

① a) weiter

$$\dot{Q}_{\text{aus}} = \dot{m}(h_1 - h_2) + \dot{Q}_R$$

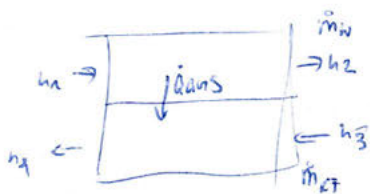
$$= 0.3 (-125.67) \frac{\text{kJ}}{\text{s}} + 100 \frac{\text{kJ}}{\text{s}}$$

aufheizen Strom ein

$$\Rightarrow \boxed{\dot{Q}_{\text{aus}} = 62.29 \text{ kW}}$$

$$1b) \bar{T}_{kf} = \frac{\int T ds}{s_3 - s_4} = \frac{T_4 - T_3}{\ln \left( \frac{T_4}{T_3} \right)} = \frac{298.15 - 288.15}{\ln \left( \frac{298.15}{288.15} \right)} \Rightarrow 293.12 = \bar{T}_{kf}$$

1c)

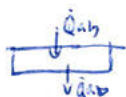


$$-\dot{S}_{\text{erz}} = \dot{m}_w (s_2 - s_1) + \dot{m}_{kf} (s_4 - s_3) + \frac{\dot{Q}_{\text{ab}}}{\bar{T}_{kf}} \leftarrow \text{aus (a)}$$

$$(s_2 - s_1) = s^\circ(T_2) - s^\circ(T_1) - R \ln \left( \frac{P_2}{P_1} \right)$$

$$(s_4 - s_3) = s_A^{\text{if}}(T_4) - s^{\text{if}}(T_3)$$

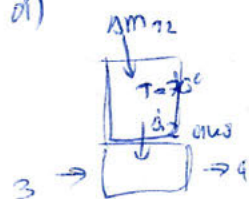
$$= c^{\text{if}} \ln \left( \frac{T_4}{T_3} \right)$$



$$\dot{S}_{\text{erz}} = \dot{Q}_{\text{ab}} \frac{T_2 - \bar{T}_{kf}}{\bar{T}_R \cdot \bar{T}_{kf}} = 62.29 \text{ kW} \cdot \frac{373.14 - 293.12}{373.14 \cdot 293.12} = 48.91 \frac{\text{W}}{\text{K}}$$

d)

Energiebilanz in Reaktor / Halboffens system



$$\Delta E = \Delta m_{12} (h_{in}) + \dot{Q}_{\text{aus}} \Delta t + m_{\text{in}} (h_{\text{inside}}) - m_{\text{out}} (h_{\text{inside}})$$

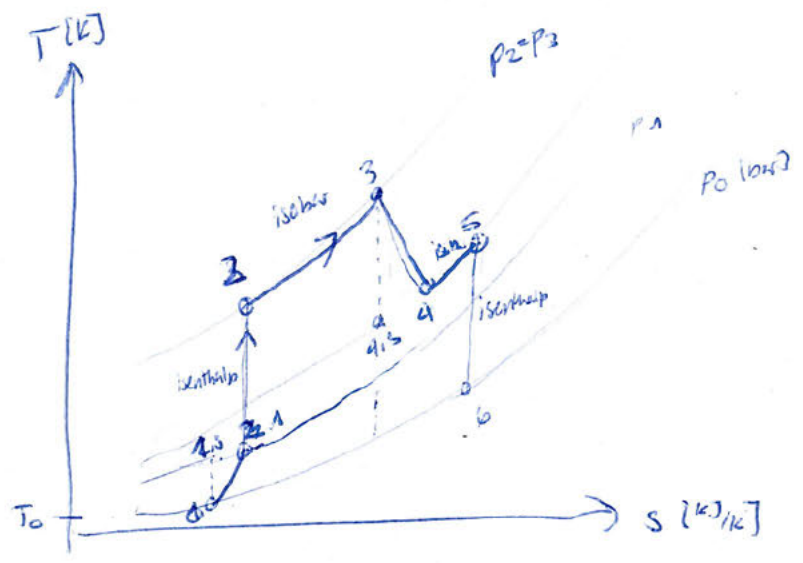
$$h_{in} = h(20^\circ\text{C}) = 83.96 \frac{\text{kJ}}{\text{kg}}$$

$$h_{\text{inside}} = 430.32 \frac{\text{kJ}}{\text{kg}} \quad (\text{aus (a), } 100^\circ\text{C}) \quad (\text{inkl. } x_2)$$

$$\Delta m_{12} = \frac{(-\dot{Q}_{\text{aus}} \Delta t + m_{\text{out}} (h_{\text{inside}}) - m_{\text{in}} (h_{in}))}{h_{in}} = \frac{(-35000 \frac{\text{kJ}}{\text{s}} + (5755 \text{ kg} \cdot 430.32 - 35000 \cdot 83.96))}{83.96} = 8175.8 \text{ kg}$$

