

Solution of the Problems – Resit Exam Paper (August 2012/13)

Question 1: Internal combustion operating on dual cycle, given

$$T_1 = 30^\circ\text{C}, \quad P_1 = 1 \text{ bar}$$

Cylinder bore: 0.25 m, Stroke length: 0.40 m

$$r_c = \frac{V_1}{V_2} = 9, \quad r_e = \frac{V_5}{V_4} = 5$$

$$\rho = \frac{r_c}{r_e} = \frac{V_1}{V_2} \times \frac{V_4}{V_5} = \frac{V_5}{V_3} \times \frac{V_4}{V_5} = \frac{V_4}{V_3}; \quad (\text{Note that } V_1 = V_5 \text{ and } V_2 = V_3)$$

$$C_p = 1.0 \frac{\text{kJ}}{\text{kg.K}}, \quad C_v = 0.71 \frac{\text{kJ}}{\text{kg.K}}$$

(a) Calculating T_i and P_i , with $i \in \{1, 2, 3, 4, 5\}$ at all strokes:

1–2: Isentropic Compression

$$P_1 V_1^n = P_2 V_2^n \implies P_2 = P_1 \left(\frac{V_1}{V_2} \right)^n = 1 \times 9^{1.25} = 15.59 \text{ bar (1 Mark)}$$

$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2} \right)^{n-1} = 9^{1.25-1} = 1.732 \implies T_2 = 525.07 \text{ K (1 Mark)}$$

3–4: Addition of Heat at Constant Pressure Calculating T_3 and T_4 :

$$\frac{T_4}{T_3} = \frac{V_4}{V_3} = \rho = \frac{r_c}{r_e} = \frac{9}{5} = 1.8 \implies T_4 = 1.8 T_3$$

The problem states that heat liberated at constant pressure is twice the heat liberated at constant volume, i.e.,

$$C_p (T_4 - T_3) = 2 C_v (T_3 - T_2).$$

Substituting T_2 and T_4 ,

$$1.0 (1.8 T_3 - T_3) = 2 \times 0.71 (T_3 - 525.07) \implies T_3 = 1202.58 \text{ K (1 Mark)}$$

$$\frac{P_3}{T_3} = \frac{P_2}{T_2} \implies P_3 = \frac{15.59}{525.07} 1202.58 = 35.70 \text{ bar (1 Mark)}$$

$$P_4 = P_3 = 35.70 \text{ bar (1 Mark)} \quad \text{and} \quad T_4 = 1.8 T_3 = 2164.65 \text{ K (1 Mark)}$$

4–5: Isentropic Expansion

$$P_4 V_4^n = P_5 V_5^n \implies P_5 = P_4 \left(\frac{V_4}{V_5} \right)^n = P_4 \frac{1}{r_e^n} = 35.70 \frac{1}{5^{1.25}} = 4.775 \text{ bar (1 Mark)}$$

$$\frac{T_5}{T_4} = \left(\frac{V_4}{V_5} \right)^{n-1} \implies T_5 = T_4 \frac{1}{r_e^{n-1}} = 1447.59 \text{ K (1 Mark)}$$

Stroke	P (bar)	T (K)
1	1.0	303.15
2	15.59	525.07
3	35.70	1202.58
4	35.70	2164.65
5	4.78	1447.59

(b) Sketch of the Pv diagram in Fig. b (2 Marks).

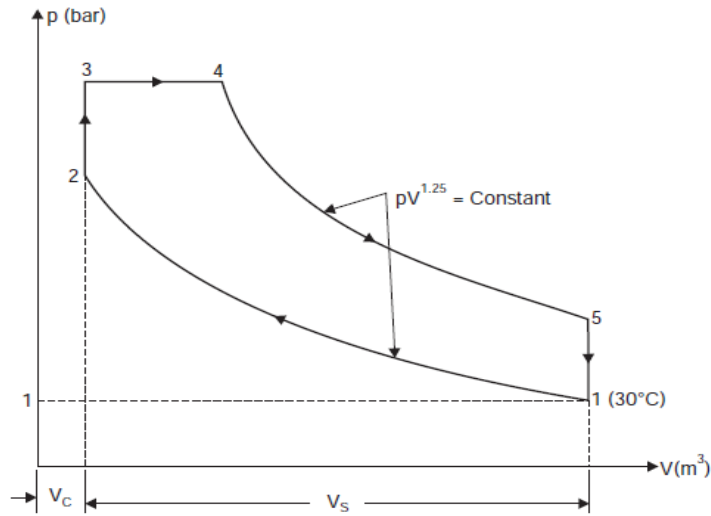


Figure 1: Pv diagram for Question 1.

