# 2017 Business Airplanes



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he U.S. economy has shown steady improvement as indicated by the 0.2%, 0.5% and 0.6% increases for November and December 2016 and January 2017, respectively, in the Conference Board Leading Economic Index, a composite measure of manufacturing activity, consumer and business demand for goods and services, stock prices and new building permits, among other factors. But you'd never know there was any improvement from looking at the general aviation market.

New aircraft sales revenues plunged by nearly \$5 billion in 2016 from one year earlier, according to the General Aviation Manufacturers Association (GAMA). Business iet deliveries fell from 718 units in 2015 to 661 units in 2016, the industry's lowest figure since 2004. Activity was strongest in North America and Europe, but a prolonged and pronounced slump in Latin America, Asia-Pacific, the Middle East and Africa dragged down total sales. North America and Europe accounted for more than 80% of turbofan deliveries and more than two-thirds of the turboprop deliveries.

GAMA reports that turboprops

fared slightly better than in the previous year, with a slight uptick in deliveries from 557 units in 2015 to 582 deliveries in 2016. North America. Asia-Pacific and Europe saw slight increases, while Latin America witnessed a minor decline. Overall, turbine aircraft deliveries have remained flat since 2009 and actually declined since 2013. More telling, turbine aircraft sales revenues fell nearly 15% in 2016 compared to the previous year.

Piston aircraft deliveries also fell by nearly 5% in 2016, although North America had a slight increase, accounting for nearly  $70\overline{\%}$  of the sales.

Yet, the size of the world's turbofan and turboprop fleet increased slightly to 36,674 aircraft, according to GAMA citing data published by Jetnet LLC.

Sales and deliveries of new aircraft historically have tracked with global economic activity. But that's no longer the case in the business aircraft industry, says Rolland Vincent of his eponymous Aviation Consulting firm in Plano, Texas. His firm surveys 500 business aircraft owners and operators every 90 days.

In collaboration with Utica, New York-based Jetnet, Vincent publishes quarterly history and forecast reports used for planning purposes by the business aircraft industry.

The Jetnet IQ report for first quarter 2017, for instance, says that 80% of North American respondents believe the economy there will grow faster in the next 12 months than in the previous year. More than 80% of North Americans believe the Donald Trump administration will be beneficial to aviation during the next year. And business jet operators are flying more than 4.3 million missions per year, the highest since 2009 and even more than in 2008 prior to the Great Recession.

Robert Stallard of Vertical Research Partners also notes that business aircraft operations grew at 2.9% in early 2017 year-over-year. For early 2016, year-over-year growth was only 1.1% versus 2015.

The economies of China and India should continue to expand, but the average GDP growth of 18 other nations, including the U.S., will hover near 2.0% in 2017, according to Vincent. These 20 nations account for most of the world's business aircraft.

Still, potential buyers are not rushing to new aircraft sales offices and asking for demo flights. In fact, Vincent projects that new turbofan aircraft deliveries will drop again this year to 640 units, accompanied by a slight decline in sales revenues. And he forecasts another 5.5% decrease to 605 units in 2018.

The reason? Oversupply. Book-to-bill ratios for Bombardier, Dassault, Embraer, Gulfstream and Textron all are below 1:1, meaning that the manufacturers are taking fewer orders for new equipment than the number of units they ship from their plants. Dassault, for example, had a book-to-bill ratio of less than 0.5 to 1 in 2015 and 2016.

Asking prices for turbofan aircraft are soft in 2017. Compare list prices in BCA's May 2016 Handbook with prices this year. Most turbofans are priced the same as last year, though a few Falcon and Gulfstream models show modest increases. To increase competitiveness, Embraer dropped the Legacy 600 in favor of the new Legacy 650E that is priced \$5.7 million less than last year's Legacy 650. And Gulfstream dropped the G150 from its lineup due to low demand.

There also is a widening gap between list prices and sale prices. For instance, Vincent says Bombardier is selling some models at a 33% discount, forcing other manufacturers to sacrifice profit margins or lose sales. While the Canadian manufacturer garnered the largest number of business aircraft deliveries in 2016 among business jet makers, any such discounting would likely result in razor-thin margins.

Textron Aviation is faring better than most others. CEO Scott Ernest's capacity discipline resulted in the best book-to-bill ratio of any of the five jet makers from 2013 through 2016. But last year it still was hovering at slightly less than 1:1, according to Vincent, hardly a banner year for business jets.

This year, the FAA revised its general aviation fleet forecast, lowering growth of the general aviation fleet to 0.1% per year for the next two decades, with new turbine aircraft deliveries offsetting a projected contraction of the piston aircraft fleet, according to its Aerospace Forecast Report Fiscal Years 2017 to 2037. GAMA also notes that the general aviation pilot population is shrinking, although there was a slight uptick in student starts in 2015. While the general aviation fleet growth is lackluster, the FAA estimates that business jet operations will increase 3.0% from 2017 to 2037 in its latest forecast.

The report also says "there is uncertainty regarding the impact of the new U.S. administration's policies on economic growth." And with both U.S. Rep. Bill Shuster (R-Penn.), chairman of the House Transportation and Infrastructure Committee, and President Trump pushing to spin off FAA ATC into to a private corporation with a board of directors dominated by the airlines, business aircraft operators potentially could face substantial airspace and airport user fees.

On a more positive note, the FAA believes that the price for turbine fuel will increase only modestly in 2017 because the price of crude oil should stabilize at about \$47 per barrel, up from \$39 per barrel in 2016. Crude oil shouldn't again reach its 2013 price of \$100 per barrel until 2026, according to the FAA Forecast.

Regardless of the price of fuel or user fees, the FAA estimates that piston

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When "Lucky" Lindy made his transatlantic crossing, he didn't have to deal with an ocean of congressional wrangling (maybe that's why they called him "Lucky"). The prevailing winds blew in his favor. But today, those winds have changed. Flying for business is more scrutinized than ever. Luckily, there's NBAA. We've made a home on the Hill, so that our members can make a living in the sky. Because business aviation enables economic growth. And at NBAA, we enable business aviation.

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aircraft deliveries will continue to decline. In 2016, piston-engine aircraft deliveries from U.S. manufacturers were down 4.2% from 2015, according to GAMA. The FAA estimates that the piston aircraft fleet will atrophy at 0.8% per year from 2017 to 2036, due to "unfavorable pilot demographics, overall increased cost of aircraft ownership" and "new aircraft deliveries not keeping pace with retirements of the aging fleet."

Nonetheless, most piston aircraft manufacturers are hiking prices this

HTF7700L turbofans, features Garmin G5000 avionics and offers double-club seating for eight passengers.

Vincent foresees a sweet spot in the business jet market for 3,000-nm to 4,000-nm super-mids, such as the Longitude. Textron's new model could spur Bombardier, Embraer and Gulfstream to look at derivatives or new models in this segment. He's also bullish on the Falcon 5X because of its cabin size, range and fuel efficiency. But ongoing problems with its Snecma Silvercrest turbofans have slowed Dassault's de-

Daher is replacing the TBM 900 with the TBM 910, a derivative upgraded with Garmin G1000 NXi avionics and other modifications. Both Daher and Pilatus increased prices in response to strong order books.

While most new piston and turbofan aircraft deliveries remain stubbornly stagnant, several developments are buoying spirits in the business aircraft industry. The European Aviation Safety Agency (EASA) issued final regulations permitting commercial single-engine turbine aircraft operations in instrument meteorological conditions (IMC). Notably, Europe is the last large business aircraft market that, with few exceptions, did not permit commercial single-engine operations in IMC.

After seven years, the 36-state International Civil Aviation Organization (ICAO) council adopted uniform CO<sub>2</sub> emission standards for aircraft. Such standardization facilitates creation of market-based measures to move toward carbon-neutral growth of aircraft operations by 2020. Reduction in CO<sub>2</sub> will be made possible by more-efficient air traffic management, use of sustainable alternative fuels, replanting rain forests and developing more-fuel-efficient aircraft.

The FAA also continues to progress through Phase II of its Piston Aviation Fuels Initiative by developing a drop-in replacement unleaded avgas by 2018. Shell Oil and Swift Fuels have been selected to partner with the FAA to develop ASTM standards for unleaded avgas that will have the least technical and financial impact on general aviation aircraft operators and establish a fuel distribution infrastructure. However, it's still not clear how much the price of that fuel will change from the cost per gallon of 100LL gasoline.

So, in the short term, look for singleengine and multiengine turboprops to be solid sellers. The piston-engine market is in for a rough ride because of aging pilot demographics, increasing direct operating costs and tougher local airport authority rules, regulations and restrictions, particularly in California. The turbofan aircraft market will remain relatively flat because of oversupply in almost all segments. But a new generation of roomy, fuel-efficient and fast U.S. transcontinental-range and transatlantic-range super-midsize to large-cabin aircraft hold the promise to lift the turbofan sector out of its doldrums. **BCA** 



year. That includes Cirrus Aircraft, Piper and Textron Aviation, but Mooney, whose future seems uncertain, is holding 2016 pricing for its M20 models. Notably, GAMA reports Mooney delivered just seven aircraft in 2016, and there is very little activity at the factory in Kerrville, Texas. However, the M20U Ovation Ultra and M20V Acclaim Ultra, models featuring left- and right-side doors, received certification in March, and development of the diesel-powered models was still pending as we closed this issue.

Not all the news for 2017 is bad, however. This year, Textron Aviation's 3,500-nm range super-midsize Cessna CE-700 Citation Longitude makes its debut in the Purchase Planning Handbook. Due for certification late this year, the Longitude's evolutionary design combines a stretched and strengthened Citation Latitude fuselage mated to proven wing and empennage structures that were modified and adapted for the mission. The aircraft is powered by well-proven Honeywell

velopment program by several years.

Gulfstream's 6,200-nm range, Mach 0.85 cruise G600 also is making its debut in this year's Handbook. A longer cabin, wider wingspan and longer-range derivative of the G500, it features active side-sticks, fly-by-wire (FBW) flight controls and Gulfstream's signature Symmetry flight deck. It's slated for certification late next year.

Bombardier's Global 7000 was due to make its debut in this year's *Handbook*. But the manufacturer declined to release performance details despite having two aircraft in flight testing. A third test aircraft, slated for first flight later this year, should be fully production conforming, Vincent believes. Look for the Global 7000 to appear in the 2018 *Handbook*.

The single-engine turboprop sector also remains stable to strong. Epic, Piper, Mahindra, Quest and Textron held prices unchanged or close to 2016 levels. Epic Aircraft is making changes to the E1000 to ensure it complies with upcoming certification requirements.

# How to Use the Airplane Charts



or an aircraft to be listed in the Purchase Planning Handbook, a production conforming article must have flown by May 1 of this year. The dimensions, weights and performance characteristics of each model listed are representative of the current production aircraft being built or for which a type certificate application has been filed. The basic operating weights we publish should be representative of actual production turboprop and turbofan aircraft because we ask manufacturers to supply us with the average weights of the last 10 commercial aircraft that have been delivered. However, spot checks of some manufacturers' BOW numbers reveal anomalies. We reserve the right to make adjustments to weights, dimensions and performance data. These data adjustments will be noted in the Remarks section for specific models as "BCA Estimated Data."

The takeoff field length distances are based on maximum takeoff weight for maximum range missions.

Please note that "all data preliminary" in the Remarks section indicates that actual aircraft weight, dimension and performance numbers may vary considerably after the model is certified and delivery of completed aircraft begins. \*\*\*All data for these aircraft is highlighted with a blue tint.\*\*\*

#### Manufacturer, Model and Type Designation

In some cases, the airplane manufacturer's name is abbreviated. The model name and the type designation also are included in this group.

#### BCA Equipped Price

- Price *estimates* are first quarter, current year dollars for the next available delivery. Some aircraft have long lead times, thus the actual price will be higher than our published price because of block point changes and inflation adjustments. Note well, manufacturers may change prices without notification.
- ► Piston-powered airplanes Computed retail price with at least the level of equipment specified in the "BCA Required Equipment List."
- ► Turbine-powered airplanes Computed retail price with at least the level of equipment specified in the "BCA Required Equipment List," if available. Some manufacturers decline to provide us with actual prices of delivered aircraft, so we may estimate them. The aircraft serial numbers aren't necessarily

consecutive because of variations in completion time and because some aircraft may be configured for non-commercial, special missions.

#### **Characteristics**

► Seating: Crew + Typical Executive Seating/High-Density Seating/Max Certification Seating — For example, 2+8/13/19 indicates that the aircraft requires two pilots, there are eight seats in the typical executive configuration, 13 seats with optional high-density seating and up to 19 passenger seats based upon FAA and/or EASA certification limits. A four-place single-engine aircraft is shown as 1+3/3, indicating that one pilot is required and there are three other seats available for passengers. We require two pilots for all turbofan airplanes, except for single-pilot certified aircraft such as the Cirrus Vision SF-50, Eclipse 550, Cessna Citation CJ series, HondaJet and Syberiet SJ30-2, which have, or will have, a large percentage of single-pilot operators. Four crewmembers are specified for ultra-long-range aircraft — three pilots and one flight attendant. However, Dassault only provides data with three crewmembers aboard for its ultra-long-range aircraft, thus the notations for the Falcon 8X.

Each occupant of a turbine-powered airplane is assumed to weigh 200 lb., thereby allowing for stowed luggage and carry-on items. In the case of pistonengine airplanes, we assume each occupant weighs 170 lb. There is no luggage allowance for piston-engine airplanes.

- ► Wing Loading MTOW divided by total wing area.
- ▶ Power Loading MTOW divided by total rated takeoff horsepower or total rated takeoff thrust.
- ► FAR Part 36 Certified Noise Levels Flyover noise in A-weighted decibels (dBA) for small and turboprop aircraft. For turbofan-powered aircraft, we provide Part 36 EPNdB (effective perceived noise levels) for Lateral, Flyover and Approach.

#### **Dimensions**

▶ External Length, Height and Span dimensions are provided for use in determining hangar and/or tie-down space requirements.

Internal Length, Height and Width are based on a completed interior, including insulation, upholstery, carpet, carpet padding and fixtures. Note well: These dimensions are not intended to be



based upon green aircraft dimensions. They must reflect the actual net dimensions with all soft goods installed. Some manufacturers provide optimistic measurements, thus prospective buyers are advised to measure aircraft themselves.

As shown in the Cabin Interior Dimensions illustration, for small airplanes other than "cabin-class" models, the length is measured from the forward bulkhead ahead of the rudder pedals to the back of the rear-most passenger seat in its normal, upright position. The upright position of the aft seat backs allows room for luggage in the cabin.

For so-called cabin-class and larger aircraft, we show two or three dimensions, depending on aircraft class. The **first** is the overall length of the passenger cabin, measured from the aft side of the forward cockpit/cabin divider to the aft-most bulkhead of the cabin. The aft-most point is defined by the rear side of a baggage compartment that is accessible to passengers in flight or the aft pressure bulkhead. The overall length is reduced by the length of any permanent mounted system or structure that is installed in the fuselage ahead of the aft bulkhead. For example, some aircraft have full fuselage cross-section fuel tanks mounted ahead of the aft pressure bulkhead.

The second length number is the net length of the cabin that routinely is occupied by passengers. It's measured from the aft side of the forward cockpit/cabin divider to an aft point defined by the rear of the cabin floor capable of supporting passenger seats, the rear wall of an aft galley or lavatory, an auxiliary pressure bulkhead or the front wall of the pressurized baggage compartment. Some aircraft have the same net and overall interior length because the

manufacturer offers at least one interior configuration with the aft-most passenger seat located next to the front wall of the aft luggage compartment.

The third length dimension is the main seating area of the cabin, including all passenger seats in the standard aircraft configuration that are certified for full-time occupancy. Some manufacturers may fit their aircraft with forward, side-facing divans, ahead of areas with individual fore-aft facing chairs. The main seating length dimension may include such forward cabin side-facing divans at the discretion of the manufacturer. The length of the lavatory, even though it may have a seat certified for full-time occupancy, may not be included in the main seating length dimension.

Interior height is measured at the center of the cabin cross-section. If the aircraft has a dropped aisle, the maximum depth below the adjacent cabin floor is shown. Some aircraft have dropped aisles of varying depths, resulting in less available interior net height in certain sections of the cabin.

Two width dimensions are shown for multiengine turbine airplanes — one at the widest part of the cabin and the other at floor level. The dimensions, however, are not completely indicative of the usable space in a specific aircraft because of individual variances in interior furnishings.

#### Power

Number of engines, if greater than one, and the abbreviated name of the manufacturer: GE — General Electric; GE/Honda — General Electric and Honda; Honeywell; CFMI — CFM International; IAE — International Aero Engines; Lyc — Textron Lycoming; P&WC — Pratt

& Whitney Canada; RR — Rolls-Royce; Snecma; TCM — Teledyne Continental; and Wms — Williams International.

▶ Output — Takeoff rated horsepower for propeller-driven aircraft or pounds thrust for turbofan aircraft. If an engine is flat rated, enabling it to produce takeoff rated output at a higher than ISA (standard day) ambient temperature, the flat rating limit is shown as ISA+XXC. Highly flat-rated engines, i.e. engines that can produce takeoff rated thrust at a much higher than standard ambient temperature, typically provide substantially improved high density altitude, climb and high-altitude cruise performance.

▶ Inspection Interval is the longest scheduled hourly major maintenance interval for the engine, either "t" for TBO or "c" for compressor zone inspection. In some

fuel required to fly 1.5 hr. at high-speed cruise.

▶ Max ramp, max takeoff and max landing weights may be the same for light aircraft that may only have a certified max takeoff weight.

▶ EOW/BOW — Empty Operating Weight is shown for piston-powered airplanes. EOW is based on the factory standard weight, plus items specified in the "BCA Required Equipment List," less fuel, loose equipment and cabin stores.

Basic Operating Weight is shown for turbine-powered airplanes. BOW is based on the average EOW weight of the last 10 commercial deliveries, plus 200 lb. for each required crewmember. Three flight crewmembers and one cabin crewmember are required for ultra-long-range aircraft, unless otherwise noted.



AIRBUS CORPORATE JETS

cases, we show a second number if the engine manufacturer has obtained an extended maintenance interval, provided that the engines are enrolled in the manufacturer's service program. OC is shown only for engines that have "on condition" repair or replace parts maintenance.

### Weights (lb.)

Weight categories are listed as appropriate to each class of aircraft.

► Max Ramp - Maximum ramp weight for taxi.

► Max Takeoff — Maximum takeoff weight as determined by structural limits.

► Max Landing — Maximum landing weight as determined by structural limits.

▶ **Zero Fuel** — Maximum zero fuel weiht, shown by "c," indicating the certified MZFW or "b," a *BCA*-computed weight based on MTOW minus the weight of

While there is no requirement to add in the weight of cabin stores, some manufacturers choose to include galley stores and passenger supplies as part of the BOW build-up. Life vests, life rafts and appropriate deep-water survival equipment are included in the weight buildup of the 80,000+ lb., ultra-longrange aircraft.

► Max Payload – Zero Fuel weight minus EOW or BOW, as appropriate. For piston-engine airplanes, Max Payload frequently is a computed value because it is based on the *BCA* ("b") computed maximum ZFW.

▶ Max Fuel — Usable fuel weight based on 6.0 lb. per U.S. gallon for avgas or 6.7 lb. per U.S. gallon for jet fuel. Fuel quantity is based upon the largest capacity tanks that are available as standard equipment.

Available Payload With Max Fuel — Max Ramp weight minus the tanks-full weight, not to exceed Zero Fuel weight minus EOW or BOW.

► Available Fuel With Max Payload — Max Ramp weight minus Zero Fuel weight, not to exceed maximum fuel capacity.

#### Limits

*BCA* lists V speeds and other limits as appropriate to the class of airplane. These are the abbreviations used on the charts:

► VNE - Never exceed speed (redline for piston-engine airplanes).

► VNO - Normal operating speed (top of the green arc for piston-engine airplanes).

► VMO - Maximum operating speed (redline for turbine-powered airplanes).

► MMO - Maximum operating Mach number (redline for turbofan-powered airplanes and a few turboprop airplanes).

► FL/VMO - Transition altitude at which VMO equals MMO (large turboprop and turbofan aircraft).

► VA - Maneuvering speed (except for certain large turboprop and all turbofan aircraft).

▶ **VDEC** — Accelerate/stop decision speed (multiengine piston and light multiengine turboprop airplanes).

► VMCA — Minimum control airspeed, airborne (multiengine piston and light multiengine turboprop airplanes).

▶ **Vso** – Maximum stalling speed, landing configuration (single-engine airplanes).

► Vx - Best angle-of-climb speed (single-engine airplanes).

► VXSE — Best angle-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12,500 lb.).

► **VY** - Best rate-of-climb speed (single-engine airplanes).

▶ WYSE — Best rate-of-climb speed, oneengine inoperative (multiengine piston and multiengine turboprop airplanes under 12.500 lb.).

► **V2** – Takeoff safety speed (large turboprops and turbofan airplanes).

▶ VREF - Reference landing approach speed (large turboprops and turbofan airplanes, four passengers, NBAA IFR reserves; eight passengers for ultralong-range aircraft).

