

Lecture notes: Trajectory tracking control for robot manipulators

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This lecture note is based on

- Chapter 8 in M. Spong **Robot modeling and control**.
- Chapter 2 of de Wit, Carlos Canudas, Bruno Siciliano, and Georges Bastin, eds. **Theory of robot control**. Springer Science & Business Media, 2012. Available online.
- Chapter 5: Position Control and Trajectory Tracking of Murray et. al. **A Mathematical Introduction to Robotic Manipulation**
- Refereces given herein.

Tracking control in the joint space: design a controller that allows the system to

- follow a given time-varying trajectory $q_d(t)$ and its successive derivatives $\dot{q}_d(t)$ and $\ddot{q}_d(t)$ — the desired velocity and acceleration.

Inverse dynamics control (or computed torque control)

- linearizing and decoupling robot manipulator dynamics;
- nonlinearities (Coriolis and centrifugal terms, gravity terms) compensated by adding these **forces** to the control input.
- CON: Compensational force can be difficult to compute. Not robust to modeling errors.

Lyapunov-based control:

- search for asymptotic stability —exponentially, if possible.

Passivity-based control:

- exploits the passivity properties of the robot manipulator dynamics.

Inverse dynamics control

Consider the dynamic model

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + N(q) = \tau.$$

Question: How to design trajectory tracking control?

- Recall: Trajectory tracking control for

$$\dot{x} = Ax + Bu$$

$$\dot{x}_d = Ax_d + Bu_d$$

$$e = x - x_d$$

- Consider use it for nonlinear trajectory tracking control: What is the linear system now?

$$\begin{aligned}\dot{e} &= \dot{x} - \dot{x}_d = Ax - Ax_d + Bu - Bu_d \\ &= Ae + B(u - u_d)\end{aligned}$$

design v s.t. $e \rightarrow 0 : v = -ke$

$$u = u_d - ke$$

$$\dot{X} = AX + Bu$$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + N(q) = \tau.$$

Let $\tau = M(q)a_q + C(q, \dot{q})\dot{q} + N(q)$. *Feedback linearization.*

Since the inertia matrix is invertible, the system dynamical model reduces to

$$\cancel{M(q)\ddot{q}} + \cancel{C(q, \dot{q})\dot{q}} + \cancel{N(q)} = \cancel{M(q)a_q} + \cancel{C(q, \dot{q})\dot{q}} + \cancel{N(q)}$$

$$\ddot{q} = a_q$$

new input

$$\ddot{q}_i = a_{q_i}$$

$$\vec{X} = \begin{bmatrix} q \\ \dot{q} \end{bmatrix}$$

$$\dot{\vec{X}} = \begin{bmatrix} 0 & I_{nm} \\ 0 & 0 \end{bmatrix} \vec{X} + \begin{bmatrix} 0 \\ I_n \end{bmatrix} a_q$$

$$\dot{q} = \dot{X}_1 = X_2$$

$$\dot{X}_2 = \ddot{q} = a_q$$

$$n: q \in \mathbb{R}^n$$

Inverse dynamic control

Given

$$\ddot{q} = a_q$$

decoupled joint dynamics.

Write into the state space form:

Inverse dynamic control

Let $q^d(t)$ be the desired trajectory (cont. differentiable at least twice)
the feedforward control input a_q^d should be

$$\ddot{q}^d = a_q^d$$

$$\dot{\vec{X}}^d = A\vec{X}^d + Ba_q^d$$

Traj. tracking of linear system

$$\vec{X} = \begin{bmatrix} q \\ \dot{q} \end{bmatrix} = \begin{bmatrix} q_1 \\ q_2 \\ \vdots \\ q_n \\ \dot{q}_1 \\ \dot{q}_2 \\ \vdots \\ \dot{q}_n \end{bmatrix}$$

$$e = \vec{X} - \vec{X}^d$$

$$a_q = a_q^d - ke$$

$$\tau = M(q)a_q + C(q, \dot{q})\dot{q} + N(q)$$

$$= M(q)[\underbrace{a_q^d - ke}] + C(q, \dot{q})\dot{q} + N(q)$$

$$a_q^d - \underbrace{k_p(q - q^d)}_{k_p(q - q^d)} - \underbrace{k_D(\dot{q} - \dot{q}^d)}_{k_D(\dot{q} - \dot{q}^d)}$$

k_p, k_D positive, def.