(c) Calculating MEP:

$$\begin{aligned}
 MEP &= \frac{1}{V_s} \left[P_3 (V_4 - V_3) + \frac{P_4 V_4 - P_5 V_5}{n - 1} - \frac{P_2 V_2 - P_1 V_1}{n - 1} \right] \\
 &= \frac{1}{r_c - 1} \left[P_3 (\rho - 1) + \frac{P_4 \rho - P_5 r_c}{n - 1} - \frac{P_2 - P_1 r_c}{n - 1} \right] \\
 &= \frac{1}{9 - 1} \left[35.70 (1.8 - 1) + \frac{35.70 \times 1.8 - 4.78 \times 9}{1.25 - 1} - \frac{15.59 - 1 \times 9}{1.25 - 1} \right] \\
 &= 10.895 \text{ bar (2 Marks)}
 \end{aligned}$$

(d) The work done per cycle is defined as

$$W_{\text{cycle}} = MEP \times V_s$$

where

$$V_s = \frac{\pi}{4} D^2 L = \frac{3.1415}{4} (0.25)^2 \times 0.4 = 0.0196 \text{ m}^3$$

Thus

$$W_{\text{cycle}} = 10.895 \text{ bar} \times 0.0196 \text{ m}^3 = 0.213542 \text{ bar.m}^3 = 21.354 \text{ kJ (1 Mark)}$$

The heat supplied to the cycle is given by

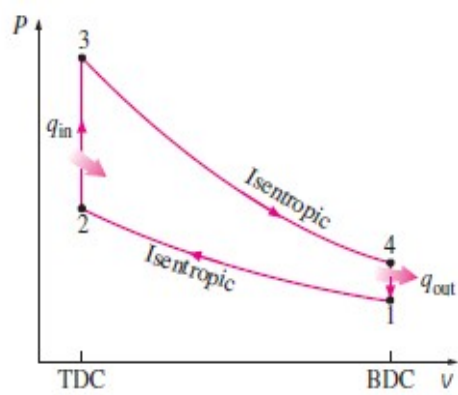
$$Q_{\text{cycle}} = mQ_s = m [C_p (T_3 - T_2) + C_p (T_4 - T_3)]$$

we need to calculate the mass (m) of air in the cylinder:

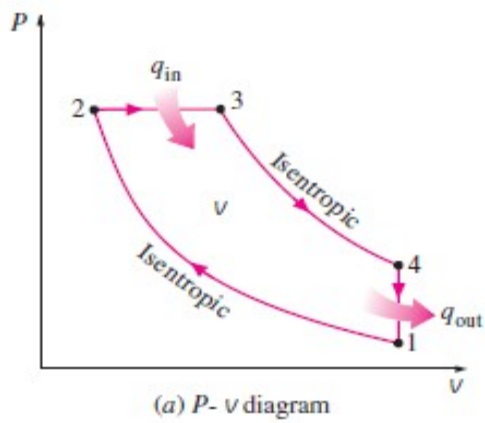
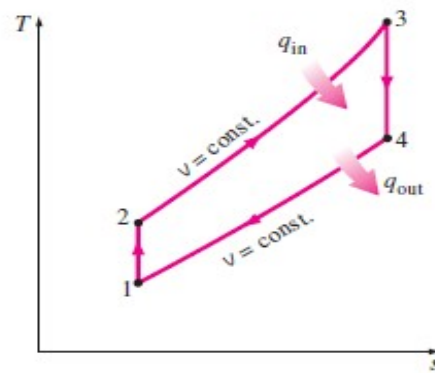
$$V_1 = V_s + V_c = \frac{r_c}{r_c - 1} V_s = \frac{9}{9 - 1} \times 0.0196 = 0.02205 \text{ m}^3$$
$$m = \frac{P_1 V_1 MW}{RT_1} = \frac{1 \text{ bar} \times 0.02205 \text{ m}^3 \times 29 \frac{\text{g}}{\text{gmol}}}{8.3144621 \times 10^{-5} \frac{\text{m}^3 \cdot \text{bar}}{\text{K} \cdot \text{gmol}} \times 303.15 \text{ K}} = 25.3697 \text{ g}$$