# ② Energie am Flugzeug Triebwerk

a) T-s



$c_p = 1.006$   
 $\kappa = 1.4$

	T [K]	p [bar]
0		
1 isentrop $n=1$	<del>243.10</del>	0.191
2 isentrop $n=1$		
3 isobar		
4 $n=0$		
5	931.9	0.5
6	314.48	0.191

plug into Energiebilanz gestrichelt

$$2\dot{m}(h_6 - h_0) = -w_6^2 + w_{\text{Luft}}^2 \rightarrow w_{\text{Luft}}^2 - 2\dot{m}(h_6 - h_0) = w_6^2$$

$\dot{m}$  might be wrong  
 $\Rightarrow$  way too small!

$$\sqrt{200^2 \frac{\text{m}^2}{\text{s}^2} + 2 \cdot 5.8 \cdot 71.9} = \boxed{w_6 = 202.07 \text{ m/s}}$$

c)  $\dot{m}$

$$\Delta e_{\text{str}} = e_{\text{str}6} - e_{\text{str}0}$$

Energiebilanz

$$\dot{E}_{\text{str}} = \dot{m} (h_6 - h_0 - T_0 (s_6 - s_0) + \frac{w_{\text{Luft}}^2 - w_6^2}{2})$$

$$e_{\text{str}} = h_6 - h_0 - T_0 (s_6 - s_0) + \frac{w_{\text{Luft}}^2 - w_6^2}{2}$$

$$e_{\text{str}} = 71.9 - 243.15 \left( \overset{0.362}{s_6 - s_0} \right) + \frac{200^2 - 5102}{2}$$

$$\underline{= 968656 \text{ kJ/kg}}$$

$$s_6 - s_0 \stackrel{!}{=} s^0(T_6) - s^0(T_0) - R \ln \left( \frac{p_6}{p_0} \right) \rightarrow 0$$

$$\stackrel{!}{=} c_p \cdot \ln \left( \frac{T_6}{T_0} \right) - R \ln \left( \frac{p_6}{p_0} \right) \rightarrow 0$$

$$\frac{c_p}{c_v} = \gamma \rightarrow c_p = \gamma \cdot c_v = 1.4 \cdot 1.006 = 1.4084$$

$$= 1.4084 \cdot \ln \left( \frac{314.95}{243.1} \right) = \boxed{-0.362 \text{ kJ/kg}} = s_6 - s_0$$

d)  $\dot{E}_{\text{exverl}} = T_0 \cdot \dot{S}_{\text{erz}}$

$$\dot{S}_{\text{erz}} = \dot{S}_{\text{vorverd.}} + \dot{S}_{\text{Brennkammer}} + \dot{S}_{\text{Turbine}}$$

$$\dot{m} \left( \dot{S}_{\text{ges}} (s_1 - s_0) + \dot{m}_k (s_4 - s_3) + \frac{\dot{Q}_B}{T_B} \right)$$

$$\dot{S}_{\text{erz}} = (s_1 - s_0) + 1/6.293 (s_4 - s_3) + \frac{\dot{Q}}{T_B}$$

$$\dot{m}_{\text{ges}} = \dot{m}_k + \dot{m}_H$$

$$\dot{m}_{\text{ges}} = \dot{m}_k + 5.293 \dot{m}_k$$

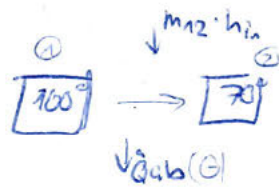
$$\boxed{\dot{m}_k = \dot{m}_{\text{ges}} / (6.293)}$$

$$(s_1 - s_0) = c_p \cdot \ln \left( \frac{T_1}{T_0} \right) - R \ln \left( \frac{p_1}{p_0} \right)$$

$$(s_4 - s_3) = c_p \cdot \ln \left( \frac{T_4}{T_3} \right) - R \ln \left( \frac{p_4}{p_3} \right)$$

