

Compact-Reconstruction WENO Schemes – Theory, Implementation and Applications

Debojyoti Ghosh

Postdoctoral Appointee

Mathematics & Computer Science

Argonne National Laboratory

James D. Baeder (UMD)

Emil M. Constantinescu (ANL)

Jed Brown (ANL)

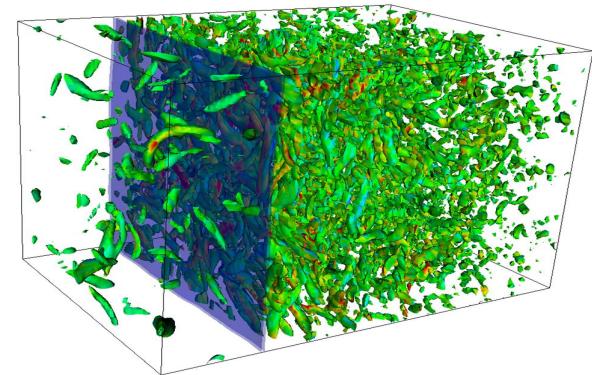
National Institute of Aerospace CFD Seminar

Hampton, VA, June 3rd, 2014

Motivation

Numerical Solution of Compressible Turbulent Flows

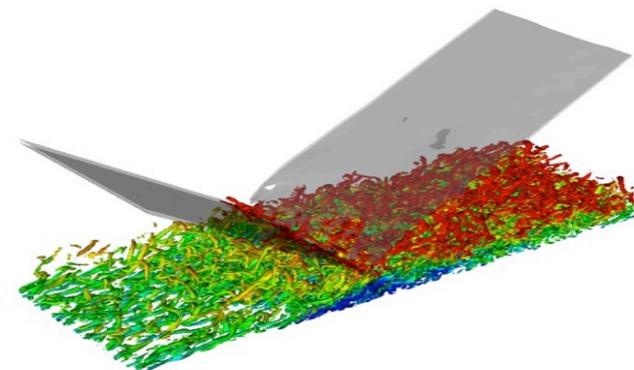
- Aircraft and Rotorcraft wake flows
- Characterized by **large range of length scales**
- Convection and interaction of eddies
- Compressibility → **Shock waves & Shocklets**
- Thin shear layers → **High gradients** in flow



Shock-Turbulence Interaction
(Stanford University CTR)

High order accurate finite-difference solver

- High spectral resolution for **accurate capturing of smaller length scales**
- Non-oscillatory solution across shock waves and shear layers
- Low dissipation errors for **preservation of flow structures over large distances**



Shock-Turbulent Boundary Layer Interaction
(Brandon Morgan, Stanford University and Nagi Mansour, NASA)

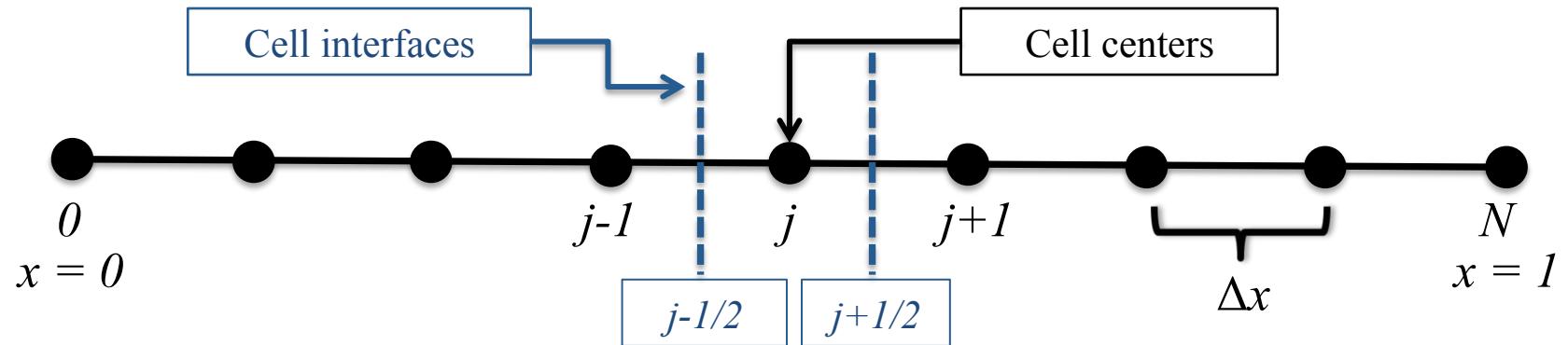
Outline

- **Compact-Reconstruction WENO Schemes**
 - Background and other high-resolution, non-oscillatory schemes in the literature
 - Description of the CRWENO scheme, especially the 5th order CRWENO schemes
 - Numerical analysis
- **Applications demonstrating the numerical properties**
 - Scalar conservation laws
 - Inviscid Euler equations
 - Aerodynamic flow problems
 - DNS of benchmark turbulent flows
- **Implementation and Computational Cost**
 - Computational efficiency on a single processor
 - Review of existing parallelizing strategies
 - Parallel implementation of the CRWENO5 scheme
 - Performance and scalability
- **Concluding Remarks**



Background

- **Hyperbolic Conservation Law** $u_t + f(u)_x = 0; f'(u) \in \Re$



- **Conservative finite-difference discretization**

$$\frac{du_j}{dt} + \frac{1}{\Delta x} [f(x_{j+1/2}, t) - f(x_{j-1/2}, t)] = 0$$

- **Reconstruction**

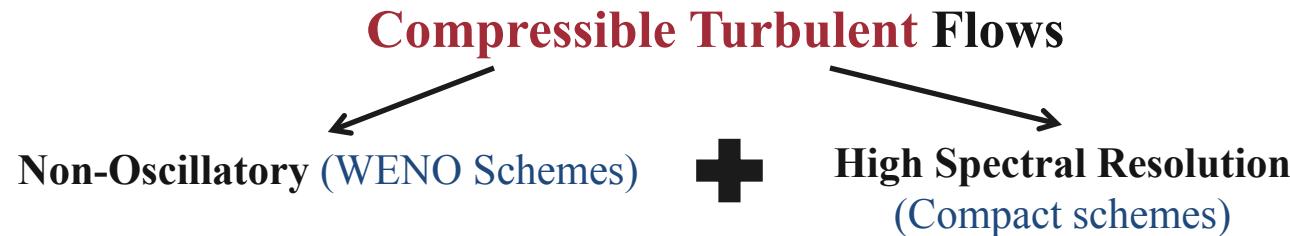
$$f_{j+1/2} = f(f_i : i = j-m, \dots, j+n)$$

