Tutorial 3

WEB-OPENING DETAIL ANALYSIS



TUTORIAL 3. WEB-OPENING DETAIL ANALYSIS

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TUTORIAL 3. WEB-OPENING DETAIL ANALYSIS

Summary

This tutorial presents the modeling and analysis processes for the reinforcement design of a beam with a circular web opening and explains the procedure for verifying results.

The essential contents for the user to experience in the example are the following:

- Detail modeling using plate elements to study the stress distribution around the vicinity of the opening
- Method of using *Rigid Link* for the structural link between the opening detail model and the model of the remaining parts with beam elements
- Method to extract the analysis results for plate elements

Extrude Elements (extension function which transforms nodes into line elements, line elements into plate elements and plate elements into solid elements) is used for the detail modeling of the opening. **Extrude Elements** is an extremely efficient tool to model complicated plate or 3-D models with minimal effort.

- 1. Preferences Setting
- 2. Enter Material and Section Properties
- 3. Structural Modeling
- 4. Enter Structure Support conditions
- 5. Enter Loading Data
- 6. Perform Structural Analysis
- 7. Interpret Analysis Results

Analysis Model and Load Cases

The summary and load cases for the structural model are shown in Fig.3.1.

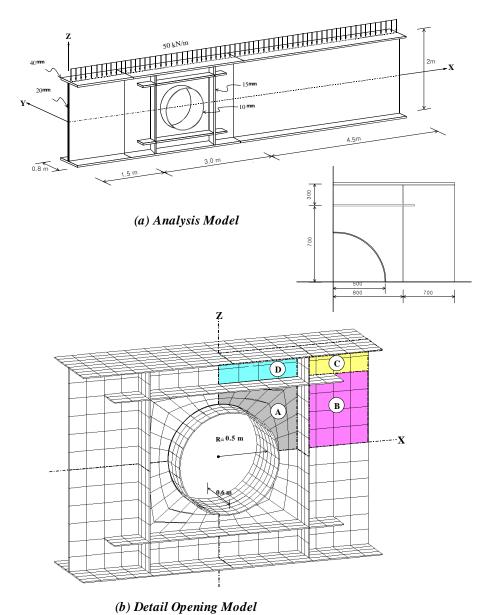


Figure 3.1 Beam Member with a Circular Web-Opening and a Detail Model

Preferences Setting

Unit System

First, open a new file. Then, use *Tools>Unit System* to specify the unit system adopted for the model.

- 1. Select *Tools>Unit System* in the main menu.
- 2. Select "mm" in the Length selection field.
- 3. Select "N(kg)" in the Force (Mass) selection field.
- 4. Click OK

For data entry and results verification, model the structure such that the beam ECS corresponds to the GCS. In other words, set \square X-Z to coincide with the web plane which is on the UCS x-y plane, and click \square Front View to adjust the working plane to correspond to the UCS x-y plane.

- 1. Click **A** X-Z in Structure > UCS > X-Z Plane from the Main Menu.
- 2. Enter "**0**, **0**, **0**" in the *Origin* field.
- 3. Enter "**0**" in the *Angle* field.
- 4. Click OK
- 5. Click Front View in the Icon Menu.

Enter Material and Section Properties

Assign the material properties for the beam and the thickness for all the parts such as vertical and horizontal stiffeners, the flange of opening reinforcing, etc.

- Material Number 1: Steel (A36)
- ➤ Thickness Number 10: 10 mm (Pipe)

15: 15 mm (Stiffeners) 20: 20 mm (Web) 40: 40 mm (Flange)

- 1. Select *Properties>Material Properties* in the Main Menu.
- 2. Click ____Add __ under the *Material* tab.
- 3. Select "**ASTM(S)**" in the *Standard* selection field.
- 4. Select "A36" in the DB selection field.
- 5. Click OK
- 6. Select the *Thickness* tab at the top of the *Properties* dialog box.
- 7. Click Add
- 8. Enter "10" in Thickness ID and "10" in In-plane & Out-of-plane.
- 9. Click Apply
- 10. Repeat steps 8 and 9 to enter successively thickness numbers "15", "20" and "40", and click OK
- 11. Select "**m**" in the unit system conversion window of the *Status Bar*. $^{\Omega}$
- 12. Click Close

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- Grid is not used in Tutorial 3. Toggle off all the Icons related to Grid.