$$\begin{aligned} Q_{\text{cycle}} &= mQ_s \\ &= 0.02537 [0.71 (1202.58 - 525, 07) + 1.0 (2164.65 - 1202.58)] \\ &= 36.67 \text{ kJ (1 Mark)} \end{aligned}$$

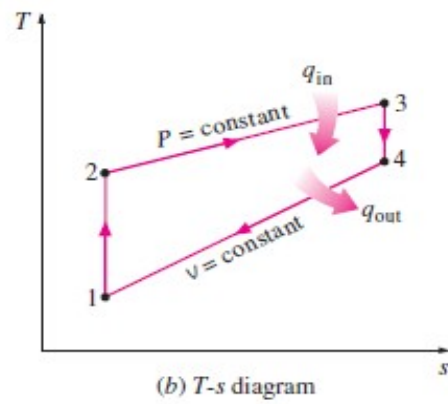
(e) PV and TS diagrams for Otto and Diesel cycles (1.5 Marks for each diagram).



Otto Cycle



Diesel Cycle



Question 2:

(a)

Amount of petroleum material
thermally equivalent to the biodiesel

$$\begin{aligned} &= \left(75000 \times 103 \times \frac{37}{43} \right) \text{ kg releasing on burning} \\ &= \left[\left(75000 \times 103 \times \frac{37}{43} \right) \times \frac{44}{14} \right] \times 10^{-3} \text{ tonnes of CO}_2 \\ &= \text{0.2 million tonnes of carbon dioxide. (8 Marks)} \end{aligned}$$

(b) Trans-esterification with methanol (2 Marks).

(c) It can be transported long distances by tanker, reducing the need for pipelines (2 Marks).

(d)

Rate of electricity production

$$\begin{aligned} &= 1.2 \times 10^9 \text{ kg} \times 55 \times 10^6 \text{ J.kg}^{-1} \times \frac{0.35}{3600 \times 365 \times 24} \text{ W} \\ &= \text{730MW (8 Marks)} \end{aligned}$$

Question 3:

(a) Symbols:

- \dot{Q} - the rate of heat added to the fluid;
- \dot{W}_s - the rate of shaft work done by the fluid;
- \dot{m} - the mass flux through the device;
- c_p - the specific heat capacity at constant pressure;
- T - the fluid temperature;
- u - the fluid velocity;
- χ_2 - the property χ evaluated at the device outlet;
- χ_1 - the property χ evaluated at the device inlet;

[3 Marks]

Assumptions:

- Steady fluid flow through the turbine;
- The fluid is an ideal gas;
- The velocity, pressure, internal energy and potential energy over each inlet or outlet can be replaced by their respective averages;
- No work due to viscosity, chemical, radiological etc;

[Half mark for each up to 2 Marks]

(b) For a steady flow device with two inlets (labelled 1 and 2) and one outlet (labelled 3), the steady flow energy equation can be written as

$$\dot{Q} - \dot{W}_s = \dot{m}_3 \left(c_p T_3 + \frac{u_3^2}{2} + g z_3 \right) - \dot{m}_1 \left(c_p T_1 + \frac{u_1^2}{2} + g z_1 \right) - \dot{m}_2 \left(c_p T_2 + \frac{u_2^2}{2} + g z_2 \right).$$

[2 Marks]

The corresponding steady mass conservation equation is

$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3.$$

[1 Marks]

(c) (i) the volume flux is related to the velocity through

$$\text{Fluid velocity, } u = \frac{\text{Volume flux}}{\text{Cross-sectional area, } A}.$$

[1 Mark of 2]

