Symmetry Breaking in Coherent Structures of Plane Couette Flow

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Agenda



Symmetry Breaking in Coherent Structures of Plane Couette Flow

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The Work

The Problem Mean Flow

The System

Navier-Stokes **Bulk Flow**

Perturbations

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Turbulence



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Why does it matter?

► Appears in physical flows - mixing, combustion, flow past bodies, etc.

Turbulence



Symmetry Breaking in Coherent Structures of Plane Couette Flow

Why does it matter?

- Appears in physical flows mixing, combustion, flow past bodies, etc.
- Difficult to predict and highly chaotic direct numerical simulation (DNS) highly infeasible.

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Turbulence Mean Flow



► The approach taken towards characterizing turbulence was by considering statistical, mean field theories such as Kolmogorov's scaling law, or the law of the wall

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Turbulence

Mean Flow



► The approach taken towards characterizing turbulence was by considering statistical, mean field theories such as Kolmogorov's scaling law, or the law of the wall

 Mean field theories cannot capture turbulent dynamics but DNS takes too long Symmetry Breaking in Coherent Structures of Plane Couette Flow

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Navier-Stokes



The Navier-Stokes equation is the linear conservation statement for Newtonian, incompressible fluids:

$$\rho \left(\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \cdot \nabla \mathbf{v} \right) = -\nabla \rho + \mu \nabla^2 \mathbf{v} + \mathbf{f}, \tag{1}$$

though in order to fully specify the flow, other conservation statements and boundary conditions are also necessary.

► Isothermal system means that we need not consider the equation of state

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though in order to fully specify the flow, other conservation statements and boundary conditions are also necessary.

- Isothermal system means that we need not consider the equation of state
- ▶ Incompressible system implies that $\nabla \cdot \mathbf{v} = 0$.

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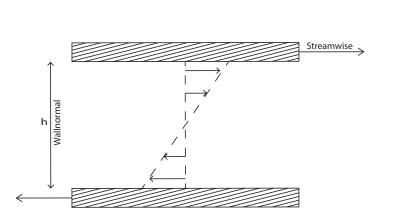
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The following geometry and assumptions are made

- 1. Fluid is constrained between two infinite parallel plates, each with some constant velocity
- 2. No-slip boundary conditions at the plates
- 3. Flow is unidirectional: [u, v, w](x, y, z) = u(y)
- 4. $\nabla p = 0$
- 5. Gravitational effects are negligible

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Under these assumptions, the Navier-Stokes equations reduce to a trivial ODE:

$$\frac{\mathrm{d}^2 u}{\mathrm{d} v^2} = 0, \tag{2}$$

with boundary conditions u(0) = 0, u(h) = V (for example), which gives the bulk flow solution

$$u(y) = u_0 \frac{y}{h}. (3)$$

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Perturbations



If we relax the unidirectional requirement on the flow, we can use the Reynolds Decomposition to decompose the new flow field into the sum of the mean flow and a perturbation from it. In this case, we can get $\mathbf{u}' = \mathbf{u}_{bulk} + \mathbf{u}$, where \mathbf{u} is a perturbation from the bulk flow. Then we get the following equation for \mathbf{u} :

$$\frac{\partial \mathbf{u}}{\partial t} + y \frac{\partial \mathbf{u}}{\partial x} + v \hat{\mathbf{x}} + \mathbf{u} \cdot \nabla \mathbf{u} = \frac{1}{Re} \nabla^2 \mathbf{u}, \nabla \cdot \mathbf{u} = 0$$
 (4)

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► Fluid flow as trajectories in an infinite dimensional phase space

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- ► Fluid flow as trajectories in an infinite dimensional phase space
- However, viscosity forces trajectories to lie on some finite-dimensional manifold

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State Space Hopf's Vision



- ► Fluid flow as trajectories in an infinite dimensional phase space
- However, viscosity forces trajectories to lie on some finite-dimensional manifold
- Can we map out this manifold? How does it depend on velocity?

