

# Tail 687 Cruise Efficiency — Problem Statement

**DATA 5100 — Fall 2025 — Group Project**

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## Executive Summary

We analyze NASA DASHlink flight-recorder data for Aircraft **Tail 687** to identify **actionable** levers that reduce **cruise fuel burn**. Focusing on altitude–speed choices under real winds and weight, we will produce interpretable, operations-ready recommendations (e.g., efficient altitude–Mach bands by weight and along-track wind) rather than generic correlations.

## Research Question & Objectives

### Primary question:

How do **altitude** and **airspeed** (Mach/TAS) combinations affect fuel efficiency during **steady-state cruise** for Tail 687, and what **cruise profiles** are optimal under varying **wind** and **aircraft weight**?

### Objectives:

1. Quantify marginal effects on total fuel flow (FF<sub>total</sub>) from altitude, Mach, along-track wind, angle of attack, and weight.
2. Map efficient operating bands (Altitude×Mach) across wind/weight regimes.
3. Screen engine-level dispersion (FF vs. N1/EGT) for potential maintenance/efficiency issues.

## Data & Analytical Sample

- **Source:** NASA DASHlink (Tail 687), 2012: **652** flights, **186** parameters, mixed sampling rates.
- **Cruise-capable flights:** **312** flights reached > **25,000 ft**; we include all 312 to leverage maximum statistical power.
- **Sampling strategy:** Restrict to **4 Hz** signals to avoid interpolation and ensure perfect time alignment.

- **Cruise definition:** Altitude  $> 25,000$  ft;  $|\text{altitude rate}| \leq 500$  ft/min; persistent windows (exclude transient level-offs — meaning we **remove short periods where the aircraft briefly levels off** (stops climbing or descending for a moment) but does not remain in steady cruise).
- We only include segments where the aircraft is consistently at cruise altitude and stable, not just passing through or pausing briefly before changing altitude again.
- **Final analytical set:** ~1,882,573 cruise records (~130 hr at 4 Hz).
- **Outliers checks:** Alt 25–35 kft; Mach 0.512–0.748 (typ. 0.69–0.72); mean N1  $\approx 91\%$  ( $\sigma \approx 3\%$ ); EGT  $\approx 557$  °C ( $\sigma \approx 22$  °C).
- **Cleaning:** Removed ~0.4% boundary points (FF outliers consistent with climb/descent bleed-through). **No missingness** in selected 4 Hz variables.

## Variables & Feature Engineering

**Target.**  $\text{FF\_total (lbs/hr)} = \text{FF}_1 + \text{FF}_2 + \text{FF}_3 + \text{FF}_4$ . (We will check engine-level dispersion for asymmetries.)

**Predictors (initial set):**

- **Engine performance:** N1 (avg), N2 (avg), EGT (avg).
- **Flight conditions & controls:** Pressure altitude, Mach, TAS, corrected angle of attack (AOAC),  $\text{PLA}_1 \dots \text{PLA}_4$ , N1T/N1C.
- **Environment:** Decompose winds into **along-track** (head/tail) and **cross-track** components.
- **Aircraft state (derived):** **Weight(t)** via initial fuel quantity minus integrated  $\text{FF\_total}$  (synchronizes 1 Hz to 4 Hz).

**Collinearity & interactions:**

We will investigate to diagnose with correlations/VIF; prefer parsimonious sets (e.g., Mach over TAS if redundant). Include **Altitude**×**Mach**, **Weight**×**Mach**, **Along-wind**×**Mach** to capture regime dependence.

## Methods & Inference Plan

**Exploratory Data Analysis:**

- Heatmaps/contours of  $\text{FF\_total}$  vs. (Altitude, Mach).

- Partial-residual diagnostics for AOAC, along-wind, weight.
- Per-engine FF vs. N1/EGT to flag asymmetric loads.

### Modeling (inference-first):

- Multiple linear regression with interpretable coefficients (e.g., +0.01 Mach  $\rightarrow$  + $\Delta$ FF\_total at fixed Altitude/Wind/Weight).
- Sensitivity across alternative specifications (N1 vs. PLA vs. N2).
- Robust SEs for time correlation (cluster-robust/HAC); optional altitude-band fixed effects.

### Interpretation:

Our plan is to report a **practical effect sizes** (lbs/hr, % change) and **decision charts** (efficient Altitude $\times$ Mach bands by wind/weight) with uncertainty. We will go through in detail in the notebook.

**Scope note.** We are aware that observational data are limited to causal claims (ATC constraints, comfort, fuel policies unobserved). We aim for physics-consistent associations and operations-testable guidance.

## Expected Deliverables

1. **Cruise efficiency maps:** Altitude $\times$ Mach grids predicting FF\_total, stratified by **weight** and **along-track wind**.
  - Stratified means the cruise efficiency maps (Altitude  $\times$  Mach grids predicting total fuel flow) are separated into groups based on aircraft weight and along-track wind. This would allow us to identify how optimal cruise settings change depending on these factors.
2. **Marginal-effects table** (95% CIs) for key levers (Altitude, Mach, AOAC, along-wind, Weight).
3. **Engine-health snapshot:** per-engine FF vs. N1/EGT dispersion.
4. **Operational playbook:** concrete targets (e.g., “At 29–31 kft and weight W, fly Mach 0.70–0.71 when headwind  $\geq$  X kt”).

## Risks & Limitations

- **Endogeneity/confounding:** route/ATC/payload; partially mitigated by conditioning on weight/wind and steady-state cruise filter.
- **Generalizability:** Tail- and era-specific.
- **Sensors/rates:** 4 Hz restriction removes some high-rate dynamics but preserves core cruise physics.
- **Actionability:** Recommendations require operational validation.

## Project Plan & Roles

Two meetings/week: (1) planning/strategy; (2) coding/QA. Minutes shared with instructor ([fischer9@seattleu.edu](mailto:fischer9@seattleu.edu)). Day-to-day via GitHub (issues/PRs).

### Initial sprint (3 tasks):

- **Task 1 — .mat data dictionary (Hemant & Duy):** enumerate fields (data, Rate, Units, Description); export CSV dictionary.
- **Task 2 — 4 Hz master table (Prithika & Duy):** finalize variable list; extract aligned 4 Hz signals per flight; stack; add flight\_id.
- **Task 3 — Cruise filter & QA (Collaborative; domain review by Prithika):** apply filters; drop non-operational segments (e.g.,  $N1 < 15\%$ ); fix impossibles; produce clean EDA outputs.

## Tools & Reproducibility

- **Python:** pandas/NumPy (ETL), SciPy (utils), statsmodels & scikit-learn (models), matplotlib & seaborn (viz).
- **Repro:** requirements.txt, data dictionary CSV, deterministic ETL script; GitHub for code/reviews.
- **Compute:** In-memory pandas is sufficient for post-filter sizes.

## References

- Boeing Commercial Airplanes. (2024). The Boeing ecoDemonstrator Program [[Backgrounder](#)]. Boeing.

- Boeing. (2024). ecoDemonstrator Program. Boeing Sustainability. [[Link](#)]
- Matthews, B. (2012). Flight Data for Tail 687 [[Dataset](#)]. NASA DASHlink (C3).

## **Appendix: Key Quantities**

- 652 total flights; **312** with cruise > 25 kft.
- **~1.88 M** 4 Hz cruise records (**~130 hr**).
- Cruise ranges: Alt 25–35 kft; Mach 0.512–0.748; mean N1  $\approx$  91%; mean EGT  $\approx$  557 °C.
- Outlier removal: **~0.45%** boundary points.