At inlet 1,

$$\begin{aligned} u_1 &= \frac{1.8 \text{ m}^3/\text{s}}{0.03 \text{ m}^2} = 60 \text{ m/s}, \\ u_2 &= \frac{0.5 \text{ m}^3/\text{s}}{0.1 \text{ m}^2} = 5 \text{ m/s}, \\ u_3 &= \frac{20 \text{ m}^3/\text{s}}{0.5 \text{ m}^2} = 40 \text{ m/s}, \end{aligned}$$

[1 Mark of 2]

(ii) The fluid density at inlet 1 and outlet 3 can be calculated directly

$$\rho_1 = \frac{p_1}{RT_1} = \frac{200000 \text{ Pa}}{300 \text{ J/(kg K)} (80 + 273.15) \text{ K}} = 1.8878 \text{ kg/m}^3,$$
$$\rho_3 = \frac{p_3}{RT_3} = \frac{110000 \text{ Pa}}{300 \text{ J/(kg K)} (30 + 273.15) \text{ K}} = 1.2095 \text{ kg/m}^3.$$

[1 Mark of 6]

The mass flux at inlet 1 and outlet 3 are given by

$$\dot{m}_1 = \rho_1 u_1 A_1 = 1.8878 \text{ kg/m}^3 \times 60 \text{ m/s} \times 0.03 \text{ m}^2 = 22.6533 \text{ kg/s},$$
$$\dot{m}_3 = \rho_3 u_3 A_3 = 1.2095 \text{ kg/m}^3 \times 40 \text{ m/s} \times 0.5 \text{ m}^2 = 24.1904 \text{ kg/s}.$$

[1 Mark of 6]

Now the mass flux through the inlet satisfies

$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3,$$

[1 Mark of 6]

i.e.

$$\dot{m}_2 = \dot{m}_3 - \dot{m}_1 = 24.1904 \text{ kg/s} - 22.6533 \text{ kg/s} = 1.5372 \text{ kg/s}.$$

[1 Mark of 6]

Consequently the density

$$\rho_2 = \frac{\dot{m}_2}{u_2 A_2} = \frac{1.5372 \text{ kg/s}}{5 \text{ m/s} \times 0.1 \text{ m}^2} = 3.0744 \text{ kg/m}^3,$$

[1 Mark of 6]

and the pressure

$$p_2 = \rho_2 RT_2 = 3.0744 \text{ kg/m}^3 \times 300 \text{ J/(kg K)} \times (70 + 273.15) \text{ K} = 316490 \text{ Pa} = 316 \text{ kPa}.$$

[1 Mark of 6]

(iii) Now we can use the SFEE in the form

$$\dot{Q} - \dot{W}_s = \dot{m}_3 \left(c_p T_3 + \frac{u_3^2}{2} + g z_3 \right) - \dot{m}_1 \left(c_p T_1 + \frac{u_1^2}{2} + g z_1 \right) - \dot{m}_2 \left(c_p T_2 + \frac{u_2^2}{2} + g z_2 \right).$$

to calculate the rate at which heat is added to the gas.

$$\begin{aligned} \dot{Q} &= -80000 + \dot{m}_3 \left((800 \times 303.15) + \frac{40^2}{2} + (9.81 \times 0.5) \right) \\ &\quad - \dot{m}_1 \left((800 \times 353.15) + \frac{60^2}{2} + (9.81 \times 0.2) \right) \\ &\quad - \dot{m}_2 \left((800 \times 343.15) + \frac{5^2}{2} + (9.81 \times 1.2) \right) \\ &= -80000 + 5886138 - 6440820 - 422024 \\ &= -1056707.0 \text{ W} = -1057 \text{ kW}. \end{aligned}$$

[2 Marks]

The rate at which heat added to the gas is negative indicating the gas in the device heats the surroundings.