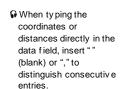
Structural Modeling

Generate 9 reference nodes in the UCS x-y plane to define the circular opening size and to locate the reinforcement (horizontal and vertical stiffeners).

The remaining zone including the circular opening is symmetrical about both axes. Only the upper-right quarter is modeled due to its symmetry (Fig. 3.1-©). The remaining 3 quarters are completed using symmetry copy (*Mirror Elements*).

- 1. Click Node Number and Lelement Number in the Icon Menu (Toggle on). ■
- 2. Click Auto Fitting in the Icon Menu.
- 3. Select *Node/Element > Create Nodes* in the Main Menu.
- 4. Enter "**0**, **0**, **0**" in the *Coordinates* (x, y, z) field.
- 5. Click Apply
- 6. Select *Translate Nodes* in the functions selection field (Fig.3.2–**1**).
- 7. Click Select All in the Icon Menu.
- 8. Confirm "Copy" in the *Mode* selection field.
- 9. Select "Unequal Distance" from the *Translation* selection field.
- 10. Confirm "x" in the Axis selection field.
- 11. Enter "0.8, 0.7" in the Distance field.
- 12. Click Apply
- 13. Click Select All in the Icon Menu.
- 14. Select "y" in the Axis selection field of Unequal Distance.
- 15. Enter "**0.7, 0.3**" in the *Distance* field.
- 16. Click Apply
- 17. Click Select Window in the Icon Menu and select node 1.
- 18. Select "Move" in the *Mode* selection field.
- 19. Select "x" in the Axis selection field.
- 20. Enter "**0.5**" in the *Distance* field and click Apply





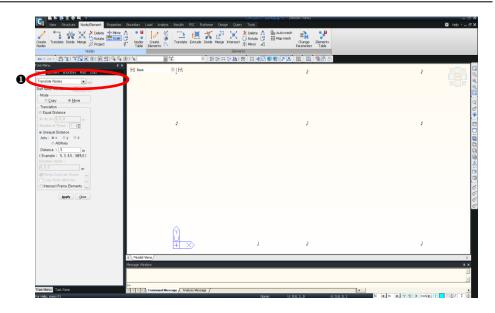


Figure 3.2 Generation of Nodes for Element Positions

While duplicating the nodes consecutively, use **!** Extrude Elements to generate elements concurrently to model beam elements for pipe-shaped stiffeners around the circumference of the opening.

These beam elements are used subsequently for the generation of the pipe-shaped stiffeners using *Extrude*, which expands the beam elements into plate elements.

- 1. Select *Element* tab in the Tree Menu or (Node/Element > Elements > Extrude from the Main Menu) (Fig.3.3–**①**).
- 2. Select **!** *Extrude Elements* in the functions selection field.
- 3. Confirm "Node→Line Element" in the *Extrude Type* selection field.
- 4. Click Select Window in the Icon Menu and select node 1.
- 5. Confirm "Beam" in the Element Type selection field.
- 6. Select "1: A36" in the Material selection field.

- The section number 999 for the beam elements is removed automatically after they have been extruded into plate elements. As such, it is not required to enter the section shape or dimensions.
- 7. Enter the section number "999" in the *Section* field. $^{\Omega}$
- 8. Select "Rotate" in the *Generation Type* selection field.
- 9. Enter "8" in the Number of Times field.
- 10. Enter "90/8" in the Angle of Rotation field.
- 11. Select "**z-axis**" in the *Axis of Rotation* selection field.
- 12. Confirm "**0**, **0**, **0**" in the *1st Point* field.
- 13. Click Apply

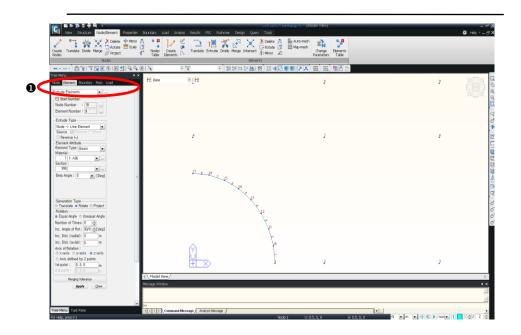


Figure 3.3 Generation of Temporary Beam Elements around the Opening Circumference

To create 8 plate elements in area \mathbb{A} of Fig. 3.1(c), the lines between nodes 2 and 5 and nodes 4 and 5 are divided into 4 equal spacings.

