

The background of the image is a scenic sunset over a beach. The sky is filled with vibrant orange, yellow, and purple clouds. The sun is low on the horizon, casting a warm glow over the water and sand. The waves are gently breaking on the shore. In the foreground, the dark silhouette of a person is visible on the sand. A large, semi-transparent circular logo is centered over the image. The logo consists of a thick white outer ring and a thinner white inner ring, both with small gaps at the top and bottom. The text "UCSB" is written in a large, black, sans-serif font across the middle of the logo. Below it, the word "HYPERLOOP" is written in a smaller, black, sans-serif font. At the bottom of the logo, there is a line of text that reads "Follow along at:" followed by the URL "ucsbhyperloop.com/texas".

UCSB

HYPERLOOP

Follow along at:
ucsbhyperloop.com/texas

TEAM



Introduction

Frame

Shell

Propulsion

I-Beam
Stabilization

Braking

Weight

Levitation

Electronics

Controls

Power

Scalability

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INTRODUCTION

- Follow along at ucsbhyperloop.com/texas
- 21 senior engineering undergraduates working to build and test a pod at Competition Weekend
- Emphasizing cost-effectiveness, scalability, and feasibility
- Estimated cost to complete design: \$40,000
 - Funding/resources already raised:
 - \$5,000 from Ingersoll Rand
 - \$5,000 from Raytheon
 - \$5,000 from private donors
 - Electronics donated by NXP Semiconductors
 - ~\$25,000 to be raised

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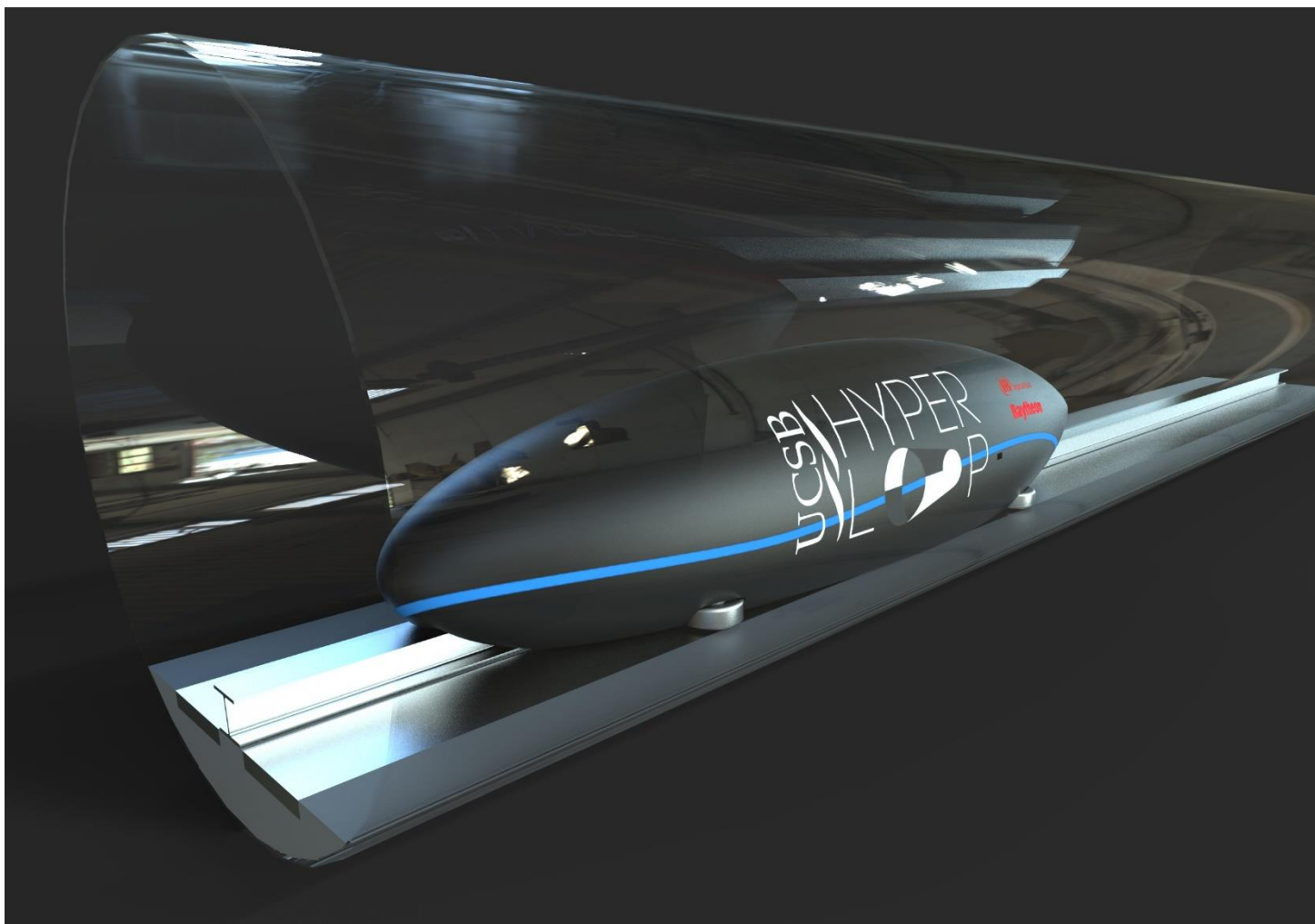
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POD OVERVIEW



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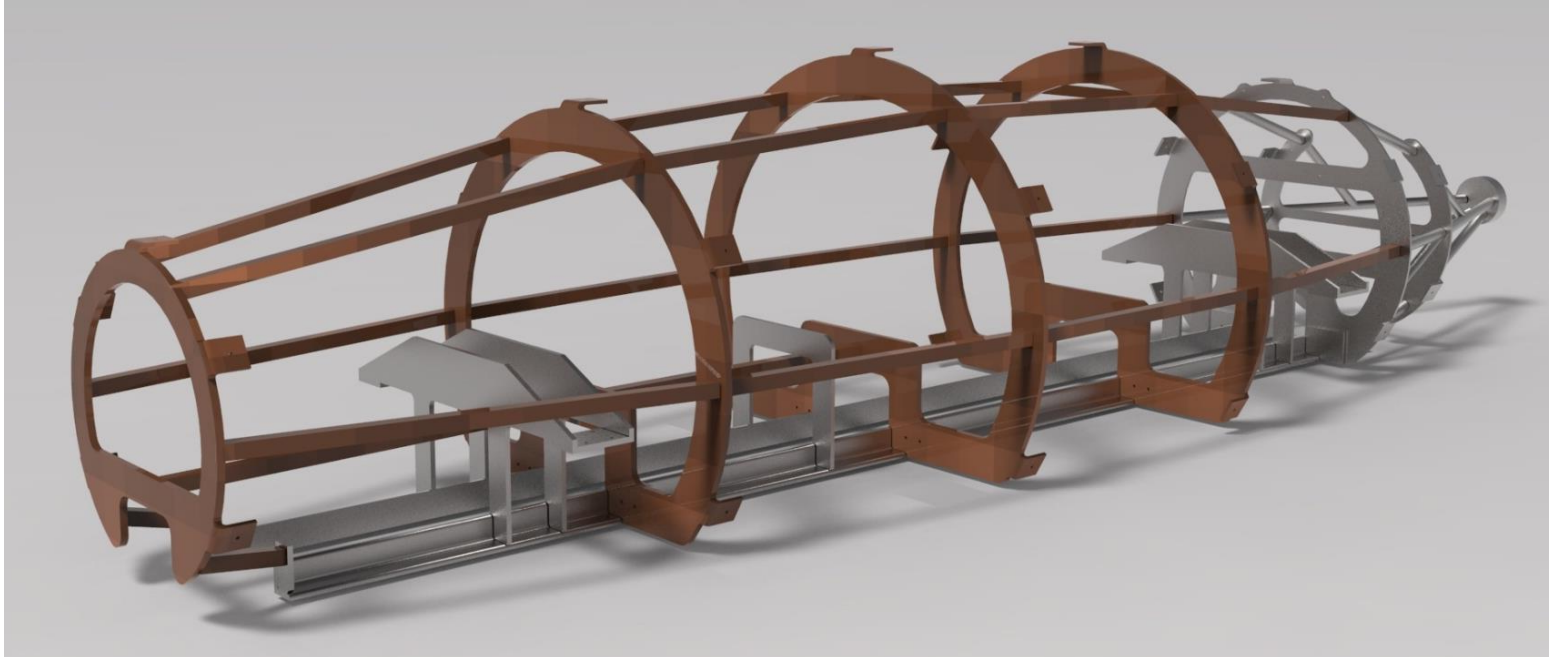
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FRAME