▶ PSI - Cabin pressure differential (all pressurized airplanes).

#### **Airport Performance**

Airplane Flight Manual takeoff runway performance is shown for sea level, standard day and for 5,000-ft. elevation/25C day density altitude. All-engine takeoff distance (TO) is shown for single-engine and multiengine piston, and turboprop airplanes with an MTOW of less than 12,500 lb. Takeoff distances and speeds assume MTOW, unless otherwise noted.

► Accelerate/Stop distance (A/S) is shown for small multiengine piston and small turboprop airplanes.

► Takeoff Field Length (TOFL), the greater of the one-engine inoperative (OEI) takeoff distance or the accelerate/stop distance, is shown for FAR Part 23 Commuter Category and FAR Part 25 airplanes. If the accelerate/stop and accelerate/stop distances are equal, the TOFL is the balanced field length.

► Landing distance (LD) is shown for FAR Part 23 Commuter Category and FAR Part 25 Transport Category airplanes. The landing weight is BOW plus four passengers and NBAA IFR fuel reserves. We assume that 80,000+ lb. ultra-long-range aircraft will have eight passengers on board.

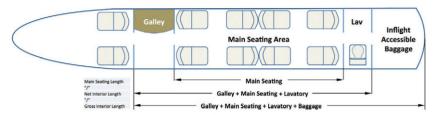
▶ V2 and VREF speeds are useful for reference when comparing the TOFL and LD numbers because they provide an indication of potential minimum-length runway performance when low RCR or runway gradient is a factor.

BCA lists two additional warm day airport performance numbers for large turboprop- and turbofan-powered airplanes. First, we publish the Mission Weight, which is the maximum allowable takeoff weight when departing a 5,000-ft. elevation/ISA+20C airport with at least four passengers aboard.

Mission Weight, when departing from a 5,000-ft./ISA+20C airport, may be less than the MTOW at sea level on a standard day because of FAR Part 25 second-segment, one-engine-inoperative, climb performance requirements. If maximum allowable mission weight at takeoff is restricted under said conditions, it's flagged with a "p." Aircraft with highly flat-rated engines are less likely to have a performance limited mission weight when departing under said warm day conditions.

Second, we publish the NBAA IFR range for said warm day conditions. assuming a transition into standardday, ISA flight conditions after takeoff. For purposes of computing NBAA IFR range, the aircraft is flown at the long-range cruise speed shown in the "Cruise" block or at the same speed as shown in the "Range" block. Notably, some aircraft may actually have slightly better range performance when departing from said warm day airport because

### Cabin Length



they have a 5,000-ft. head start on the climb to cruise altitude.

#### Climb

The all-engine time to climb provides an indication of overall climb performance, especially if the aircraft has an all-engine service ceiling well above our sample time-to-climb altitudes. We provide the all-engine time to climb to one of three specific altitudes, based on type of aircraft departing at MTOW from a sea-level, standard-day airport: (1) FL 100 (10,000 ft.) for normally aspirated single-engine and multiengine piston aircraft, plus pressurized singleengine piston aircraft and unpressurized turboprop aircraft; (2) FL 250 for pressurized single-engine and multiengine turboprop aircraft; or (3) FL 370 for turbofan-powered aircraft. These data are published as time-to-climb in minutes/climb altitude. For example, if a non-pressurized twin-engine piston aircraft can depart from a sea-level airport at MTOW and climb to 10,000 ft. in 8 min., the time to climb is expressed as 8/FL 100.

We also publish the initial all-engine climb feet per nautical mile gradient, plus initial engine-out climb rate and gradient, for single-engine and multiengine pistons and turboprops with MTOWs of 12,500 lb. or less.

The one-engine-inoperative (OEI) climb rate for multiengine aircraft at MTOW is derived from the Airplane Flight Manual. OEI climb rate and gradient are based on landing gear retracted and wing flaps in the takeoff configuration used to compute the published takeoff distance. The climb gradient for such airplanes is obtained by dividing the product of the climb rate (fpm) in the Airplane Flight Manual times 60 by the VY or VYSE climb speed. as appropriate.

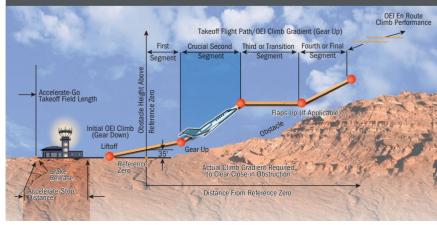
The OEI climb gradients we show for FAR Part 23 Commuter Category and FAR Part 25 Transport Category aircraft are the second-segment net climb performance numbers published in the AFMs. Please note: The AFM net second-segment climb performance numbers are adjusted downward by 0.8% to compensate for variations in pilot technique and ambient conditions.

The OEI climb gradient is computed at the same flap configuration used to calculate the takeoff field length.

#### Ceilings (ft.)

► Maximum Certificated Altitude — Maximum allowable operating altitude

# FAR Part 25 and Part 23 Commuter Category OEI Climb Performance



determined by airworthiness authorities. 
All-Engine Service Ceiling — For turbofan aircraft: maximum altitude at which at least a 300-fpm rate of climb can be attained, assuming the aircraft departed a sea-level, standard-day airport at MTOW and climbed directly to altitude. For piston and turboprop aircraft: 100 fpm rate of climb.

▶ OEI (Engine Out) Service Ceiling ▶ Sea-Level Cabin (SLC) Altitude — Maximum cruise altitude at which a 14.7-psia, sea-level cabin altitude can be maintained in a pressurized airplane.

#### Cruise

Cruise performance is computed using EOW with four occupants or BOW with four passengers and one-half fuel load. Ultra-long-range aircraft carry eight passengers for purposes of computing cruise performance.

Assume 170 lb. for each occupant of a piston-engine airplane and 200 lb. for each occupant of a turbine-powered aircraft.

- ▶ Long Range True air speed (TAS), fuel flow in pounds/hour, flight level (FL) cruise altitude and specific range for longrange cruise by the manufacturer.
- ► Recommended (Piston-Engine Airplanes)

   TAS, fuel flow in pounds/hour, FL cruise altitude and specific range for normal cruise performance specified by the manufacturer.
- ► High Speed TAS, fuel flow in pounds/hour, FL cruise altitude and specific range for short-range, high-speed performance specified by the aircraft manufacturer.

Speed, fuel flow, specific range and altitude in each category are based on one mid-weight cruise point and these data reflect standard-day conditions. They are not an average for the overall mission and they are not representative of the above standard-day temperatures at cruise altitudes commonly encountered in everyday operations.

BCA imposes a 12,000-ft. maximum cabin altitude requirement on CAR3/FAR Part 23 normally aspirated aircraft. Non-pressurized turbocharged piston-engine airplanes are limited to FL 250, providing they are fitted with supplemental oxygen systems having sufficient capacity for all occupants for the entire duration of the mission. Pressurized CAR3/FAR Part 23 aircraft are limited to a maximum cabin altitude of 10,000 ft. For FAR Part 25 aircraft, the maximum cabin altitude for computing

cruise performance is 8,000 ft.

To conserve space, we use flight levels (FL) for all cruise altitudes, which is appropriate considering that we assume standard-day ambient temperature and pressure conditions. Cruise performance is subject to *BCA*'s verification.

#### Range

BCA shows various paper missions for each aircraft that illustrate range versus payload tradeoffs, runway and cruise performance, plus fuel efficiency. Similar to the cruise profile calculations, BCA limits the maximum altitude to 12,000 ft. for normally aspirated, non-pressurized CAR3/FAR Part 23 aircraft, 25,000 ft. for turbocharged non-pressurized airplanes with supplemental oxygen, 10,000 ft. cabin altitude for pressurized CAR 3/FAR Part 23 airplanes and 8,000 ft. cabin altitude for FAR Part 23 Commuter Category or FAR Part 25 aircraft.

- ▶ Seats-Full Range (Single-Engine Piston Airplanes) Based on typical executive configuration with all seats filled with 170-lb. occupants, with maximum available fuel less 45-min. IFR fuel reserves. We use the lower of seats full or maximum payload.
- ► Tanks-Full Range (Single-Engine Piston Airplanes) Based on one 170-lb. pilot, full fuel less 45-min. IFR fuel reserves.
- ► Max Fuel With Available Payload (Single-Engine Turboprops) — Based on BOW, plus full fuel and the maximum available payload up to maximum ramp weight. Range is based on arriving at

destination with NBAA IFR fuel reserves, but only a 100-mi. alternate is required.

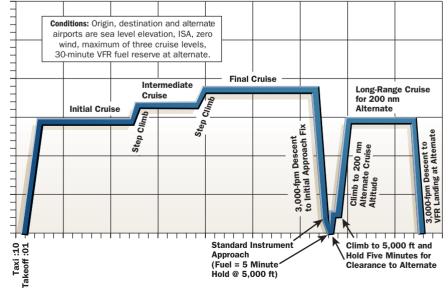
► Ferry (Multiengine Piston Airplanes and Single-Engine Turboprops) — Based on one 170-lb. pilot, maximum fuel less 45-min. IFR fuel reserves.

**Please note:** None of the missions for piston-engine aircraft includes fuel for diverting to an alternate. However, single-engine turboprops are required to have NBAA IFR fuel reserves, but only a 100-mi. alternate is required.

NBAA IFR range format cruise profiles, having a 200-mi. alternate, are used for turbine-powered aircraft with MTOWs equal to, or greater than, 22,000 lb. Turbine aircraft having MTOWs less than 22,000 lb. only need a 100-mi. NBAA alternate. The difference in alternate requirements should be kept in mind when comparing range performance of various classes of aircraft.

- ► Available Fuel With Max Payload (Multiengine Turbine Airplanes) Based on aircraft loaded to maximum zero fuel weight with maximum available fuel up to maximum ramp weight, less NBAA IFR fuel reserves at destination.
- ► Available Payload With Max Fuel (Multiengine Turbine Airplanes) Based on BOW plus full fuel and maximum available payload up to maximum ramp weight. Range based on NBAA IFR reserves at destination.
- ► Full/Max Fuel With Four Passengers (Multiengine Turbine Airplanes) Based on BOW plus four 200-lb. passengers and the lesser of full fuel or maximum available fuel up to maximum ramp

#### **NBAA IFR RANGE PROFILE**



weight. Ultra-long-range aircraft must have eight passengers on board.

► Ferry (Multiengine Turbine Airplanes) — Based on BOW, required crew and full fuel, arriving at destination with NBAA IFR fuel reserves.

We allow 2,000-ft. increment step climbs above the initial cruise altitude to improve specific range performance, even though current air traffic rules in North America provide for 4.000-ft, altitude semicircular directional traffic separation above FL 290. The altitude shown in the range section is the highest cruise altitude for the trip — not the initial cruise or mid-mission altitude.

The range profiles are in nautical miles, and the average speed is computed by dividing that distance by the total flight time or weight-off-wheels time en route. The Fuel Used or Trip Fuel includes the fuel consumed for start, taxi, takeoff, cruise, descent and landing approach but not after-landing taxi or reserves.

The Specific Range is obtained by



dividing the distance flown by the total fuel burn. The Altitude is the highest cruise altitude achieved on the specific mission profile shown.

#### Missions

Various paper missions are computed to illustrate the runway requirements, speeds, fuel burns and specific range. plus cruise altitudes. The mission ranges are chosen to be representative for the airplane category. All fixeddistance missions are flown with four passengers on board, except for ultra-long-range airplanes, which have eight passengers on board. The pilot is counted as a passenger on board piston-engine airplanes. If an airplane cannot complete a specific fixed distance mission with the appropriate payload, BCA shows a reduction of payload in the remarks section or marks the fields NP (Not Possible) at our option.



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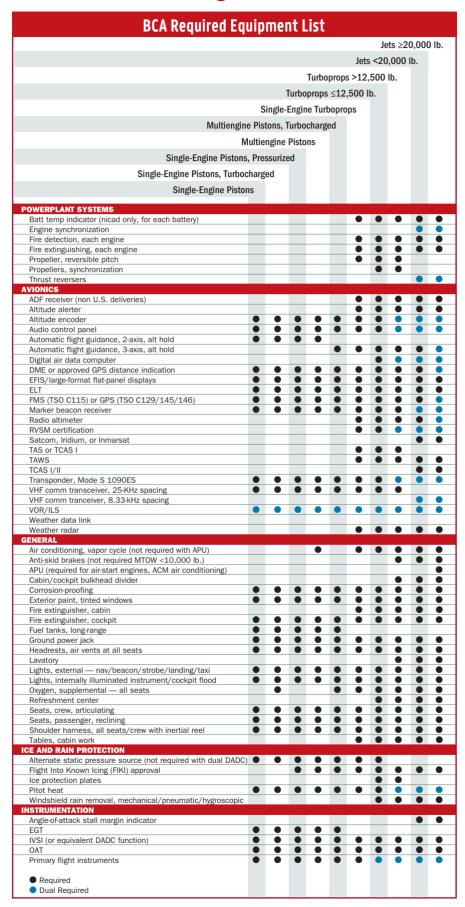
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**AVIATION WEEK** 



Runway performance is obtained from the Approved Airplane Flight Manual. Takeoff distance is listed for single-engine airplanes; accelerate/stop distance is listed for piston twins and light turboprops; and takeoff field length, which often corresponds to balanced field length, is used for FAR Part 23 Commuter Category and FAR Part 25 large Transport Category airplanes.

Flight Time (takeoff to touchdown, or weight-off-wheels, time) is shown for turbine airplanes. Some piston-engine manufacturers also include taxi time, resulting in a chock-to-chock, Block Time measurement. Fuel Used, though, is the actual block fuel burn for each type of aircraft, but it does not include fuel reserves. The cruise altitude shown is that which is specified by the manufacturer for fixed-distance missions.

- ≥ 200 nm (Piston-engine airplanes).
- ► 500 nm (Piston-engine airplanes).
- ▶ 300 nm (Turbine-engine airplanes, except ultra-long-range).
- ► 600 nm (Turbine-engine airplanes, except ultra-long-range).
- ▶ 1,000 nm (All turbine-engine airplanes).
- ▶ 3,000 nm − (Ultra-long-range turbine-engine airplanes).
- ▶ 6,000 nm − (Ultra-long-range turbine-engine airplanes).

#### Remarks

In this section, *BCA* generally includes the base price, if it is available or applicable; the certification basis and year; and any notes about estimations, limitations or qualifications regarding specifications, performance or price. All prices are in 2017 dollars, FOB at a U.S. delivery point, unless otherwise noted. The certification basis includes the regulation under which the airplane was originally type certified, the year in which it was originally certified and, if applicable, subsequent years during which the airplane was re-certified. "BCA Estimated Data" indicates that we made adjustments to data provided by manufacturers.

#### General

The following abbreviations are used throughout the tables: "NA" means not available; "—" indicates the information is not applicable and "NP" signifies that specific performance is not possible. **BCA** 

# Single-Engine Pistons normally aspirated

Manufacturer			Cirrus Design	Piper	Textron Aviation	Cirrus Design
Model			SR20	Arrow PA-28R-201	Cessna Skylane CE-182T	SR22
BCA Equipped	Price		\$389,900	\$466,880	\$480,000	\$539,900
		Seating	1+3/4	1+3/3	1+3/3	1+3/4
haracter-		Wing Loading	21.7	16.2	17.8	23.5
tics		Power Loading	14.65	13.75	13.48	11.61
		Noise (dBA)	83.4	77.7	77.7	83.7
kternal	Length		26.0	24.7	29.0	26.0
imensions		Height	8.9	7.9	9.3	8.9
t.)		Span	38.3	35.4	36.0	38.3
ternal		Length	8.0	7.7	7.2	8.0
mensions		Height	4.1	3.7	4.0	4.1 4.1
.)		Width	4.1 Lyc	3.5	3.5	4.1 Cont
		Engine	IO-390-C3B6	Lyc 10-360-C1C6	Lyc IO-540-AB1A5	IO-550-N
wer		Output (hp)	215	200	230	310
		Inspection Interval	2,000t	2,000t	2,000t	2,000t
		Max Ramp	3,160	2,758	3,110	3,610
		Max Takeoff	3,150	2,750	3,100	3,600
		Max Landing	3,150	2,750	2,950	3,600
		Zero Fuel	3,043b	2,636b	2,976b	3,400c
		EOW	2,120	1,798	1,965	2,260
eights (lb.)		Max Payload	923	838	1,011	1,140
		Useful Load	1,040	960	1,145	1,350
		Max Baggage	130	200	200	130
		Max Fuel	336	432	522	552
		nilable Payload w/Max Fuel	704	528	623	798
		nilable Fuel w/Max Payload	117	122	135	210
		Vne	201	183	175	205
imits		Vno	164	146	140	176
		VA	133	118	110	140
	TO (SL elev./ISA temp.)		2,530	1,600	1,514	1,756
rport	TO (5,000-ft. elev@25C)		4,305	3,250	2,708	3,016
erfor-			62	55	49	64 88
ance		Vx Vy	81 88	78 90	65 80	108
	Ti	me to Climb (min.)/Altitude	20/FL 100	16/FL 100	15/FL 100	11/FL 100
imb	111	Initial Gradient (ft./nm)	540	560	694	775
ailing (ft )		Service	17,500	16,200	18,100	17,500
eiling (ft.)		TAS	135	124	125	160
		Fuel Flow	53	51	61	68
	Long Range	Altitude	FL 080	FL 100	FL 100	FL 080
		Specific Range	2.547	2.431	2.049	2.353
		TAS	145	130	135	171
		Fuel Flow	61	68	69	92
ruise	Recommended	Altitude	FL 080	FL 090	FL 100	FL 080
		Specific Range	2.369	1.912	1.957	1.859
		TAS	152	137	144	180
		Fuel Flow	71	76	83	107
	High Speed	Altitude	FL 080	FL 060	FL 060	FL 080
		Specific Range	2.129	1.803	1.735	1.682
		Nautical Miles	672	537	795	1,118
		Average Speed	135	121	131	162
	Seats Full	Fuel Used	275	156	414	492
andoe		Specific Range/Altitude	2.444/FL 080	3.442/FL 070	1.920/FL 120	2.272/FL 080
anges		Nautical Miles	672	926	912	1,118
	Tanks Full	Average Speed	135	121	131	162
	IANKS FÜH	Fuel Used	275	408	471	492
		Specific Range/Altitude	2.444/FL 080	2.270/FL 070	1.936/FL 120	2.272/FL 080
		Runway	1,685	1,600	1,216	1,303
	200 nm	Block Time	1+26	1+29	1+37	1+09
	200 11111	Fuel Used	112	125	123	127
issions		Specific Range/Altitude	1.786/FL 080	1.600/FL 070	1.626/FL 120	1.575/FL 080
occupants)		Runway	1,685	1,600	1,369	1,519
	500 nm	Block Time	3+30	3+50	3+52	2+49
	Joo mill	Fuel Used	245	278	269	305
		Specific Range/Altitude	2.041/FL 080	1.799/FL 090	1.859/FL 120	1.639/FL 080
		Suggested Base Price	\$389,900	\$466,880	\$480,000	\$539,900
emarks			FAR 23, 1999/2017 Includes Garmin Perspective+ avionics.	CAR 3, 1976/2001 Garmin G500 standard.	FAR 23, 1996/2001 A 23-6 Garmin G1000 NXi with GFC 700 autopilot.	FAR 23, 2000 Includes Garmin Perspectiv avionics.

