## Problem 1 [100 points] Feedback Linearization

Consider the dynamics of one-link robotic arm, given by

dynamics as:

constant parameters.

rank(2) < rank(3).

feedback linearizable

$$egin{align} J_1 \ddot{q}_{\,1} + F_1 \dot{q}_{\,1} + rac{K}{N} \Big( q_2 - rac{q_1}{N} \Big) &= au, \ J_2 \ddot{q}_{\,2} + F_2 \dot{q}_{\,2} + K \left( q_2 - rac{q_1}{N} 
ight) &= -mgd \cos q_2, \ \end{pmatrix}$$

where  $q_1,q_2$  denote angular positions (in rad), and au denotes the actuation torque. The parameters  $\{J_i,F_i\}_{i=1}^2$  respectively denote the moments of inertia and rotational damping coeffcients of the actuator shaft (for i=1) and the link (for i=2), respectively. The parameters K, N, m, d denote the torsional spring constant, the transmission gear ratio, mass of the link, and distance of the center of gravity of the link from a fixed frame of reference, respectively. The **state vector** is  $x:=(q_1,q_2,\dot{q}_1,\dot{q}_2)^{ op}\in[0,2\pi)^2\times\mathbb{R}^2$  and the control input is  $u:= au\in\mathbb{R}$ .

## (i) Write the dynamics of the robotic arm in **standard control-affine form** $\dot{x}=f(x)+g(x)u$ , i.e., **explicitly write** the drift vector field f(x) and the input vector field g(x), both of size $4 \times 1$ , in terms of the state components $x_1, x_2, x_3, x_4$ and the parameters.

(a) [(5+5) + 20 = 30 points] Deciding feedback linearizability

1. We can do some algebra on each equation to isolate the  $\ddot{q_1}$  and  $\ddot{q_2}$  terms on the RHS, and given au=u, we can express the

$$\dot{x} = \begin{bmatrix} \dot{x}_{1} \\ \dot{x}_{2} \\ \dot{x}_{3} \\ \dot{x}_{4} \end{bmatrix} = \begin{bmatrix} \dot{q}_{1} \\ \dot{q}_{2} \\ \ddot{q}_{1} \\ \ddot{q}_{2} \end{bmatrix} = \begin{bmatrix} x_{4} \\ \frac{-F_{1}q_{1} - \frac{K\left(q_{2} - \frac{q_{1}}{N}\right)}{N}}{J_{1}} \\ \frac{-F_{2}q_{2} - K\left(q_{2} - \frac{q_{1}}{N}\right) - dgm\cos q_{2}}{J_{2}} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{J_{1}} \\ 0 \end{bmatrix} u$$

$$= \begin{bmatrix} x_{3} \\ -\frac{F_{1}x_{3}}{J_{1}} - \frac{Kx_{2}}{J_{1}N} + \frac{Kx_{1}}{J_{1}N^{2}} \\ F_{2}x_{4} & Gdm\cos(x_{2}) & Kx_{2} + Kx_{1} \end{bmatrix} (f(x)) + \begin{bmatrix} 0 \\ 0 \\ \frac{1}{J_{1}} \\ 0 \end{bmatrix} (g(x)) u$$

$$(1)$$

$$\left[-\frac{F_2x_4}{J_2}-\frac{Gdm\cos{(x_2)}}{J_2}-\frac{Kx_2}{J_2}+\frac{Kx_1}{J_2N}\right] \qquad \left[\begin{array}{c} \overline{J_1} \\ 0 \end{array}\right]$$
 (ii) Determine the region in the state space  $[0,2\pi)^2 \times \mathbb{R}^2$ , over which this system is full state static feedback linearizable. (**Hint:** use Lec. 16 p. 11, also Step 1 in p. 12.)

