



**POLITECNICO**  
**MILANO 1863**

# IAC Project

Indycar Autonomous Challenge

**Alessio Russo 945781**  
**Carmelo Valore 944851**

Accademic year 19/20

*Industrial and Information Engineering*  
*Computer Science and Engineering*  
*September 10, 2020*

# Contents

<b>1</b>	<b>Dynamic model</b>	<b>2</b>
1.1	Simple dynamic model . . . . .	2
1.2	Pacejka tyre model . . . . .	4
1.3	Aerodynamic force . . . . .	5
1.4	Fuel consumption . . . . .	5
1.5	Tyre wear . . . . .	6
1.6	Banking . . . . .	6
1.7	Friction ellipse and wear . . . . .	7
1.8	Slipstream . . . . .	9
1.9	Simulations . . . . .	9
1.9.1	Simulation 1 - Non-accelerating straight path . . . . .	9
1.9.2	Simulation 2 - Accelerating straight path . . . . .	10
1.9.3	Simulation 3 - Accelerating straight path on a banked road . . . . .	11
1.9.4	Simulation 4 - Accelerating-Breaking straight path . . . . .	13
1.9.5	Simulation 5 - Constant steer and increasing velocity path . . . . .	15
1.9.6	Simulation 6 - Sinusoidal wave path . . . . .	18
1.9.7	Simulation 7 - Constant speed and increasing steering angle . . . . .	19
1.10	Appendix A . . . . .	22
<b>2</b>	<b>Control</b>	<b>24</b>
2.1	Linearized model . . . . .	24
2.2	Longitudinal control . . . . .	28
2.3	Lateral control . . . . .	31
2.3.1	Simulation 8 - Lap of the track . . . . .	36

## Dynamic model

In this chapter all the steps towards the realization of our model are detailed. The final model is a dynamic bicycle model that takes into consideration the non-linear tyre model using the Pacejka magic formula, the fuel consumption, the aerodynamic forces and slipstream effect, the tyre wears and finally also the banking of the road.

### 1.1 Simple dynamic model

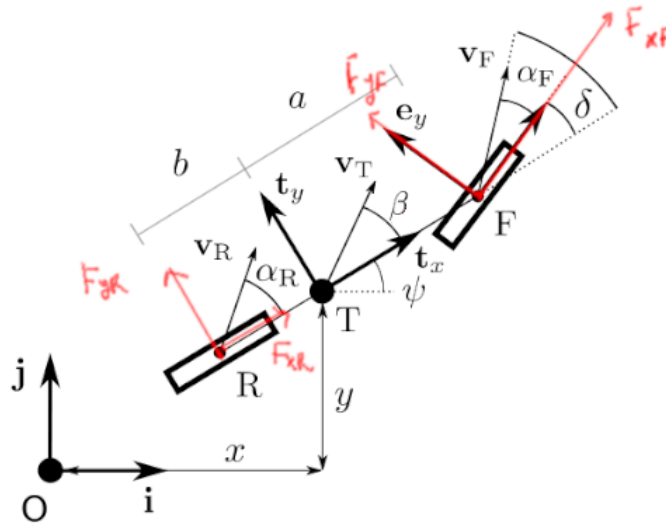


Figure 1.1: Bicycle model

$$\begin{aligned} m_T \ddot{x} &= F_{xF} \cos(\psi + \delta) + F_{xR} \cos(\psi) - F_{yF} \sin(\psi + \delta) - F_{yR} \sin(\psi) \\ m_T \ddot{y} &= F_{xF} \sin(\psi + \delta) + F_{xR} \sin(\psi) + F_{yF} \cos(\psi + \delta) + F_{yR} \cos(\psi) \\ I_T \ddot{\psi} &= F_{xF} a \sin(\delta) + F_{yF} a \cos(\delta) - F_{yR} b \end{aligned} \quad (1.1)$$

$$\begin{aligned}
\alpha_F &= \arctan\left(\frac{\dot{y} + a\dot{\psi}\cos(\psi)}{\dot{x} - a\dot{\psi}\sin(\psi)}\right) - (\delta + \psi) \\
\alpha_R &= \arctan\left(\frac{\dot{y} - b\dot{\psi}\cos(\psi)}{\dot{x} + b\dot{\psi}\sin(\psi)}\right) - \psi
\end{aligned} \tag{1.2}$$

At first, as state vector this has been used:

$$\begin{aligned}
z1 &= x \\
z2 &= y \\
z3 &= \psi \\
z4 &= \dot{x} \\
z5 &= \dot{y} \\
z6 &= \dot{\psi}
\end{aligned} \tag{1.3}$$

So that

$$\begin{aligned}
\dot{z}_1 &= z_4 \\
\dot{z}_2 &= z_5 \\
\dot{z}_3 &= z_6 \\
\dot{z}_4 &= \frac{F_{xF}\cos(z_3 + \delta) + F_{xR}\cos(z_3) - F_{yF}\sin(z_3 + \delta) - F_{yR}\sin(z_3)}{m_T} \\
\dot{z}_5 &= \frac{F_{xF}\sin(z_3 + \delta) + F_{xR}\sin(z_3) + F_{yF}\cos(z_3 + \delta) + F_{yR}\cos(z_3)}{m_T} \\
\dot{z}_6 &= \frac{F_{xF}a\sin(\delta) + F_{yF}a\cos(\delta) - F_{yR}b}{I_T}
\end{aligned} \tag{1.4}$$

With slip angles

$$\begin{aligned}
\alpha_F &= \arctan\left(\frac{z_5 + az_6\cos(z_3)}{z_4 - az_6\sin(z_3)}\right) - (\delta + z_3) \\
\alpha_R &= \arctan\left(\frac{z_5 - bz_6\cos(z_3)}{z_4 + bz_6\sin(z_3)}\right) - z_3
\end{aligned} \tag{1.5}$$

Now instead of using  $\dot{x}$  and  $\dot{y}$ ,  $v_T$  and  $\beta$  have been used. The transformations are the following:

$$\begin{aligned}
\dot{x} &= v_T\cos(\psi + \beta) \\
\dot{y} &= v_T\sin(\psi + \beta) \\
\ddot{x} &= \dot{v}_T\cos(\psi + \beta) - v_T(\dot{\psi} + \dot{\beta})\sin(\psi + \beta) \\
\ddot{y} &= \dot{v}_T\sin(\psi + \beta) + v_T(\dot{\psi} + \dot{\beta})\cos(\psi + \beta)
\end{aligned} \tag{1.6}$$

Substituting and simplyfing with the help of Matlab

$$\begin{aligned}
 v_T &= \frac{F_{xF}\cos(\beta - \delta) + F_{xR}\cos(\beta) + F_{yF}\sin(\beta - \delta) + F_{yR}\sin(\beta)}{m_T} \\
 \dot{\beta} &= \frac{-F_{xF}\sin(\beta - \delta) - F_{xR}\sin(\beta) + F_{yF}\cos(\beta - \delta) + F_{yR}\cos(\beta) - m_T v_T \dot{\psi}}{m_T v_T} \\
 \ddot{\psi} &= \frac{F_{xF}\sin(\delta) + F_{yF}\cos(\delta) - F_{yR}b}{I_T}
 \end{aligned} \tag{1.7}$$

$$\begin{aligned}
 \alpha_F &= \arctan\left(\frac{v_T \sin(\beta) + a \dot{\psi}}{v_T \cos(\beta)}\right) - \delta \\
 \alpha_R &= \arctan\left(\frac{v_T \sin(\beta) - b \dot{\psi}}{v_T \cos(\beta)}\right)
 \end{aligned} \tag{1.8}$$

The new state and the state equations are

$$\begin{aligned}
 x_1 &= x \\
 x_2 &= y \\
 x_3 &= \psi \\
 x_4 &= v_T \\
 x_5 &= \beta \\
 x_6 &= \dot{\psi} \\
 \dot{x}_1 &= x_4 \cos(x_3 + x_5) \\
 \dot{x}_2 &= x_5 \sin(x_3 + x_5) \\
 \dot{x}_3 &= x_6 \\
 \dot{x}_4 &= \frac{F_{xF}\cos(x_5 - \delta) + F_{xR}\cos(x_5) + F_{yF}\sin(x_5 - \delta) + F_{yR}\sin(x_5)}{m_T} \\
 \dot{x}_5 &= \frac{-F_{xF}\sin(x_5 - \delta) - F_{xR}\sin(x_5) + F_{yF}\cos(x_5 - \delta) + F_{yR}\cos(x_5) - m_T x_4 x_6}{m_T x_4} \\
 \dot{x}_6 &= \frac{F_{xF}\sin(\delta) + F_{yF}\cos(\delta) - F_{yR}b}{I_T}
 \end{aligned} \tag{1.9}$$

## 1.2 Pacejka tyre model

The following Pacejka tyre model (Magic Formula '94) has been used, taking as inputs the tyre slip angle  $\alpha$  ( $\alpha_F$  and  $\alpha_R$ ) and the vertical load  $F_z$  on the tyre (respectively  $l_F * F_z$  and  $l_R * F_z$ , where  $l_F$  and  $l_R$  are coefficients to distribute the load between front wheel and rear wheel, such that  $l_F + l_R = 1, l_F \geq 0, l_R \geq 0$ ).

$$F_y = D \sin(C \arctan(B_{x1} - E(B_{x1} - \arctan(B_{x1})))) + V \tag{1.10}$$

With

$$\begin{aligned}
C &= a_0 \\
D &= F_z(a_1 F_z + a_2)(1 - a_{15}\gamma^2) \\
BCD &= a_3 \sin(2 \arctan(\frac{F_z}{a_4}))(1 - a_5|\gamma|) \\
B &= BCD/CD \\
E &= (a_6 F_z + a_7)(1 - (a_{16}\gamma + a_{17})\text{sign}(\alpha + H)) \\
H &= a_8 F_z + a_9 + a_{10}\gamma \\
V &= a_{11} F_z + a_{12} + (a_{13} F_z + a_{14})\gamma F_z \\
B_{x1} &= B(\alpha + H)
\end{aligned} \tag{1.11}$$

Where  $a_i, i \in \{0, \dots, 17\}$ , are the parameters of the Pacejka model, whose value and meaning can be seen in the Appendix A.

### 1.3 Aerodynamic force

The following changes have been done in the previous model to take into account for the aerodynamic force  $F_A = \frac{1}{2}\rho C_x S v^2$  in the same direction of  $v_T$  but in the opposite side, and  $F_{Lift} = \frac{1}{2}\rho C_z S v^2$  that "pushes" the vehicle to be stucked on the ground. In here  $S$  is the area of the vehicle on which the air goes through,  $C_x, C_z$  are drag coefficients,  $\rho$  is the density of the air and  $v$  is the velocity of the vehicle

$$\begin{aligned}
\ddot{x}_4 &= \frac{F_{xF} \cos(x_5 - \delta) + F_{xR} \cos(x_5) + F_{yF} \sin(x_5 - \delta) + F_{yR} \sin(x_5) - \frac{1}{2}\rho C_x S x_4^2}{m_T}
\end{aligned} \tag{1.12}$$

$$F_z = mg + \frac{1}{2}\rho C_z S x_4^2 \tag{1.13}$$

### 1.4 Fuel consumption

The following changes have been done in the previous model to take fuel consumption into account. A simplified version has been used, in which the Power is computed ( $P_e$ ) and multiplied by a coefficient ( $C_{fuel}$ ) that expresses the relation among mass loss (in terms of fuel consumption) and power provided

$$\begin{aligned}
P_e &= (F_{xF} + F_{xR})x_4, F_{xF} \geq 0, F_{xR} \geq 0 \\
\dot{m} &= P_e C_{fuel}
\end{aligned} \tag{1.14}$$

## 1.5 Tyre wear

The **following** have been added in the previous model to take into account for wear of the rubber compound of the tyre. The model is called Archard model, and it makes use of the vertical pression ( $P_z = \frac{F_z}{Area}$ ), the longitudinal sliding velocity of the wheels ( $v_{xF}$  and  $v_{xR}$ ) and some parameters (such as  $K_{wear}$  and  $H$ ). The model output is the wear depth over the time ( $\dot{h}$ ), that will be then converted into  $mm^3$  of wasted material.

The model has been converted in order to take into account, instead of the sliding velocity, the forces on the wheels, thus the parameters have been re-modulated too.

Here the modified Archard model formulation is shown

$$\dot{h}_i = \frac{K_{wear} P_{load} \sqrt{F_{xi}^2 + F_{yi}^2}}{H} \quad (1.15)$$

where  $i \in \{F, R\}$

## 1.6 Banking

Taking into account the shape of the road we introduce other terms in the model equations. As can be seen in figure 1.2 and 1.3 we are able to find the term whose projection will be summed up in the previous dynamic equations, that is the  $mg$  term, that multiplied by  $\sin(\gamma)$  will be directed as the perpendicular to the vehicle direction.

As can be seen from those figures, the contribution of this lateral force acting on the vehicle will add some new terms in the equations of the model. Along the direction of  $v_T$  the contribution of the force is  $mg \sin(\gamma) \sin(\beta)$ , while on the orthogonal direction it is represented by the force  $mg \sin(\gamma) \cos(\beta)$ .