# Single-Engine Pistons normally aspirated

Manufacturer			Mooney	GippsAero	Textron Aviation		
Model			Ovation Ultra M20U	Airvan GA-8	Beechcraft Bonanza G36 G36		
BCA Equipped P	rice		\$689,000	\$726,960	\$815,000		
	Seating		1+3/4	1+6/7	1+4/5		
Character-		Wing Loading	19.3	20.7	20.2		
stics		Power Loading	10.86	13.33	12.17		
		Noise (dBA)	NA	84.9	76.7		
xternal	Length		26.9	29.3	27.5		
Dimensions		Height	8.3	12.8	8.6		
ft.)		Span	36.1	40.7	33.5		
nternal		Length	8.1	11.6	12.6		
Dimensions		Height	3.7	3.7	4.2		
t.)		Width	3.6	4.2	3.5		
		Engine	Cont	Lyc IO-540-K1A5	Cont		
ower			IO-550-G-AP		IO-550-B		
OWCI		Output (hp)	310	300	300		
		Inspection Interval	2,200t	2,000t	1,900t		
		Max Ramp	3,374	4,014	3,663		
		Max Takeoff	3,368	4,000	3,650		
		Max Landing	3,200	4,000	3,650		
		Zero Fuel	3,197b	3,849b	3,509b		
		EOW	2,260	2,241	2,600		
leights (lb.)		Max Payload	937	1,608	909		
		Useful Load	1,114	1,773	1,063		
		Max Baggage	120	180	670		
		Max Fuel	600	540	444		
		ailable Payload w/Max Fuel	514	1,233	619		
	Ava	ailable Fuel w/Max Payload	177	166	154		
		VNE	195	185	203		
imits	Vno		174	143	165		
		VA	127	121	139		
	TO (SL elev./ISA temp.)		2,300	1,860	1,913		
irport	TO (5,000-ft. elev.@25C)		3,400	3,670	3,450		
erfor-	Vso		59	57	59		
nance	Vx Vy		75	70	84		
			105	86	100		
limb	Time to Climb (min.)/Altitude		10/FL 100	15/FL 100	14/FL 100		
(5: )		Initial Gradient (ft./nm)	NA	787	730		
eiling (ft.)		Service	NA	20,000	18,500		
		TAS	163	127	160		
	Long Range	Fuel Flow	50	78	71		
		Recommended	Altitude	FL 120	FL 120	FL 080	
			Recommended	Specific Range	3.260	1.628	2.254
				TAS	186	135	167
Cruise				Fuel Flow	84	88	86
		Altitude	FL 121	FL 080	FL 080		
		Specific Range	2.214	1.534	1.942		
		TAS	196	142	174		
	High Speed	Fuel Flow	114	101	94		
		Altitude	FL 080	FL 060	FL 080		
		Specific Range	1.719 1,075	1.406	1.851		
		Nautical Miles	1,075	487 124	217		
	Seats Full	Average Speed	438	339	153		
ondoo		Fuel Used Specific Range/Altitude			115		
anges		Nautical Miles	2.454/FL 121 1,465	1.437/FL 120 690	1.887/FL 040 859		
		Average Speed	1,465	125	859 159		
	Tanks Full		558	464	403		
		Fuel Used					
		Specific Range/Altitude Runway	2.625/FL 121 1,230	1.487/FL 120 1,860	2.132/FL 080 1,664		
		·	1,230				
	200 nm	Block Time Fuel Used	1115	1+38 157	1+11 130		
lingiana							
lissions		Specific Range/Altitude	1.739/FL 050 1,290	1.274/FL 120	1.538/FL 060		
occupants)		Runway	· · · · · · · · · · · · · · · · · · ·	1,860	1,870		
	500 nm	Block Time	2+58	3+55	2+54		
		Fuel Used	221	339	304		
		Specific Range/Altitude	2.262/FL 100	1.475/FL 120	1.645/FL 060		
		Suggested Base Price	\$689,000	\$726,960	\$815,000		
Remarks	Certification Basis		CAR 3/FAR 23, 1955/94; STC SA02483CH Includes Garmin G1000; composite fuselage shell with left and right doors.	FAR 23 A 54 Includes Garmin G500. All data preliminary. 2016 data.	CAR 3, 1956/69/83/2005 A/C system standard; Garmin G1000 NXi.		

### Single-Engine Pistons turbocharged

Manufacturer			Cirrus	Textron Aviation	Textron Aviation	GippsAero	Mooney	
Model			SR22T SR 22	Cessna Turbo Stationair HD	Cessna TTx CE-T240	GA8 Airvan TC GA8-320 TC	Acclaim Ultra	
BCA Equipped	Prico		\$639,900	<b>CE-T206H</b> \$665,000	\$715,000	\$761,030	<b>MO20V</b> \$769,000	
CA Equipped	FIICE	Seating	1+3/4	1+5/5	1+3/3	1+6/7	1+3/3	
Character-		Wing Loading	23.5	21.8	25.5	20.7	19.2	
stics		Power Loading	11.43	12.22	11.61	13.13	12.03	
		Noise (dBA)	80.3	82.6	81.4	85.4	78.0	
xternal		Length	26.0	28.3	25.2	28.3	26.9	
imensions		Height	8.9	9.3	9.0	9.3	8.3	
ft.)		Span	38.3	36.0	36.0	36.0	36.4	
nternal		Length	8.0	9.3	7.9	11.6	8.1	
Dimensions		Height	4.1	4.1	4.1	3.7	3.7	
ft.)		Width	4.1	3.7	4.0	4.2	3.6	
		Engine	Cont TSIO-550-K	Lyc TIO-540-AJ1A	Cont TSIO-550-C	Lyc TIO-540-AH1A	Cont TSIO-550-G	
Power		Output (hp)	315	310	310	320	280	
		Inspection Interval	2,000t	2,000t	2,000t	1,800t	2,200t	
		Max Ramp	3,610	3,806	3,600	4,214	3,374	
		Max Takeoff	3,600	3,789	3,600	4,200	3,368	
		Max Landing	3,600	3,600	3,420	4,000	3,200	
		Zero Fuel	3,400c	3,618b	3,300c	4,053b	3,173b	
		EOW	2,342	2,336	2,535	2,349	2,378	
Veights (lb.)		Max Payload	1,058	1,282	765	1,704	795	
		Useful Load	1,268	1,470	1,065	1,865	996	
		Max Baggage	130	180	120	180	120	
		Max Fuel	552	522	612	540	612	
		ilable Payload w/Max Fuel	716	948	453	1,325	384	
	Ava	ilable Fuel w/Max Payload	210	188	300	161	201	
imita		VNE	205	182	230	185	195	
imits.		Vno Va	176 140	149 125	181 158	143 121	174 127	
		TO (SL elev./ISA Temp.)	1,517	1,970	1,900	1,840	1,900	
Airport		TO (5,000-ft. elev.@25C)	2,268	2,845	2,460	2,788	3,300	
Perfor-		Vs0	64	59	61	61	60	
nance		Vx	88	70	82	71	80	
		Vy	103	88	110	81	105	
Oliver Iv	Tin	ne to Climb (min.)/Altitude	7/FL 100	12/FL 100	7/FL 100	13/FL 100	7/FL 100	
Climb		Initial Gradient (ft./nm)	782	724	701	825	770	
Ceilings (ft.)		Certificated	25,000	25,000	25,000	20,000	25,000	
Jenniga (rt.)		Service	25,000	27,000	25,000	20,000	25,000	
	TAS		171	137	208	125	215	
	Long Range	Long Range	Fuel Flow	76	85	78	68	99
		Altitude	FL 250	FL 240	FL 250	FL 200	FL 250	
		Specific Range	2.250	1.612	2.667	1.838	2.172	
		TAS Fuel Flow	98	155 99	227 130	130 78	227 128	
Cruise	Recommended	Altitude	FL 250	FL 240	FL 250	FL 200	FL 180	
		Specific Range	2.051	1.566	1.746	1.667	1.773	
		TAS	213	164	235	135	242	
		Fuel Flow	110	114	152	98	130	
	High Speed	Altitude	FL 250	FL 200	FL 250	FL 200	FL 250	
		Specific Range	1.936	1.439	1.546	1.378	1.862	
		Nautical Miles	1,021	512	666	233	500	
	Seats Full	Average Speed	171	137	202	125	178	
	Jeats I un	Fuel Used	486	387	345	220	259	
Ranges		Specific Range/Altitude	2.101/FL 250	1.323/FL 200	1.930/FL 250	1.059/FL 200	1.931/FL 160	
		Nautical Miles	1,021	655	1,270	618	1,122	
	Tanks Full	Average Speed	171	138	204	125	200	
		Fuel Used	486	459	572	459	539	
		Specific Range/Altitude	2.101/FL 250	1.427/FL 240	2.220/FL 250	1.346/FL 200	2.082/FL 250	
		Runway	1,405	1,396	1,730	1,743	1,300	
	200 nm	Block Time Fuel Used	1+08 197	1+23 163	1+03 159	1+35 125	1+05 139	
Missions		Specific Range/Altitude	1.015/FL 100	1.227/FL 150	1.258/FL 150	1.600/FL 120	1.439/FL 120	
4 occupants)		Runway	1,699	1,597	1,880	1,743	1,380	
1 occupants)		Block Time	2+28	3+22	2+24	3+30	2+54	
	500 nm	Fuel Used	360	385	338	373	259	
		Specific Range/Altitude	1.389/FL 180	1.299/FL 240	1.479/FL 250	1.340/FL 200	1.931/FL 250	
		Suggested Base Price	\$639,900	\$665,000	\$715,000	\$597,500	\$769,000	
Remarks		Certification Basis	FAR 23, 2010 Includes Garmin Perspective+ Global avionics.	FAR 23, 1998 Utility version w/2,183-lb. EOW, \$658,650; Garmin G1000 NXi w/GFC 700 a/p; new interior.	FAR 23 Includes Garmin G2000, SVT, AP, TAWS, TAS, ESP, A/C, Ti LE, leather.	FAR 23, 1998 Garmin G500; KC 225 All data preliminary. 2016 data.	CAR 3, 1955/89/2006 Includes Garmin G1000 new composite fuselag shell with left and right doors.	

# Single-Engine Pistons pressurized

Manufacturer		Piper Aircraft	Piper Aircraft	
			Matrix	M350
Model			PA-46R-350	PA-46-350P
<b>BCA</b> Equipped	Price		\$916,680	\$1,178,610
01		Seating	1+4/5	1+4/5
Character- istics		Wing Loading Power Loading	24.8 12.40	24.8 12.40
131103		Noise (dBA)	81.0	81.0
External		Length	28.9	28.9
Dimensions		Height	11.3	11.3
(ft.)		Span	43.0	43.0
Internal		Length	12.4	12.4
Dimensions		Height	3.9	3.9
(ft.)		Width	4.2	4.2
		Engine	Lyc TIO-540-AE2A	Lyc TIO-540-AE2A
Power		Output (hp)	350	350
		Inspection Interval	2,000t	2,000t
		Max Ramp	4,358	4,358
		Max Takeoff	4,340	4,340
		Max Landing	4,123	4,123
		Zero Fuel EOW	4,123c 2,969	4,123c 3,146
Weights (lb.)		Max Payload	1,154	977
-Holbino (ib.)		Useful Load	1,389	1,212
		Max Baggage	200	200
		Max Fuel	720	720
	Avai	lable Payload w/Max Fuel	669	492
		lable Fuel w/Max Payload	235	235
		Vne	198	198
Limits		Vno	168	168
LITTICS		VA	133	133
		PSI	5.5	5.5
		TO (SL elev./ISA temp.)	2,090	2,090
Airport		TO (5,000-ft. elev.@25C)	2,977 58	2,977 58
Performance		Vso Vx	81	81
		VX Vy	110	110
	Tim	e to Climb (min.)/Altitude	8/FL 100	8/FL 100
Climb		Initial Gradient (ft./nm)	703	703
		Certificated	25,000	25,000
Ceilings (ft.)		Service	25,000	25,000
		Sea-Level Cabin	_	12,300
		TAS	156	156
	Long Range	Fuel Flow	66	66
		Altitude	FL 250	FL 250
		Specific Range TAS	2.364	2.364
		Fuel Flow	108	108
Cruise	Recommended	Altitude	FL 250	FL 250
		Specific Range	1.880	1.880
		TAS	213	213
	High Count	Fuel Flow	120	120
	High Speed	Altitude	FL 250	FL 250
		Specific Range	1.775	1.775
		Nautical Miles	867	535
	Seats Full	Average Speed	151 457	138
Ranges		Fuel Used Specific Range/Altitude	457 1.897/FL 200	312 1.715/FL 120
Ranges		Nautical Miles	1,343	1,343
		Average Speed	158	159
	Tanks Full	Fuel Used	658	670
		Specific Range/Altitude	2.041/FL 250	2.004/FL 250
		Runway	2,090	2,090
	200 nm	Block Time	1+07	1+06
	2031111	Fuel Used	168	167
Missions		Specific Range/Altitude	1.190/FL 120	1.198/FL 200
(4 occupants)		Runway	2,090	2,090
	500 nm	Block Time Fuel Used	2+31 350	2+31 350
		Specific Range/Altitude	1.429/FL 250	1.429/FL 250
		Suggested Base Price	\$916,680	\$1,178,610
Remarks		Certification Basis	FAR 23, 1983/88 Garmin G1000; FIKI optional.	FAR 23, 1983/88 Garmin G1000; FIKI optional.
	I			

# Multiengine Pistons normally aspirated

Manufacturer			Vulcanair SpA	Vulcanair SpA
Model			P.68C P 68C	Victor P 68R
BCA Equipped	Price		\$830,800	\$848,200
- Equipped	11100	Seating	1+5/6	1+5/6
Character-		Wing Loading	22.9	22.7
stics		Power Loading	11.49	11.37
		Noise (dBA)	74.7	78.8
External		Length	31.3	31.3
Dimensions		Height	11.2	11.2
ft.)		Span	39.4	39.4
nternal		Length	10.6	10.6
Dimensions		Height	3.9	3.9
ft.)		Width	3.8	3.8
		Engines	2 Lyc IO-360-A1B6	2 Lyc IO-360-A1B6
Power		Output (hp each)	200	200
		Inspection Interval	2,000t	2,000t
		Max Ramp	4,630	4,548
		Max Takeoff	4,594	4,548
		Max Landing	4,365	4,321
		Zero Fuel	4,167c	4,374b
/eights (lb.)		EOW	3,153	3,197
roiginto (ib.)		Max Payload	1,014	1,177
		Useful Load	1,477	1,351
		Max Fuel	1,063	1,063
		lable Payload w/Max Fuel	415	289
	Avai	lable Fuel w/Max Payload	463	174
imito		VNE	194	197
imits		Vno	154	157
		TO (SL elev./ISA Temp.)	132 1,312	127 1,260
		TO (5,000-ft. elev.@25C)	4,000	4,000
		A/S (SL elev./ISA)	2,150	1,410
irport	4	/S (5,000-ft. elev.@25C)	2,950	2,370
erformance		VMCA	60	60
criorinance		VDEC	70	70
		Vxse	82	82
		Vyse	88	88
	Tim	e to Climb (min.)/Altitude	12/FL 100	12/FL 100
limb	Init	ial Engine-Out Rate (fpm)	217	217
IIIII	Initial All	-Engine Gradient (ft./nm)	1,100	920
	Initial Eng	ine-Out Gradient (ft./nm)	147	147
		Certificated		
eilings (ft.)		All-Engine Service	18,000	20,000
		Engine-Out Service	5,000	5,650
		TAS	144	144
	Long Range	Fuel Flow Altitude	94	94
			FL 080	FL 080
		Specific Range TAS	1.532 155	1.532 155
		Fuel Flow	108	108
Cruise	Recommended	Altitude	FL 080	FL 080
		Specific Range	1.435	1.435
		TAS	162	162
		Fuel Flow	116	116
	High Speed	Altitude	FL 080	FL 080
		Specific Range	1.397	1.397
		Nautical Miles	300	300
	Max Payload	Average Speed	140	140
	i ujibau	Trip Fuel	315	315
anges		Specific Range/Altitude	0.952/FL 080	0.952/FL 080
		Nautical Miles	1,000	1,000
	Ferry	Average Speed	145	145
		Trip Fuel	975	975
		Specific Range/Altitude Runway	1.026/FL 080	1.026/FL 080
		Block Time	1,450 1+28	1,450 1+28
	200 nm	Fuel Used	140	140
lissions		Specific Range/Altitude	1.429/FL 080	1.429/FL 080
occupants)		Runway	1,500	1,500
- coupanto)		Block Time	3+25	3+25
	500 nm	Fuel Used	375	375
		Specific Range/Altitude	1.333/FL 080	1.333/FL 080
		Suggested Base Price	\$830,800	\$848,200
emarks		Certification Basis	FAR 23, 1976/80 Garmin G950; STEC 55X DFCS. BCA estimated	EASA 23, 2009 Garmin G950; STEC 55X DFCS. BCA estimated
			data.	data.

# Multiengine Pistons normally aspirated

Manufacturer		-	Textron Aviation
Model			Beechcraft Baron G58
	Price		<b>G58</b>
<b>BCA</b> Equipped	Price	Seating	\$1,400,000 1+4/5
Character-		Wing Loading	27.6
istics		Power Loading	9.17
		Noise (dBA)	77.6
External		Length Height	29.8
Dimensions		9.8	
(ft.) Internal		37.8 12.6	
Dimensions		4.2	
(ft.)		Height Width	3.5
(10.)			2 Cont
Power		Engines	10-550-C
		Output (hp each)	300
		Inspection Interval Max Ramp	1,900t 5,524
		Max Takeoff	5,500
		Max Landing	5,400
		Zero Fuel	5,215b
Weights (lb.)		EOW	3,970
		Max Payload	1,245
		Useful Load Max Fuel	1,554 1,164
	Δ	wax Fuel wailable Payload w/Max Fuel	390
		vailable Fuel w/Max Payload	309
		Vne	223
Limits		195	
		VA VA	165
		TO (SL elev./ISA Temp.) TO (5,000-ft. elev.@25C)	2,345
		4,144 3,009	
Airport		4,335	
Performance		84	
		VDEC	85
		Vxse	100
		Vyse	101
		Time to Climb (min.)/Altitude	10/FL 100 390
Climb		Initial Engine-Out Rate (fpm) I All-Engine Gradient (ft./nm)	988
		Engine-Out Gradient (ft./nm)	232
		Certificated	_
Ceilings (ft.)		All-Engine Service	20,688
		Engine-Out Service	7,284
		TAS	185
	Long Range	Fuel Flow Altitude	144 FL 080
		Specific Range	1.285
		TAS	192
Cruise	Recommended	Fuel Flow	174
Cruise	Recommended	Altitude	FL 080
		Specific Range	1.103
		TAS	200
	High Speed	Fuel Flow Altitude	190 FL 080
		Specific Range	1.053
		Nautical Miles	333
	Max Payload	Average Speed	178
ъ.	mux r uyiouu	Trip Fuel	293
Ranges		Specific Range/Altitude Nautical Miles	1.137/FL 040
		Average Speed	1,480
	Ferry	Trip Fuel	1,081
		Specific Range/Altitude	1.369/FL 120
		Runway	2,862
	200 nm	Block Time	1+02
Missions		Fuel Used Specific Range/Altitude	226 0.885/FL060
(4 occupants)		Runway	0.885/FL 060 2,941
( r oooupunto)	F00	Block Time	2+31
	500 nm	Fuel Used	531
		Specific Range/Altitude	0.942/FL 060
		Suggested Base Price	\$1,400,000
Remarks		Certification Basis	CAR 3, 1957/69/83/2005 A/C system standard; Garmin G1000.

#### Multiengine Pistons turbocharged

		ONS turbochar		
Manufacture	r		Vulcanair SpA P 68C-TC	Piper Aircraft Seneca V
Model			P 68C-TC	PA-34-220T
<b>BCA</b> Equipped	l Price		\$877,500	\$999,900
Character		Seating Wind Loading	1+5/5	1+4/5
Character- istics		Wing Loading Power Loading	20.7 10.94	22.8 10.80
151105		Noise (dBA)	74.7	75.6
External		Length	31.3	28.6
Dimensions		Height	11.2	9.9
(ft.)		Span	39.4	38.9
Internal		Length	10.6	10.4
Dimensions		Height	3.9	3.6
(ft.)		Width	3.8	4.1
		Engines	2 Lyc TIO-360-C1A6D	2 Cont TSIO-360-RB
Power		Output (hp each)	210	220
		Inspection Interval	2,000t	1,800t
		Max Ramp	4,630	4,773
		Max Takeoff	4,594	4,750
		Max Landing Zero Fuel	4,365 4,140b	4,513 4,479c
		EOW	3,197	3,491
Weights (lb.)		Max Payload	943	988
		Useful Load	1,433	1,282
		Max Fuel	1,062	732
		ailable Payload w/Max Fuel	371	550
	Ava	ailable Fuel w/Max Payload	490	294
Limita		VNE	194	204
Limits		Vno Va	154 132	164 139
		TO (SL elev./ISA temp.)	1,260	1,707
		TO (5,000-ft. elev.@25C)	2,200	2,435
Airport		A/S (SL elev./ISA)	1,800	2,510
Perfor-		A/S (5,000-ft. elev.@25C)	2,400	3,117
mance		VMCA	66	66
manoo		VDEC	NA 70	73
		Vxse Vyse	78 88	83 88
	Tir	me to Climb (min.)/Altitude	10/FL 100	7/FL 100
au .		itial Engine-Out Rate (fpm)	240	253
Climb		II-Engine Gradient (ft./nm)	1,400	996
	Initial En	gine-Out Gradient (ft./nm)	NA	173
		Certificated	20,000	25,000
Ceilings (ft.)		All-Engine Service	20,000	25,000
		Engine-Out Service TAS	10,000 144	16,500 167
		Fuel Flow	104	108
	Long Range	Altitude	FL 080	FL 230
		Specific Range	1.385	1.546
		TAS	155	196
Cruise	Recommended	Fuel Flow	125	144
		Altitude	FL 080	FL 250
		Specific Range TAS	1.240 162	1.361 200
	mag.	Fuel Flow	150	156
	High Speed	Altitude	FL 080	FL 230
		Specific Range	1.080	1.282
		Nautical Miles	1,100	866
Range	Ferry	Average Speed	145	160
		Trip Fuel Specific Range/Altitude	960 1.146/FL 080	648 1.336/FL 180
		Runway	1.140/ FL 000 NA	1,520
	200	Block Time	1+28	1+10
	200 nm	Fuel Used	260	213
Missions		Specific Range/Altitude	0.769/FL 080	0.939/FL 120
(4 occupants)		Runway	NA 2 : 25	1,610
	500 nm	Block Time	3+25	2+41
		Fuel Used Specific Range/Altitude	485 1.031/FL 080	476 1.050/FL 200
		Suggested Base Price	\$877,500	\$999,900
Remarks		Certification Basis	FAR 23, 1982 Garmin G950 glass cockpit: STEC 55X DFGS. BCA estimated data.	FAR 23, 1971/80/97 Garmin G1000 standard.