1. From Lec 16 pg 11, condition (i) we check is that the M matrix has full rank. We see that:

$$=\begin{bmatrix} 0 & -\frac{1}{J_1} & -\frac{F_1}{J_1^2} & -\frac{F_1^2}{J_1^3} - \frac{K}{J_1^2N^2} \\ 0 & 0 & 0 & -\frac{K}{J_1J_2N} \\ \frac{1}{J_1} & \frac{F_1}{J_1^2} & \frac{F_1^2}{J_1^3} + \frac{K}{J_1^2N^2} & \frac{F_1^3}{J_1^4} + \frac{2F_1K}{J_1^3N^2} \\ 0 & 0 & \frac{K}{J_1J_2N} & \frac{K(F_1J_2+F_2J_1)}{J_1^2J_2^2N} \end{bmatrix}$$
 2.  $rank(M)=4=n$ , it is **full rank** over  $[0,2\pi)^2 \times \mathbb{R}^2$  since **no entries are a function of any element of the state**, they are all constant parameters.

(3)

(5)

(8)

(9)

(13)

(14)

(16)

(17)

(18)

(19)

(20)

(21)

(22)

(23)

(24)

(26)

(27)

(28)

x2 хЗ

x4

5

- $egin{bmatrix} g(x^0) & adj_f g(x^0) \ \end{bmatrix} = egin{bmatrix} 0 \ 0 \ 0 \ \end{bmatrix}$

3. We also need to check condition (ii) from the same reference, we see that:

$$\begin{bmatrix} g(x^0) & adj_f^2 g(x^0) \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$
(6)

$$\begin{bmatrix} adj_f g(x^0) & adj_f^2 g(x^0) \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$
 (7)

(b) [10 + 10 = 20 points] Finding  $\lambda(x)$  and r(i) Show that the scalar field  $\lambda(x)$  in the Theorem given in Lec. 16 p. 10, for this system, can be taken as  $\lambda(x)=x_2$ . (Hint: use Lec. 16 p. 13, Step 2.)

6. Then from Lec 16 pg 11, we've checked for the 2 constructive conditions, and from 16 pg 12, Step 1, we've verified the system is

4. For each of the lie brackets in 3., we see that the lie bracket does **not** add rank, and each 2 pair of  $g(x^0)$ ,  $adj_fg(x^0)$ ,  $adj_f^2g(x^0)$ 

are **not** full rank (they are rank(2)). Combining each pair of  $g(x^0)$ ,  $adj_fg(x^0)$ ,  $adj_f^2g(x^0)$  with the pair's lie bracket would still be

5. Then by definition the span is closed under Lie bracket, **is involutive** over  $[0,2\pi)^2 imes\mathbb{R}^2$ 

1. We can see that for  $\lambda(x)=x_2$ ,  $rac{\partial \lambda}{\partial x}=egin{bmatrix} 0 \\ 1 \\ 0 \\ 0 \end{bmatrix}$ 2. We also can see that since  $g(x^0)[2]=adj_fg(x^0)[2]=adj_f^2g(x^0)[2]=0$ :

$$\mathcal{L}_g\lambda(x)=\left[egin{array}{cccc} 0 & 0 & rac{1}{J_1} & 0 \end{array}
ight] \left[egin{array}{cccc} 0 \ 1 \ 0 \ 0 \end{array}
ight]=0$$
  $\mathcal{L}_{adj_fg(x^0)}\lambda(x)=\left[-rac{1}{J_1} & 0 & rac{F_1}{J_1^2} & 0 \end{array}
ight] \left[egin{array}{cccc} 0 \ 1 \ 0 \ 0 \end{array}
ight]=0$ 

$$\mathcal{L}_{adj_f^2g(x^0)}\lambda(x) = egin{bmatrix} -rac{F_1}{J_1^2} & 0 & rac{F_1^2}{J_1^3} + rac{K}{J_1^2N^2} & rac{K}{J_1J_2N} \end{bmatrix} egin{bmatrix} 0 \ 1 \ 0 \ 0 \end{bmatrix} = 0$$
 (10)

$$\mathcal{L}_{adj_f^3g(x^0)}\lambda(x) = egin{bmatrix} -rac{K_1}{J_1^3} - rac{K}{J_1^2N^2} & -rac{K}{J_1J_2N} & rac{F_1^3}{J_1^4} + rac{2F_1K}{J_1^3N^2} & rac{K(F_1J_2+F_2J_1)}{J_1^2J_2^2N} \end{bmatrix} egin{bmatrix} 0 \ 1 \ 0 \ 0 \end{bmatrix} = -rac{K}{J_1J_2N} 
eq 0 \ (11)$$

