Solution of the Problems – Exam (August 2012/13)

Question 1: Internal combustion operating on dual cycle, given

$$T_1 = 30^{\circ} \text{C}$$
, $P_1 = 1 \text{ bar}$

Cylinder bore: 0.25 m, Stroke length: 0.40 m

$$r_c = 9 \; , \; r_e = 5$$

$$C_p = 1.0 \frac{kJ}{kq.K}$$
, $C_v = 0.71 \frac{kJ}{kq.K}$

(a) Calculating T_i and P_i , with $i \in \{1, 2, 3, 4, 5\}$ at all strokes:

1-2: Isentropic Compression

$$P_1V_1^n = P_2V_2^n \Longrightarrow P_2 = P_1 \left(\frac{V_1}{V_2}\right)^n = 1 \times 9^{1.25} = 15.59 \text{ bar}$$

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{n-1} = 9^{1.25-1} = 1.732 \Longrightarrow T_2 = 525.07 \text{ K}$$

3–4: Addition of Heat at Constant Pressure Calculating T_3 and T_4 :

$$\frac{T_4}{T_3} = \frac{V_4}{V_3} = \rho = \frac{r_c}{r_e} = \frac{9}{5} = 1.8 \Longrightarrow T_4 = 1.8T_3$$

The problem states that heat liberated at constant pressure is twice the heat liberated at constant colume, i.e.,

$$C_p(T_4 - T_3) = 2C_v(T_3 - T_2).$$

Substituting T_2 and T_4 ,

$$1.0 (1.8T_3 - T_3) = 2 \times 0.71 (T_3 - 525.07) \Longrightarrow T_3 = 1202.58 \text{ K}$$

$$\frac{P_3}{T_3} = \frac{P_2}{T_2} \Longrightarrow P_3 = \frac{15.59}{525.07} 1202.58 - 35.70 \text{ bar}$$

$$P_4 = P_3 \quad \text{and} \quad T_4 = 1.8T_3 = 2164.65 \text{ K}$$

4-5: Isentropic Expansion

$$P_4V_4^n = P_5V_5^n \Longrightarrow P_5 = P_4 \left(\frac{V_4}{V_5}\right)^n = P_4\frac{1}{r_e^n} = 35.70\frac{1}{5^{1.25}} = 4.775 \text{ bar}$$

$$\frac{T_5}{T_4} = \left(\frac{V_4}{V_5}\right)^{n-1} \Longrightarrow T_5 = T_4\frac{1}{r_e^{n-1}} = 1447.59 \text{ K}$$

Stroke	P (bar)	T(K)
1	1.0	303.15
2	15.59	525.07
3	35.70	1202.58
4	35.70	2164.65
5	4.78	1447.59

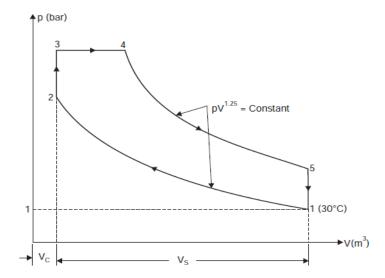


Figure 1: Pv diagram for Question 1.

- (b) Skecth of the Pv diagram in Fig. b.
- (c) Calculating MEP:

$$\begin{split} MEP &= \frac{1}{V_s} \left[P_3 \left(V_4 - V_3 \right) + \frac{P_4 V_4 - P_5 V_5}{n-1} - \frac{P_2 V_2 - P_1 V_1}{n-1} \right] \\ &= \frac{1}{r_c - 1} \left[P_3 \left(\rho - 1 \right) + \frac{P_4 \rho - P_5 r_c}{n-1} - \frac{P_2 - P_1 r_c}{n-1} \right] \\ &= \frac{1}{9-1} \left[35.70 \left(1.8 - 1 \right) + \frac{35.70 \times 1.8 - 4.78 \times 9}{1.25 - 1} - \frac{15.59 - 1 \times 9}{1.25 - 1} \right] \\ &= 10.895 \, \mathrm{bar} \end{split}$$

(d) The work done per cycle is defined as

$$W = MEP \times V_s$$

where

$$V_s = \frac{\pi}{4}D^2L = \frac{3.1415}{4}(0.25)^2 \times 0.4 = 0.0196 \text{ m}^3$$

Thus

$$W = 10.895 \text{ bar } \times 0.0196 \text{ m}^3 = 0.213542 \text{ bar.m}^3 = 21.354 \text{ kJ}$$

