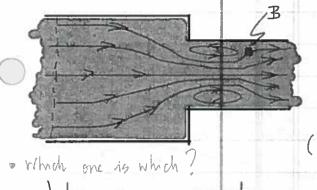
Chapter 4: Integral analysis - D (Fluids Moving.)

As engineers you can study the flow properties (pressure, velocity) in a flow field in two different ways:

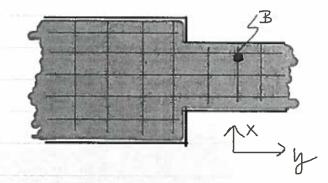
- 1) Lagrangian approach: Track the trajectory of a fluid particle and study how this one evolves in time. => (like a pathine).
- 2) Eulerian approach: Solve a set of equations for the flow properties at any point in the flow field.

D Using a pictorial representation: let's consider a sudden contraction:



(Natro)

1) Lagrangian approach



2) Eulerian approach.

if someone wanted to determine, lots say, the pressure at point 3, knowing the velocity field and pressure at the inlet, it could be done in 2 different ways:

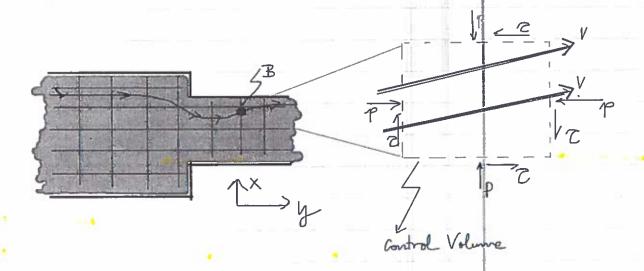
1) Find the pathline that passes by point 3. As the particle mones along the pathline, the pressure changes can be tracked using Euler's equation (will see it later on). Or if the flow is steady then Bernoulli's equation can be used. To LAGRANTIAN APPROACH.

Problem: Importable to keep track of all possible pathlines, to evaluate the fluid properties at each point. => For unteady flows, even worse, since different pathlines will pass through the save point.

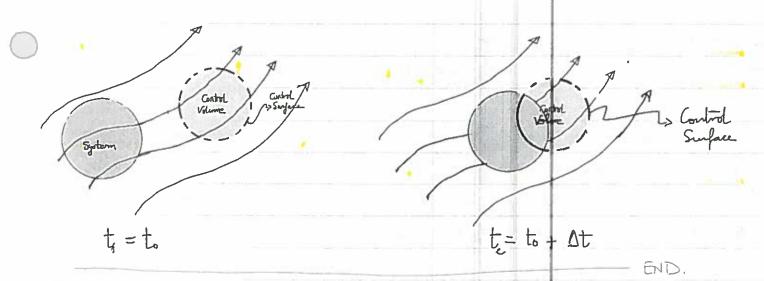
I) Tendop a solution to the flow field that provides the flow properties at any point. If for example, the pressure was available as a function of location p(x17), then one could get Ps just by replacing (XB, YB) coordinates into the pressure solution. Evertime APPROACH

D In order to tackle the perform following in Educial approach, the oquations of movement must be re-written accordingly. (Similar to solid mechanics, the fundamental equations are developed using the "free body" concept.)

Mening that: imagine we could isolate a portion of the flow field, and the offects of the surroundings where introduced by forces acting on the limiting surfaces endosing our portion of flow field.



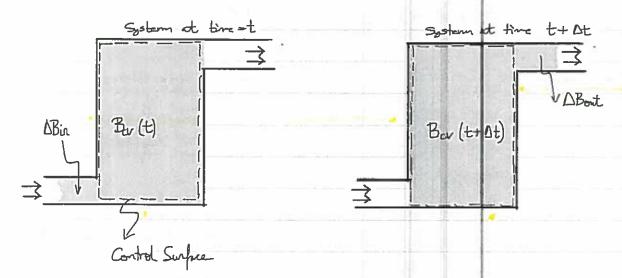
The volume enclosing a point is called "control volume". The effects of the surroundings are replaced by pressure forces and shear stress acting on the surface of the control volume. Also, it must be considered that here is a flow through the control volume (CV). D Sometimes the Federian approach is also called as "control volume" approach.



o Raynolds transport theorem

L> it relates the Eulerian and Lagrangian approaches, from a mathematical standpoint. The Reynolds transport theorem is derived by considering the rate of change of a property (B) of a system as it person through the control volume.

Let's consider the following progression of a system as it flows through a control volume:



The property B of the system are be any of the relevant variables: Mass, linear momentum, energy, argular momentum, ...

Unfortunately the basic equations (Newton's 2rd law) are applicable for a body of given mass, or a fluid particle, moving in the field = Lagrangian approach.

Next, a procedure to transform the governing laws from a Lagrangian perspective to an Eulerian perspective, will be introduced. => Therefore, equations can be applied in a control volume. Later, by considering the volume enclosed in the CV tend to zero, the Eulerian forms of the differential equations will be derived.

* The Fulerian form of the equations is very important because provides the basic equations for numerically solving the flow field => CFD (computational Fluid Dynamics).

O System and Control Volume (Notes)

System = is a continuous muss of fluid that always contains the same matter.

The shape of the system may charge with time, but the mass is constant
since it always consists of the same matter.