- 13'7" (length) x 3'4" (width) x 2'7" (height)
- Divided into front, base, and rear frame
 - Lightweight, wooden front frame reinforces shell
 - Aluminum base frame supports all major subsystems
 - Steel tube rear frame interfaces with SpaceX pusher



SHELL

- Tapered bullet shape
- E-Glass reinforced polyester
 - Quasi-isotropic material dissipates force evenly in all directions
- Withstands tube pressure breach with a factor of safety of 4000



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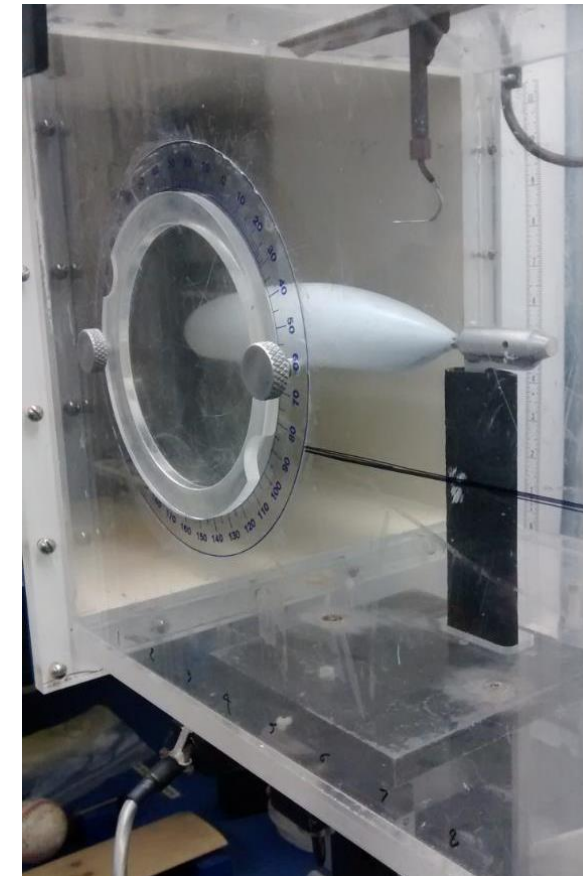
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WIND TUNNEL TESTING

- 3-D printed ABS plastic pod model
- Reynolds number in evacuated tube is 8.5×10^3
- Estimated drag coefficient = 1.5 lbs
 - Drag force = 6.6 N



Pod model mounted
in wind tunnel

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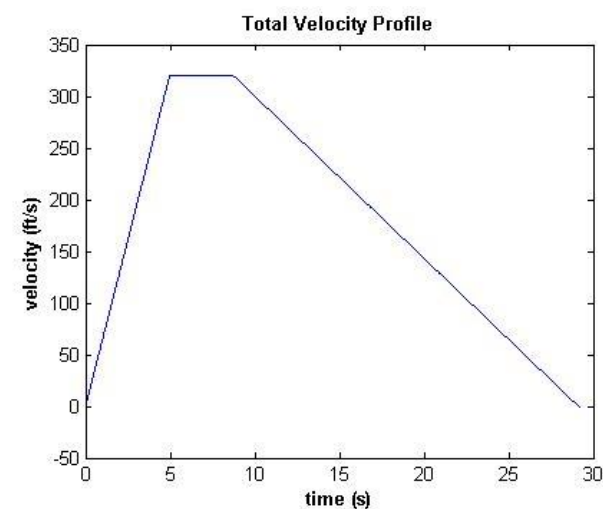
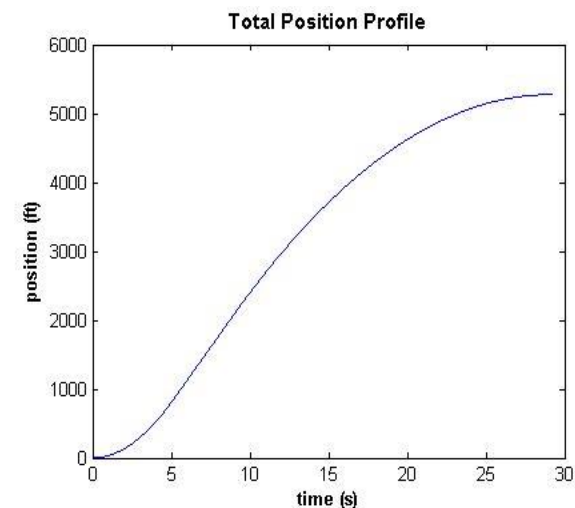
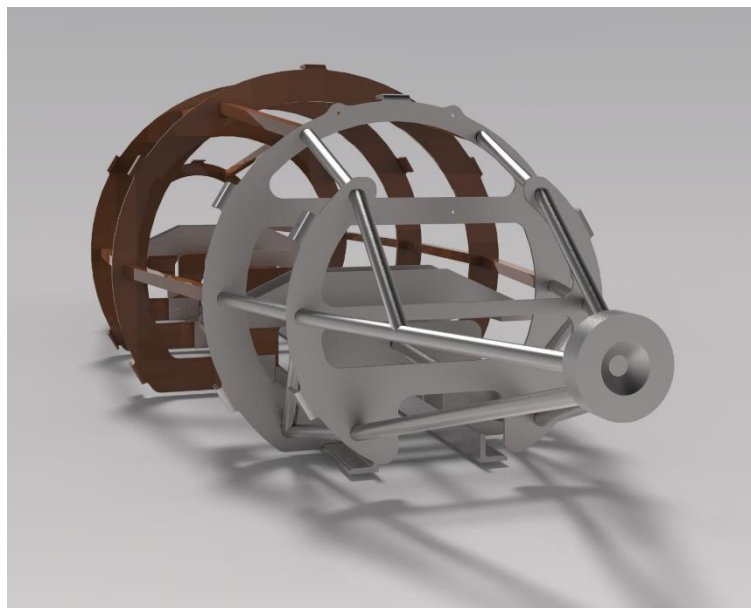
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TRAJECTORY

- Top speed of 218 mph (320 fps)
- Total run time of 29.16 s
 - Acceleration – 4.98 s
 - Coasting – 3.75 s
 - Braking – 20.43 s



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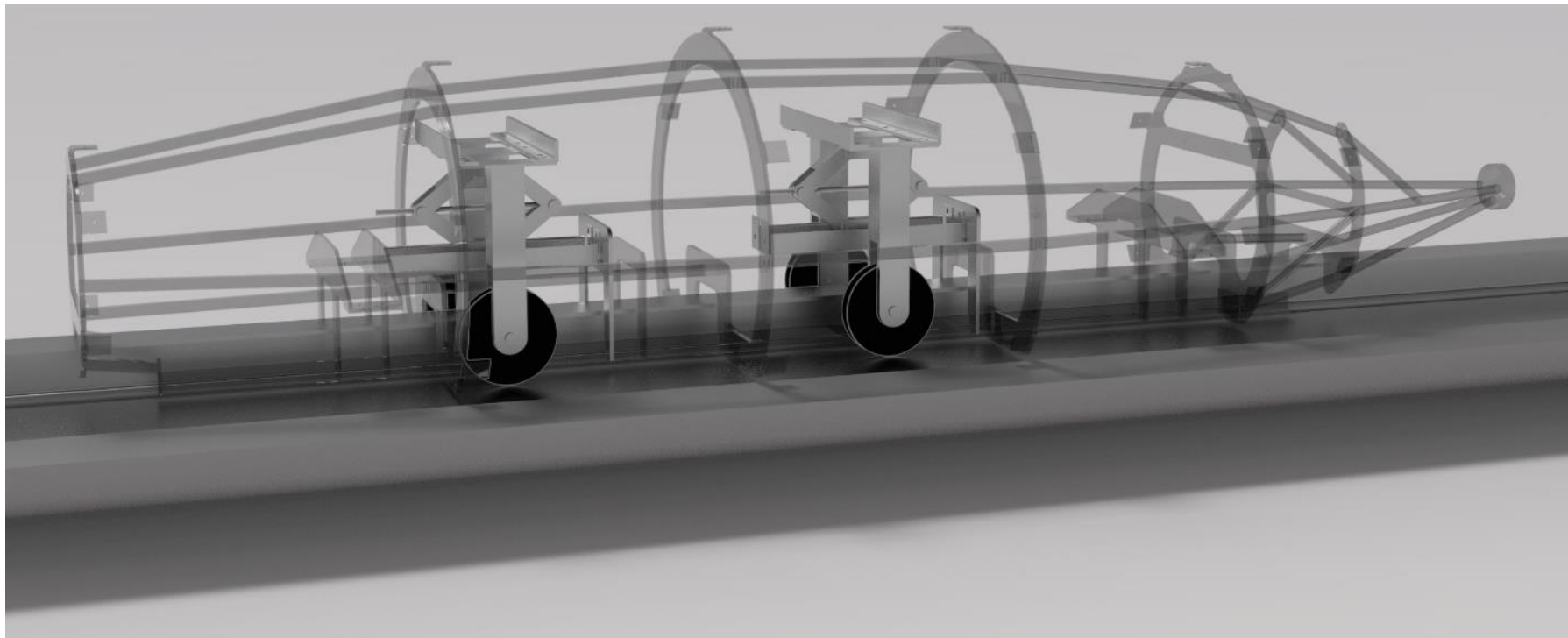
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SERVICE WHEELS