[1 Mark]

Question 4: (a) Calculating enthalpies:

- (i) *State 1:* At $P_1 = 1.4$ bar, the $T_{\text{sat}} = -18.80^\circ\text{C} \ll T_1 = -10^\circ\text{C}$. Thus the refrigerant fluid is a *superheated vapour* and the thermodynamic table for *R-134a* gives $H_1 = 243.40$ kJ/kg (2 Marks) and $S_1 = 0.9606$ kJ/(kg.K).
- (ii) *State 2:* Isentropic compression from $P_1 = 1.4$ bar to $P_2 = 7$ bar $\implies S_{2s} = S_1$. The fluid remains at superheated vapour state,

T ($^\circ\text{C}$)	H (kJ/kg)	S (kJ/(kg.K))
40	275.93	0.9539
50	286.35	0.9867

From linear interpolation, we can find $H_{2s} = 278.06$ kJ/kg. Now, using the compressor efficiency,

$$\eta = \frac{H_{2s} - H_1}{H_2 - H_1} = 0.67 \implies H_2 = 295.13 \text{ kJ/kg (2 Marks)}.$$

With the updated H_2 , we should return to the superheated vapour table at $P_2 = 7$ bar,

T ($^\circ\text{C}$)	H (kJ/kg)	S (kJ/(kg.K))
50	286.35	0.9867
60	286.35	1.0182

through linear interpolation, $S_2 = 1.0135$ kJ/(kg.K) and $T_2 = 58.49^\circ\text{C}$.

- (iii) *State 3:* For $P_3 = 7$ bar and $T_3 = 24^\circ\text{C}$, the fluid is at subcooled state (at this pressure, the saturation temperature is $26.72^\circ\text{C} > T_3$) and the thermodynamic table gives $H_3 = H_f (T = 297.15 \text{ K}) = 82.90$ kJ/kg (2 Marks) and $S_3 = 0.3113$ kJ/(kg.K)
- (iv) *State 4:* Throttling (isenthalpic) process at $P_4 = 1.4$ bar $\implies H_4 = H_3 = 82.90$ kJ/kg (2 Marks). At this pressure, from the thermodynamic table,

$$\begin{aligned} H_f &= 25.77 \text{ kJ/kg} & S_f &= 0.1055 \text{ kJ/(kg.K)} \\ H_g &= 236.04 \text{ kJ/kg} & S_g &= 0.9322 \text{ kJ/(kg.K)} \end{aligned}$$

$H_4 \gg H_f \implies$ two-phase fluid, thus in order to calculate S_4 , we first need to calculate the quality of the fluid

$$x_4 = \frac{H_4 - H_f}{H_g - H_f} = \frac{82.90 - 25.77}{236.04 - 25.77} = 0.2717$$

Now,

$$x_4 = \frac{S_4 - S_f}{S_g - S_f} = 0.2717 \implies S_4 = 0.3301 \text{ kJ/(kg.K)}$$

- (b) The coefficient of performance, β , is,

$$\beta = \frac{H_1 - H_4}{H_2 - H_1} = \frac{243.40 - 82.90}{295.13 - 243.40} = 3.10 \text{ (3 Marks)}$$

(c) The refrigeration capacity, \dot{Q}_{in} , is,

$$\begin{aligned}\dot{Q}_{in} &= \dot{m}(H_1 - H_4) \\ &= 6 \frac{\text{kg}}{\text{min}} \times (243.40 - 82.90) \frac{\text{kJ}}{\text{kg}} \times \frac{60 \text{ min}}{1 \text{ h}} \times \frac{1 \text{ ton}}{1.4 \times 10^4 \text{ kJ/h}} = 4.127 \text{ tons} \quad (3 \text{ Marks})\end{aligned}$$

(d) The schematic and TS diagrams are given bellow (3 Marks each).