# **Single-Engine Turboprops**

			A1 1-	0		.,	
1odel			Airvan 10 GA-10	Cessna Caravan CE-208	M500 PA-46-500TP	Kodiak Kodiak 100	Cessna Grand Caravan CE-208B
<b>CA</b> Equipped	Price		\$999,500*	\$1,950,000	\$1,999,900	\$2,454,725	\$2,527,900
		Seating	1+9/—	1+9/13*	1+4/5	1+6/9	1+9/13*
haracter-		Wing Loading	28.6	28.6	27.8	30.2	31.3
tics		Power Loading	10.56	11.85	10.18	9.67	10.16
kternal		Noise (dBA) Length	79.0 33.5	79.0 37.6	76.8 29.6	84.4 33.8	84.1 41.6
mensions		Height	12.7	14.9	11.3	15.3	14.8
:.)		Span	40.6	52.1	43.0	45.0	52.1
ternal		Length	16.1	12.7	12.3	15.8	16.7
imensions		Height	3.8	4.5	3.9	4.8	4.5
t.)		Width	4.2	5.3	4.1	4.5	5.3
<i></i>		Engine	RR	P&WC	P&WC	P&WC	P&WC
ower		-	M250 B-17F/2	PT6A-114A	PT6A-42A	PT6A-34	PT6A-140
31101		Output (shp)/Flat Rating	450/ISA+31C	675/ISA+31C	500/ISA+55C	750/ISA+7C	867/ISA+24C
		Inspection Interval Max Ramp	3,500t 4,775	3,600t 8,035	3,600t 5,134	4,000t 7,305	3,600t 8,842
		Max Takeoff	4,750	8,000	5,092	7,255	8,807
		Max Landing	4,750	7,800	4,850	7,255	8,500
		Zero Fuel	4,182b	7,432b	4,850c	6,490c	8,152b
.:		BOW	2,475	4,930	3,634	4,417	5,510
eights (lb.)		Max Payload	1,707	2,502	1,216	2,073	2,642
		Useful Load	2,300	3,105	1,500	2,888	3,332
		Max Fuel	1,025	2,224	1,160	2,144	2,246
		ailable Payload w/Max Fuel	1,275	881	340	744	1,086
	Av	ailable Fuel w/Max Payload	594	604	284	815	691
		VMO	175	175	188	180	175
nits		VA	150	150	127	143	148
		PSI PSI		_	5.6	- 4.400	- 0.400
port		TO (SL elev./ISA temp.) TO (5,000-ft. elev.@25C)	1,600 2,973	2,055 2,973	2,438 3,691	1,468 2,396	2,160 3,661
rfor-		Vso	61	61	69	60	61
ance		Vx Vx	90	90	95	73	86
arice		V	107	107	125	101	108
	Ti	me to Climb (min.)/Altitude	9/FL 100	9/FL 100	19/FL 250	9/FL 100	9/FL 100
imb		Initial Gradient (ft./nm)	771	771	753	915	816
Ceilings (ft.)		Certificated	20,000	25,000	30,000	25,000	25,000
		Service	25,000	25,000	30,000	25,000	25,000
		Sea-Level Cabin	_	_	12,600	_	_
		TAS	157	157	179	164	156
	Long Range	Fuel Flow	281	281	135	251	328
	Long Kange	Altitude	FL 100	FL 100	FL 280	220	FL 100
ruise		Specific Range	0.559	0.559	1.326	0.653	0.476
		TAS	186	186	258	175	185
	High Speed	Fuel Flow Altitude	379 FL 100	379 FL 100	242 FL 280	335 FL 120	437 FL 100
		Specific Range	0.491	0.491	1.066	0.522	0.423
		Nautical Miles	965	288	834	1,005	291
	Full Fuel	Average Speed	156	153	171	175	155
BAA IFR	(with available	Trip Fuel	1,795	581	748	2,130	676
inges	payload)	Specific Range/Altitude	0.538/FL 100	0.496/FL 100	1.115/FL 280	0.472/120	0.430/FL 100
00-nm		Nautical Miles	970	970	834	1,236	816
ernate)	F	Average Speed	156	156	171	164	156
, i	Ferry	Trip Fuel	1,800	1,800	748	2,130	1,772
		Specific Range/Altitude	0.539/FL 100	0.539/FL 100	1.115/FL 280	0.580/FL 200	0.460/FL 100
		Runway	1,468	1,468	1,550	1,468	1,428
	300 nm	Flight Time	1+40	1+40	1+22	1+47	1+41
	2.2	Fuel Used	648	648	379	587	750
		Specific Range/Altitude	0.463/FL 100	0.463/FL 100	0.792/FL 280	0.511/FL 120	0.400/FL 100
ssions		Runway	1,675	1,675	1,625	1,468	1,792
passen-	600 nm	Flight Time	3+17	3+17	2+32	3+30	3+19
rs)		Fuel Used Specific Range/Altitude	1,260 0.476/FL 100	1,260 0.476/FL 100	660 0.909/FL 280	1,140 0.526/FL 120	1,462 0.410/FL 100
		Runway	NP	0.476/FL 100 NP	1,700	1,467	0.410/FL 100 NP
		Flight Time	NP	NP NP	4+18	5+47	NP
	1,000 nm	Fuel Used	NP	NP	985	1,878	NP
		Specific Range/Altitude	NP	NP	1.015/FL 280	0.532/FL 120	NP
		Suggested Base Price	NA	NA	\$1,999,900	\$2,075,000	NA
emarks		Certification Basis	FAR 23, 1984/98 *BCA estimated price. Garmin G1000 with GFC700 autopilot. 2016 data.	FAR 23, 1984/98 *Export only. Garmin G1000 with GFC700 autopilot.	FAR 23 A 52 *1,000 nm, 3 passengers. Garmin G1000 with SVS.	FAR 23, 2007 Normal category Includes Garmin G1000; GFC700 with coupled GA; Summit interior option.	FAR 23, 1986/201 *Export only. Includes cargo poo Garmin G1000 wit GFC700 autopilot

# **Single-Engine Turboprops**

/lanufacturer			Piper Aircraft	Epic Aircraft	Daher	Daher	Pilatus
/lodel			M600 PA-46-600TP	Epic E1000	TBM 910 TBM 700 N	TBM 930 TBM 700 N	PC-12 NG PC-12/47E
<b>CA</b> Equipped	l Price		\$2,899,000	\$2,995,000	\$3,683,260	\$3,979,750	\$4,923,000
OH Equipped	111100	Seating	1+4/5	1+5/6	1+5/6	1+5/6	1+7/10
Character-		Wing Loading	28.7	36.9	38.2	38.2	37.6
stics		Power Loading	10.00	6.25	8.70	8.70	8.71
		Noise (dBA)	76.8	76.0	76.2	76.2	77.0
xternal		Length	29.6	35.8	35.2	35.2	47.3
Dimensions		Height	11.3	12.5	14.3	14.3	14.0
ft.)		Span	43.2	43.0	42.1	42.1	53.3
nternal		Length	12.3	10.5	15.0	15.0	16.9
Dimensions		Height	3.9	4.9	4.1	4.1	4.8
ft.)		Width	4.1	4.6	4.0	4.0	5.0
		Engine	P&WC	P&WC	P&WC	P&WC	P&WC
ower		_	PT6A-42A	PT6A-67A	PT6A-66D 850/ISA+49C	PT6A-66D 850/ISA+49C	PT6A-67P
		Output (shp)/Flat Rating Inspection Interval	600/ISA+55C 3,600t	1,200/ISA+35C 3,500t	3,500t	3,500t	1,200/ISA+35C 3,500t
		Max Ramp	6,050	7,500	7,430	7,430	10,495
		Max Takeoff	6,000	7,500	7,394	7,394	10,450
		Max Landing	5,800	7,500	7,024	7,024	9,921
		Zero Fuel	4,850c	5,400c	6,032c	6,032c	9,039c
		BOW	3,850	4,600	4,829	4,829	6,782
eights (lb.)		Max Payload	1,000	800	1,203	1,203	2,257
		Useful Load	2,200	2,900	2,601	2,601	3,713
		Max Fuel	1,742	1,876	2,001	2,001	2,704
	Ava	ailable Payload w/Max Fuel	458	1,024	584	584	1,009
		ailable Fuel w/Max Payload	1,200	2,100	1,398	1,398	1,456
	Ave	VMO	250	280	266	266	240
mits		VA VA	151	170	160	160	163
		PSI	5.6	6.7	6.2	6.2	5.8
		TO (SL elev./ISA temp.)	2,635	1,600	2,380	2,380	2,600
irport		TO (5,000-ft. elev.@25C)	3,998	NA NA	3,475	3,475	4,270
erfor-		Vso	62	65	65	65	67
ance		Vx	95	124	100	100	120
		Vy	122	144	124	124	130
	Ti	me to Climb (min.)/Altitude	21/FL 250	10/FL 250	13/FL 250	13/FL 250	20/FL 250
imb		Initial Gradient (ft./nm)	785	1,500	1,000	1,000	860
		Certificated	30,000	34,000	31,000	31,000	30,000
eilings (ft.)		Service	30,000	34,000	31,000	31,000	30,000
		Sea-Level Cabin	12,600	18,000	14,390	14,390	13,100
		TAS	184	265	252	252	225
	Lang Danga	Fuel Flow	155	268	241	241	268
	Long Range	Altitude	FL 280	FL 280	FL 310	FL 310	FL 300
ruise		Specific Range	1.187	0.989	1.046	1.046	0.840
ruise		TAS	274	330	330	330	285
	High Speed	Fuel Flow	324	402	412	412	497
	mgn opccu	Altitude	FL 280	FL 280	FL 260	FL 260	FL 200
		Specific Range	0.846	0.821	0.801	0.801	0.573
	Full Fuel	Nautical Miles	1,406	1,650	1,514	1,514	1,608
	(with available	Average Speed	179	265	252	252	261
Baa ifr	payload)	Trip Fuel	1,324	1,599	1,599	1,599	2,282
anges		Specific Range/Altitude	1.062/FL 280	1.032/FL 310	0.947/FL 310	0.947/FL 310	0.705/FL 300
00-nm		Nautical Miles	1,406	1,594	1,594	1,594	1,650
ternate)	Ferry	Average Speed	179	252	252	252	264
	•	Trip Fuel	1,324	1,598	1,598	1,598	2,294
		Specific Range/Altitude	1.062/FL 280	0.997/FL 310	0.997/FL 310	0.997/FL 310	0.719/FL 300
		Runway	1,593	1,765	1,765	1,765	1,563
	300 nm	Flight Time	1+21	1+00	1+00	1+00	1+10
		Fuel Used	429 0.699/FL 280	440 0.682/FL 280	440 0.682/FL 280	440 0.682/FL 280	549 0.546/FL 260
		Specific Range/Altitude		2,005	2,005	2,005	0.546/FL 260 1.753
issions		Runway Flight Time	1,687 2+31	2,005 1+55	2,005 1+55	2,005 1+55	1,753 2+16
passen-	600 nm	Fugit time Fuel Used	735	830	830	830	975
rs)		Specific Range/Altitude	0.816/FL 280	0.723/FL 280	0.723/FL 280	0.723/FL 280	0.615/FL 270
		Runway	1,812	2,380	2,380	2,380	2,026
		Flight Time	4+06	3+10	3+10	3+10	3+46
	1,000 nm	Fuel Used	1,142	1,320	1,320	1,320	1,520
		Specific Range/Altitude	0.876/FL 280	0.758/FL 290	0.758/FL 290	0.758/FL 290	0.658/FL 280
		Suggested Base Price	\$2,899,000	NA	\$3,658,336	\$3,899,887	\$4,095,000
emarks		Certification Basis	FAR 23 A 62, 2016 Garmin G3000 with SVS and enhanced AFCS.	FAR 23 pending Garmin G1000 NXi.	FAR 23, 1990/2006/07/14 Pilot door standard; 5-blade propeller; G1000 NXi; AOA-ESP-USP; satcom; weather; 5-year system warranty.	FAR 23, 1990/2006/07/14 All features of TBM 900 plus advanced interior; Garmin G3000; 5-year system warranty.	FAR 23, 1996/2005/C Honeywell APEX avionics; SmartView ADS-B Out; BMW executive interior; Hartzell 5-blade propeller.

# $\textbf{Multiengine Turboprops} \leq 12,500\text{-LB. MTOW}$

			Vulcanair SpA Viator	Nextant Aerospace G90XT	Evektor Outback	Textron Aviation  Beechcraft King Air C90GTx
Model			AP68TP-600	C90	EV-55	C90ĞTi
BCA Equipped Price			\$2,485,900	\$2,750,000	\$3,000,000	\$3,595,000
		Seating Wing Loading	1+7/10 33.0	1+7/10 34.4	1+9/14 37.4	1+7/8 34.4
Characteristics		Power Loading	10.08	9.55	9.46	9.53
	Noise (dBA)		71.7	71.7	NA NA	74.8
External		Length	37.0	35.5	46.6	35.5
Dimensions (ft.)		Height	11.9	14.3	16.8	14.3
		Span	39.4	NA 10.1/10.1	53.2	50.3
Internal		<b>Length:</b> OA/Net Height	11.9/17.2 4.1	12.4/12.4 4.8	16.5/20.0 4.5	12.4/12.4 4.8
Dimensions (ft.)		Width: Max/Floor	3.7/3.7	4.5/4.1	5.3/4.7	4.5/4.1
			2 RR	2 GE Czech	2 P&WC	2 P&WC
Power		Engines	250 B17C	H75-100	PT6A-21	PT6A-135A
I OWEI	0	Output (shp each)/Flat Rating	328/ISA+25C	550/ISA+8C	536/ISA+15C	550/ISA+30C
		Inspection Interval Max Ramp	3,500t 6,669	4,000t 10,560	3,600t 10,207	3,600t 10,545
		Max Takeoff	6,613	10,500	10,207	10,485
		Max Landing	6,283	9,700	10,141	9,832
		Zero Fuel	5,621c	9,650c	9,810c	9,378c
Moighte (lb.)		BOW	3,850	7,200	5,965	7,265
Weights (lb.)		Max Payload	1,771	2,450	3,845	2,113
		Useful Load	2,819	3,360	4,242	3,280
		Max Fuel	1,487	2,573	3,413	2,573
		wailable Payload w/Max Fuel	1,332	787	829	707
	A	vailable Fuel w/Max Payload Vmo	1,048 200	910 208	397 205	1,167 226
Limits		VMO _ VA	141	169	140	169
		PSI	- 141	5.0		5.0
		TO (SL elev./ISA temp.)	2,034	2,100	1,378	1,984
		TO (5,000-ft. elev.@25C)	2,950	2,800	1,837	3,375
	A/S (SL elev,/ISA temp.) A/S (5,000-ft, elev,@25C) Vlvca Vocc Vssc		2,034	3,800	1,722	3,690
Airport			2,953	5,100	2,395	5,855
Performance			77	92 97	66	80 97
			85 90	101	79 92	100
		Vyse	105	111	95	108
		Time to Climb (min.)/Altitude	7/FL 100	18/FL 250	6/FL 010	18/FL 250
Olima In		Initial Engine-Out Rate (fpm)	270	460	290	460
Climb		All-Engine Gradient (ft./nm)	1,500	1,900	1,107	1,900
	Initial	Engine-Out Gradient (ft./nm)	180	260	219	260
		Certificated	25,000	30,000	24,000	30,000
Ceilings (ft.)		All-Engine Service	25,000	30,000	24,000	30,000
		Engine-Out Service Sea-Level Cabin	8,050	22,000	15,420	19,230
		Sea-Level Cabin TAS	169	11,065 213	 180	11,065 208
	Lang Danga	Fuel Flow	261	292	432	332
	Long Range	Altitude	FL 100	FL 280	FL 010	FL 260
Cruico		Specific Range	0.648	0.729	0.417	0.627
Cruise		TAS	214	283	220	270
	High Speed	Fuel Flow	375	578	610	612
		Altitude	FL 100	FL 240	FL 200	FL 200
		Specific Range Nautical Miles	0.571 543	0.490 324	0.361 NP	0.441 260
	Max Payload	Average Speed	180	203	NP NP	229
	(with available fuel)	Trip Fuel	781	600	NP NP	620
	(with available fact)	Specific Range/Altitude	0.695/FL 100	0.540/FL 220	NP/—	0.419/FL 270
		Nautical Miles	837	1,300	1,046	1,026
	Max Fuel	Average Speed	179	207	217	252
	(with available payload)	Trip Fuel	1,220	1,782	3,008	2,044
NBAA IFR Ranges	payloau)	Specific Range/Altitude	0.686/FL 100	0.730/FL 280	0.348/FL 100	0.502/FL 270
(100-nm alternate)		Nautical Miles	837	1,290	1,046	975
	Full Fuel	Average Speed	179	207	217	252
	(with 4 passsengers)	Trip Fuel	1,220 0.686/FL 100	1,769 0.729/FL 280	3,008 0.348/FL 100	1,949 0.500/FL 270
		Specific Range/Altitude Nautical Miles	0.686/FL 100 837	1,369	0.348/FL 100 1,051	1,045
		Average Speed	179	203	218	255
	Ferry	Trip Fuel	1,220	1,850	3,008	2,053
		Specific Range/Altitude	0.686/FL 100	0.740/FL 280	0.349/FL 100	0.509/FL 270
		Runway	1,247	3,010	3,163	3,004
	300 nm	Flight Time	1+35	1+06	1+26	1+13
	333 1111	Fuel Used	419	584	943	748
		Specific Range/Altitude	0.716/FL 100	0.514/FL 220	0.318/FL 100	0.401/FL 210
Missions		Runway Flight Time	1,558 3+18	3,350 2+12	1,289 2+22	3,347 2+22
4 passengers)	600 nm	Fuel Used	866	1,162	1,773	1,353
i passengers)		Specific Range/Altitude	0.693/FL 100	0.516/FL 280	0.338/FL 100	0.443/FL 230
		Runway	NP	3,500	1,565	3,690
	1,000 nm	Flight Time	NP	3+39	4+ 36	3+57
	1,000 HM	Fuel Used	NP	1,938	2,881	1,990
		Specific Range/Altitude	NP/NP	0.516/FL 280	0.347/FL 100	0.503/FL 270
		Suggested Base Price	\$2,485,900	NA	NA NA	NA 0400 40007
Remarks		Certification Basis	FAR 23, 1986 Garmin G950; STEC 2100 autopilot. BCA estimated data.	ST01902CH; SA3593NM; SA4010NM; SA3593NM; SA01902CH; SA01456WI-D; SA02133SE.	EASA/FAR 23 pending 2016 data.	CAR 3, 1959/2007 Pro Line Fusion standard.; STC SA10747SC weight increase; SA02054SE winglets; SA3593NM swept props; SA4010NM dual aft strakes; 1,000-nm mission, 755-lb. pld.