- Powered service wheels for transport and potential pod recovery
- Wheels extend 1/8" below hover engines
- Motorized rear-left support wheel



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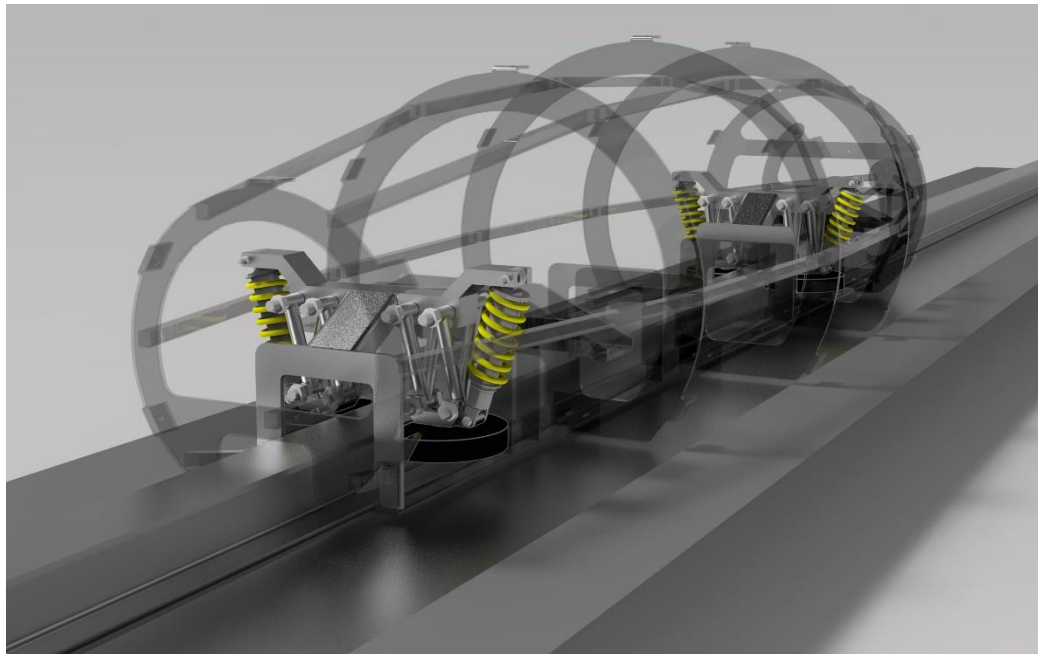
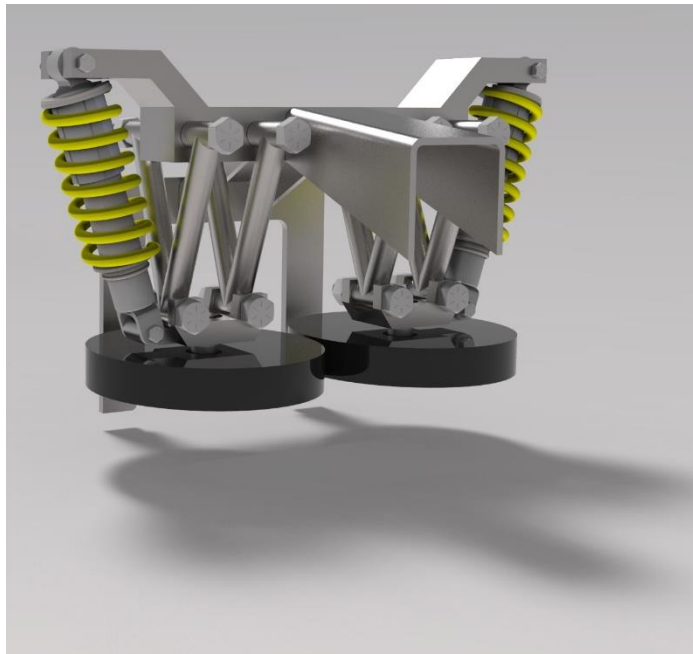
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I-BEAM STABILIZATION

- Benchmarked from roller coaster design and car/motorcycle suspension systems
- Spring-damper resists movement from the parallel linkage
 - Handles lateral forces



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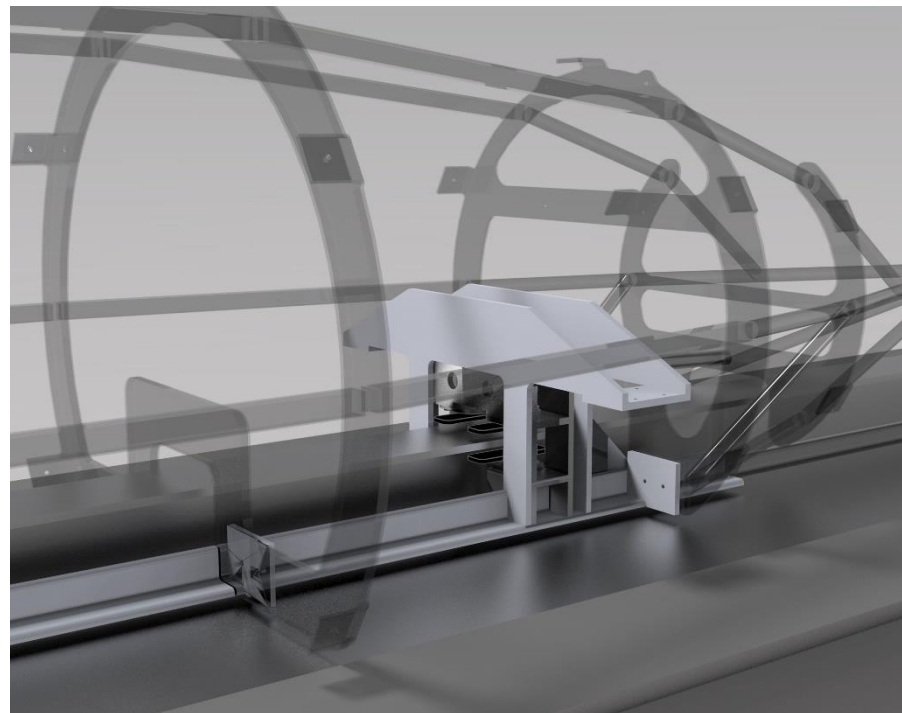
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BRAKING

- Pneumatic braking assembly with four actuated brake pads
 - Brake pads clamp onto the I-beam
 - Located at rear of pod



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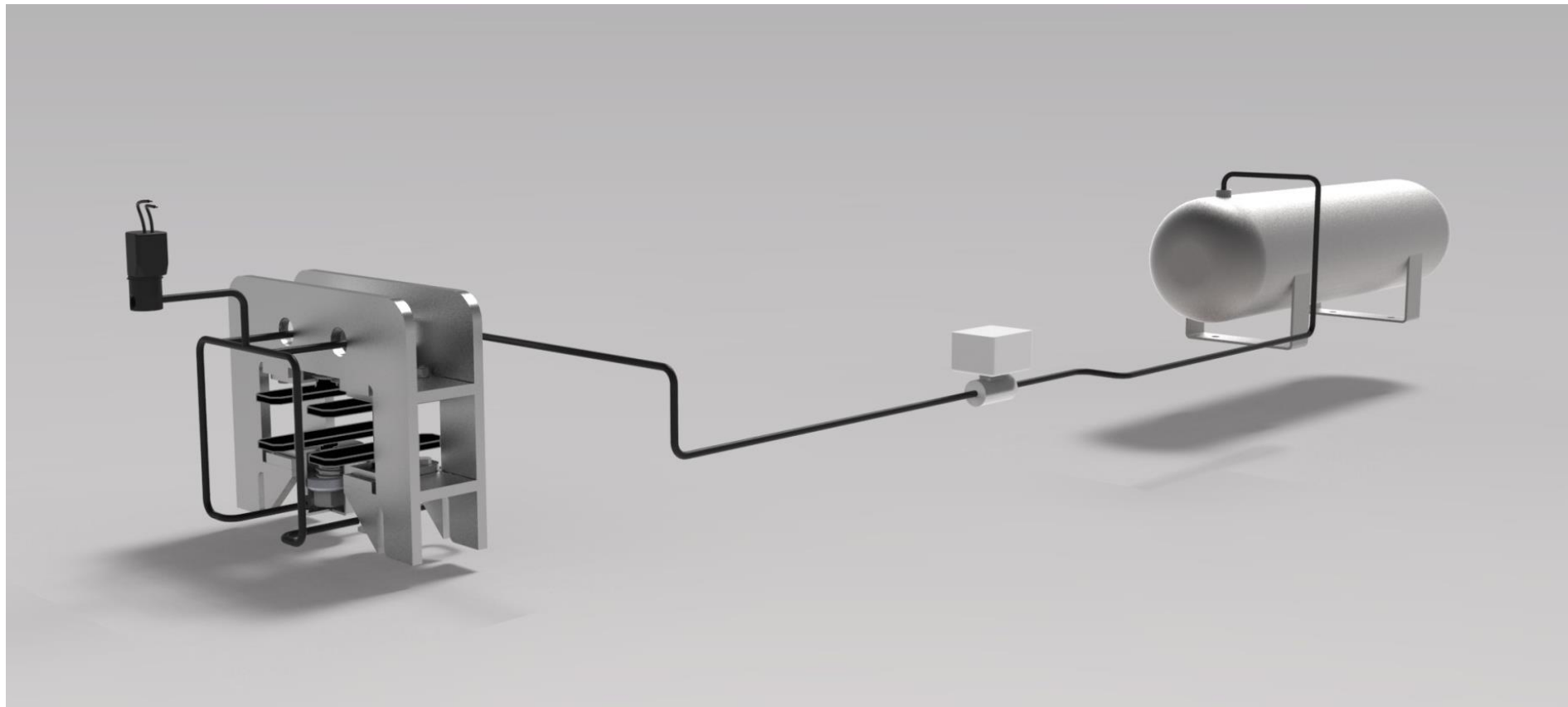
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BRAKING

