

Dan Richards

Objectives

- Create unit tests for Chapter 3, 4, & 5
 - More efficient debugging & grading
 - Easy to update true values

Library



Format

- Unit Test folder
- Unit tests for each chapter
 - Test files—names contain "test"
 - True value files

Format

```
def test_update_Mz():
194
          # initialize elements of the architecture
195
196
          mav = MavDynamics(SIM.ts_simulation)
197
          # set state
198
          mav__state = np.zeros((13,1),dtype=float)
199
          mav._state[3,0] = 25.0 # set u0
          mav._state[6,0] = 1.0 # set e0
200
201
          # set forces and moments
          forces_moments = np.zeros((6,1),dtype=float) # fx, fy, fz, Mx, My, Mz
202
203
          forces_moments[5,0] = 0.1 #set Mz
          # get update results
204
          mav_update(forces_moments) # propagate the MAV dynamics
205
206
          #compare against true values
          assert np.allclose(mav._state, trueValues.state_Mz)
```

```
state_Mz = np.array(
                              [[ 2.50000000e-01],
122
                               [-6.69747058e-20],
                               [-1.25415774e-14],
123
124
                               [ 2.50000000e+01],
                               [-7.17806614e-05],
125
126
                               [ 3.04437995e-11],
                               [ 1.00000000e+00],
127
128
                               [ 2.09665008e-07],
                               [ 3.07878132e-13],
129
                               [ 1.43561323e-06],
130
                                8.38660032e-05],
131
                               [ 2.46302506e-10],
132
                               [ 5.74245291e-04]])
133
```

oytest Fixtures

return mav, delta, wind

40

```
@pytest.fixture # create variables that can be used in each test function
    def sim():
23
         # initialize mav
24
25
         mav = MavDynamics(SIM.ts_simulation)
26
         # set state
27
         mav._state = np.zeros((13,1),dtype=float)
         mav._state[3,0] = 25.0 # set u0
28
                                                           def test_motor thrust_torque(sim):
                                                     46
29
         mav._state[6,0] = 1.0 # set e0
                                                             mav, delta, wind = sim
                                                     47
30
         mav_{va} = 25.0
                                                               delta.throttle = 0.9
                                                     48
31
         # initialize inputs
                                                               thrust, torque = mav._motor_thrust_torque(delta.throttle)
                                                     49
         delta = MsgDelta()
32
                                                               assert np.isclose(thrust, trueValues.thrust)
                                                     50
         delta.elevator = 0.0
33
                                                               assert np.isclose(torque, trueValues.torque)
                                                     51
34
         delta.aileron =
35
         delta.rudder =
                           0.0
         delta.throttle = 0.7
36
37
38
         # initialize wind
         wind = np.zeros((6.1))
39
```

Chapter 3 Tests

- chap3.MavDynamics._derivatives
 - f_X , f_y , f_z , M_X , M_y , M_Z
- chap3.MavDynamics.update
 - f_X , f_y , f_z , M_X , M_y , M_Z

Chapter 4 Tests

- chap4.MavDynamics._motor_thrust_torque
 - delta.throttle
- chap4.MavDynamics._forces_moments
 - delta.aileron, delta.elevator, delta.rudder
- chap4.MavDynamics._update_velocity_data
 - Wind N, E, S, W

Chapter 5 Tests

- chap5.trim.compute_trim
- chap5.compute_model.f_euler
- chap5.compute_model.df_dx
- chap5.compute_model.df_du
- chap5.compute_model.compute_ss_model
- chap5.compute_model.compute_tf_model

How to Test

Run in terminal using commands:

```
cd /Users/mavsim_python
pytest -q unit_tests/unitTestsChap4.py
```