(a)(ii). (Hint: use Lec. 15 p. 12 for computing the relative degree)

3. We can also see that since  $adj_f^3g(x^0)[2] 
eq 0$ :

1. We can show first that for our system, 
$$y=h(x)=\lambda(x)=x_2$$
, and: 
$$\mathcal{L}_g\mathcal{L}_f^0h=\mathcal{L}_gh=\mathcal{L}_g\lambda(x)=\begin{bmatrix}0&0&\frac{1}{J_1}&0\end{bmatrix}\begin{bmatrix}0\\1\\0\\0\end{bmatrix}=0 \tag{12}$$

(ii) By directly computing the relative degree r (a positive integer), prove that the state equation derived in part (a)(i), augmented with the output equation  $y = \lambda(x) = x_2$ , indeed has relative degree 4, that is, satisfies the r = n condition in the region determined in part

 $\mathcal{L}_g\mathcal{L}_f^1h=\mathcal{L}_g\left\langlerac{\partial h}{\partial x},f
ight
angle=\mathcal{L}_g\left\langle\left[egin{array}{cccc}0&1&0&0
ight],f
ight
angle=\mathcal{L}_gx_4=\left[egin{array}{cccc}0&0&rac{1}{J_1}&0
ight]\left|egin{array}{cccc}0&0&0&0&1\end{array}
ight.$ 

 $\mathcal{L}_g \mathcal{L}_f^2 h = \mathcal{L}_g \left\langle rac{\partial}{\partial x} \mathcal{L}_f^1 h, f 
ight
angle = \mathcal{L}_g \left\langle rac{\partial}{\partial x} x_4, f 
ight
angle = \mathcal{L}_g \left\langle \left[egin{array}{c} 0 \ 0 \end{array}
ight], f 
ight
angle$ 

4. Then from Lec 16 pg 13, Step 2, we see that  $\lambda(x)=x_2$  satisfies the desired system of PDEs.

$$=\mathcal{L}_g f[3] = \left\langle rac{\partial}{\partial x} f[3], g 
ight
angle = \left\langle \left[ rac{K}{J_2 N} \quad rac{-K + dgm \sin{(x_2)}}{J_2} \quad 0 \quad -rac{F_2}{J_2} 
ight], \left[ egin{matrix} 0 \ 0 \ rac{1}{J_1} \ 0 \end{array} 
ight] 
ight
angle = 0 \ (15)$$

2. From 1., we see that 
$$\mathcal{L}_g\mathcal{L}_f^0h=\mathcal{L}_g\mathcal{L}_f^1h=\mathcal{L}_g\mathcal{L}_f^2h=0$$
, and  $\mathcal{L}_g\mathcal{L}_f^3h\neq 0$ .  
3. This means t hat it will at  $r=3+1$ ,  $y=\lambda(x)=x_2$ ,  $y^r=\mathcal{L}_f^4\lambda(x)+\mathcal{L}_g\mathcal{L}_f^3\lambda(x)u$  has a **nonzero u** coefficient.  
4. From 2. and Lec 15 pg 12, we prove that this SISO system has relative degree  $3+1=4=n$ 

- $lpha(x) = rac{-\left\langle rac{\partial}{\partial x} \mathcal{L}_f^3 h, f 
  ight
  angle}{\left\langle rac{\partial}{\partial x} \mathcal{L}_f^3 h, g 
  ight
  angle}$  $=F_{1}x_{3}+rac{F_{2}^{3}J_{1}Nx_{4}}{J_{2}^{2}K}+rac{F_{2}^{2}GJ_{1}Ndm\cos{(x_{2})}}{J_{2}^{2}K}+rac{F_{2}^{2}J_{1}Nx_{2}}{I^{2}}-rac{F_{2}^{2}J_{1}x_{1}}{I^{2}}$
- $+rac{2F_{2}GJ_{1}Ndmx_{4}\sin{(x_{2})}}{J_{2}K}-rac{2F_{2}J_{1}Nx_{4}}{J_{2}}+rac{F_{2}J_{1}x_{3}}{J_{2}} \ +rac{G^{2}J_{1}Nd^{2}m^{2}\sin{(x_{2})}\cos{(x_{2})}}{J_{2}K}-rac{GJ_{1}Ndm\,\mathrm{x_{4}}^{2}\left(t
  ight)\cos{(x_{2})}}{\kappa}$ 
  - (25)

