## Solution of the Problems – Exam (May 2012/13)

## **Question 1:** Reheat Rankine cycle with two turbines:

(a) In order to fill the Table we need to calculate the thermodynamic properties for each stage of the cycle

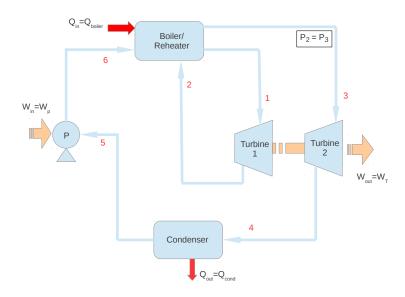


Figure 1: Reheat Rankine cycle with 2 turbines.

**Stage 1:** The fluid leaving the boiler towards the first turbine is at 40 bar and 370°C. This is well above the saturation temperature  $(T_{\text{sat}} = 250.3^{\circ}\text{C})$  and we can thus assume that the fluid is a superheated steam. At such pressure, the superheated steam tables (SHST) give,

T (°C)	H (kJ/kg)	S(kJ/(kg.K))		
350	3092.5	6.582		
400	3213.6	6.769		

Thus, with linear interpolation at  $T_1=370^{\rm o}{\rm C}$ :  $H_1=3140.94\frac{kJ}{kg}$  and  $S_1=6.6568\frac{kJ}{kg.K}$ 

**Stage 3:** The reheated steam is at 7 bar  $(T_{\rm sat}=165^{\rm o}{\rm C})$  and 370°C is also superheated and from SHST,

$$\begin{array}{cccc} {\bf T~(^{o}C)} & {\bf H~(kJ/kg)} & {\bf S~(kJ/(kg.K))} \\ 350 & 3163.7 & 7.473 \\ 400 & 3268.7 & 7.635 \end{array}$$
 At  $T_1=370^{o}C$ :  $H_3=3205.7\frac{kJ}{kg}$  and  $S_3=7.5378\frac{kJ}{kg.K}$ 

Stage 2: After an isentropic expansion in the first turbine, therefore

$$P_{2s} = P_3 = 7 \text{ bar}$$
 and  $S_{2s} = S_1 = 6.6568 \frac{kJ}{kq.K}$ 

The properties of the fluid can be calculated from the saturated steam tables:

$$\begin{array}{cccc} & (kJ/kg) & (kJ/(kg.K)) \\ H_f & 697.1 & S_f & 1.9918 \\ H_{fg} & 2064.9 & S_{fg} & 4.7134 \\ H_g & 2762.0 & S_g & 6.7052 \end{array}$$

In order to calculate  $H_2$ , we need to calculate  $H_{2s}$  first,

$$H_{2s} = H_f + x_{2s}H_{fg}$$

Thus we should start calculating the quality of the steam flow,  $x_{2s}$ , given by

$$x_{2s} = \frac{S_{2s} - S_f}{S_{fg}} = \frac{6.6568 - 1.9918}{4.7134} = 0.9897$$

And now,

$$H_{2s} = 697.1 + 0.9897 \times 2064.9 = 2554.84 \frac{kJ}{kg}$$

We can finally calculate  $H_2$  as,

$$H_2 = H_1 - (H_1 - H_{2s}) \eta_{T1} = 3140.94 - (3140.94 - 2754.84) \times 0.84 = 2816.61 \frac{kJ}{ka}$$

**Stage 4:** The fluid that left the second turbine is a saturated steam at 0.10 bar with:

$$P_{4s} = 0.10 \text{ bar}$$
 and  $S_{4s} = S_3 = 7.5378 \frac{kJ}{kg.K}$ 

From the saturated steam tables:

The quality of the steam is

$$x_{4s} = \frac{S_{4s} - S_f}{S_{fg}} = \frac{7.5378 - 0.649}{7.501} = 0.9184$$

and the enthalpy,

$$H_{4s} = H_f + x_{4s}H_{fg} = 191.8 + 0.9184 \times 2392.8 = 2389.35 \frac{kJ}{kg}$$

