Time-Optimal Control of a Pseudo-Helicopter

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Aims

- Design model predictive controllers with a minimum time objective for the Quanser 3 DOF helicopter
- Compare the performance of the controllers

Objectives

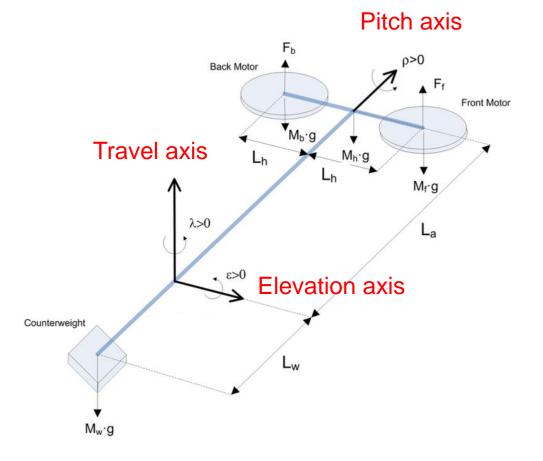
- Improve and validate mathematical model of the helicopter
- Determine optimal path
- Design tracking and nontracking controllers
- Investigate performance of the controllers in the absence and presence of disturbances

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Introduction to the Pseudo-Helicopter

- 3 degrees of freedom: elevation, pitch, travel
- Highly coupled, nonlinear system
- Constraints on motor voltage, elevation, and pitch



Introduction to the Pseudo-Helicopter

Inputs:

Outputs:

- Front motor voltage V_f

Elevation

 ϵ

Back motor voltage V_h

Pitch

ρ

Travel

λ

Mathematical Model and Model Validation

Euler-Lagrange equations:

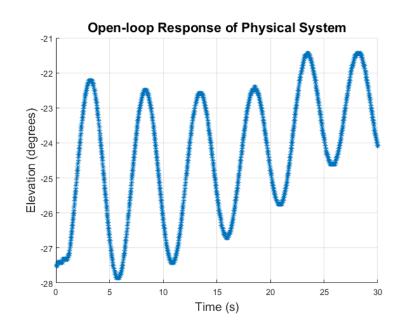
$$\frac{\partial}{\partial q_i} L(t, \boldsymbol{q}, \dot{\boldsymbol{q}}) - \frac{d}{dt} \frac{\partial}{\partial \dot{q}_i} L(t, \boldsymbol{q}, \dot{\boldsymbol{q}}) = Q_i \text{ for } i = 1, 2, \dots, n$$

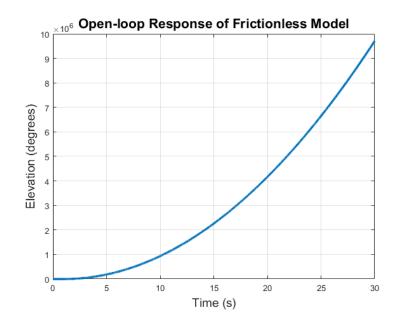
where
$$\mathbf{q} = [\epsilon \quad \rho \quad \lambda]^T$$
 and $L(t, \mathbf{q}, \dot{\mathbf{q}}) = T - V$

$$\rightarrow \dot{x} = f(x, u)$$

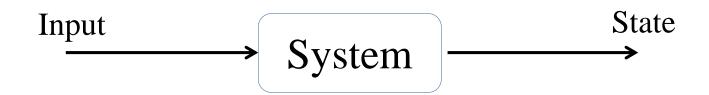
where
$$\mathbf{x} = [\epsilon \quad \rho \quad \lambda \quad \dot{\epsilon} \quad \dot{\rho} \quad \dot{\lambda}]^T$$
 and $\mathbf{u} = [V_f \quad V_b]^T$

Mathematical Model and Model Validation

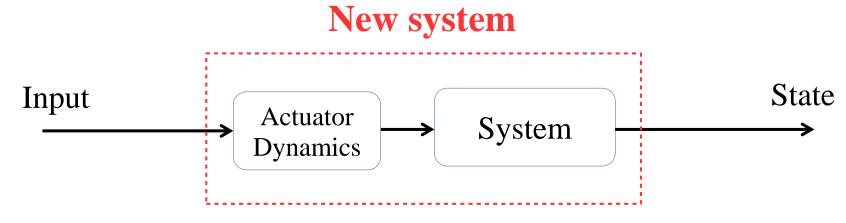




Mathematical Model and Model Validation



No Actuator Lag



First Order Actuator Lag

Mathematical Model and Model Validation

 $\tilde{Q}_{i} = Q_{i} + F_{v_{i}}\dot{q}_{i} + F_{c_{i}}sign(q_{i})$

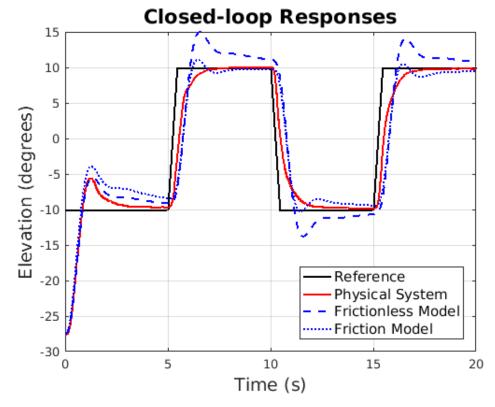
Mechanical damping

Kinetic friction

- Friction in pitch and travel directions negligible
- Simple friction model preferred
- sign function replaced with smooth approximation