- Pressurized air tank provides pneumatic brake force



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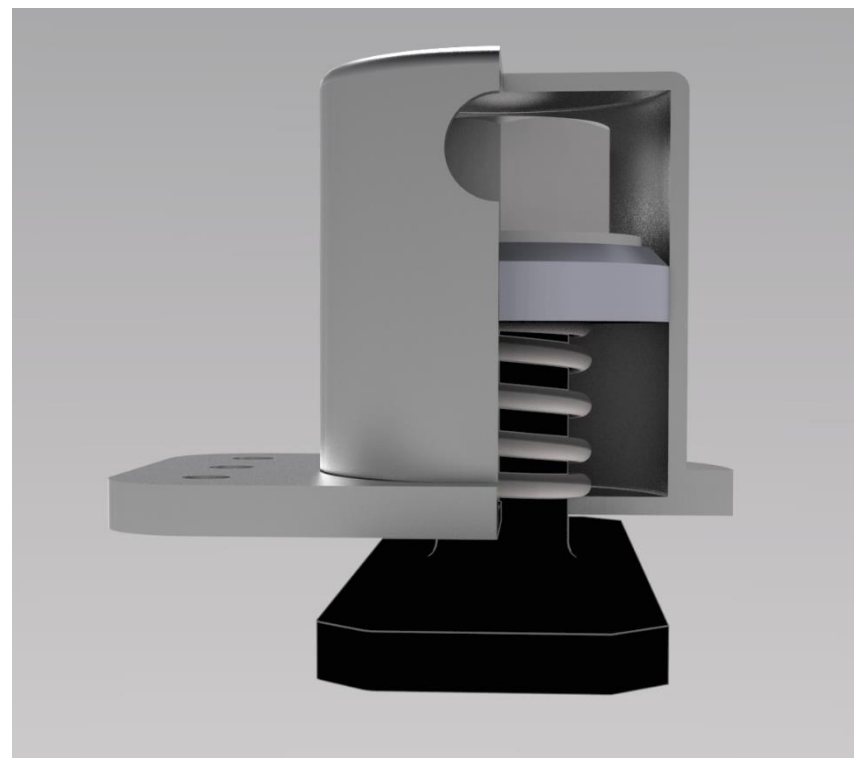
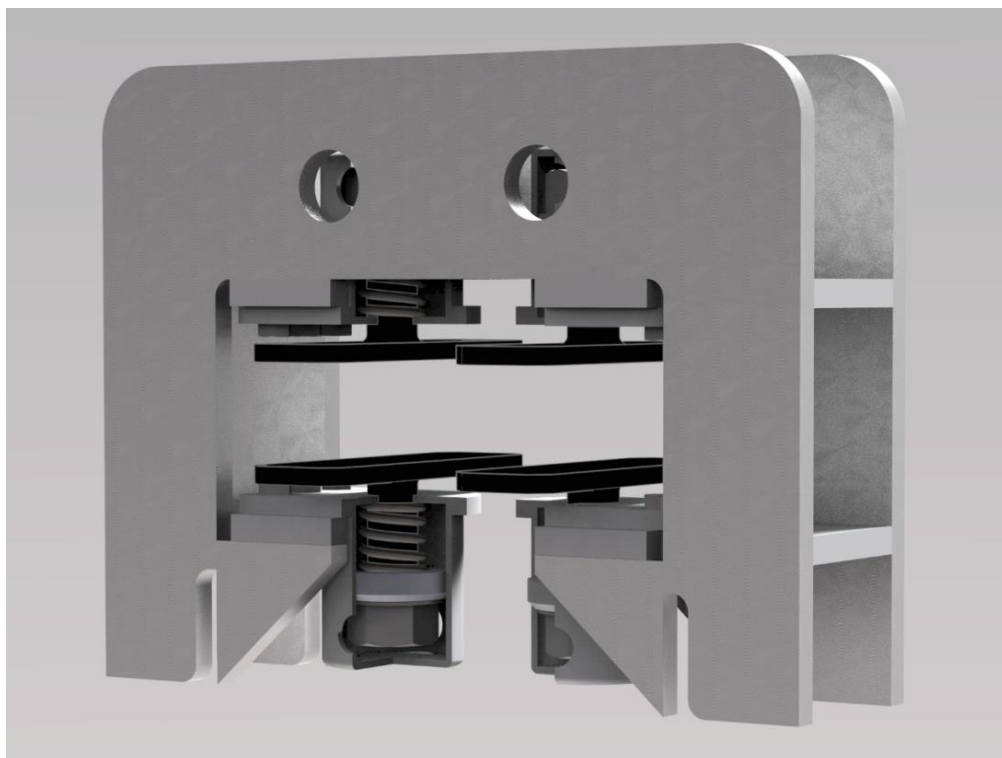
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BRAKING

- Braking automatically activated by solenoid valves if power fails
- Ball valve manually disengages the brakes



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POD WEIGHT

Subsystem	Weight
Frame	66 lbs
Shell	83 lbs
Service propulsion wheels	96 lbs
I-Beam Stabilization	60 lbs
Braking	27 lbs
Maglev Engines	60 lbs
Battery and electronics	63 lbs
Total Weight	455 lbs

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MAGLEV LEVITATION

- System utilizes four Arx Pax Magnetic Field Architecture (MFA) hover engines
- Ground clearance
 - 0.20" (5mm)
- Four engine payload
 - 550 lbs



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MAGLEV LEVITATION

- Best chance of success for competition while still adhering to the future scalability of the Hyperloop
 - Levitation + Propulsion + Braking + Control
 - Operate at high speeds and in low-pressure environments



