In the last lecture, using the physical considerations of axisymmetry, no warping/bulging of c/s, axial homogeneity, and torsion, we obtain simplied equilibrium eqns and displacement functions.

Solving the simplified equations

As u_r and u_z are functions of r and z, respectively, we can rewrite σ_{zz} as

$$\sigma_{zz} = \lambda \left(u_s' + \frac{u_r}{r} + u_z' \right) + 2\mu u_z'$$

$$= \left(\lambda + 2\mu \right) u_z' + \lambda \left(u_r' + \frac{u_r}{r} \right)$$
depends
$$u_{pon} z \qquad u_{pon} r$$

But from one of the equilibrium equations, we have that $\frac{\partial \sigma_{zz}}{\partial z} = 0 \implies \sigma_{zz} \text{ does not depend upon } z$

$$\Rightarrow$$
 $\sigma_{zz} = f(r)$ } must be a function of r only

.. The term uz must be a constant

$$\Rightarrow$$
 $U_z = C_1 z + C_2$

Also, $U_z = 0$ at the cylinder's mid-section (z=0) [disp BC] Thereforce, $c_z = 0 \implies u_z = c_1 z \leftarrow expression$ for u_z

Expression for ur

We now look at the radial component of equilibrium equation

The expressions of our and too become

$$\overline{\sigma_{rr}} = \lambda \left(u_r' + \frac{u_r}{r} + u_z' \right) + 2\mu u_r' \qquad \left[U_{se} \ u_z = C_1 z \right]$$

$$= \lambda \left(u_r' + \frac{u_r}{r} + C_1 \right) + 2\mu u_r'$$

$$\sqrt[5]{6} = \lambda \left(u_r' + \frac{u_r}{r} + c_1 \right) + 2\mu \frac{u_r}{r}$$

We will make use of the simplified equilibrium eqn,

$$\frac{3r}{9a^{2}} + \frac{\lambda}{a^{2} - a^{2}} = 0$$

Subtracting orr - too, we get

$$\sigma_{rr} - \sigma_{ao} = 2\mu \left(u_r' - \frac{u_r}{r} \right)$$

The partial derivative or can be obtained as:

$$\frac{\partial \sigma_{rr}}{\partial r} = \lambda \frac{d}{dr} \left(u_r' + \frac{u_r}{r} \right) + 2 u u_r'' \quad \left[:: c_1 \text{ is constant} \right]$$

Plugging these values into the equilibrium eqn, we get

$$\Rightarrow \lambda \frac{d}{dr} \left(\frac{du_r}{dr} + \frac{u_r}{r} \right) + 2\mu \frac{d}{dr} \left(\frac{du_r}{dr} + \frac{u_r}{r} \right) = 0$$

$$\Rightarrow \qquad \left(\lambda + 2\mu\right) \left(\frac{d}{dr}\left(\frac{dur}{dr} + \frac{ur}{r}\right)\right) = 0$$

$$\Rightarrow \frac{d}{dr} \left(\frac{du_r}{dr} + \frac{u_r}{r} \right) = 0$$

$$\frac{du_r}{dr} + \frac{u_r}{r} = constant, say C$$

$$\frac{1}{c_{rr}} = \frac{1}{c_{rr}} = constant$$

The above expression also implies that $\epsilon_{rr} + \epsilon_{oo} = constant$

To simplify a bit more, lets rewrite:

$$\frac{du_r}{dr} + \frac{u_r}{r} = C$$

$$\Rightarrow \frac{1}{r} \frac{d}{dr} (ru_r) = C$$

$$\Rightarrow ru_r = \frac{Cr^2 + D}{a}$$
 are unknown integration
$$\Rightarrow u_r = \frac{Cr + D}{a}$$
 constants

These unknown integration constants C and D are obtained from boundary conditions.

If we substitute the expression of ur and up into the stress-strain relations, we will obtain expressions for radial stress or and hoop stress on.

