

Objective function:

$$\boxed{L(\tilde{z}, \tilde{y}) \in \mathbb{R}}$$

$\tilde{z} \in \mathbb{R}^n$  State variable  
 $\tilde{y} \in \mathbb{R}^{m \times 1}$  Control variable

Static equality constraint:

$$\boxed{f(\tilde{z}, \tilde{y}) = 0_{n \times 1}}$$

Here  $f: \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^n$

Assume that both  $L(\tilde{z}, \tilde{y})$  &  $f(\tilde{z}, \tilde{y})$  poses continuous pts  
of upto  $\infty$  orden. wrt  $\tilde{z}$  &  $\tilde{y}$  (though we are interested in derivative  
only upto 2nd orden)

Thus far we have learnt about the "necessary condition" for  
minimization of  $L(\tilde{z}, \tilde{y})$  subject to the constraint  $f(\tilde{z}, \tilde{y}) = 0_{n \times 1}$

in terms of Hamiltonian function

$$H(\tilde{z}, \tilde{y}, \lambda) \rightarrow \mathbb{R}^n$$

$$= L(\tilde{z}, \tilde{y}) + (\lambda^T)_{m \times n} f(\tilde{z}, \tilde{y})$$

$$\text{if it is } H_{\tilde{z}} = 0_{n \times n} \text{ & } H_{\tilde{y}} = 0_{m \times m}$$

$$\begin{aligned} & H_{\tilde{z}} = 0_{n \times n} & H_{\tilde{y}} = 0_{m \times m} \\ & \tilde{z}^* + (\lambda^T)_{m \times n} f_{\tilde{z}} = 0_{n \times n} & \tilde{y}^* + (\lambda^T)_{m \times n} f_{\tilde{y}} = 0_{m \times m} \\ & f_{\tilde{z}} = \tilde{z}^* & f_{\tilde{y}} = \tilde{y}^* \end{aligned}$$

Now we will emphasize on the  
"Sufficient condition" for minimization of  $L(\tilde{z}, \tilde{y})$  subject to the  
constraint  $f(\tilde{z}, \tilde{y}) = 0_{n \times 1}$

$$\left( \frac{\partial^2 L}{\partial \tilde{a}^2} \right)_{\tilde{f}=0} > 0$$

@ Stationary pts,  $\tilde{z}^*, \tilde{y}^*$

then at  $(\tilde{z}^*, \tilde{y}^*)$   $L(\tilde{z}, \tilde{y})$  achieved  
minima subject to constraint  
 $f(\tilde{z}, \tilde{y}) = 0_{n \times 1}$

$$\left( \frac{\partial \chi}{\partial y^2} \right)_{f=0_{n \times n}} = \left\{ - \left( \underline{f}_2 \right)^T \left( \underline{f}_2^T \right)^{-1} \right\} \cdot \begin{pmatrix} H_{22} \\ H_{22} & H_{44} \end{pmatrix}$$

$\left\{ - \left( \underline{f}_2 \right)^T \left( \underline{f}_2^T \right)^{-1} \right\} > 0.$

$\left( \frac{\partial \chi}{\partial y^2} \right)_{f=0} = H_{44} - H_{22} \underline{f}_2^{-1} \underline{f}_2 - \underline{f}_2^T \left( \underline{f}_2^T \right)^{-1} H_{22}$ $+ \underline{f}_2^T \left( \underline{f}_2^T \right)^{-1} H_{22} \left( \underline{f}_2^{-1} \right) \underline{f}_2 > 0$
--

here

$$H_{22} = \frac{\partial}{\partial z} \left( \left( \frac{\partial H}{\partial z} \right)^T \right)_{n \times n} = \frac{\partial^2 H}{\partial z \partial z} \in R^{n \times n}$$

$$\frac{\partial H}{\partial z} = H_{24} = \frac{\partial^2 H}{\partial z \partial y} = \frac{\partial}{\partial z} \left( \left( \frac{\partial H}{\partial y} \right)^T \right) \in R^{n \times n}$$

$$H_{44} = \frac{H_{22}}{n \times n} = \frac{H_{22}}{n \times n}$$

i.e.  $H_{22} = \frac{H_{22}}{n \times n}$ ,  $H_{44} = \frac{H_{22}}{n \times n}$ ,  $H_{22} = \frac{H_{22}}{n \times n}$ ,  $H_{24} = \frac{H_{24}}{n \times n}$

$$f \underline{f}_2 = 1 \times n \quad \text{and} \quad \underline{f}_2 = n \times 1$$

Legendre  
- Clebsche  
Condition!

Q. problem: minimize  $\lambda(\tilde{z}, \tilde{y})$  w.r.t  $\tilde{y}$  subject to constraint  
 $f(\tilde{z}, \tilde{y}) = 0_{n \times 1}$

First find out the stationary pt.  $(\tilde{z}^*, \tilde{y}^*)$

second check whether  $\lambda(\tilde{z}, \tilde{y})$  is minimized at  $(\tilde{z}^*, \tilde{y}^*)$   
 satisfies the "sufficient condition" for minimization!

{ if it satisfies then  $(\tilde{z}^*, \tilde{y}^*)$  pt. of minima  
 else  $(\tilde{z}^*, \tilde{y}^*)$  is NOT the pt. of minima!

Step I: Objective function:  $\lambda(\tilde{z}, \tilde{y}) \in \mathbb{R}$

Equality constraint:  $f(\tilde{z}, \tilde{y}) = 0_{n \times 1} \in \mathbb{R}^{n \times 1}$   
 $\sim \in \mathbb{R}^{n \times 1}$

Step II: Check whether  $\sim$  has continuous pd's w.r.t  $\tilde{z}, \tilde{y}$   
 w.r.t  $\tilde{z} f' \sim$

Check "  $\lambda(\tilde{z}, \tilde{y})$ "

if : so then proceed Else Not

Step III: form the Hamiltonian,

$$H(\tilde{z}, \tilde{y}, \lambda) = \underbrace{\lambda(\tilde{z}, \tilde{y})}_{\in \mathbb{R}^{n \times 1}} + \lambda^T \cdot \underbrace{f(\tilde{z}, \tilde{y})}_{\text{Lagrangian}}$$

Step IV: at stationary pt.  $(\tilde{z}^*, \tilde{y}^*)$

$$\lambda_{\tilde{z}} = 0_{n \times n}, \quad \lambda_{\tilde{y}} = 0_{n \times n}$$

$$\lambda_{\tilde{z}} + \lambda^T \cdot \underline{f_{\tilde{z}}} = 0_{n \times n} \quad \lambda_{\tilde{y}} + \lambda^T \cdot \underline{f_{\tilde{y}}} = 0_{n \times n}$$

$$\lambda^T = -\lambda_{\tilde{y}} \underline{f_{\tilde{y}}}^{-1}$$

use these 8 eqns to find out  $\tilde{z}^*$  &  $\tilde{y}^*$ .  
 $(n+m=n)$

Step V: Now check for Legendre-Clebsche condition at  $(\bar{z}^*, \bar{y}^*)$ .  
 i.e.  $(\bar{z}^*, \bar{y}^*)$  will be the pt. of minima if