# **Multiengine Turboprops** ≤12,500-LB. MTOW

			Textron Aviation Beechcraft King Air 250	Viking Air 400 Series	Piaggio Aero Industries Avanti Evo
Model			B200GT	DHC-6-400	P180
RCA Equipped Price			\$5,995,000	\$6,500,000	\$7,695,000
		Seating   Wing Loading	1+8/10 40.3	1+11/19 29.8	1+7/9 70.3
Characteristics		Power Loading	7.35	10.08	70.3
		Noise (dBA)	TBD	85.6	75.0
xternal		Length	43.8	51.8	47.3
		Height	14.8	19.5	13.0
imensions (ft.)	Span		57.9	65.0	46.0
ternal		Length: OA/Net	16.7/16.7	18.4/24.5	17.5/17.5
imensions (ft.)		Height	4.8	4.9	5.8
mensions (re.)		Width: Max/Floor	4.5/4.1	5.4/4.4	6.1/3.5
		Engines	2 P&WC PT6A-52	2 P&WC PT6A-34	2 P&WC PT6A-66B
ower	,	Output (shp each)/Flat Rating	850/ISA+37C	620/ISA+27C	850/ISA+28C
	,	Inspection Interval	3,600t	3,600t	3,600t
		Max Ramp	12,590	12,525	12,150
		Max Takeoff	12,500	12,500	12,100
		Max Landing	12,500	12,300	11,500
		Zero Fuel	11,000c	11,655b	9,800c
oidhte (lh.)		BOW	8,830	8,100	8,375
eights (lb.)		Max Payload	2,170	3,555	1,425
		Useful Load	3,760	4,425	3,775
		Max Fuel	3,645	3,549	2,802
		wailable Payload w/Max Fuel	115	876	973
	A	wailable Fuel w/Max Payload	1,590	870	2,350
vito		VMO	260	170	260
nits		VA PSI	182 6.5	136	202 9.0
		TO (SL elev./ISA temp.)	2,111	1,490	3,262
		TO (SL elev./ISA temp.)	3,099	1,490 NA	3,262 4,700
		A/S (SL elev./ISA temp.)	3,687	2,220	5,750
port		A/S (5,000-ft. elev.@25C)	4,859	NA	7,400
rformance		VMCA	86	66	100
Tiomanice	Voec Vxse		94	NA	106
			115	NA	132
		Vyse	121	NA	140
Climb		Time to Climb (min.)/Altitude	13/FL 250	NA/FL 100	10/FL 250
		Initial Engine-Out Rate (fpm)	682	340	670
110		I All-Engine Gradient (ft./nm)	1,170	NA	1,106
	Initial	Engine-Out Gradient (ft./nm)	364	NA	287
		Certificated	35,000	25,000	41,000
Ceilings (ft.)		All-Engine Service	35,000	26,700	39,400
		Engine-Out Service	26,000 15,293	11,600	23,800 24,000
	Sea-Level Cabin TAS		256	— NA	318
		Fuel Flow	430	NA NA	408
	Long Range  High Speed	Altitude	FL 350	FL 100	FL 410
		Specific Range	0.595	NA NA	0.779
ruise		TAS	310	180	400
		Fuel Flow	750	580	792
		Altitude	FL 260	FL 100	FL 310
		Specific Range	0.413	0.310	0.505
		Nautical Miles	321	NP	1,070
	Max Payload	Average Speed	267	NP	315
	(with available fuel)	Trip Fuel	870	NP	1,715
		Specific Range/Altitude	0.369/FL 330	NP	0.624/FL 390
	Max Fuel	Nautical Miles	1,403	NA	1,450
	(with available	Average Speed	291	NA NA	311
AA IFR Ranges	payload)	Trip Fuel	2,941 0,477/FL 220	NA NA /EL 100	2,167
		Specific Range/Altitude Nautical Miles	0.477/FL 330 1,038	NA/FL 100 NA	0.669/FL 410 1,510
0-nm alternate)	Full Fuel	Average Speed	288	NA NA	317
	(with 4 passsengers)	Trip Fuel	2,225	NA NA	2,167
	, , p.20000118010)	Specific Range/Altitude	0.467/FL 330	NA/FL 100	0.697/FL 410
		Nautical Miles	1,420	NA NA	1,530
		Average Speed	293	NA	318
	Ferry	Trip Fuel	2,942	NA	2,167
		Specific Range/Altitude	0.483/FL 330	NA/FL 100	0.706/FL 410
		Runway	3,504	NA	2,350
	300 nm	Flight Time	1+03	NA	0+53
	300 11111	Fuel Used	869	NA	688
		Specific Range/Altitude	0.345/FL 250	NA/FL 100	0.436/FL 310
o i o n o		Runway	3,587	NA NA	2,550
ssions	600 nm	Flight Time	2+03	NA	1+44
assengers)		Fuel Used	1,494	NA NA (FL 100	1,144
		Specific Range/Altitude	0.402/FL 290	NA/FL 100	0.524/FL 350
		Runway Flight Time	3,677	NA NA	2,700
	1,000 nm	Flight Time Fuel Used	3+28 2,147	NA NA	3+02 1,603
		Specific Range/Altitude	0.466/FL 330	NA/FL 100	0.624/FL 390
		Suggested Base Price	NA	NA NA	\$7,395,000
marks		Certification Basis	FAR 23, 1973/80/2008/11 Rockwell Collins Pro Line Fusion standard; Wi-Fi optional; STC SA02131SE.	EASA/FAR 23 A 57, 2010 2016 data.	EASA 23, 2014; FAR 23, 2015 Includes Rockwell Collins Pro Lir 21 avionics; TCAS I; Iridium satcc RVSM approved; optional 390-ll capacity internal tank: \$275,00

# Multiengine Turboprops >12,500-LB. MTOW

Manufacture			Textron Aviation Beechcraft King Air 250 EP	Textron Aviation  Beechcraft King Air 350i	Textron Aviation  Beechcraft King Air 350HW	Textron Aviation Beechcraft King Air 350iER
Model			B200GT	B300	Band B300	B300ER
<b>BCA</b> Equipped	d Price		\$6,231,025	\$6,995,000	\$7,329,055	\$8,445,625
٠.		Seating	1+8/10	1+9/11	1+9/14	1+9/11
Character-		Wing Loading	43.3	48.4	53.2	53.2
istics		Power Loading Noise (dBA)	7.89 85.3	7.14 72.1	7.86 81.5	7.86 81.5
External		Length	43.8	46.7	46.7	46.7
Dimensions		Height	14.8	14.3	14.3	14.3
(ft.)		Span	57.9	57.9	57.9	57.9
Internal		Length: OA/Net	16.7/16.7	19.5/19.5	19.5/19.5	19.5/19.5
Dimensions		Height	4.8	4.8	4.8	4.8
(ft.)		Width: Max/Floor	4.5/4.1	4.5/4.1	4.5/4.1	4.5/4.1
(11.)			2 P&WC	2 P&WC	2 P&WC	2 P&WC
_		Engines	PT6A-52	PT6A-60A	PT6A-60A	PT6A-60A
Power	0	Output (shp each)/Flat Rating	850/ISA+37C	1,050/ISA+10C	1,050/ISA+10C	1,050/ISA+10C
		Inspection Interval	3,600t	3,600t	3,600t	3,600t
		Max Ramp	13,510	15,100	16,600	16,600
		Max Takeoff	13,420	15,000	16,500	16,500
		Max Landing Zero Fuel	12,500 11,000c	15,000 12,500c	15,675 13,000c	15,675 13,000c
		BOW	8,865	9,955	9,290	10,215
Weights (lb.)		Max Payload	2,135	2,545	3,710	2,785
		Useful Load	4,645	5,145	7,310	6,385
		Max Fuel	3,645	3,611	3,611	5,192
		Available Payload w/Max Fuel	1,000	1,534	3,699	1,193
	A	Available Fuel w/Max Payload	2,510	2,600	3,600	3,600
		MMO	0.58	0.58	0.58	0.58
Limits		Trans. Alt. FL/VMo	FL 210/259	FL 210/263	FL 240/245	FL 240/245
		VA PSI	182 6.5	182 6.6	182 6.6	182 6.5
		TO (SL elev./ISA temp.)	4,005	3,300	4,057	4,057
		TOFL (5,000-ft. elev.@25C)	5,780	5,376	5,140	7,675
Airport		Mission Weight	13,220	14,196	13,686	16,100
Perfor-		NBAA IFR Range	1,430	1,549	1,445	2,257
mance		V2	109	109	111	111
		VREF	97	100	104	104
		Landing Distance	2,780	2,390	2,720	2,728
Climb		Time to Climb (min.)/Altitude Initial Engine-Out Rate (fpm)	15/FL 250 580	15/FL 250 552	23/FL 250 274	18/FL 250 337
Climb		Engine-Out Gradient (ft./nm)	255	304	172	182
	TAIR 20 IIIIdai	Certificated	35,000	35,000	35,000	35,000
0-111-2- (61.)		All-Engine Service	35,000	35,000	35,000	35,000
Ceilings (ft.)	Engine-Out Service		24,400	21,500	17,100	17,100
		Sea-Level Cabin	15,293	15,293	15,293	15,293
		TAS	233	235	232	238
	Long Range	Fuel Flow	369	362	392	402
		Altitude	FL 350	FL 330	FL 330 0.592	FL 330
Cruise		Specific Range TAS	0.631 308	0.649 312	303	0.592 303
		Fuel Flow	750	773	766	764
	High Speed	Altitude	FL 260	FL 240	FL 240	FL 240
		Specific Range	0.411	0.404	0.396	0.397
		Nautical Miles	802	896	1,254	1,316
	Max Payload	Average Speed	275	273	258	261
	(with available fuel)	Trip Fuel	1,802	1,891	2,838	2,880
		Specific Range/Altitude	0.445/FL 330	0.474/FL 350	0.442/FL 350	0.457/FL 350
	Max Fuel	Nautical Miles	1,393	1,485	1,260	2,223
NBAA IFR	(with available	Average Speed Trip Fuel	283 2,947	280 2,944	258 2,884	269 4,528
Ranges	payload)	Specific Range/Altitude	0.473/FL 330	0.504/FL 350	0.437/FL 350	0.491/FL 350
(100-nm		Nautical Miles	1,414	1,533	1,437	2,271
alternate)	Full Fuel	Average Speed	285	285	276	271
ancornato)	(with 4 passsengers)	Trip Fuel	2,950	2,951	2,930	4,533
		Specific Range/Altitude	0.479/FL 330	0.519/FL 350	0.490/FL 350	0.501/FL 350
		Nautical Miles	1,442	1,560	1,473	2,338
	Ferry	Average Speed	289	289	282	276
		Trip Fuel	2,956	2,958	2,942 0.501/FL 350	4,543
		Specific Range/Altitude	0.488/FL 330 3,524	0.527/FL 350 2,586	0.501/FL 350 2,634	0.515/FL 350 2,795
		Runway Flight Time	3,524 1+05	2,586 1+02	2,634 1+06	2,795 1+05
	300 nm	Fuel Used	848	881	954	919
		Specific Range/Altitude	0.354/FL 250	0.341/FL 250	0.314/FL 250	0.326/FL 250
Missions		Runway	3,611	2,702	2,746	2,927
(4 passen-	600 nm	Flight Time	2+05	2+02	2+07	2+07
(4 passen- gers)	Joo mili	Fuel Used	1,472	1,470	1,561	1,529
5010)		Specific Range/Altitude	0.408/FL 290	0.408/FL 290	0.384/FL 290	0.392/FL 290
		Runway Flight Time	3,702	2,827 3+27	2,883	3,048 3+35
	1,000 nm	Flight Time Fuel Used	3+31 2,123	2,102	3+33 2,227	2,195
		Specific Range/Altitude	0.471/FL 330	0.476/FL 330	0.449/FL 330	0.456/FL 330
		Suggested Base Price	NA	NA	NA	NA
Remarks			FAR 23, 1973/80/2008/11 Commuter category Rockwell Collins Pro Line	FAR 23, 1989 Commuter category	FAR 23, 1989/2007 Commuter category 17,500-lb. MTOW optional; Rockwell Collins Pro Line	FAR 23, 1989/2007 Commuter category
		Certification Basis	Fusion; Wi-Fi optional; STC SA11103SC for IGW; 14,000-lb. MTOW also available.	Rockwell Collins Pro Line Fusion; Wi-Fi standard; RVSM approved.	Fusion; Wi-Fi standard; factory-installed Slick interior available for special missions; RVSM approved.	Rockwell Collins Pro Line Fusion; Wi-Fi standard; RVSM approved.

Manufacturer			Cirrus Design	Eclipse Aerospace	
/lodel			Vision SF-50	Eclipse 550 EA-500	
<b>CA</b> Equipped	Price		\$1,960,000	\$2,995,000	
en Equippou	11100	Seating	1+4/6	1+4/5	
haracter-		Wing Loading	30.7	41.0	
ics		Power Loading	1.67	3.33	
rt o wo o l	Noise (EPNdB)	: Lateral/Flyover/Approach	NA/NA/NA 30.7	69.2/78.9/81.9 33.5	
ternal		Length Height	10.9	11.0	
mensions		Span	38.7	37.9	
.) ternal		Length: OA/Net	11.5/9.8	12.3/10.0	
mensions	н	eight/Dropped Aisle Depth	4.1/NA	4.2/NA	
.)		Width: Max/Floor	5.1/3.1	4.7/3.0	
		Internal: Cu. ft./lb.	24/NA	16/260	
ggage		External: Cu. ft./lb.	30/NA	NA/NA	
		Engine(s)	1 Wms Intl	2 P&WC	
wer			FJ33-5A	PW610F	
		utput (lb. each)/Flat Rating Manu. Service Plan Interval	1,800/ISA+10C 3,500t/—	900/ISA+10C 3,500t/—	
	inspection interval/	Max Ramp	6,040	6,034	
		Max Takeoff	6,000	6,000	
		Max Landing	5,550	5,600	
		Zero Fuel	4,900c	4,922c	
eights (lb.)		BOW	3,772	3,923	
J ()		Max Payload Useful Load	1,128 2,268	999 2,111	
		Max Fuel	2,208	1,680	
	Av	ailable Payload w/Max Fuel	268	431	
		ailable Fuel w/Max Payload	1,140	1,112	
		Ммо	0.530	0.640	
nits		Trans. Alt. FL/VM0	FL 183/250	FL 200/285	
		TOFL (SL elev./ISA temp.)	6.4 2,036	8.7 2,394	
		TOFL (5,000-ft. elev.@25C)	3,679	4,171	
rport		Mission Weight	6,000	5,893	
erfor-		NBAA IFR Range	1,125	1,015	
ance		V2	90	102*	
		VREF	87 1,628		
		Landing Distance Time to Climb/Altitude	1,628 NA/FL 370	2,340 25/FL 370	
imb	FAF	R 25 Engine-Out Rate (fpm)	NA NA	500	
		ngine-Out Gradient (ft./nm)	NA	294	
		Certificated	28,000	41,000	
eilings (ft.)		All-Engine Service	28,000	41,000	
		Engine-Out Service Sea-Level Cabin	NA NA	25,000 21,500	
		TAS	256	334	
	Laure Daniel	Fuel Flow	358	321	
	Long Range	Altitude		FL 280	FL 410
ruise		Specific Range	0.715	1.040	
		TAS Fuel Flow	300 466	369 462	
	High Speed	Altitude	FL 280	FL 350	
		Specific Range	0.644	0.799	
		Nautical Miles	550	530	
	Max Payload	Average Speed	251	307	
	(with available fuel)	Trip Fuel	845	677	
		Specific Range/Altitude	0.651/FL 280	0.783/FL 410	
	Max Fuel	Nautical Miles	1,167	1,125	
BAA IFR	(with available	Average Speed	248 1,602	319 1,254	
inges	payload)	Specific Range/Altitude	0.728/FL 280	0.897/FL 410	
00-nm		Nautical Miles	796	825	
ernate)	Four Passengers	Average Speed	250	317	
	(with available fuel)	Trip Fuel	1,076	965	
		Specific Range/Altitude	0.740/FL 280	0.855/FL 410	
		Nautical Miles Average Speed	1,219 218	1,190 312	
	Ferry	Trip Fuel	1,680	1,263	
		Specific Range/Altitude	0.726/FL 280	0.942/FL 410	
		Runway	1,857	2,038	
	300 nm	Flight Time	1+10	0+58	
	COO IIIII	Fuel Used	568	456	
		Specific Range/Altitude	0.528/FL 280	0.658/FL 350	
ssions		Runway Flight Time	2,171 2+15	2,258 1+46	
passen-	600 nm	Fuel Used	1,033	837	
s)		Specific Range/Altitude	0.581/FL 280	0.717/FL 390	
		Runway	2,437	2,318	
	1,000 nm	Flight Time	3+36	3+04	
	_,000	Fuel Used	1,642	1,137	
		Specific Range/Altitude	0.609/FL 280	0.880/FL 410	
marks		Certification Basis	FAR 23, 2016 Some data preliminary.	FAR 23, 2006/15 1,000-nm mission flown with 3 passengers. *V50 used in lieu of V2. 2016 data.	

Manufacturer Model BCA Equipped			0				Nextant Aerospace
BCA Fauinned			Cessna Citation Mustang	Phenom 100 EV	Cessna Citation M2	HondaJet	Nextant 400 XTi
	I Price		<b>CE-510</b> \$3,350,000	<b>EMB-500</b> \$4,495,000	<b>CE-525</b> \$4,500,000	<b>HA-420</b> \$4,850,000	<b>BE 400A</b> \$5,304,500
Character-	111100	Seating	1+5/5/—	1+5/7/7	1+7/7/—	1+5/6/6	2+7/9/—
istics		ng Loading/Power Loading	41.2/2.96	53.1/3.09	44.6/2.72	60.0/2.60	67.6/2.67
External	Noise (EPNdB):	: Lateral/Flyover/Approach Length	73.9/85.0/86.0 40.6	70.4/81.4/86.1 42.1	85.9/73.2/88.5 42.6	85.4/72.9/87.5 42.6	76.9/91.5/88.8 48.4
Dimensions		Height	13.4	14.3	13.9	14.9	13.9
ft.)		Span	43.2	40.4	47.3	39.8	43.5
nternal	Length	: Main Seating/Net/Gross	6.7/9.8/9.8	9.0/11.0/11.0	8.8/11.0/11.0	12.1/12.1/NA	15.5/15.5/—
Dimensions	He	eight/Dropped Aisle Depth	4.5/0.3	4.9/0.3	4.8/0.4	4.8/NA	4.8/flat floor
ft.)		Width: Max/Floor	4.6/3.1	5.1/3.6	4.8/3.1	5.0/NA	4.9/4.0
Baggage		Internal: Cu. ft./lb.	6/98	10/99	—/— 40/70F	NA/NA	27/410
		External: Cu. ft./lb.	57/620 2 P&WC	60/418 2 P&WC	46/725 2 Wms Intl	66/500 2 GE Honda	26/450 2 Wms Intl
Power		Engines	PW615F	PW 617F-E	FJ44-1AP-21	HF-120-H1A	FJ44-3AP
- OWEI		utput (lb. each)/Flat Rating	1,460/ISA+10C	1,730/ISA+8C	1,965/ISA+7C	2,037/ISA+10C	3,052/ISA+7C
	inspection interval/i	Manu. Service Plan Interval Max Ramp	3,500t/— 8,730	3,500t/— 10,748	3,500t/5,000 10,800	NA/— 10,680	5,000t/— 16,500
		Max Takeoff	8,645	10,703	10,700	10,600	16,300
		Max Landing	8,000	9,877	9,900	9,860	15,700
		Zero Fuel	6,750c	9,072c	8,400c	8,800c	13,000c
Veights (lb.)		BOW Max Payload	5,600 1,150	7,298 1,774	6,990 1,410	7,279 1,521	10,950 2,050
		Useful Load	3,130	3,450	3,810	3,401	5,550
		Max Fuel	2,580	2,804	3,296	2,845	4,912
		ailable Payload w/Max Fuel	550	646	514	556	638
	Ava	ailable Fuel w/Max Payload Mmo	1,980 0.630	1,676 0.700	2,400 0,710	1,880 0.720	3,500 0.780
imits		Trans. Alt. FL/VMO	FL 271/250	280/275	FL 305/263	FL 302/270	FL 290/320
		PSI/Sea-Level Cabin	8.3/21,280	8.3/21,280	8.5/22,027	8.8/23,060	9.1/24,000
		TOFL (SL elev./ISA temp.)	3,110	3,199	3,210	3,934	3,821
irport		TOFL (5,000-ft. elev.@25C) Mission Weight	6,600 8,645	5,663 10,703	5,580 10,700	6,108 10,600	5,088 14,500p
erfor-		NBAA IFR Range	984	1,092	1,204	1,223	1,197
nance		V2	97	99	111	120	116
		VREF	88	95	101	105	105
		Landing Distance Time to Climb/Altitude	2,137 20/FL 370	2,473 19/FL 370	2,340 18/FL 370	2,795 15/FL 370	2,960 16/FL 370
limb	FAR	25 Engine-Out Rate (fpm)	432	597	618	933	305
	FAR 25 Engine-Out Gradient (ft./nm)		267	316	334	400	158
		Certificated	41,000	41,000	41,000	43,000	45,000
eilings (ft.)		All-Engine Service Engine-Out Service	41,000 26,900	41,000 24,045	41,000 26,800	43,000 27,000	45,000 27,500
		TAS/Fuel Flow (lb./hr.)	319/498	340/543	323/516	360/558	406/740
Cruise	Long Range	Altitude/Specific Range	FL 390/0.641	FL 410/0.626	FL 410/0.626	FL 430/0.645	FL 450/0.549
Ciuise	High Speed	TAS/Fuel Flow (lb./hr.)	339/609	406/955	401/920	420/972	447/968
	<u> </u>	Altitude/Specific Range Nautical Miles	FL 350/0.557 716	FL 330/0.425 466	FL 350/0.436 812	FL 330/0.432 600	FL 430/0.462 1,024
	Max Payload	Average Speed	294	325	361	347	367
	(with available fuel)	Trip Fuel	1,300	1,036	1,706	1,230	2,411
IBAA IFR		Specific Range/Altitude	0.551/FL 410	0.450/FL 410	0.476/FL 410	0.488/FL 430	0.425/FL 450
langes	Max Fuel	Nautical Miles	1,141	1,194	1,357	1,282	1,895
AR Part 23.	(with available	Average Speed Trip Fuel	304 1,947	333 2,196	372 2,675	361 2,273	384 3,953
00-nm	payload)	Specific Range/Altitude	0.586/FL 410	0.544/FL 410	0.507/FL 410	0.564/FL 430	0.479/FL 450
Iternate;		Nautical Miles	963	1,092	1,183	1,065	1,801
AR Part 25,	Four Passengers	Average Speed	301	333	370	361	383
00-nm	(with available fuel)	Trip Fuel	1,664 0.579/FL/10	2,038 0.536/FL 410	2,352 0.503/FL/410	1,976	3,706 0.486/FL 450
lternate)		Specific Range/Altitude Nautical Miles	0.579/FL 410 1,204	1,254	0.503/FL 410 1,400	0.539/FL 430 1,358	0.486/FL 450 1,981
	F	Average Speed	315	329	378	358	381
	Ferry	Trip Fuel	1,965	2,220	2,705	2,290	3,986
		Specific Range/Altitude	0.613/FL 410	0.565/FL 410	0.518/FL 410	0.593/FL 430	0.497/FL 450
		Runway Flight Time	2,498 1+00	2,909 0+53	2,625 0+52	3,564 0+53	3,015 0+48
	300 nm	Fuel Used	670	753	804	676	786
		Specific Range/Altitude	0.448/FL 370	0.398/FL 390	0.373/FL 370	0.444/FL 430	0.382/FL 390
Missions		Runway	2,700	3,121	2,692	3,732	3,044
4 passen-	600 nm	Flight Time Fuel Used	1+56 1,135	1+45 1,236	1+38 1,362	1+38 1,179	1+30 1,323
ers)		Specific Range/Altitude	0.529/FL 390	0.485/FL 390	0.441/FL 390	0.509/FL 430	0.454/FL 430
		Runway	3,110	3,179	3,009	3,909	3,101
	1,000 nm	Flight Time	3+19	2+54	2+42	2+40	2+28
	,	Fuel Used Specific Range/Altitude	1,754 0.570/FL 410	1,919 0.521/FL 410	2,018 0.496/FL 410	1,863 0.537/FL 430	2,145 0.466/FL 450
emarks		Certification Basis	FAR 23, 2006	FAR 23, 2008	FAR 23, 2013	FAR 23, 2015	FAR 25, 1981/85 STC 02371LA; STC 10959SC; STC 03960AT