Arx Pax HE3.0 Hover Engine

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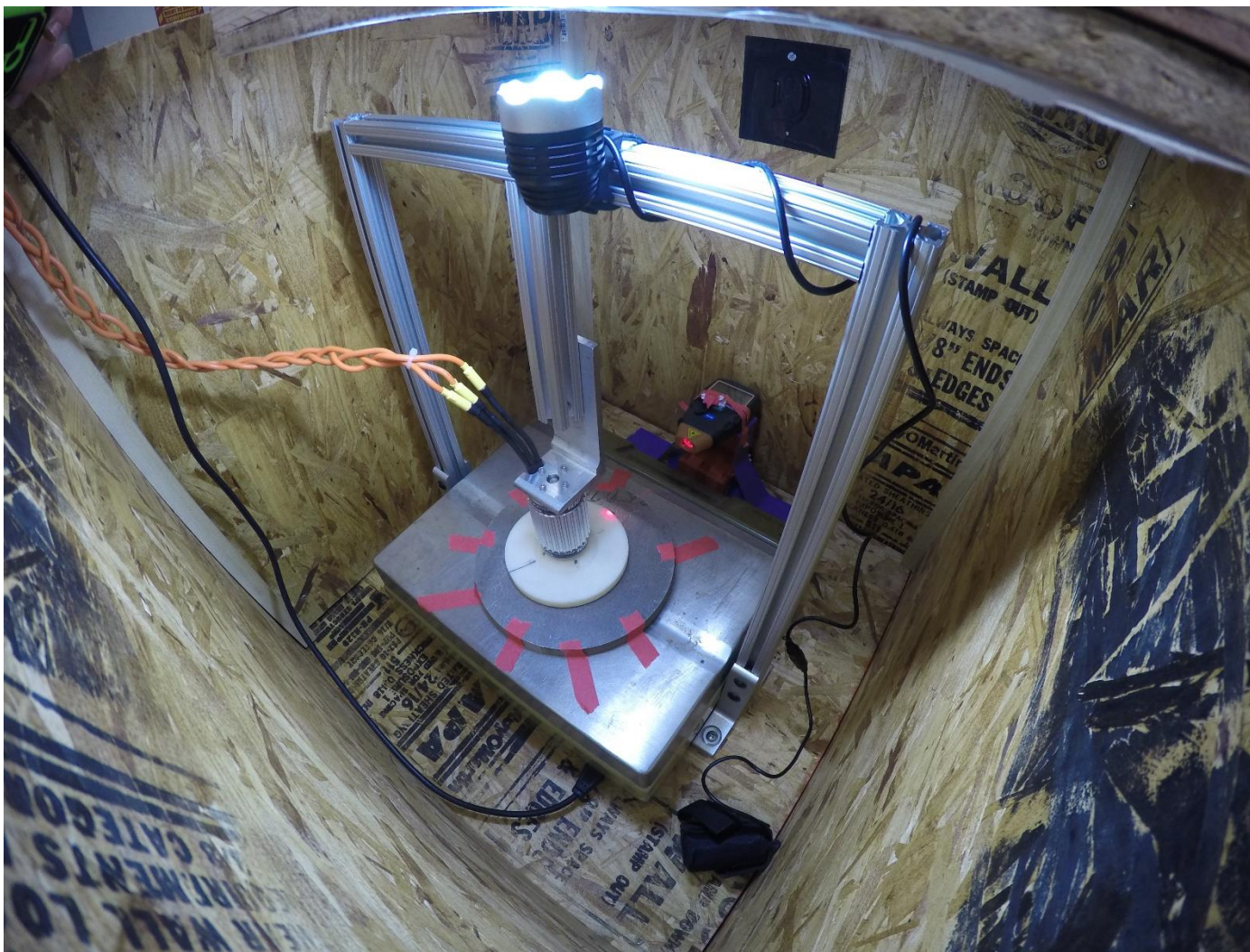
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MAGLEV PROTOTYPING



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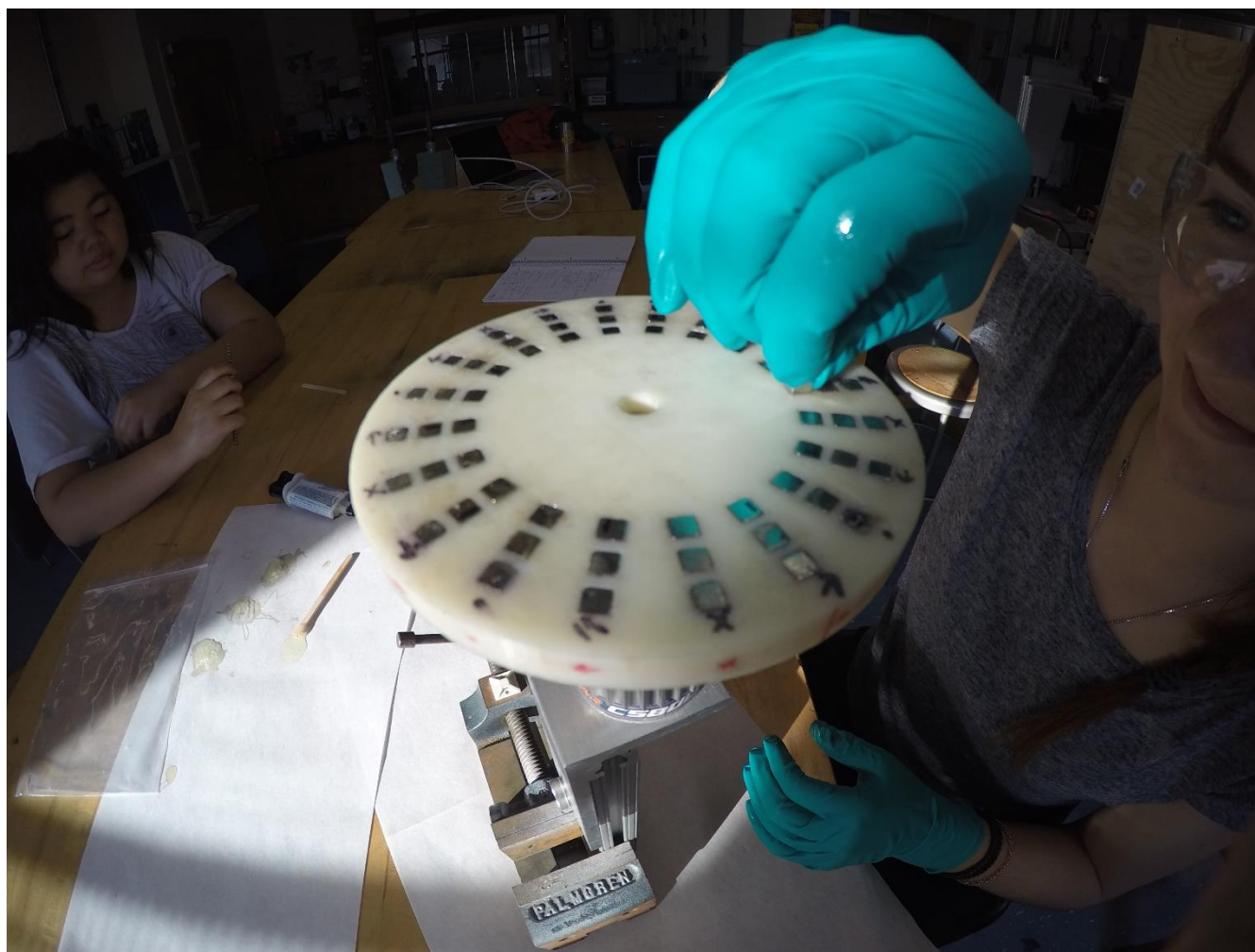
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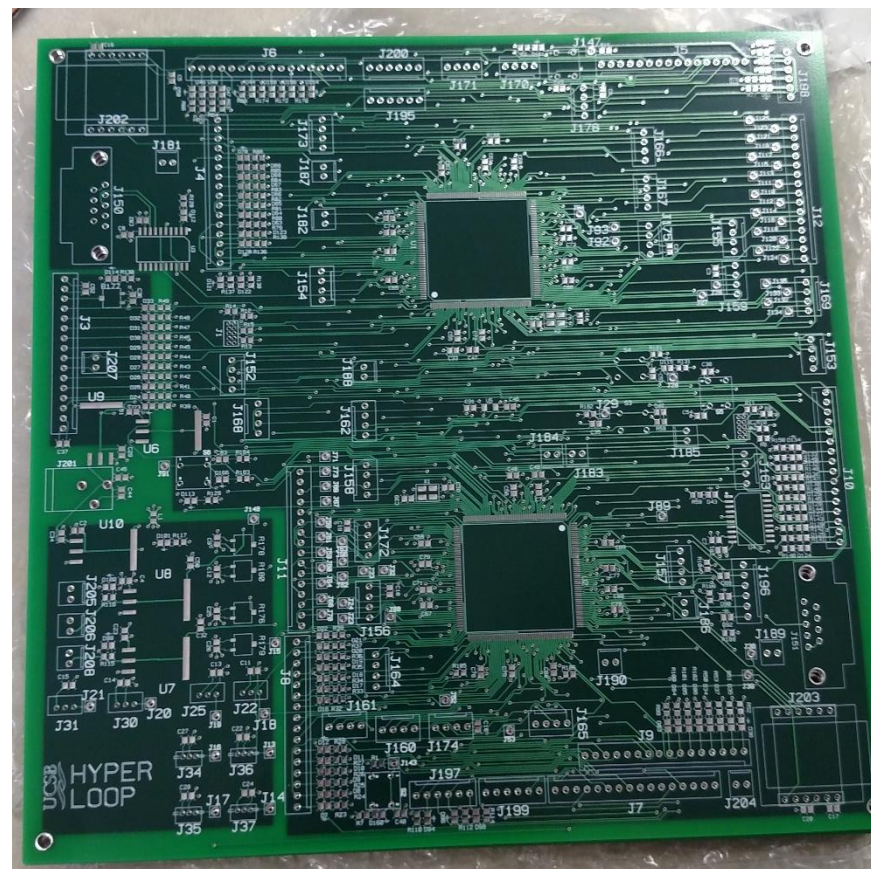
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PRINTED CIRCUIT BOARD