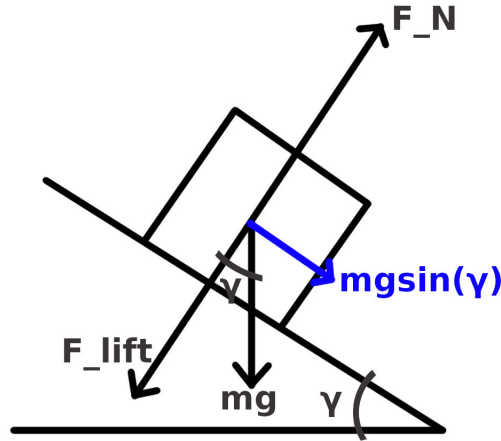


Figure 1.2: Contribution of  $mg$

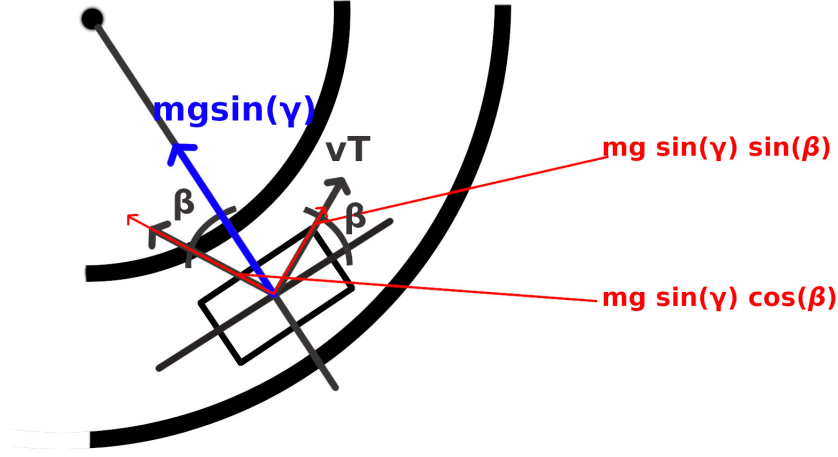


Figure 1.3: Terms affecting the model equations

In the end, the final equations will be:

$$\begin{aligned} \dot{x}_4 &= \frac{F_{xF} \cos(x_5 - \delta) + F_{xR} \cos(x_5) + F_{yF} \sin(x_5 - \delta) + F_{yR} \sin(x_5) - \frac{1}{2} \rho C_x S v^2 + mg \sin(\gamma) \sin(x_5)}{m_T} \\ \dot{x}_5 &= \frac{-F_{xF} \sin(x_5 - \delta) - F_{xR} \sin(x_5) + F_{yF} \cos(x_5 - \delta) + F_{yR} \cos(x_5) - m_T x_4 x_6 + mg \sin(\gamma) \cos(x_5)}{m_T x_4} \end{aligned} \quad (1.16)$$

## 1.7 Friction ellipse and wear

Coming to the conclusion of our model, we took in consideration also the physical relation between longitudinal and lateral forces through the friction ellipse

$$\left( \frac{F_x}{F_{x,max}} \right)^2 + \left( \frac{F_y}{F_{y,max}} \right)^2 = 1 \quad (1.17)$$

In particular,  $F_{x,max}$  and  $F_{y,max}$  are respectively the maximum longitudinal and lateral forces, that are calculated through the Pacejka parameters, being  $D + V$  the point of max of the tyre model. Here we recall that:



$$\begin{aligned}
D_{lat} &= F_z(a_1 F_z + a_2)(1 - a_{15}\gamma^2) \\
V_{lat} &= a_{11}F_z + a_{12} + (a_{13}F_z + a_{14})\gamma F_z \\
D_{long} &= F_z(b_1 F_z + b_2) \\
V_{long} &= b_{11}F_z + b_{12}
\end{aligned} \tag{1.18}$$

Thus as we can see, the maximum longitudinal and lateral forces are functions of the vertical load  $F_z$ .

(Note: different parameters are used for longitudinal and lateral Pacejka, in particular  $a_i$  is referred to the lateral one while  $b_i$  to the longitudinal one).

In this way the ellipse is defined, but, taking in consideration the wear  $h$  the ellipse is scaled. Thus, at the end,  $F_{x,max}$  and  $F_{y,max}$  are functions of  $F_z$  and  $h$ , specifically:

$$\begin{aligned}
F_{x,max} &= (D_{long} + V_{long}) \frac{1}{w_1 h + w_2} \\
F_{y,max} &= (D_{lat} + V_{lat}) \frac{1}{w_1 h + w_2}
\end{aligned} \tag{1.19}$$

Where  $w_1$  and  $w_2$  are parameters opportunely chosen. In this way the more the wear, the more the ellipse is shrunk.

The ellipse is saying us which is the maximum lateral force wrt the given  $F_x$  in input. Thus, finally, the output of the lateral Pacejka is scaled, so to have the peak value  $(D_{lat} + V_{lat})$  equal to the value given by the ellipse. To do so, the D value of the Pacejka is directly fed as input to the model through the output of the ellipse.

Following image tries to clarify the steps detailed so far.

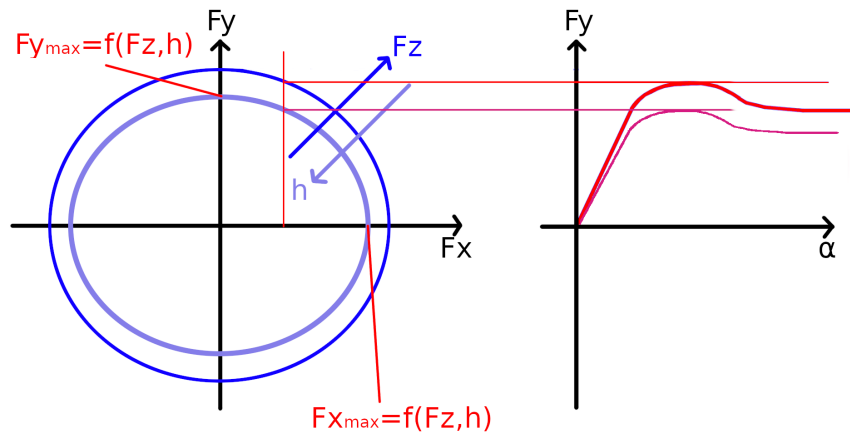


Figure 1.4: Friction ellipse and wear effects

## 1.8 Slipstream

In our model we also considered the slipstream effect. Some thresholds have been defined to establish if the vehicle is undergoing the slipstream condition. These thresholds take into account both front and rear distance of two different vehicles and also their alignment degree: below these values, the slipstream is considered to be effective. This leads to have a lower air density coefficient  $\rho$  in the aerodynamic force formula, that physically represents a lower opponent aerodynamic force. All the used coefficients can be found in the Appendix A.

## 1.9 Simulations

Different simulations have been run to validate qualitatively the presented model and to highlight its most relevant features. Among the others, 5 simulations are considered to be the most representative and are shown in the following. For each simulation only the most relevant graphs have been shown. For a complete overview of the simulations we refer to this link [https://github.com/cvalore/simulations\\_logs](https://github.com/cvalore/simulations_logs) where it is possible to find the logs and all graphs related to these simulations. Note: In all the simulations the input  $\delta$  is reported, that is the wheel-angle although we directly fed the steer-angle. They can be reciprocally easy calculated knowing that the steering ratio  $\frac{steer\_angle}{wheel\_angle} = 10$ .

### 1.9.1 Simulation 1 - Non-accelerating straight path

In this preliminary simulation the aim was to test the correctness of the various figures of merit in a flying start situation with an initial velocity of  $20[m/s]$  and nothing else. Provided inputs are  $F_x = 0[N]$ ,  $\delta = 0[rad]$ , simulation time =  $30[s]$ . Banking is not considered.

Although not very significant, this test guarantees that all the internal forces are working correctly. Indeed, the decreasing of the acceleration and velocity due to the aerodynamic force can be seen in the figure 1.5. All the other figures are not useful, since all angles, lateral forces, power and mass decreasing factor are zero; also the trajectory is not reported since it is simply a straight line.

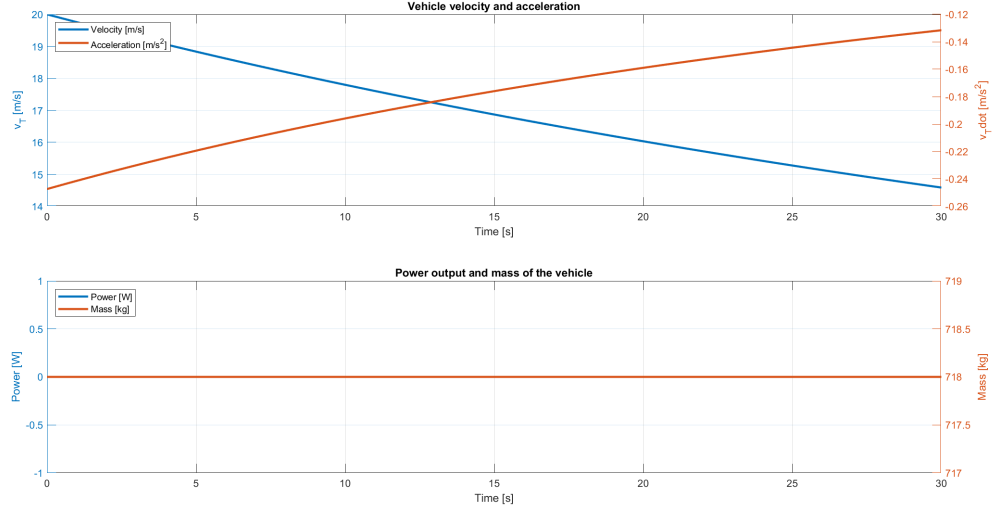


Figure 1.5: Simulation 1 - velocity, acceleration, power and mass figure

### 1.9.2 Simulation 2 - Accelerating straight path

In this preliminary simulation the aim was to test the correctness of the various figures of merit in a situation in which the initial velocity of the vehicle is  $0[m/s]$  and there is a constant acceleration. Provided inputs are  $F_x = 1000[N]$ ,  $\delta = 0[rad]$ , simulation time =  $30[s]$ . Banking is not considered. Also in this case not every simulation result is shown, since angles and trajectory are, as before, respectively zero and a straight line. The velocity and acceleration figure is shown and it is consistent with the given simulation data; the first one is increasing since there is a constant force applied over all the time interval that makes the acceleration always positive, the second one is decreasing since the applied force is not changing over the time and the aerodynamic force is increasing with the square of the velocity. The figures reported, figures 1.6 and 1.7 are showing as the tyres wear is increasing with time and how the power is effecting the mass of the vehicle (actually the mass of the fuel) that is decreasing.

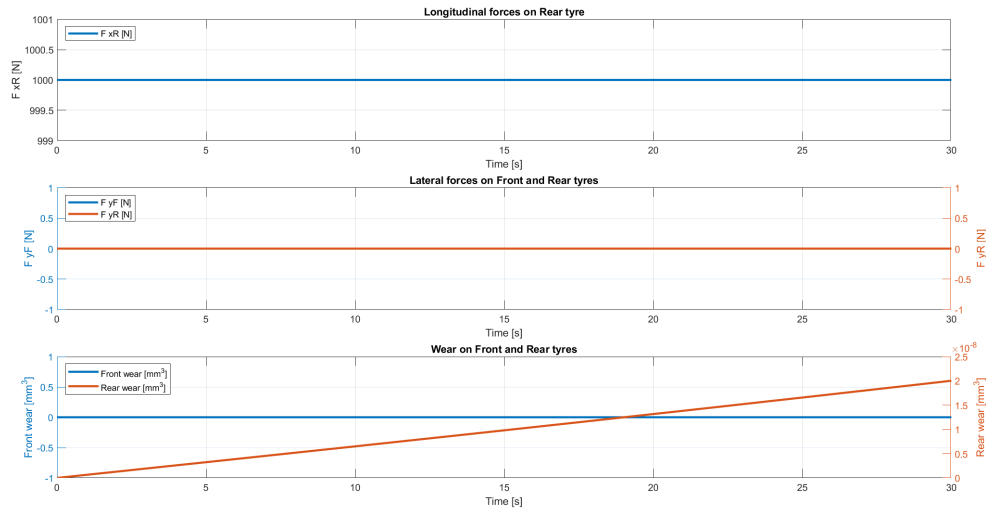


Figure 1.6: Simulation 2 - forces and tyres wear figure

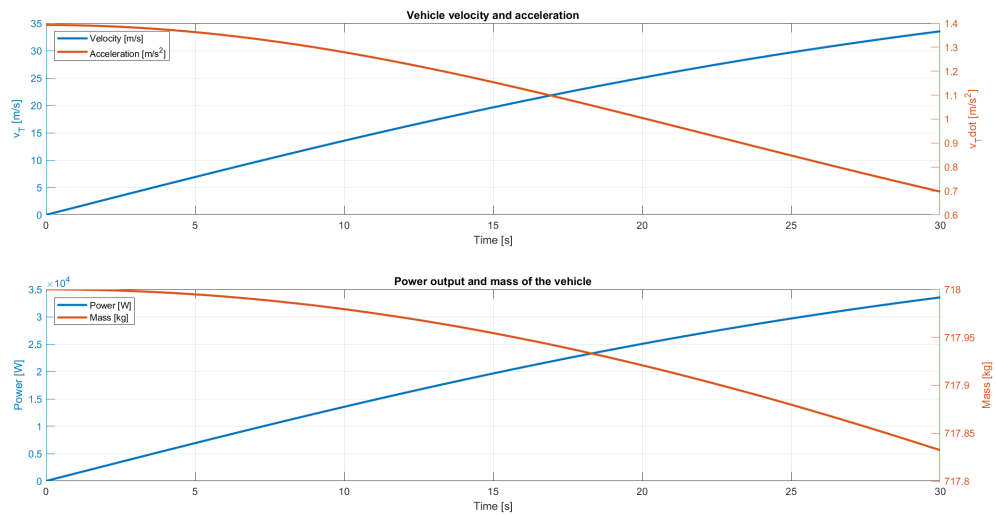


Figure 1.7: Simulation 2 - velocity, acceleration, power and mass figure

### 1.9.3 Simulation 3 - Accelerating straight path on a banked road

In this simulation the aim was to test the correctness of the various figures of merit when the ego vehicle traverses a banked road. Provided inputs are:  $F_x = 500[\text{N}]$ ,  $\delta = 0[\text{rad}]$ , simulation time =  $50[\text{s}]$ .