Pass

```
test_df_du
   def test df du():
      # initialize elements of the architecture
      mav = MavDynamics(SIM.ts_simulation)
      Va = 25.
      gamma = 0.*np.pi/180.
      trim_state, trim_input = compute_trim(mav, Va, gamma)
      mav._state = trim_state
      x_{euler} = np.array([[-5.95888304e-15]],
                       [ 0.00000000e+00],
                       [-1.00000000e+02],
                       [ 2.49687427e+01],
                       [ 0.00000000e+00],
                       [ 1.24975516e+00]
                       [ 0.00000000e+00],
                       [ 5.00109104e-02],
                       [ 0.00000000e+00]
                       [ 0.00000000e+00],
                       [ 0.00000000e+00],
                      [ 0.00000000e+00]])
      B = cm.df_du(mav, x_euler, trim_input)
      assert np.allclose(B, trueValues.B)
      assert False
       [ 0.00000000e+00, 0.00
                                                                                                                                    [0., 0., 0., 0.]
                            [0., 0., 0., 0.],\n [0., 0., 0.... [0., 0., 0., 0.],\n
                                                                                   [0., 0., 0., 0.],\n
                                                                                                         [0., 0., 0., 0.],\n
                                                                                                                              [0., 0., 0., 0.],\n
       [0., 0., 0., 0.],\n
 [0., 0., 0., 0.]]))
          where <function allclose at 0x10e02b8b0> = np.allclose
          and array([[0., 0., 0., 0.],\n [0., 0., 0., 0.],\n
                                                                                     [0., 0., 0., 0.],\n
                                                                                                           [0., 0., 0.... [0., 0., 0., 0.],\n
                                                                [0., 0., 0., 0.],\n
                                                                                                                                             [0., 0
                                    [0., 0., 0., 0.],\n [0., 0., 0., 0.]]) = trueValues.B
 0., 0.],\n
               [0., 0., 0., 0.],\n
 nit_tests/unitTestsChap5.py:122: AssertionError
                                                                       - Captured stdout call --
Optimization terminated successfully (Exit mode 0)
         Current function value: 2.586543198071751e-06
         Iterations: 13
         Function evaluations: 255
         Gradient evaluations: 13
elevator= -0.1247780701597401 aileron= 0.0018361809638628682 rudder= -0.000302608037876341 throttle= 0.6767522859047431
trim_state= [[-5.95888304e-15 0.00000000e+00 -1.00000000e+02 2.49687427e+01
  0.00000000e+00 1.24975516e+00 9.99687380e-01 0.00000000e+00
  2.50028494e-02 0.00000000e+00 0.00000000e+00 0.00000000e+00
 FAILED unit_tests/unitTestsChap5.py::test_df_dx - assert False
FAILED unit_tests/unitTestsChap5.py::test_df_du - assert False
 failed, 4 passed in 1.32s
```

```
def test_df_du():
      # initialize elements of the architecture
     mav = MavDynamics(SIM.ts_simulation)
     Va = 25.
     gamma = 0.*np.pi/180.
      trim_state, trim_input = compute_trim(mav, Va, gamma)
     mav._state = trim_state
     x_{euler} = np.array([[-5.95888304e-15],
                    [ 0.00000000e+00],
                    [-1.00000000e+02],
                    [ 2.49687427e+01],
                    [ 0.00000000e+00],
                    [ 1.24975516e+00],
                    [ 0.00000000e+00],
                    [ 5.00109104e-02],
                    [ 0.00000000e+00],
                    [ 0.00000000e+00],
                    [ 0.00000000e+00],
                    [ 0.00000000e+00]])
     B = cm.df_du(mav, x_euler, trim_input)
      assert np.allclose(B, trueValues.B)
      assert False
                                   00...20e-06,\n
               -9.87160320e-06],\n
      [0., 0., 0., 0.],\n [0., 0., 0., 0.],\n
[0., 0., 0., 0.]]))
     + where <function allclose at 0x10e02b8b0> = np.allclose
+ and array([[0., 0., 0., 0.],\n [0., 0., 0.],\n [0., 0., 0.],\n
                                                                          [0., 0., 0., 0.],\n [0., 0., 0... [0., 0., 0., 0.],\n
                                                                                                                                [0., 0.]
, 0., 0.], n = [0., 0., 0., 0.], n = [0., 0., 0.], n = [0., 0., 0., 0.]) = trueValues.B
unit_tests/unitTestsChap5.py:122: AssertionError
                                 Optimization terminated successfully
                             (Exit mode 0)
        Current function value: 2.586543198071751e-06
        Iterations: 13
        Function evaluations: 255
        Gradient evaluations: 13
elevator= -0.1247780701597401 aileron= 0.0018361809638628682 rudder= -0.000302608037876341 throttle= 0.6767522859047431
trim_state= [[-5.95888304e-15 0.00000000e+00 -1.00000000e+02 2.49687427e+01
  0.00000000e+00 1.24975516e+00 9.99687380e-01 0.00000000e+00
  2.50028494e-02 0.00000000e+00 0.0000000e+00 0.00000000e+00
  0.00000000e+00]]
                      FAILED unit_tests/unitTestsChap5.py::test_df_dx - assert False
FAILED unit_tests/unitTestsChap5.py::test_df_du - assert False
2 failed, 4 passed in 1.32s
```