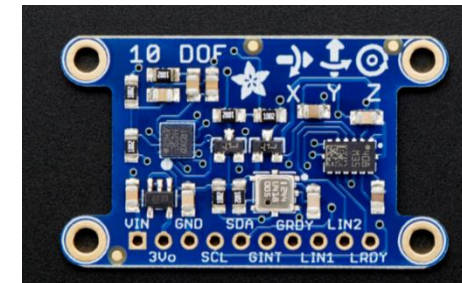
- 8.1" x 8.1" printed circuit board
 - (2) LPC NXP4088 microcontrollers
 - Actuate maglev engines, braking, and service wheels
 - Interface with sensors
 - Communicate telemetry at 1 Hz
 - Interfaces with SpaceX Network Access Panel
 - Execute pod-stop command
 - Manage control systems
- Already fabricated and awaiting assembly



SENSORS

- **Consolidated Board:** *Adafruit 10-DOF IMU Breakout (2)*

- Positioned at front and back of the pod
- Combines four sensors:
 - Accelerometer
 - Gyroscope
 - Barometer
 - Thermometer



- **Photoelectric sensor:** *Omron E3FB-DP13 2M (2)*

- Telemetry and navigation
- Positioned at top of pod
- Detects reflective strips on the top half of the tube



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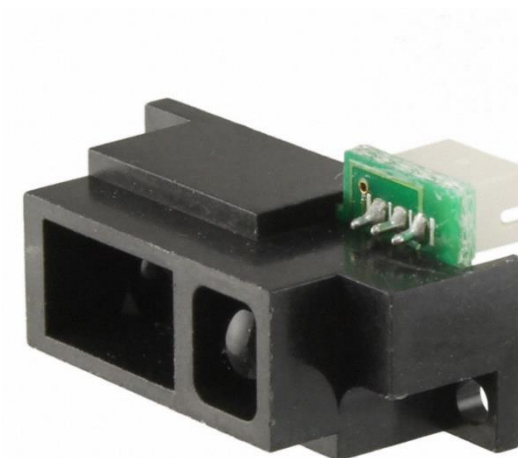
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SENSORS

- **Short range:** *Sharp GP2Y0A51SK0F* (4)
 - Telemetry and stability system
 - Positioned at four bottom corners of pod
 - Determines height relative to the bottom of the tube
- **Long range:** *Sharp GP2Y0A02YK* (4)
 - Telemetry
 - Positioned at pod's left and right sides
 - Gives position relative to sides of the tube



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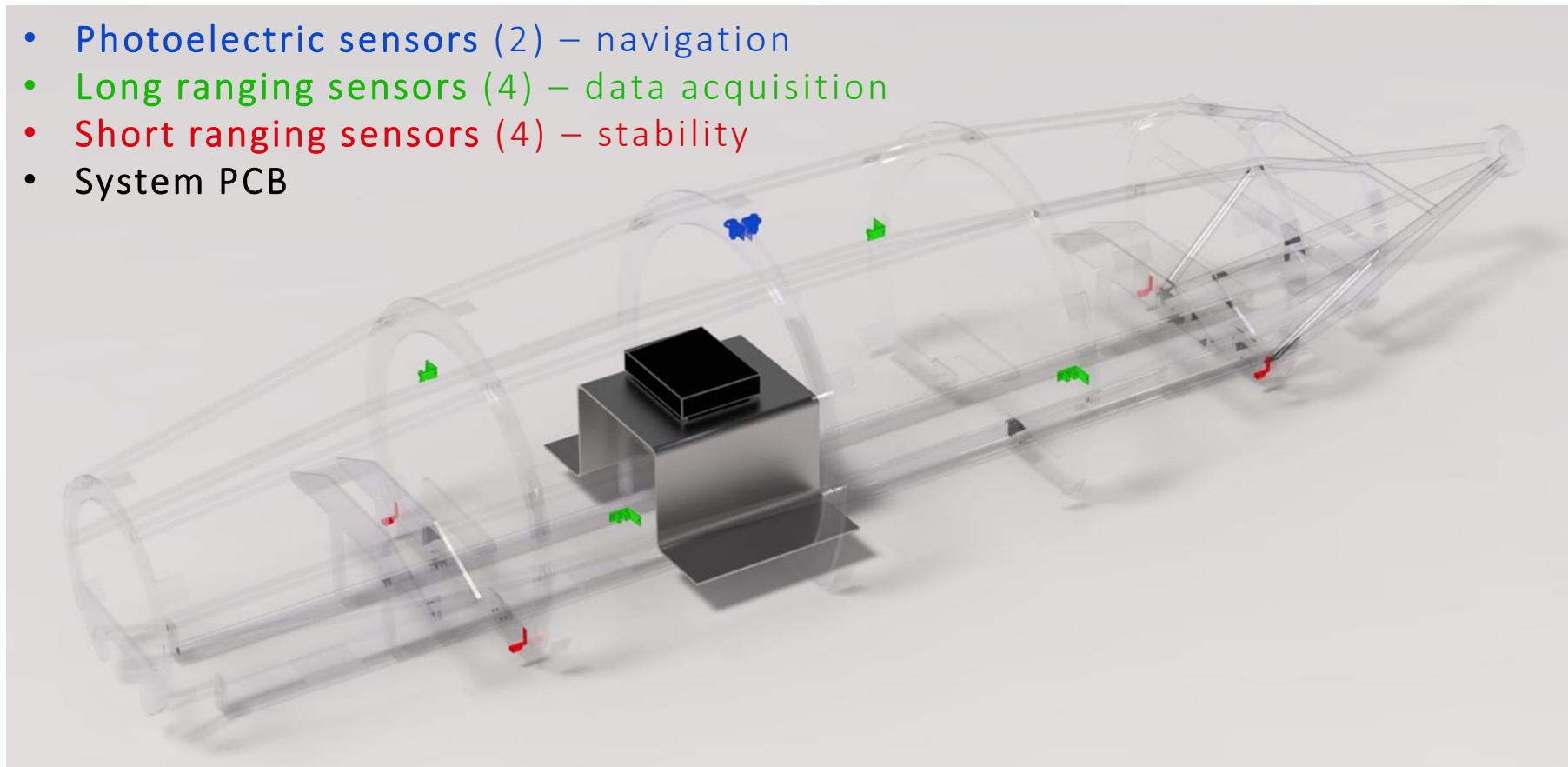
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SENSOR LOCATIONS

- Photoelectric sensors (2) – navigation
- Long ranging sensors (4) – data acquisition
- Short ranging sensors (4) – stability
- System PCB



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PROTOTYPING

- Currently prototyping sensors on LPC NXP4088 Developer's Kit
- Testing cabling constraints on a full-sized model



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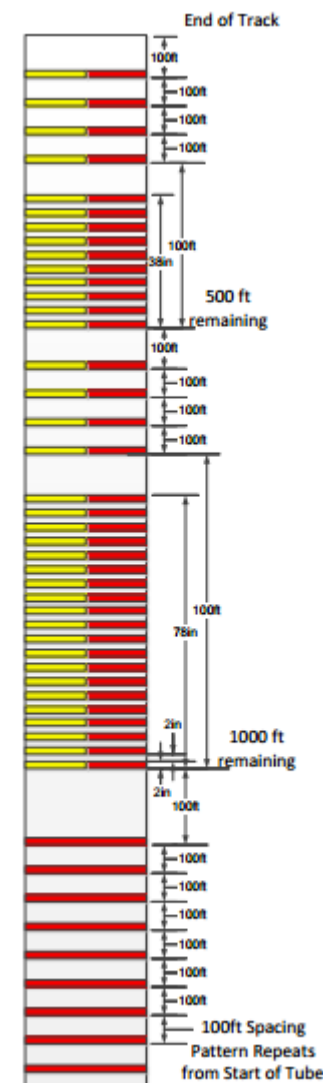
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CONTROL SYSTEMS

- Navigation
 - Double integrate accelerometer
 - Recalibrate with diffuse-reflective photoelectric sensor
- Stability
 - 2 gyroscopes at front and back of pod
 - 4 ranging sensors at bottom corners of pod
 - Correct each engine's levitation to maintain stability and correct disturbances
- Braking
 - Microcontroller activates/disengages brake system solenoid valve to apply/release brakes
 - Automatically applied if connection lost for >5 secs



