad \hat{x} = \begin{bmatrix} 0 \\ 0 \end{bmatrix}, \, \hat{u} = \begin{bmatrix} 0 \\ \frac{\pi}{2} \end{bmatrix}$$

$$\hat{\beta} = \frac{\partial f}{\partial u} \Big|_{\hat{X}/\hat{G}} = \begin{bmatrix} 0 & -\sin(u_2(k)) \\ u_2(k) & u_1(k) \end{bmatrix} \Big|_{\hat{X}/\hat{G}} = \begin{bmatrix} 0 & -1 \\ \frac{2}{2} & 0 \end{bmatrix}$$

$$\hat{C} = \left[ \frac{\partial h}{\partial \chi_1} \quad \frac{\partial h}{\partial \chi_2} \right] \Big|_{\hat{X}, \hat{u}} = \begin{bmatrix} 2 & 0 \end{bmatrix}$$

$$\Delta X_{kH} = \hat{A} \Delta X_{k} + \hat{B} \Delta N_{k}$$

$$\delta J_{k} = \hat{C} \Delta X_{k} + 0$$



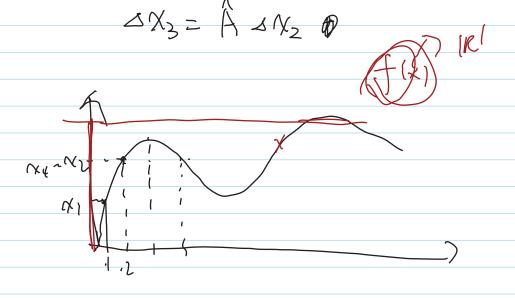
$$x_{k+1} = f(x_k, u_k) \qquad f(x_{k+1}) \qquad f(x_{$$

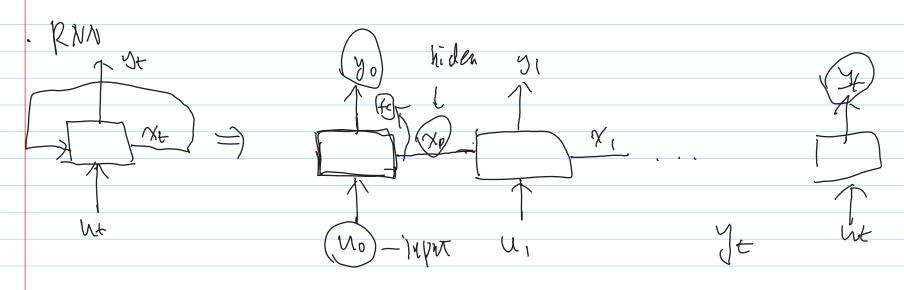
)

0 1/2

$$\chi_{2}=f(\chi_{1}) \qquad \qquad \chi_{k}=\chi_{k}-\chi$$

$$\chi_{3}=f(\chi_{2})$$





 $\chi_{\circ} \rightarrow \chi_{1} \rightarrow \chi_{1} \dots$ 

X0 X1

M= JM(X)

M= JPhut

TEMU

### **Outline**

State space model: definition and examples

From continuous-time to discrete time model

From nonlinear to linear model

System solution and stability

General linear state space model:

$$x(k+1) = A(k)x(k) + B(k)u(k),$$
  
$$y(k) = C(k)x(k) + D(k)u(k)$$

- If system matrices (A(k), B(k), C(k), D(k)) change over time k, then system is called **Linear Time Varying (LTV)** system
- If system matrices are constant w.r.t. to time, then the system is called a Linear Time Invariant (LTI) System

$$x(k+1) = Ax(k) + Bu(k),$$
  

$$y(k) = Cx(k) + Du(k)$$

Derivation of Solution to LTI state space system:

$$x(k+1) = Ax(k) + Bu(k),$$
  
$$y(k) = Cx(k) + Du(k)$$

• given initial state  $\underline{x}(0) = \hat{x}$ , and control sequence  $u(0), ..., u(k), k \ge 1$ 

0, we have  $x(k) = A^{k} \hat{x} + \sum_{j=0}^{k-1} A^{k-j-1} Bu(j)$ 

$$\chi(1) = A \chi(0) + B u(0) = A \hat{\chi} + B u(0)$$

$$\chi(2) = A \chi(1) + B u(1) = A^{2} \hat{\chi} + A B u(0) + B u(1)$$

$$\chi(3) = A \chi(2) + B u(2) = A^{3} \hat{\chi} + A^{2} B u(0) + A B u(1) + B u(2)$$

$$\vdots$$

$$\xi = A^{k} \hat{\chi} + A^{k-1} B u(0) + A^{k-2} B u(1) + \dots + B u(k-1)$$

$$= A^{k} \hat{\chi} + A^{k-1} B u(0) + A^{k-2} B u(1) + \dots + B u(k-1)$$

Given system (D, R, C, O)

- · Stability
- · Emphloriti- any
- · ple sevesbil ty

- A large portion of control applications can be transformed into a regulation problem
  - Regulation problem: keep certain function of the state x(k) or output y(k) close to a known constant reference value under disturbances and model uncertainties

• Keep inverted pendulum at upright position ( $\theta = 0$ )

Maintain a desired attitude of spacecraft or aircraft

For example:

- Air conditioner regulate temperate close to setpoint (e.g. 75F)
- Cruise control maintain a constant speed despite uncertain road conditions
- Converter maintains a desired voltage level for different loads

• If reference  $y_{ref}(t)$  is changing, this is no longer a regulation problem (becomes a **tracking problem**)

- Stabiling: 1) Bounded input Bounded output & external (Z) 'Convergence' internal · stability of a system - equi(ibrium: J Autonomous system.  $\dot{X} = f(x)$ ,  $\dot{X}_{\alpha+1} = f(x_{\alpha})$ )  $\hat{X} \text{ is equilibrium: } f(\hat{X}) = 0 , \quad f(\hat{X}) = \hat{X}$ [controlled system:  $\dot{x} = f(x, u)$ 

L) controller: M(·), => closed-Loop system  $\dot{x} = f(x, M(x))$  $=\int_{cL}(x)$ à is equilibrium of fel if fel (2)=0 in discrete time: fac(x)=x

M(x)= 2  $\dot{X} = f(x, z) = f_{cl}(x)$  ■ Internal Stability (with  $u(k) \equiv 0$ , i.e. concerned with zero-input state response)

$$x(k+1) = Ax(k) + Bu(k),$$
  
$$y(k) = Cx(k) + Du(k)$$

- Asymptotic stable:  $||x(k)|| \to 0$ , as  $k \to \infty$ , for all initial state  $\hat{x}$
- Marginal stable:  $||x(k)|| \le M$ , for all k = 1, 2, ...
- Recall state space solution for linear systems:

$$x(k) = A^{k}\widehat{x} + \sum_{j=0}^{k-1} A^{k-j-1}Bu(j)$$

■ Therefore, or linear system, the key for stability analysis is to understand how  $A^k$  behave as  $k \to \infty$ 

■ Case 1: diagonal matrix: e.g.  $A = \begin{bmatrix} \lambda_1 & 0 & 0 \\ 0 & \lambda_2 & 0 \\ 0 & 0 & \lambda_3 \end{bmatrix}$ 

■ Case 2: diagonalizable matrix, i.e.  $\exists T$  such that  $A = TDT^{-1}$ 

- Case 3: Unfortunately, *not all* square matrices are diagonalizable
  - e.g.:  $\begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix}$  is **not diagonalizable**

■ Theorem (Internal stability): LTI (A, B) is asymptotically stable if all eigs of A satisfies  $|\lambda_i| < 1$ 

More discussions