As can be seen, when the ego vehicle starts traversing a banked road about at time  $40[\text{s}]$ , slip

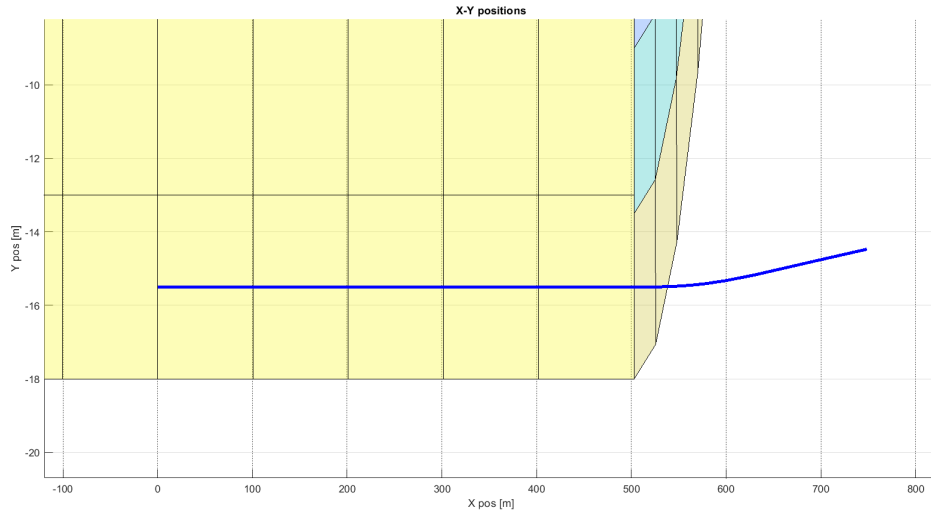


Figure 1.8: Simulation 3 - trajectory of the vehicle on the banked road

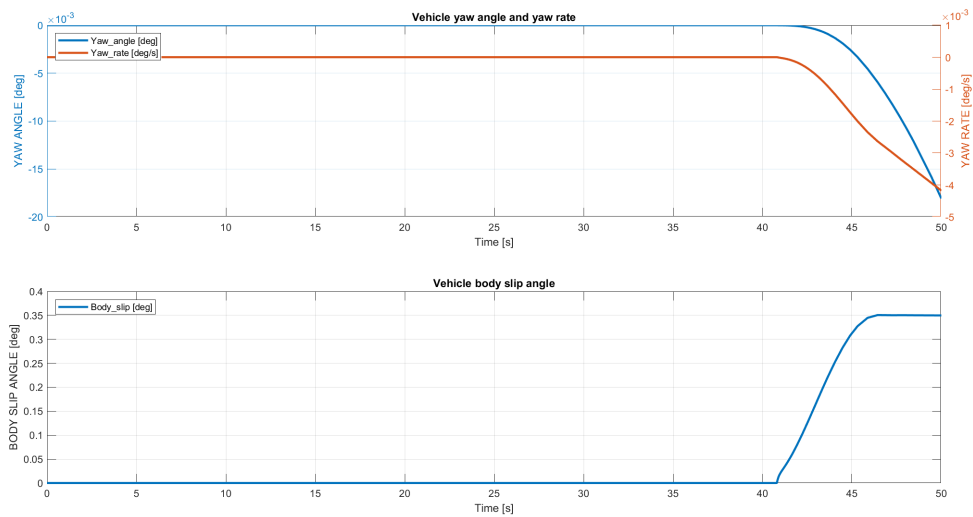


Figure 1.9: Simulation 3 - angles figure

angles, lateral forces and other related figures of merit assume non-zero values. This is clearly consistent with the model dynamics that also takes into account the projection of the vertical load on the inclined road.

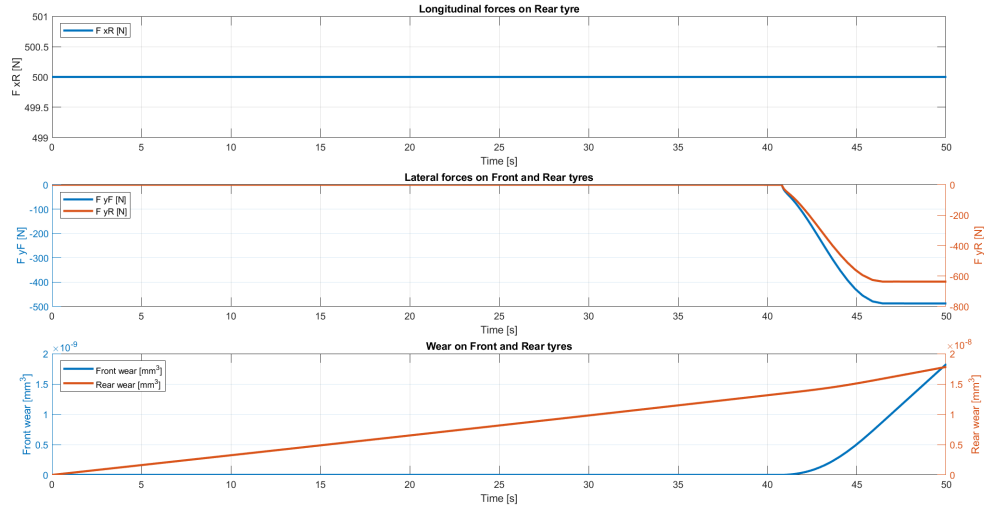


Figure 1.10: Simulation 3 - forces and tyres wear figure

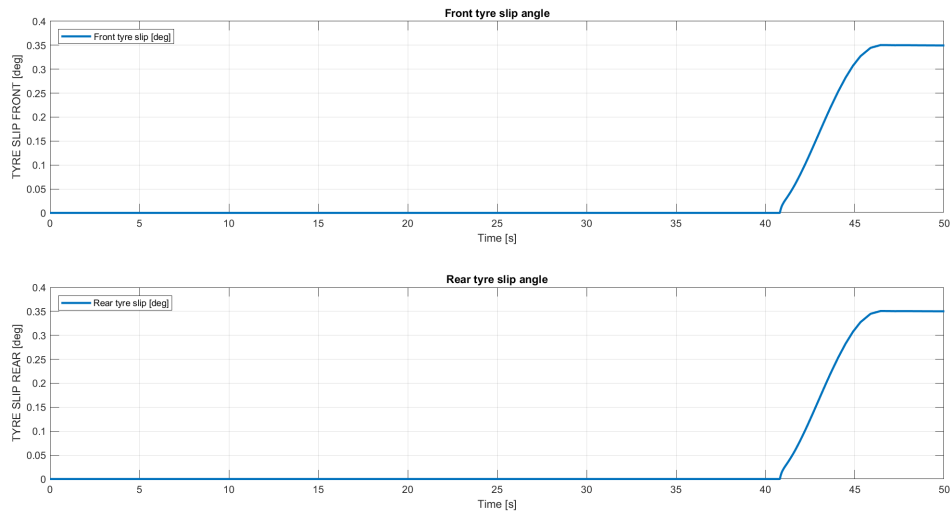


Figure 1.11: Simulation 3 - tyre slips figure

#### 1.9.4 Simulation 4 - Accelerating-Breaking straight path

In this simulation the aim was to show how the considered figures of merit change when the car is moving on a straight line with positive acceleration and then a braking force is applied to reduce its speed. Provided inputs are:  $F_x$  in the form of a step function with  $F_x = 1250[N]$  for  $15[s]$ ,  $F_x = -700[N]$  for the following  $15[s]$ ,  $\delta = 0[rad]$ , simulation time =  $30[s]$ . Banking is not considered.

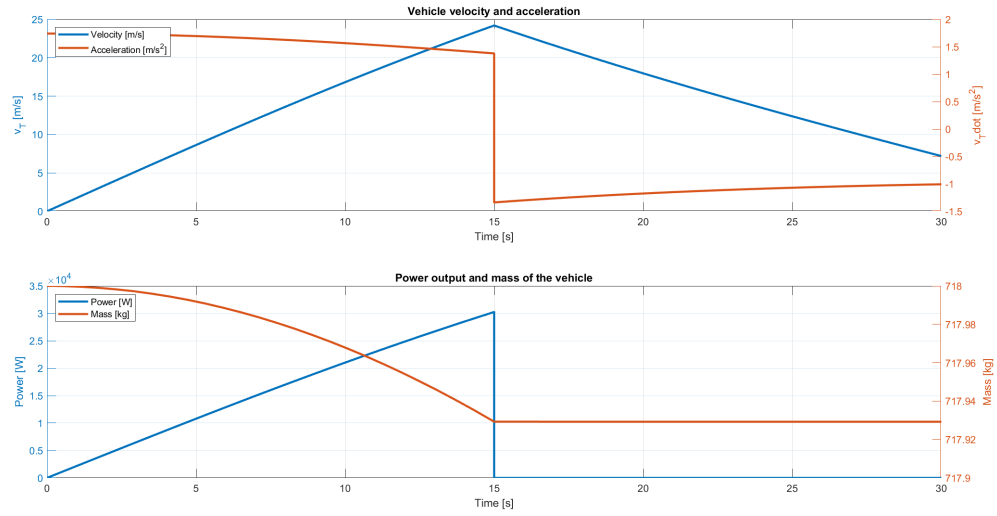


Figure 1.12: Simulation 4 - velocity, acceleration, power and mass figure

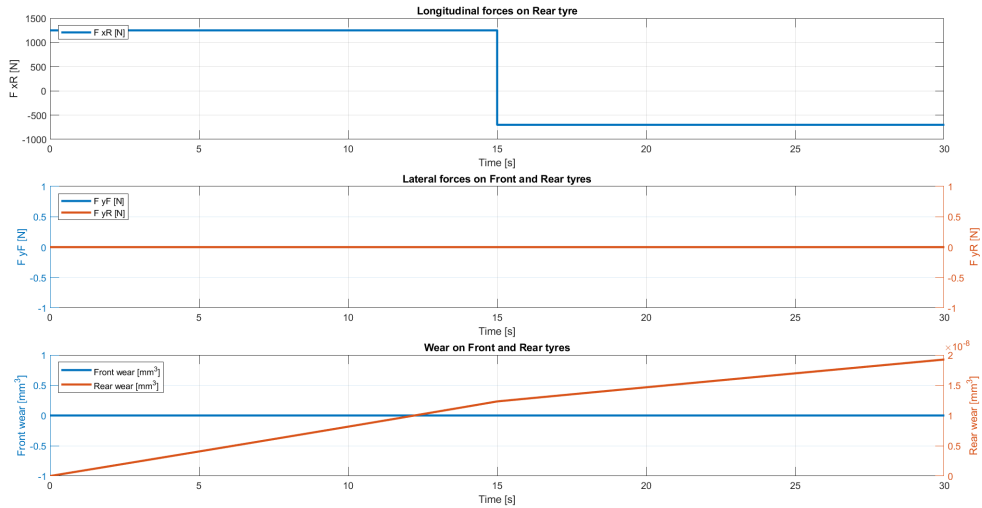


Figure 1.13: Simulation 4 - forces and tyres wear figure

In this simulation the images related to the figures of merit such as trajectory, slip angle and yaw angles are omitted because always keep a zero value. The shown images are consistent with the vehicle dynamics: as we can see the velocity increases as far as  $F_x$  is positive and starts decreasing in the second half of the simulation when  $F_x$  holds a negative value. Another important aspect that can be highlighted is the model of the fuel tank that is coherent with the shown results: when the input force  $F_x$  has negative values the power produced by the vehicle is considered null as well as

the fuel consumption.

### 1.9.5 Simulation 5 - Constant steer and increasing velocity path

The aim of simulation 5 was to test the correctness of the various figures of merit in a "circle trajectory" situation with increasing velocity. The car starts with an initial speed of  $0[m/s]$  and then constantly accelerates and steers to the left. Provided inputs are  $F_x = 1500[N]$ ,  $\delta = 0[rad]$  for the first  $0.5[s]$  and then constantly increases up to about  $0.03[rad]$ , simulation time =  $45[s]$ .

In this case the shown figures represent the trajectory, the angles and the lateral forces. As can be seen, the yaw angle constantly increases, since the vehicle is constantly turning, and up to a certain time point also the yaw rate is increasing before settling down. Also lateral forces have a positive growth since it is a left-turning manoeuvre.