test_df_du _

```
def test_df_du():
    # initialize elements of the architecture
    mav = MavDynamics(SIM.ts_simulation)
    Va = 25.
    gamma = 0.*np.pi/180.
    trim_state, trim_input = compute_trim(mav, Va, gamma)
    mav._state = trim_state
    x_{euler} = np.array([[-5.95888304e-15],
                        [ 0.00000000e+00],
                        [-1.00000000e+02],
                        [ 2.49687427e+01],
                        [ 0.00000000e+00],
                        [ 1.24975516e+00],
                        [ 0.00000000e+00],
                        [ 5.00109104e-02],
                        [ 0.00000000e+00],
                         [ 0.00000000e+00],
                         [ 0.00000000e+00],
                         [ 0.00000000e+00]])
    B = cm.df_du(mav, x_euler, trim_input)
    assert np.allclose(B, trueValues.B)
    assert False
```

```
test_df_du
   def test_df_du():
       # initialize elements of the architecture
      mav = MavDynamics(SIM.ts_simulation)
      Va = 25.
      gamma = 0.*np.pi/180.
      trim_state, trim_input = compute_trim(mav, Va, gamma)
      mav._state = trim_state
      x_{euler} = np.array([[-5.95888304e-15],
                        [ 0.00000000e+00],
                        [-1.00000000e+02],
                        [ 2.49687427e+01],
                        [ 0.00000000e+00],
                        [ 1.24975516e+00],
                        [ 0.00000000e+00],
                        [ 5.00109104e-02],
                        [ 0.00000000e+00],
                        [ 0.00000000e+00],
                        [ 0.00000000e+00],
                        [ 0.00000000e+00]])
      B = cm.df_du(mav, x_euler, trim_input)
       assert False
       0.00000000e+00],\n
                                                                                                                                       [ 0.00000000e+00, 0.00
00...20e-06,\n
                                                                                              -2.48813413e+01]]), array([[0., 0., 0., 0.],\n
                                                                                                                                             [0., 0., 0., 0.]
                              [0., 0., 0., 0.],\n [0., 0., 0.... [0., 0., 0., 0.],\n
       [0., 0., 0., 0.],\n
                                                                                        [0., 0., 0., 0.],\n
                                                                                                               [0., 0., 0., 0.],\n
                                                                                                                                      [0., 0., 0., 0.],\n
 [0., 0., 0., 0.]]))
           where <function allclose at 0x10e02b8b0> = np.allclose
and array([[0., 0., 0., 0.],\n [0., 0., 0., 0.],\n [0., 0., 0., 0.],\n
                                                                                           [0., 0., 0., 0.],\n [0., 0., 0... [0., 0., 0., 0.],\n
                                                                                                                                                      [0., 0
                [0., 0., 0., 0.], n [0., 0., 0., 0.], n [0., 0., 0., 0.]]) = trueValues.B
 0., 0.],\n
unit_tests/unitTestsChap5.py:122: AssertionError
Optimization terminated successfully
                                  (Exit mode 0)
          Current function value: 2.586543198071751e-06
          Iterations: 13
          Function evaluations: 255
          Gradient evaluations: 13
elevator= -0.1247780701597401 aileron= 0.0018361809638628682 rudder= -0.000302608037876341 throttle= 0.6767522859047431
trim_state= [[-5.95888304e-15 0.00000000e+00 -1.00000000e+02 2.49687427e+01
  0.00000000e+00 1.24975516e+00 9.99687380e-01 0.00000000e+00
  2.50028494e-02 0.00000000e+00 0.0000000e+00 0.00000000e+00
  0.00000000e+00]]
                         FAILED unit_tests/unitTestsChap5.py::test_df_dx - assert False
FAILED unit_tests/unitTestsChap5.py::test_df_du - assert False
2 failed, 4 passed in 1.32s
```