The heat supplied to the cycle is given by

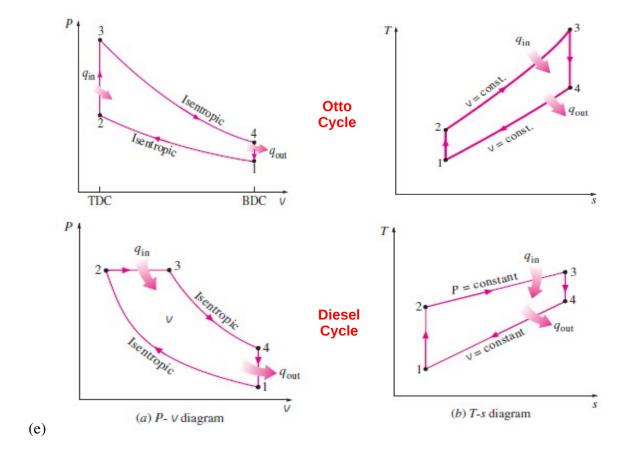
$$Q_{\text{cycle}} = mQ_s = m \left[C_p \left(T_3 - T_2 \right) + C_p \left(T_4 - T_3 \right) \right]$$

we need to calculate the mass (m) of air in the cylinder:

$$\begin{split} V_1 &= V_s + V_c = \frac{r_c}{r_c - 1} V_s = \frac{9}{9 - 1} \times 0.0196 = 0.02209 \text{ m}^3 \\ m &= \frac{P_1 V_1 MW}{RT_1} = \frac{1 \text{ bar} \times 0.02209 \text{ m}^3 \times 29 \frac{\text{g}}{\text{gmol}}}{8.3144621 \times 10^{-5} \frac{\text{m}^3.\text{bar}}{\text{K.gmol}} \times 303.15 \text{ K}} = 25.4141 \text{ g} \end{split}$$

Heat supplied =
$$mQ_s$$

= $0.02541 [0.71 (1202.58 - 525, 07) + 1.0 (2164.65 - 1202.58)]$
= 36.67 kJ



Question 2:

(a)

Amount of petroleum material

thermally equivalent to the biodiesel
$$= \left(75000 \times 103 \times \frac{37}{43}\right) \text{ kg releasing on burning}$$

$$= \left[\left(75000 \times 103 \times \frac{37}{43}\right) \times \frac{44}{14}\right] \times 10^{-3} \text{ tonnes of CO}_2$$

$$= 0.2 \text{ million tonnes of carbon dioxide.}$$

- (b) Trans-esterification with methanol.
- (c) It can be transported long distances by tanker, reducing the need for pipelines.
- (d)

Rate of electricity production =
$$= 1.2\times109 \text{kg}\times55\times106 \text{J.kg}^{-1}\times\frac{0.35}{3600\times365\times24} \text{W}$$
 = 730MW

Question 3:

(a) Symbols:

 \dot{Q} - the rate of heat added to the fluid;

 \dot{W}_s - the rate of shaft work done by the fluid;

 \dot{m} - the mass flux through the device;

 c_p - the specific heat capacity at constant pressure;

T - the fluid temperature;

u - the fluid velocity;

 χ_2 - the property χ evaluated at the device outlet;

 χ_1 - the property χ evaluated at the device inlet;

[3 Marks]

Assumptions:

- Steady fluid flow through the turbine;
- The fluid is an ideal gas;
- The velocity, pressure, internal energy and potential energy over each inlet or outlet can be replaced by their respective averages;
- No work due to viscosity, chemical, radiological etc;

[Half mark for each up to 2 Marks]

(b) For a steady flow device with two inlets (labelled 1 and 2) and one outlet (labelled 3), the steady flow energy equation can be written as

$$\dot{Q} - \dot{W}_s = \dot{m}_3 \left(c_p T_3 + \frac{u_3^2}{2} + g z_3 \right) - \dot{m}_1 \left(c_p T_1 + \frac{u_1^2}{2} + g z_1 \right) - \dot{m}_2 \left(c_p T_2 + \frac{u_2^2}{2} + g z_2 \right).$$

[2 Marks]

The corresponding steady mass conservation equation is

$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3$$
.

[1 Marks]

(c) (i) the volume flux is related to the velocity through

Fluid velocity,
$$u = \frac{\text{Volume flux}}{\text{Cross-sectional area, } A}$$
.