High-Order Finite Difference Schemes

- **Weighted Essentially Non-Oscillatory Schemes**
 - Liu, Osher & Chan (*JCP*, 1994) and Jiang & Shu (*JCP*, 1996)
 - Solution-dependent combination of r interpolation stencils of r^{th} -order accuracy – **weights depend on the local smoothness of the solution**
 - Optimal weights in smooth regions allow **$(2r-1)^{\text{th}}$ order accuracy**
 - Near-zero weights for stencils with discontinuities → **non-oscillatory behavior**
 - Poor spectral resolution, even for very high orders of accuracy
- **Compact Finite-Difference Schemes**
 - Lele (*JCP*, 1992) (Non-conservative discretization of the flux derivative)
 - Coupling between neighboring flux derivative approximations
 - Requires the solution of a **banded (tridiagonal/penta-diagonal) system of equations** (linear compact schemes → constant coefficients → LU decomposition can be pre-computed)
 - **High spectral resolution** for the same order of convergence
 - Linear compact schemes need some form of limiting for solutions with discontinuities

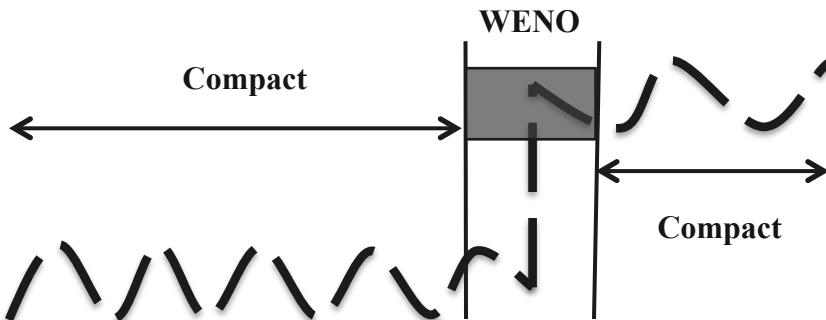


High-Resolution Non-Oscillatory Schemes



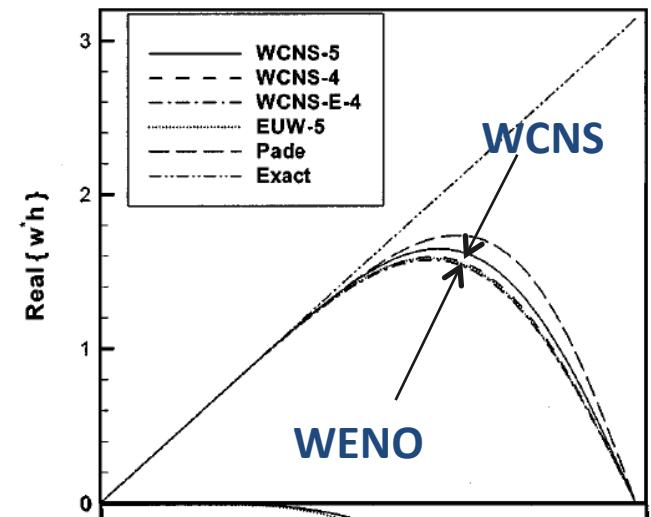
Hybrid Compact-WENO Schemes

- Smoothness indicator **marks out regions near discontinuities**
- Adams & Sherriff (1996), Pirozzoli (2002), Ren et. al. (2003)*
- Disadvantages:** arbitrary parameter for switching, loss of spectral resolution near discontinuities

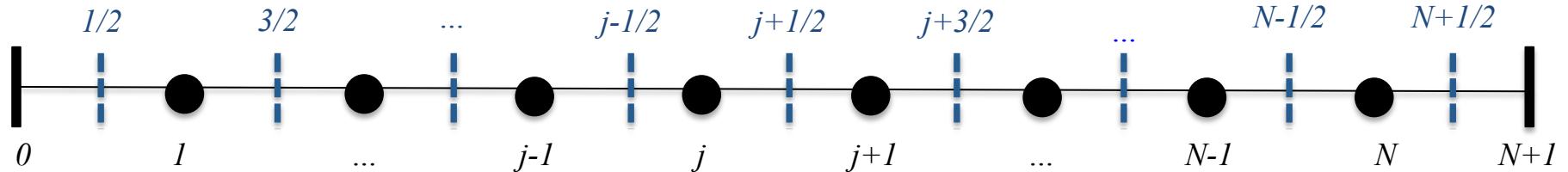


Weighted Compact Non-Linear Schemes

- Non-conservative finite-difference formulation over a staggered mesh
- Deng & Zhang (2000), Wang & Huang (2002), Zhang, Jiang & Shu (2008)*



Compact-Reconstruction WENO (CRWENO) Schemes



General form of a **conservative compact scheme**:

$$A(\hat{f}_{j+1/2-m}, \dots, \hat{f}_{j+1/2}, \dots, \hat{f}_{j+1/2+m}) = B(f_{j-n}, \dots, f_j, \dots, f_{j+n}) \quad \rightarrow \quad [A]\hat{\mathbf{f}} = [B]\mathbf{f}$$

At each interface, r possible (r) -th order compact interpolations, combined using optimal weights c_k to yield $(2r-1)$ -th order compact interpolation scheme:

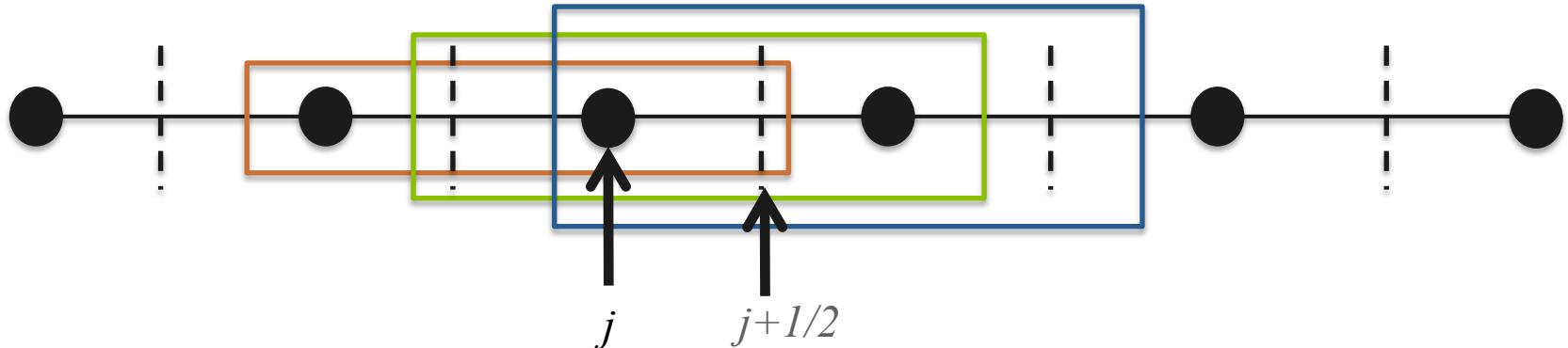
$$\sum_{k=1}^r c_k A_k^r(\hat{f}_{j+1/2-m}, \dots, \hat{f}_{j+1/2+m}) = \sum_{k=1}^r c_k B_k^r(f_{j-n}, \dots, f_{j+n})$$

$$\Rightarrow A^{2r-1}(\hat{f}_{j+1/2-m}, \dots, \hat{f}_{j+1/2+m}) = B^{2r-1}(f_{j-n}, \dots, f_{j+n})$$