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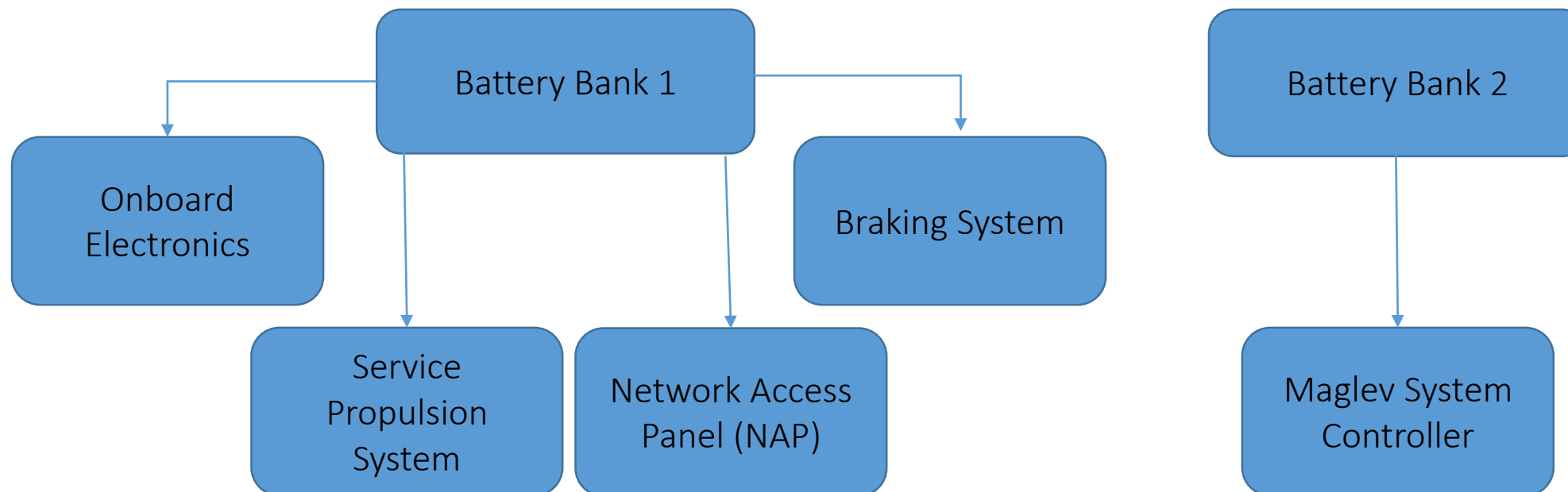
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POWER SYSTEM

- **Battery Bank 1**
 - Onboard electronics, NAP, braking, service propulsion
- **Battery Bank 2**
 - Magnetic levitation system



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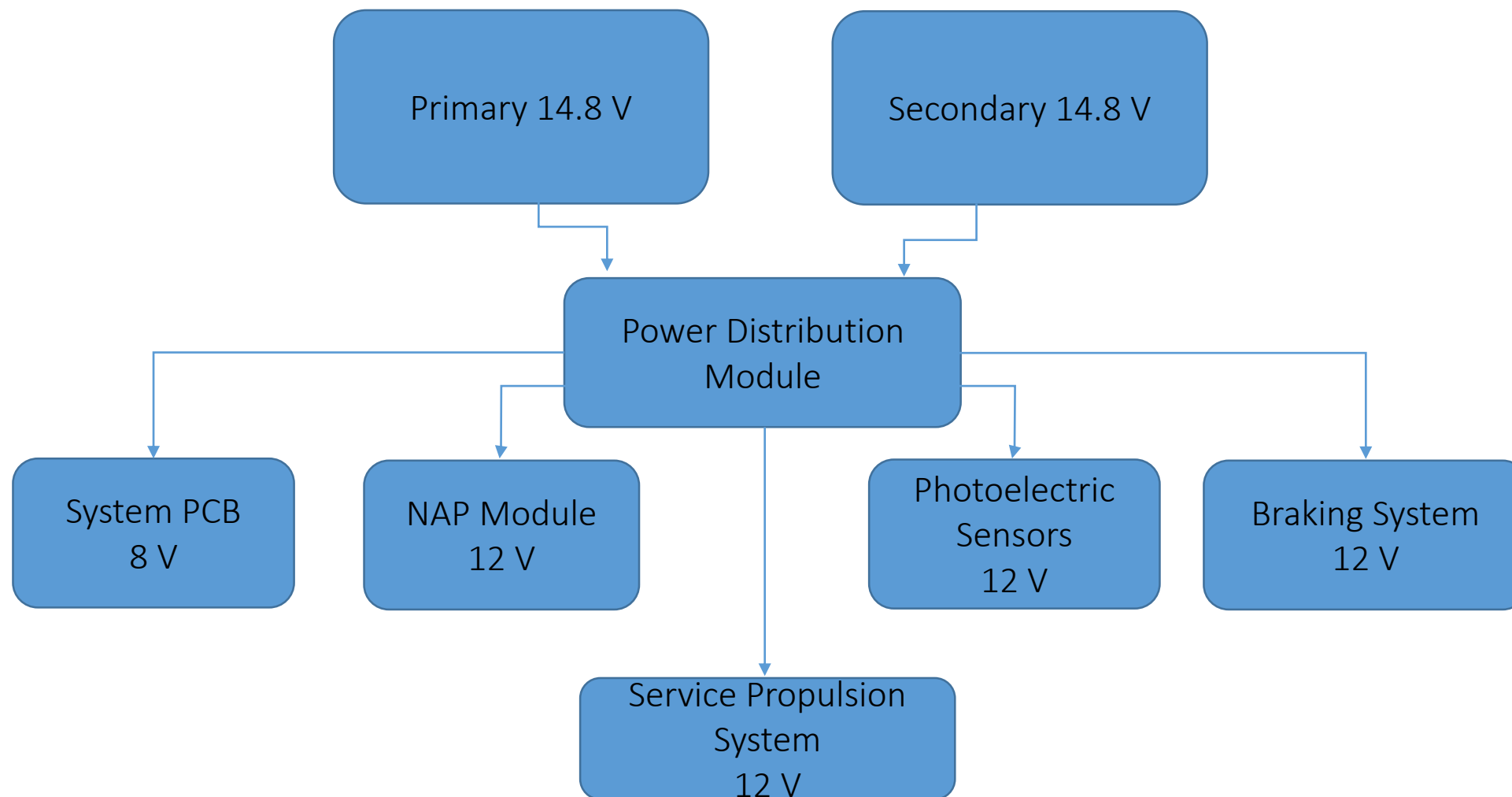
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BATTERY BANK 1



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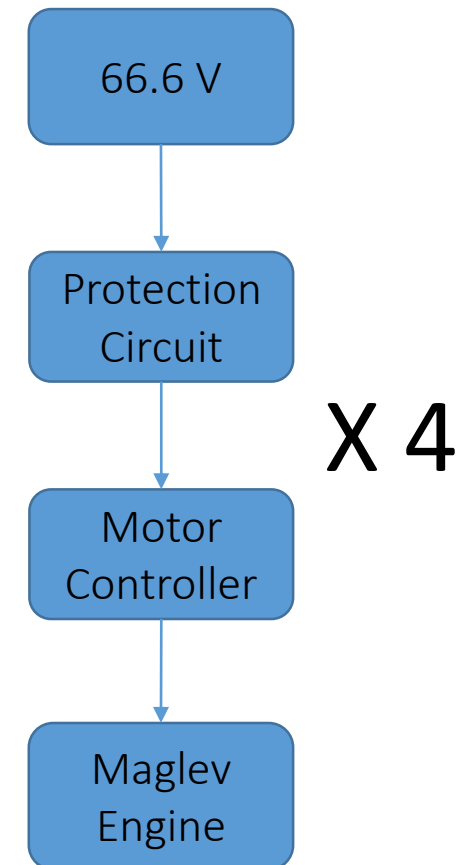
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BATTERY BANK 2

- Independent power supply for each engine
 - 20Ahr total capacity for 26.2 min of levitation
- Protection circuit to prevent battery over-discharge
- Motor controller communicates with System PCB



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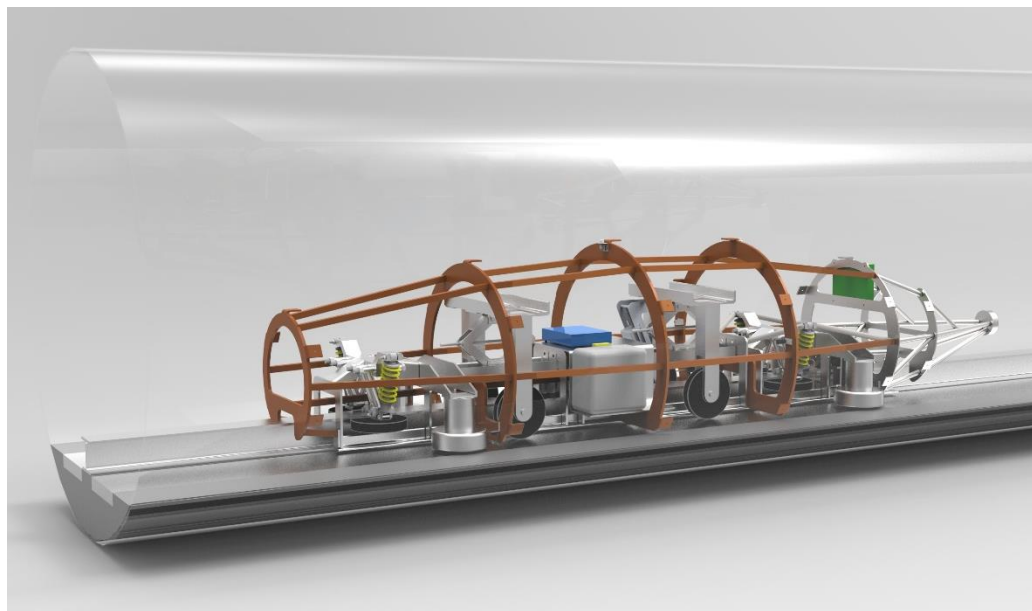
BATTERY SELECTION