$$\left( \frac{\partial^2 L}{\partial q^2} \right)_{L=0} > 0.$$

$$\text{i.e. } H_{qq} - f_q^{-1} f_q = f_q^T (f_q^T)^{-1} H_{qq} f_q$$

$$= H_{qq} - f_q^{-1} f_q - f_q^T (f_q^T)^{-1} H_{qq} (f_q^{-1}) f_q > 0$$

$$\left. \begin{array}{l} H_{qq} \\ = H_{qq} \\ = H_{qq} \\ = H_{qq} \\ = H_{qq} \end{array} \right\}_{\substack{N \times N \\ M \times N \\ N \times N \\ N \times N \\ N \times N}}$$

$$\left. \begin{array}{l} f_q \\ = f_q \\ = f_q \\ = f_q \end{array} \right\}_{\substack{N \times 1 \\ N \times 1 \\ N \times 1 \\ N \times 1}}$$

$$\text{BUT } \left. \begin{array}{l} H_{qq} \\ \approx H_{qq} \\ \approx H_{qq} \\ \approx H_{qq} \end{array} \right\}_{\substack{N \times N \\ 1 \times 1 \\ 1 \times 1 \\ N \times N}}$$

$$+ \lambda + R^{N \times 1}$$

if Legendre-Clebsche condition is satisfied @  $(\bar{z}^*, \bar{y}^*)$   
 the pt. we have is the minima pt.  $(\bar{z}^*, \bar{y}^*)!!$

## ④ Inequality Constraint:

Objective function:  $\lambda(\tilde{z}, \tilde{y}) \in \mathbb{R}$

Inequality constraint:  $f(\tilde{z}, \tilde{y}) \leq \underset{\sim}{\rho}_{px} \quad \text{NOT } = 0$

(defines a feasible region in which the search for any minimum point is to be performed)

Such search requires a numerical procedure called Non-linear programming!

- if a minimum pt.  $(\hat{z}, \hat{y})$  exists
- then it has two possibility

either  $f(\hat{z}, \hat{y}) < \underset{\sim}{\rho}_{px}$  or  $f(\hat{z}, \hat{y}) = \underset{\sim}{\rho}_{px}$

in such case use unconstrained optimization!

- Consider the Augmented objective function:

$$J(\tilde{z}, \tilde{y}) = \lambda(\tilde{z}, \tilde{y}) + \underset{\sim}{\lambda}_{px} f(\tilde{z}, \tilde{y})$$

## ⑤ Necessary condition for minimization:

$$\frac{\partial L}{\partial \tilde{y}} + \lambda^T \frac{\partial f}{\partial \tilde{y}} = 0, \quad \text{when } \lambda = 0, \quad f(\tilde{z}, \tilde{y}) < 0$$

$$\lambda > 0, \quad f(\tilde{z}, \tilde{y}) = 0$$

The search

\* The Search for minimum in the presence of inequality constraints  
~~could yield~~ will result in any of the following:

i. Minima pt. lies inside the feasible region  
Both necessary & sufficient condition satisfied!

ii. Minima pt. lies on the boundary

Satisfies  
necessary  
But not "sufficient"  
Condition  
of minimization

does not  
Satisfy

"Necessary condition"  
of minimization

iii. apt. on the boundary of  
feasible region can be  
a minima pt. even if  
it does NOT satisfy the  
necessary conditions!!

$$0.8x_1 + 0.6x_2 = 1.6$$

$$0 = -0.8x_1 - 0.6x_2$$

Problem: minimize  $\lambda(x, y)$  subject to constraint  $f(x, y) \leq 0$

wrt  $y$

Step 1 - first find out the stationary pt. of  $\lambda(x, y)$

$$\cancel{\lambda(x^*, y^*)} =$$

$$\lambda_{yy} = \frac{\partial \lambda}{\partial y} = 0 \rightsquigarrow (x^*, y^*)$$

- Then consider only those stationary pts. which lie in the feasible region.

- find  $\lambda_{yy} = \lambda_{yy}$  if at  $(x^*, y^*) \lambda_{yy} > 0 \Rightarrow (x^*, y^*)$  is a pt. of minima.

if there is no stationary pts which lies in feasible region.

Then minima must occur on the boundaries  $\Rightarrow$  solve by inspection!

Consider a plant  $\dot{z} = f(z, u, t)$

$z \sim$  state variable

$u \sim$  control "

$$IC: z(t_i) = z_i \sim \text{Initial condition}$$

problem: determine the control history  $u(t)$ ,  $t_i \leq t \leq t_f$   
such that the following objective function  
 $J$  is minimized w.r.t  $u(t)$

given terminal  
(fixed)  
out.

$$J = \phi[z(t_f), t_f] + \int_{t_i}^{t_f} [H(z, u, \lambda, t) - \lambda^T \cdot \dot{z}] dt$$

Consider the Hamiltonian,

$$H(z, u, t, \lambda) = L(z, u, t) + \lambda^T \cdot f(z, u, t)$$

( $\lambda(t)$  is an additional  
variable called  
costate variable)

① Necessary condition for minimization of  $J$  w.r.t  $u(t)$

Step I  $\dot{\lambda}^T = -\frac{\partial H}{\partial z}$  the 1st order differential (costate)  
beginning of the BC to  
be solved for \*

Step II  $\frac{\partial H}{\partial u} = 0$  costate vector  
 $\lambda(t)$

Step IV  $\dot{z} = f(z, u, t) + \lambda^T(t_i) = z'_i$

for a generalized LTI system.

$$z(t) = e^{-At} z(t=0) + \int_0^t e^{-A(t-\tau)} B \cdot \lambda(\tau) d\tau$$

NOTE: No constraint on control variable  $g(t)$   
 NO other constraint on control interval  $t_i \leq t \leq t_f$   
 NO constraint on  $t_f$  (apart from the fact that it is fixed)  
 ↗ terminal time.

- Suppose  $\exists$  a trajectory  $\tilde{g}^*(t)$  with corresponding control history  $\tilde{g}^*(t)$ ,  $t_i \leq t \leq t_f$  which satisfies the necessary conditions for minimization.
- such a trajectory is called an extremal trajectory corresponding to the extremal control history.

$$g(t) = \tilde{g}^*(t) + \int_{t_i}^t B(t-t') \dot{g}(t') dt'$$

$$= \tilde{g}^*(t) + \int_{t_i}^t B(t-t') A(t-t') \dot{g}^*(t') dt'$$

$$= \tilde{g}^*(t) + e^{-A(t-t_i)} \tilde{g}^*(t=t_i)$$

for a generalized LTI system:

$$g(t) = e^{-A(t-t_i)} \tilde{g}^*(t=t_i) + \int_{t_i}^t e^{-A(t-t')} B(t-t') \dot{g}^*(t') dt'$$

where  $e^{-At} = \lambda^{-1} \left\{ (\delta - A)^{-1} \right\}$

LTI based on state space  
method

where  $\tilde{g}^*(t) = (A^{-1})^T v$

where  $v = (A^{-1})^T g(t_i)$

$$\tilde{g}^*(t) = (A^{-1})^T v + (A^{-1})^T (t-t_i) u$$

① Consider a dynamic plant governed by :

$$\dot{\underline{x}} = f(\underline{x}, \underline{u}, t) \quad \textcircled{i}$$

