

# DroMOOC

## Sensor fusion and state estimation Basic Level

### Observers and Kalman filter (Part II)

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# State estimation in presence of random disturbances (discrete time)

## Problem

The linear system (possibly non invariant) is affected by noise:

$$\mathbf{x}_{k+1} = \mathbf{F} \mathbf{x}_k + \mathbf{G} \mathbf{u}_k + \mathbf{v}_k$$

$$\mathbf{z}_k = \mathbf{C} \mathbf{x}_k + \mathbf{w}_k$$

with  $\mathbf{v}_k$ ,  $\mathbf{w}_k$  white centered gaussian non correlated noise:

$$E\{\mathbf{v}_k\} = 0 \quad E\{\mathbf{v}_k \mathbf{v}_j^T\} = \mathbf{Q}(t) \delta(k-j), \quad \mathbf{Q} = \mathbf{Q}^T \geq 0$$

$$E\{\mathbf{w}_k\} = 0 \quad E\{\mathbf{w}_k \mathbf{w}_j^T\} = \mathbf{R}(t) \delta(k-j), \quad \mathbf{R} = \mathbf{R}^T > 0$$

## Objective

The goal is to reconstruct  $\mathbf{x}_k$  from  $\mathbf{u}_k$  and  $\mathbf{z}_k$

- without bias on the estimation error
- with minimal variance of the estimation error

# Solution - discrete-time Kalman filter

## Discrete-time Kalman filter

$$\hat{\mathbf{x}}_{k+1|k+1} = \mathbf{F} \hat{\mathbf{x}}_{k|k} + \mathbf{G} \mathbf{u}_k + \mathbf{K}_{k+1} (\mathbf{z}_{k+1} - \hat{\mathbf{z}}_{k+1|k})$$

$$\hat{\mathbf{x}}_{k_0|k_0} = E \{ \mathbf{x}_{k_0} \}$$

⇒ this is an **observer** which performs an *estimation* of the state  $\hat{\mathbf{x}}_{k|k}$ , with gain  $\mathbf{K}_k$  varying in time (even in case of LTI systems).

## Notations

- Predicted state  $\hat{\mathbf{x}}_{k+1|k}$ 
  - ▶ Prediction error  $\boldsymbol{\varepsilon}_{k+1|k} = \mathbf{x}_{k+1} - \hat{\mathbf{x}}_{k+1|k}$ 
    - ★ Variance of the prediction error  $\boldsymbol{\Sigma}_{k+1|k} = E \{ \boldsymbol{\varepsilon}_{k+1|k} \boldsymbol{\varepsilon}_{k+1|k}^T \}$
- Estimated state  $\hat{\mathbf{x}}_{k+1|k+1}$ 
  - ▶ Estimation error  $\boldsymbol{\varepsilon}_{k+1|k+1} = \mathbf{x}_{k+1} - \hat{\mathbf{x}}_{k+1|k+1} \Rightarrow$  **without bias**:  
 $E \{ \boldsymbol{\varepsilon}_{k+1|k+1} \} = 0$ 
    - ★ Variance of the estimation error  $\boldsymbol{\Sigma}_{k+1|k+1} = E \{ \boldsymbol{\varepsilon}_{k+1|k+1} \boldsymbol{\varepsilon}_{k+1|k+1}^T \}$   
 $\Rightarrow$  **minimized**

# Discrete-time Kalman filter recursive equations

## Initialization

- Initial state estimate  $\hat{\mathbf{x}}_{0|0}$
- Initial prediction error variance  $\Sigma_{0|0}$

## Tuning

- $\hat{\mathbf{x}}_{0|0}$  initial guess
- $\Sigma_{0|0} \propto 1/\text{confidence in } \hat{\mathbf{x}}_{0|0} \text{ value}$
- $\mathbf{Q} \propto 1/\text{confidence in state equations}$
- $\mathbf{R} \propto 1/\text{confidence in measurements}$

## Prediction (before measurement $k + 1$ )

- 1  $\hat{\mathbf{x}}_{k+1|k} = \mathbf{F} \hat{\mathbf{x}}_{k|k} + \mathbf{G} \mathbf{u}_k$
- 2  $\Sigma_{k+1|k} = \mathbf{F} \Sigma_{k|k} \mathbf{F}^T + \mathbf{Q}$
- 3  $\hat{\mathbf{z}}_{k+1|k} = \mathbf{C} \hat{\mathbf{x}}_{k+1|k}$