**Stage 5:** The fluid after the condenser is a saturated water with the following characteristics:

$$P_5 = 0.10 \text{ bar}, \quad x_5 = 0.0, \quad V_5 = V_f = 0.0101 \\ \frac{m^3}{kg}, \quad H_5 = 191.8 \\ \frac{kJ}{kg} \text{ and } S_5 = 0.649 \\ \frac{kJ}{kg.K}$$

**Stage 6:** Finally, the fluid leaving the isentropic boiler feed pump has the following characteristics:

$$P_{6s} = P_6 = 40 \text{ bar}, \quad S_{6s} = S_5 = S_5 = 0.649 \frac{kJ}{kg.K} \text{ and } H_{6s} = H_{fg} = 1712.9 \frac{kJ}{kg}$$

As the efficiency of the pump is 61%:

$$H_6 = H_5 + V_5 \frac{(P_6 - P_5)}{\eta_P} = 191.8 \frac{kJ}{kg} + \frac{0.0101 \frac{m^3}{kg} \times (40 - 0.10) \text{ (bar)}}{0.610} = 257.86 \frac{kJ}{kg}$$

Thus the Table becomes:

Stage	P	T	State	Н	S	
	(bar)	$(^{o}\mathbf{C})$		$(\mathbf{kJ}.\mathbf{kg}^{-1})$	$(\mathbf{kJ.}(\mathbf{kg.K})^{-1})$	
1	40	370	superheated	(a) 3140.94	(b) 6.6568	
			steam			
2	_	_	(c) saturated	_	_	
			steam			
3	7	370	superheated	(d) 3205.7	(e) 7.5358	
			steam			
4	0.10	_			_	
5	0.10	_	(f) saturated	(g) 191.8	(h) 0.649	
			liquid( or water)			
6	40	_	(i) saturated (j) 257.86		_	
			liquid( or water)			

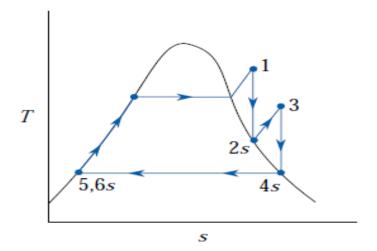
(b) The thermal efficiency of the cycle is:

$$\eta_{\text{Thermal}} = \frac{(H_1 - H_{2s}) \, \eta_{\text{T1}} + (H_3 - H_{4s}) \, \eta_{\text{T2}} - V_5 \, (P_6 - P_5) \, \eta_{\text{P}}^{-1}}{(H_1 - H_6) + (H_3 - H_2)}$$

$$= \frac{(3140.94 - 2754.84) \, 0.84 + (3205.7 - 2389.35) \, 0.80 - \frac{0.0101 \, (40 - 0.10)}{0.610} \frac{100}{1.0}}{(3140.44 - 257.86) + (3205.7 - 2816.61)}$$

$$= 0.2785$$

(c) Sketch of the Ts diagram



## **Question 2:**

(a) 
$$421 \text{ billion kWh} = 421 \times 10^{12} \text{ J.s}^{-1} \times 3600 \text{ s} = 1.5 \times 10^{18} \text{ J of electricity.}$$

Efficiencies of generation depend on turbine performance not on whether nuclear and or chemical fuels were used, so the electricity above must have been obtained from about:

$$\left(1.5\times10^{18}/0.35\right)\,\mathrm{J}$$
 of heat  $=4.3\times10^{18}\,\mathrm{J}$  of heat

If this had been raised from natural gas, the carbon dioxide release would have been:

$$[4.3 \times 10^{18} \text{ J/} (889103 \text{ J.mol}^{-1})] \times 0.044 \text{ kg.mol}^{-1} \times 10^{-3} \text{ tonne.kg}^{-1} = 213 \text{ million tonnes}$$

- (b) A supertanker, carrying 1 million barrels of oil, would if it blew up have a blast equivalent to that from the atomic bomb at Hiroshima. A blast of this magnitude was actually observed at a refinery fire in Venezuela in 2012 (either of these).
- (c) Rate of supply of coke =  $\left[300 \times 10^6 \text{ J. s}^{-1} / \left(25 \times 10^6 \text{ J.kg}^{-1}\right)\right] = 12 \text{ kg.s}^{-1}$  rate of production of carbon dioxide =  $12 \text{ kg.s}^{-1} \times (44/12) \times 7 \times 24 \times 3600 \times 0.001$  tonne per week = 26611 tonne per week.