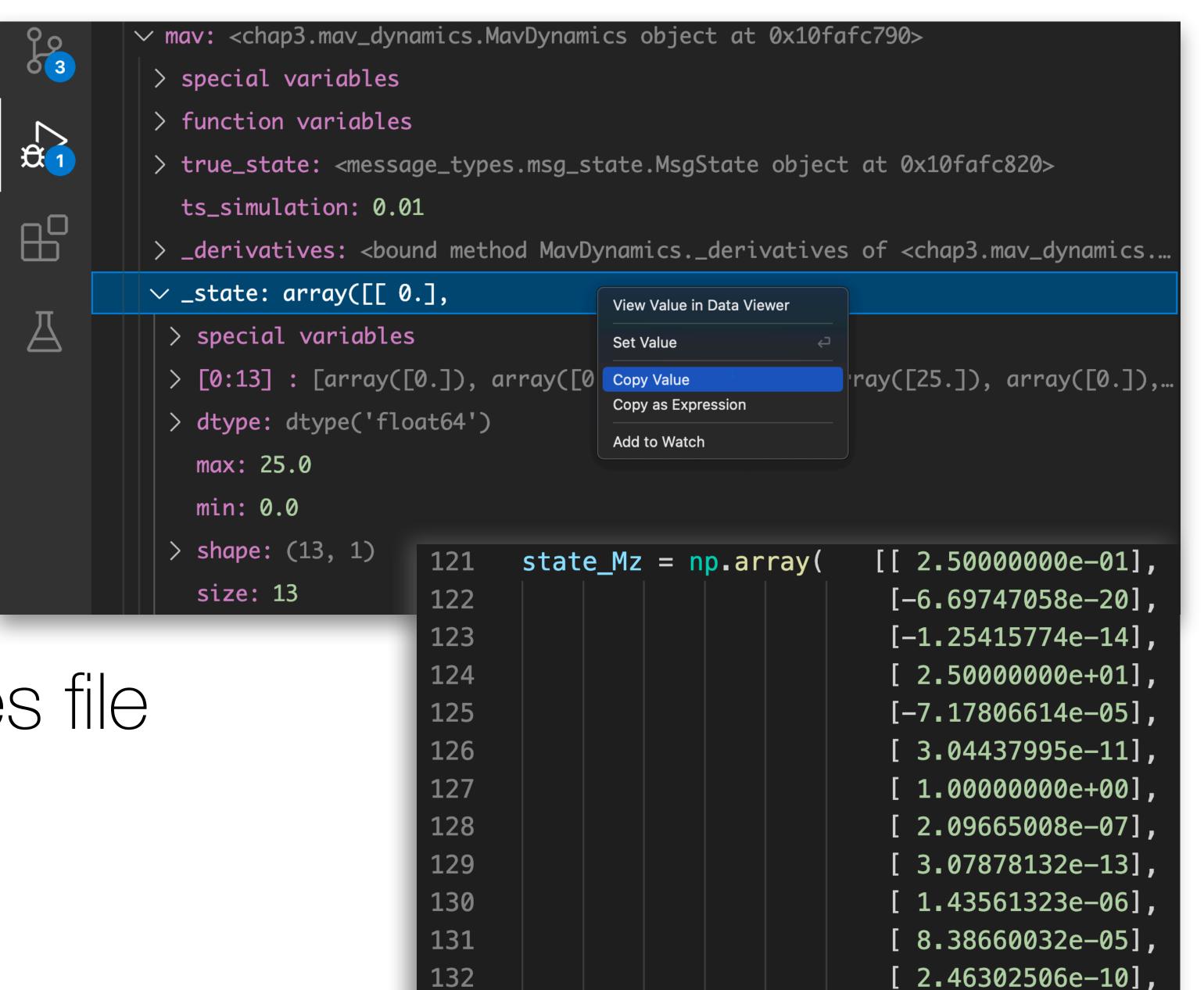
```
test_df_du _
   def test_df_du():
      # initialize elements of the architecture
      mav = MavDynamics(SIM.ts_simulation)
      Va = 25.
      gamma = 0.*np.pi/180.
      trim_state, trim_input = compute_trim(mav, Va, gamma)
      mav._state = trim_state
      x_{euler} = np.array([[-5.95888304e-15],
                     [ 0.00000000e+00],
                     [-1.00000000e+02],
                     [ 2.49687427e+01],
                     [ 0.00000000e+00],
                     [ 1.24975516e+00],
                     [ 0.00000000e+00],
                     [ 5.00109104e-02],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00]])
      B = cm.df_du(mav, x_euler, trim_input)
      assert np.allclose(B, trueValues.B)
      assert False
      00...20e-06,\n
                          [0., 0., 0., 0.],\n
                                                                                                   [0., 0., 0., 0.],\n
                                                                                                                      [0., 0., 0., 0.],\n
[0., 0., 0., 0.]]))
          where <function allclose at 0x10e02b8b0> = np.allclose
and array([[0., 0., 0., 0.],\n [0., 0., 0.],\n [0., 0., 0.],\n
                                                                             [0., 0., 0., 0.],\n [0., 0., 0.... [0., 0., 0., 0.],\n
                                                                                                                                     [0., 0.]
              [0., 0., 0., 0.], n [0., 0., 0., 0.], n [0., 0., 0., 0.]]) = trueValues.B
, 0., 0.],\n
      Optimization terminated successfully
                              (Exit mode 0)
         Current function value: 2.586543198071751e-06
         Iterations: 13
         Function evaluations: 255
         Gradient evaluations: 13
elevator= -0.1247780701597401 aileron= 0.0018361809638628682 rudder= -0.000302608037876341 throttle= 0.6767522859047431
trim_state= [[-5.95888304e-15 0.00000000e+00 -1.00000000e+02 2.49687427e+01
  0.00000000e+00 1.24975516e+00 9.99687380e-01 0.00000000e+00
  2.50028494e-02 0.00000000e+00 0.00000000e+00 0.00000000e+00
  0.00000000e+00]]
```