## Estimation (update after measurement $k + 1$ )

- 4  $\mathbf{K}_{k+1} = \Sigma_{k+1|k} \mathbf{C}^T (\mathbf{C} \Sigma_{k+1|k} \mathbf{C}^T + \mathbf{R})^{-1}$
- 5  $\hat{\mathbf{x}}_{k+1|k+1} = \hat{\mathbf{x}}_{k+1|k} + \mathbf{K}_{k+1} (\mathbf{z}_{k+1} - \hat{\mathbf{z}}_{k+1|k})$
- 6  $\Sigma_{k+1|k+1} = (\mathbf{I}_n - \mathbf{K}_{k+1} \mathbf{C}) \Sigma_{k+1|k}$

# Convergence and steady-state Kalman filter for LTI systems

## Assumptions

- A0** The system is LTI, i.e.  $F$ ,  $G$ ,  $C$ ,  $Q$ ,  $R$  are constant matrices
- A1** If  $(C, F)$  is detectable, and  $(F, H)$  is stabilisable with  $H$  s.t.  $Q = HH^T$ , the filter is asymptotically stable.

## Results

- ①  $\Sigma_{k+1|k}$  tends to a **constant** matrix  $\Sigma_p$ , unique positive definite solution of the Riccati algebraic matrix equations:

$$\Sigma_p = F\Sigma_p F^T - F\Sigma_p C^T (C\Sigma_p C^T + R)^{-1} C\Sigma_p F^T + Q$$

- ②  $K_k$  tends to a **constant** matrix:

$$K = \Sigma_p C^T (C\Sigma_p C^T + R)^{-1}$$

- ③  $\Sigma_{k|k}$  tends to  $\Sigma_e = (I_n - KC) \Sigma_p$

- ④ In steady state, the estimation error becomes white noise if the model is perfect.

The Kalman filter then becomes an observer which gain  $K$  is **constant** and **entirely determined by the choice of matrices  $Q$  and  $R$** .

# Illustration of theoretical results

## Example 3 - sine wave with noise

Consider a simple integrator system with  $v_k$  and  $w_k$  state and measurement noise of variances  $Q_{real}$  and  $R_{real}$ , and  $u_k$  an sinusoidal input:

$$x_{k+1} = x_k + u_k + v_k$$

$$z_k = x_k + w_k$$

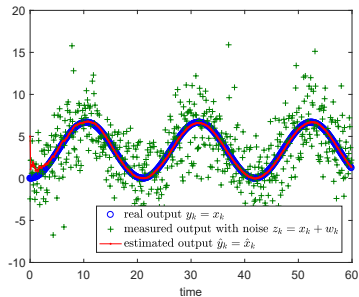
- ①  $\hat{x}_{k+1|k} = \hat{x}_{k|k} + u_k$
- ②  $\Sigma_{k+1|k} = \Sigma_{k|k} + Q$
- ③  $\hat{z}_{k+1|k} = \hat{x}_{k+1|k}$
- ④  $K_{k+1} = \Sigma_{k+1|k} (\Sigma_{k+1|k} + R)^{-1}$
- ⑤  $\hat{x}_{k+1|k+1} = \hat{x}_{k+1|k} + K_{k+1} (z_{k+1} - \hat{z}_{k+1|k})$
- ⑥  $\Sigma_{k+1|k+1} = (1 - K_{k+1}) \Sigma_{k+1|k}$

## Steady state ( $\hat{x}_k = \hat{x}_{k|k}$ )

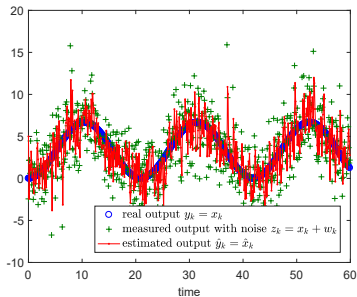
$$\hat{x}_{k+1} = \hat{x}_k + u_k + K(z_k - \hat{x}_k - u_k)$$

$$K = \frac{1}{2} \left( -\frac{Q}{R} + \sqrt{\left(\frac{Q}{R}\right)^2 + 4\frac{Q}{R}} \right)$$