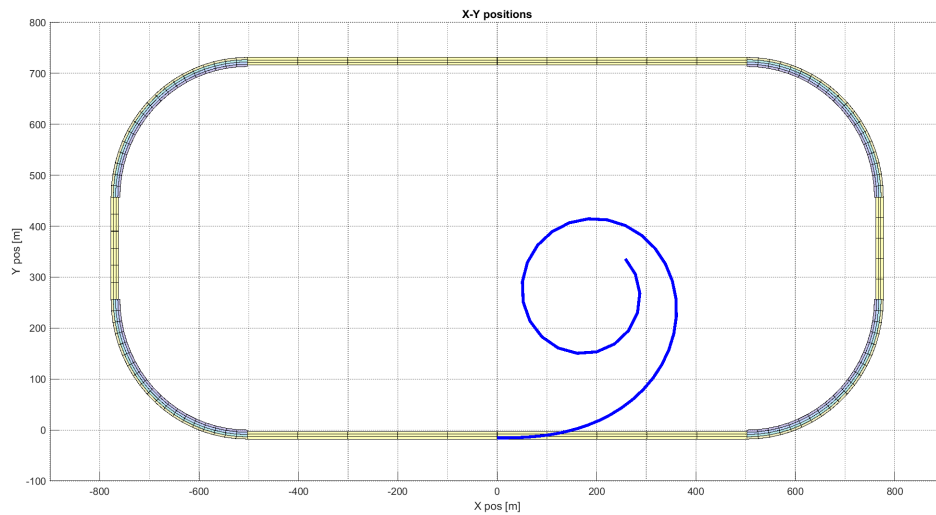


Figure 1.14: Simulation 5 - trajectory of the vehicle



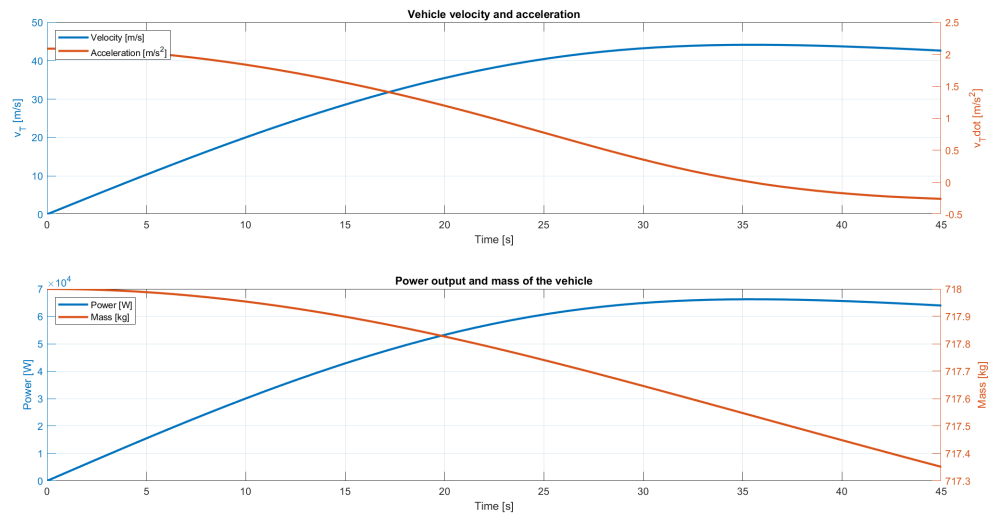


Figure 1.15: Simulation 5 - velocity, acceleration, power and mass figure

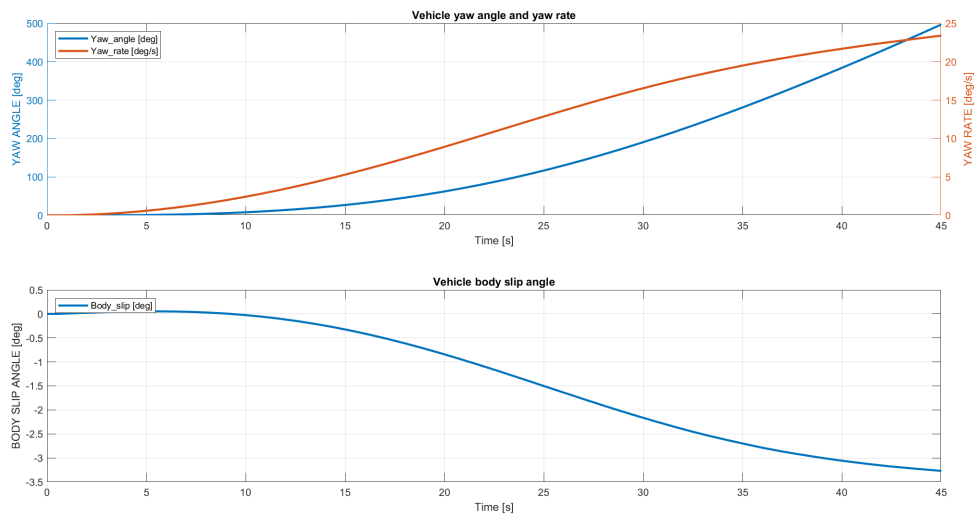


Figure 1.16: Simulation 5 - angles figure

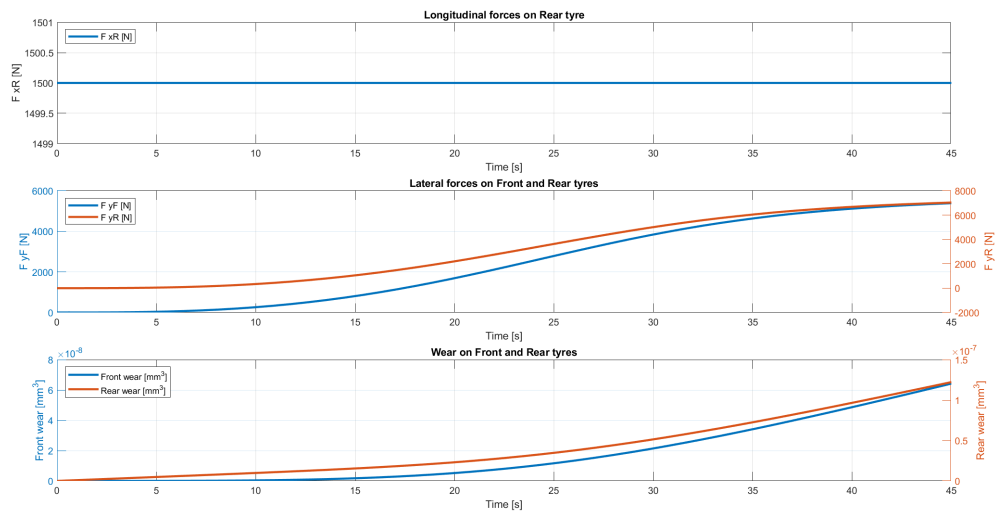


Figure 1.17: Simulation 5 - forces and tyres wear figure

### 1.9.6 Simulation 6 - Sinusoidal wave path

The aim of this simulation was to test the correctness of the various figures of merit in a "sinusoidal wave trajectory" situation. The car starts with an initial speed of  $0[m/s]$  and then constantly accelerates and steers following a sinusoidal wave with amplitude about  $0.02[rad]$  and frequency  $0.22[rad/s]$ . Provided inputs are  $F_x = 500[N]$ ,  $\delta = 0.02\sin(0.22t)$ , simulation time =  $30[s]$ . For the given simulation time and frequency, just a single wave is simulated. The shown figures represent the trajectory, the angles and the lateral forces. Power and mass figures are not shown since consistent with simulation scenario and already shown in previous simulations. All the figures provided exhibit a wave behavior consistent with the simulation data.

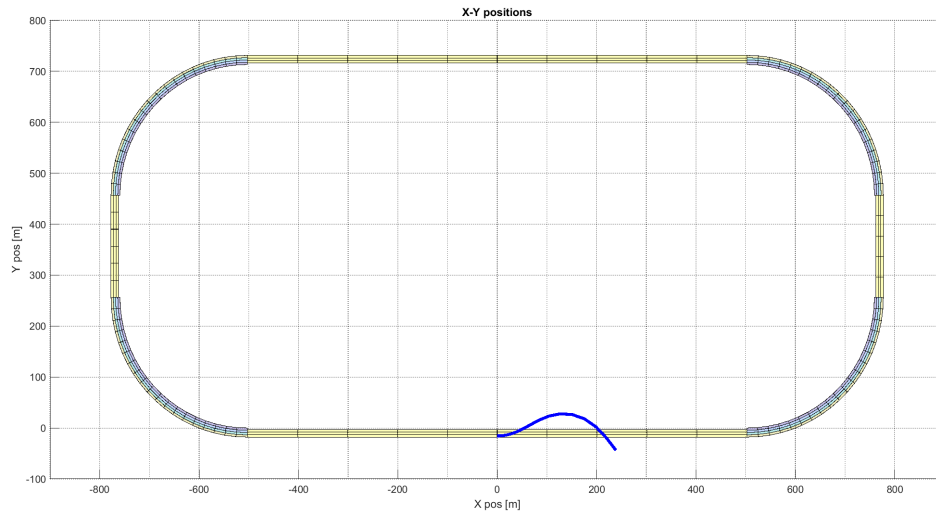


Figure 1.18: Simulation 6 - trajectory of the vehicle

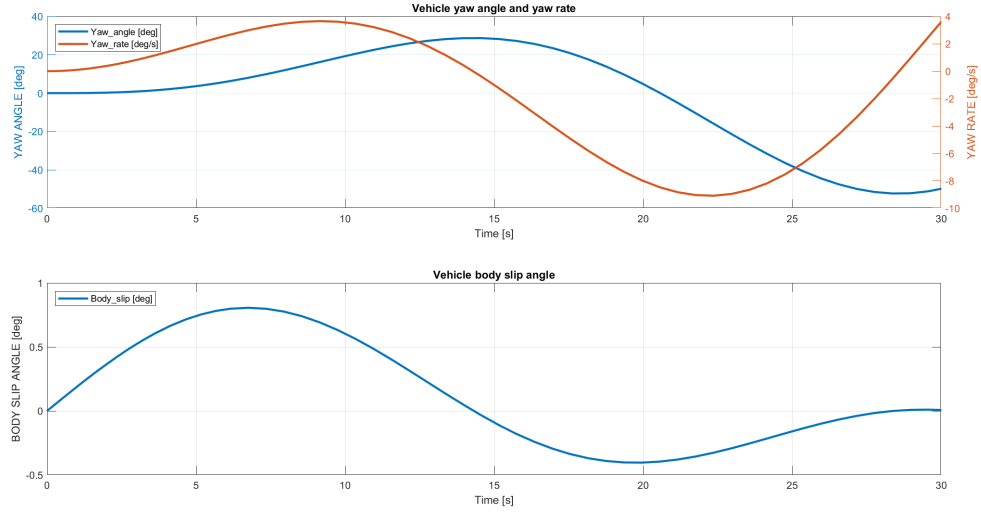


Figure 1.19: Simulation 6 - angles figure

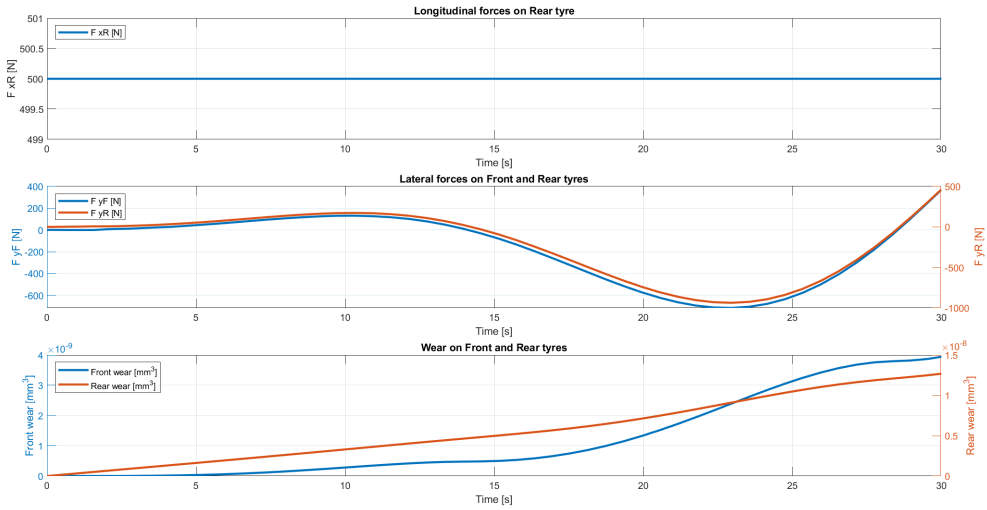


Figure 1.20: Simulation 6 - forces and tyres wear figure

### 1.9.7 Simulation 7 - Constant speed and increasing steering angle

The aim of this simulation was to test the correctness of the various figures of merit in a sort of circular trajectory with a constant speed. The car starts with a speed of  $30[m/s]$  and after few seconds reaches its desired speed of  $50[m/s]$  speed. The applied steering angle is a linear function with slope of  $3[deg/s]$  and the simulation is run for  $100[s]$ : this will result in a final steering angle

on the wheels of  $30[\text{deg/s}]$ . The velocity is kept constant by using a simple PID controller. As it is clear also from the figure representing the trajectory, the presented vehicle shows an under-steering behaviour because it enlarges the trajectory even if the applied steering angle is increasing.