- 1. Click Auto Fitting (Toggle off).
- 2. Select the *Node* tab in the Tree Menu (Fig.3.4–**①**).
- 3. Select *Divide Nodes* in the functions selection field.
- 4. Enter "4" in the Number of Divisions field of Equal Distance.
- Click the *Nodes to Divide* field once and click successively nodes 2 and 5 and nodes 4 and 5.

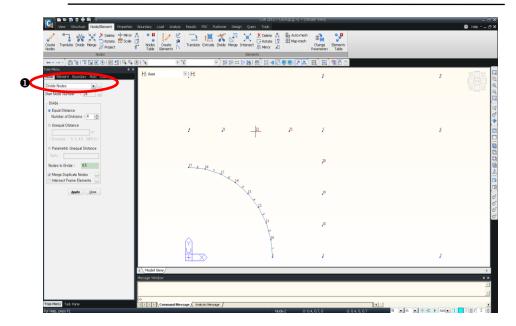


Figure 3.4 Division of nodes to create Plate Elements

Connect the generated nodes counter-clockwise to create the 8 plate elements in area (a) of Fig.3.1(c). The ECS thus-created consistently enables the user to use **Divide Elements** effectively when dividing the elements afterwards.

- Select "Plate" in the *Element Type* selection field and confirm "4 Nodes".
- 3. Confirm "1: A36" in the *Material Name* selection field.
- 4. Enter "20" in the *Thickness No.* field.
- 5. Click the *Nodal Connectivity* field and connect nodes **1, 2, 18, 10** to create plate element **9**.

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- 6. Connect nodes 10, 18, 19, 11 to create plate element 10.
- 7. Similarly, create successively the remaining plate elements 11 to 16.
- 8. Click Shrink in the Icon Menu (Toggle on).
- 9. Click **Q** Zoom out.

Figure 3.5 Generation of Plate Elements around the Circular Opening

- Government Use the Size tab of Display Option to adjust Zoom In and Zoom Out Factor.

Second ECS is defined according to the order

ECS.

in which nodes are assigned during the

generation of elements. It is adv isable to follow a consistent order at all times. Refer to Model Numerical Analy sis> Types of elements and related items>Plate Elements in Analy sis & Design Manual for the

Create 3 plate elements forming the boundaries of the B, C, D zones as shown in Fig.3.1(c) by connecting the corner nodes.

- 1. Click *Intersect Node* to remove the check (✓).
- 2. Connect nodes **2**, **3**, **6**, **5** to create plate element **17**. $^{\Omega}$
- 3. Connect nodes **5**, **6**, **9**, **8** to create plate element **18**.
- 4. Connect nodes 4, 5, 8, 7 to create plate element 19.

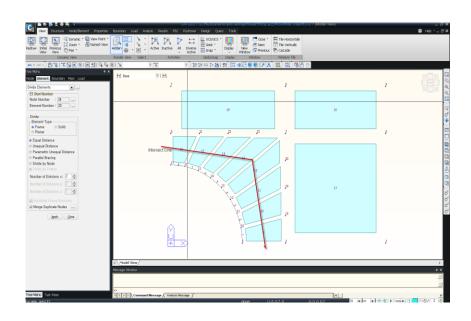


Figure 3.6 Generation of the Remaining Plate Elements of the Web

Divide the plate elements already created into appropriate sizes to form fine meshes.

- ⊕ To finish y our selection of elements, using the Intersect Line command, double-click at the last point.
- 1. Select *Divide Elements* in the functions selection field (Fig.3.7–**1**).
- 2. Use Select Intersect in View > Select > Intersect Line from the Main Menu to select the plate elements 9 to 16 in area (A) (Fig. 3.6) .