$$A = \begin{bmatrix} 0 & I \\ 0 & 0 \end{bmatrix} \quad B = \begin{bmatrix} 0 \\ I \end{bmatrix} \left(\begin{bmatrix} 1 & k_p & k_d \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} \right)$$

$$A - BK = \begin{bmatrix} 0 & I \\ -k_p & -k_d \end{bmatrix}$$

Inverse dynamic control with integral action

$$\tau = M(q)\ddot{q} + C(q, \dot{q})\dot{q} + N(q)$$

$$a_q = \ddot{q}_d + K_D(\dot{q}_d - \dot{q}) + K_P(q_d - q) + K_I \int_0^t (q_d - q) d\tau$$

lead to the error equation:

$$e = q - q_d$$

$$e^{(3)} + K_D \ddot{e} + K_P \dot{e} + K_I e = 0$$

$$K_x = \begin{bmatrix} K_{x1} & & 0 \\ & K_{x2} & \\ 0 & & \ddots \\ & & & K_{xm} \end{bmatrix}$$

$x \in \{D, P, I\}$

which is exponentially stable with a choice of K_P, K_D, K_I . — why?

$$e_i = q_i - q_{d,i} \in \mathbb{R}$$

$$\Rightarrow \boxed{e_i^{(3)} + k_{Di} \ddot{e}_i + k_{Pi} \dot{e}_i + k_{Ii} e_i = r_i(t)}$$

Frequency Domain

$$s^3 E_i(s) + k_{Di} s^2 E_i(s) + k_{Pi} s E_i(s) + k_{Ii} E_i(s) = R_i(s)$$

$$E_i(s) = \frac{1}{s^3 + k_{Di} s^2 + k_{Pi} s + k_{Ii}} R_i(s) \Rightarrow \lim_{t \rightarrow \infty} e_i(t)$$

$$\Rightarrow \lim_{t \rightarrow \infty} e_i(t)$$

$$e_i^{(3)} = -K_D \ddot{e}_i - K_P \dot{e}_i - K_I e_i$$

$$\lim_{s \rightarrow 0} E_i(s) = \lim_{s \rightarrow 0} \frac{R(s)}{s^3 + K_D s^2 + K_P s + K_I}$$

$$R(s) = C \quad \downarrow \quad \lim_{s \rightarrow 0} \frac{C}{s^3 + K_D s^2 + K_P s + K_I} = \lim_{s \rightarrow 0} \frac{C}{K_I}$$

$C=0 \Rightarrow$ choose K_D, K_P, K_I to make the system stable.

$$\Rightarrow \text{Time domain: } \hat{z} = \begin{bmatrix} e_i \\ \dot{e}_i \\ \ddot{e}_i \end{bmatrix}$$

$$\dot{\hat{z}} = A \hat{z}$$

$$= \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -K_I & -K_P & -K_D \end{bmatrix} \hat{z}$$

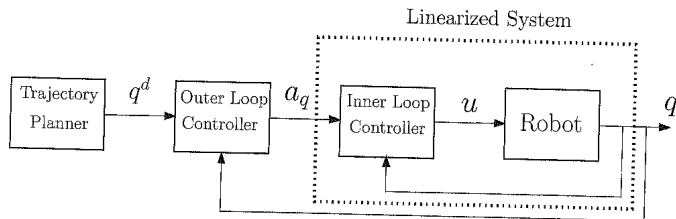
$$\dot{e}_i = \dot{z}_1 = z_2$$

$$\ddot{e}_i = \dot{z}_2 = z_3$$

$$e_i^{(3)} = \dot{z}_3 = -K_D z_3 - K_P z_2 - K_I z_1$$

Inverse dynamic control

the control architecture



- Outer: input $q^d, \dot{q}^d, q, \dot{q}$, output a_q
- Inner: input q, \dot{q}, a_q , and output u (or τ).

NOTE: This controller needs **exact knowledge** of the model parameters and requires an additional number of numerical computations.

Lyapunov-based control

introduce a change of coordinates:

$e = q - q_d$ — the tracking error.

$$\dot{\xi} = \dot{q}_d - \Lambda(q - q_d) = \dot{q}_d - \Lambda e;$$

$$\sigma = \dot{q} - \dot{\xi} = \dot{e} + \Lambda e; \quad = \dot{q} - \dot{q}_d + \Lambda e = \underline{\dot{e}} + \Lambda e$$

where Λ — a constant, positive definite square matrix.

Consider the following controller:

$$u = \underbrace{M(q)\ddot{\xi} + C(q, \dot{q})\dot{\xi} + N(q)}_{\text{desired dynamics}} - \underbrace{K_D \sigma}_{\text{damping}}$$

where K_D — a constant, positive definite matrix.