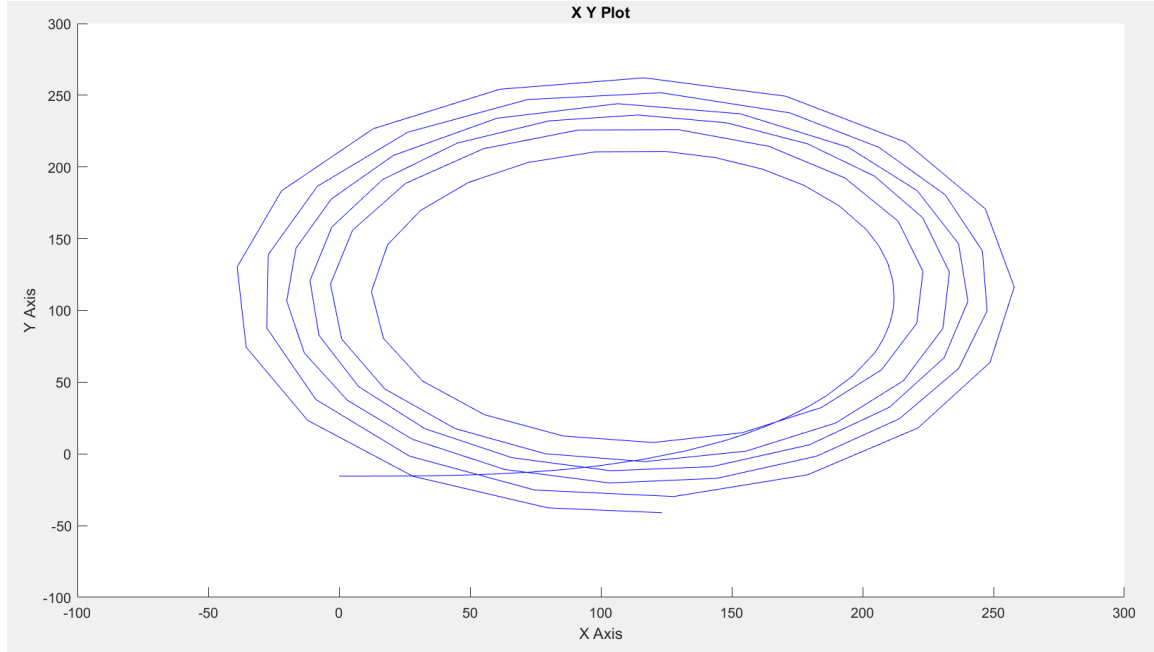


Figure 1.21: Simulation 7 - trajectory of the vehicle

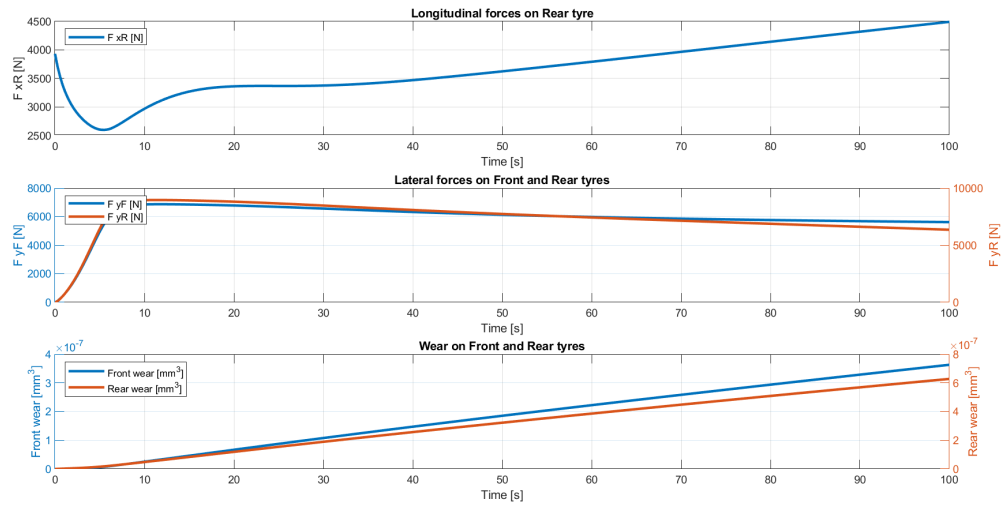


Figure 1.22: Simulation 7 - forces and tyres wear figure

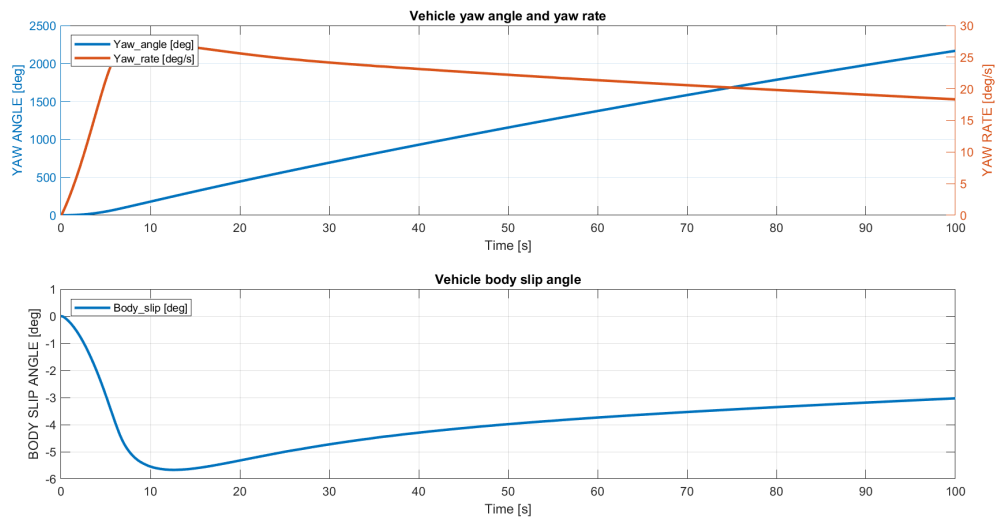


Figure 1.23: Simulation 7 - angles figure

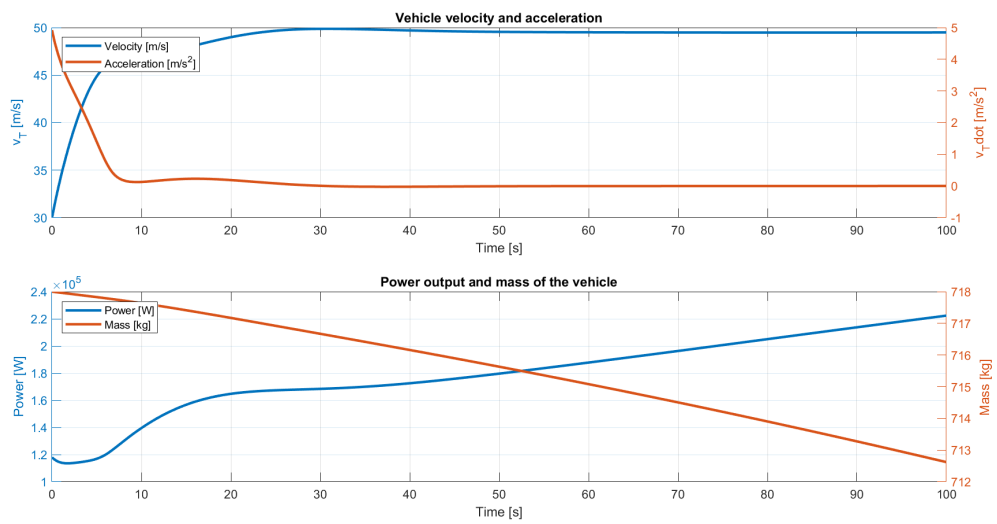


Figure 1.24: Simulation 7 - velocity, acceleration, power and mass figure

## 1.10 Appendix A

In this section we will define the values given to the different parameters of the model.

Parameters of lateral Pacejka tyre model

$$\begin{aligned}
 a0 &= 1.47[-] \text{--- Shape factor} \\
 a1 &= 0[1/kN] \text{--- Load influence on lateral friction coefficient (*1000)} \\
 a2 &= 2050[-] \text{--- Lateral friction coefficient (*1000)} \\
 a3 &= 2500[N/deg] \text{--- Change of stiffness with slip} \\
 a4 &= 10[kN] \text{--- Change of progressivity of stiffness / load} \\
 a5 &= 0[\%/deg/100] \text{--- Camber influence on stiffness} \\
 a6 &= 0[-] \text{--- Curvature change with load} \\
 a7 &= -2[-] \text{--- Curvature factor} \\
 a8 &= 0[deg/kN] \text{--- Load influence on horizontal shift} \\
 a9 &= 0[deg] \text{--- Horizontal shift at load = 0 and camber = 0} \\
 a10 &= 0[-] \text{--- Camber influence on horizontal shift} \\
 a11 &= 0[N] \text{--- Vertical shift} \\
 a12 &= 0[N] \text{--- Vertical shift at load = 0} \\
 a13 &= 0[N/deg/kN] \text{--- Camber influence on vertical shift, load dependent} \\
 a14 &= 0[N/deg] \text{--- Camber influence on vertical shift} \\
 a15 &= 0[1/deg] \text{--- Camber influence on lateral friction coefficient} \\
 a16 &= 0[-] \text{--- Curvature change with camber} \\
 a17 &= 0[-] \text{--- Curvature shift} \\
 \gamma &= 0[rad] \text{--- Camber angle}
 \end{aligned}$$

Parameters of longitudinal tyre model - only the ones used

$$\begin{aligned}
 b1 &= 0[1/kN] \text{--- Load influence on longitudinal friction coefficient (*1000)} \\
 b2 &= 2080[-] \text{--- Longitudinal friction coefficient (*1000)} \\
 b11 &= 0[N] \text{--- Vertical shift} \\
 b12 &= 0[N] \text{--- Vertical shift at load = 0}
 \end{aligned}$$

$m_{vehicle} = 590[kg]$ ——Mass of the vehicle

$m_{fuel} = 58[kg]$ ——Mass of the fuel

$m_{passenger} = 70[kg]$ ——Mass of the passenger

$m_T = m_{vehicle} + m_{fuel} + m_{passenger}$

$g = 9.81[\frac{m}{s^2}]$ ——Gravity acceleration

$l_F = 0.414[-]$ ——Distribution of load on the front wheel

$l_R = 0.586[-]$ ——Distribution of load on the rear wheel

$a = 1.767[m]$ ——Distance between center of vehicle and front wheel

$b = 1.353[m]$ ——Distance between center of vehicle and rear wheel

$L = a + b[m]$ ——Wheelbase

$I_T = 606[kgm^2]$ ——Moment of Inertia of the vehicle

$C_x = 0.725[-]$ ——Drag coefficient

$C_z = 0.778[-]$ ——Lift coefficient

$\rho_{noslipstream} = 1.225[\frac{kg}{m^3}]$ ——Density of air without slipstream effect

$\rho_{slipstream} = 0.8[\frac{kg}{m^3}]$ ——Density of air with slipstream effect

$S = 1[m^2]$ ——Area on which the air goes through

$C_{fuel} = 3 \times 10^{-7}[\frac{s^2}{m^2}]$ ——Fuel consumption coefficient

$K_{wear} = 3.85 * 10^{-17}[\frac{m^3 s^3}{kg^2}]$ ——Tyre wear parameter

$K_{wear\_vel} = 10^{-4.8}[\frac{m^3 s^3}{kg^2}]$ ——Tyre wear parameter

$TyreContactAreaFront = 0.072137[m^2]$ ——Contact area between front tyre and asphalt

$TyreContactAreaRear = 0.082758[m^2]$ ——Contact area between rear tyre and asphalt

$w_1 = 10^{-4.5}[-]$ ——Parameter to scale ellipse due to wear

$w_2 = 1[-]$ ——Parameter to scale ellipse due to wear



## Chapter 2

# Control

In this chapter controllers are realized in order to achieve the possibility for the vehicle to follow a given trajectory according to a certain velocity profile. For this goal, two different controllers are realized, a longitudinal and a lateral one.

Both of them were projected on a linearized version of our model with linear tyre model and then tested on the real simulator.

### 2.1 Linearized model

Here the linearized model with linear tyre model is presented.

First of all in the linear tyre model the lateral forces of the tyres are proportional to the slip angles up to a coefficient called cornering stiffness coefficient ( $C_i$ ). This was estimated by looking at the mean vertical forces on the tyres and looking the slip-lateral force relation of the simulator Pacejka model. As can be seen from figures 2.1 and 2.2, by using a  $C_F = 100000$  [N/rad] and  $C_R = 110000$  [N/rad], linear tyre model well approximate non-lateral one specifically for small slips angles, assumption done to linearize.

Given the model presented so far without considering banking and aereodynamic force:

$$\begin{aligned}
 x1 &= x \\
 x2 &= y \\
 x3 &= \psi \\
 x4 &= v_T \\
 x5 &= \beta \\
 x6 &= \dot{\psi} \\
 \dot{x}_1 &= x_4 \cos(x_3 + x_5) \\
 \dot{x}_2 &= x_5 \sin(x_3 + x_5) \\
 \dot{x}_3 &= x_6 \\
 \dot{x}_4 &= \frac{F_{xF} \cos(x_5 - \delta) + F_{xR} \cos(x_5) + F_{yF} \sin(x_5 - \delta) + F_{yR} \sin(x_5)}{m_T} \\
 \dot{x}_5 &= \frac{-F_{xF} \sin(x_5 - \delta) - F_{xR} \sin(x_5) + F_{yF} \cos(x_5 - \delta) + F_{yR} \cos(x_5) - m_T x_4 x_6}{m_T x_4} \\
 \dot{x}_6 &= \frac{F_{xF} a \sin(\delta) + F_{yF} a \cos(\delta) - F_{yR} b}{I_T} \\
 \alpha_F &= \arctan\left(\frac{x_4 \sin(x_5) + a x_6}{x_4 \cos(x_5)}\right) - \delta \\
 \alpha_R &= \arctan\left(\frac{x_4 \sin(x_5) - b x_6}{x_4 \cos(x_5)}\right)
 \end{aligned} \tag{2.1}$$

We can linearize it choosing an operating point assuming velocity is constant and we are travelling in straight path:

Operating point for states

$$\begin{aligned}
 x1 &= x_{op} \\
 x2 &= y_{op} \\
 x3 &= \psi_{op} = 0 \\
 x4 &= v_{T_{op}} = v_{T,const} \\
 x5 &= \beta_{op} = 0 \\
 x6 &= \dot{\psi}_{op} = 0
 \end{aligned} \tag{2.2}$$

Operating point for inputs - considering also linear lateral forces

$$\begin{aligned}
 \delta_{op} &= 0 \\
 F_{xF_{op}} &= 0 \\
 F_{xR_{op}} &= 0 \\
 F_{yF_{op}} &= 0 \\
 F_{yR_{op}} &= 0
 \end{aligned}$$

Expanding the system in a Taylor series:

$$f_{lin} = f(x_{op}, u_{op}) + \nabla f(x_{op}, u_{op})[x - x_{op} | u - u_{op}]' \tag{2.3}$$

And considering linear tyre model and linear version of the slip angles:

$$\begin{aligned}
 F_{yF} &= -C_F \alpha_F \\
 F_{yR} &= -C_R \alpha_R \\
 \alpha_F &= x_5 + \frac{a}{v_{T,const}} x_6 - \delta \\
 \alpha_R &= x_5 - \frac{b}{v_{T,const}} x_6
 \end{aligned} \tag{2.4}$$