↑ State vector  
↓ control or I/P  
vector  
 $\in \mathbb{R}^{n \times 1}$

Let  $\underline{x}^*(t)$  → external trajectory

$\underline{u}^*(t)$  → corresponding control history

$$\dot{\underline{x}}^*(t) = f(\underline{x}^*(t), \underline{u}^*(t), t)$$

available from necessary condition, for minimisation

$$\text{of } J = \phi(\underline{x}(t_f), t_f)$$

$$\textcircled{ii} \quad + \int_{t_0}^{t_f} f_x(\underline{x}, \underline{u}, t) dt \quad \text{wrt } \underline{u}(t)$$

subject to I.C.  
 $\underline{x}(t_i) = \underline{x}_i$

$$t_i, t_f$$

✓ Consider small control & state deviation,

$$\left\{ \begin{array}{l} \underline{y}(t) = \underline{u}(t) - \underline{u}^*(t) ; t_0 \leq t \leq t_f \\ \underline{z}(t) = \underline{x}(t) - \underline{x}^*(t) ; t_0 \leq t \leq t_f \end{array} \right.$$

Linearising ① about  $\underline{x}^*(t)$  &  $\underline{u}^*(t)$  [assuming  $f(\cdot)$  to be  
differentiable]

$$\dot{\underline{z}}(t) = \underline{A}(t) \cdot \underline{z}(t) + \underline{B}(t) \cdot \underline{y}(t)$$

$$\text{f. } \underline{z}(t_0) = \underline{z}_0$$

$$\text{here } \underline{A}(t) = \frac{\partial f}{\partial \underline{x}} \Big|_{(\underline{x}^*, \underline{u}^*)}$$

$$\underline{B}(t) = \frac{\partial f}{\partial \underline{u}} \Big|_{(\underline{x}^*, \underline{u}^*)}$$

Linearizing (i) about  $\tilde{g}^*(t)$  &  $\tilde{y}^*(t)$

$$J(\tilde{x}, \tilde{y}) - J(\tilde{x}^*, \tilde{y}^*) \approx s^2 J(2, 2) = \hat{J}(2, 2)$$

$\phi(2(t), t)$

$$\hat{J}(2, 2) = \frac{1}{2} \tilde{z}^T(t) \cdot \Phi \cdot \tilde{z}(t)$$

(i)

$$+ \int_0^t \left\{ \begin{array}{l} \tilde{z}^T(t) \\ \tilde{y}^T(t) \end{array} \right\} \left\{ \begin{array}{l} \tilde{z}'(t) \\ \tilde{y}'(t) \end{array} \right\} \begin{bmatrix} \Phi(t) & S(t) \\ S^T(t) & R(t) \end{bmatrix} \begin{bmatrix} \tilde{z}(t) \\ \tilde{y}(t) \end{bmatrix} dt$$

$J(2, 2, t)$

$$\text{where } \Phi_f = \frac{\partial^2 \Phi}{\partial \tilde{x}^2} \Big|_{(\tilde{x}^*, \tilde{y}^*)}$$

$$= \frac{\partial^2 \lambda}{\partial \tilde{x}^2} \Big|_{(\tilde{x}^*, \tilde{y}^*)}$$

$$S(t) = \frac{\partial^2 \lambda}{\partial \tilde{x} \partial \tilde{y}} \Big|_{(\tilde{x}^*, \tilde{y}^*)}$$

$$R(t) = \frac{\partial^2 \lambda}{\partial \tilde{y}^2} \Big|_{(\tilde{x}^*, \tilde{y}^*)}$$

$$\Phi_f(t), S(t) \text{ & } R(t)$$

are symmetric, real, Sq. matrix.

$$H(2, 2, 2, t)$$

$$= L(2, 2, t)$$

$$+ \tilde{s}^2 J(2, 2, t)$$

our target is to minimize  $\hat{J}(2, 2)$  w.r.t  $\tilde{y}(t)$

where  $\tilde{z} \tilde{y} \tilde{y}$  satisfies (6)

for applying "necessary condition" for minimization of  $\hat{J}(2, 2)$  w.r.t  $\tilde{y}(t)$

We obtain Riccati eqn:

$$\dot{P} = -\Phi - \left( A - BR^{-1}S^T \right)^T P - P \left( A - BR^{-1}S^T \right) + P B R^{-1} B^T P + S^T R^{-1} S^T$$

$$P(t_f) = \Phi_f : BC$$

linear feedback law,  $y(t) = -R^{-1} [B^T P + S^T] z(t)$

• Riccati eqn reflects the "necessary condition" for existence of an optimal control law for linear system with quadratic performance index.

① "Sufficient Condition" for minimization of  $\hat{J}(z, y)$  w.r.t  $g(t)$  (in ②) (subject to ③)

$$H(z, y, \lambda, t) = \lambda[z, y, t] + \lambda^T f(z, y, t)$$

if  $H_{uu} \Big|_{(z^*(t), y^*(t))} > 0$  +ve definite

then the H.Eq is satisfied on the

extremal trajectory  $(z^*(t), y^*(t))$

if  $H_{uu} \Big|_{(z^*(t), y^*(t))} = 0$

Singular requires additional constraints on  $z(t)$  /  $y(t)$  to determine optimal trajectory!

- for an LQR system ( $\textcircled{1}$  +  $\textcircled{2}$ ) the "sufficient condition" becomes

$$P(t) \geq 0.$$

$\frac{\partial}{\partial t}$

$$\boxed{P(t) > 0 \text{ & } t_0 \leq t \leq t_f.}$$

### ① Steady State Riccati Solution:

Consider an LTI plant:  $\dot{x}(t) = A \cdot x(t) + B \cdot u(t)$  asymptotically stable ( $x(t) \rightarrow 0$  as  $t \rightarrow \infty$ )

$$x(t=0) = x_0$$

$A, B$  are  
const coeff.  
matrices

$$J = \int_0^\infty \frac{1}{2} \left[ \dot{x}^T Q x + \dot{x}^T S^T x \right] dt + x^T S x$$

Then  $J = \int_0^\infty \frac{1}{2} \left\{ x^T (Q + S^T S) x \right\} dt$

Quadratic objective func<sup>n</sup>  
with control intv. requires  $P = 0$   
with  $Q, R, S$  constant coeff. matrices.  
symmetric  $\cancel{\text{if}} \quad \cancel{\text{if}}$   
Riccati eqn.

•  $P \sim$  can be obtained from

$$y(t) = -R^{-1} [B^T P + S^T] \cdot \dot{x}(t)$$

$P(t) > 0$  sufficient condition for LQR system.

stabilized by feedback matrices

unstable at  $t = t_f$

! negative terms

## Lec-12 General formulation for bounded I/P.

Consider a general plant with

$$\text{vector state eqns} \quad \dot{\underline{x}} = f(\underline{x}, \underline{u}, t) \quad \rightarrow \textcircled{1}$$

subject to IC:  $\underline{x}(t_i) = \underline{x}_i$   $\rightarrow$  here  $\underline{x}_i$  &  $t_i$   
 $\textcircled{1}$  are both constant  
fixed.

$$\text{terminal BC: } g(\underline{x}(t_f), t_f) = 0 \quad \rightarrow \textcircled{2}$$

$$\text{Ineq. constraint: } h(\underline{x}, \underline{u}, t) \geq 0 \quad \rightarrow \textcircled{3}$$

A problem where  $t_f$  is not mentioned in advance is called an open control intv. problem.