- Lithium polymer (LiPo)
 - Battery of choice for quadcopter/drone applications
 - Optimal capacity/weight ratio
 - High discharge rating
 - Easily obtainable off-shelf packs
 - Fire safety
 - Fireproof bags encase each battery bank
 - Stainless steel LiPo charge box to prevent battery puncture in the event of a crash



SCALABILITY

- 1/5 of pod weight (87 lbs) relates to the I-Beam
 - I-Beam stability assembly (60 lbs)
 - Pneumatic brakes (27 lbs)
- Maglev engines vs air bearings
 - Tilting the maglev engines can provide braking and propulsion
 - Highly scalable controls software architecture



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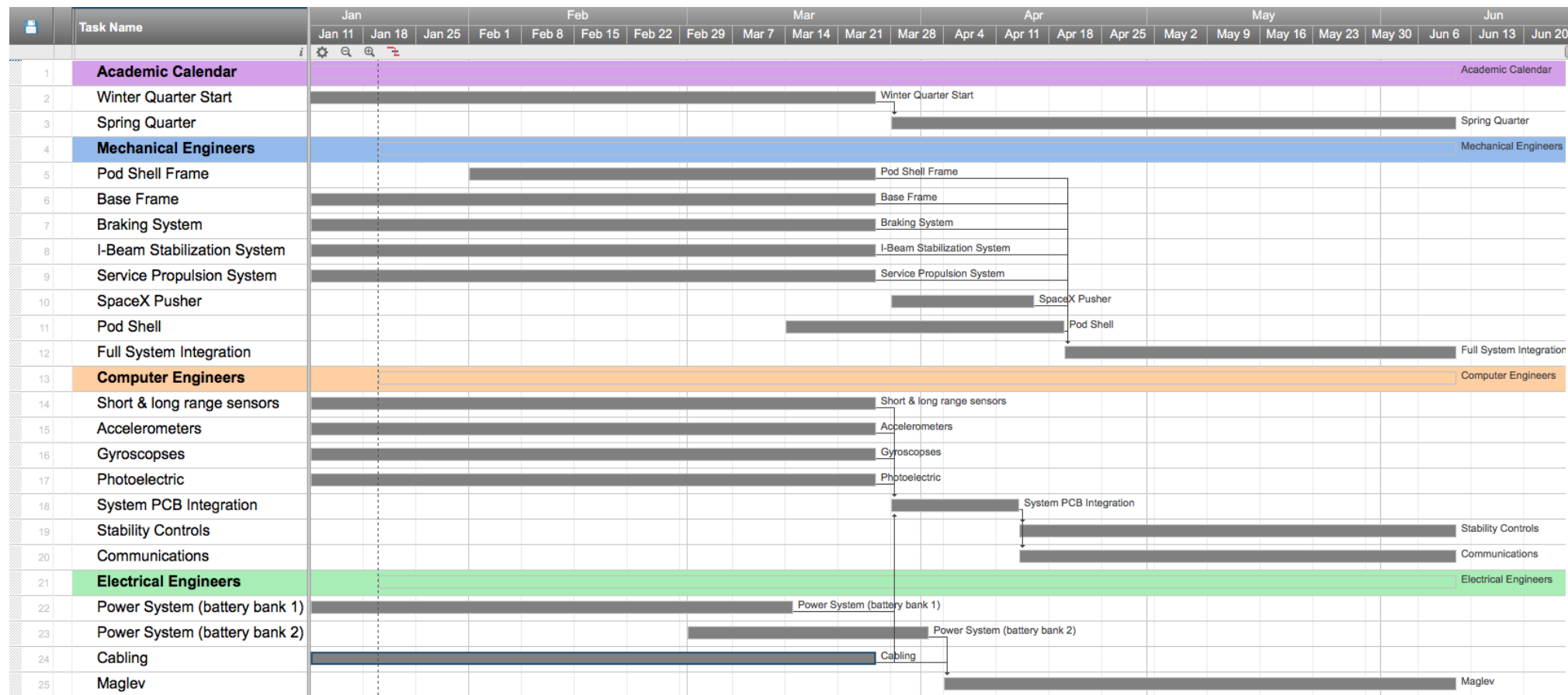
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PRODUCTION SCHEDULE



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CONCLUSION

- Work began in October 2015
 - With school schedules—effectively 12 weeks of work
- Accomplishments
 - **Raised \$15,000—need \$25,000 to reach goal of \$40,000**
 - Printed Circuit Board is in assembly
 - Prototyping sensors with NXP Developer's Kit
 - 3D printed model
 - Used for extensive wind tunnel testing
 - Magnetic levitation testing & prototyping
 - Styrofoam and PVC pipe model of frame completed
 - Beginning to work with cabling
 - Established a finance/marketing team

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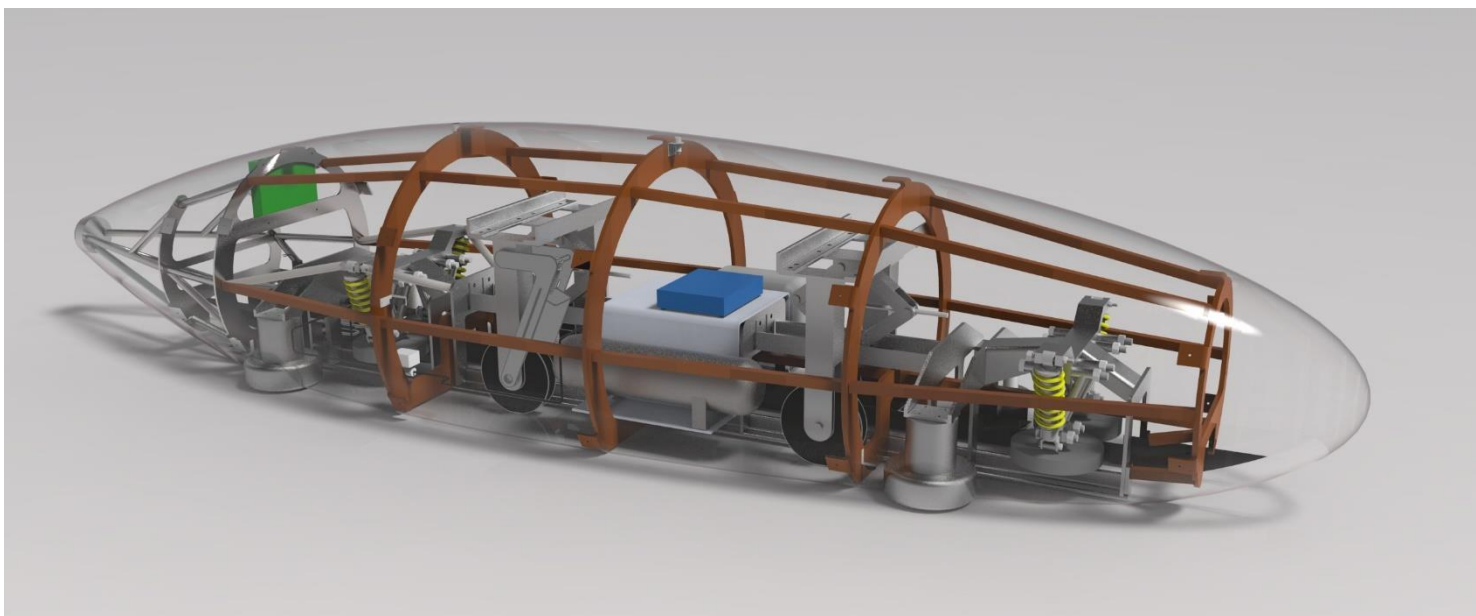
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CONCLUSION

- Thank you to our mentors and our sponsors.
- Please find us at Booth 64 or contact us:
 - ucsbhyperloop.com
 - ucsbhyperloop@gmail.com
 - @UCSBHyperloop

Raytheon

IR *Ingersoll Rand*



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