We obtain the linearized model:

$$\begin{aligned}
 x_1 &= x \\
 x_2 &= y \\
 x_3 &= \psi \\
 x_4 &= v_T \\
 x_5 &= \beta \\
 x_6 &= \dot{\psi} \\
 \dot{x}_1 &= x_4 \\
 \dot{x}_2 &= v_{T,const}(x_3 + x_5) \\
 \dot{x}_3 &= x_6 \\
 \dot{x}_4 &= \frac{F_{xF} + F_{xR}}{m_T} \\
 \dot{x}_5 &= -\frac{C_F + C_R}{m_T v_{T,const}} x_5 - \frac{m_T v_{T,const} + \frac{aC_F - bC_R}{v_{T,const}}}{m_T v_{T,const}} x_6 + \frac{C_F}{m_T v_{T,const}} \delta \\
 \dot{x}_6 &= -\frac{aC_F - bC_R}{I_T} x_5 - \frac{a^2 C_F + b^2 C_R}{I_T v_{T,const}} x_6 + \frac{aC_F}{I_T} \delta
 \end{aligned} \tag{2.5}$$

This model was the one used to project the controllers

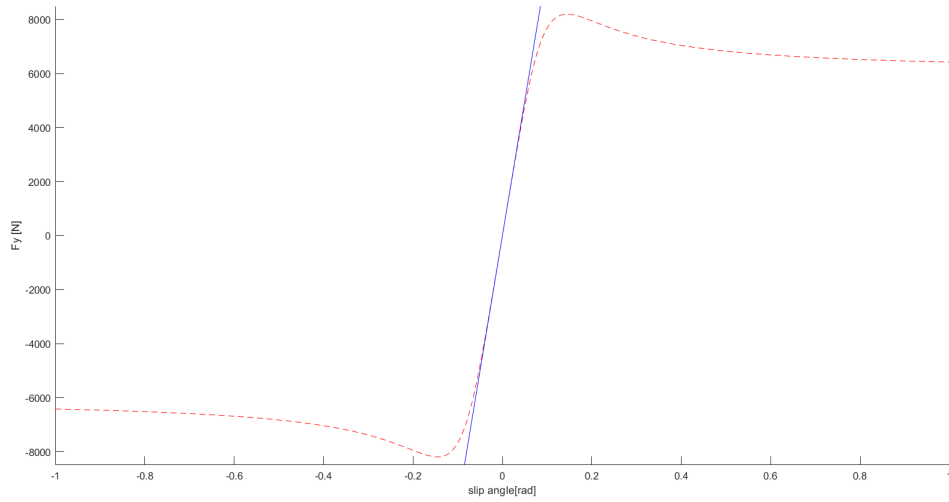
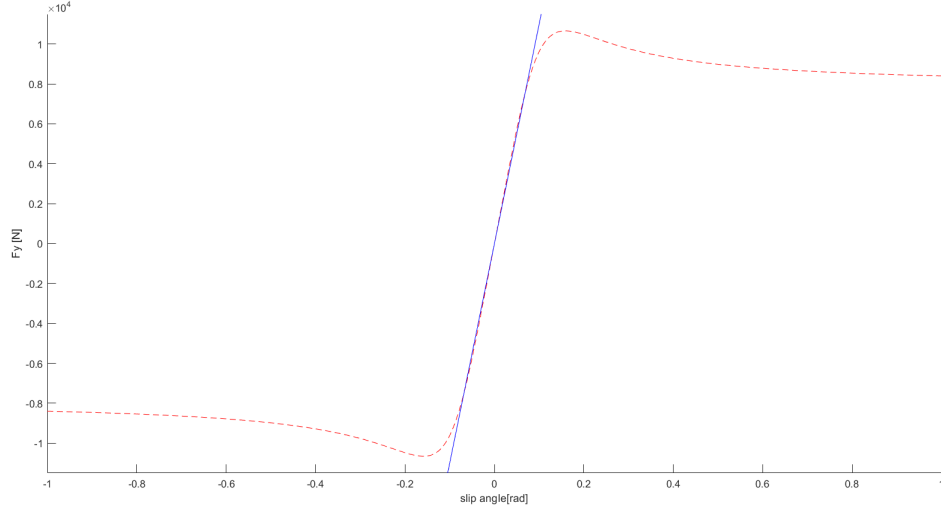


Figure 2.1: Linear vs non-linear modely on front tyre with  $C_F = 100000$

Figure 2.2: Linear vs non-linear modely on rear tyre with  $C_R = 110000$ 

## 2.2 Longitudinal control

The goal of the longitudinal controller is to make the vehicle follow a given speed profile. At first, this profile simply denotes the speed the vehicle must have in each point of the predefined trajectory expressed using the curvilinear abscissa. Given that, using the position of the vehicle, the desired speed is calculated by looking up in the profile table searching for the nearest point with respect to the vehicle position; the actual velocity of the vehicle is subtracted to the wanted one, and this error is fed to a controller that gives to the model the longitudinal force on the rear wheel ( $F_{xR}$ ) in order to minimize this error, and thus obtaining the desired velocity. The block scheme is represented in figure 2.3

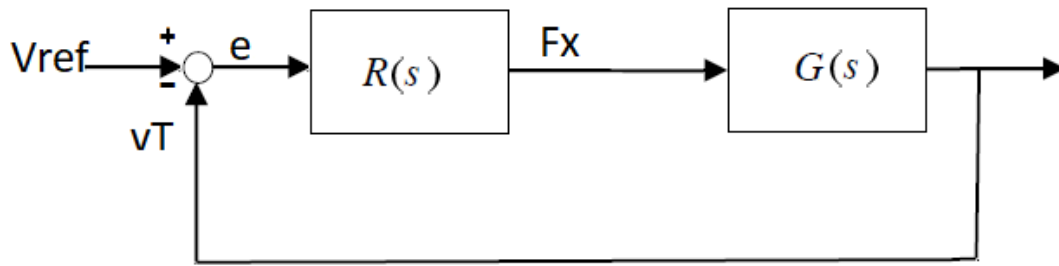


Figure 2.3: Block scheme

From the linear model we calculated the transfer function from  $F_{xR}$  to the velocity of the vehicle, the plant, and it is

$$G(s) = \frac{1}{m_T s}$$

The projected controller is of order two and is of the form

$$R(s) = \frac{k}{s} * \frac{(1 + \frac{s}{2\pi f_z})^2}{1 + \frac{s}{2\pi f_p}}$$

Tuning was done in order to achieve these requirements:

- Reject ramp disturbances of velocity profile: to obtain this an integrator  $\frac{1}{s}$  was inserted. Thus given the transfer function of the plant, two integrators are present and the requirements is satisfied
- Obtain a bandwidth of at least 1 [Hz]
- Closed loop system stable

Tuned parameters are:  $f_z = 0.06$  [Hz],  $f_p = 0.03$  [Hz] and  $k = 5.2 \times 10^3$ . The double zeros were chosen to be coincident for tuning simplicity and were added to obtain a raise of the phase margin thus the system to be stable. The pole at frequency  $f_p$  was added to obtain the desired bandwidth and to have a causal regulator.

From figures 2.4 and 2.5 we can see the bode plot of  $G(s)$  and of the open loop function  $L(s) = R(s)G(s)$

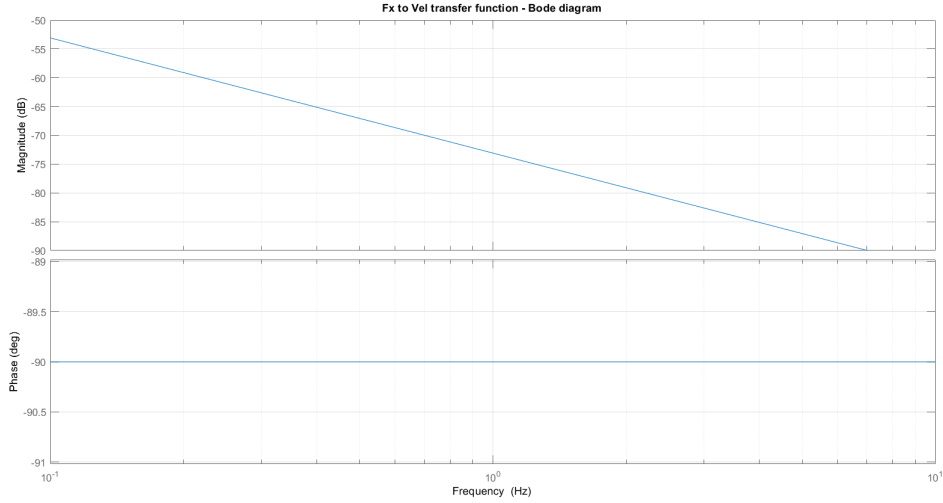
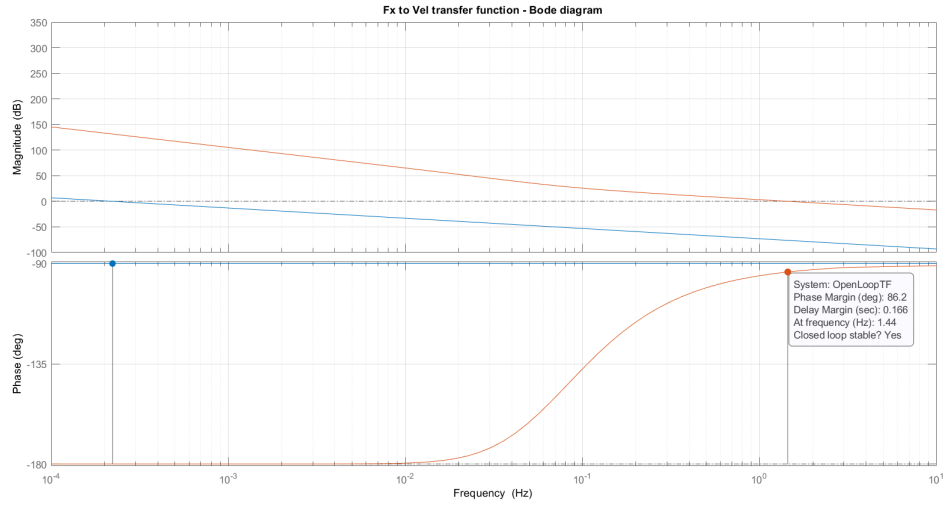


Figure 2.4: Bode diagram of the plant  $G(s)$

Moreover from figures 2.5 we can see that for the bode criterion the closed loop system is stable since the phase margin is  $86$  [deg]  $> 0$ , and that the bandwidth is  $1.44$  [Hz]

Figure 2.5: Bode diagram of the open loop function  $L(s)$  vs the plant  $G(s)$ 

From figures 2.6 and 2.7 the response of the system with the projected regulator to a sinusoidal wave and to the generated speed profile to follow during the trajectory. As can be seen, the reference velocity is well followed.

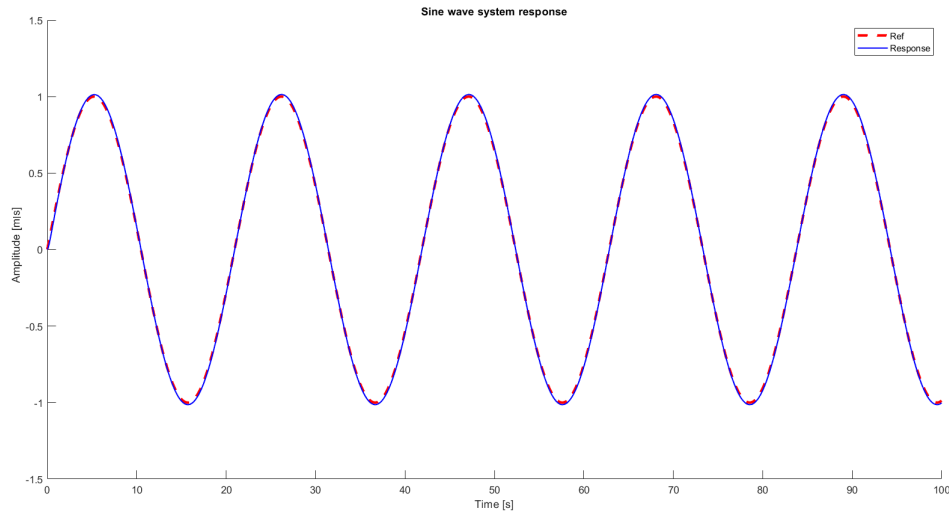


Figure 2.6: System response to sin wave

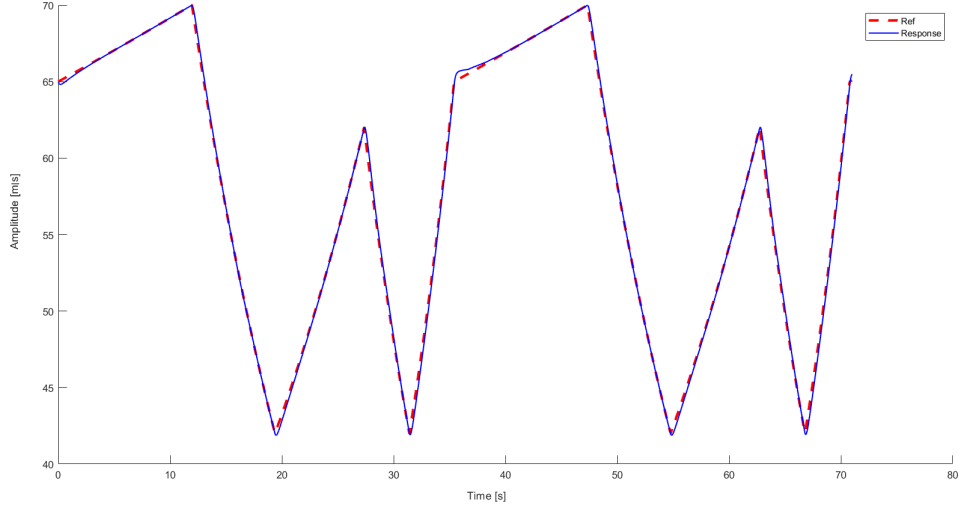


Figure 2.7: System response to speed profile

## 2.3 Lateral control

The goal of the lateral controller is to give to the vehicle the correct steering angle in order to minimize the lateral error with respect to the predefined trajectory. The trajectory is generated, as for the speed profile, as a function of the curvilinear abscissa, meaning that for each value of it, desired  $x$ ,  $y$  and yaw angle are expressed. Anyway, the lateral error is not computed referring to the centre of mass of the vehicle but based on a lookahead distance. Thus, the controller first calculates the coordinates of the virtual point which is an approximation of where the vehicle will be in the next instants. The virtual point is computed using the lookahead distance and the current yaw angle, then it checks the distance between the nearest point on the trajectory and this virtual point and fed this error to the regulator. Concerning the lookahead distance, we choose to set it as variable with the speed up to a constant of 1/3 of seconds.