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State Space 65,000 dimensions



► Divide computational domain into 3D grid

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State Space 65,000 dimensions



► Divide computational domain into 3D grid

► Track components of vector field at each grid point

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State Space

65,000 dimensions



- ► Divide computational domain into 3D grid
- ► Track components of vector field at each grid point
- ► The state space vector can then be written as $\mathbf{v} = (u_{11}, v_{11}, w_{11}, u_{12}, \dots)^T$

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State Space 65.000 dimensions



► Divide computational domain into 3D grid

► Track components of vector field at each grid point

► The state space vector can then be written as $\mathbf{v} = (u_{11}, v_{11}, w_{11}, u_{12}, \dots)^T$

► The dimension of this space is then given by $3N_xN_yN_z$

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Coherent Structures Equilibria



► Apart from the trivial laminar state, there exist other equilibrium states for **u**.

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Coherent Structures Equilibria



- ► Apart from the trivial laminar state, there exist other equilibrium states for **u**.
- ► These appear to wall off the turbulent regions of the phase space

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Equilibria



- ► Apart from the trivial laminar state, there exist other equilibrium states for **u**.
- ► These appear to wall off the turbulent regions of the phase space
- ► The following solutions were computed by John Gibson and Predrag Civtanović at Georgia Tech.

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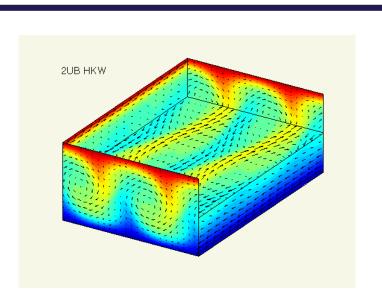
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Coherent Structures Equilibria





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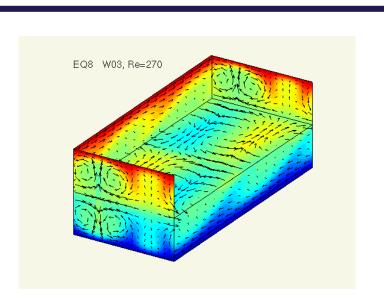
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Coherent Structures Equilibria





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Coherent Structures Periodic Orbits



► Limit cycles can also exist

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- ► Limit cycles can also exist
- ➤ These exist in the middle of the turbulent region of phase space

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- ► Limit cycles can also exist
- ► These exist in the middle of the turbulent region of phase space
- ► Do they have some influence on the turbulent dynamics?

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Periodic Orbits



► Limit cycles can also exist

- ► These exist in the middle of the turbulent region of phase space
- ▶ Do they have some influence on the turbulent dynamics?
- Can they help predict tubulent behavior?

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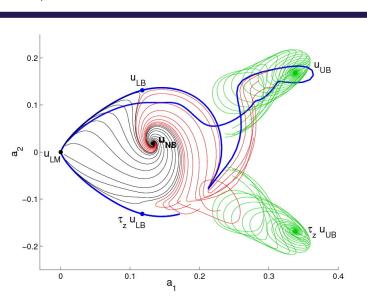
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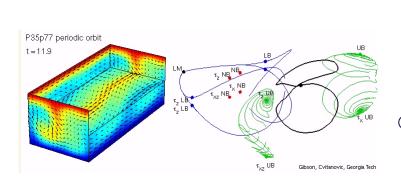
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Symmetry Breaking in Coherent Structures of Plane Couette Flow

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The generators of symmetry for **u** are as follows

 \blacktriangleright $[u, v, w](x, y, z) \rightarrow [-u, v, w](-x, y, z)$

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Symmetry Breaking in Coherent Structures of Plane Couette Flow

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- $\blacktriangleright [u,v,w](x,y,z) \rightarrow [u,-v,w](x,-y,z)$

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- \blacktriangleright $[u, v, w](x, y, z) \rightarrow [u, v, -w](x, y, -z)$
- ▶ $[u, v, w](x, y, z) \rightarrow [-u, -v, -w](x, y, z)$
- ▶ $[u, v, w](x, y, z) \rightarrow [-u, v, w](x + l_x, y, z + l_z)$

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Symmetry S Group



We will limit the explored symmetries to the S group, generated by the following operations:

 \bullet $s_1[u, v, w](x, y, z) = [u, v, -w](x + 0.5L_x, y, -z)$

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Symmetry S Group



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- \triangleright $s_3[u, v, w](x, y, z) = [-u, -v, -w](-x, -y, -z + 0.5L_z)$

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- ► I use the software system Channelflow, developed by John Gibson, to simulate the fluid flow and to solve for periodic orbits and equilibria.
- ► The minimum viable cell is used to speed up computation, with nondimensionalized lengths of $L_x = 1.75\pi$, $L_z = 1.2\pi$ and parallel plates at $y = \pm 1$

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- Periodic boundary conditions are imposed in the streamwise and spanwise directions, Dirichlet boundary conditions at the plates.