[1 Mark of 2]

At inlet 1,

$$u_1 = \frac{1.8 \text{ m}^3/\text{s}}{0.03 \text{ m}^2} = 60 \text{ m/s},$$

$$u_2 = \frac{0.5 \text{ m}^3/\text{s}}{0.1 \text{ m}^2} = 5 \text{ m/s},$$

$$u_3 = \frac{20 \text{ m}^3/\text{s}}{0.5 \text{ m}^2} = 40 \text{ m/s},$$

[1 Mark of 2]

(ii) The fluid density at inlet 1 and outlet 3 can be calculated directly

$$\begin{split} \rho_1 = & \frac{p_1}{RT_1} = \frac{200000\,\mathrm{Pa}}{300\,\mathrm{J/(kg\;K)}\,(80 + 273.15)\;K} = 1.8878\,\mathrm{kg/m^3}, \\ \rho_3 = & \frac{p_3}{RT_3} = \frac{110000\,\mathrm{Pa}}{300\,\mathrm{J/(kg\;K)}\,(30 + 273.15)\;K} = 1.2095\,\mathrm{kg/m^3}. \end{split}$$

[1 Mark of 6]

The mass flux at inlet 1 and outlet 3 are given by

$$\dot{m}_1 = \rho_1 u_1 A_1 = 1.8878 \text{ kg/m}^3 \times 60 \text{ m/s} \times 0.03 \text{ m}^2 = 22.6533 \text{ kg/s},$$

 $\dot{m}_3 = \rho_3 u_3 A_3 = 1.2095 \text{ kg/m}^3 \times 40 \text{ m/s} \times 0.5 \text{ m}^2 = 24.1904 \text{ kg/s}.$

[1 Mark of 6]

Now the mass flux through the inlet satisfies

$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3$$

[1 Mark of 6]

i.e.

$$\dot{m}_2 = \dot{m}_3 - \dot{m}_1 = 24.1904 \,\mathrm{kg/s} - 22.6533 \,\mathrm{kg/s} = 1.5372 \,\mathrm{kg/s}.$$

[1 Mark of 6]

Consequently the density

$$\rho_2 = \frac{\dot{m}_2}{u_2 A_2} = \frac{1.5372 \, \mathrm{kg/s}}{5 \, \mathrm{m/s} \times 0.1 \, \mathrm{m}^2} = 3.0744 \, \mathrm{kg/m}^3,$$

[1 Mark of 6]

and the pressure

$$p_2 = \rho_2 R T_2 = 3.0744 \,\mathrm{kg/m^3} \times 300 \,\mathrm{J/(kg \ K)} \times (70 + 273.15) \,\, K = 316490 \,\mathrm{Pa} = 316 \,\mathrm{kPa}.$$

[1 Mark of 6]

(iii) Now we can use the SFEE in the form

$$\dot{Q} - \dot{W}_s = \dot{m}_3 \left(c_p T_3 + \frac{u_3^2}{2} + g z_3 \right) - \dot{m}_1 \left(c_p T_1 + \frac{u_1^2}{2} + g z_1 \right) - \dot{m}_2 \left(c_p T_2 + \frac{u_2^2}{2} + g z_2 \right).$$

to calculate the rate at which heat is added to the gas.

$$\dot{Q} = -80000 + \dot{m}_3 \left((800 \times 303.15) + \frac{40^2}{2} + (9.81 \times 0.5) \right)$$

$$- \dot{m}_1 \left((800 \times 353.15) + \frac{60^2}{2} + (9.81 \times 0.2) \right)$$

$$- \dot{m}_2 \left((800 \times 343.15) + \frac{5^2}{2} + (9.81 \times 1.2) \right)$$

$$= -80000 + 5886138 - 6440820 - 422024$$

$$= -1056707.0 \,\mathbf{W} = -1057 \,\mathbf{kW}.$$

[2 Marks]

The rate at which heat added to the gas is negative indicating the gas in the device heats the surroundings. [1 Mark]

Question 4: The schematic and Ts diagrams are given bellow (Fig.).

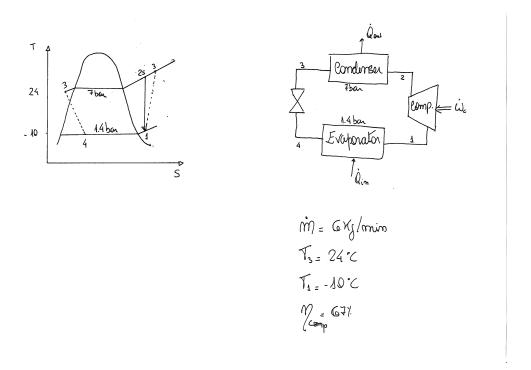


Figure 2: Schematic and Ts diagram for Question 4.