1e)  $\Delta S_{12}$  ~~4/5/2~~

Entropiebilanz

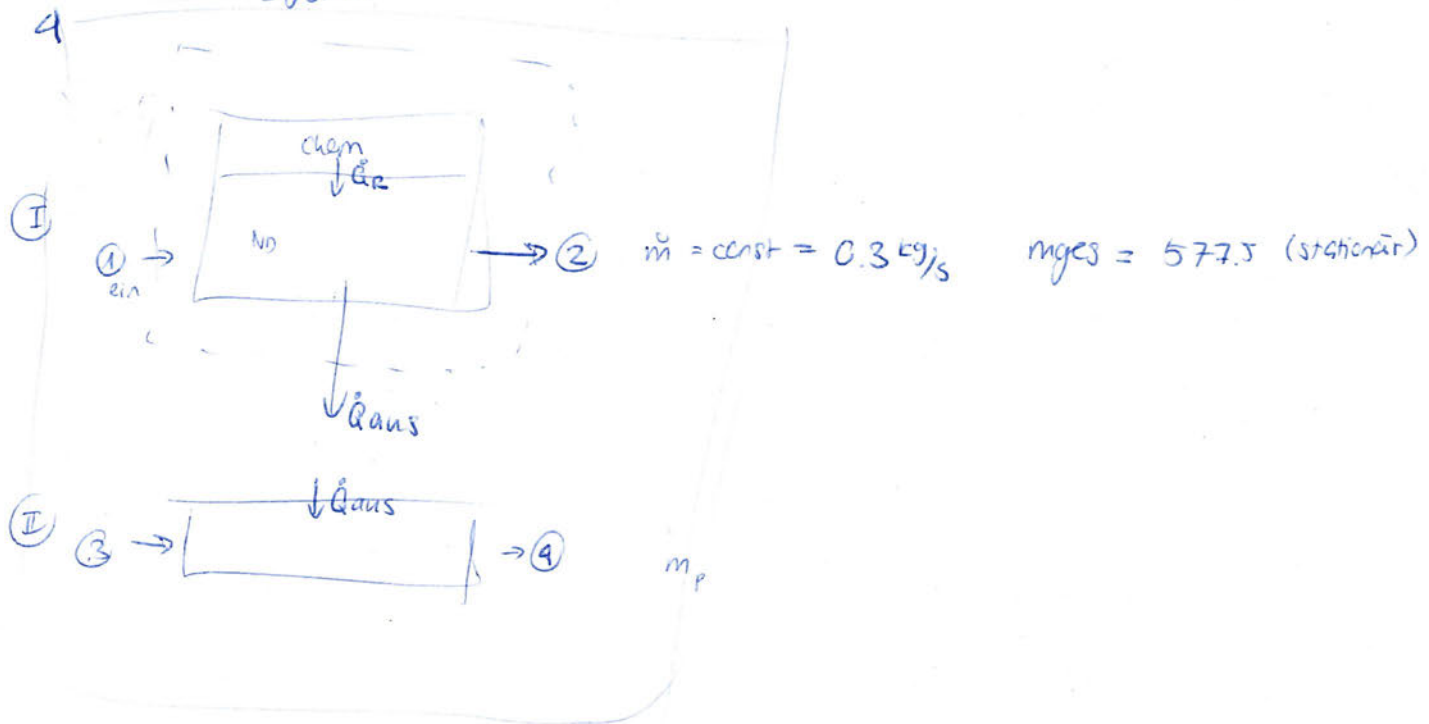


~~AS = 4/5/2~~

$$\Delta S = (m_{tot}^{+m_{12}} s_2 \text{ (at } T=70^\circ) - m_1 s_1 \text{ (at } T=100) + \frac{\dot{Q}_{ab}}{T_{UW}} \quad \text{4/5/2}$$

2.1

	T [K]	P [bar]	x
1	343.14		0.005
2	373.14		0.005
3	288.15	P <sub>B</sub>	
4	298.15	P <sub>B</sub>	



a)  $\dot{Q}_{\text{aus}}$  mit Energiebilanz (I)

$$0 = \dot{m} (h_1 - h_2) + \dot{Q}_2 - \dot{W}_r^0 - \dot{Q}_{\text{aus}}$$

$$(h_1 - h_2)^{\text{Wasser}} = h_1(T=343.14) - h_2(T=373.14)$$

Dampfanteil  $x_1 = x_2$

$$h_1 = h_{f1} + x_1(h_{g1} - h_{f1}) \quad \text{hat } T_1$$

$$h_1 = 304.649$$

$$h_2 = h_{f2} + x_2(h_{g2} - h_{f2})$$

$$h_2 = 430.82$$

$$T_{f1} = 292.98 \quad h_{f2} = 419.04 \quad = x_1 - x_2$$

$$h_{g1} = 2622.68 \quad h_{g2} = 2676.1 \text{ m} \rightarrow \dot{m}$$

$\Rightarrow$  Seite 3

3  
b)  $w_6^2$  with 576

$$0 = \dot{m}(h_6 - h_5) + \cancel{\dot{Q}_{16}} + \dot{m} \left( \frac{w_5^2 - w_6^2}{2} \right)$$