For 10% mitigation of the carbon 10% of the heat must come from the citrus peel.  $10.8~{\rm kg.s^{-1}}$  of coke plus:  $(1.2\times25/7)~{\rm kg.s^{-1}}$  of citrus peel =  $4.3~{\rm kg.s^{-1}}$  of citrus peel Ratio coke to citrus peel = (10.8/4.3)=2.5

(d) Suppliers are monitored by the Forest Stewardship Council (FSC) for replacement of trees felled with new plantings.

**Question 3:** (a) The inlet has circumference 1 m. Therefore the radius of the inlet  $r_1 = 1/2\pi = 0.15915$  m and the area of the inlet  $A_1 = \pi r_1^2 = 0.07958$  m<sup>2</sup>.

Similarly the outlet has circumference  $0.6 \,\mathrm{m}$ . Therefore the radius of the outlet  $r_2 = 0.6/2\pi = 0.09549 \,\mathrm{m}$  and the area of the outlet  $A_2 = \pi r_2^2 = 0.02865 \,\mathrm{m}^2$ . [1 Mark of 4] Evaluating the mass flux at the inlet gives

$$\rho_1 = \frac{\dot{m}}{u_1 A_1} = \frac{4 \,\mathrm{kg \, s^{-1}}}{30 \,\mathrm{m \, s^{-1}} \times 0.07958 \,\mathrm{m^2}} = 1.6755 \,\mathrm{kg \, m^{-3}}.$$

[1 Mark of 4]

Rearranging the SFEE to give the gas velocity at the outlet implies

$$u_2^2 = u_1^2 + \frac{2(\dot{Q} - \dot{W}_s)}{\dot{m}} + 2(h_1 - h_2).$$

Therefore

$$u_2^2 = 30^2 + \frac{2(-15000 - 30000)}{4} + 2(70000 - 40000)$$
  
= 900 - 22500 + 60000  
= 38400 m<sup>2</sup> s<sup>-2</sup>.

giving a fluid velocity at the outlet of

$$u_2 = 195.959 \,\mathrm{m \, s^{-1}}$$

[1 Mark of 4]

Now the gas density at the outlet can be calculated from the mass flux

$$\rho_2 = \frac{\dot{m}}{u_2 A_2} = \frac{4 \,\mathrm{kg \, s^{-1}}}{195.959 \,\mathrm{m \, s^{-1}} \times 0.02865 \,\mathrm{m}^2} = 0.7125 \,\mathrm{kg \, m^{-3}}.$$

Finally the difference in gas density the turbine is given by

$$\Delta \rho = \rho_1 - \rho_2 = 1.6755 - 0.7125 = 0.9630 \,\mathrm{kg} \,\mathrm{m}^{-3}.$$

[1 Mark of 4]

(b) The differential forms of for mass and energy conservation are

$$\frac{\mathrm{d}V}{V} - \frac{\mathrm{d}u}{u} - \frac{\mathrm{d}A}{A} = 0,$$
$$\mathrm{d}h + u\,\mathrm{d}u = 0.$$

[1 Mark of 2]

Eliminating du between these two expressions gives

$$\frac{\mathrm{d}V}{V} + \frac{\mathrm{d}h}{u^2} - \frac{\mathrm{d}A}{A} = 0$$

## [1 Mark of 2]

The speed of sound is the distance travelled during a unit of time by a sound wave propagating through a compressible medium. The Mach number is the non-dimensional ratio of the speed of a body moving through a fluid to the local speed of sound. For an isentropic process, the speed of sound is given by

$$c = \left(\frac{\partial p}{\partial \rho}\right)^{1/2},\,$$

while the Mach number is defined to be

$$Ma = \frac{u}{c},$$

[4 Marks]