The optimal control problem is the problem of determining a control history  $\underline{u}(t)$ ,  $t_i \leq t \leq t_f$ , such that the following objective function is minimized w.r.t. control I/P  $\underline{u}(t)$ .

$$J' = \phi[\underline{x}(t_f), t_f] + \int_{t_i}^{t_f} L(\underline{x}, \underline{u}, t) dt \quad \textcircled{5}$$

NOTE:  $\underline{x} \in \mathbb{R}^{n \times 1}$ ,  $\underline{u} \in \mathbb{R}^{m \times 1}$

$$\text{so, } f: \mathbb{R}^{n \times 1} \times \mathbb{R}^{m \times 1} \times \mathbb{R} \longrightarrow \mathbb{R}^{n \times 1}$$

$$g: \mathbb{R}^{n \times 1} \times \mathbb{R} \longrightarrow \mathbb{R}^{k \times 1}$$

$$\phi: \mathbb{R}^{n \times 1} \times \mathbb{R} \longrightarrow \mathbb{R}$$

$$L: \mathbb{R}^{n \times 1} \times \mathbb{R}^{m \times 1} \times \mathbb{R} \longrightarrow \mathbb{R}$$

$f, g, \phi, L$  are continuously diffble wrt their arguments. w.r.t.

The control history  $\underline{u}(t)$ , which minimizes  $J'$  subject to the state equation  $\textcircled{1}$ , initial & terminal BCs ( $\textcircled{2}$  &  $\textcircled{3}$ ),  $L$  inequality constraint  $\textcircled{3}$  is called an Optimal control history of the

If the corresponding trajectory  $\hat{x}(t)$ , firstly is called optimal trajectory of the system.

Ques. What does it indicate?

Ans. It indicates that the system has been controlled by the best possible way.

Ques. What is the best way?

Ans. The best way is the one which minimizes the cost function.

Ques. What is the cost function?

Ans. The cost function is the sum of all the costs of the system.

Ques. What are the costs?

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Ques. What are the costs?

① Optimal Control with Unbounded I/P : (i.e. there is no ineq. constraint ②)

control variable  $\tilde{u}(t)$  is completely unrestricted in the optimization process.

$$\tilde{z}_i = f(\tilde{z}, \tilde{u}, t) \quad \text{vector state eqn}$$

$$\tilde{z}(t_i) = \tilde{z}_i \quad \text{IC}$$

$$J = \phi[\tilde{z}(t_f), t_f] + \int_{t_i}^{t_f} \lambda(\tilde{z}, \tilde{u}, t) dt$$

objective function to be minimized.

Suppose  $\exists$  a trajectory  $\tilde{z}^*(t)$  with corresponding control history  $\tilde{u}^*(t)$ ,  $t_i \leq t \leq t_f$  which satisfies the "necessary condition" for minimization of  $J$  w.r.t  $\tilde{u}(t)$  subject to constraint ① & ②

such a trajectory will be called extremal trajectory corresponding to the extremal control history,  $\tilde{u}^*(t)$

The "necessary condition" for an extremal trajectory  $\tilde{z}^*(t)$  to exist in the control interv.  $t \in [t_i, t_f]$  are :

$\tilde{z}^*(t)$  ~ extremal trajectory

$\tilde{u}^*(t)$  ~ corresponding control history

$\tilde{\lambda}^*(t)$  ~ corresponding costate vector ]

$$(\dot{\lambda}^*)^T = -\frac{\partial L}{\partial \tilde{z}} - (\lambda^*)^T \frac{\partial f}{\partial \tilde{z}}$$

$$\frac{\partial L}{\partial \tilde{u}} + (\lambda^*)^T \frac{\partial f}{\partial \tilde{u}} = 0$$

$$\tilde{z}_i = f(\tilde{z}, \tilde{u}, t)$$

$$+\left(\frac{\partial \lambda_a}{\partial \dot{z}}\right)_{t=t_f} \delta \dot{z}_f + \left(\lambda_a - \frac{\partial \lambda_a}{\partial \dot{z}} \dot{z}^*\right)_{t=t_f} \delta t_f = 0$$

↳ transversality condition

$$\text{where } \lambda_a = \lambda + \frac{\partial \phi}{\partial t} + \frac{\partial \phi}{\partial z} \cdot \dot{z} + \dot{z}^T (f - \ddot{z})$$

augmented  
Langrangian.  $\lambda_a = H + \frac{\partial \phi}{\partial t} + \frac{\partial \phi}{\partial z} \cdot \dot{z} - \dot{z}^T \ddot{z}$

$$\text{where, } H = \lambda + \dot{z}^T f$$

In terms of Hamiltonian,  $H(z, y, \dot{z}, t)$

$$= \lambda(z, y, t) + \dot{z}^T f(z, y, t)$$

$$\dot{z} = A \cdot z + B \cdot y$$

the "necessary conditions" becomes:

Step I  $\rightarrow \dot{z}^* = - \left( \frac{\partial H}{\partial \dot{z}} \right)^T |_{(z^*, y^*, \dot{z}^*, t)}$

Step II  $\rightarrow \frac{\partial H}{\partial y} |_{(z^*, y^*, \dot{z}^*, t)}$

Step III  $\rightarrow \dot{z}^* \left( = \left( \frac{\partial H}{\partial \dot{z}} \right)^T \right) |_{(z^*, y^*, \dot{z}^*, t)} \\ = f(z^*, y^*, t)$

$$= c \int_0^{t_f} \left[ \frac{A(t-t_i)}{B} \dot{z}^*(t) dt + \right]$$

$$+ \left[ \int \left( \frac{\partial \phi}{\partial \dot{z}} \right)_{t=t_f}^T - \dot{z}^*(t_f) \right]^T \delta \dot{z}_f$$

$$+ \left( H + \frac{\partial \phi}{\partial t} \right)_{t=t_f} \delta t_f = 0$$

Euler Langrange  
equations

①

NOTE: We need  $(2n+1)$  BCs to solve the Euler-Lagrange equation.

of these  $n$  BCs are specified by

$$ICs \quad \tilde{g}(t_i) = \tilde{g}_{i,n}$$

We need  $(n+1)$  more BCs.

## ① Boundary Conditions:

1. Fixed terminal time:

$$\delta t_f = 0$$

We need  $n$  more BCs.

a. Fixed terminal state:  $\tilde{g}(t_f) = \tilde{g}_f \Rightarrow \delta \tilde{g}_f = 0$   
 ① is trivially satisfied.

b. Free terminal state:

$$\tilde{\lambda}^*(t_f) = \left( \frac{\partial \Phi}{\partial \dot{x}} \right)_{t=t_f}^T$$

c. Terminal state on a hypersurface: terminal state lying anywhere in the region specified by

$$F[\tilde{g}(t_f)] = \Omega_{Kx1}$$

$$F: \mathbb{R}^{n \times 1} \rightarrow \mathbb{R}^{K \times 1} \quad \text{with } 1 \leq K \leq n-1$$

$$\tilde{\lambda}^*(t_f) = \left( \frac{\partial \tilde{\Phi}}{\partial \dot{x}} \right)_{t=t_f}^T$$

$$\text{where } \tilde{\Phi} = \phi + \tilde{c}^T F[\tilde{g}(t_f)]$$

$$\text{here } \tilde{c} = (c_1, c_2, c_3, \dots, c_K)^T$$

are constant Langrange multipliers to be determined.