- 3. Select "Planar" in the *Element Type* selection field.
- 4. Confirm "Equal Distance" in the *Divide* selection field.
- 5. Enter "3" in the *Number of Divisions* x field ("3" means one element into three elements).
- 6. Enter "1" in the *Number of Divisions y* field ("1" means one element into one element).
- 7. Click Apply
- 8. Click Select Single in the Icon Menu to select element 17 of area ®.
- 9. Enter "4" in both the *Number of Divisions x* and y fields.
- 10. Click Apply
- 11. Select elements **18** and **19** in areas © and ® respectively.
- 12. Confirm "4" in the Number of Divisions x field.
- 13. Enter "2" in the Number of Divisions y field.
- 14. Click Apply



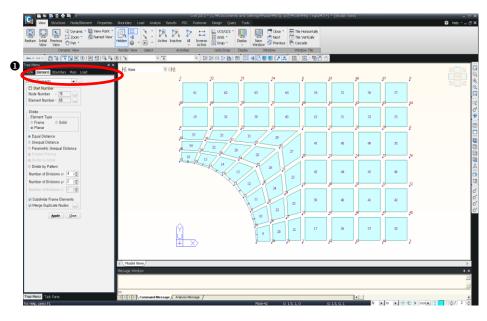


Figure 3.7 Division of Web Plate Elements

Create temporary beam elements at the locations of the reinforcing stiffeners and the flanges in order to generate the vertical and horizontal stiffeners and plate elements by extruding the beam elements into plate elements.

- 1. Select *Create Elements* in the functions selection field (Fig.3.8–**①**).
- 2. Select "General beam/Tapered beam" in the *Element Type* selection field.
- 3. Enter section number "998" in the Section No. field.
- 4. Check (✓) *Intersect Node*.
- 5. Click the *Nodal Connectivity* field once and connect nodes **4** and **58** to generate the temporary beams.
- 6. Connect nodes **2** and **8** to generate the temporary beams.
- 7. Enter section number "997" in the Section No. field.
- 8. Click the *Nodal Connectivity* field once and connect nodes **7** and **9** to create temporary beams at the upper flange position.

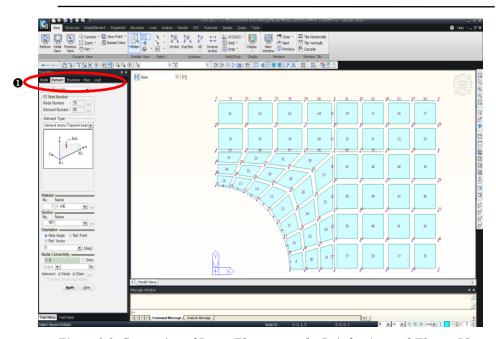


Figure 3.8 Generation of Beam Elements at the Reinforcing and Flange Plates

Use *Mirror Elements* to generate the elements in the remaining 3 quarters of the opening detail model.

- 1. Click Node Number and Element Number in the Icon Menu (Toggle off).
- 2. Click Select All and Auto Fitting in the Icon Menu.
- 3. Select *Mirror Elements* in the functions selection field (Fig.3.9–**1**).
- 4. Confirm "Copy" in the *Mode* selection field.
- 5. Select "z-x plane" in the *Reflection* selection field .
- 6. Confirm "y: 0" and click Apply
- 7. Click Select All in the Icon Menu.
- 8. Select "y-z plane" in the *Reflection* selection field.
- 9. Confirm "x: 0" and click Apply
- 10. Click Close

 $\ensuremath{\mbox{\wp}}$ when selecting plane that

you are mirroring elements

about, consider the plane as the perpendicular plane to

the plane you want to mirror. Ex) The first mirroring work

here is to mirror elements in the x-y plane (UCS). For this

the z-x plane in the reflection

case, the perpendicular

plane is the x-z plane, therefore, Step 5 chooses

selection field.

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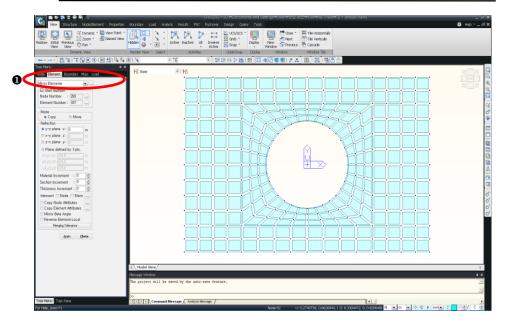


Figure 3.9 Complete Model of the Web

Extrude the temporary beam elements into plate elements to complete the reinforcing flange of the circular opening, the vertical and horizontal stiffeners and the flanges of the beam as shown in Fig.3.11.