For an isentropic process the specific volume  $V=1/\rho$ , is a function of just pressure and therefore satisfies

$$\mathrm{d}V = \frac{\mathrm{d}V}{\mathrm{d}p}\mathrm{d}p.$$

In this expression the derivative can be written in terms of the speed of sound c since

$$\frac{\mathrm{d}V}{\mathrm{d}p} = \frac{\partial V}{\partial \rho} \frac{\partial \rho}{\partial p} = -\frac{V^2}{c^2},$$

and therefore

$$\mathrm{d}V = -\frac{V^2}{c^2} \mathrm{d}p.$$

[2 Marks of 3]

For a general thermodynamic process

$$dh = T ds + V dp.$$

However for an isentropic process the entropy remains constant ds = 0, and the enthalpy is a function of just pressure. Changes in enthalpy are not related to changes in entropy, and

$$\mathrm{d}h = V\,\mathrm{d}p.$$

[1 Mark of 3]

Hence if we eliminate dV and dh,

$$-\frac{V}{c^2}\mathrm{d}p + \frac{V}{u^2}\mathrm{d}p - \frac{\mathrm{d}A}{A} = 0.$$

Collecting together terms involving dp gives

$$\frac{\mathrm{d}A}{A} = \left(\frac{V}{u^2} - \frac{V}{c^2}\right) \mathrm{d}p = \frac{V}{u^2} \left(1 - \frac{u^2}{c^2}\right) \mathrm{d}p.$$

Given the definition of Mach number

$$\frac{\mathrm{d}A}{A} = \frac{V}{u^2} \left( 1 - \mathrm{Ma}^2 \right) \mathrm{d}p.$$

For an isentropic process the speed of sound can be used to eliminate dp, giving

$$\frac{\mathrm{d}A}{A} = \frac{V}{u^2} \left( 1 - \mathrm{Ma}^2 \right) c^2 \mathrm{d}\rho.$$

Rearranging and using the definition of Mach number and specific density gives

$$\frac{1}{(1 - \mathrm{Ma}^2) A} \mathrm{d}A = \frac{1}{\rho \mathrm{Ma}^2} \mathrm{d}\rho.$$

If we're interested in changes along a pipe whose length is parameterized by x, then

$$\frac{1}{(1 - \mathrm{Ma}^2) A} \frac{\mathrm{d}A}{\mathrm{d}x} = \frac{1}{\rho \mathrm{Ma}^2} \frac{\mathrm{d}\rho}{\mathrm{d}x},$$

as required. [5 Marks]

For a supersonic diffuser  $(1 - \text{Ma}^2) < 0$ , while  $\frac{\text{d}A}{\text{d}x} > 0$ , A > 0,  $\rho > 0$  and  $\text{Ma}^2 > 0$ . Therefore  $\frac{\text{d}\rho}{\text{d}x} < 0$  and the gas density falls as gas flows along a supersonic diffuser. [2 Marks]

**Question 4:** Figure 2 shows the Ts diagram for the refrigeration cycle.

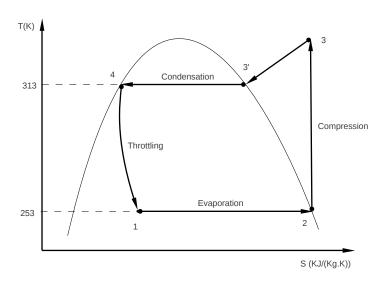


Figure 2: Refrigeration cycle – Question 4

From the given thermodynamic table for *Freon-12*,

	T	$P_s$	$V_g$	$H_f$	$H_g$	$S_f$	$S_g$	Specific Heat
	$(^{o}C)$	(bar)	$(m^3/kg)$	(kJ/kg)	(kJ/kg)	(kJ/(kg.K))	(kJ/(kg.K))	(kJ/(kg.K))
ĺ	-20	1.509	0.1088	17.8	178.61	0.073	0.7082	_
	40	9.607	_	74.53	203.05	0.2716	0.682	0.747

$$H_2 = 178.61 \; \mathrm{kJ/kg}, \;\; H_3' = 203.05 \; \mathrm{kJ/kg}, \;\; H_{f4} = H_1 = 74.53 \; \mathrm{kJ/kg}$$