4. Finally, for the new control system in new coordinates:

1. We see that for the elements of  $z = \tau(x)$ 

3. We can derive  $\tau(\cdot)$ :

1. We can derive  $\alpha(\cdot)$ :

(ii) Give physical interpretation of the components of the vector 
$$z$$
 (think of their units).

1. We see that for the elements of  $z=\tau(x)$ 
2.  $z_1$  is joint position
3.  $z_2$  is velocity of that joint

- $au^1(z) = egin{pmatrix} x_1 \ x_2 \ x_3 \ x \end{pmatrix} = egin{bmatrix} rac{rac{IV(F_2z_2 + J_2z_3 + Kz_1 + agm\cos{(z_1)})}{K}}{21} \ rac{N(F_2z_3 + J_2z_4 + Kz_2 dgmz_2\sin{(z_1)})}{K} \end{bmatrix}$ 
  - $egin{pmatrix} \dot{z_2} \ \dot{z_3} \ \dot{z_4} \end{pmatrix} = egin{bmatrix} 0 & 0 & 1 & 0 \ 0 & 0 & 0 & 1 \ 0 & 0 & 0 & 0 \end{bmatrix} egin{pmatrix} z_2 \ z_3 \ z_4 \end{pmatrix} + egin{pmatrix} 0 \ 0 \ 1 \end{pmatrix} (k_1 & k_2 & k_3 & k_4) egin{pmatrix} z_1 \ z_2 \ z_3 \ z_4 \end{pmatrix}$ (29) $=egin{bmatrix} egin{bmatrix} 0 & 1 & 0 & 0 \ 0 & 0 & 1 & 0 \ 0 & 0 & 0 & 1 \ k_1 & k_2 & k_3 & k_4 \end{bmatrix} egin{pmatrix} z_1 \ z_2 \ z_3 \ z_4 \end{pmatrix}$ (30)

3

 $\mathcal{L}_g \mathcal{L}_f^3 h = \mathcal{L}_g \left\langle rac{\partial}{\partial x} \mathcal{L}_f^2 h, f 
ight
angle = \mathcal{L}_g \left\langle rac{\partial}{\partial x} f[3], f 
ight
angle$  $I = \left\langle rac{\partial}{\partial x} \mathcal{L}_f^3 h, g 
ight
angle = rac{K}{I_1 I_2 N} 
eq 0$ 

control system in the feedback linearized coordinates with the new 4 imes 1 state vector z and the new control  $v \in \mathbb{R}$ , where z := au(x)and  $u = \alpha(x) + \beta(x)v$ .

(c) [(5+5+5+5) + 5 + 25 = 50 points] Normal form and closed loop simulation

(i) Use the Steps 3 and 4 in Lec. 16 p. 14-15 to derive the feedback linearizing tuple  $(\tau(\cdot), \alpha(\cdot), \beta(\cdot))$ . Also clearly write down the

