TFES Lab (ME EN 4650)

Airfoil Aerodynamics Lab: Raw Data Sheet

T_{atm} :	(°C)
P_{atm} :	(mm Hg)
Fan:	(Hz)

Freestream Conditions*				
(Pitot-static tube upstream of airfoil)				
Freestream Velocity				
P _{total} : Pitot-static probe (TOTAL)				
P _{static} : Pitot-static probe (STATIC)				
$P_{ m output} = rac{1}{2} ho U_{\infty}^2$				
P _{output} :	(in H ₂ O)			
*act massure coloct knob to CLI; O				

^{*}set pressure select knob to CH: 0

*set Pitot-static probe to x=1.5", y=9.5"

Lift and Drag				
(pivoting airfoil attached to lift/drag balance)				
α (deg)	Data Filename*			
0				
1				
2				
3				
4				
5				
6				
7				
8				
9				
10				
11				
12				
13				
14				
15				
16				

^{*}data are output in units of kg

^{*}data file: 1st column=LIFT, 2nd column=DRAG

Airfoil Pressure Distrubtion						
(airfoil plug with pressure taps along surface)						
P _{total} : Airfoil pressure taps						
P _{static} : Pitot-static probe (STATIC)						
$P_{output} = P_x - P_\infty$						
Before stall: α = (deg)						
x/c	Pressure Channel	P _{output} (in H₂O)				
		Top Surface	Bottom Surface			
0.05	1					
0.1	2					
0.2	3					
0.3	4					
0.4	5					
0.5	6					
0.6	7					
0.7	8					
0.8	9					
	After sta	all: α = (deg)			
x/c	Pressure Channel	P_{output} (in H_2O)				
A/C		Top Surface	Bottom Surface			
0.05	1					
0.1	2					
0.2	3					
0.3	4					
0.4	5					
0.5	6					
0.6	7					
0.7	8					

8.0

^{*}set airfoil at α =0°