2. Free terminal time:  $\underline{\underline{t_f}}$  is not specified  
 $\delta t_f \neq 0$

① provides BC

to determine  $\underline{\underline{y}}$

a. Fixed terminal state:  $\underline{\underline{z}}(t_f) = \underline{\underline{z}_f} \rightarrow \delta \underline{\underline{z}}_f = 0$

solving for  $\underline{\underline{y}}$

$$\left( H + \frac{\partial \Phi}{\partial t} \right)_{t=t_f} = 0$$

b. Free terminal state:  $\underline{\underline{z}}^*(t_f) = \left( \frac{\partial \Phi}{\partial \underline{\underline{z}}} \right)_{t=t_f}^T$

$$\left( H + \frac{\partial \Phi}{\partial t} \right)_{t=t_f} = 0$$

c. time varying terminal state: terminal BC is a moving point in the state space described by time varying function,

$$h(t) : \mathbb{R} \rightarrow \mathbb{R}^{n \times 1}$$

$$\delta \underline{\underline{z}}_f = \left( \frac{dh}{dt} \right)_{t=t_f} \delta t_f \quad \text{put this in i)}$$

d. terminal state on a moving hypersurface:

The terminal state lies anywhere in a region of the state space ~~defined~~ specified by:

$$F[\underline{\underline{z}}(t_f), t_f] = 0$$

$$F: \mathbb{R}^{n \times 1} \times \mathbb{R} \rightarrow \mathbb{R}^{K \times 1}, 1 \leq K \leq n-1$$

$$\underline{\underline{z}}^*(t_f) = \left( \frac{\partial \Phi}{\partial \underline{\underline{z}}} \right)_{t=t_f}^T$$

where  $\Phi = \phi + \zeta^T$ .  $F[\underline{\underline{z}}(t_f), t_f]$

here  $\zeta = (c_1, c_2, \dots, c_K)^T$

$$\left( H + \frac{\partial \Phi}{\partial t} \right)_{t=t_f} = 0$$

$c_1, c_2, \dots, c_K$  are time varying coefficients to be treated as additional state variable.

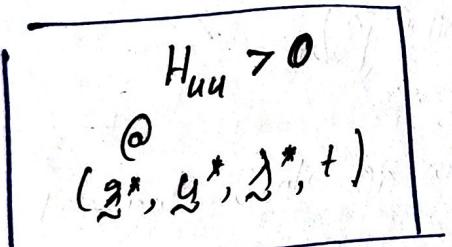
e. Terminal state partially fixed:

↓

Some state variables are fixed.  
or  
↓  
some state variables are free.

lies on a hypersurface.

 **Sufficient Condition**: for the existence of an extremal trajectory  $\tilde{x}^*(t)$  & corresponding control history  $\tilde{u}^*(t)$ , which minimizes  $J = \int L(\tilde{x}, \tilde{u}, t) dt$  subject to constraint  $\tilde{g}_i = f_i(\tilde{x}, \tilde{u}, t) = 0$

  $H_{uu} > 0$  @  $(\tilde{x}^*, \tilde{u}^*, \lambda^*, t)$

$H_{uu}$  non-neg.  $\Rightarrow$  Legendre-Clebsch condition.

When  $H_{uu} = 0$  on  $(\tilde{x}^*, \tilde{u}^*, \lambda^*, t)$  such a problem is said to be singular control.

NOTE: if  $H$  is a linear function of  $\tilde{u}$  then  $H_{uu} = 0$ .

$f$  requires additional constraints (such as linear constraints on  $\tilde{x}(t)$  &  $\dot{\tilde{x}}(t)$ ) to determine optimal trajectory.

① Consider a general plant with vector state evn<sup>n</sup>:  $\dot{z} = f(z, y, t) \in \mathbb{R}^{n \times 1}$  → ②  
 Subject to IC:  $z(t_i) = z_i' \rightarrow ③$

Ineq. constraint:  $h(z, y, t) \geq 0 \rightarrow ④$

Consider the minimization of  $J = \phi[z(t_f), t_f]$

$$H(z, y, \lambda, t) = l(z, y, t) + \int_{t_i}^{t_f} \lambda^T f(z, y, t) dt$$

wrt  $y(t) \in \mathbb{R}^{m \times 1}$

Pontryagin's minimum principle states that if  $\hat{y}(t)$  minimizes  $J$  w.r.t  $y(t)$  subject to constraint ②, i.e. if  $\hat{y}(t)$  is the optimal trajectory applied on the control  $\hat{y}(t)$ , then a variation  $\delta \hat{y}$  in the control  $\hat{y}(t)$  applied on the optimal trajectory  $\hat{y}(t)$  produces either an increase or a decrease in the Hamiltonian,  $H$ , i.e.  $H(\hat{z}, \hat{y} + \delta \hat{y}, \hat{\lambda}, t) > H(\hat{z}, \hat{y}, \hat{\lambda}, t)$

$\forall y \in U, t \in [t_i, t_f]$

an optimal control  $\hat{y}(t)$  is the one which minimises

$$H(z, y, \lambda, t)$$

Optimal trajectory  $\hat{z}(t)$

$U \subset \mathbb{R}^{m \times 1}$  is the set of all admissible controls  $y(t)$  which satisfies ④.

\* NOTE: • Pontryagin's minimum principle is a more general necessary condition than Euler-Lagrange evn<sup>n</sup>  
 • The minimum principle does not require  $H$  to be a smooth function of control variable  $y(t)$

The Euler-Lagrange eqn leads to the minimum principle if the Hamiltonian possesses continuous Adds upto and Order ant  $\mathcal{H}(t)$

- ① the Pontryagin's minimum principle is used in minimization of  $J$  in Singular Control Problem

$H_{uu} = 0$ , so Euler-Lagrange eqn is of no use to determine the optimal control profile.

Legendre Clebsch condition: the "sufficient condition" is NOT satisfied. Additional condition needs to be specified.

- iv) Consider the following Singular control problem:

state eqn of the system:  $\dot{x}_i = f(x, t) + \frac{\partial L(x, t)}{\partial u_i} u_i$

control affine system

[a system linear in  $u_i$ ]

Lagrangian to be minimized;  $\mathcal{L}(x, u, t) = l_1(x, t) + \frac{1}{2} u^T B(t) u$

$$\text{so, } H(x, u, \lambda, t) = l_1(x, t) + \lambda^T (f + Bu)$$

$H_{uu} = 0$  so we need more conditions to find the

Optimal trajectory  $\lambda(t)$  & corresponding optimal control  $u(t)$

When  $\frac{|y(t)|}{|u_m|} < u_m$  a practical solution can be obtained using Pontryagin's min. principle.

PMP yields

$$[\hat{l}_2(t) + \hat{\lambda}^T B(\theta, t)] \hat{g}(t) \leq [\hat{l}_2(t) + \hat{\lambda}^T B(\theta, t)] u_m,$$

$$\hat{u}_i(t) = \begin{cases} -u_m, & (\hat{l}_2 + \hat{\lambda}^T B)_i > 0 \\ 0, & (\hat{l}_2 + \hat{\lambda}^T B)_i = 0 \\ u_m, & (\hat{l}_2 + \hat{\lambda}^T B)_i < 0 \end{cases}, \quad i=1, \dots, m$$

Switching condition for optimal control

$G(t)$

$(\hat{l}_2 + \hat{\lambda}^T B)_i \sim$  Switching Function

$$\hat{u}_i(t) = -u_m \operatorname{sgn} (\hat{l}_2 + \hat{\lambda}^T B)_i, \quad i=1, \dots, m$$

NOTE:  $\oplus \Rightarrow \hat{u}_i = -u_m$  then  $\hat{u}_j = 0 \forall i \neq j$

or  $\hat{u}_i = u_m$  then  $\hat{u}_j = 0 \forall i \neq j$

Optimality requires the norm of the control I/O vector to be on the boundary which is either maximum or the minimum value.