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- Periodic boundary conditions are imposed in the streamwise and spanwise directions, Dirichlet boundary conditions at the plates.
- This restricts the size of observable features in the physical flow

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The Hunt for the Red October



► How do we hunt through the immense phase space? (> 60,000 dimensions)

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The Hunt for the Red October



- ▶ How do we hunt through the immense phase space? (> 60,000 dimensions)
- Start from a random initial condition

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The Hunt for the Red October



▶ How do we hunt through the immense phase space? (> 60,000 dimensions)

- Start from a random initial condition.
- Integrate flow forward in time

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The Hunt for the Red October



- ▶ How do we hunt through the immense phase space? (> 60,000 dimensions)
- Start from a random initial condition.
- Integrate flow forward in time
- Look at recurrence diagrams

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 We first need to define a distance function between two flow states Symmetry Breaking in Coherent Structures of Plane Couette Flow

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- We first need to define a distance function between two flow states
- ► The physically motivated energy norm is used

$$d(\mathbf{u}, \mathbf{v}) = \frac{1}{V} \int_{V} \mathbf{u} \cdot \mathbf{v} dV \tag{5}$$

- We can now compare the flow at time t to time t + T by calculating d(u(t), u(t + T))
- Minima of these plots should indicate the presence of a periodic orbit

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Root Finding



➤ Once a promising candidate is found, it can be passed on to the root finder, to solve

$$\mathbf{u} - f^{\mathsf{T}}(\mathbf{u}) = 0 \tag{6}$$

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Root Finding



 Once a promising candidate is found, it can be passed on to the root finder, to solve

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► The root finder primarily uses the Newton method, using the generalized minimum residual method to determine the Newton step. Symmetry Breaking in Coherent Structures of Plane Couette Flow

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Root Finding



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- ▶ If the Newton step isn't good enough, a different step x_H is used, which minimizes $||Ax_h b||^2$.

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$$\mathbf{u} - f^{\mathsf{T}}(\mathbf{u}) = 0 \tag{6}$$

- ► The root finder primarily uses the Newton method, using the generalized minimum residual method to determine the Newton step.
- ▶ If the Newton step isn't good enough, a different step x_H is used, which minimizes $||Ax_h b||^2$.
- ▶ As opposed to the Newton step x_N which solves $Ax_N = b$

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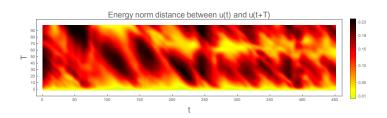
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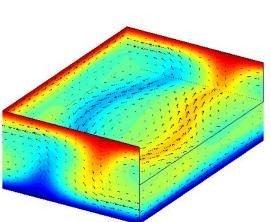
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This orbit has period T=85.47, and has symmetries that are not part of S.



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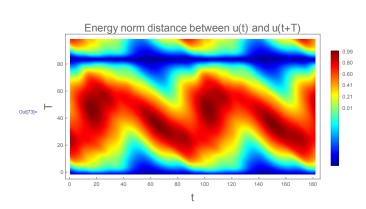
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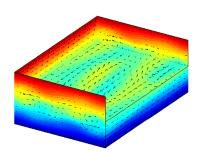
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Future Work

► Locate more periodic orbits in the symmetric subspace



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Future Work



- Remove symmetry constraints and attempt to locate asymmetric orbits

► Locate more periodic orbits in the symmetric subspace

Symmetry Breaking in Coherent Structures of Plane Couette Flow

The Problem

The System

Symmetry

The Work

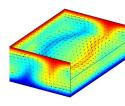
Recurrence Plots







A random field



A highly symmetric field

Symmetry Breaking in Coherent Structures of Plane Couette Flow

The Problem

Mean Flow

The System

Navier-Stokes

Perturbation

State Space Hopf's Vision

Coherent Structures

Symmetry

The Simulation

Root Finding

The Work

Recurrence Plots

I'd like to thank the following people for making the progress so far possible:

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