- (a) State 1: At $T_1 = 10^{\circ}$ C and $P_1 = 1.4$ bar, the thermodynamic table for R-134a gives $H_1 = 243.40$ kJ/kg and $S_1 = 0.9606$ kJ/(kg..K).
- (b) State 2: For isentropic compression to $P_2 = 7$ bar, $S_{2s} = S_1$. From linear interpolation, we can find $H_{2s} = 278.06$ kJ/kg. Now, using the (given) compressor efficiency,

$$\eta = \frac{H_{2s} - H_1}{H_2 - H_1} = 0.67 \Longrightarrow H_2 = 245.13 \text{ kJ/kg}, \ \ S_2 = 1.0135 \text{ kJ/(kg.K)}$$

- (c) State 3: For $P_3=7$ bar and $T_3=24$ °C, the thermodynamic table gives $H_3=H_f\left(T=297.15~\mathrm{K}\right)=82.90~\mathrm{kJ/kg}$ and $S_3=0.3113~\mathrm{kJ/(kg.K)}$
- (d) State 4: Throttling process $\implies H_4 = H_3 = 82.90 \text{ kJ/kg}$ and $S_4 = 0.33011 \text{ kJ/(kg.K)}$

The coefficient of performance, β is given by

$$\beta = \frac{H_1 - H_4}{H_2 - H_1} = 3.10$$

And the refrigerant capacity is

$$\dot{Q}_{in} = \dot{m} \left(H_1 - H_4 \right) = \left(6 \frac{\text{kg}}{\text{min}} \right) \left(243.40 - 82.90 \right) \frac{\text{kJ}}{\text{kg}} \left[\frac{1 \text{ ton}}{211 \text{ kJ/min}} \right] = 4.564 \text{ tons}$$

Question 5: The specific humidity ω is the ratio of the mass of water vapour m_v , to the mass of dry air m_a and satisfies the equation

$$\omega = \frac{m_v}{m_a}.$$

As both water vapour and dry air behave like ideal gases, in some arbitrary volume V,

$$\omega = \frac{m_v}{m_a} = \frac{\rho_v}{\rho_a} = \frac{p_v}{R_v T} \frac{R_a T}{p_a} = \frac{R_a p_v}{R_v p_a}.$$

The partial pressures of dry air and water vapour satisfy $p_a = p - p_v$. Hence

$$\omega = \frac{R_a p_v}{R_v \left(p - p_v \right)}.$$

The saturation pressure of water $p_{v,\text{sat}}$ is the maximum partial pressure of water vapour a gas can contain at a give temperature before water starts condensing out of the gas. The relative humidity φ is the ratio of the partial pressure of water vapour to the saturation pressure of water

$$\varphi = \frac{p_v}{p_{v,\text{sat}}}.$$

Hence eliminating p_v from the previous expression gives

$$\omega = \frac{R_a \varphi p_{v,\text{sat}}}{R_v \left(p - \varphi p_{v,\text{sat}} \right)}.$$

[5 Marks]

If the inlet from inside is labelled 1, the inlet from outside is labelled 2 and the outlet with the mixture is labelled 3, then

$$\dot{m}_{a_1} + \dot{m}_{a_2} = \dot{m}_{a_3}$$
, Mass conservation of dry air, $\omega_1 \dot{m}_{a_1} + \omega_2 \dot{m}_{a_2} = \omega_3 \dot{m}_{a_3}$, Mass conservation of water vapour, $h_1 \dot{m}_{a_1} + h_2 \dot{m}_{a_2} = h_3 \dot{m}_{a_3}$, Energy conservation,

where \dot{m}_a is a mass flux of air, h is the enthalpy and ω is the specific humidity. [3 Marks of 7]

Eliminating \dot{m}_3 from the water vapour conservation equation gives

$$\omega_1 \dot{m}_{a_1} + \omega_2 \dot{m}_{a_2} = \omega_3 \left(\dot{m}_{a_1} + \dot{m}_{a_2} \right) = \omega_3 \dot{m}_{a_1} + \omega_3 \dot{m}_{a_2}.$$