Manufacturer		Textron Aviation Syberjet  Cessna Citation CJ3+ SJ30i		Pilatus Aircraft	Embraer	Textron Aviation		
Model	lodel		Cessna Citation CJ3+ SJ30i CE-525B SJ30-2		SVJ PC-24	Phenom 300 EMB-505	Cessna Citation CJ4 CE-525C	
BCA Equipped	d Price		\$7,995,000	\$8,306,452	\$8,900,000	\$8,995,000	\$8,995,000	
Character-		Seating	1+8/9/—	1+5/6/—	1+8/11/NA	1+7/10/10	2+8/9/—	
stics		ng Loading/Power Loading	47.2/2.46	73.2/3.03	53.1/2.60	60.0/2.74	51.8/2.36	
External	Noise (EPNab):	Lateral/Flyover/Approach Length	88.7/74.0/88.6 51.2	78.5/86.2/91.8 46.8	NA/NA/NA 55.2	69.9/88.8/88.5 51.2	92.8/75.6/89.5 53.3	
Dimensions		Height	15.2	14.2	17.3	16.7	15.3	
ft.)		Span	53.3	42.3	55.8	52.2	50.8	
nternal	l ength	: Main Seating/Net/Gross	12.3/15.7/—	12.5/12.5/—	NA/NA/23.0	14.8/17.2/17.2	12.9/17.3/17.3	
Dimensions		eight/Dropped Aisle Depth	4.8/0.4	4.4/NA	5.1/flat floor	4.9/0.3	4.8/0.4	
ft.)		Width: Max/Floor	4.8/3.1	4.8/2.8	5.5/3.8	5.1/3.6	4.8/3.3	
		Internal: Cu. ft./lb.	-/-	6/100	90/NA	10/77	7/40	
Baggage		External: Cu. ft./lb.	65/1,000	53/500	NA/NA	74/573	71/1,000	
		Engines	2 Wms Intl	2 Wms Intl	2 Wms Intl	2 P&WC	2 Wms Intl	
ower		-	FJ44-3A	FJ44-2A	FJ44-4A	PW 535E	FJ44-4A	
		utput (lb. each)/Flat Rating Manu. Service Plan Interval	2,820/ISA+11C 4,000t/5,000	2,300/ISA+8C 3,500t/—	3,400/NA 5,000t/NA	3,360/ISA+15C 5,000t/—	3,621/ISA+11C 5,000t/5,000	
	inspection interval/	Max Ramp	14,070	14,050	17,750	18,497	17,230	
		Max Takeoff	13,870	13,950	17,650	18,387	17,110	
		Max Landing	12,750	12,725	16,250	17,042	15,660	
		Zero Fuel	10,510c	10,500c	NA	14,220c	12,500c	
eights (lb.)		BOW	8,540	8,917	NA	11,583	10,280	
		Max Payload	1,970	1,583	2,500	2,637	2,220	
		Useful Load Max Fuel	5,530 4,710	5,133 4,850	NA 5,965	6,914 5,353	6,950 5,828	
	Ava	illable Payload w/Max Fuel	820	283	915	1,561	1,122	
		ilable Fuel w/ Max Payload	3,560	3,550	NA NA	4,277	4,730	
		Ммо	0.737	0.830	NA	0.780	0.770	
imits		Trans. Alt. FL/VMo	FL 293/278	FL 295/320	NA/NA	FL 263/320	FL 279/305	
		PSI/Sea-Level Cabin	8.9/23,586	12.0/41,000	NA/23,500	9.4/25,560	9.0/24,005	
	,	TOFL (SL elev./ISA temp.) TOFL (5,000-ft. elev.@25C)	3,180 4,750	3,939 8,784	2,690 4,430	2,354 5.400	3,140 5,180	
irport	·	Mission Weight	13,870	13,125	17,750	18,387	16,788	
erfor-		NBAA IFR Range	1,827	1,915	NA NA	2,019	1,948	
nance		V2	114	112	NA	113	117	
		VREF	99	104	NA	104	99	
		Landing Distance	2,422	2,657	NA	2,220	2,281	
limb	EAD.	Time to Climb/Altitude 25 Engine-Out Rate (fpm)	15/FL 370 808	16/FL 370 312	NA/FL 370 NA	15/FL 370 872	14/FL 370 839	
limb		gine-Out Gradient (ft./nm)	425	167	NA NA	437	430	
	1711(20 E11	Certificated	45,000	49,000	45,000	45,000	45,000	
eilings (ft.)		All-Engine Service	45,000	44,000	45,000	45,000	45,000	
		Engine-Out Service	26,250	25,800	26,000	30,137	28,200	
	Long Range	TAS/Fuel Flow (lb./hr.)	352/624	436/684	NA/NA	383/757	377/812	
Cruise		Altitude/Specific Range	FL 450/0.564	FL 450/0.637	NA/NA	FL 450/0.506	FL 450/0.464	
	High Speed	TAS/Fuel Flow (lb./hr.) Altitude/Specific Range	415/1,197 FL 350/0.347	475/1,188 FL 360/0.400	NA/NA FL 300/NA	444/1,312 FL 350/0.338	442/1,470 FL 370/0.301	
		Nautical Miles	1,172	1,635	NA	1,351	1,425	
	Max Payload	Average Speed	368	402	NA NA	397	407	
	(with available fuel)	Trip Fuel	2,552	2,908	NA NA	3,362	3,753	
DAA 150	<u> </u>	Specific Range/Altitude	0.459/FL 450	0.562/FL 470	NA/NA	0.402/FL 450	0.380/FL 450	
IBAA IFR	Max Fuel	Nautical Miles	1,814	2,598	NA	1,883	1,913	
anges	(with available	Average Speed	377	410	NA	406	413	
FAR Part 23,	payload)	Trip Fuel	3,846	4,241	NA	4,469	4,904	
00-nm		Specific Range/Altitude	0.472/FL 450	0.613/FL 490	NA/NA	0.421/FL 450	0.390/FL 450	
Iternate;	Four Passengers	Nautical Miles Average Speed	1,825 276	2,205 408	NA NA	1,936 411	1,927 416	
AR Part 25,	(with available fuel)	Trip Fuel	3,767	3,713	NA NA	4,510	4,920	
00-nm Iternate)	(	Specific Range/Altitude	0.484/FL 450	0.594/FL 490	NA/NA	0.429/FL 450	0.392/FL 450	
Iternate)		Nautical Miles	1,900	2,667	ŃΑ	1,985	1,955	
	Ferry	Average Speed	383	411	NA	417	420	
	Tolly	Trip Fuel	3,872	4,246	NA	4,473	4,955	
		Specific Range/Altitude	0.491/FL 450	0.628/FL 490	NA/NA	0.444/FL 450	0.395/FL 450	
		Runway Flight Time	2,608 0+49	2,822 0+45	NA NA	2,613 0+47	2,429 0+46	
	300 nm	Flight Time Fuel Used	969	846	NA NA	1,058	1,087	
		Specific Range/Altitude	0.310/FL 370	0.355/FL 410	NA/NA	0.284/FL 390	0.276/FL 390	
Missions		Runway	2,609	3,025	NA	2,747	2,444	
4 passen-	600 nm	Flight Time	1+35	1+26	NA	1+29	1+27	
ers)	000 11111	Fuel Used	1,571	1,313	NA	1,735	1,865	
5.5)		Specific Range/Altitude	0.382/FL 410	0.457/FL 450	NA/NA	0.346/FL 410	0.322/FL 410	
		Runway Flight Time	2,720 2+36	3,336 2+21	NA NA	2,808 2+26	2,490 2+23	
	1,000 nm	Fuel Used	2,315	1,980	NA NA	2,471	2,823	
		Specific Range/Altitude	0.432/FL 430	0.505/FL 450	NA/NA	0.405/FL 450	0.354/FL 430	
Remarks		Certification Basis	FAR 23, 2004/14 Commuter category Garmin G3000.	FAR 23 Commuter category	EASA CS 23, FAR 23 Commuter category pending Pricing in 2017 dollars; FJ44-4 with quiet power mode APU function.	FAR 23, 2009 Commuter category Performance-based upon optional increased weights.	FAR 23, 2010 Commuter category	

wodei		Textron Aviation Cessna Citation X Elite	Bombardier	Textron Aviation Cessna Citation XLS+	Bombardier	Textron Aviation Cessna Citation Latitude CE-680A	
		CE-750	Learjet 70 Model 45	CE-560XL	Learjet 75 Model 45		
<b>CA</b> Equippe	d Price		\$6,500,000	\$11,300,000	\$12,750,000	\$13,800,000	\$16,350,000
haracter-		Seating	2+8/11/—	2+6/7/7	2+9/12/—	2+8/9/9	2+9/9/10
tics		ng Loading/Power Loading Lateral/Flyover/Approach	68.5/2.67 83.8/71.2/90.3	69.6/2.79 87.4/74.3/93.4	54.6/2.45 86.8/72.2/92.8	69.6/2.79 87.4/74.3/93.4	56.8/2.61 87.7/73.5/87.7
xternal	Noise (EFNub).	Length	72.3	56.0	52.5	58.0	62.3
imensions		Height	19.3	14.0	17.2	14.0	20.9
t.)		Span	63.9	50.9	56.3	50.9	72.3
ternal	Length	: Main Seating/Net/Gross	17.0/23.9/23.9	10.6/17.7/17.7	14.3/18.5/18.5	13.4/19.8/19.8	15.9/21.8/21.8
imensions		eight/Dropped Aisle Depth	5.7/0.7	4.9/flat floor	5.7/0.7	4.9/flat floor	6.0/flat floor
t.)		Width: Max/Floor	5.5/3.9	5.1/3.2	5.5/3.9	5.1/3.2	6.4/4.1
		Internal: Cu. ft./lb.	variable/variable	15/150	10/100	15/150	26/NA
aggage		External: Cu. ft./lb.	82/775	50/500	80/700	50/500	100/1,000
		Engines	2 RR	2 Hon	2 P&WC	2 Hon	2 P&WC
ower	0.	utput (lb. each)/Flat Rating	AE3007C1 6,764/ISA+15C	TFE731-40BR 3,850/ISA+23C	PW545C 4,119/ISA+10C	TFE731-40BR 3,850/ISA+23C	PW306D 5,907/ISA+16C
		Manu. Service Plan Interval	4,500t*/—	6,000t/—	5,000t/—	6,000t/—	6,000t/—
	mopodadii intoi vaij ii	Max Ramp	36,400	21,750	20,400	21,750	31,050
		Max Takeoff	36,100	21,500	20,200	21,500	30,800
		Max Landing	31,800	19,200	18,700	19,200	27,575
		Zero Fuel	24,400c	16,000c	15,100c	16,000c	21,200c
eights (lb.)		BOW	22,100	13,900	12,860	14,050	18,656
		Max Payload Useful Load	2,300 14,300	2,100 7,850	2,240 7,540	1,950 7,700	2,544 12,394
		Useful Load Max Fuel	14,300	6,062	7,540 6,740	6,062	12,394
	Ava	ilable Payload w/Max Fuel	1,369	1,788	800	1,638	1,000
		ilable Fuel w/Max Payload	12,000	5,750	5,300	5,750	9,850
		Ммо	0.920	0.810	0.750	0.810	0.800
imits		Trans. Alt. FL/VMO	FL 307/350	FL 270/330	FL 265/305	FL 270/330	FL 298/305
		PSI/Sea-Level Cabin	9.3/25,230	9.4/25,700 4,440	9.3/25,230 3,560	9.4/25,700 4,440	9.7/25,400 3,580
	١,	TOFL (SL elev./ISA temp.) OFL (5,000-ft. elev.@25C)	5,140 7,350	5,191	3,560 5,430	5,272	5,070
irport	· ·	Mission Weight	34,980p	20,632	20,200	20,782	30,675
erfor-		NBAA IFR Range	2,980	2,045	1,740	2,026	2,700
ance		V2	137	125	118	125	115
		VREF	112	112	106	113	95
		Landing Distance	2,730	2,326	2,740	2,338	2,085
limb	FAD	Time to Climb/Altitude 25 Engine-Out Rate (fpm)	18/FL 370 486	15/FL 370 430	15/FL 370 765	15/FL 370 430	15/FL 370 652
ZIIIIID		gine-Out Gradient (ft./nm)	213	207	389	207	340
		Certificated	51,000	51,000	45,000	51,000	45,000
eilings (ft.)		All-Engine Service	43,000	45,200	45,000	44,700	43,000
		Engine-Out Service	26,000	28,400	28,600	27,900	26,260
	Long Range	TAS/Fuel Flow (lb./hr.)	470/1,529	437/970	353/865	437/977	368/1,114
Cruise		Altitude/Specific Range TAS/Fuel Flow (lb./hr.)	FL 470/0.307 513/2,229	FL 470/0.451 452/1,080	FL 450/0.408 431/1,238	FL 470/0.447 451/1,079	FL 430/0.330 432/1,765
	High Speed	Altitude/Specific Range	FL 410/0.230	FL 470/0.419	FL 410/0.348	470/0.418	FL 390/0.245
		Nautical Miles	2,703	1,728	1,150	1,728	2,135
	Max Payload	Average Speed	462	425	385	425	394
	(with available fuel)	Trip Fuel	9,973	4,575	3,663	4,575	7,901
DAAJED		Specific Range/Altitude	0.271/FL 470	0.378/FL 470	0.314/FL 450	0.378/FL 470	0.270/FL 450
IBAA IFR	Max Fuel	Nautical Miles	3,070	1,881	1,719	1,881	2,645
langes	(with available	Average Speed	462	426	395	426	401
FAR Part 23, 00-nm	payload)	Trip Fuel	11,055	4,901	5,233	4,901	9,586
		Specific Range/Altitude Nautical Miles	0.278/FL 490 3,125	0.384/FL 470 2,045	0.328/FL 450 1,719	0.384/FL 470 2,026	0.276/FL 450 2,678
ternate; NP Part 25	Four Passengers	Nautical Miles Average Speed	3,125 463	426	1,719 395	427	2,678
AR Part 25,	(with available fuel)	Trip Fuel	11,078	5,064	5,168	5,058	9,594
00-nm ternate)		Specific Range/Altitude	0.282/FL 490	0.404/FL 470	0.333/FL 450	0.401/FL 470	0.279/FL 450
terriate)		Nautical Miles	3,221	2,150	1,785	2,129	2,731
	Ferry	Average Speed	463	427	403	427	405
	,	Trip Fuel	11,118	5,099	5,268	5,093	9,628
		Specific Range/Altitude	0.290/FL 490	0.422/FL 490	0.339/FL 450	0.418/FL 490	0.284/FL 450
		Runway Flight Time	3,536 0+41	3,588 0+45	2,734 0+46	3,598 0+45	2,760 0+46
	300 nm	Fuel Used	1,837	1,072	1,246	1,075	1,610
		Specific Range/Altitude	0.163/FL 370	0.280/FL 470	0.241/FL 390	0.279/FL 470	0.186/FL 390
lissions		Runway	3,580	3,632	2,758	3,642	2,845
passen-	600 nm	Flight Time	1+16	1+24	1+29	1+23	1+29
ers)		Fuel Used	2,855	1,805	2,094	1,810	2,573
		Specific Range/Altitude	0.210/FL 430	0.332/FL 470 3 691	0.287/FL 410 3.028	0.331/FL 470 3.701	0.233/FL 430 2.951
		Runway Flight Time	3,672 2+03	3,691 2+18	3,028 2+26	3,701 2+18	2,951 2+25
	1,000 nm	Fuel Used	4,469	2,787	3,211	2,792	3,989
		Specific Range/Altitude	0.224/FL 430	0.359/FL 470	0.311/FL 430	0.358/FL 470	0.251/FL 430
emarks		Certification Basis	FAR 25, 1996/2002; JAR 25 1999/2002 *Engine flight hour inspection interval.	FAR/EASA CS 25	FAR 25, 2008	FAR/EASA CS 25	FAR 25, 2015 Garmin G5000.

