
```
read_data
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```
model1 = polyfit(time1,altitude1,3);
model2 = polyfit(time2,altitude2,3);
model3 = polyfit(time3,altitude3,3);
model4 = polyfit(time4,altitude4,3);
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m_altitude1 = model1(1)*time1.^3 + model1(2)*time1.^2 + model1(3)*time1 +
    model1(4);
m_altitude2 = model2(1)*time2.^3 + model2(2)*time2.^2 + model2(3)*time2 +
    model2(4);
m_altitude3 = model3(1)*time3.^3 + model3(2)*time3.^2 + model3(3)*time3 +
    model3(4);
m_altitude4 = model4(1)*time4.^3 + model4(2)*time4.^2 + model4(3)*time4 +
    model4(4);
```

```
figure % figure 1
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```
scatter(time1,altitude1,'g')
hold on
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```
scatter(time2,altitude2,'r')
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```
scatter(time3,altitude3,'b')
```

```
scatter(time4,altitude4,'c')
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```
plot(time1,m_altitude1,'k','LineWidth',2)
plot(time2,m_altitude2,'k','LineWidth',2)
plot(time3,m_altitude3,'k','LineWidth',2)
plot(time4,m_altitude4,'k','LineWidth',2)
```

```
hold off
```

```
legend('TL 22','CL 22','TL 23','CL 23','Location','Southeast','FontSize',16)
xlabel('Time (s)','FontSize',16)
ylabel('Altitude (ft)','FontSize',16)
title('Post-Burnout RRC3 Altitude Data for CRT Flights','FontSize',20)
%legend('Test Launch 22','Competition Launch 22','Test Launch 23','Competition
    Launch 23','Location','southeast','FontSize',16)
```

```
hold off
```

```
vmode11 = [3*model1(1) 2*model1(2) model1(3)];
vmode12 = [3*model2(1) 2*model2(2) model2(3)];
vmode13 = [3*model3(1) 2*model3(2) model3(3)];
vmode14 = [3*model4(1) 2*model4(2) model4(3)];
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m_velocity1 = vmodel1(1)*time1.^2 + vmodel1(2)*time1 + vmodel1(3);
m_velocity2 = vmodel2(1)*time2.^2 + vmodel2(2)*time2 + vmodel2(3);
m_velocity3 = vmodel3(1)*time3.^2 + vmodel3(2)*time3 + vmodel3(3);
m_velocity4 = vmodel4(1)*time4.^2 + vmodel4(2)*time4 + vmodel4(3);

figure % figure 2
plot(time1,m_velocity1,'g','LineWidth',2)

hold on
plot(time2,m_velocity2,'r','LineWidth',2)
plot(time3,m_velocity3,'b','LineWidth',2)
plot(time4,m_velocity4,'c','LineWidth',2)

xlabel('Time into Flight (s)')
ylabel('Velocity (ft/s)')
title('Velocity v. Time into Flight for CRT Flights','FontSize',16)
legend('Test Launch 22','Competition Launch 22','Test Launch 23','Competition
Launch 23','Location','northeast','FontSize',12)

hold off

R = 287; % ideal gas constant of air
g0 = 9.81; % gravitational constant
a = -0.0065;
rho_ref = 1.225;
T_ref = 288.15;

% densities for altitudes
rho1 = (1/16.0185)*(1+(a*(altitude1*0.3048)/T_ref)).^((-g0/(a*R))-1); % ft/
lb^3
rho2 = (1/16.0185)*(1+(a*(altitude2*0.3048)/T_ref)).^((-g0/(a*R))-1);
rho3 = (1/16.0185)*(1+(a*(altitude3*0.3048)/T_ref)).^((-g0/(a*R))-1);
rho4 = (1/16.0185)*(1+(a*(altitude4*0.3048)/T_ref)).^((-g0/(a*R))-1);

% distance to apogee
d1 = 10700 - altitude1; % ft
d2 = 10345 - altitude2;
d3 = 12315 - altitude3;
d4 = 10067 - altitude4;

TL_t = [time1 time3]; % single matrix for test launch times
CL_t = [time2 time4]; % single matrix for comp launch times
TL_v = [m_velocity1 m_velocity3]; % single matrix for test launch velocities