Solution for radial stress or and hoop stress occ

We find that, just like $E_{rr} + E_{qq} = constant$, $E_{rr} + E_{qq} = constant$

$$\overline{\sigma_{rr}} + \overline{\sigma_{00}} = \lambda \left(u_{r}^{\dagger} + \frac{u_{r}}{r} + c_{1} \right) + 2\mu u_{r}^{\dagger} + \lambda \left(u_{r}^{\prime} + \frac{u_{r}}{r} + c_{1} \right) + 2\mu u_{r}^{\dagger} \\
= 2\lambda \left(u_{r}^{\dagger} + \frac{u_{r}}{r} \right) + 2\mu \left(u_{r}^{\prime} + \frac{u_{r}}{r} \right) + 2\lambda c_{1} \\
= 2(\lambda + \mu) \left(u_{r}^{\prime} + \frac{u_{r}}{r} \right) + 2\lambda c_{1} \\
\underbrace{\varepsilon_{rr} + \varepsilon_{00}}$$

=
$$a(\lambda + \mu) C + 2\lambda c_1 = A$$
 (a constant)

So the sum of radial stress and hoop stress turns out to be a constant throughout the hollow cylinder

If we now solve the simplified equilibrium equation directly in terms of stress components, we get:

$$\frac{d\sigma_{rr}}{dr} + \frac{\sigma_{rr} - \sigma_{oo}}{r} = 0$$

$$\Rightarrow \frac{d\sigma_{rr}}{dr} + 2\frac{\sigma_{rr}}{r} - \frac{\sigma_{rr} + \sigma_{oo}}{r} = 0$$
Note the partial deriv.

has become total deriv.

since σ_{rr} is function of σ_{rr} only σ_{rr}

$$\Rightarrow \frac{d\sigma_{rr}}{dr} + \frac{2\sigma_{rr}}{r} = \frac{A}{r}$$

$$\Rightarrow \frac{1}{\Upsilon^2} \frac{d}{dr} \left(\sigma_{rr} \Upsilon^2 \right) = \frac{A}{\Upsilon}$$

$$\Rightarrow \quad \sigma_{rr} = \frac{A}{2} + \frac{B}{r^2} \quad \text{and} \quad \sigma_{oo} = A - \sigma_{rr}$$

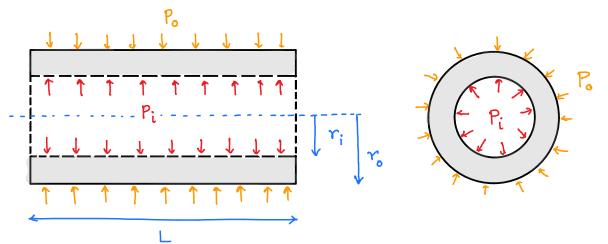
*A and B are unknown integration = A constants.

$$= \frac{A}{a} - \frac{B}{r^2}$$

If we now satisfy boundary conditions, then we will have "unique" elasticity solutions

(a) External and internal pressure loading

Cylindrical vessel is loaded by uniform pressures (Pi, Po) on the inside and outside surfaces of the cylinder. We assume the ends of the cylinder are traction-free A real pressure vessel will have its ends closed in some fashion and we will deal with that case in axial loading.



For this case, the boundary conditions will be defined by the outward surface normal. The inner curve surface (at $r=r_i$), the outward surface normal points in the -r-direction

$$\left[\underline{n}\right]_{(r_i, 0, z)} = \begin{bmatrix} -1 \\ 0 \\ 0 \end{bmatrix}$$

while the internal pressure Pi acts in the + r direction

So,
$$\left[t^{ext}\right]_{(r_i, O, z)} = \begin{bmatrix} p_i \\ o \\ o \end{bmatrix}$$

Upon writing the boundary condition equation $\underline{\underline{u}} = \underline{t}^{\text{ext}}$ in cylindrical coordinate system for the inner surface, we get:

$$\begin{bmatrix} \sigma_{rr} & \tau_{re} & \tau_{rz} \\ \tau_{re} & \tau_{ae} & \tau_{ez} \\ \tau_{rz} & \tau_{ez} & \sigma_{zz} \end{bmatrix} \begin{bmatrix} -1 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} \rho_{i} \\ 0 \\ 0 \end{bmatrix} \Rightarrow \begin{bmatrix} -\sigma_{rr} \\ -\tau_{re} \\ 0 \end{bmatrix}$$

er,
$$\sigma_{rr}(r_i) = -p_i$$
, $\tau_{ro} = \tau_{rz} = 0$

A similar analysis for the outer curve surface would lead to $\tau_{rr}(r_o) = -P_o , \quad \tau_{ro} = \tau_{rz} = 0$