Apply WENO algorithm on the optimal weights c_k – scale them according to local smoothness

$$\sum_{k=1}^r \omega_k A_k^r(\hat{f}_{j+1/2-m}, \dots, \hat{f}_{j+1/2+m}) = \sum_{k=1}^r \omega_k B_k^r(f_{j-n}, \dots, f_{j+n}) \quad \alpha_k = \frac{c_k}{(\beta_k + \varepsilon)^p}; \quad \omega_k = \alpha_k / \sum_k \alpha_k$$

5th Order CRWENO Scheme (CRWENO5)



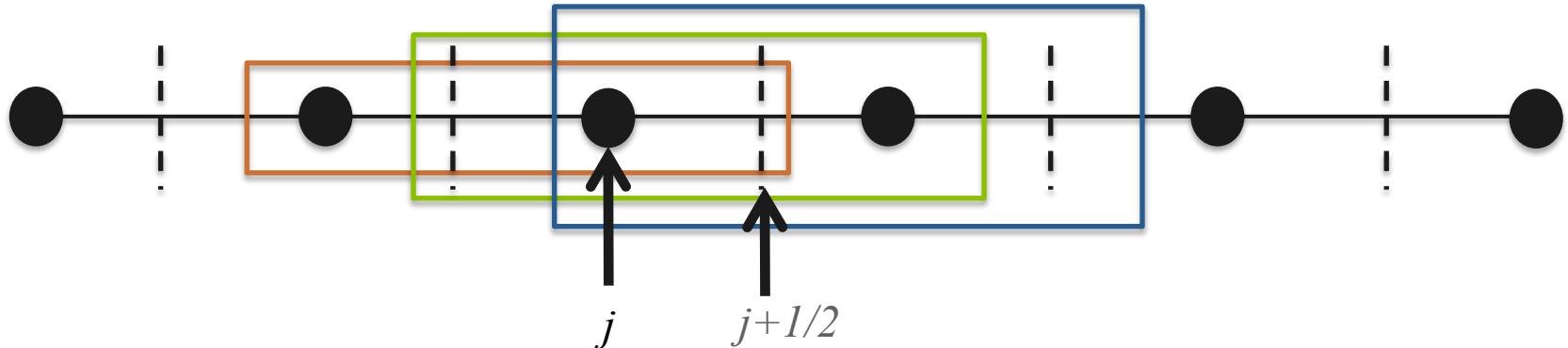
$$\frac{2}{3}f_{j-1/2} + \frac{1}{3}f_{j+1/2} = \frac{1}{6}f_{j-1} + \frac{5}{6}f_j \quad \xrightarrow{\text{orange}} \quad c_1 = \frac{2}{10}$$

$$\frac{1}{3}f_{j-1/2} + \frac{2}{3}f_{j+1/2} = \frac{5}{6}f_j + \frac{1}{6}f_{j+1} \quad \xrightarrow{\text{green}} \quad c_2 = \frac{5}{10}$$

$$\frac{2}{3}f_{j+1/2} + \frac{1}{3}f_{j+3/2} = \frac{1}{6}f_j + \frac{5}{6}f_{j+1} \quad \xrightarrow{\text{blue}} \quad c_3 = \frac{3}{10}$$

$$\frac{3}{10}f_{j-1/2} + \frac{6}{10}f_{j+1/2} + \frac{1}{10}f_{j+3/2} = \frac{1}{30}f_{j-1} + \frac{19}{30}f_j + \frac{10}{30}f_{j+1}$$

5th Order CRWENO Scheme (CRWENO5)



$$\frac{2}{3}f_{j-1/2} + \frac{1}{3}f_{j+1/2} = \frac{1}{6}f_{j-1} + \frac{5}{6}f_j$$



$$c_1 \begin{pmatrix} 2 \\ 0 \\ 0 \end{pmatrix} \omega_1$$

$$\frac{1}{3}f_{j-1/2} + \frac{2}{3}f_{j+1/2} = \frac{5}{6}f_j + \frac{1}{6}f_{j+1}$$



$$c_2 \begin{pmatrix} 2 \\ -1 \\ 0 \end{pmatrix} \omega_2$$

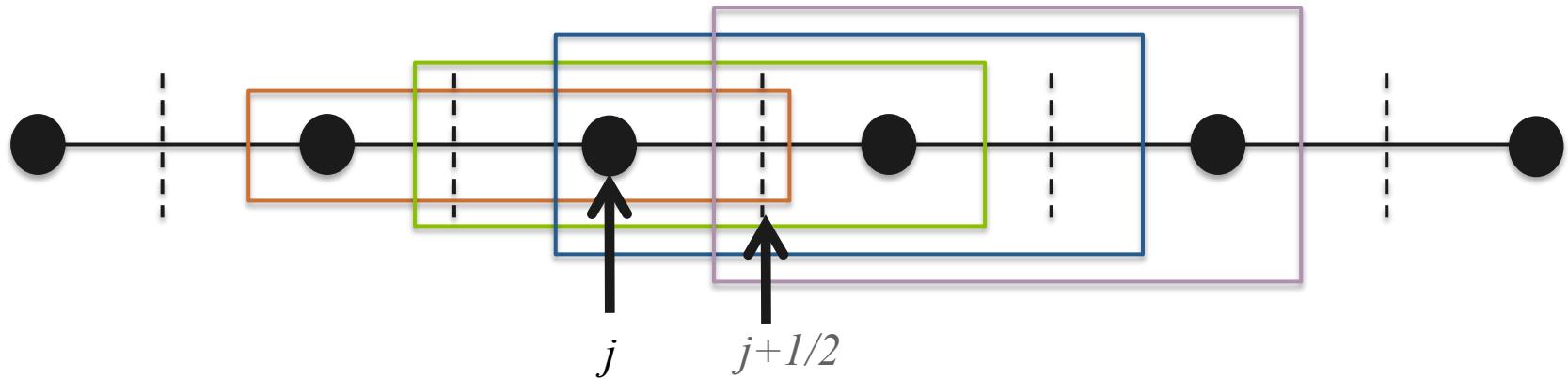
$$\frac{2}{3}f_{j+1/2} + \frac{1}{3}f_{j+3/2} = \frac{1}{6}f_j + \frac{5}{6}f_{j+1}$$



$$c_3 \begin{pmatrix} 2 \\ -1 \\ 10 \end{pmatrix} \omega_3$$

$$\left(\frac{2}{3}\omega_1 + \frac{1}{3}\omega_2 \right) f_{j-1/2} + \left(\frac{1}{3}\omega_1 + \frac{2}{3}(\omega_2 + \omega_3) \right) f_{j+1/2} + \frac{1}{3}\omega_3 f_{j+3/2} = \frac{\omega_1}{6}f_{j-1} + \frac{5(\omega_1 + \omega_2)}{6}f_j + \frac{\omega_2 + 5\omega_3}{6}f_{j+1}$$