- 1. Click loo View in the Icon Menu.
- 2. Click GCS in View > Grids/Snap > UCS/GCS from the Main Menu.
- 3. Select the *Works* tab (Fig.3.10–**①**).
- Double-click section number '999' (pipe-shaped stiffener) in *Properties* >Section.
- 5. Click Extrude Elements in the Main Menu > Node/Element > Extrude.
- Select "Line Elem.→ Planar Elem." in the Extrude Type selection field
- 7. Select "10: 0.010000" in the *Thickness* selection field.
- 8. Confirm "Translate" in the Generation Type selection field.
- 9. Type " $\mathbf{0}$, - $\mathbf{0}$.1, $\mathbf{0}$ " in the dx, dy, dz field of Equal Distance.
- 10. Enter "3" in the Number of Times field.
- 11. Click Apply
- 12. Click Select Identity-Elements in the Icon Menu.
- 13. Select "**Section**" in the Select Type field.
- 14. Click section number "998" (vertical, horizontal stiffeners).
- 15. Click Add
- 16. Click Close
- 17. Select "15: 0.015000" in the Thickness selection field.
- 10 Clials Apply
- 19. Repeat steps 12 to 16 to enter section number "**997**" (flange of the beam).
- 20. Select "40: 0.040000" in the Thickness selection field.
- 21. Enter "4" in the Number of Times field.
- 22. Click Apply

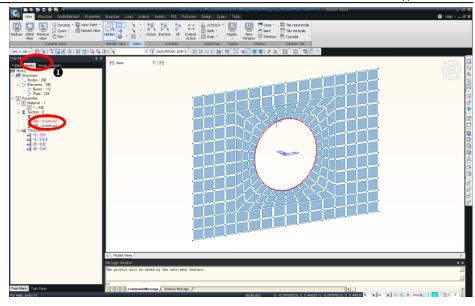


Figure 3.10 Section selection using Works Tree

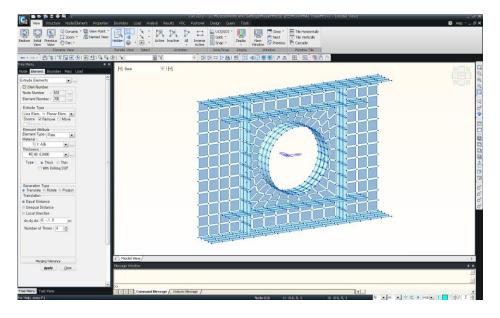


Figure 3.11 Complete One Side of the Opening Detail Model

To generate the flanges and stiffeners of the opposite face, select all the parts, except for the web, and use **Mirror Elements** to complete the opening detail model.

- 1. Click Select All in the Icon Menu.
- 2. After selecting the thickness number "20: 0.02" in the *Tree Menu* > *Works* tab > *Properties*>*Thickness*, right-click the mouse.
- 3. Select *Unselect* from the Context menu.
- 4. Select Mirror Elements in the Main Menu > Node/Element > Elements > Mirror.
- 5. Select "z-x plane" in the *Reflection* field.
- 6. Click Apply

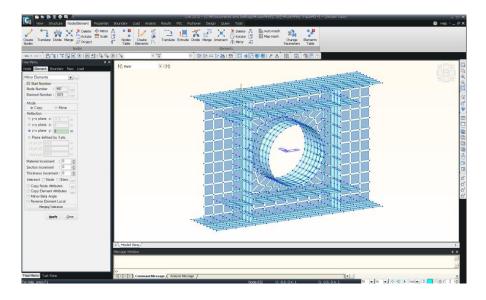


Figure 3.12 The Complete Opening Detail Model

After completing the opening reinforcing detail model, extend both ends of the beam elements to the supports to specify the support conditions.

Before creating the beam elements, create the nodes where support conditions are to be assigned.