Control Volume = is a volume located in space and through which meter our pass. The system can pass through the CV. The shape and position of the CV is problem dependent. (There is no unique CV).

The CV is enclosed by the control surface, through which

muss enters and leaves the control volume. The CV can deform with time, move, rotate, and the mass in the CV can charge.

In the previous drawing the control notours is enclosed by the control surface (dashed live). The system is identified with the blue-colored regions. the system consists of the material inside the CV and the material at time = t; going in. Therefore the property (B) of the system can be withen as: 3 3ys (t) = Ber (t) + Bin at time = t+At, the system has moved and now consists of the material in the control volume and the material passing out, so Boys (to+At) = Ber (t+At) + ABout therefore, one can write the rate of drange of the property B as. which by replacing the previous terms: (Notes) dbiys _ lin Ber (t+Dt) + \Dont - Ber (t) - \Dbin \

at \Dt > 0 upon rearrangement; dBoys = lim Ber (t+Dt) - Ber (t) + lan Bont _ lum \(\Delta \to \Dt \rightarrow \Dt \rightarr

term (1) = Rate of change of the property 3' inside the control volume $\lim_{\Delta t \to 0} \frac{B_{cv}(t+\Delta t) - B_{cv}(t)}{\Delta t} = \frac{3B_{cv}}{2t}$

term (2)= outflow of the property B through the control surface.

(Notes)

Lin ABout _ Bout (= flux of the property B' through the CS).

At >0 At

term 3 = inflow of the property 3 through control surface.

(Notes)

lim Abin = Bon (= influx of property B' through CS).

therefore, be rate of change of property 3 can be rewritten as follows:

$$\frac{dB_{yy}}{dt} = \frac{2B_{uy}}{\partial t} + \dot{B}_{out} - \dot{B}_{uy} = \frac{2B_{uy}}{\partial t} + \dot{B}_{net}$$

At this point we just need a specific expression for Ber and Brut. The amount of property 3 contained in a control volume at a given instart is

$$B_{cr} = \int_{cr} b dm = \int_{cr} b p dt$$

where dt is the CV differential. One could imagine dm" as the differential of "material" or "fluid" contained inside the CV. Therefore "b" represents the amount of property B per wit muss (material or fluid contained in the CV).

(Netr)

if B represents mass: B=M; -> b=1

B represent linear momentum: B=p; -> b= F

B represents angular momentum: B= H; ---- b= Fx F

B represents energy: $B=E; \longrightarrow b=e$ B represents entropy: $B=S; \longrightarrow b=s$

therefore one can rewrite that: $\frac{2}{2}B_{cv} = \frac{2}{2t} \int_{0}^{\infty} b \, dt$

On the other hand the not flow rate of the property B through the control volume is given by:

Notice that here the integral is not a volume of integral of the control volume, but a surface integral of the overall control surface.

Further, here I represents the velocity vestor of the fluid crossing the control surface. Thus the velocity is with respect to the velocity of the control surface. Also, do represents the differential voctor of the CS. Recall that this one points outwards of the surface

Finally the nate of change of property B can be written as:

Regnolds Transport theorem

expressed in a different way: (Notes) Rate of change Rate of change Not outflow of proper by 3 of property 3 of property B through carrol in control volume of system Surface Important: Special care should be taken when evaluating the dot product in the Raynolds transport theorem. A is always directed outwards, the dot product will be positive when it is outwards, and regative when it is inwords. Control Surface Control Volume $\overline{V_1} \cdot \overline{A_1} = -V_1 A_1$

Purticular vare: Conservation of Mass & Continuity equation

The continuity equation derives from the conservation of mass, which in Lagrangian form, simply states that the mass of the systam is constant

May, = te. (Here the property B= M, thus b=1).

The Eulerian form is derived applying the Reynolds transport theren.

$$\frac{dM}{dt}\Big)_{system} = \frac{2}{3t} \int_{cv} \rho \, dV + \int_{cs} \rho \, \vec{F} \cdot d\vec{A} = 0$$

This is the man conservation equation, which states that the total mans of the system down't charge with thre.

Torm (1) indicates the rate of drange of man within the CV.

Term (2) represents the net rate of mens flux out through the control surpre.
This equation is also called the continuity equation, and can be rewritten as:

* In the special case where the flow is incompressible, p= te, and it can be pulled out from the integrals, and thus simplified. Therefore:

$$\frac{\partial V}{\partial t} + \int_{cs} \vec{v} \cdot d\vec{A} = 0$$

* For a non-deformable control volume of fixed size and shape, t= de. Thurfore, communition of man becomes:

$$\int_{cs} \vec{V} \cdot d\vec{A} = 0$$

if the velocity field can be approximated as aniform at the inlet and outlet of the CV, the previous equation can be further simplified to:

Note: So for we haven't imposed the condition of the flow to be steady, we have only applied the restriction of incompressibility. Therefore:

$$\sum_{cs} \vec{v} \cdot \vec{A} = 0$$

$$\int_{cs} \vec{v} \cdot d\vec{A} = 0$$

state corrervation of mass for an incompressible flow that can be steady or unsteady.

* The integral term: Jos of the integrand: [13/t]

Thus one can before the volume flow rate Q, through a section of the control volume as:

* for a steady case, the mass flow rate into a control volume must be equal to be mass flow rate out of the control volume.