Such an optimal control law switching between the two extreme I/O values is typical of singular problems & is called Bang Bang control.

Lagrangian of a time optimal control problem

Time optimal transfer, ( $J = t_f$ )

initial condition  $x(0) = x_0$

final condition  $x(t_f) = x_f$

$$\begin{aligned} \text{Maximize } & J = t_f \\ \text{subject to } & \dot{x} = f(x, u) \\ & x(0) = x_0 \end{aligned}$$

the action of taking set of mass at various times  
and the number of controls required is

number of controls required is  $n$ .  
number of controls required is  $n$ .

number of controls required is  $n$ .

① Space Flight: primarily governed by gravity field.

Unless thrust is applied by rocket engines, a spacecraft moves in well defined trajectories.

is perturbed by the presence of small forces of torque disturbances due to environment (atmosphere, solar radiation pressure, geo-magnetic etc.)

generated by one or more bodies (which are themselves moving under mutual gravitational attraction)

though small could cause an appreciable change in long period of time!!

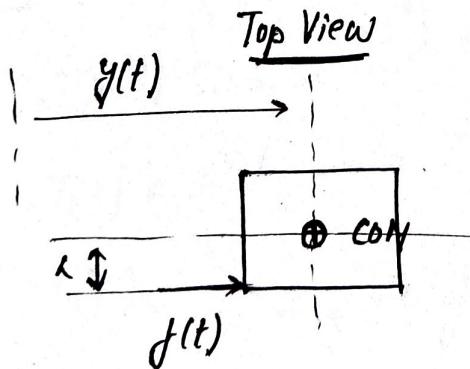
② Motion of a "spacecraft": can be divided into

(by assuming that it is a rigid body)

two points → Translation of COM  
[Orbital dynamics]

Rotation of spacecraft about its COM  
[attitude dynamics]

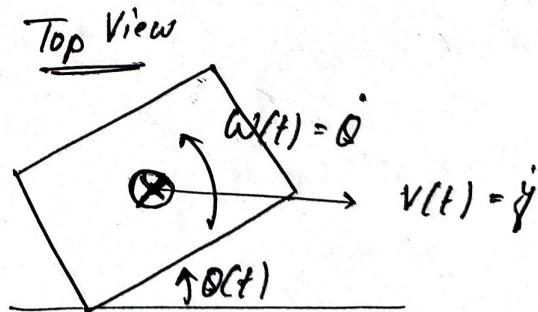
③ Consider a Sliding block on a Horizontal, Frictionless Table:



Translation

$$\text{translational dynamics: } m \frac{d^2 y}{dt^2} = f$$

System parameters:  $m$ ,  $J$ ,  $L$ ,  $f$



Rotation

$$\text{Rotational dynamics: } J \frac{d^2 \theta}{dt^2} = f L$$

Linear System  
Order of System = 4

Let us select the state variables:

$$q_1 = y$$

$$q_2 = \frac{dy}{dt}$$

$$q_3 = \theta$$

$$q_4 = \frac{d\theta}{dt}$$

State Equations

Translational  $\rightarrow q_1 = q_2$

$$\begin{matrix} \text{Kinematics} \\ \text{Kinetics} \end{matrix} \quad \sim q_2 = f/m$$

Rotational  $\rightarrow q_3 = q_4$

$$\begin{matrix} \text{Kinematics} \\ \text{Kinetics} \end{matrix} \quad \sim q_4 = I\ddot{\theta}/J$$

Want to obtain

Want to obtain

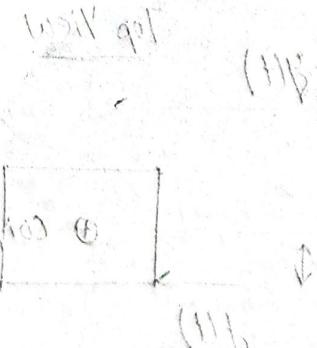
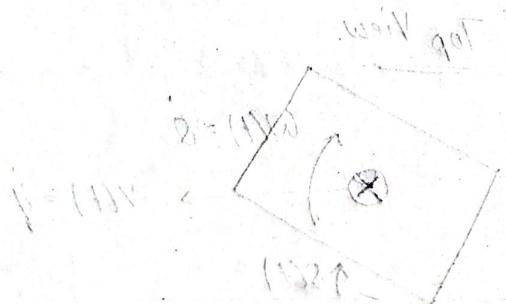
(Kinematics)

Want to obtain

(Kinetics)

(Kinematics)

Want to obtain (Kinetics)



$$\Rightarrow V = \frac{Fb}{m} t \quad (\text{kinematics})$$

$$V = \frac{M}{I} \theta t \quad (\text{kinetics})$$

# ① Space Flight Control:

is classified into

$$\begin{array}{l} \mathbf{r}(t) \in \mathbb{R}^{3 \times 1} \\ \mathbf{v}(t) \in \mathbb{R}^{3 \times 1} \\ \mathbf{a}(t) \in \mathbb{R}^{3 \times 1} \\ \omega(t) \in \mathbb{R}^{3 \times 1} \end{array}$$

Spacecraft's COM relative  
to a "fixed" reference  
frame.

for has its origin  
taken to be at the  
centre of a large body  
(such as a planet, the Moon, or  
the Sun) assumed to be  
at rest during the time of flight.

## Navigation

consisting of a  
control of position,  $\mathbf{r}(t)$

$\in \mathbb{R}^{3 \times 1}$  if linear

velocity  $\mathbf{v}(t), \mathbf{v}(t) \in \mathbb{R}^{3 \times 1}$

(i.e. translation) of the

## Attitude Control

consists of control of  
rigid vehicle's  
angular orientation

(or attitude),  $\omega(t)$   
 $\in \mathbb{R}^{3 \times 1}$

(called  
quaternion)

angular velocity vector,  
 $\omega(t) \in \mathbb{R}^{3 \times 1}$

about the COM relative  
to a "fixed" ref. frame.

has its origin

at the COM if its  
axes are fixed w.r.t.  
distant stars,

the  $\omega(t)$  of  $\mathbf{q}(t)$   
of the body frame  
(a frame rigidly  
attached with the  
spacecrafts) are  
measured relative  
to this ref.  
frame.

## Translational Motion by com

$$\ddot{\mathbf{r}} = \mathbf{v} \quad \text{kinematical eqn}$$

$$\dot{\mathbf{v}} = \mathbf{g}(\mathbf{r}, t) + \mathbf{T}/m \quad \text{kinematical eqn}$$

$\mathbf{g}(\mathbf{r}, t) \sim$  acc. due to gravity.

$\mathbf{T}(t) \in \mathbb{R}^{3 \times 1} \rightarrow$  Thrust by rocket  
engine.

$m(t) \sim$  mass of the vehicle.

The attitude is  
controlled by  
ext. applied torque  
I/Ps &/or internally  
Variable angular momentum via  
photons.

Rotational Motion of the body:

$$\ddot{\omega} = A(\omega) \quad \text{attitude kinematics}$$

$$\dot{\omega} = \underline{J}^{-1} \tau - \underline{J}^{-1} (\omega \times \underline{J} \omega) \quad \text{attitude kinetics}$$

$$A(\cdot) : \mathbb{R}^{3 \times 1} \rightarrow \mathbb{R}^{1 \times 1}$$

Non linear attitude  
kinematics  
functional.