The project was done according to the already presented linearized model but with another simplification: the longitudinal dynamics was neglected. Moreover a change of state has been done in order to have as resulting states just the lateral velocity  $V_y$  and the yaw rate  $\dot{\psi}$ , longitudinal velocity  $V_x$  is thus considered constant since varying with a much slower dynamic with respect to the lateral one

$$\begin{aligned}\dot{V}_y &= -\frac{C_F + C_R}{m_T V_x} V_y + \left( \frac{C_R l_R - C_F l_F}{m_T V_x} - V_x \right) \dot{\psi} + \frac{C_F}{m_T} \delta \\ \ddot{\psi} &= \frac{C_R l_R - C_F l_F}{I_T V_x} V_y - \frac{C_F l_F^2 - C_R l_R^2}{I_T V_x} \dot{\psi} + \frac{C_F l_F}{I_T} \delta\end{aligned}\quad (2.6)$$

To this model two states were added to take into consideration the dynamic of the error. The first state  $e_{cg}$  represent the lateral positioning error with respect to the center of gravity; the second state  $\Delta\psi$  is the heading angle error, representing the difference between the desired yaw angle and the actual one. The reference path is described by the coordinates  $X_t$ ,  $Y_t$  and the curvature  $\rho_t$ .



$$\begin{aligned} \dot{e}_{cg} &= V_y \cos(\Delta\psi) + V_x \sin(\Delta\psi) \\ \dot{\Delta\psi} &= \dot{\psi} - \rho_t V_t \end{aligned} \quad (2.7)$$

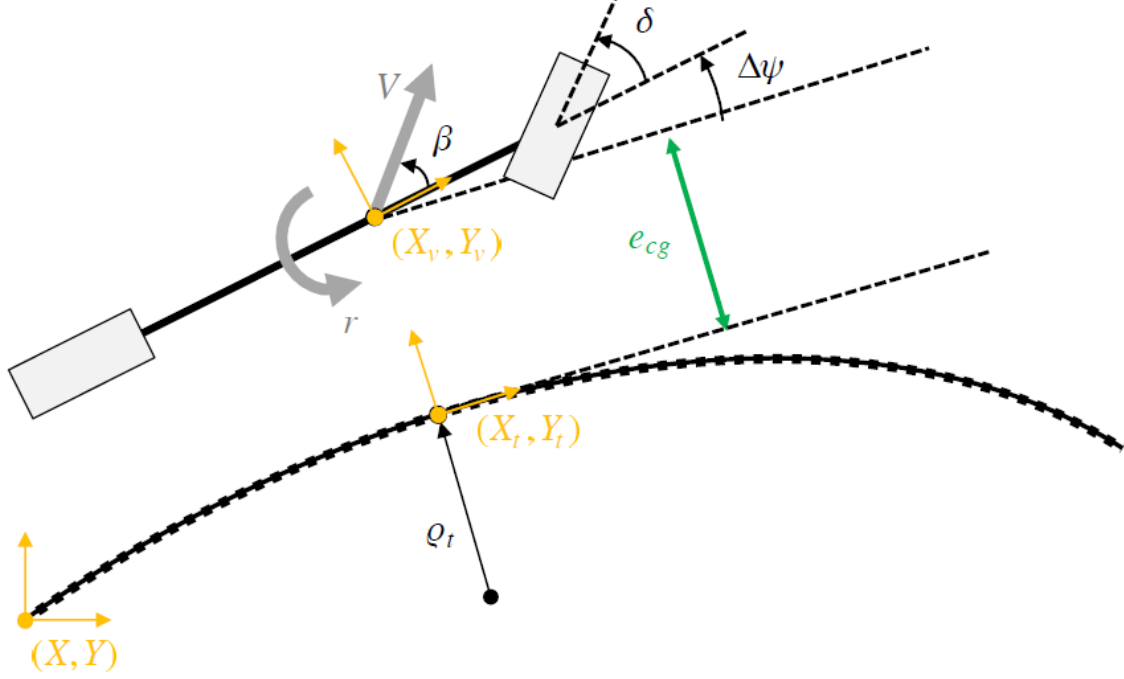


Figure 2.8: Lateral positioning error and heading angle error

Assuming small angles and that the velocity is well tracked we have:

$$\begin{aligned} \dot{e}_{cg} &= V_y + V_x \Delta\psi \\ \dot{\Delta\psi} &= \dot{\psi} - \rho_t V_x \end{aligned} \quad (2.8)$$

These two equations are added to the previous state space model. The output of the system would simply be  $e_{cg}$  but since we decided to use the lookahead error we have as output

$$y = e_{la} = e_{cg} + d_{la} \Delta\psi \quad (2.9)$$

Where  $d_{la}$  is the chosen distance of lookahead, set to  $V_x/k$  where  $k$  is  $3 [s^{-1}]$  so to take the distance in which will be 1/3 of seconds from now as already said.

This is the model used to project the regulator, thus we calculated the transfer function between

the steering angle  $\delta$  and the error  $e_{la}$  and it is

$$G_{e,la}(s) = \frac{C_F V_x}{k s^2} \frac{N(s)}{D(s)}$$

$$N(s) = (l_f m V_x^2 + J_z k V_x) s^2 + (C_r V_x l_r + C_r l_f V_x + C_r k l_r^2 + C_r k l_f l_r) s + (C_r k V_x l_r + C_r k l_f V_x)$$

$$D(s) = (J_z V_x^2 m) s^2 + (C_f J_z V_x + C_r J_z V_x + C_f V_x l_f^2 m + C_r V_x l_r^2 m) s + (C_f C_r l_f^2 + C_f C_r l_r^2 - C_f V_x^2 l_f m + C_r V_x^2 l_r m + 2 C_f C_r l_f l_r)$$
(2.10)

The block scheme of the system is the one in figure 2.9 and as can be seen the curvature can be interpreted as external disturbances, in particular a step disturbance.  $G_\rho$  is the transfer function between the curvature and the lookahead error and it is

$$G_\rho(s) = -\frac{V_x^2 s + V_x^2 k}{k s^2}$$
(2.11)

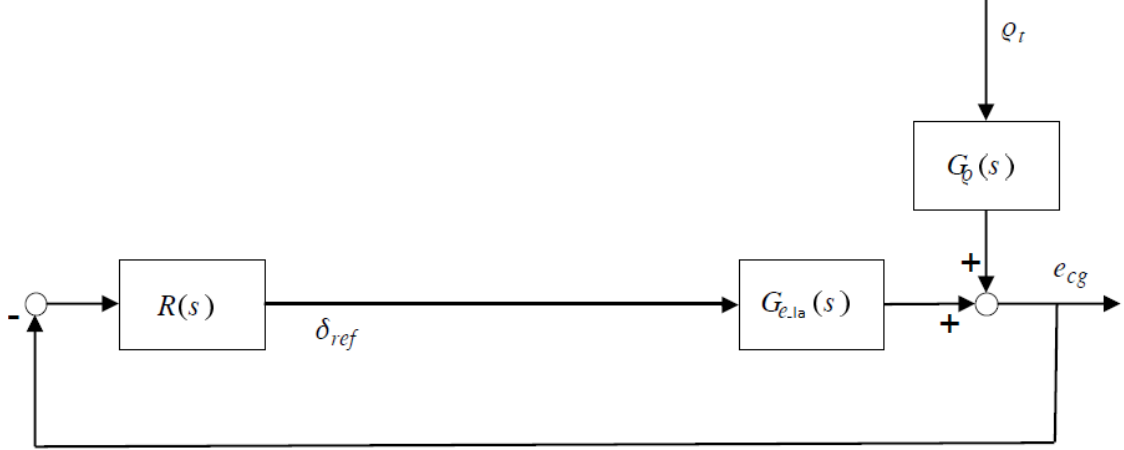


Figure 2.9: Block scheme of lateral project

The controller was projected to satisfy the following requirements

- Reject step disturbances of curvature: to obtain this an integrator  $\frac{1}{s}$  was inserted.
- Obtain a bandwidth of at least 1 [Hz]

- Closed loop system stable

Thus it is of the form

$$R(s) = \frac{k}{s} * \frac{(1 + \frac{s}{2\pi f_z})^2}{1 + \frac{s}{2\pi f_p}}$$

Tuning leads to choose  $f_z = 0.01$  [Hz],  $f_p = 2$  [Hz] and  $k = 2 \times 10^{-4}$ .

Anyways this is not the controller used for the simulator; indeed when testing this one on the non linear model, we have come to the conclusion that another integrator was needed to overwhelm non linearity condition. Thus the pole at 2 Hz was removed and another integrator was inserted. The final regulator structure is:

$$R(s) = \frac{k}{s^2} * (1 + \frac{s}{2\pi f_z})^2$$

Tuning leads to choose  $f_z = 0.01$  [Hz] and  $k = 1.2 \times 10^{-4}$ . Different operating points have been chosen, and as can be seen there's not a lot of change in bandwidth to lead us to use a scheduling coefficient. Figures 2.10 and 2.11 are the bode diagram of the plant  $G(s)$  and of the open loop function  $L(s)$  at different velocities from which can be seen that the closed loop system is stable since the phase margin is  $40$  [deg]  $> 0$  and that satisfies the requirements; moreover the system step response is represented in figure 2.12.

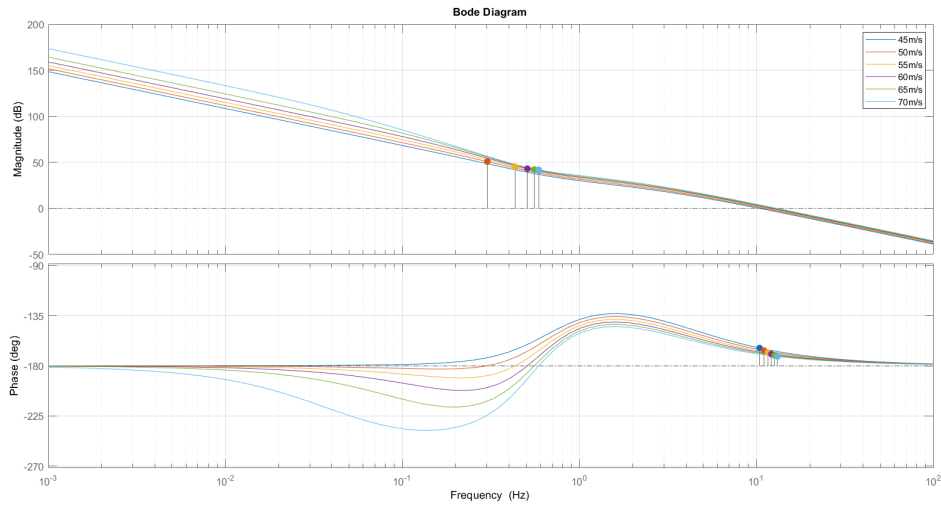


Figure 2.10: Plant  $G(s)$ , transfer function from  $\delta$  to  $e_{la}$  at different velocities

Finally, an open loop component was added, figure 2.13, to directly acts on the steering angle to compensate curvature error since it is known and acts as disturbance. The main idea is that the curvature disturbance is exploited to anticipate the regulator and to directly give the compensating steering command. Designing it simply, we add it as a gain:

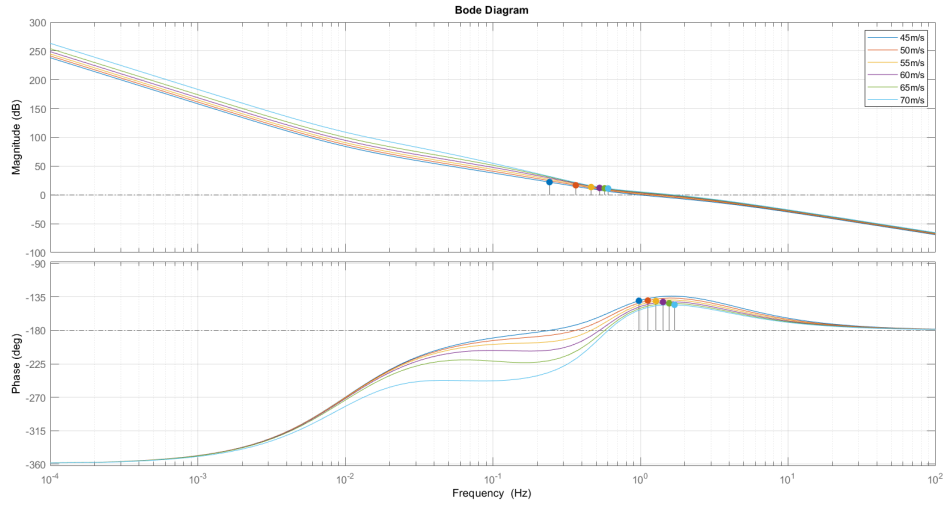
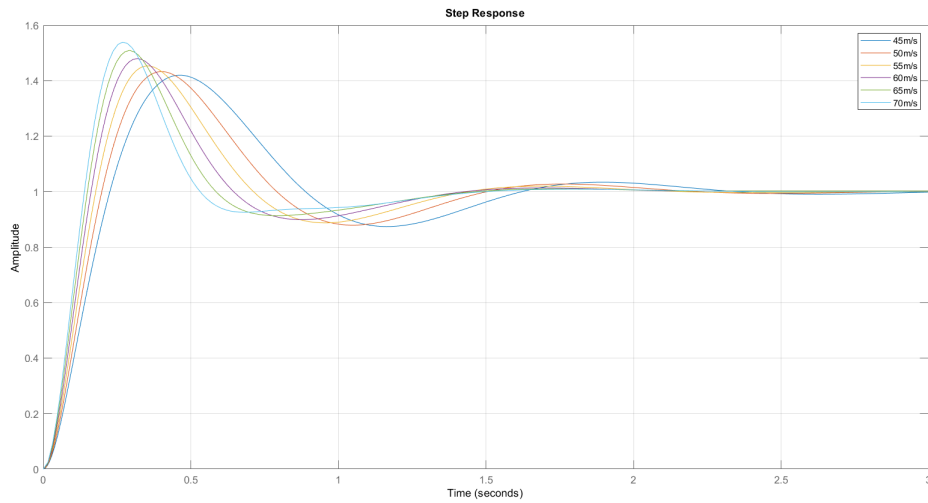
Figure 2.11: Open loop function  $L(s)$  at different velocities