Effectively, we obtain the BCs

$$\sigma_{rr}(r_i) = -P_i$$
 $\sigma_{rr}(r_o) = -P_o$

Plugging the above BCs in the expression for orr, we get:

$$\frac{A}{2} + \frac{B}{r_i^2} = -P_i, \qquad \frac{A}{2} + \frac{B}{r_o^2} = -P_o$$

Solving for (A, B), we get

$$A = Q \frac{\gamma_{i}^{2} P_{i} - \gamma_{o}^{2} P_{o}}{\gamma_{o}^{2} - \gamma_{i}^{2}}, \qquad B = \frac{\gamma_{i}^{2} \gamma_{o}^{2} (P_{o} - P_{i})}{\gamma_{o}^{2} - \gamma_{i}^{2}}$$

Therefore, the radial and hoop slesses become

$$\overline{\sigma_{rr}} = \frac{A}{2} + \frac{B}{r^2} = \frac{P_i r_i^2}{r_o^2 - r_i^2} \left(1 - \frac{r_o^2}{r^2}\right) - \frac{P_o r_o^2}{r_o^2 - r_i^2} \left(1 - \frac{r_i^2}{r^2}\right)$$

$$\sigma_{00} = \frac{A}{2} - \frac{B}{r^{2}} = \frac{P_{i} r_{i}^{2}}{r_{o}^{2} - r_{i}^{2}} \left(1 + \frac{r_{o}^{2}}{r^{2}}\right) - \frac{P_{o} r_{o}^{2}}{r_{o}^{2} - r_{i}^{2}} \left(1 - \frac{r_{i}^{2}}{r^{2}}\right)$$

Note that both radial and hoop stresses vary with the radial coordinate r and depend linearly on the pressure loodings P; & Po They are, however, independent of the axial loading.

To study the behavior of orr and on, we consider several special cases.

(a) Inner and outer pressures are both equal $(P_i = P_o = P)$ $\nabla_{rr} = \frac{P r_i^2}{r_o^2 - r_i^2} \left(1 - \frac{r_o^2}{r^2}\right) - \frac{P r_o^2}{r_o^2 - r_i^2} \left(1 - \frac{r_i^2}{r^2}\right)$ $= P \frac{1}{r_o^2 - r_i^2} \left[\frac{r_i^2 - r_i^2}{r^2} - \frac{r_i^2 r_o^2}{r^2} - \frac{r_i^2 r_o^2}{r^2} \right]$

Similarly, Too = -P

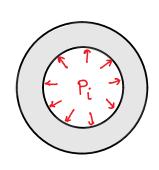
$$\sigma_{rr} = \sigma_{oo} = -p$$

(b) When only internal pressure is present (p = 0)

$$\overline{\nabla_{rr}} = \frac{P_i \Upsilon_i^2}{\Upsilon_0^2 - \Upsilon_i^2} \left(1 - \frac{\Upsilon_0^2}{\Upsilon^2} \right)$$

At $r=r_i$, $\sigma_{rr}=-p$ (inner surface)

At r=ro, or = 0 (outer surface)

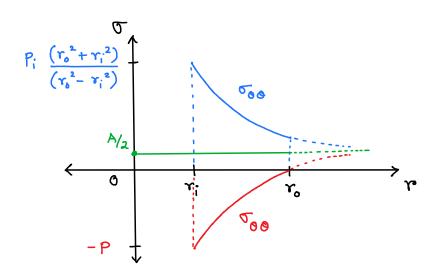


The inner pressure results in a compressive radial stress which varies from $\sigma_{rr} = -p$ at inner wall to $\sigma_{rr} = 0$ at the outer wall

$$\nabla_{00} = \frac{P_{i} r_{i}^{2}}{r_{o}^{2} - r_{i}^{2}} \left(1 + \frac{r_{o}^{2}}{r^{2}}\right) \qquad \text{At } r = r_{i} , \quad \nabla_{00} = P_{i} \frac{\left(r_{o}^{2} + r_{i}^{2}\right)}{\left(r_{o}^{2} - r_{i}^{2}\right)}$$

$$\text{At } r = r_{o} , \quad \nabla_{00} = P_{i} \frac{2r_{i}^{2}}{\left(r_{o}^{2} - r_{i}^{2}\right)}$$

The hoop stress τ_{00} is tensile throughout with peak magnitude occurring at the inner surface $r=r_i$. So the highest stresses for an internally pressurized cylinder occur on the inner wall. The variation of τ_{rr} and τ_{00} with r



is shown below:

* Keep in mind that the radial coordinate r is physically realizable only between and r.