Calculating mass flow rate,

Cooling Effect = 
$$20 = \dot{m} (H_2 - H_1) = \dot{m} (178.61 - 74.53) \Longrightarrow \dot{m} = 0.192 kg/s$$

As stage 2-3 is isentropic  $\Rightarrow S_3 = S_2$  and

$$S_3' + C_p \ln\left(\frac{T_3}{T_3'}\right) = 0.7082$$
  
 $0.682 + 0.747 \ln\left(\frac{T_3}{313.15}\right) = 0.7082 \Longrightarrow T_3 = 330.95 \text{ K}$ 

Now we can calculate the enthalpy of state 3:

$$H_3 = H_3' + C_p (330.95 - 313.15) = 203.05 + 0.747 (330.95 - 313.15) = 216.34kJ/kg$$

The power required by the machine is

$$\dot{m}(H_3 - H_2) = 7.25 \text{ kW} = 7244.91 \text{ W}$$

In order to calculate the piston displacement V, we first need to compute the volumetric efficiency,

$$\eta_{\text{vol}} = 1 + C - C \left(\frac{P_d}{P_s}\right)^{1/n} = 1 + 0.03 - 0.03 \left(\frac{9.607}{1.509}\right)^{1.13} = 0.876$$

The volume of refrigerant at the intake condition is:

$$V_{\text{ref}}^{(\text{intake})} = \dot{m} \times v_g = 0.192 \times .1088 = 0.02089 \frac{\text{m}^3}{\text{s}}$$

The swept volume can now be calculated as

$$V_{\text{swept}} = \frac{V_{\text{ref}}^{(\text{intake})}}{\eta_{\text{vol}}} = \frac{0.0289}{0.876} = 2.3847 \times 10^{-2} \frac{\text{m}^3}{\text{s}}$$

Finally, the piston displacement is:

$$V = \frac{V_{\text{swept}} \times 60}{300} = 4.769 \times 10^{-3} \text{ m}^3$$

**Question 5:** The specific humidity  $\omega$  is the ratio of the mass of water vapour  $m_v$ , to the mass of dry air  $m_a$  and satisfies the equation

$$\omega = \frac{m_v}{m_a}.$$

As both water vapour and dry air behave like ideal gases, in some arbitrary volume V,

$$\omega = \frac{m_v}{m_a} = \frac{\rho_v}{\rho_a} = \frac{p_v}{R_v T} \frac{R_a T}{p_a} = \frac{R_a p_v}{R_v p_a}.$$

The ratio of specific gas constants  $R_a/R_v=0.622$ , while the partial pressures of dry air and water vapour satisfy  $p_a=p-p_v$ . Hence

$$\omega = \frac{0.622p_v}{p - p_v}.$$

[4 Marks]

The saturation pressure of water

$$p_q = \varphi p_v$$
.

Hence eliminating  $p_v$  from the previous expression gives

$$\omega = \frac{0.622\varphi p_g}{p - \varphi p_g}.$$

[2 Marks]

The heating and humidification are split into two steady process. Firstly a heater (with inlet properties labelled 1 and outlet properties labelled 2) and secondly a humidifier (with inlet properties labelled 2 and outlet properties labelled 3).