The closed-loop system is

$$M(q)\ddot{e} + C(q, \dot{q})\dot{e} + N(q) = M(q)\ddot{\xi} + C(q, \dot{q})\dot{\xi} + N(q) - K_D \sigma$$

$$M(q)(\ddot{e} - \ddot{\xi}) + C(q, \dot{q})(\dot{e} - \dot{\xi}) + K_D \sigma = 0$$

$$M(q)\ddot{e} + C(q, \dot{q})\dot{e} + K_D \sigma = 0$$

further, we simplify the closed loop system with the definition of $\dot{\xi}$ and σ :

Control: Key idea: If $\sigma \rightarrow 0$ as $t \rightarrow \infty$, then $e \rightarrow 0$

$$\underline{\sigma = e + \Lambda \dot{e}}$$

Proof of stability: The Lyapunov candidate :

$$V = \frac{1}{2} \sigma^T M(q) \sigma + e^T P e$$

where $P = P^T = \Lambda^T K_D > 0$. $\Rightarrow K_D$ is diagonal matrix. Λ : constant pos. def matrix.
show this candidate is a valid Lyapunov function.

$\Lambda^T K_D$ symmetric matrix

$$P = \Lambda^T K_D \\ P^T = K_D^T \Lambda \Rightarrow \Lambda^T K_D = K_D^T \Lambda$$

$$\underline{M(q) \dot{\sigma}} + C(q, \dot{q}) \sigma + K_D \sigma = 0$$

$$V \geq 0. \quad \text{and } V=0. \quad \text{if } \delta=0, \quad e=0$$

$$\begin{aligned}
 \dot{V} &= \frac{\partial}{\partial \delta} \left(\frac{1}{2} \delta^T M(q) \delta \right) \cdot \frac{d\delta}{dt} + \frac{\partial}{\partial e} (e^T P e) \cdot \dot{e} + \frac{1}{2} \delta^T \dot{M}(q) \delta \\
 &= \frac{1}{2} (\delta^T \dot{M}^T(q) + \delta^T \dot{M}(q)) \cdot \frac{d\delta}{dt} + (e^T P^T + e^T P) \cdot \dot{e} + \frac{1}{2} \delta^T \dot{M}(q) \delta \\
 &= \delta^T M(q) \ddot{\delta} + 2e^T P \dot{e} + \frac{1}{2} \delta^T \dot{M}(q) \delta \\
 &= \delta^T (-K_D \delta - C(q, \dot{q}) \delta) + 2e^T P \dot{e} + \frac{1}{2} \delta^T \dot{M}(q) \delta \\
 &= \frac{1}{2} \delta^T (\dot{M}(q) - 2C(q, \dot{q})) \delta - \delta^T K_D \delta + 2e^T P \dot{e} \quad \Leftarrow \delta = \dot{e} + \eta e \\
 &= (\dot{e} + \eta e)^T (-K_D) (\dot{e} + \eta e) + 2e^T \Lambda^T K_D \dot{e} \quad \Leftarrow p = \Lambda^T K_D \\
 &= -\dot{e}^T K_D \dot{e} - e^T \Lambda^T K_D \eta e + (-2e^T \Lambda^T K_D \dot{e}) + 2e^T \Lambda^T K_D \dot{e} \\
 &= -\dot{e}^T K_D \dot{e} - e^T \Lambda^T K_D \eta e \leq 0 \\
 \dot{V} = 0 &\Rightarrow \dot{e} = 0 \quad \text{and} \quad e = 0
 \end{aligned}$$

Passivity-based control: A general theorem

$$\int_{t=0}^T -\dot{q}^T \tau(s) ds \geq -\beta.$$

$$M(q)\dot{\phi} + C(q, \dot{q})\phi + K_D\phi = 0. \quad \text{special case}$$

when $\tau = 0$
and $\phi = 0$

Consider the differential equation

$$M(q)\dot{r} + C(q, \dot{q})r + K_V r = \Phi$$

where $K_V = K_V^T > 0$ is a positive definite matrix.

Suppose $\int_{t=0}^T -r^T(t)\Phi(t)dt \geq -\beta$ for all $T > 0$ and for some $\beta \geq 0$, then as $t \rightarrow \infty$, $r(t) \rightarrow 0$.

Proof

$$M(q)\dot{r} + C(q, \dot{q})r + k_v r = \phi \quad \star$$

Consider a Lyapunov candidate V defined by

$$V = \underbrace{\frac{1}{2} r^T M(q) r}_{\geq 0} + \underbrace{\beta - \int_{t=0}^T r^T(t) \phi(t) dt}_{\geq 0 \text{ passingy}}$$

Clearly, $V \geq 0$.