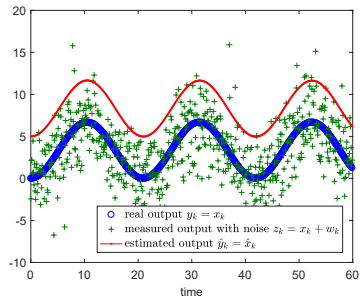
# Illustration of theoretical results (example 3)



(a)  $R = R_{real}$  (reference  $Q/R$ )



(b)  $R \ll R_{real}$  (large  $Q/R$ )



(c)  $R \gg R_{real}$  (small  $Q/R$ )

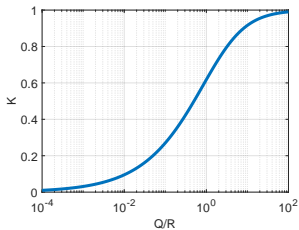


Figure: Influence of  $R$  as tuning parameter (with  $Q = Q_{real}$ )

# Back to Example 1 - Quadrotor pitch angle estimation

## Problem formulation

- **Available measurements to reconstruct pitch angle**

- ① Gyro measurements of pitch rate  $\dot{\theta}_{Gyr}^m(t) = GYR_y^m(t)$
- ② Accelerometer measurements of pitch angle

$$\theta_{Acc}^m(t) = \arctan \left( \frac{-ACC_x^m}{\sqrt{(ACC_y^m)^2 + (ACC_z^m)^2}} \right)$$

- **Towards a state-space formulation**

- ① Gyro measurements  $\dot{\theta}_{Gyr}^m$  represent real pitch rate  $\dot{\theta}$  affected by a white noise  $v_{Gyr} \sim \mathcal{N}(0, \sigma_{Gyr}^2)$  and an almost constant bias  $b_{Gyr}$  with  $v_b \sim \mathcal{N}(0, \sigma_b^2)$ :

$$\dot{\theta}_{Gyr}^m(t) = \dot{\theta}(t) + b_{Gyr}(t) + v_{Gyr}(t)$$

$$\dot{b}_{Gyr}(t) = v_b(t)$$

- ② Accelerometer measurements  $\theta_{Acc}^m(t)$  represent real pitch angle  $\theta$  affected by a measurement noise  $w_\theta \sim \mathcal{N}(0, \sigma_{\theta_{acc}}^2)$ :

$$\theta_{Acc}^m(t) = \theta(t) + w_\theta(t)$$

## State space

- State:  $\mathbf{x}(t) = [\theta, b_{Gyr}]^T$
- Input:  $u(t) = \dot{\theta}_{Gyr}^m$
- Output:  $z(t) = \theta_{Acc}^m$
- State noise:  $\mathbf{v} = [v_{Gyr}, v_b]^T$
- Measurement noise:  $w_\theta$



# Back to Example 1 - Quadrotor pitch angle estimation

## Continuous-time state-space model

$$\begin{cases} \dot{\theta}(t) = \dot{\theta}_{Gyr}^m(t) - b_{Gyr}(t) - v_{Gyr}(t) \\ \dot{b}_{Gyr}(t) = v_b(t) \\ \theta_{Acc}^m(t) = \theta(t) + w_{\theta}(t) \end{cases}$$
  
$$\Leftrightarrow \begin{cases} \dot{x}(t) = \begin{bmatrix} 0 & -1 \\ 0 & 0 \end{bmatrix} x(t) + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u(t) + v(t) \\ z(t) = \begin{bmatrix} 1 & 0 \end{bmatrix} x(t) + w_{\theta}(t) \end{cases}$$

## Discrete-time state-space model

$$F = \begin{bmatrix} 1 & -T_s \\ 0 & 1 \end{bmatrix}, \quad G = \begin{bmatrix} T_s \\ 0 \end{bmatrix}$$

## State space

- State:  $x = [\theta, b_{Gyr}]^T$
- Input:  $u = \dot{\theta}_{Gyr}^m$
- Output:  $z = \theta_{Acc}^m$
- State noise:  
 $v = [v_{Gyr}, v_b]^T$
- Measurement noise:  $w_{\theta}$