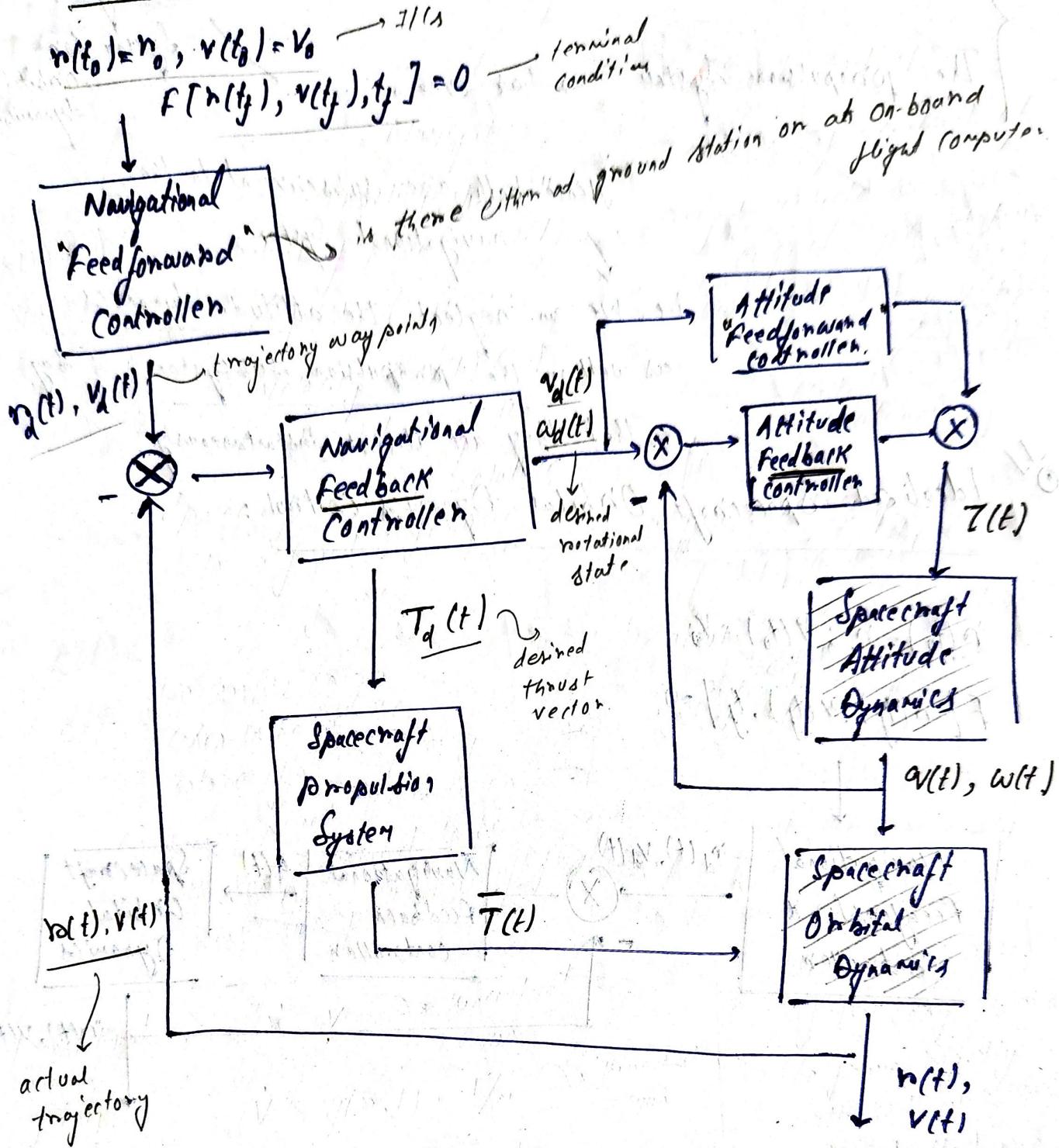
$$\underline{J} \in \mathbb{R}^{3 \times 3}$$

$\underline{J}(t) \in \mathbb{R}^{3 \times 3}$  (assumed to be const.  
for rigid body)

applied torque I/p.

# ① General Spacecraft Control System:

$$f(\cdot) : \mathbb{R}^{3 \times 1} \times \mathbb{R}^{3 \times 1} \rightarrow \mathbb{R}$$



- Navigational control J/P's are the commands issued to propulsion subsystem

for producing a derived thrust vector  $T_d(t)$

to attitude control system to achieve

desired rotational state :  $w_d(t), w(t)$

- ① { The attitude controller ~ has smaller timescale → part of navigation system (on orbital dynamics)  
 The propulsion system ~ has even " "

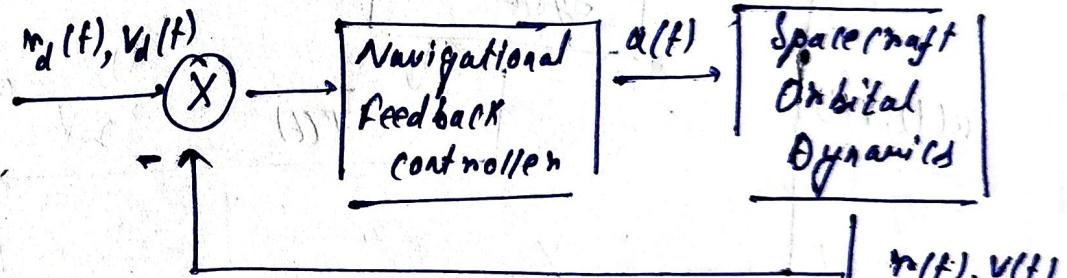
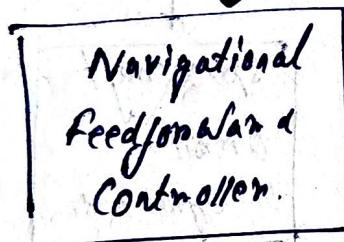
Hence both are subservient to the navigational system.

i.e. we can neglect the attitude dynamics as well as the propulsion subsystem & say that they act almost instantaneously.

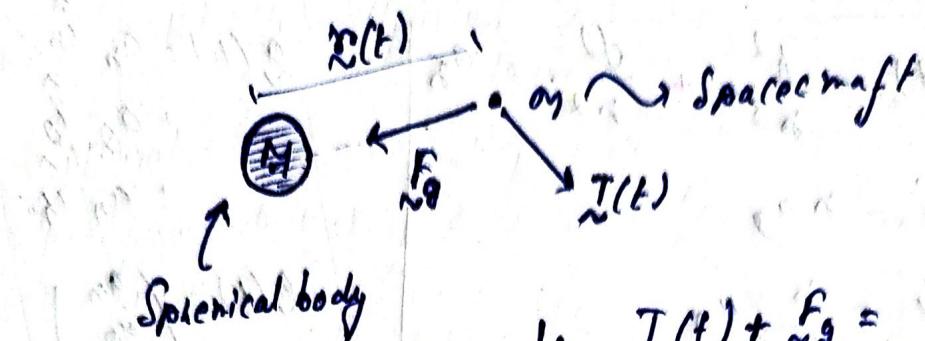
## ② Idealized Spacecraft Orbital ~~Dynamics~~ Control:

$$r(t_0) = r_0; v(t_0) = v_0$$

$$F[r(t_0), v(t_0), t_0] = 0$$



① Consider Orbiting around a Spherical body



$$1. \quad \ddot{r}(t) + \frac{F_g}{m} = m(t) \ddot{v}$$

$$\Rightarrow \frac{\ddot{r}(t)}{m(t)} + \frac{-GM\hat{r}}{r^2(t)m(t)} = \ddot{v}$$

