## CHAPTER 12

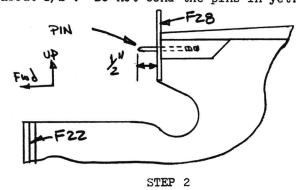
### CANARD INSTALLATION

OVERVIEW: In this chapter you will align and temporarily install the canard on the fuselage. The canard is aligned for zero sweep, zero dihedral, and zero incidence. Alignment pins are installed in the F28 bulkhead and alignment tabs installed on the canard to match. The canard tabs (aluminum tabs installed on the canard in chapter ) are drilled in place and nut plates are installed to facilitate removal in service. The following prefab parts used in this chapter are available from a distributor: CN1 canard atch bushing (2 reqd) and CN2 3/16" ID flanged bushing (2 reqd).

#### STEP 1

## INSTALLING THE ALIGNMENT PINS

This should only take 1/2 hour. Cut the heads off of two AN-3-20 or -20A bolts and sand or file a round nose on the unthreaded ends. Drill two holes in the F28 bulkhead and wood doublers as shown about 1-1/2-inch deep. Check that the pins will slip into the holes and stick out only about 1/2". Do not bond the pins in yet.



# ALIGNING THE CANARD

This step may crowd your shop slightly, but only takes about an hour, and you need an assistant occasionally to help measure things.

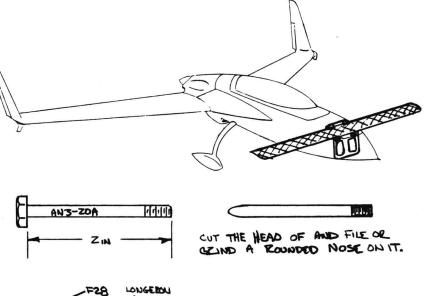
Get your canard out of storage and position it (without electors) on the fuse-lage as shown on page A-7. Use a pair of "C" clamps to hold the LIFT tabs against the bulkhead (be careful not to bend or scratch them). Check for zero sweep-back by measuring from the canard tips to the wing fittings as shown. Both should measure the same. Use shims between the canard lift tabs and the forward bulkhead to adjust the sweep if necessary.

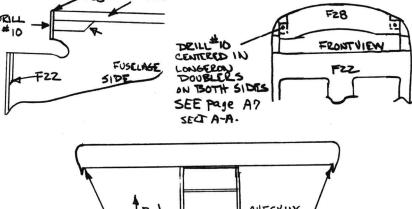
Next level across the top fuselage longerons and the canard spar cap. Lastly (and most important), use your canard incidence blocks to accurately set the incidence at zero with the top longerons. The patterns for the incidence block is found

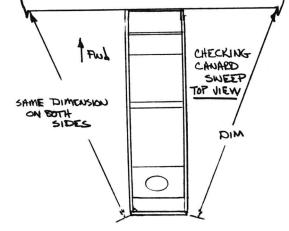
Next, level across the top fuselage longerons and the canard spar cap. Lastly (and most important), use your canard incidence blocks to accurately set the incidence at zero with the top longerons. The patterns for the incidence block is found on page A- 13 If your canard has a slight twist, set the average incidence equal to the longerons. Recheck sweep, dihedral, and incidence and clamp firmly in position. Use the #10 pilot holes in the canard tabs and drill #10 through the F22 bulkhead. Go cut two tabs of high density foam (dark red) as shown.

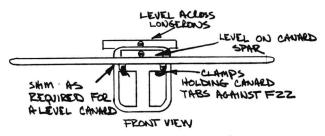
bulkhead. Go cut two tabs of high density foam (dark red) as shown.

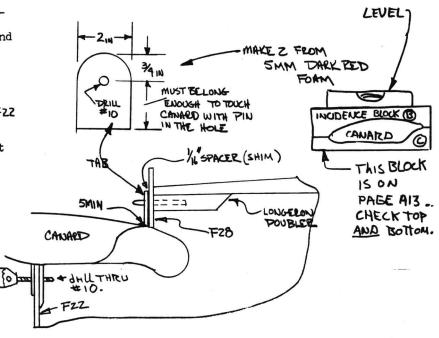
Trim the tabs so that they fit against the top of the canard and against a 1/16" shim on the bulkhead with the pin in the hole as shown. With everything trimmed to fit, 5MIN the tabs to the canard. When hard, carefully remove the canard without knocking the tabs loose.











Measure the thickness of any shim that was used, to correct the sweep of the canard (any shim between F22 and the lift tabs). Remove the shim and dull the the canard (any shim between F22 and the lift tabs). Remove the shim and dull the forward surface of the F22 bulkhead where the lift tabs rest, and lay up additional BID pads to the thickness of the shim (BID is about .013/ply). After cure, back drill #10 through the bulkhead and new glass. Thus, you have replaced the shim with additional glass pad. The tabs on the canard are glassed as shown. shim with additional glass pad. The tabs on the canard are glassed as shown.

Reopen the #10 holes with a drill (you can see the hole through the glass layup).

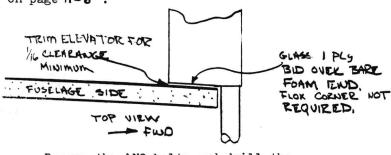
Drill these holes out to accomodate a 3/16"

I.D. flanged bushing (available from a distributor). 5MIN the bushings in position. Reinstall the canard using AN3 (3/16" dia) bolts to locate the aluminum lift tabs. Recheck your sweep and incidence.

incidence.

CN2

Now, mount the ELEVATORS on the canard, trimming their inboard length to obtain about 1/16" clearance with the fuselage sides. Check that adequate clearance (0.1") exists around the ELEVATOR tubes (at the fuselage sides) when the elevators are moved through their full travel of 20° trailing edge up and 22° trailing edge down. The elevator angle template pattern is found on page 11-6.



Remove the AN3 bolts and drill the aluminum lift tabs and F22 bulkhead up to 1/4-inch dia. Remove the canard.

USING YOUR 5/8 Counterbore Tecl, drill the two 1/4" holes in the F22 bulkhead up to 5/8" for 5/8 0.D.x1/4 I.D. aluminum bushings. These are CNL canard atch bushings available from a distributor. They are supplied long, so you can trim to the exact thickness of your bulkhead. Do not trim, nor bond the bushings in yet, because during chapter 13 you will be installing additional glass plies on the foreward face of F22 to tie it into the nose sides. This will move your canard forward from where it now sits.

Now, remove the 3/16" pins from the F28 bulkhead and apply one ply BID to the entire area shown, using 1/4" flox corners at the edges of the bare foam. When cured, drill out the #10 holes and permanently install the 3/16" alignment pins with wet flox.

NOTE HARDWARE SHOWN ON A ? TO MOUNT LIFT TABS TO BULKHEAD. THESE ARE PERMINANTY INSTALLED DURING CHAP 13.

> NOTE: IF you are installing a VariEZE canard on your Long-EZ, ignore its waterline. The Long-EZ has 0.6 degree MOTE CANARD INCIDENCE. USING The B) and C) Incidence Templates ON Pg A13 will assure correct Incidence of either canard.