Manufacture Model	<u> </u>		Embraer Legacy 450 EMB-545	Textron Aviation Cessna Citation Sovereign+ CE-680	Embraer Legacy 500 EMB-550	Textron Aviation Cessna Citation X+ CE-750	Textron Aviation Cessna Citation Longitud CE-700
BCA Equipped	d Price		\$16,570,000	\$17,895,000	\$19,995,000	\$23,365,000	\$23,995,000
Character-		Seating	2+7/9/9	2+9/12/12	2+8/12/12	2+9/12/—	2+8/12/12
stics		ng Loading/Power Loading	74.0/2.73	56.7/2.60	79.4/2.73	69.4/2.60	NA/NA
External	NOISE (EPNOB):	Lateral/Flyover/Approach Length	84.2/72.8/89.9 64.6	87.8/71.9/87.9 63.5	85.5/73.1/89.9 68.1	87.7/72.4/89.3 73.6	NA/NA/NA 73.2
Dimensions		Height	21.1	20.3	21.2	19.2	19.4
ft.)		Span	66.4	72.3	66.4	69.2	68.9
nternal		: Main Seating/Net/Gross	17.4/20.6/24.0	17.4/25.3/25.3	21.3/24.1/27.5	18.3/25.2/25.2	16.5/25.2/28.1
Dimensions	Не	eight/Dropped Aisle Depth	6.0/flat floor	5.7/0.7	6.0/flat floor	5.7/0.7	6.0/flat floor
ft.)		Width: Max/Floor	6.8/4.7	5.5/3.9	6.8/4.7	5.5/3.9	6.4/4.1
Baggage		Internal: Cu. ft./lb. External: Cu. ft./lb.	40/330 110/882	35/415 100/1,000	45/330 110/882	22/NA 82/775	112/1,115 NA/NA
			2 Hon	2 P&WC	2 Hon	2 RR	2 Hon
Power		Engines	HTF7500E	PW306D	HTF7500E	AE3007C2	HTF7700L
OWCI		tput (lb. each)/Flat Rating	6,540/ISA+18C	5,907/ISA+16C	7,036/ISA+18C	7,034/ISA+15C	7,600/ISA+19C
	inspection interval/ i	Manu. Service Plan Interval Max Ramp	0C/— 35,891	6,000t/— 31,025	0C/— 38,537	4,500t*/— 36,900	OC/— NA
		Max Takeoff	35,759	30,775	38,360	36,600	NA NA
		Max Landing	32,518	27,575	34,524	32,000	NA
		Zero Fuel	25,904c	21,000c	26,499	24,978c	NA NA
Veights (lb.)		BOW May Daylood	22,983 2,921	18,235 2,765	23,699 2,800	22,114 2,864	NA 2,725
		Max Payload Useful Load	12,908	12,790	14,838	14,786	2,725 NA
		Max Fuel	12,108	11,390	13,058	12,931	NA NA
		ilable Payload w/Max Fuel	800	1,400	1,780	1,855	1,600
	Ava	ilable Fuel w/Max Payload	9,987	10,025	12,038	11,922	NA 0.040
imite		Ммо Trans. Alt. FL/Vмо	0.830 FL 395/320	0.800 FL 208/305	0.830 FL 295/320	0.935 FL 307/350	0.840 NA/NA
imits		PSI/Sea-Level Cabin	9.7/26,520	FL 298/305 9.3/25,230	FL 295/320 9.7/26,520	FL 307/350 9.3/25,230	9.7/25,400
		TOFL (SL elev./ISA temp.)	3,907	3,530	4,084	5,250	4,900
	T	OFL (5,000-ft. elev.@25C)	5,189	4,760	5,523	7,317	NA
Airport		Mission Weight	35,759	30,250	38,360	35,645	NA 2.500
erfor-		NBAA IFR Range V2	2,919 117	3,093 117	3,131 120	3,396 139	3,520 NA
nance		VREF	101	96	102	116	NA NA
		Landing Distance	2,090	2,144	2,114	2,727	NA
		Time to Climb/Altitude	14/FL 370	13/FL 370	14/FL 370	13/FL 370	13/FL 370
Climb		25 Engine-Out Rate (fpm)	634	735	856	614	NA NA
	FAR 25 EN	gine-Out Gradient (ft./nm) Certificated	324 45,000	377 47,000	387 45,000	267 51,000	NA 45,000
Ceilings (ft.)	All-Engine Service		44,000	45,000	44,000	47,000	45,000
		Engine-Out Service	24,476	29,740	28,189	25,900	27,500
	Long Range	TAS/Fuel Flow (lb./hr.)	438/1,404	368/1,059	440/1,441	470/1,470	457/1,591
Cruise	20118 11411180	Altitude/Specific Range	FL 450/0.312	FL 450/0.347	FL 450/0.305	FL 470/0.320	FL 450/0.287
	High Speed	TAS/Fuel Flow (lb./hr.) Altitude/Specific Range	462/1,621 FL 430/0.285	448/1,756 FL 390/0.255	467/1,741 FL 430/0.268	520/2,453 FL 410/0.212	476/1,933 FL 430/0.246
		Nautical Miles	2,170	2,484	2.603	2,838	3,074
	Max Payload	Average Speed	428	396	438	463	452
	(with available fuel)	Trip Fuel	8,084	8,170	9,908	9,952	11,600
NBAA IFR		Specific Range/Altitude	0.268/FL 450	0.304/FL 470	0.263/450	0.285/FL 490	0.265/FL 450
Ranges	Max Fuel	Nautical Miles	2,904 431	2,996 400	2,998 440	3,241 464	3,422 453
FAR Part 23,	(with available	Average Speed   Trip Fuel	10,285	9,658	11,151	11.108	12,763
100-nm	payload)	Specific Range/Altitude	0.282/FL 450	0.310/FL 470	0.269/FL 450	0.292/FL 490	0.268/FL 450
alternate;		Nautical Miles	2,904	3,069	3,125	3,372	3,500
AR Part 25,	Four Passengers	Average Speed	431	402	433	465	454
200-nm	(with available fuel)	Trip Fuel   Specific Range/Altitude	10,285 0.282/FL 450	9,679 0.317/FL 470	11,222 0.278/FL 450	11,157 0.302/FL 490	12,787 0.274/FL 450
lternate)		Nautical Miles	2,973	3,138	3,153	3,463	3,568
	F	Average Speed	430	405	440	465	454
	Ferry	Trip Fuel	10,313	9,708	11,250	11,195	12,810
		Specific Range/Altitude	0.288/FL 450	0.323/FL 470	0.280/FL 450	0.309/FL 490	0.279/FL 450
		Runway Flight Time	3,674 0+45	2,591 0+45	2,822 0+45	3,725 0+41	2,744 0+44
	300 nm	Flight Time   Fuel Used	1,543	1,506	1,545	1,827	1,516
		Specific Range/Altitude	0.194/FL 450	0.199/FL 390	0.194/FL 450	0.164/FL 370	0.198/FL 450
Missions		Runway	2,696	2,600	2,817	3,775	2,880
4 passen-	600 nm	Flight Time	1+26	1+26	1+26	1+16	1+23
(ers)		Fuel Used   Specific Range/Altitude	2,478 0.242/FL 450	2,404 0.250/FL 430	2,478 0.242/FL 450	2,937 0.204/FL 430	2,457 0.244/FL 450
		Specific Kange/Altitude Runway	2,873	2,650	2,963	3,849	3,025
	1,000 nm	Flight Time	2+21	2+21	2+21	2+02	2+16
	1,000 11111	Fuel Used	3,710	3,750	3,750	4,680	3,746
Remarks		Specific Range/Altitude  Certification Basis	0.270/FL 450  RBAC/FAR/EASA CS 25, 2015	0.267/FL 430 FAR 25, 2013 Garmin G5000.	0.267/FL 450  RBAC/FAR/EASA CS 25, 2014	0.214/FL 430  FAR 25, 2014 Garmin G5000. *Engine flight hour inspection interval.	0.267/FL 450  FAR 25 pending Garmin G5000.

Manufacturer Model		Gulfstream Aerospace Gulfstream 280 G280	Embraer Legacy 650E EMB-135BJ*	Bombardier Challenger 350 BD-100-1A10	Dassault Falcon 2000S Falcon 2000EX	Bombardier Challenger 650 CL-600-2B16	
CA Equipped	Price		\$24,500,000	\$25,900,000	\$26,673,000	\$29,550,000	\$32,350,000
haracter-		Seating	2+9/10/19	2+13/14/19	2+10/11/19	2+10/10/19	2+12/13/19
stics		ng Loading/Power Loading	80.0/2.60	97.2/2.97	77.6/2.77	77.7/2.93	98.6/2.61
	Noise (EPNdB):	: Lateral/Flyover/Approach	75.2/89.5/90.5	86.9/78.0/91.7	87.6/75.3/89.6	75.1/91.8/90.5	86.2/81.2/90.3
xternal		Length	66.8	86.4	68.7	66.3	68.4
imensions		Height	21.3	21.8	20.0	23.2	20.7
t.)		Span	63.0	69.5	69.0	70.2	64.3
ternal	Length	: Main Seating/Net/Gross	17.7/25.8/32.3	30.3/42.4/49.1	16.6/25.2/28.6	17.1/26.2/31.0	15.4/25.6/28.3
		eight/Dropped Aisle Depth	6.1/4.5	6.0/2.5	6.0/flat floor	6.2/flat floor	6.0/flat floor
imensions			· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·	·	·
t.)		Width: Max/Floor	6.9/5.4	6.9/5.2	7.2/5.1	7.7/6.3	7.9/6.9
aggage		Internal: Cu. ft./lb.	154/1,980	286/1,441	106/750	131/1,600	112/900
aggage		External: Cu. ft./lb.	—/—	—/—	—/—	8/92	—/—
		Engines	2 Hon	2 RR	2 Hon	2 P&WC	2 GE
ower		Liigiiics	HTF7250G	AE 3007A2	HTF 7350	PW308C	CF34-3B
JWEI	0ι	utput (lb. each)/Flat Rating	7,624/ISA+17C	9,020/ISA+15C	7,323/ISA+15C	7,000/ISA+15C	9,220*/ISA+15C
	Inspection Interval/N	Manu. Service Plan Interval	OC/—	OC/—	OC/—	7,000c/—	OC/—
		Max Ramp	39,750	53,727	40,750	41,200	48,300
		Max Takeoff	39,600	53,572	40,600	41,000	48,200
		Max Landing	32,700	44,092	34,150	39,300	38,000
		Zero Fuel	28,200c	36,156c	28,200c	29,700c	32,000c
		BOW	24,200	31,217	24,800	24,750	27,250
eights (lb.)		Max Payload	4,000	4,939	3,400	4,950	4,750
						16,450	21,050
		Useful Load	15,550	22,510	15,950		
		Max Fuel	14,600	20,600	14,045	14,600	19,852
		nilable Payload w/Max Fuel	950	1,910	1,905	1,850	1,198
	Ava	ailable Fuel w/Max Payload	11,550	17,571	12,550	11,500	16,300
		Ммо	0.850	0.800	0.830	0.862	0.850
nits		Trans. Alt. FL/VMo	FL 280/340	FL 276/320	FL 290/320	FL 250/370	FL 222/348
		PSI/Sea-Level Cabin	9.2/25,000	8.4/21,650	8.8/23,338	9.3/25,300	8.8/23,000
		TOFL (SL elev./ISA temp.)	4,750	5,741	4,829	4,325	5,640
	1	TOFL (5,000-ft. elev.@25C)	7,320	7,979	6,451	6,055	9,233
rport		Mission Weight	39,600	53,572	39.495	39,950	47.802
erfor-		NBAA IFR Range	3,600	3,953	3,250	3,600	4,011
		V2	137	144	133	123	147
ance		VREF	115	115	111	106	117
			2,373	2,346	2,302	2,295	2,365
	Landing Distance						
	540	Time to Climb/Altitude	14/FL 370	21/FL 370	14/FL 370	16/FL 370	21/FL 370
imb	FAR 25 Engine-Out Rate (fpm)		845	633	552	528	581
	FAR 25 En	gine-Out Gradient (ft./nm)	371	259	249	257	237
	Certificated		45,000	41,000	45,000	47,000	41,000
eilings (ft.)	All-Engine Service		45,000	41,000	44,000	43,265	38,250
		Engine-Out Service	27,500	23,128	27,800	22,187	20,000
	Long Range	TAS/Fuel Flow (lb./hr.)	459/1,488	425/1,901	459/1,590	437/1,400	424/1,832
	Long Range	Altitude/Specific Range	FL 450/0.308	FL 410/0.224	FL 450/0.289	FL 470/0.312	FL 410/0.231
ruise		TAS/Fuel Flow (lb./hr.)	482/1,925	459/2,570	470/1,832	482/2,075	470/2,448
	High Speed	Altitude/Specific Range	FL 430/0.250	FL 370/0.179	FL 430/0.257	FL 410/0.232	FL 370/0.192
		Nautical Miles	2,577	3,076	2,719	2,450	3,011
	Max Payload	Average Speed	448	417	447	426	417
	(with available fuel)		9,591	15,238	10,689	9,640	14,256
	(with available fuel)	Trip Fuel					
BAA IFR		Specific Range/Altitude	0.269/FL 450	0.202/FL 410	0.254/FL 450	0.254/FL 450	0.211/FL 410
	Max Fuel	Nautical Miles	3,636	3,839	3,235	3,445	3,974
anges	(with available	Average Speed	452	417	449	429	419
AR Part 23,	payload)	Trip Fuel	12,757	18,380	12,206	12,740	17,939
0-nm	-ujiouuj	Specific Range/Altitude	0.285/FL 450	0.209/FL 410	0.265/FL 450	0.270/FL 470	0.222/FL 410
ernate;		Nautical Miles	3,646	3,919	3,250	3,540	4,011
R Part 25,	Four Passengers	Average Speed	451	415	448	430	419
0-nm	(with available fuel)	Trip Fuel	12,761	18,422	12,212	12,740	17,953
	,	Specific Range/Altitude	0.286/FL 450	0.213/FL 410	0.266/FL 450	0.278/FL 470	0.223/FL 410
ernate)		Nautical Miles	3,724	3,980	3,307	3,615	4,085
		Average Speed	452	414	450	430	419
	Ferry	Trip Fuel	12,789	18,450	12,236	12,740	17,982
			0.291/FL 450			0.284/FL 470	0.227/FL 410
		Specific Range/Altitude		0.216/FL 410	0.270/FL 450		
		Runway	2,957	3,346	3,611	2,795	3,389
	300 nm	Flight Time	0+47	0+49	0+47	0+47	0+47
	J00II	Fuel Used	1,505	1,773	1,583	1,525	1,595
		Specific Range/Altitude	0.199/FL 450	0.169/FL 410	0.190/FL 450	0.197/FL 470	0.188/FL 410
issions		Runway	2,997	3,518	3,656	2,855	3,421
	600 mm	Flight Time	1+26	1+34	1+26	1+27	1+27
passen-	600 nm	Fuel Used	2,412	3,146	2,577	2,465	2,835
rs)		Specific Range/Altitude	0.249/FL 450	0.191/FL 410	0.233/FL 450	0.243/FL 470	0.212/FL 410
		Runway	3,136	3,573	3,718	2,920	3,483
		Flight Time	2+18	2+33	2+18	2+20	2+19
	1,000 nm	Fuel Used	3,645	4,815	3,925	3,755	4,532
		Specific Range/Altitude	3,645 0.274/FL 450	0.208/FL 410	3,925 0.255/FL 450	0.266/FL 470	4,532 0.221/FL 410
marks		Certification Basis	FAR 25, 2012; EASA CS 25, 2013	FAR 25, 2011 *Factory modification DCA 145-000- 00020/2008	FAR 25 A 98; JAR 25 Chg 15 Rockwell Collins Pro Line 21 Advanced.	FAR/EASA CS 25, 2013 EASy II flight deck; 2017 delivery price.	FAR 25, 1980/83, 87/95/2006/15 Rockwell Collins Pro Line 21 Advance *9,220-lb. max take 8,729-lb. normal take

Manufacturei Model			Dassault Falcon 2000LXS Falcon 2000EX	Gulfstream Aerospace Gulfstream 450 GIV-X	Dassault Falcon 900LX Falcon 900EX	Gulfstream Aerospace Gulfstream 500 GVII-G500
CA Equipped	l Price		\$34,700,000	\$43,150,000	\$44,300,000	\$44,650,000
haracter-		Seating	2+8/10/19	2+14/16/19	2+12/12/19	2+13/19/19
stics		ng Loading/Power Loading	81.2/3.06	78.4/2.69	92.9/3.27	80.9/2.54
xternal	Noise (EPNdB):	Lateral/Flyover/Approach Length	76.4/91.7/90.5 66.3	76.2/89.5/92.3 89.3	78.2/90.3/92.1 66.3	NA/NA/NA 91.2
imensions		Height	23.2	25.2	24.8	25.5
t.)		Span	70.2	77.8	70.2	86.3
iternal	Length	: Main Seating/Net/Gross	17.1/26.2/31.0	25.8/37.0/45.1	23.5/33.2/39.3	26.3/41.5/47.6
imensions		eight/Dropped Aisle Depth	6.2/flat floor	6.0/flat floor	6.2/flat floor	6.2/flat floor
t.)		Width: Max/Floor	7.7/6.3	7.0/5.4	7.7/6.3	7.6/6.1
		Internal: Cu. ft./lb.	131/1,600	169/2,000	127/2,866	230/2,250
aggage		External: Cu. ft./lb.	8/92	—/—	-/-	-/-
		Enginee	2 P&WC	2 RR	3 Hon	2 P&WC
ower		Engines	PW308C	Tay Mk 611-8C	TFE731-60	PW814GA
J., J.		tput (lb. each)/Flat Rating	7,000/ISA+15C	13,850/ISA+15C	5,000/ISA+17C	15,144/ISA+15C
	inspection interval/it	Manu. Service Plan Interval Max Ramp	7,000c/— 43,000	12,000t or 0C/— 75,000	6,000c/— 49,200	0C/— 77,250
		Max Takeoff	42,800	74,600	49,000	76,850
		Max Landing	39,300	66,000	44,500	64,350
		Zero Fuel	29,700c	49,000c	30,864c	52,100c
laidhta (lh.)		BOW	24,750	43,200	26,750	46,600
eights (lb.)		Max Payload	4,950	5,800	4,114	5,500
		Useful Load	18,250	31,800	22,450	30,650
	_	Max Fuel	16,660	29,281	20,905	28,850
		ilable Payload w/Max Fuel	1,590	2,519	1,545	1,800
	Ava	ilable Fuel w/Max Payload Mmo	13,300 0.862	26,000 0.880	18,336 0.870	25,150 0.925
imits		Trans. Alt. FL/VMo	FL 250/370	FL 280/340	FL 250/370	0.925 NA/NA
		PSI/Sea-Level Cabin	9.3/25,300	9.6/26,700	9.6/25,300	10.7/31,900
		TOFL (SL elev./ISA temp.)	4,675	5,600	5,360	5,200
	Т	OFL (5,000-ft. elev.@25C)	6,840	8,200	7,615	7,930
irport		Mission Weight	42,010	74,600	48,255	76,850
erfor-		NBAA IFR Range	4,100	4,328	4,685	5,000
nance		V2	126	150	134	NA
		VREF	106 2,295	123 2,663	111 2,455	NA NA
		Landing Distance Time to Climb/Altitude	2,295 17/FL 370	2,063 16/FL 370	2,455 19/FL 370	15/FL 370
limb	FAR	25 Engine-Out Rate (fpm)	463	712	723	NA NA
		gine-Out Gradient (ft./nm)	221	285	324	NA NA
		Certificated	47,000	45,000	51,000	51,000
Ceilings (ft.)	All-Engine Service		42,315	42,400	39,630	NA
		Engine-Out Service	21,010	25,000	24,980	NA
	Long Range	TAS/Fuel Flow (lb./hr.)	437/1,485	459/2,585	431/1,665	488/2,440
ruise	. 0 . 0	Altitude/Specific Range	FL 450/0.294	FL 450/0.178	FL 430/0.259	FL 450/0.200
	High Speed	TAS/Fuel Flow (lb./hr.)	483/2,325 FL 390/0.208	476/3,055	474/2,225	516/3,467
		Altitude/Specific Range Nautical Miles	2,915	FL 410/0.156 3,549	FL 390/0.213 3,790	FL 410/0.149 4,129
	Max Pavload	Average Speed	427	452	422	4,123
	(with available fuel)	Trip Fuel	11,438	22,622	16,340	22,365
	( ,	Specific Range/Altitude	0.255/FL 450	0.157/FL 450	0.232/FL 430	0.185/FL 470
BAA IFR		Nautical Miles	3,990	4,216	4,565	5,000
anges	Max Fuel (with available	Average Speed	430	453	421	480
AR Part 23,	payload)	Trip Fuel	14,798	26,023	18,909	26,172
00-nm	pajioaaj	Specific Range/Altitude	0.270/FL 470	0.162/FL 450	0.241/FL 430	0.191/FL 490
ternate;		Nautical Miles	4,065	4,328	4,650	5,075
AR Part 25,	Four Passengers	Average Speed	430	452	420	480
00-nm	(with available fuel)	Trip Fuel Specific Range/Altitude	14,798 0.275/FL 470	26,087 0.166/FL 450	18,909 0.246/FL 430	26,200 0.194/FL 490
ternate)		Nautical Miles	4,155	4,382	4,740	5,137
		Average Speed	431	453	419	480
	Ferry	Trip Fuel	14,798	26,116	18,909	26,222
		Specific Range/Altitude	0.281/FL 470	0.168/FL 450	0.251/FL 430	0.196/FL 490
		Runway	2,795	3,225	2,730	NA
	300 nm	Flight Time	0+47	0+46	0+47	0+45
		Fuel Used	1,525	2,599	1,595	2,274
		Specific Range/Altitude Runway	0.197/FL 470 2,855	0.115/FL 450 3,258	0.188/FL 470 2,865	0.132/FL 490 NA
lissions		Flight Time	2,855 1+27	3,258 1+25	2,805	1+22
passen-	600 nm	Fuel Used	2,465	4,113	2,625	3,561
ers)		Specific Range/Altitude	0.243/FL 470	0.146/FL 450	0.229/FL 470	0.168/FL 490
		Runway	2,920	3,304	2,880	NA
	1,000 nm	Flight Time	2+20	2+18	2+20	2+12
	2,000 1/111	Fuel Used	3,755	6,176	4,070	5,313
emarks		Specific Range/Altitude  Certification Basis	0.266/FL 470  FAR/EASA CS 25, 2013 EASy II flight deck; 2017 delivery price.	0.162/FL 450 FAR/EASA CS 25, 2004	0.246/FL 450  FAR/EASA 25, 1979/2010 EASy II flight deck; 2017 delivery price.	0.188/FL 490  FAR/EASA 25 pending