$$2. \quad \boxed{\ddot{r}(t) = \ddot{v}(t)}$$

$$\Rightarrow \ddot{v} = g(t) - GM \frac{\ddot{r}(t)}{r^3(t)}$$

$$\Rightarrow \boxed{\ddot{v} = g(t) - \mu \frac{\ddot{r}(t)}{r^3(t)}} \rightarrow \textcircled{i}$$

where  $\mu = GM$

$\begin{matrix} \text{Grav.} \\ \text{Constant} \end{matrix}$

$$\therefore r(t) = |\ddot{r}(t)|$$

When  $g(t) = 0$  : Unforced Case

we will have  $\star$   
orbit

define  $\dot{h}(t) = \dot{r}(t) \times \dot{v}(t)$   $\dot{h}(t) \perp r(t)$  &  $\dot{h}(t) \perp \dot{v}(t)$   
angular momentum vector with magnitude  $h(t) = |\dot{h}(t)|$

lets define,  $\hat{r}_r = \frac{\dot{r}(t)}{|\dot{r}(t)|}$ ,  $\hat{h}_h = \frac{\dot{h}(t)}{h(t)}$ ,  $\hat{n}$   
unit vector in radial,  $\hat{h}_h = \hat{r}_r \times \hat{r}_n$ ,  $\hat{h}_h$  normal  
 $\hat{h}_c = \hat{h}_h \times \hat{r}_r$ ,  $\hat{h}_c$  circumferential  
 $\hat{h}_d$  direction

Resolving eqn (i) in  $\hat{i}$ ,  $\hat{j}$ ,  $\hat{k}$  directions we have:

$$\boxed{\ddot{r}(t) = \frac{h^2}{m^3} + \frac{p}{mr^2} = a_n}$$

~~$\ddot{r}(t)$~~

Charge in  
 $|r(t)|$

$\dot{h} = m a_0$

$\dot{p} = \frac{m}{h} a_n$

Charge in the orbital plane's orientation &

$$a(t) = a_n \hat{i}_n + a_0 \hat{j}_0 + a_n \hat{k}_n$$

$$a(t) = \begin{pmatrix} a_n \\ a_0 \\ a_n \end{pmatrix}$$

if  $a_n = 0 \sim \dot{a} = 0 \Rightarrow \alpha = \text{const.}$

if  $a_0 = 0 \sim \dot{h} = 0 \Rightarrow |h(t)| = \text{const.}$

if  $a_n = 0 = a_0 \sim h(t) = \text{const.}$

$$\frac{n = h/c^2}{\text{frequency}}$$

$$n = \sqrt{\frac{U}{c^3}}$$

$$U = GM$$

# Bang-Bang Control problem

Given :  $\dot{\tilde{z}} = f(\tilde{z}, \tilde{y}, t)$

$$\tilde{z}(t_i) = \tilde{z}_i$$

$$\tilde{z}(t_f) = \tilde{z}_f$$

Bound on I/P  $|y(t)| \leq u_m$

Objective function,  $J = \phi[\tilde{z}(t_f), t_f] + \int_{t_i}^{t_f} l(\tilde{z}, \tilde{y}, t) dt$

Step I :

Form the Hamiltonian,  $H = \lambda_i(\tilde{z}, \tilde{y}, t) + \sum_{j=1}^m \lambda_j^T f_j(\tilde{z}, \tilde{y}, t)$

Check whether  $H_{uu} = 0$  (singular problem)

if so then proceed.

apply PMP:  $H(\hat{z}, \hat{u}_0, \hat{\lambda}, +) \geq H(\hat{z}, \hat{u}, \hat{\lambda}, t) \sim (\rightarrow \hat{u}(t))$   
 $\hat{u}(t) = \begin{cases} -u_m & \text{for } \hat{u}(t) < -u_m \\ 0 & \text{for } -u_m \leq \hat{u}(t) \leq u_m \\ u_m & \text{for } \hat{u}(t) > u_m \end{cases}$

Step II :

Compare  $\dot{\tilde{z}} = f(\tilde{z}, \tilde{y}, t)$

with  $\dot{\tilde{z}} = f(\tilde{z}, t) + B(\tilde{z}, t) \cdot \hat{u}(t)$  [control affine system]

$\dot{\tilde{z}} = f(\tilde{z}, t)$  with

$$f(\tilde{z}, t) = \sum_{i=1}^n l_i(\tilde{z}, t) \cdot u_i(t) \quad \text{no of switches.}$$

Find  $B$  &  $l_2(t)$  expression for  $\hat{u}$  is needed to determine ~~whether~~  $\hat{u}(t)$

Step III : Find the costate variable  $\hat{\lambda}(t)$  corresponding to the extremal trajectory  $\hat{z}(t)$  & extremal control  $\hat{u}(t)$

$$\text{here } e^{\hat{\lambda}t} = \left( \begin{array}{c} 1 \\ \vdots \\ 1 \end{array} \right) \hat{\lambda} = \hat{\lambda}_1 \hat{z}_1 + \hat{\lambda}_2 \hat{z}_2 + \dots + \hat{\lambda}_n \hat{z}_n$$

$$\dot{\hat{z}} = -\left( \frac{\partial H}{\partial \tilde{z}} \right)^T \mid (\hat{z}^e, \hat{u}^e, \hat{\lambda}^e, t)$$

in this process you may need to assume

$$\hat{\lambda}_K(t_f) = c_K$$

Step IV: Now apply what Pontryagin's Minimum principle yields  
Under the bound  $|u_i(t)| \leq u_M$

② So,  $\hat{u}_i(t) = \begin{cases} -u_M, & (l_2 + \hat{\lambda}^T B)_i > 0, \\ 0, & (l_2 + \hat{\lambda}^T B)_i = 0, i=1, \dots, M \\ u_M, & (l_2 + \hat{\lambda}^T B)_i < 0 \end{cases}$

switching condition

Step V: decide the bounds on  $C_K$   $K = 1, \dots, M$  such that the  $Q_K$  are satisfied after switching happened at  $t_i$ . Decide whether you need to  $\hat{u}_i(t)$  or  $-u_M$  based on  $\hat{g}(t_i)$  or  $\hat{g}(t_f)$ .  $\hat{u}_i(t)$  is decided from  $\hat{\lambda}(t)$ .

Also you need to determine the Switching time  $\hat{t}$ .  
 This is decided from  $\hat{\lambda}(t)$

Step VI:  $\hat{u}(t) = e^{-\hat{\lambda}(t)} \hat{g}(t)$  from  $t_i \leq t \leq t_f$   
 Find  $\hat{g}(t)$  for  $t \leq t \leq t_f$

Step VII:  $\hat{t}$  can be found out by solving  $\hat{g}(t_f) = g$

(1)  $\int_{t_i}^{t_f} \hat{g}(t) dt = BC$  on  $\hat{g}(t)$

$$\begin{aligned} \hat{g}_1 &= A \cdot \hat{g}_1 + B \cdot \hat{g}_2 \\ \hat{g}_2 &= e^{-A(t-t_i)} \hat{g}_2(t=t_i) + \int_{t_i}^{t_f} e^{-A(t-\tau)} B \cancel{\hat{g}}_2(\tau) d\tau \end{aligned}$$

• plot graph of  $\hat{g}_1(t)$ ,  $\hat{g}_2(t)$  &  $u(t)$