- The unspecified axis coordinates are recognized as 0.
- 1. Select Create Nodes in the Main Menu > Node/Element > Nodes > Create Nodes.
- 2. Enter "-3" in the *Coordinates* (x, y, z) field.
- 3. Enter "1" in the Number of Times field.
- 4. Enter "9" in the Distances (dx, dy, dz) field.
- 5. Click Apply

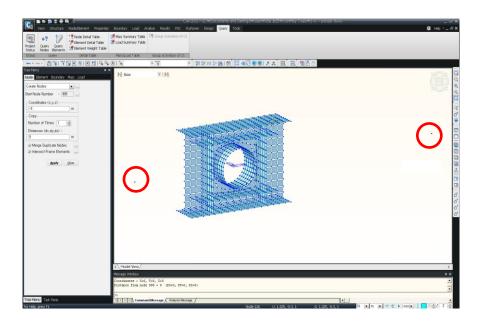


Figure 3.13 Creation of Nodes at the Beam Supports

- 1. Select Create Elements in the Main Menu > Node/Element > Create Elements.
- Select "General beam/Tapered beam" in the *Element Type* selection field.
- 3. Confirm "1: A36" in the Material Name selection field.
- 4. Click the button to the right of the **Section Name** selection field.
- 5. Select "mm" in the unit system conversion window of *Status Bar*.
- 6. Click Add
- 7. Confirm "I-Section" in the DB/User tab.
- 8. Select "User".
- 9. Enter "I 2000×800×20/40" in the *Name* field.
- 10. Enter "2000", "800", "20" and "40" in the H, B1, t_w and $t_f I$ fields, respectively.
- 11. Click OK
- 12. Click Close
- 13. Select "m" in the unit system conversion window of *Status Bar*.
- 14. Select "1: I 2000×800×20/40" in the Section Name selection field.
- 15. Click the Nodal Connectivity field once.
- 16. Connect nodes **997** and **183** and nodes **3** and **998** (Fig. 3.14) to create beam elements **1073** and **1074** respectively.

Enter Structure Support Conditions

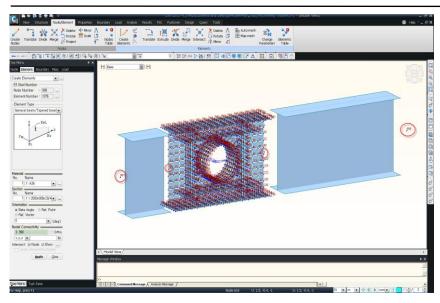


Figure 3.14 Creation of Beam Elements at Both Ends of the Opening Detail Model

Specify the pin joint support conditions at both ends of the beam

- 1. Select *Boundary* in the tab (Fig.3.15–**0**).
- 2. Confirm *Supports* in the functions selection field.
- 3. Check (\checkmark) "**D-AII**" and "**RX**" for boundary conditions.
- 4. Click Select Window in the Icon Menu.
- 5. Select both ends of the beam (nodes **997**, **998**).
- 6. Click Apply

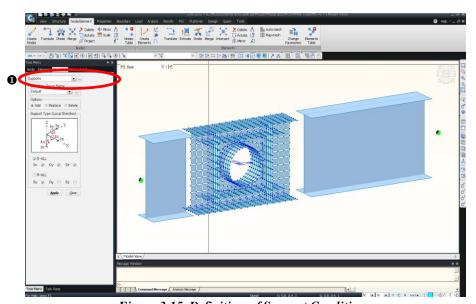


Figure 3.15 Definition of Support Conditions

Use **Rigid Link** to attribute the continuity conditions between the beams modeled as line elements and the detail model composed of plate elements.

- 1. Click **Zoom Window** (Toggle on) to magnify the opening detail model and click **Zoom Window** once again to Toggle off.
- 2. Select *Rigid Link* in the functions selection field.
- Click the *Master Node Number* field once and click the node (Fig. 3.16-1) to which the left beam extends in the Model window to enter "183" automatically.
- 4. Click Rigid Body in the *Typical Types* selection field.
- 5. Click Select Plane in the Main Menu > View > Select.
- 6. Select "YZ Plane".
- 7. Click the node at the left-end of the opening detail model.
- 8. Click Close
- 9. Click Apply
- 10. Repeat the steps 3~9 to specify the rigid body connection condition of the master node/slave nodes at the right end of the detail model (Fig. 3.16-2).