(a) The partial vapour pressure at the inlet 1, is

$$p_{v_1} = \varphi_1 p_{g_1} = \varphi p_{\text{sat @ 10^{\circ}C}} = 0.25 \times 1.4028 \,\text{kPa} = 0.3507 \,\text{kPa}.$$

[1 Mark]

Hence the partial pressure of dry air is given by

$$p_{a_1} = p_1 - p_{v_1} = 100 \,\text{kPa} - 0.3507 \,\text{kPa} = 99.6493 \,\text{kPa}.$$

[1 Mark]

The specific humidity is given by

$$\omega_1 = \frac{0.622 p_{v_1}}{p_1 - p_{v_1}} = \frac{0.622 \times 0.3507 \, \text{kPa}}{100 \, \text{kPa} - 0.3507 \, \text{kPa}} = 0.00219 \, \text{kg H}_2\text{O/ kg dry air}.$$

[1 Mark]

(b) Applying the mass and energy balances on the heating section gives

Dry air mass balance: 
$$\dot{m}_{a_1} = \dot{m}_{a_2} = \dot{m}_a$$
,

Water mass balance: 
$$\dot{m}_{a_1}\omega_1 = \dot{m}_{a_2}\omega_2$$
,  $\Rightarrow \omega_1 = \omega_2$ ,

Energy balance: 
$$\dot{Q} = \dot{m}_a h_2 - \dot{m}_a h_1$$
.

[2 Marks]

The total specific enthalpy at 1, the inlet is

$$\begin{aligned} h_1 &= c_p T_1 + \omega_1 h_{g_1} = & (1.005 \, \text{kJ/(kg K)} \times (12 + 173.15) \, \text{K}) \\ &+ (0.00219 \times 2523 \, \text{kJ/kg}) \\ &= & 292.0988 \, \text{kJ/kg}. \end{aligned}$$

The total specific enthalpy at 2, the outlet of the heating is

$$\begin{aligned} h_2 &= c_p T_2 + \underbrace{\omega_2}_{=\omega_1} h_{g_2} = (1.005 \, \text{kJ/(kg K)} \times (20 + 173.15) \, \, \text{K}) \\ &\quad + (0.00219 \times 2537 \, \text{kJ/kg}) \\ &\quad = 300.1693 \, \text{kJ/kg} \end{aligned}$$

[2 Marks]

The specific volume of dry air at 1, is given by

$$V_1 = \frac{R_a T_1}{p_{a_1}} = \frac{287.058 \,\text{J/(kg K)} \left(12 + 273.15\right) \,\text{K}}{99649.3 \,\text{Pa}} = 0.8215 \,\text{m}^3\text{/kg}.$$

Therefore the mass flux of dry air through the inlet

$$\dot{m}_a = \frac{q_1}{V_1} = \frac{40 \text{ m}^3/\text{min}}{0.8185 \text{ m}^3/\text{kg}} = 48.6886 \text{ kg/min},$$

where  $q_1 = 40 \,\mathrm{m}^3$ /min is the total volume flux through the inlet. [1 Mark] Hence the energy conservation equation gives the rate at which heat is transferred to the air

$$\dot{Q} = \dot{m}_a \left(h_2 - h_1\right) = 48.6958 \, \text{kg/min} \left(300.1693 \, \text{kJ/kg} - 292.0988 \, \text{kJ/kg}\right) = 392.9488 \, \text{kJ/min}.$$

[1 Mark]

(c) The mass balance for water in the humidifying section can be expressed as

$$\dot{m}_{a_2}\omega_2 + \dot{m}_w = \dot{m}_{a_2}\omega_3$$

or

$$\dot{m}_w = \dot{m}_a \left( \omega_3 - \omega_2 \right).$$

[2 Marks]

Here  $\omega_2=\omega_1$ , while the specific humidity at 3, the outlet is given by

$$\omega_3 = \frac{0.622 \varphi_3 p_{g_3}}{p_3 - \varphi_3 p_{g_3}} = \frac{0.622 \times 0.55 \times 2.9858}{100 - (0.55 \times 2.9858)} = 0.0104\,\mathrm{kg}\,\mathrm{H_2O/\,kg}\,\mathrm{dry}\,\mathrm{air}.$$

[1 Mark]

Therefore the required mass flow rate of steam is

$$\dot{m}_w = \dot{m}_a \left( \omega_3 - \omega_2 \right) = 48.6886 \, \text{kg/min} \left( 0.0104 - 0.00219 \right) = 0.3990 \, \text{kg/min}$$

[2 Marks]