 $+rac{GJ_{1}Ndmx_{2}\sin{(x_{2})}}{J_{2}}-rac{GJ_{1}Ndm\cos{(x_{2})}}{J_{2}}-rac{GJ_{1}dmx_{1}\sin{(x_{2})}}{J_{2}} \ -rac{J_{1}KNx_{2}}{J_{2}}+rac{J_{1}Kx_{1}}{J_{2}}+rac{Kx_{2}}{N}-rac{Kx_{1}}{N^{2}}$ 2. We can derive  $\beta(\cdot)$ :

$$z = au(x) = egin{pmatrix} \mathcal{A}(x) \ \mathcal{L}_f \lambda(x) \ \mathcal{L}_f^3 \lambda(x) \end{pmatrix} \ & \begin{bmatrix} z_1 \ z_2 \ z_3 \ z_4 \end{bmatrix} = egin{bmatrix} & x_2 \ & x_4 \ & -rac{F_2 x_4}{J_2} - rac{G d m \cos{(x_2)}}{J_2} - rac{K x_2}{J_2} + rac{K x_1}{J_2 N} \ & -rac{F_2 \left(-rac{F_2 x_4}{J_2} - rac{G d m \cos{(x_2)}}{J_2} - rac{K x_2}{J_2} + rac{K x_1}{J_2 N} 
ight)}{J_2} + \left(rac{G d m \sin{(x_2)}}{J_2} - rac{K}{J_2}
ight) x_4 + rac{K x_3}{J_2 N} \end{bmatrix}$$

 $\left( egin{array}{c} z_1 \ \dot{z}_2 \ \dot{z}_3 \end{array} 
ight) = \left[ egin{array}{cccc} 0 & 1 & 0 & 0 \ 0 & 0 & 1 & 0 \ 0 & 0 & 0 & 1 \end{array} 
ight] \left( egin{array}{c} z_1 \ z_2 \ z_3 \end{array} 
ight) + \left( egin{array}{c} 0 \ 0 \ 0 \end{array} 
ight] v$ 

 $eta(x) = rac{1}{\mathcal{L}_{q}\mathcal{L}_{f}^{3}\lambda(x)} = rac{J_{1}J_{2}N}{K}$ 

4.  $z_3$  is the acceleration of that joint, as we derived in (a)(i) for  $\ddot{q}_2$ 5.  $z_4$  is the Lie derivative of  $z_3=\ddot{q_2}$  along f, intuitively it is how the acceleration term  $z_3=\ddot{q_2}$  changes as we move along the system dynamics vector field f.

2.  $z_1$  is joint position

3.  $z_2$  is velocity of that joint

diffeomorphism, and  $x = \tau^{-1}(z)$ )

1. We can define  $x = \tau^{-1}(z)$ :

to integrate out z(t):

$$\tau^1(z) = \begin{pmatrix} x_2 \\ x_3 \\ x_4 \end{pmatrix} = \begin{bmatrix} z_1 \\ \frac{N(F_2 z_3 + J_2 z_4 + K z_2 - dgm z_2 \sin{(z_1)})}{K} \\ z_2 \end{bmatrix}$$
 (2) We can use  $\tau$  to translate the initial state in  $x(t=0)$  to  $z(t=0)$ , and then use the z coordinate closed-loop dynamics  $(v=k^{\top}z)$ 

values = the negative characteristic polynomial coefficients: k = [-24, -50, -35, -10], and we can plot a simulator output as

Feedback linearization, mode = derivs\_alphabetau

below, one simulating by z, the other using u and original system dynamics (as expected they are equivalent). Derivation in

(iii) Use a simple pole placement controller  $v=k^{ op}z$  where  $k:=(k_1,k_2,k_3,k_4)^{ op}\in\mathbb{R}^4$  to stabilize the feedback linearized states to

the origin. For this stablizing controller, **submit the plots** for  $x_1, x_2, x_3, x_4, u$  versus time t = 0:0.01:5 with initial condition  $x(t=0)=(\pi/6,\pi/3,1,2)^{ op}$ . (Hint: recall that over the region in which the system is feedback linearizable, the map au must be a

3. Using the matrix in 2. (the companion matrix), we can see that if we want the eigenvalues to be at -1, -2, -3, -4, then we set the k

derivation\_1c.py, simulation in 1c.py.

from IPython.display import Image

Image(filename='plot.png')

400

300

100

In [1]:

Out[1]:

0 -1005 time (s) **+** Q ≠ ∠ 🖺 x=4.801 y=8 Feedback linearization, mode = derivs\_z 400 x2 хЗ 300 200 100 0 -100

2

time (s)