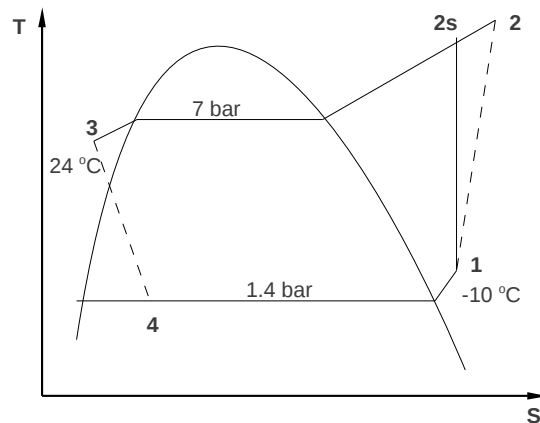
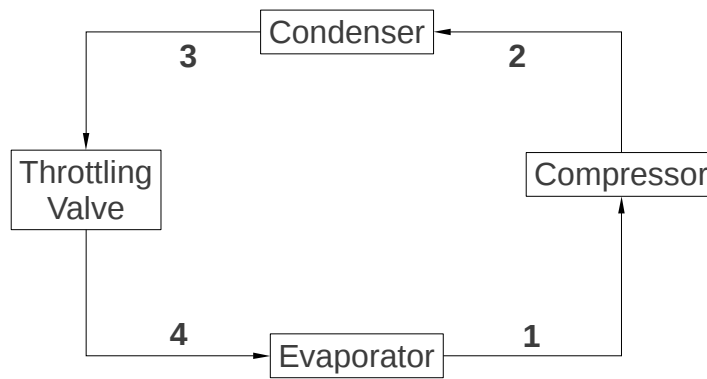


Figure 2: Schematic and TS diagram for Question 4.

Question 5: The specific humidity ω is the ratio of the mass of water vapour m_v , to the mass of dry air m_a and satisfies the equation

$$\omega = \frac{m_v}{m_a}.$$

As both water vapour and dry air behave like ideal gases, in some arbitrary volume V ,

$$\omega = \frac{m_v}{m_a} = \frac{\rho_v}{\rho_a} = \frac{p_v}{R_v T} \frac{R_a T}{p_a} = \frac{R_a p_v}{R_v p_a}.$$

The partial pressures of dry air and water vapour satisfy $p_a = p - p_v$. Hence

$$\omega = \frac{R_a p_v}{R_v (p - p_v)}.$$

The saturation pressure of water $p_{v,\text{sat}}$ is the maximum partial pressure of water vapour a gas can contain at a give temperature before water starts condensing out of the gas. The relative humidity φ is the ratio of the partial pressure of water vapour to the saturation pressure of water

$$\varphi = \frac{p_v}{p_{v,\text{sat}}}.$$

Hence eliminating p_v from the previous expression gives

$$\omega = \frac{R_a \varphi p_{v,\text{sat}}}{R_v (p - \varphi p_{v,\text{sat}})}.$$

[5 Marks]

If the inlet from inside is labelled 1, the inlet from outside is labelled 2 and the outlet with the mixture is labelled 3, then

$$\begin{aligned} \dot{m}_{a1} + \dot{m}_{a2} &= \dot{m}_{a3}, & \text{Mass conservation of dry air,} \\ \omega_1 \dot{m}_{a1} + \omega_2 \dot{m}_{a2} &= \omega_3 \dot{m}_{a3}, & \text{Mass conservation of water vapour,} \\ h_1 \dot{m}_{a1} + h_2 \dot{m}_{a2} &= h_3 \dot{m}_{a3}, & \text{Energy conservation,} \end{aligned}$$

where \dot{m}_a is a mass flux of air, h is the enthalpy and ω is the specific humidity. **[3 Marks of 7]**

Eliminating \dot{m}_3 from the water vapour conservation equation gives

$$\omega_1 \dot{m}_{a1} + \omega_2 \dot{m}_{a2} = \omega_3 (\dot{m}_{a1} + \dot{m}_{a2}) = \omega_3 \dot{m}_{a1} + \omega_3 \dot{m}_{a2}.$$

Collecting together terms involving \dot{m}_1 and terms involving \dot{m}_2 gives

$$\begin{aligned} \omega_1 \dot{m}_{a1} - \omega_3 \dot{m}_{a1} &= \omega_3 \dot{m}_{a2} - \omega_2 \dot{m}_{a2} \\ \Rightarrow (\omega_1 - \omega_3) \dot{m}_{a1} &= (\omega_3 - \omega_2) \dot{m}_{a2}, \end{aligned}$$

and hence rearranging gives

$$\frac{\dot{m}_{a2}}{\dot{m}_{a1}} = \frac{\omega_1 - \omega_3}{\omega_3 - \omega_2} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3}.$$

[2 Marks of 7]