The radial displacement ur was obtained as: $u_r = \frac{Cr}{a} + \frac{D}{r}$

To obtain the constants (C, D), we can make use of the stress-strain relation

$$\sigma_{rr} = \lambda \left(u_r' + \frac{u_r}{r} + c_1 \right) + \lambda \mu u_r'$$

$$= \lambda \left(C + c_1 \right) + \lambda \mu \left(\frac{C}{\lambda} - \frac{D}{r^2} \right)$$

$$= \left[\left(\lambda + \mu \right) C + \lambda c_1 \right] - \lambda \mu \frac{D}{r^2}$$

Comparing the above requation with expression obtained for orr

$$\sigma_{rr} = \frac{A}{A} + \frac{B}{r^2}$$

$$\left[\left(\lambda + M\right) C + \lambda c_{i}\right] = \frac{A}{2} = \frac{\gamma_{i}^{2} P_{i} - \gamma_{o}^{2} P_{o}}{\gamma_{o}^{2} - \gamma_{i}^{2}}$$

$$C = -\frac{\lambda}{\lambda + \mu} c_1 + \frac{1}{\lambda + \mu} \frac{p_i r_i^2 - p_o r_o^2}{r_o^2 - r_i^2}$$

$$D = -\frac{B}{2M} = -\frac{1}{2M} \frac{\gamma_i^2 \gamma_o^2 (P_o - P_i)}{\gamma_o^2 - \gamma_i^2}$$

Therefore, radial displacement ur becomes

$$u_{r} = \frac{1}{2} \left(\frac{\lambda}{(\lambda + \mu)} c_{1} + \frac{1}{(\lambda + \mu)} \frac{p_{i} r_{i}^{2} - p_{o} r_{o}^{2}}{r_{o}^{2} - r_{i}^{2}} \right) r - \frac{1}{2 \mu r} \frac{r_{i}^{2} r_{o}^{2} (p_{o} - p_{i}^{2})}{r_{o}^{2} - r_{i}^{2}}$$

When both internal and external pressures are zero, i.e. $P_i = P_o = 0$, we get

$$u_r = \frac{-\lambda}{a(\lambda + \mu)} e_1 r = - v c_1 r \qquad \left(\frac{\lambda}{a(\lambda + \mu)} \right)$$

We see that radial strain is not zero but it turns out as

$$\epsilon_{rr} = \frac{du_r}{dr} = -vc_1$$

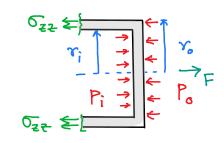
This expression is exactly what we expect, that is, radial displacement has arisen only due to Poisson's effect in the absence of radial pressure loading.

We now need to obtain axial strain E_{zz} and the end-to-end rotation Ω in terms of prescribed external loads (axial force F, torque T, and pressures P_i and P_o)

Relating axial force and axial strain

Typically, the ends of pressure vessels are capped. In such a case the end caps are

- (a) pushed out by internal pressure p:
- (b) pushed in by external pressure po
- (c) pulled out by external force F
- (d) balanced by internal traction due to ozz



Upon doing force balance, we can write

$$F + P_i \pi r_i^2 - P_o \pi r_o^2 = \iint_{A_o} \sigma_{zz} dA$$

$$\Rightarrow F = \iint_{A_0} \left[\lambda \left(u_r' + \frac{u_r}{r} + u_z' \right) + 2\mu u_z' \right] dA - P_i \pi r_i^2 + P_0 \pi r_0^2$$

$$= \iint_{A_0} \left[\lambda C + (\lambda + 2\mu) c_1 \right] dA - P_i \pi r_i^2 + P_o \pi r_o^2$$

$$= \left[\lambda C + (\lambda + 2\mu) c_1 \right] \iint_{A_0} dA - P_i \pi r_i^2 + P_0 \pi r_0^2$$