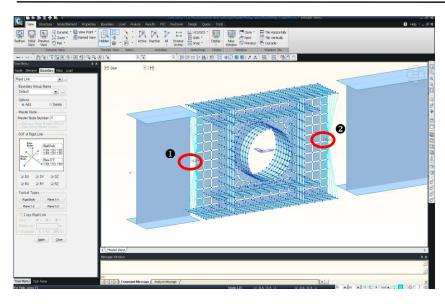


Figure 3.16 Rigid Link Setup

Enter Loading Data

Define Load Cases

- 1. Select *Load* in the tab (Fig.3.18–**1**).
- 2. Click the button ____ to the right of *Load Case Name*.
- 3. Enter the contents shown in Fig.3.17 in the Static Load Cases dialog box.
- 4. Click Add
- 5. Click Close

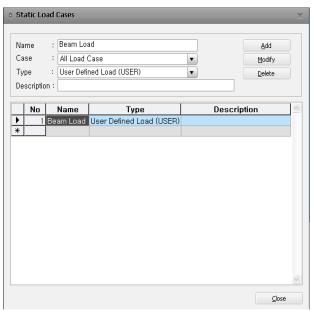


Figure 3.17 Load Cases

Define Uniformly Distributed Load

- 1. Click **Q Zoom Fit** in the Icon Menu.
- 2. Click Select Single in the Icon Menu.
- 3. Select the beams at both ends of the opening detail model (Fig.3.18– Θ).
- 4. Select *Element Beam Loads* in the functions selection field.

- 5. Confirm "Beam Load" in the *Load Case Name* selection field.
- 6. Enter "-50000" in the w field of *Value*.
- 7. Click Apply
- 8. Toggle off the *Hidden* icon

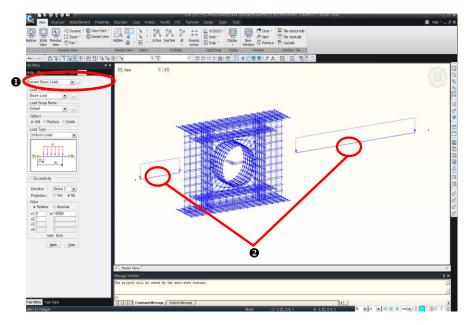


Figure 3.18. Assigning Uniformly Distributed Load on the top of the Beams

Define Concentrated Loads

- 1. Click Select Plane in the Main Menu > View > Select.
- 2. Select "XZ Plane".
- 3. Select any node in the plane of the web of the opening detail model.
- 4. Click Close
- 5. Click Activate in the Icon Menu.
- 6. Click **Zoom Window** in the Icon Menu to magnify the detail model.
- 7. Click **Select Polygon** in the Icon Menu.

When selecting elements by Select Poly gon or Select Intersect in the Icon Menu, double-click to end the selection.

- 8. Select the nodes where concentrated loads are applied as shown in Fig.3.19-1.
- 9. Select *Nodal Loads* in the functions selection field.
- 10. Confirm "Beam Load" in the Load Case Name selection field.
- 11. Enter "-50000*3/16" in the FZ field.
- 12. Click Apply
- 13. Click Select Window to select 2 unloaded nodes at both ends of the detail model.
- 14. Enter "-50000*3/16/2" in the FZ field.
- 15. Click Apply (Fig. 3.19).
- 16. Click Activate All in the Icon Menu.
- 17. Click Close
- 18. After Selecting "Element Beam Load" in the *Tree menu > Works* tab > Static Load > Static Load Case, right-click the mouse.
- 19. Select *Display Loads* from the Context Menu.
- 20. Confirm the Element Beam Element Load input.
- 21. Confirm the Point Load similarly following the steps 17 to 19 (Fig. 3.20).

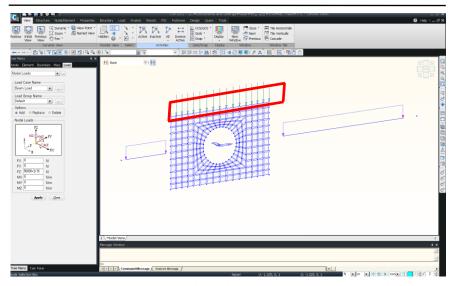


Figure 3.19 Concentrated Loads on the Opening Detail Model

Fig. 3.20 shows the screen display after checking the uniform distributed and point loads above using *Works Tree*.

Works Tree systematically organizes the model data by attributes for easy manipulation of data.