Figure 2.12: Step response of the system at different velocities

$$C(s) = C = -G_{e,la}(0)^{-1}G_p(0) = \frac{mV_x^2(C_R l_R - C_F l_F) + C_F C_R (l_F + l_R)^2}{C_F C_R (l_F + l_R)}$$

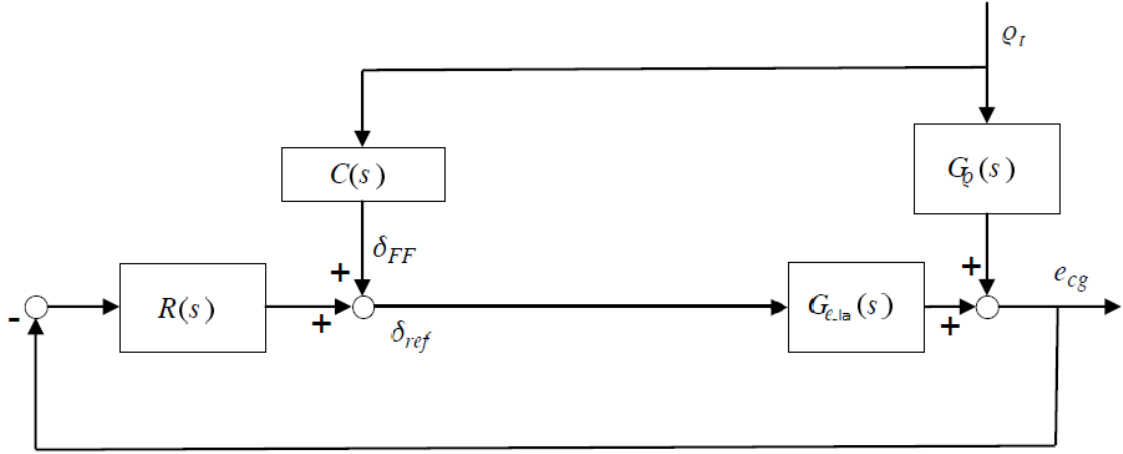


Figure 2.13: Final block scheme of lateral project with feed forward compensation

### 2.3.1 Simulation 8 - Lap of the track

Once projected, both controllers were tested in a first lap simulation on the non linear model in Simulink, focusing on the correctness of the speed profile and in the trajectory following with respect to the lateral error of the center of gravity. For completeness also the trajectory followed, the lateral error on lookahead and the controller steering angle  $\delta$  are reported. From what can be seen from those simulations, the controllers works well in the non linear settings of our model, since the speed profile is very well followed despite of the high velocity and the lateral error on center of gravity is in absolute value less then 0.3 [m] = 30 [cm]. Lap time is around 70 [s]

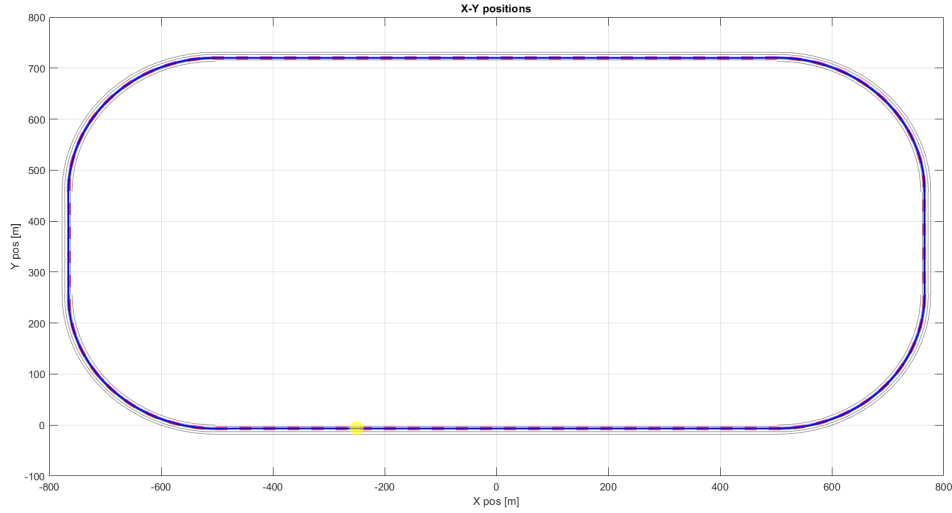


Figure 2.14: Trajectory of the simulation

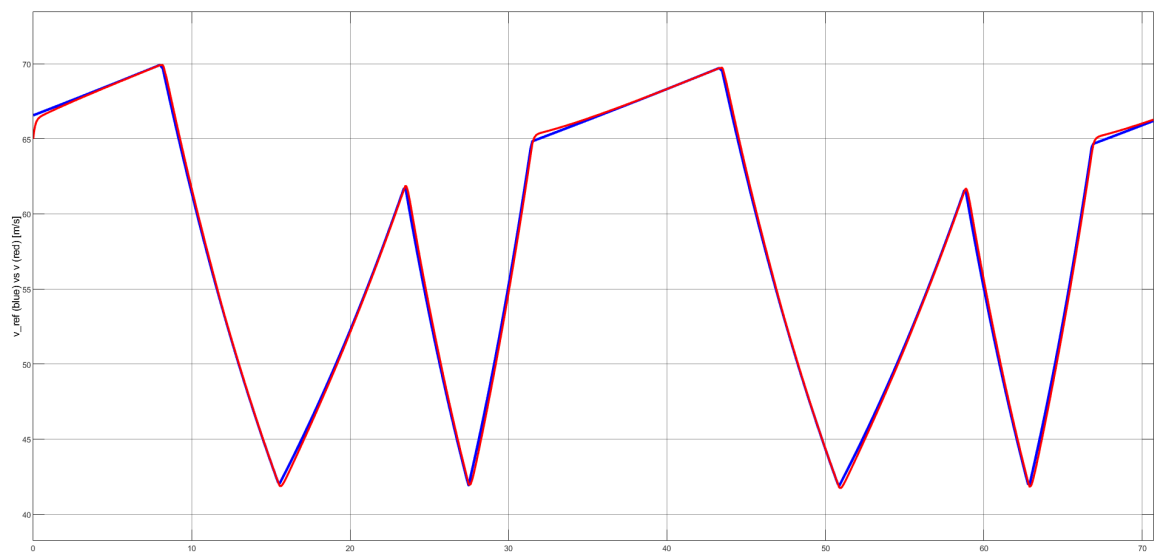


Figure 2.15: Speed profile tracking

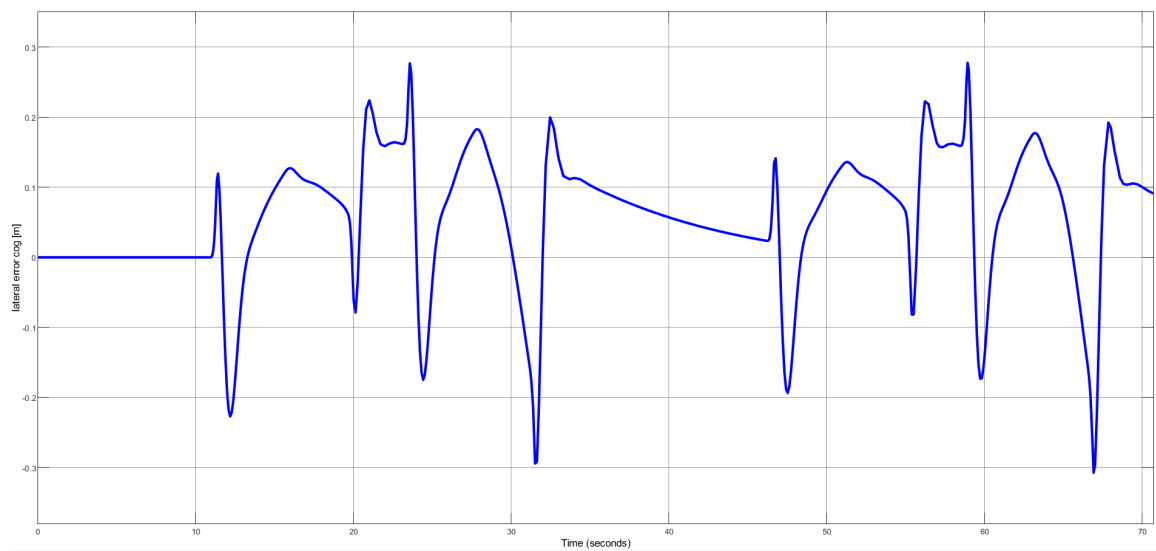


Figure 2.16: Lateral error on center of gravity

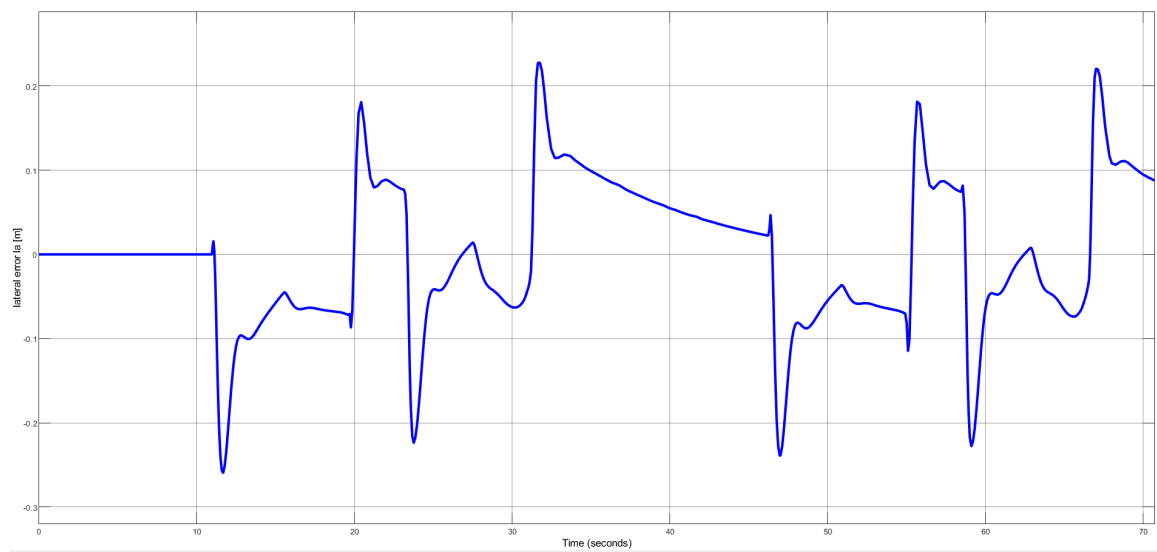


Figure 2.17: Lateral error on lookahead

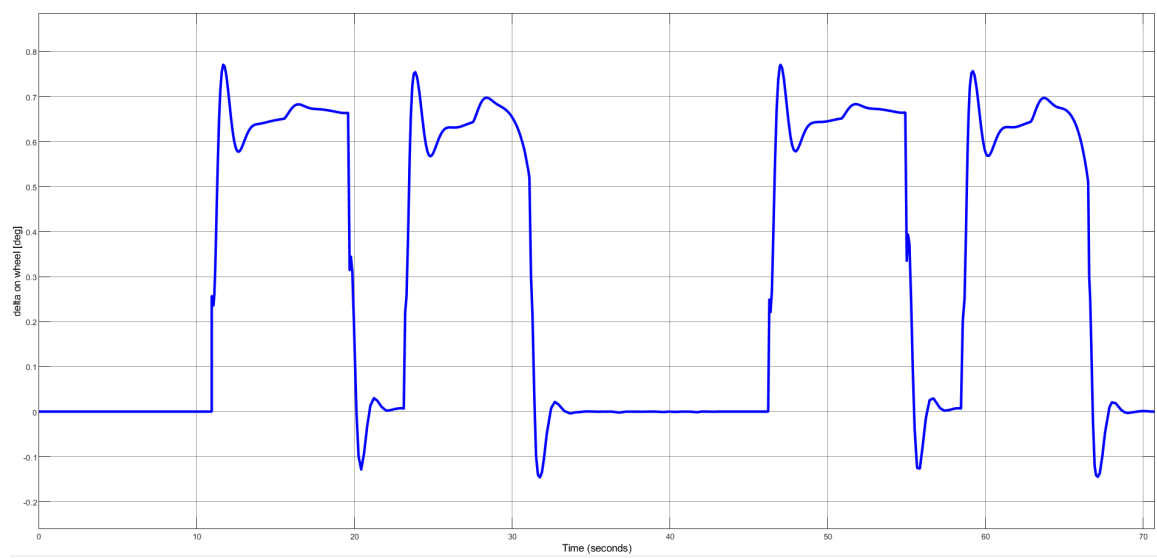


Figure 2.18: Controlled steering angle