Collecting together terms involving \dot{m}_1 and terms involving \dot{m}_2 gives

$$\omega_1 \dot{m}_{a_1} - \omega_3 \dot{m}_{a_1} = \omega_3 \dot{m}_{a_2} - \omega_2 \dot{m}_{a_2} \Rightarrow (\omega_1 - \omega_3) \, \dot{m}_{a_1} = (\omega_3 - \omega_2) \, \dot{m}_{a_2},$$

and hence rearranging gives

$$\frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} = \frac{\omega_1 - \omega_3}{\omega_3 - \omega_2} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3}.$$

[2 Marks of 7]

Similarly eliminating \dot{m}_3 from the energy conservation equation gives

$$h_1\dot{m}_{a_1} + h_2\dot{m}_{a_2} = h_3\left(\dot{m}_{a_1} + \dot{m}_{a_2}\right) = h_3\dot{m}_{a_1} + h_3\dot{m}_{a_2}.$$

Collecting together terms involving \dot{m}_1 and terms involving \dot{m}_2 gives

$$h_1 \dot{m}_{a_1} - h_3 \dot{m}_{a_1} = h_3 \dot{m}_{a_2} - h_2 \dot{m}_{a_2}$$

$$\Rightarrow (h_1 - h_3) \, \dot{m}_{a_1} = (h_3 - h_2) \, \dot{m}_{a_2},$$

and hence rearranging gives

$$\frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} = \frac{h_1 - h_3}{h_3 - h_2} = \frac{h_3 - h_1}{h_2 - h_3} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3},$$

as required.

[2 Marks of 7]

(a) The partial pressure of dry air at 1 is given by

$$p_{a_1} = p_1 - p_{v,\text{sat}_1} = 100000 \,\text{Pa} - 1818.747 \,\text{Pa} = 98181.253 \,\text{Pa}.$$

The specific volume of dry air

$$V_1 = \frac{R_a T_1}{p_{a_1}} = \frac{287.1 \,\text{kJ/(kg K)} (16 + 273.15)}{98181.253 \,\text{Pa}} = \,\text{m}^3\text{/kg}$$

[1 Mark of 3]

The mass flux of dry air through inlet 1 is given by

$$\dot{m}_{a_1} = \frac{\text{volume flux}}{V_1} = \frac{1 \text{ m}^3/\text{s}}{0.8455 \text{ m}^3/\text{kg}} = 1.1827 \text{ kg/s}.$$

[1 Mark of 3]

The mass flux of dry air through inlet 2 is given by

$$\dot{m}_{a_2} = \dot{m}_{a_3} - \dot{m}_{a_1} = 1.8 \,\text{kg/s} - 1.1827 \,\text{kg/s} = 0.6173 \,\text{kg/s}$$

[1 Mark of 3]

(b) The gas entering the system through inlet 1 is saturated ($\varphi_1 = 1$), and therefore the specific humidity

$$\begin{split} \omega_1 = & \frac{R_a \varphi_1 p_{v, \text{sat}_1}}{R_v \left(p_1 - \varphi_1 p_{v, \text{sat}_1} \right)} = \frac{287.1 \, \text{kJ/(kg K)} \times 1 \times 1818.747 \, \text{Pa}}{461.5 \, \text{kJ/(kg K)} \left(1000000 \, \text{Pa} - 1 \times 1818.747 \, \text{Pa} \right)} \\ = & 0.0115 \, \text{kg H}_2 \text{O/ kg dry air.} \end{split}$$

[1 Mark]

(c) Using

$$\frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3}.$$

Rearranging

$$(\omega_2 - \omega_3) \frac{\dot{m}_{a_2}}{\dot{m}_{a_3}} = (\omega_3 - \omega_1).$$

Collecting terms involving ω_3 gives

$$\omega_2 \frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} + \omega_1 = \omega_3 + \omega_3 \frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} = \omega_3 \left(1 + \frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} \right).$$

Therefore

$$\omega_3 = \frac{\omega_1 + \omega_2 \frac{\dot{m}_{a_2}}{\dot{m}_{a_1}}}{1 + \frac{\dot{m}_{a_2}}{\dot{m}_{a_1}}}$$

[2 Marks of 4]

The mass flux ratio

$$\frac{\dot{m}_{a_2}}{\dot{m}_{a_1}} = \frac{0.6173}{1.1827} = 0.5219$$

[1 Mark of 4]

Therefore

$$\omega_3 = \frac{0.0115 + (0.5219 \times 0.0182)}{1 + 0.5219} = 0.0138 \,\mathrm{kg}\,\mathrm{H_2O}/\,\mathrm{kg}$$
 dry air.

[1 Mark of 4]