Manufacture	r		Bombardier	Embraer	Dassault	Airbus
Model			Global 5000 BD-700-1A11	Lineage 1000E ERJ 190-100 ECJ	Falcon 7X Falcon 7X	A320 Prestige A320-214
<b>BCA</b> Equipped	d Price		\$50,441,000	\$53,000,000	\$53,800,000	\$95,000,000
	4 1 1100	Seating	3+13/15/19	3+13/19/19	3+12/14/19	4+18/179/—
Character-	Wi	ng Loading/Power Loading	90.6/3.14	120.7/3.25	92.0/3.64	130.3/3.18
istics	Noise (EPNdB):	: Lateral/Flyover/Approach	88.7/83.5/89.7	92.7/86.4/92.5	82.3/90.1/92.6	85.5/93.4/95.5
External		Length	96.8	118.9	76.7	123.3
Dimensions		Height Span	25.5	34.7	25.7	38.6
(ft.)	Landthe Main Constant (N)		94.0	94.2	86.0	111.8
Internal		: Main Seating/Net/Gross	27.2/40.7/45.7	67.2/76.6/84.3	26.2/39.1/46.5	90.3/90.3/—
Dimensions	H	eight/Dropped Aisle Depth	6.2/flat floor	6.6/flat floor	6.2/flat floor	7.4/flat floor
(ft.)		Width: Max/Floor	7.9/6.5	8.8/8.0	7.7/6.3	12.1/11.7
Baggage		Internal: Cu. ft./lb.	195/1,000	323/2,293	140/2,004	NA/NA
		External: Cu. ft./lb.	_/_	120/705	_/ <u>_</u>	985/NA
		Engines	2 RR BR700-710A2-20	2 GE CF34-10E7-B	3 P&WC PW307A	2 CFMI CFM56-5B4/3*
Power	01	utput (lb. each)/Flat Rating	14,750/ISA+20C	18,500/ISA+15C	6,402/ISA+17C	27,000/ISA+29C
		Manu. Service Plan Interval	OC/—	OC/—	7,200c/—	OC/—
		Max Ramp	92,750	120,593	70,200	172,850
		Max Takeoff	92,500	120,152	70,000	171,950
		Max Landing	78,600	100,972	62,400	145,500
		Zero Fuel	58,000c	80,469c	41,000c	137,800c
Weights (lb.)		BOW	50,861	70,548	36,600	109,000
		Max Payload	7,139 41,889	9,921 50,045	4,400 33,600	28,800 63,850
		Useful Load Max Fuel	41,889 38,959	48,217	33,600	53,450
	Διε	ailable Payload w/Max Fuel	2,930	1,828	1,660	10,400
		ailable Fuel w/Max Payload	34,750	40,124	29,200	35,050
	λ	Ммо	0.890	0.820	0.900	0.820
Limits		Trans. Alt. FL/VM0	FL 303/340	FL 289/320	FL 270/370	FL 250/350
		PSI/Sea-Level Cabin	10.3/30,125	8.8/23,190	10.2/29,200	8.3/NA
		TOFL (SL elev./ISA temp.)	5,540	6,076	5,710	6,920
Airport	1	TOFL (5,000-ft. elev.@25C)	7,223	9,500	8,045	9,355
Airport		Mission Weight	90,370	112,038	69,140	171,950
Perfor-		NBAA IFR Range V2	5,475 133	3,965 140	5,795 133	4,300 NA
mance		V2 Vref	107	110	106	NA NA
		Landing Distance	2,189	2,038	2,120	2,400
		Time to Climb/Altitude	18/FL 370	29/FL 350	19/FL 370	23/FL 360
Climb	FAR	25 Engine-Out Rate (fpm)	704	NA	597	NA
	FAR 25 En	gine-Out Gradient (ft./nm)	318	NA	269	NA
		Certificated	51,000	41,000	51,000	39,000
Ceilings (ft.)		All-Engine Service	44,600	35,000	40,215	NA
		Engine-Out Service	20,600	19,178	25,480	NA 451 (4.720
	Long Range	TAS/Fuel Flow (lb./hr.) Altitude/Specific Range	470/2,856 FL 450/0.165	454/4,184 FL 380/0.109	459/2,260 FL 430/0.203	451/4,730 FL 370/0.095
Cruise		TAS/Fuel Flow (lb./hr.)	499/3,582	471/5,033	497/3,205	473/5,860
	High Speed	Altitude/Specific Range	FL 410/0.139	FL 350/0.094	FL 390/0.155	350/0.081
		Nautical Miles	4,920	3,493	5,000	2,100
	Max Payload	Average Speed	463	442	453	428
	(with available fuel)	Trip Fuel	33,374	35,569	26,820	27,936
NIDAAJED		Specific Range/Altitude	0.147/FL 470	0.098/FL 400	0.186/FL 450	0.075/FL 350
NBAA IFR	Max Fuel	Nautical Miles	5,486	4,532	5,670	3,852
Ranges	(with available	Average Speed	464	446	454	438
(FAR Part 23,	payload)	Trip Fuel	35,723	43,962	29,560	46,930
100-nm		Specific Range/Altitude	0.154/FL 470	0.103/FL 410	0.192/FL 470	0.082/FL 390
alternate;	Four Passengers	Nautical Miles	5,475 463	4,602 446	5,760 454	4,330 438
FAR Part 25,	(with available fuel)	Average Speed Trip Fuel	35,719	44,240	29,560	438
200-nm	(man dvalidatic ruci)	Specific Range/Altitude	0.153/FL 470	0.104/FL 410	0.195/FL 470	0.090/FL 390
alternate)		Nautical Miles	5,526	4,640	5,840	4,380
	Eo	Average Speed	464	446	454	438
	Ferry	Trip Fuel	35,743	44,264	29,560	48,108
		Specific Range/Altitude	0.155/FL 470	0.105/FL 410	0.198/FL 470	0.091/FL 390
		Runway	2,487	3,002	2,500	3,670
	300 nm	Flight Time	0+46	0+48	0+46	0+55
		Fuel Used	2,773 0.108/FL/450	3,426 0.088/FL 390	2,075 0.145/FL 450	4,265 0.070/FL 350
		Specific Range/Altitude Runway	0.108/FL 450 2,575	0.088/FL 390 3,133	2,515	3,700
Missions		Flight Time	1+23	1+26	1+25	1+34
(4 passen-	600 nm	Fuel Used	4,445	5,862	3,285	7,080
gers)		Specific Range/Altitude	0.135/FL 490	0.102/FL 410	0.183/FL 470	0.085/FL 390
		Runway	2,697	3,251	2,640	3,760
	1,000 nm	Flight Time	2+13	2+20	2+17	2+28
	2,2 30	Fuel Used	6,752 0.148/FL 470	9,063 0.110/FL 410	4,945 0.202/FL 470	10,970 0.091/FL 390
Remarks		Specific Range/Altitude  Certification Basis	FAR 25, 1998/2004; EASA 25, 2004 Global Vision flight deck	FAR/EASA 25, 2008	FAR/EASA 25, 2007 EASy II flight deck; DFCS; 2017 delivery price.	FAR 25, 1999 *Also available with 26,500- lbf IAEV2527M-A5 engines; includes 2 additional center tanks and VIP cabin. BCA estimated data.

# **Ultra-Long-Range Jets**

Manufacturer		Gulfstream Aerospace Gulfstream 600	Dassault	Gulfstream Aerospace	Bombardier Global 6000	Gulfstream Aerospac	
Model	odel		Falcon 8X Falcon 7X	Falcon 8X Falcon 7X GUIfstream 550 GV-SP		Gulfstream 650 GVI	
BCA Equipped Price		<b>GVII-600</b> \$56,200,000	\$58,400,000	\$61,500,000	<b>BD-700-1A10</b> \$62,310,000	\$67,400,000	
	Seating	4+16/19/19	3+12/14/19	4+16/18/19	4+13/15/19	4+16/19/19	
Character-	Wing Loading/Power Loading	78.9/2.92	95.9/3.62	80.1/2.96	97.5/3.37	77.6/2.95	
stics Noise	e (EPNdB): Lateral/Flyover/Approach	NA/NA/NA	81.5/88.9/90.6	79.3/90.2/90.8	88.7/83.5/89.7	77.5/89.8/88.3	
xternal	Length	96.1	80.2	96.4	99.4	99.8	
imensions	Height	25.3	26.1	25.8	25.5	25.7	
	Span	94.1	86.3	93.5	94.0	99.6	
t.)	<u> </u>						
iternal	Length: Main Seating/Net/Gross	30.2/45.2/51.3	29.8/42.7/50.1	30.3/42.6/50.1	27.3/43.3/48.3	32.7/46.8/53.6	
imensions	Height/Dropped Aisle Depth	6.2/flat floor	6.2/flat floor	6.0/flat floor	6.2/flat floor	6.3/flat floor	
t.)	Width: Max/Floor	7.6/6.1	7.7/6.3	7.0/5.4	7.9/6.5	8.2/6.7	
loddodo	Internal: Cu. ft./lb.	230/2,250	140/2,004	226/2,500	195/1,000	235/2,500	
aggage	External: Cu. ft./lb.	-/-	—/—	—/—	—/—	—/—	
	Engines	2 P&WC	3 P&WC	2 RR	2 RR	2 RR	
ower	Eligilles	PW815GA	PW307D	BR700-710C4-11	BR700-710A2-20	BR700-725A1-12	
ower	Output (lb. each)/Flat Rating	15,680/ISA+15C	6,722/ISA+17C	15,385/ISA+15C	14,750/ISA+20C	16,900/ISA+15C	
Inspection	n Interval/Manu. Service Plan Interval	OC/—	7,200c/—	8,000t or OC/—	OC/—	10,000t/—	
	Max Ramp	92,000	73,200	91,400	99,750	100,000	
	Max Takeoff	91,600	73,000	91,000	99,500	99,600	
	Max Landing	76,800	62,400	75,300	78,600	83,500	
	Zero Fuel	57,440c	41,000c	54,500c	58,000c	60,500c	
laidhta (lb.)	BOW	51,440	36,800	48,700	52,560	54,500	
eights (lb.)	Max Payload	6,000	4,200	5,800	5,440	6,000	
	Useful Load	40,560	36,400	42,700	47,190	45,500	
	Max Fuel	38,760	35,141	40,994	44,716	44,200	
	Available Payload w/Max Fuel	1,800	1,259	1,706	2,474	1,300	
	Available Fuel w/Max Payload	34,560	32,200	36,900	41,750	39,500	
	Ммо	0.925	0.900	0.885	0.890	0.925	
imits	Trans. Alt. FL/VMo	NA/NA	FL 270/370	FL 270/340	FL 303/340	FL 290/340	
	PSI/Sea-Level Cabin	10.7/31,900	10.4/30,300	10.2/29,200	10.3/30,125	10.7/31,900	
	TOFL (SL elev./ISA temp.)	5,700	5,880	5,910	6,476	5,858	
	TOFL (5,000-ft. elev.@25C)	NA	8,555	9,070	7,880	9,000	
	Mission Weight	91,600	72,591	91,000	94,513p	99,600	
rport	NBAA IFR Range	6,200	6,415	6,738	5,594	6,912	
erformance	V2	NA	138	147	142	146	
	VREF	NA	107	112	110	114	
	Landing Distance	NA	2,245	2,240	2,243	2,680	
	Time to Climb/Altitude	17/FL 370	20/FL 370	18/FL 370	21/FL 370	19/FL 370	
imb	FAR 25 Engine-Out Rate (fpm)	NA NA	774	594	474	NA NA	
IIIII	FAR 25 Engine-Out Gradient (ft./nm)	NA NA	339	242	200	NA	
	Certificated	51,000	51,000	51,000	51,000	51,000	
eiling (ft.)	All-Engine Service	42,700	40,075	42,700	42,400	42,700	
Siii 18 (1 c.)	Engine-Out Service	25,000	26,645	25,820	18,000	25,000	
	TAS	488	459	459	470	488	
	Fuel Flow	2,769	2,254	2,563	3,046	2,825	
Long I	Range Altitude	FL 450	FL 430	FL 450	FL 450	FL 450	
	Specific Range	0.176	0.204	0.179	0.154	0.173	
Cruise	TAS	516	480	488	499	516	
	Fuel Flow	3,891	2,508	3,228	3,796	3,136	
High S	Speed Altitude	FL 410	FL 430	FL 430	FL 410	FL 450	
	Specific Range	0.133	0.191	0.151	0.131	0.165	
	Nautical Miles	5,286	5,555	5,767	5,882	5,934	
Max Pa		481	452	452	464	481	
	lable fuel) Trip Fuel	31,622	29,507	33,993	40.415	36,285	
(mar aran	Specific Range/Altitude	0.167/FL 450	0.188/FL 470	0.170/FL 490	0.146/FL 470	0.164/FL 490	
	Nautical Miles	6,200	6,325	6,698	6,200	6,981	
Max	Fuel						
BAA IFR (with a	vailable Average Speed	481	453	454	464	482	
navl	oad) Irip Fuel	35,918	32,558	38,202	41,472	41,129	
anges	Specific Range/Altitude	0.173/FL 490	0.194/FL 470	0.175/FL 490	0.149/FL 470	0.170/FL 510	
00-nm	Nautical Miles	6,217	6,235	6,708	6,124	6,912	
	ssengers Average Speed	481	453	453	464	481	
(with avail	lable fuel) Trip Fuel	35,924	32,204	38,205	41,437	40,820	
	Specific Range/Altitude	0.173/FL 490	0.194/FL 470	0.176/FL 490	0.148/FL 470	0.169/FL 510	
	Nautical Miles	6,353	6,475	6,853	6,233	7,105	
Fe	rry Average Speed	481	454	454	464	482	
	Inp Fuel	35,966	32,653	38,251	41,487	41,168	
	Specific Range/Altitude	0.177/FL 490	0.198/FL 470	0.179/FL 510	0.150/FL 470	0.173/FL 510	
	Runway	NA NA	2,685	3,436	2,852	3,241	
1,00	0 nm Flight Time	2+12	2+17	2+20	2+13	2+10	
2,00	Fuel Used	5,728	4,994	5,599	6,842	5,942	
	Specific Range/Altitude	0.175/FL 490	0.200/FL 470	0.179/FL 490	0.146/FL 470	0.168/FL 510	
	Runway	NA	3,540	3,599	3,858	3,591	
issions 3.00	0 nm Flight Time	6+19	6+39	6+42	6+20	6+17	
passengers)	Fuel Used	16,060	14,122	15,474	19,538	16,280	
	Specific Range/Altitude	0.187/FL 490	0.212/FL 470	0.194/FL 490	0.154/FL 470	0.184/FL 510	
	Runway	NA	5,645	5,277	6,293	5,241	
	Flight Time	12+29	13+12	13+15	12+39	12+28	
6,00	0 nm Fuel Used	34,432	30,729	33,428	41,053	34,622	
	Specific Range/Altitude	0.174/FL 490	0.195/FL 470	0.179/FL 490	0.146/FL 490	0.173/FL 510	
emarks	Certification Basis	FAR, EASA CS 25	FAR/EASA 25, 2016 EASy III flight deck;	FAR 25, 1997/2003;	FAR 25, 1998/2003; JAR 25 BEVS and new Global	FAR, EASA CS 25, 201	
sine in to	Columbiation Dasis	pending	DFCŠ; 2017 delivery price.	EASA 25 CS, 2004	Vision flight deck standard.	, 2.13.1 00 20, 20.	

# **Ultra-Long-Range Jets**

Manufacturer Model			Gulfstream Aerospace Gulfstream 650ER	Boeing BBJ	Airbus ACJ319	Boeing BBJ MAX8	Boeing BBJ MAX9
	Drico		<b>GVI</b> \$69,400,000	<b>737-700IGW</b> \$79.000.000	<b>A319-133</b>	737-8	737-9
BCA Equipped	Price	Canting			\$87,000,000	\$95,300,000	\$103,300,000
Character-	Wir	Seating Ig Loading/Power Loading	4+16/19/19 80.7/3.07	4+19/55/149 127.5/3.13	4+19/19/156 127.8/3.12	4+19/71/189 135.1/3.24	4+19/75/220 145.2/3.48
stics		Lateral/Flyover/Approach	78.7/89.6/88.3	85.4/94.9/95.8	85.4/94.6/94.2	NA/NA/NA	NA/NA/NA
xternal	Holse (Li Hub).	Length	99.8	110.3	111.0	129.7	138.3
imensions		Height	25.7	41.2	38.6	40.3	40.3
t.)		Span	99.6	117.4	111.8	117.8	117.8
-	Longth	: Main Seating/Net/Gross	32.7/46.8/53.6	72.7/79.2/—	78.0/78.0/—	91.9/98.5/98.5	100.6/107.2/107.2
ternal		ight/Dropped Aisle Depth	6.3/flat floor	79.3/flat floor	7.4/flat floor	7.1/flat floor	7.1/flat floor
imensions	110						
t.)		Width: Max/Floor	8.2/6.7	11.6/10.7	12.2/11.6	11.6/10.7	11.6/10.7
aggage		Internal: Cu. ft./lb.	235/2,500	NA/NA	160/NA	NA/NA	NA/NA
-00-0-		External: Cu. ft./lb.	_/_	159/NA	NA/NA	713/NA	874/NA
		Engines	2 RR BR700-725A1-12	2 CFMI CFM56-7B27E	2 CFMI CFM56-5B7/3*	2 CFMI LEAP-1B	2 CFMI LEAP-1B
ower	0	tput (lb. each)/Flat Rating	16,900/ISA+15C	27,300/ISA+15C	27,000/ISA+29C	28,000/ISA+15C	28,000/ISA+15C
		lanu. Service Plan Interval	10,000t/—	0C/—	0C/—	0C/—	0C/—
	mspection interval/ iv	Max Ramp	104,000	171,500	169,530	181,700	195,200
		Max Takeoff	103,600	171,000	168,650	181,200	194,700
		Max Landing	83,500	134,000	137,790	152,800	163,900
		Zero Fuel	60,500c	126,000c	128,970c	145,400c	156,500c
		BOW	54,500	98,040	96,450**	110,000	118,080
eights (lb.)		Max Payload	6,000	27,960	32,520	35,400	38,420
		Useful Load	49,500	73,460	73,080	71.700	77,120
		Max Fuel	48,200	71,737	72,560	69,814	73,325
	Ava	ilable Payload w/Max Fuel	1,300	1,723	520	1,886	3,795
		ilable Fuel w/Max Payload	43,500	45,500	40,560	36,300	38,700
		Ммо	0.925	0.820	0.820	0.820	0.820
imits		Trans. Alt. FL/VMO	FL 290/340	FL 260/340	FL 250/350	FL 260/340	FL 260/340
		PSI/Sea-Level Cabin	10.7/31,900	9.0/24,000	8.3/22,000	9.0/24,000	9.0/24,000
		TOFL (SL elev./ISA temp.)	6,299	6,085	6,170	6,630	8,200
	T	OFL (5,000-ft. elev.@25C)	11,139	10,330	8,360	NA	NA
irport		Mission Weight	103,600	171,000	168,650	NA	NA
•		NBAA IFR Range	7,437	6,297	6,000	NA	NA
erformance		V2	148	141	137	NA	NA
		VREF	114	117	111	122	124
		Landing Distance	2,680	2,360	2,220	2,440	2,570
		Time to Climb/Altitude	21/FL 370	25/FL 370	22/360	24/FL 350	26/FL 330
limb	FAR 25 Engine-Out Rate (fpm)		NA	NA	NA	NA	NA
	FAR 25 Engine-Out Gradient (ft./nm)		NA NA	NA	NA	NA	NA
		Certificated	51,000	41,000	41,000	41,000	41,000
eiling (ft.)		All-Engine Service	41,000	NA	36,000	NA	NA NA
		Engine-Out Service	25,000	NA 450	18,000	NA 455	NA 457
		TAS Fuel Flow	488 2,883	452 4,679	447 4,695	455 NA	457 NA
	Long Range	Altitude	2,003 FL 450	FL 390	4,695 FL 370	FL 380	FL 360
		Specific Range	0.169	0.097	0.095	NA	NA
Cruise		TAS	516	470	470	471	471
		Fuel Flow	3,136	5,550	5,830	NA NA	NA NA
	High Speed	Altitude	FL 450	FL 370	FL 370	FL 360	FL 360
		Specific Range	0.165	0.085	0.081	NA NA	NA NA
		Nautical Miles	6,459	3,306	2,679	2,692	2,628
	Max Payload	Average Speed	481	437	434	NA	NA
	(with available fuel)	Trip Fuel	40,285	39,508	33,677	NA	NA
		Specific Range/Altitude	0.160/FL 490	0.084/FL 390	0.080/FL 370	NA/FL 370	NA/FL 350
		Nautical Miles	7,507	6,285	6,134	6,521	6,300
	Max Fuel	Average Speed	482	443	442	NA NA	NA NA
IBAA IFR	(with available	Trip Fuel	45,129	66,854	66,673	NA	NA
anges	payload)	Specific Range/Altitude	0.166/FL 510	0.094/FL 410	0.092/FL 410	NA/FL 390	NA/FL 390
200-nm		Nautical Miles	7,437	6,270	6,002	6,555	6,376
lternate)	Eight Passengers	Average Speed	482	443	442	NA	NA
,	(with available fuel)	Trip Fuel	44,820	66,723	65,558	NA	NA
		Specific Range/Altitude	0.166/FL 510	0.094/FL 410	0.092/FL 410	NA/FL 390	NA/FL 410
		Nautical Miles	7,636	6,348	6,200	6,619	6,441
	Form	Average Speed	482	442	442	NA	NA
	Ferry	Trip Fuel	45,168	66,886	67,207	NA	NA
		Specific Range/Altitude	0.169/FL 510	0.095/FL 410	0.092/FL 410	NA/FL 390	NA/FL 410
		Runway	3,241	3,485	4,075	NA	NA
	1,000 nm	Flight Time	2+10	2+27	2+26	NA	NA
	1,000 11111	Fuel Used	5,942	10,478	10,370	NA	NA
		Specific Range/Altitude	0.168/FL 510	0.095/FL 410	0.096/FL 410	NA/NA	NA/NA
		Runway	3,591	4,290	4,280	NA	NA
lissions	3.000 nm	Flight Time	6+17	6+54	6+54	NA	NA
passengers)	0,000 mm	Fuel Used	16,280	29,534	30,070	NA	NA
		Specific Range/Altitude	0.184/FL 510	0.102/FL 410	0.100/FL 410	NA/NA	NA/NA
		Runway	5,241	5,855	6,160	NA	NA
	6,000 nm	Flight Time	12+28	13+34	13+35	NA NA	NA NA
	-,000	Fuel Used	34,622	63,311	65,528	NA NA	NA NA
		Specific Range/Altitude	0.173/FL 510	0.095/FL 410	0.092/FL 410	NA/NA	NA/NA
emarks		Certification Basis	FAR 25, 2014	FAR 25 A 77, 1967/98 Split scimitar winglets. 2016 data.	FAR 25, 1999 *Also available with 26,500-lbf IAEV2527M- A5 engines; includes 6 additional center tanks plus VIP cabin. **Spec weight. BCA estimated data.	FAR 25 A TBD All data preliminary. 2016 data.	FAR 25 A TBD All data preliminary. 2016 data.