Mathematical Model and Model Validation

- Open-loop:
 - RMS deviation of each model from experimental data showed a 533% improvement for the friction model
- Closed-loop:
 - Compared closed-loop responses of the models and the physical system using same LQR gains matrix



Optimal Path Planning

- Objective: get from start point to Cost function: end point in minimum time

$$V(x,u,t) = t_f$$

Start point:

$$x_0 = [0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad 0]^T$$

End point:

$$x_f \in \begin{bmatrix} 0 & 0 & \pi & 0 & 0 & 0 & -\infty & -\infty \\ 0 & 0 & \pi & 0 & 0 & \infty & \infty \end{bmatrix}$$

Hard constraints:

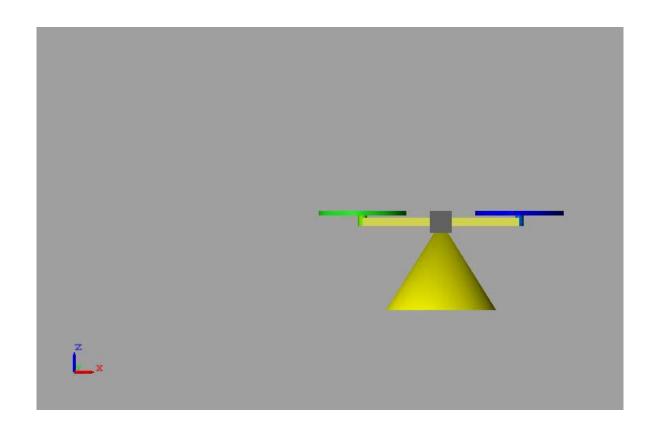
$$\epsilon \in [-27.5^{\circ}, 27.5^{\circ}]$$

$$\rho \in [-90^{\circ}, 90^{\circ}]$$

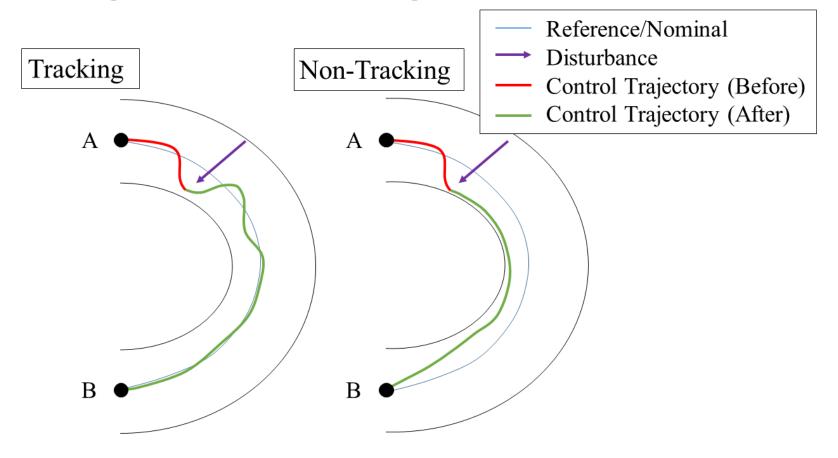
$$u \in \begin{bmatrix} -24 & 24 \\ -24 & 24 \end{bmatrix}$$

Optimal Path Planning

$$t_f = 2.1$$
 seconds



Tracking vs Non-Tracking



Tracking vs Non-Tracking

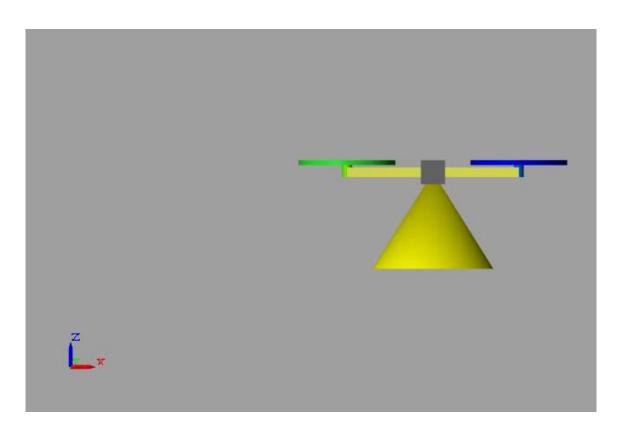
Linear tracking receding horizon controller:

- Objective: track the optimal path using a linear state space model as the predictive controller
- Cost function:

$$V(x, u, t) = \int_{t_{k_0}}^{t_{k_f}} ||y - y_r||^2 dt$$

Tracking vs Non-Tracking

$$||y_f - y_{des}||_2 = 1.124$$



Tracking vs Non-Tracking

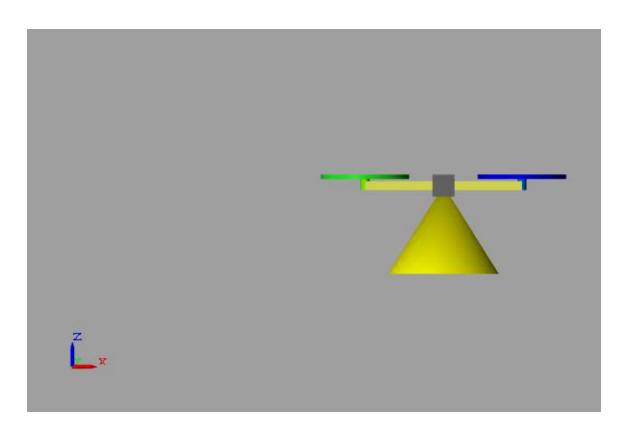
Nonlinear non-tracking variable horizon controller:

- Objective: get from start point to end point in the minimum time possible using nonlinear frictionless model as predictive plant
- Cost function:

$$V(\boldsymbol{x},\boldsymbol{u},t)=t_f$$

Tracking vs Non-Tracking

$$||y_f - y_{des}||_2 = 0.646$$



Disturbance Handling

- White noise disturbance:
 - Added Gaussian white
 noise with signal to
 noise ratio 2
 - $-\dot{x} = f(x, u) + \xi(t)$

- Bias disturbance:
 - Added disturbance in the travel direction

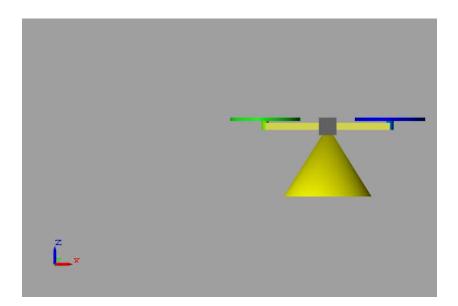
$$- \ddot{\lambda} = f_{\lambda}(x, u) + \xi(t)$$

$$\xi(t) = \begin{cases} 2 & if \ t < 0.2 \\ 0 & otherwise \end{cases}$$

White noise disturbance

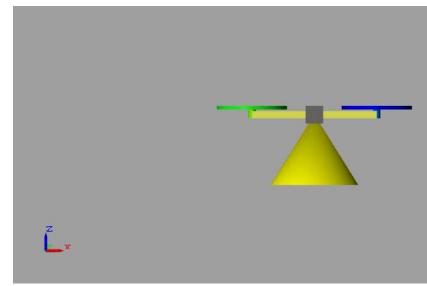
Disturbance Handling

Tracking



$$||y_f - y_{des}||_2 = 2.539$$

Non-Tracking

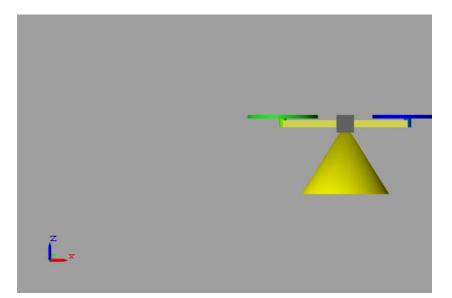


$$||y_f - y_{des}||_2 = 0.7718$$

Bias disturbance

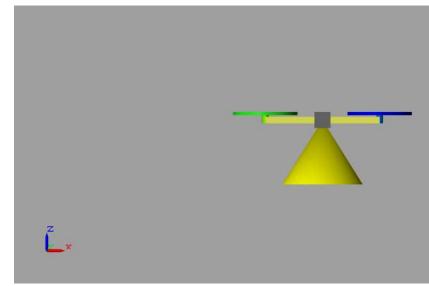
Disturbance Handling

Tracking



$$\|y_f - y_{des}\|_2 = 0.868$$

Non-Tracking



$$\|y_f - y_{des}\|_2 = 0.336$$

Future Work

- Reduce computational time, possibly by parallelising the code or calculating multiple inputs each iteration
- Reduce the effects of plant-model mismatch by either introducing an input disturbance model or by using integral action
- Improve non-tracking controller performance by switching to a minimum energy cost function when in the vicinity of the desired end point

Conclusions

- Modified the nonlinear model by adding previously ignored dynamics, improving open-loop performance by 533%
- Determined the optimal path for a minimum time control objective
- Designed tracking and non-tracking controllers for the control objective
- Compared the performance of the controllers without disturbance, with non-tracking offering 74% better performance
- Compared the performance of the controllers with disturbance:

	Tracking	Non-tracking
White Noise	126% reduction	20% reduction
Bias	23% improvement	48% improvement