=
$$\left[\lambda C + (\lambda + 2\mu) c_1 \right] \pi (r_0^2 - r_i^2) - P_i \pi r_i^2 + P_0 \pi r_0^2$$

Substitute the obtain expression for C

$$C = -\frac{\lambda}{\lambda + \mu} C_1 + \frac{1}{\lambda + \mu} \frac{P_i r_i^2 - P_o r_o^2}{r_o^2 - r_i^2}$$

$$F = \left[\left(\lambda + 2\mu \right) - \frac{\lambda^{2}}{\lambda + \mu} \right] \frac{\pi \left(r_{o}^{2} - r_{i}^{2} \right)}{C/s \text{ area}} c_{1} - P_{i} \pi r_{i}^{2} \left[1 - \frac{\lambda}{\lambda + \mu} \frac{1}{\pi \left(r_{o}^{2} - r_{i}^{2} \right)} \right]$$

$$+ P_{o} \pi r_{o}^{2} \left[1 - \frac{\lambda}{\lambda + \mu} \frac{1}{\pi \left(r_{o}^{2} - r_{i}^{2} \right)} \right]$$

Relating torque and end-to-end rotation

A typical C/s of the hollow cylinder would have its cross-sectional normal point along the z-direction. Therefore, stress components ozz, Tzo, Tzr act on it.

We had already mentioned that
$$T_{zr} = 0$$

The moment M about the Ys's centroid can be calculated as:

$$\underline{M} = \iint \Upsilon \underline{e}_{r} \times \underline{t}^{z}(r) dA$$

$$= \iint \Upsilon \underline{e}_{r} \times \left[\sigma_{z} \underline{e}_{z} + \tau_{0z} \underline{e}_{0} \right] dA$$

Before we obtain torque from M let's look at the variation of Toz in the cross-section. This Toz occurs as a result of applying torque.

We know,
$$\gamma_{0z} = \frac{\Omega}{L} \gamma \Rightarrow \gamma_{0z} = G_1 \gamma_{0z}$$

$$= G_1 \Omega \gamma \Rightarrow \gamma_{0z} = G_1 \gamma_{0z}$$
linearly with γ_0

The torque T is simply the component of moment along the z-axis, i.e

$$T = \underbrace{M \cdot e_z}_{A_0} = \iint (re_r \times [\sigma_{zz} e_z + \tau_{0z} e_0]) \cdot e_z dA$$

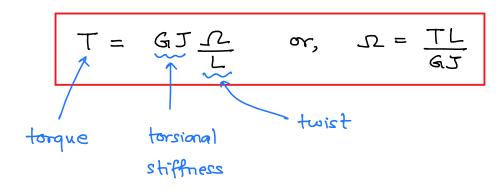
$$= \iint r \tau_{0z} dA$$

$$= \iint r^2 G R dA = G R \iint r^2 dA$$
The quantity
$$\iint r^2 dA \text{ is called}$$

$$Polar moment of inertia, J$$

polar moment of inertia (a geometric quantity)

So, finally torque T is equal to



Thin cylinder approximation

For thin cylinder, $\gamma_o \approx \gamma_i = \gamma_m$ (Outer wall radius \approx Inner wall and $t = \gamma_0 - \gamma_1$

$$t = \gamma_0 - \gamma_i$$

The radial and hoop stress then become:

$$\sigma_{rr} (r=r_i) = -P_i$$

$$\frac{\sigma_{00} (r = r_i^2)}{r_0^2 - r_i^2} \left(1 + \frac{r_0^2}{r_i^2}\right) = \frac{P_i}{r_0^2 - r_i^2} \frac{(r_0^2 + r_i^2)}{(r_0 + r_i^2)} \frac{2r_m^2}{r_0^2 - r_i^2}$$

$$=\frac{P_{i}r_{m}}{t}$$

For
$$t << r_m \Rightarrow \frac{r_m}{t} >> 1$$

So, Too >> orr -> Hence, in comparison to Too, the small radial stress component is normally ignored