Low Dissipation CRWENO5 scheme (CRWENO5-LD)



$$\frac{2}{3}f_{j-1/2} + \frac{1}{3}f_{j+1/2} = \frac{1}{6}f_{j-1} + \frac{5}{6}f_j$$



$$c_1 = \frac{3}{20}$$

$$\frac{1}{3}f_{j-1/2} + \frac{2}{3}f_{j+1/2} = \frac{5}{6}f_j + \frac{1}{6}f_{j+1}$$



$$c_2 = \frac{9}{20}$$

$$\frac{2}{3}f_{j+1/2} + \frac{1}{3}f_{j+3/2} = \frac{1}{6}f_j + \frac{5}{6}f_{j+1}$$



$$c_3 = \frac{7}{20}$$

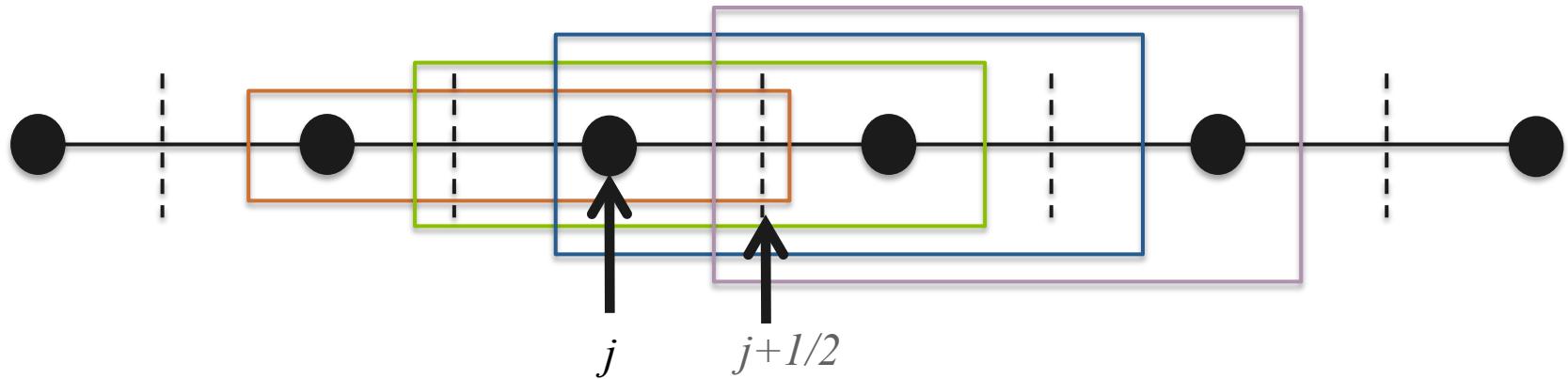
$$\frac{1}{3}f_{j+1/2} + \frac{2}{3}f_{j+3/2} = \frac{5}{6}f_{j+1} + \frac{1}{6}f_{j+2}$$



$$c_4 = \frac{1}{20}$$

$$\frac{5}{20}f_{j-1/2} + \frac{12}{20}f_{j+1/2} + \frac{3}{20}f_{j+3/2} = \frac{3}{120}f_{j-1} + \frac{67}{120}f_j + \frac{49}{120}f_{j+1} + \frac{1}{120}f_{j+2}$$

Low Dissipation CRWENO5 scheme (CRWENO5-LD)



$$\frac{2}{3}f_{j-1/2} + \frac{1}{3}f_{j+1/2} = \frac{1}{6}f_{j-1} + \frac{5}{6}f_j$$



$$c_1 \times \frac{3}{20} \quad \omega_1$$

$$\frac{1}{3}f_{j-1/2} + \frac{2}{3}f_{j+1/2} = \frac{5}{6}f_j + \frac{1}{6}f_{j+1}$$



$$c_2 \times \frac{9}{20} \quad \omega_2$$

$$\frac{2}{3}f_{j+1/2} + \frac{1}{3}f_{j+3/2} = \frac{1}{6}f_j + \frac{5}{6}f_{j+1}$$



$$c_3 \times \frac{7}{20} \quad \omega_3$$

$$\frac{1}{3}f_{j+1/2} + \frac{2}{3}f_{j+3/2} = \frac{5}{6}f_{j+1} + \frac{1}{6}f_{j+2}$$



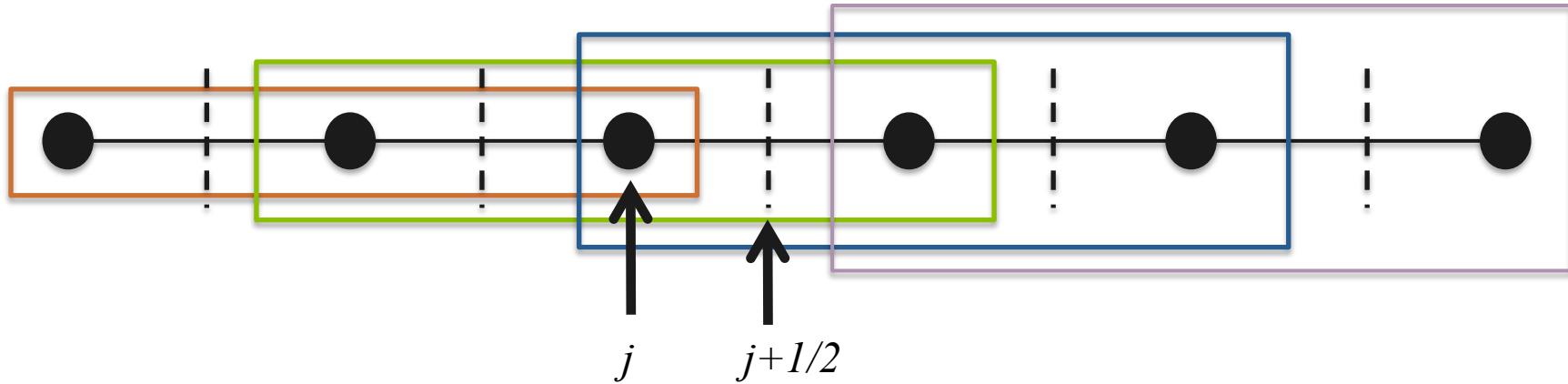
$$c_4 \times \frac{1}{20} \quad \omega_4$$

$$\left(\frac{2}{3}\omega_1 + \frac{1}{3}\omega_2 \right) f_{j-1/2} + \left(\frac{1}{3}\omega_1 + \frac{2}{3}(\omega_2 + \omega_3) + \frac{1}{3}\omega_4 \right) f_{j+1/2} + \left(\frac{1}{3}\omega_3 + \frac{2}{3}\omega_4 \right) f_{j+3/2} = \frac{\omega_1}{6}f_{j-1} + \frac{5(\omega_1 + \omega_2) + \omega_3}{6}f_j + \frac{\omega_2 + 5(\omega_3 + \omega_4)}{6}f_{j+1} + \frac{\omega_4}{6}f_{j+2}$$