Similarly eliminating \dot{m}_3 from the energy conservation equation gives

$$h_1\dot{m}_{a1} + h_2\dot{m}_{a2} = h_3(\dot{m}_{a1} + \dot{m}_{a2}) = h_3\dot{m}_{a1} + h_3\dot{m}_{a2}.$$

Collecting together terms involving \dot{m}_1 and terms involving \dot{m}_2 gives

$$\begin{aligned} h_1\dot{m}_{a1} - h_3\dot{m}_{a1} &= h_3\dot{m}_{a2} - h_2\dot{m}_{a2} \\ \Rightarrow (h_1 - h_3)\dot{m}_{a1} &= (h_3 - h_2)\dot{m}_{a2}, \end{aligned}$$

and hence rearranging gives

$$\frac{\dot{m}_{a2}}{\dot{m}_{a1}} = \frac{h_1 - h_3}{h_3 - h_2} = \frac{h_3 - h_1}{h_2 - h_3} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3},$$

as required.

[2 Marks of 7]

(a) The partial pressure of dry air at 1 is given by

$$p_{a1} = p_1 - p_{v,\text{sat}_1} = 100000 \text{ Pa} - 1818.747 \text{ Pa} = 98181.253 \text{ Pa}.$$

The specific volume of dry air

$$V_1 = \frac{R_a T_1}{p_{a1}} = \frac{287.1 \text{ kJ/(kg K)} (16 + 273.15)}{98181.253 \text{ Pa}} = \text{m}^3/\text{kg}$$

[1 Mark of 3]

The mass flux of dry air through inlet 1 is given by

$$\dot{m}_{a1} = \frac{\text{volume flux}}{V_1} = \frac{1 \text{ m}^3/\text{s}}{0.8455 \text{ m}^3/\text{kg}} = 1.1827 \text{ kg/s}.$$

[1 Mark of 3]

The mass flux of dry air through inlet 2 is given by

$$\dot{m}_{a2} = \dot{m}_{a3} - \dot{m}_{a1} = 1.8 \text{ kg/s} - 1.1827 \text{ kg/s} = 0.6173 \text{ kg/s}$$

[1 Mark of 3]

(b) The gas entering the system through inlet 1 is saturated ($\varphi_1 = 1$), and therefore the specific humidity

$$\begin{aligned} \omega_1 &= \frac{R_a \varphi_1 p_{v,\text{sat}_1}}{R_v (p_1 - \varphi_1 p_{v,\text{sat}_1})} = \frac{287.1 \text{ kJ/(kg K)} \times 1 \times 1818.747 \text{ Pa}}{461.5 \text{ kJ/(kg K)} (100000 \text{ Pa} - 1 \times 1818.747 \text{ Pa})} \\ &= 0.0115 \text{ kg H}_2\text{O/ kg dry air}. \end{aligned}$$

[1 Mark]

(c) Using

$$\frac{\dot{m}_{a2}}{\dot{m}_{a1}} = \frac{\omega_3 - \omega_1}{\omega_2 - \omega_3}.$$

Rearranging

$$(\omega_2 - \omega_3) \frac{\dot{m}_{a2}}{\dot{m}_{a1}} = (\omega_3 - \omega_1).$$

Collecting terms involving ω_3 gives

$$\omega_2 \frac{\dot{m}_{a2}}{\dot{m}_{a1}} + \omega_1 = \omega_3 + \omega_3 \frac{\dot{m}_{a2}}{\dot{m}_{a1}} = \omega_3 \left(1 + \frac{\dot{m}_{a2}}{\dot{m}_{a1}} \right).$$

Therefore

$$\omega_3 = \frac{\omega_1 + \omega_2 \frac{\dot{m}_{a2}}{\dot{m}_{a1}}}{1 + \frac{\dot{m}_{a2}}{\dot{m}_{a1}}}$$

[2 Marks of 4]

The mass flux ratio

$$\frac{\dot{m}_{a2}}{\dot{m}_{a1}} = \frac{0.6173}{1.1827} = 0.5219$$

[1 Mark of 4]

Therefore

$$\omega_3 = \frac{0.0115 + (0.5219 \times 0.0182)}{1 + 0.5219} = 0.0138 \text{ kg H}_2\text{O/ kg dry air}.$$

[1 Mark of 4]