$$2(h_6 - h_5) = w_5^2 - w_6^2 \quad \Rightarrow \quad w_6^2 = w_5^2 - 2(h_6 - h_5)$$

$$(h_6 - h_5) = c_p (T_6 - T_5) = 1.006 \cdot (319.4 - 431.9) = -118.25$$

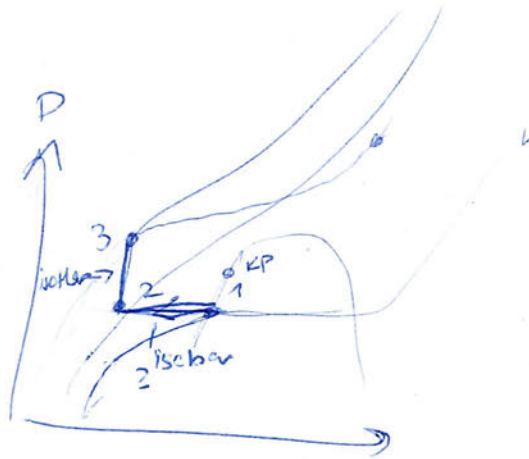
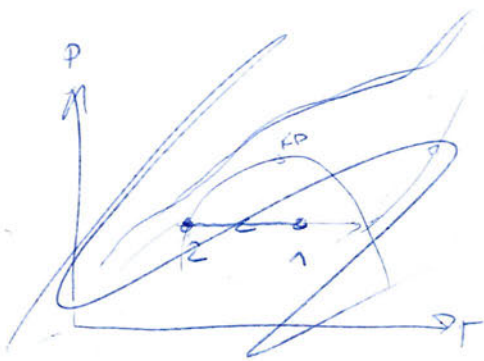
$$w_6 = \sqrt{220^2 - 2 \cdot -118.25} = 220.26 \text{ m/s}$$

$$d) \quad \epsilon_{\text{K}} = \frac{\dot{Q}_{\text{zu}}}{\dot{W}_r} = \frac{|\dot{Q}_k|}{|\dot{Q}_{\text{ab}}| + \dot{W}_k}$$

e) die Temperatur würde die ~~Reaktor~~ bei der gleichen Temp wie das Kühlmedium im Gleichgewicht sein. ( $T^* = T_i - b$ )  
 Kalt kann wegen des 2. H nicht ~~zu~~ zu warm übertragen werden.

3.8.5





b)  $\dot{m}$  durch Energiebilanz ~~im~~ verdichteter ~~gesamtes~~ system

$$\dot{W}_K = \dot{W}_{K^{rev}}$$

$$\dot{W}_{K^{rev}} = \int_1^2 v dp \Rightarrow \dot{m} = \frac{\dot{W}_K}{\dot{W}_{K^{rev}}}$$