Smoothness Indicators

Weights are calculated based on **smoothness indicators** of corresponding explicit stencils (**same as WENO5 scheme**)



$$\left. \begin{array}{l} \beta_1 = IS(f_{j-2}, f_{j-1}, f_j) \\ \beta_2 = IS(f_{j-1}, f_j, f_{j+1}) \\ \beta_3 = IS(f_j, f_{j+1}, f_{j+2}) \\ \beta_4 = IS(f_{j+1}, f_{j+2}, f_{j+3}) \end{array} \right\} \quad \alpha_k = \frac{c_k}{(\beta_k + \varepsilon)^p}; \quad \omega_k = \frac{\alpha_k}{\sum_k \alpha_k}; \quad k = 1, 2, 3$$

$\beta_4 = \max(\beta_3, \beta_4)$ (Avoid downwind interpolation)

CRWENO5-LD

Numerical Analysis – Accuracy and Convergence

Underlying optimal (linear) schemes:

$$f_{j+1/2} = \frac{1}{30}f_{j-2} - \frac{13}{60}f_{j-1} + \frac{47}{60}f_j + \frac{27}{60}f_{j+1} - \frac{1}{20}f_{j+2} \quad \text{WENO5}$$

$$\frac{3}{10}f_{j-1/2} + \frac{6}{10}f_{j+1/2} + \frac{1}{10}f_{j+3/2} = \frac{1}{30}f_{j-1} + \frac{19}{30}f_j + \frac{10}{30}f_{j+1} \quad \text{CRWENO5}$$

$$\frac{5}{20}f_{j-1/2} + \frac{12}{20}f_{j+1/2} + \frac{3}{20}f_{j+3/2} = \frac{3}{120}f_{j-1} + \frac{67}{120}f_j + \frac{49}{120}f_{j+1} + \frac{1}{120}f_{j+2} \quad \text{CRWENO5-LD}$$

Dissipation and dispersion errors

WENO5

$$\left. \frac{1}{60} \frac{\partial^6 f}{\partial x^6} \right|_j \Delta x^5$$

CRWENO5

$$\left. \frac{1}{600} \frac{\partial^6 f}{\partial x^6} \right|_j \Delta x^5$$

CRWENO5-LD

$$\left. \frac{1}{1200} \frac{\partial^6 f}{\partial x^6} \right|_j \Delta x^5$$

Dissipation

WENO5

$$\left. \frac{1}{140} \frac{\partial^7 f}{\partial x^7} \right|_j \Delta x^6$$

CRWENO5

CRWENO5-LD

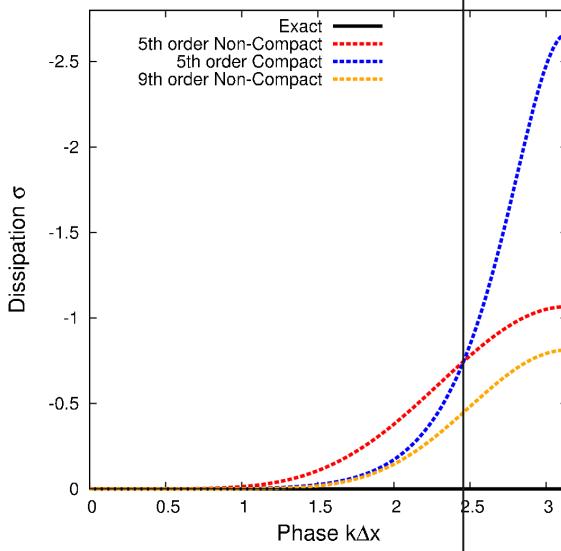
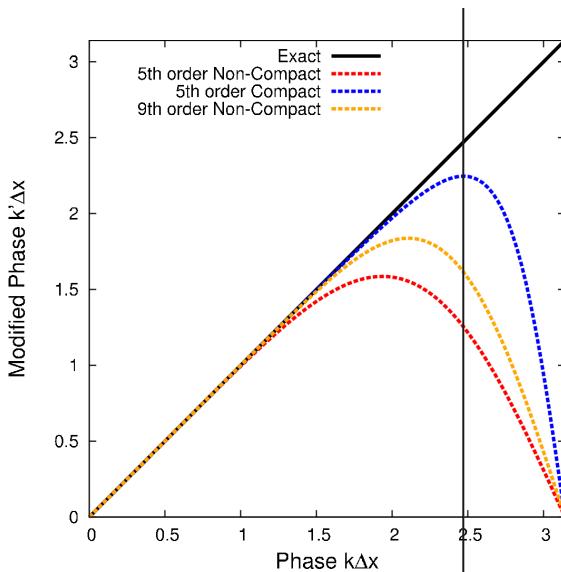
$$\left. \frac{1}{2100} \frac{\partial^7 f}{\partial x^7} \right|_j \Delta x^6$$

Dispersion

→ WENO5 requires **~ 1.5 times more grid points per dimension** to yield a solution of comparable accuracy as the CRWENO5 scheme.



Numerical Analysis – Spectral Resolution

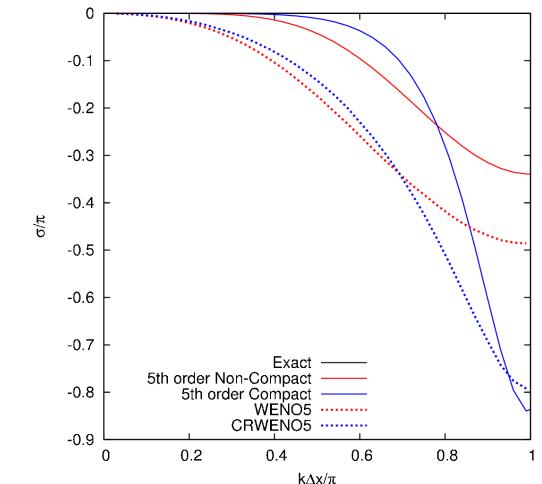
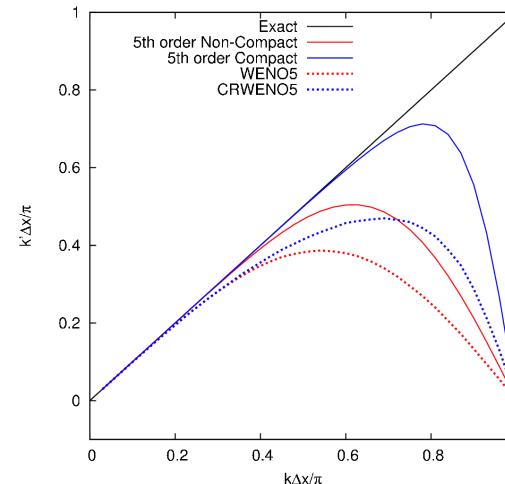