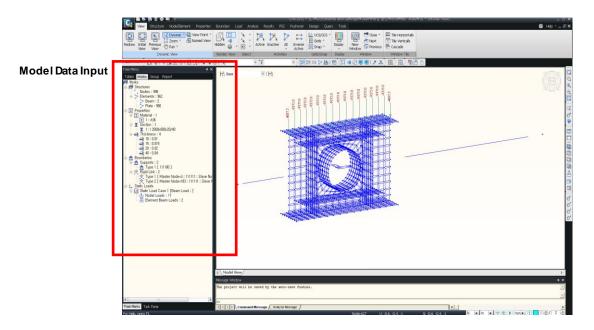


Figure 3.20 Complete Model

Perform Structural Analysis

Click Analysis in the Icon Menu to analyze the model. After completing the analysis, the program switches automatically to the post-processing mode, which provides access to the interpretation of analysis and design results.

Interpret Analysis Results

Verify Member Stresses

The opening detail model is modeled with plate elements. The analysis results and interpretation of results focus on the deformed shape and the variation of stresses in the vicinity of the opening.

- 1. Click Hidden (Toggle on) in the Icon Menu.
- 2. Click Shrink (Toggle off) in the Icon Menu.
- 3. Select Results>Stresses>Plane-Stress/Plate Stresses in the Main Menu.
- 4. Select "Sig-XX" in the *Components* selection field.
- Check (✓) "Contour" and "Legend" in the Type of Display selection field.
- 6. Convert to "kN" and "cm" in the unit conversion window.
- 7. Click Apply

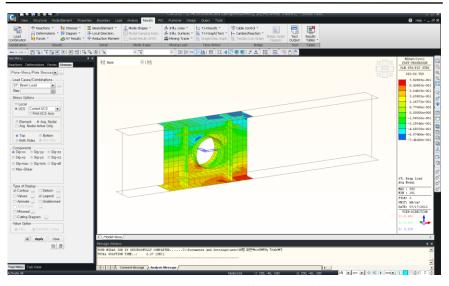


Figure 3.21 Stress Contour for Plate Elements

Auto-Compute Member Stresses

It is necessary to compute the element forces from the internal forces at each node in plate or solid elements for design purposes.

Use *Local Direction Force Sum* to compute the element forces automatically at the boundaries between the beam elements and the detail model.

- 1. Click Initial View in the Icon Menu.
- 2. Convert to "m" in the unit conversion window.
- 3. Click **Zoom Window** in the Icon Menu to magnify the boundary of the detail model and the right side line element (Fig. 3.22).
- 4. Select View > Display > Display Option > Draw > Hidden Option (Model).
- 5. Select *Outline* in *Type* of *Option Value* and click OK
- 6. Select Results>Local Direction Force Sum in the Main Menu.
- 7. Select *Plate Edge Polygon Select* in *Mode*.
- 8. Confirm "ST: Beam Load" in the Load Case selection field.
- 9. Click Hidden (Toggle off) in the Icon Menu.
- 10. Click nodes **980**, **971**, **607**, **616**, **980** successively as shown in Fig.3.22.
- 11. Click Hidden (Toggle on) in the Icon Menu.
- 12. Click Calculate in the Local Direction Force Sum dialog box.

The sum of all the nodal forces, contained in the specified section, is computed at the centroid of the section according to the local coordinates (Fig. $3.22-\Phi$) defined on the section for which element forces are to be computed. The computed value of the strong axis bending moment, My, for the member at the right end of the detail model is $506.25~\text{kN}\cdot\text{m}$.

 The member forces computed by *Local Direction Force Sum* are compared with the member forces of the linear element on the right side.

- 1. Select Results>Forces>Beam Forces/Moments in the Main Menu.
- 2. Click Apply
- 3. Click the button in to the right of *Contour* in *Type of Display* and check (✓) "Reverse Contour".
- 4. Click OK in the *Contour Details* dialog box.
- 5. Move the mouse cursor to the middle of element **1074** and snap. Use *Fast Query* to confirm "My **506.25** kN·m" at the i end.
- 6. Change *Components* in the *Beam Forces/Moments* dialog bar to compare the member forces of *Local Direction Force Sum* with those of *Bubble Tip*.

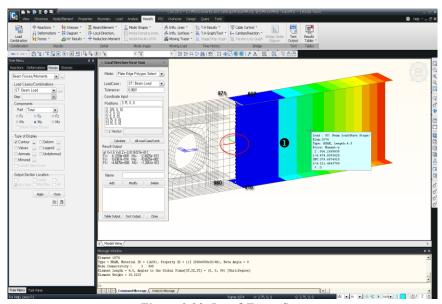


Figure 3.22 Local Force Sum