CL_v = [m_velocity2 m_velocity4]; % single matrix for competition launch
velocities
TL_d = [d1 d3]; % single matrix for test launch distances to apogee
CL_d = [d2 d4]; % single matrix for comp launch distances to apogee
TL_a = [m_altitude1 m_altitude3];
CL_a = [m_altitude1 m_altitude3];
TL_rrc3v = [rrc3_v1 rrc3_v3]; % filtered velocity from test launch rrc3's
CL_rrc3v = [rrc3_v2 rrc3_v4]; % filtered velocity from comp launch rrc3's

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```

v = linspace(0,750,750*10);

TL_model = polyfit(TL_v,TL_d,4);
CL_model = polyfit(CL_v,CL_d,4);

TL_model_prediction = run_model(TL_model,v); % test launch model curve for
graph
CL_model_prediction = run_model(CL_model,v); % competition launch model curve
for graph

TL_model_prediction = TL_model(1)*v.^4 + TL_model(2)*v.^3 + TL_model(3)*v.^2+
TL_model(4)*v + TL_model(5);
CL_model_prediction = CL_model(1)*v.^4 + CL_model(2)*v.^3 + CL_model(3)*v.^2+
CL_model(4)*v + CL_model(5);

figure % figure 3
scatter(m_velocity1,d1,'g','o')
hold on
scatter(m_velocity2,d2,'r','o')
scatter(m_velocity3,d3,'b','*')
scatter(m_velocity4,d4,'c','*')
plot(v,TL_model_prediction,'k')
plot(v,CL_model_prediction,'k')

xlabel('Velocity (ft/s)','FontSize',16)
ylabel('Distance to Apogee (ft)','FontSize',16)
title('Velocity v. Distance to Apogee','FontSize',20)
legend('Test Launch 22','Competition Launch 22','Test Launch 23','Competition
Launch 23','Location','southeast','FontSize',12)

hold off

%TLMP = TL_model(1)*TL_v.^3 + TL_model(2)*TL_v.^2+ TL_model(3)*TL_v +
TL_model(4); % predictions for every TL velocity data point
%CLMP = CL_model(1)*CL_v.^3 + CL_model(2)*CL_v.^2+ CL_model(3)*CL_v +
CL_model(4); % predictions for every CL velocity data point
%TLMP = TL_model(1)*TL_v.^4 + TL_model(2)*TL_v.^3 + TL_model(3)*TL_v.^2+
TL_model(4)*TL_v + TL_model(5);
%CLMP = CL_model(1)*CL_v.^4 + CL_model(2)*CL_v.^3 + CL_model(3)*CL_v.^2+
CL_model(4)*CL_v + CL_model(5);

v1 = [];
v2 = [];
v3 = [];
v4 = [];

% obtain actual velocities from RRC3 rather than ones from derivative of
% fit
for i = 1:(length(time1)-1)
    v1(i) = (altitudel(i+1) - altitudel(i))/0.05;
end

for i = 1:(length(time2)-1)
    v2(i) = (altitude2(i+1) - altitude2(i))/0.05;

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end

for i = 1:(length(time3)-1)
    v3(i) = (altitude3(i+1) - altitude3(i))/0.05;
end

for i = 1:(length(time4)-1)
    v4(i) = (altitude4(i+1) - altitude4(i))/0.05;
end

TL_vexp = [v1 v3];
CL_vexp = [v2 v4];

TLMP = TL_model(1)*TL_rrc3v.^4 + TL_model(2)*TL_rrc3v.^3 +
    TL_model(3)*TL_rrc3v.^2 + TL_model(4)*TL_rrc3v + TL_model(5);
CLMP = CL_model(1)*CL_rrc3v.^4 + CL_model(2)*CL_rrc3v.^3 +
    CL_model(3)*CL_rrc3v.^2 + CL_model(4)*CL_rrc3v + CL_model(5);

TL_residuals = TL_d - TLMP;
CL_residuals = CL_d - CLMP;

Altitude_residuals = altitudel - m_altitudel;

figure
scatter(TL_v,TL_residuals)
xlabel('Velocity (ft)')
ylabel('Actual Distance to Apogee - Predicted Distance to Apogee (ft)')
title('Residuals v. Live Velocity','FontSize',20)

figure
scatter(TL_rrc3v,TL_v)
xlabel('Filtered RRC3 Velocity Readings (ft/s)')
ylabel('Modeled Velocity (ft/s)')

function model_prediction = run_model(model,data)
    model_prediction = model(1)*data.^3 + model(2)*data.^2 + model(1)*data
    +model(4);

end

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