Temp. ~~unverändert~~

$$\frac{T_3}{T_2} = \frac{p_3}{p_2}$$

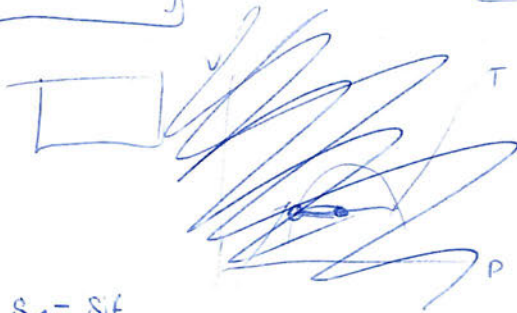
$$\Delta U = q - W$$

c)  $x_1$  direkt nach drossel

$$T_{verd} = T_1 - \Delta T_K =$$

$$T_1 = T_i - \Delta T_K$$

$$4 \rightarrow 1 \text{ adiabatisch}$$



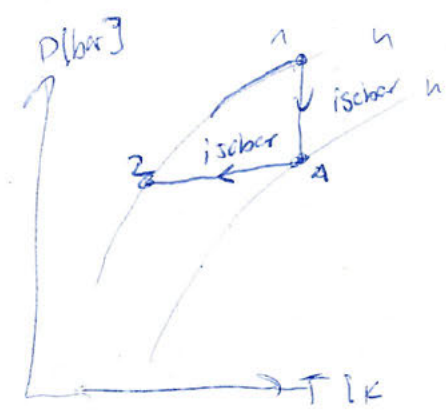
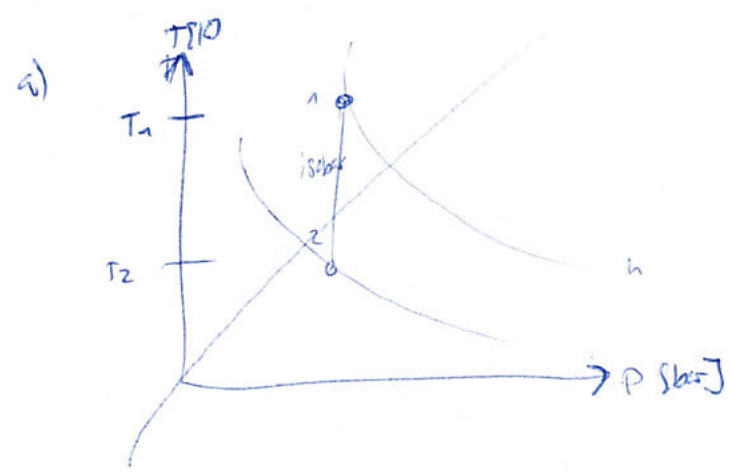
$$X = \frac{s_1 - s_i}{s_g - s_i}$$

$$\frac{ds}{dT} = 0 \Rightarrow \frac{T_1 - T_K}{T_K \cdot T_K} = s_{02}$$



③ EIS  
a)  $p_{01}$

④ TAB A-10



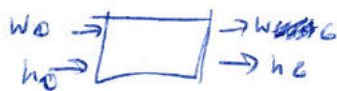
b)  $m_R$  mit Energiebilanz um

	T [K]	p
1 <del>NO</del> <del>vap sat</del>		
2 vap sat	Ausstab A-10 bei sat	$p_2 = p_1$
3 VSupen.	TAB MS bei 8 bar,	8
4 liq sat	31.33	8

b)  $W_0$

Energiebilanz umgestrichenes System

$$W_0 = W_{\text{Luft}} = 200 \frac{\text{m}}{\text{s}}$$



$$0 = \dot{m} (h_0 - h_e + \left( \frac{W_{\text{Luft}}^2 - W_0^2}{2} \right)) + \dot{Q} - \dot{W}$$

$$h_0 - h_e = h(T = 243.14) - h(T = T_6)$$

$$\dot{Q} = c_p (T_6 - T_0) = 1.006 \cdot (314.4 - 243.14) = 71.9 \frac{\text{kJ}}{\text{kg}} = (h_e - h_0)$$

→ find  $T_6$  via Schubdüse 5→6

$$\left( \frac{T_6}{T_5} \right) = \left( \frac{P_6}{P_5} \right)^{\frac{\kappa-1}{\kappa}} \Rightarrow T_6 = T_5 \cdot \left( \frac{P_6}{P_5} \right)^{\frac{\kappa-1}{\kappa}}$$

$$= 431.7 \cdot \left( \frac{0.191}{0.5} \right)^{\frac{0.4}{1.4}} = 344.4 \text{ K} = T_6$$

$\dot{m}_{\text{ges}}$  finden mit  $\dot{m}_{\text{H}_2} \rightarrow 3, \dot{m}_{\text{K}}$

$$\frac{\dot{m}_{\text{H}_2}}{\dot{m}_{\text{K}}} = 5.293$$

Energiebilanz um Brennkammer

$$0 = \dot{m}_e (h_2 - h_3) + \dot{Q}_B$$

$$\begin{aligned} (h_2 - h_3) &= \\ -c_p (T_2 - T_3) - c_p (T_B) &= 1.006 \cdot 1289 \left[ \frac{\text{kJ}}{\text{kg}} \right] \\ &= 1296.7 \left[ \frac{\text{kJ}}{\text{kg}} \right] \end{aligned}$$

$$\dot{m}_K = \frac{\dot{Q}_B}{h_2 - h_3} = \frac{1195}{1296.7} = 0.9215 \text{ kg/s}$$

$$\left. \begin{aligned} \dot{m}_{\text{tot}} + \dot{m}_K + \dot{m}_M &= 5.799 \text{ kg/s} \\ &\approx 5.8 \text{ kg/s} = \dot{m} \end{aligned} \right\}$$

$$\dot{m}_{\text{H}_2} = 5.293 \dot{m}_K = 4.8778 \text{ kg/s}$$