**4.2** Consider a 10<sup>4</sup> kg spaceship in deep space. Suppose the spaceship's thruster produces 100 N of constant thrust. If the spaceship is moving at 50 m/s and the thruster is anti-aligned with the spaceship's velocity, how long should the thruster fire to bring the spaceship to rest with respect to absolute space?

Final

$$a = \frac{F_{throst}}{m}$$
 $a = \frac{F_{throst}}{m}$ 
 $a = \frac{mv_0 = 0}{m}$ 
 $a = \frac{mv_0 = 0}{m}$ 

**4.6** Suppose frame  $\mathcal{B} = (O', \mathbf{b}_1, \mathbf{b}_2, \mathbf{b}_3)$  is traveling in the  $\mathbf{e}_1$  direction at a constant speed of  $v_0$  with respect to stationary frame  $\mathcal{I} = (O, \mathbf{e}_1, \mathbf{e}_2, \mathbf{e}_3)$ , as shown in Figure 4.19. Mass  $m_P$  is connected to point O' by a spring with spring constant k and rest length  $r_0$ . The spring can freely pivot about O'. Assume that the positions of O and O' are the same at time t = 0.

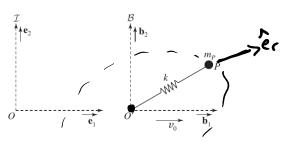


Figure 4.19 Problem 4.6.

- a. Using the coordinates of your choice, find the position  $\mathbf{r}_{P/O'}$  and velocity  ${}^{\mathcal{I}}\mathbf{v}_{P/O'}$  of the mass with respect to O' in  $\mathcal{I}$ . [HINT: Introduce a polar frame at O'.]
- b. Find the position  $\mathbf{r}_{P/O}$  and velocity  ${}^{\mathcal{I}}\mathbf{v}_{P/O}$  of the mass with respect to O in  $\mathcal{I}$ .
- c. Draw a free-body diagram for mass  $m_P$ . (There is no gravity in this problem.)
- d. Find the angular momentum  ${}^{\mathcal{I}}\mathbf{h}_{P/O'}$  of the mass with respect to O' in  $\mathcal{I}$ .
- e. Show that the angular momentum of the mass with respect to O' in  $\mathcal{I}$  is conserved, but the angular momentum with respect to O in  $\mathcal{I}$  is not.

$$\sum M_{o'} = \frac{d}{dt} \left( \overline{h}_{o'} \right) = \frac{d}{dt} \left( \overline{r}_{l'o'} \times m \overline{v}_{l'} \right)$$

$$\overline{r}_{l'o'} \times m \overline{v}_{l'} = r e_{l'} \times m \left( r e_{l'o'} \times m \overline{v}_{l'} \right)$$

$$= m r e_{l'} \times e_{l'} + m r^{2} e_{l'} \cdot e_{l'} \cdot e_{l'}$$

$$= m r^{2} e_{l'} \cdot e_$$

$$r_0^2 \dot{\theta}_0 = \frac{1}{\sqrt{2}} r_0^2 \dot{\theta}_0 \implies \dot{\theta} = \frac{r_0^2 \dot{\theta}_0}{r_0^2}$$

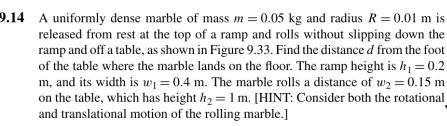
$$\overline{ZM_o} = \overline{r_{0/o}} \times \overline{F_s} = (\overline{r_{0/o}} + \overline{r_{0/o}}) \times -kr\hat{e}_r$$

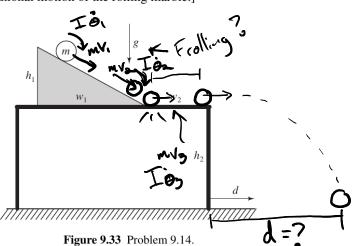
$$\overline{ZM_o} = \overline{r_{0/o}} \times -kr\hat{e}_r$$

$$\overline{ZM_o} = (vt\hat{b}_1) \times (-kr\hat{e}_r)$$

$$= (vt\hat{b}_1) \times (-kr)(\cos\theta\hat{b}_1 + \sin\theta\hat{b}_2)$$

$$\overline{ZM_o} = -krvt\sin\theta\hat{b}_3$$





$$e = \frac{-v'}{v_0}$$

$$O = \frac{-v'}{v_0}$$
  $\frac{1}{2}mv_0^2 + w = 0$   $w = -\frac{1}{2}mv_0^2$ 

$$mv_x + Fat = 0$$
 $mv_x + 0 = m\overline{v_3}$ 