Differentiate V along the system traj.

$$\dot{V} = \underbrace{r^T M(q) \dot{r}} + \frac{1}{2} r^T \dot{M}(q) r - r^T \phi$$

$$= r^T (\phi - C(q, \dot{q})r - k_v r) + \frac{1}{2} r^T \dot{M}(q) r - r^T \phi$$

$$= \underbrace{r^T (-C(q, \dot{q}))r}_{\text{cancel}} - \underbrace{r^T k_v r}_{\text{cancel}} + \frac{1}{2} r^T \dot{M}(q) r$$

$$= -r^T k_v r \leq 0. \quad \text{and } \dot{V}=0 \text{ iff } r=0 \Rightarrow \begin{cases} e=0 \\ \dot{e}=0 \end{cases}$$

Insight for passivity-based control

Let $q^d(t)$ be a twice differentiable function, define

$$e(t) = q(t) - q^d(t).$$

If the following condition holds: $r(t) = \cancel{f(e(t))}$ for some proper mapping $f(\cdot)$ such that $r(t) \rightarrow 0$ means $\underline{e(t) \rightarrow 0}$ and $\underline{\dot{e}(t) \rightarrow 0}$, then select some control to make the system behaves like

$$M(q)\dot{r} + C(q, \dot{q})r + K_v r = \Phi$$

$$\begin{aligned} t &\rightarrow \infty \\ r &\rightarrow 0 \end{aligned}$$

$$\begin{aligned} &\xrightarrow{r = f(e, \dot{e})} \\ &\leftarrow ? \rightarrow \begin{cases} e \rightarrow 0 \\ \dot{e} \rightarrow 0 \end{cases} \end{aligned}$$

Passivity based motion control

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + N(q) = \tau \quad \dots (1)$$

select τ and a definition of r to make the system dynamics as

$$M(q)\dot{r} + C(q, \dot{q})r + K_v r = 0 \quad \dots (2) \quad r = f(e, \dot{e})$$

Note: A special case for $\Phi = 0$. Let $\tau = M(q)a + C(q, \dot{q})v + N(q) - K_v r$

$$M(q)\ddot{q} + C(q, \dot{q})\dot{q} + N(q) = M(q)a + C(q, \dot{q})v + N(q) - K_v r$$

$$M(q)(\ddot{q} - a) + C(q, \dot{q})(\dot{q} - v) + K_v r = 0 \quad \dots (3)$$

$$\begin{aligned} \dot{r} &= \ddot{q} - a \\ \boxed{r} &= \dot{q} - v \end{aligned} \Rightarrow \begin{cases} a = \ddot{q} - \dot{r} \\ v = \dot{q} - r \end{cases}$$

$r = f(e, \dot{e})$ such that $r \rightarrow 0, e \rightarrow 0, \dot{e} \rightarrow 0$.

$$= \dot{e} + \lambda e$$

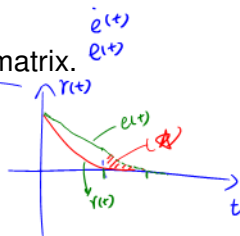
What is the mapping $r = f(e, \dot{e})$?

Select $v = \dot{q}^d - \Lambda e$ where $\Lambda > 0$ is a positive definite matrix.

$$r = \dot{e} + \Lambda e$$

$$\boxed{r \rightarrow 0 \quad \text{and} \Rightarrow \quad \dot{e} \rightarrow 0 \quad e \rightarrow 0} \quad \checkmark$$

$r \rightarrow 0 \quad \dot{r} = 0$



$M(q)\ddot{r} + C(q, \dot{r})\dot{r} + K_r r = 0 \rightarrow$ Lyapunov to show that

$$\lim_{t \rightarrow \infty} r = 0 \quad \text{and} \quad \dot{r} = 0$$

$$\begin{cases} \dot{r} = \ddot{e} + \Lambda \dot{e} = 0 \\ r = \dot{e} + \Lambda e = 0 \end{cases} \Rightarrow \begin{cases} \ddot{e} + \Lambda \dot{e} = 0 \\ \dot{e} + \Lambda e = 0 \end{cases} \quad \checkmark$$

$$\vec{x} = \begin{bmatrix} e \\ \dot{e} \end{bmatrix}$$

$$\dot{\vec{x}} = \begin{bmatrix} \dot{e} \\ \ddot{e} \end{bmatrix} = \begin{bmatrix} -\Lambda e \\ -\Lambda \dot{e} \end{bmatrix} = \underbrace{\begin{bmatrix} -\Lambda & 0 \\ 0 & -\Lambda \end{bmatrix}}_A \begin{bmatrix} e \\ \dot{e} \end{bmatrix} = A \vec{x}$$

$$\lim_{t \rightarrow \infty} \vec{x}(t) = 0$$

What happens when $r \equiv 0$?

Conclusion

- Simplification of dynamic model (a simplified inertia matrix) for ease the computation with computed torque control is studied in [1].
- Lyapunov-based control [2].
- Passivity-based control [3].

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