Linear Analysis

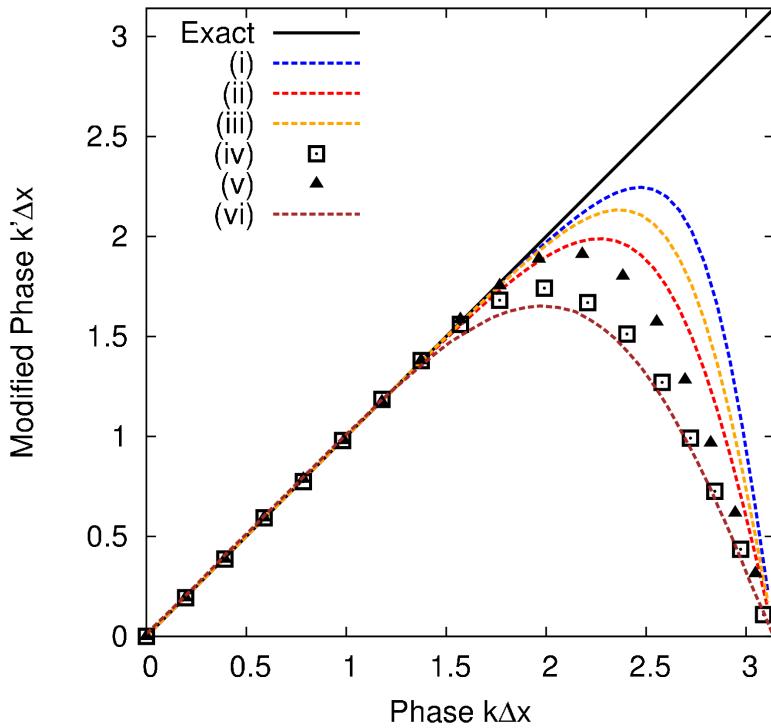
- 5th order compact scheme (CRWENO5) shows higher spectral resolution than the 5th (WENO5) and 9th order (WENO9) non-compact schemes.
- Lower dissipation at small and moderate wavenumbers for the compact scheme
- Higher dissipation for the compact scheme at incorrectly aliased wavenumbers

Non-Linear Analysis

- Fauconnier & Dick (*J. Comput. Appl. Math.*, 2013)
- Same qualitative trends, but resolution and dissipation compromised due to non-linear weights



Numerical Analysis – Comparison



- i. CRWENO5
- ii. 6th order central compact (Lele, 1992)
- iii. 8th order central compact (Lele, 1992)
- iv. WENO-SYMB0 ($r=3$) (Martin, et. al., 2006)
- v. WENO-SYMB0 ($r=4$) (Martin, et. al., 2006)
- vi. WCNS5 (Deng & Zhang, 2000)

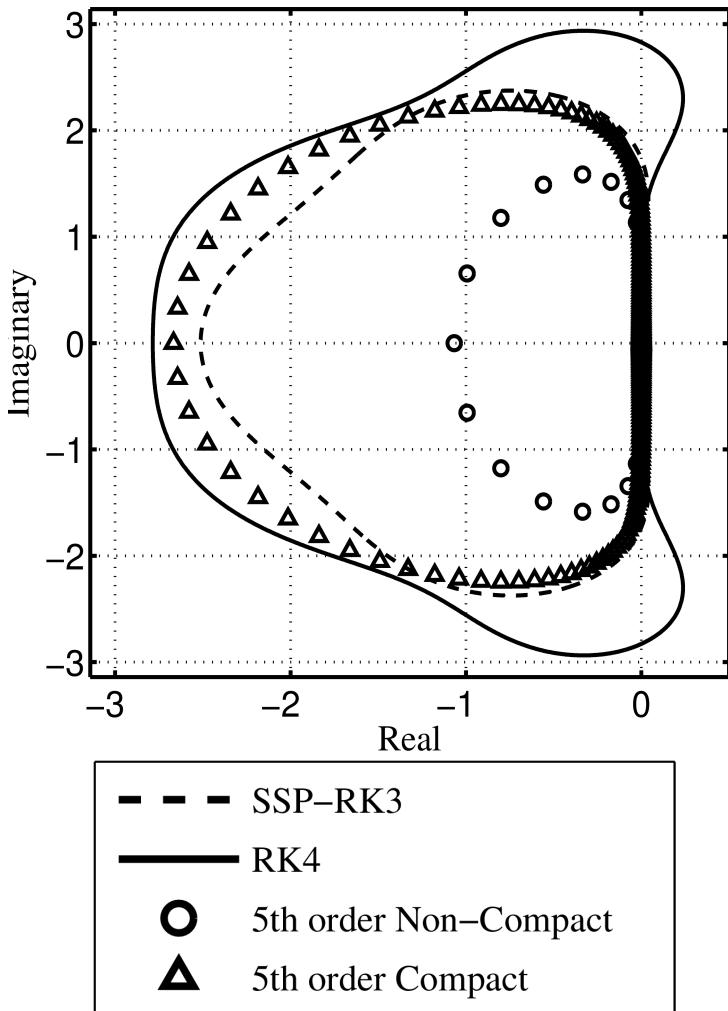
Comparison of **spectral resolution** and **bandwidth resolving efficiency** – CRWENO5 scheme with high-resolution schemes in literature

Bandwidth Resolving Efficiency

WENO5 (Jiang & Shu, 1996)	0.35
WENO7 (Balsara & Shu, 2000)	0.42
WENO9 (Balsara & Shu, 2000)	0.48
CRWENO5	0.61
6th-order central compact (Lele, 1992)	0.50
8th-order central compact (Lele, 1992)	0.58
WENO-SYMB0 ($r = 3$) (Martin, et. al., 2006)	0.49
WENO-SYMB0 ($r = 4$) (Martin, et. al., 2006)	0.56