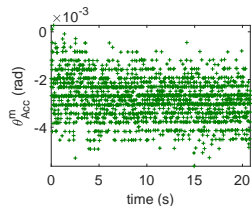
## Numerical values

- $T_s = 0.01s$
- $\hat{x}_{0|0} = ?$
- $\Sigma_{0|0} = ?$
- $Q = ?$
- $R = ?$

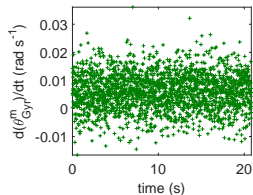
## Noise variance

$$Q = E \left\{ v(t) v(t)^T \right\} = \begin{bmatrix} \sigma_{Gyr}^2 & 0 \\ 0 & \sigma_b^2 \end{bmatrix}$$
$$R = E \left\{ w_{\theta}(t) w_{\theta}(t)^T \right\} = \sigma_{\theta_{acc}}^2$$

# Back to Example 1 - Quadrotor pitch angle estimation



(a)  $z_k = \theta_{Acc}^m$  (rad)



(b)  $u_k = \dot{\theta}_{Gyr}^m$  (rad/s)

Figure: Analysis of sensor readings w/o mvt

## Tuning

First tuning using sensor readings without movement ( $\theta = 0$ ):

- Noise variances

- ▶ Using  $u_k = \dot{\theta}_{Gyr}^m$

- ★  $\sigma_{Gyr}^2 = E \left\{ (\dot{\theta}_{Gyr}^m)^2 \right\} = 4.3 \cdot 10^{-5}$

- ★  $\sigma_b^2 = 10^{-9}$  (we suppose that the model is quite accurate)

- ▶ Using  $z_k = \theta_{Acc}^m$

- ★  $\sigma_{\theta_{acc}}^2 = E \left\{ (\theta_{Acc}^m)^2 \right\} = 7 \cdot 10^{-7}$

- Initialization

- ▶  $\hat{x}_{0|0} = [E \{ \theta \} \quad E \{ b_\theta \}]^T = [0 \quad 6 \cdot 10^{-3}]^T$

- ▶  $\Sigma_{0|0} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$

⇒ This initial tuning can be adjusted to provide satisfactory results.

## Back to Example 1 - Quadrotor pitch angle estimation

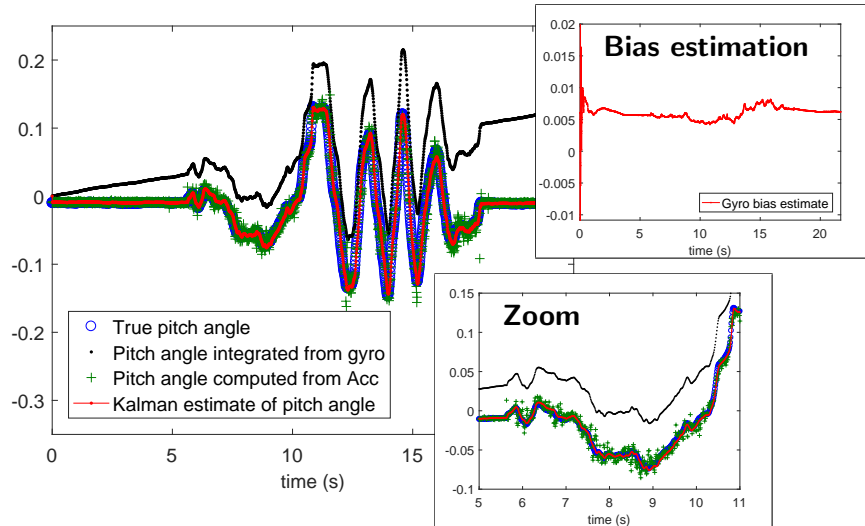


Figure: Pitch angle estimation - estimated state

## Back to Example 1 - Quadrotor pitch angle estimation

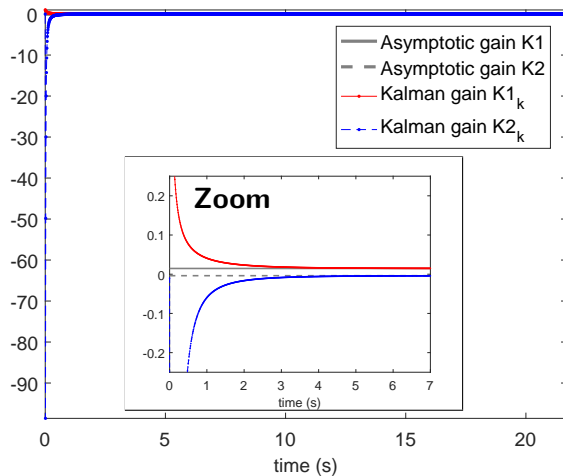


Figure: Pitch angle estimation - Kalman filter gains