FAILED unit_tests/unitTestsChap5.py::test_df_dx - assert False
FAILED unit_tests/unitTestsChap5.py::test_df_du - assert False
2 failed, 4 passed in 1.32s

```
test_df_du _
   def test_df_du():
      # initialize elements of the architecture
     mav = MavDynamics(SIM.ts_simulation)
     Va = 25.
     gamma = 0.*np.pi/180.
      trim_state, trim_input = compute_trim(mav, Va, gamma)
     mav._state = trim_state
      x_{euler} = np.array([[-5.95888304e-15],
                     [ 0.00000000e+00],
                     [-1.00000000e+02],
                     [ 2.49687427e+01],
                     [ 0.00000000e+00],
                     [ 1.24975516e+00],
                     [ 0.00000000e+00],
                     [ 5.00109104e-02],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00],
                     [ 0.00000000e+00]])
     B = cm.df_du(mav, x_euler, trim_input)
     assert np.allclose(B, trueValues.B)
      assert False
      00...20e-06,\n
      [0., 0., 0., 0.],\n [0., 0., 0., 0.],\n [0., 0., 0., 0., 0., 0., 0.],\n [0., 0., 0., 0.],\n
                                                                                                 [0., 0., 0., 0.],\n [0., 0., 0., 0.],\n
[0., 0., 0., 0.]]))
     + where <function allclose at 0x10e02b8b0> = np.allclose
+ and array([[0., 0., 0., 0.],\n [0., 0., 0.],\n [0., 0., 0.],\n
                                                                            [0., 0., 0., 0.],\n [0., 0., 0.... [0., 0., 0., 0.],\n
                                                                                                                                  [0., 0.]
, 0., 0.], n = [0., 0., 0., 0.], n = [0., 0., 0.], n = [0., 0., 0., 0.]) = trueValues.B
unit_tests/unitTestsChap5.py:122: AssertionError
                                                   Optimization terminated successfully
                              (Exit mode 0)
         Current function value: 2.586543198071751e-06
         Iterations: 13
         Function evaluations: 255
         Gradient evaluations: 13
elevator= -0.1247780701597401 aileron= 0.0018361809638628682 rudder= -0.000302608037876341 throttle= 0.6767522859047431
trim_state= [[-5.95888304e-15 0.00000000e+00 -1.00000000e+02 2.49687427e+01
  0.00000000e+00 1.24975516e+00 9.99687380e-01 0.00000000e+00
  2.50028494e-02 0.00000000e+00 0.00000000e+00 0.00000000e+00
```

Update Values

- Debugger
- Breakpoint
- Copy Value
- Paste in true values file



133

[5.74245291e-04]])