Time-Integration and Stability

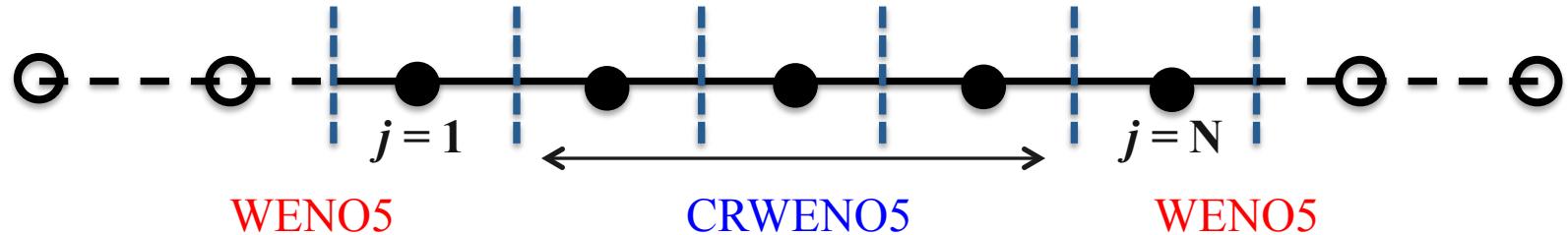


Linear Stability Analysis

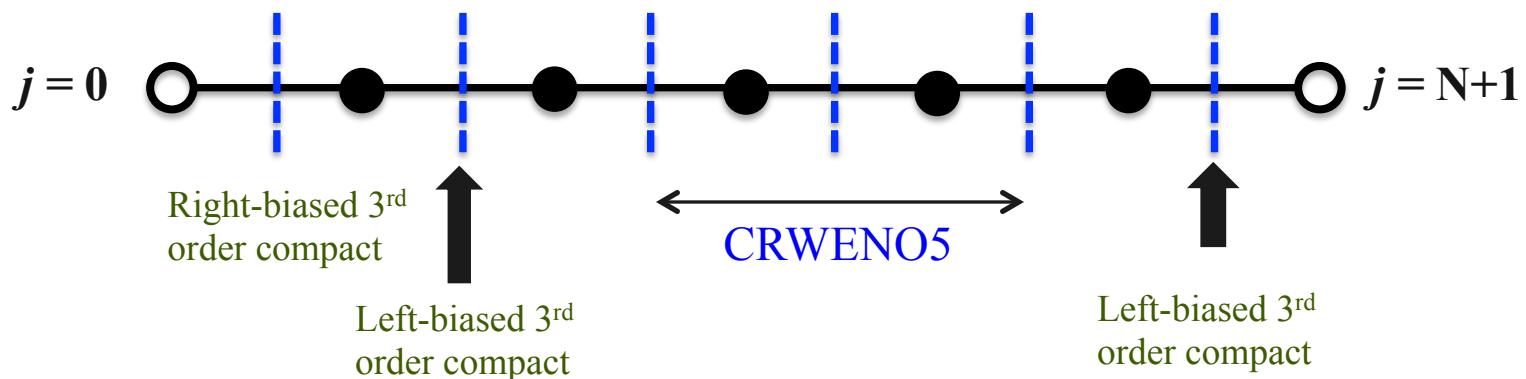
- CRWENO5 scheme has a lower linear stability limit than the WENO5 scheme → Smaller timestep restriction for explicit time integration
- CFL restriction for **3rd order SSP Runge-Kutta** scheme
 - WENO5: ~ 1.5
 - CRWENO5: ~ 0.9
- CFL restriction for **4th order Runge Kutta** scheme
 - WENO5: ~ 1.8
 - CRWENO5: ~ 1.1
- Time step size limit is **~ 1.6 times smaller** for CRWENO than WENO5

Boundary Treatment

Cell interface aligned with physical boundary – Boundary conditions implemented on “ghost cells”



Cell center coincident on physical boundary – **Reduced-order / interior-biased interpolation scheme** at interfaces near the boundary



Applications

Scalar Conservation Laws

Inviscid Euler Equations

Aerodynamic Applications

DNS of Benchmark Turbulent Flows



Scalar Conservation Laws

Schemes validated for the **linear advection equation** and the **inviscid Burgers equation**

Smooth problems

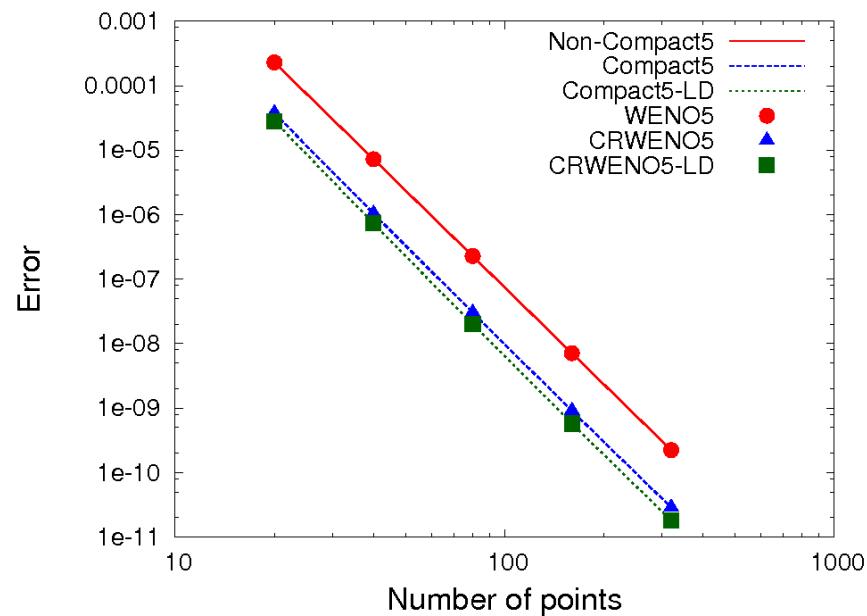
- 5th order convergence verified for the new schemes
- Errors for CRWENO5 order of magnitude lower than WENO5, errors for CRWENO5-LD half those of CRWENO5 (consistent with Taylor series analysis)

Linear advection equation

$$u_0(x) = \sin(2\pi x); 0 \leq x \leq 1$$

(Periodic domain)

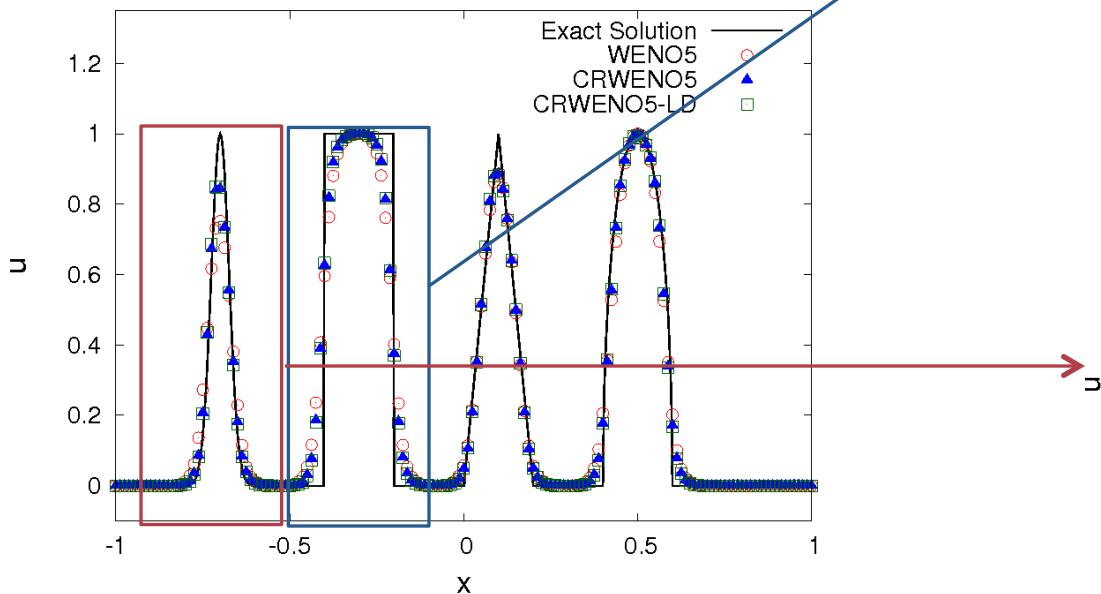
Solution obtained after 1 cycle
with SSP-RK3 time-stepping



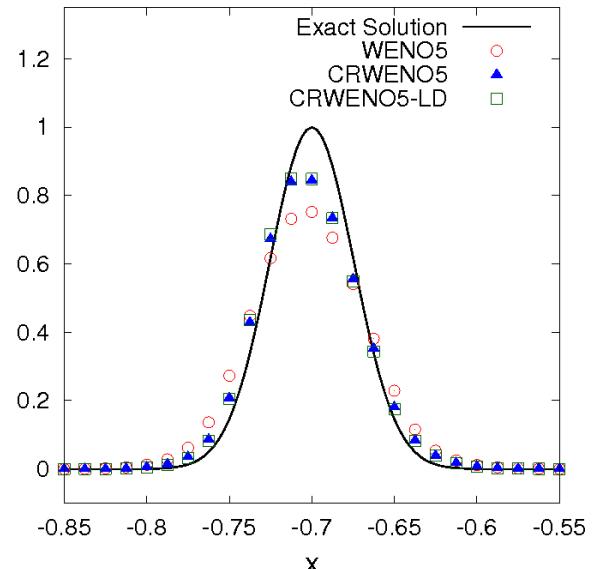
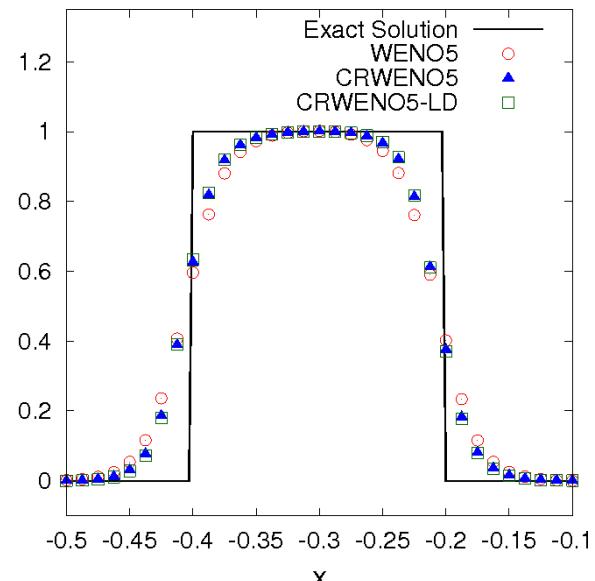
Scalar Conservation Laws

Discontinuous problems

- Non-oscillatory behavior validated across discontinuities
- CRWENO schemes show better resolution of discontinuous data



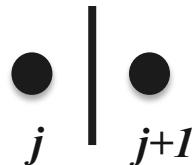
Linear Advection equation



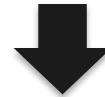
CRWENO for Euler Equations

Extension of scalar reconstruction schemes to vectors

- Component-wise reconstruction
- Characteristic-based reconstruction



$$a\alpha_{j-1/2}^k + b\alpha_{j+1/2}^k + c\alpha_{j+3/2}^k = \tilde{a}\alpha_{j-1}^k + \tilde{b}\alpha_j^k + \tilde{c}\alpha_{j+1}^k \quad (\alpha_i^k = \mathbf{l}_{j+1/2}^k \cdot \mathbf{f}_i)$$



\mathbf{U}^{avg} (Roe averaged)

$$\hookrightarrow \lambda_{j+1/2}^k, \mathbf{l}_{j+1/2}^k, \mathbf{r}_{j+1/2}^k$$

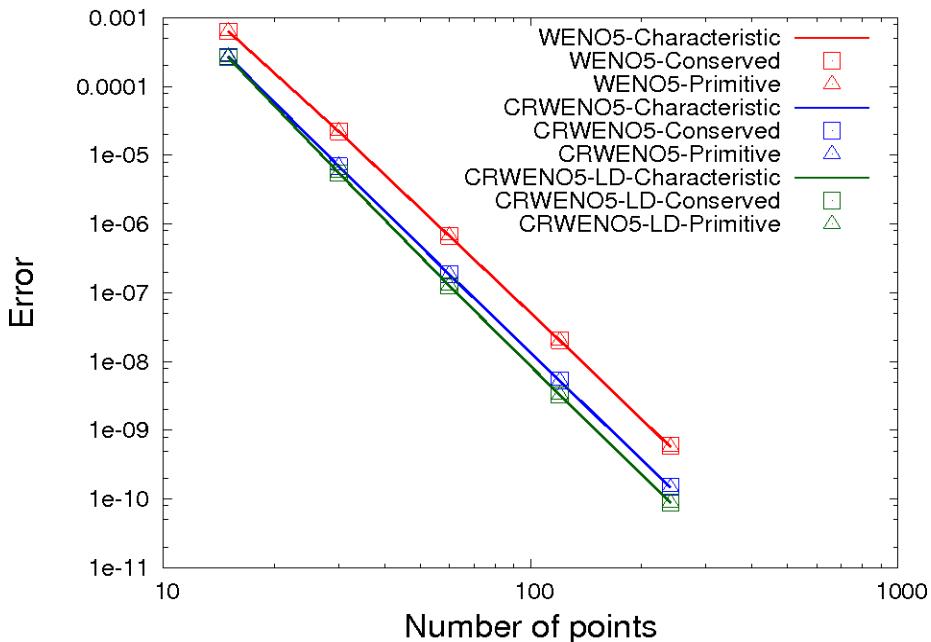
Eigenvalues, left and right eigenvectors

$$\begin{aligned} & a(l_{j+1/2}^{k1} f_{j-1/2}^1 + l_{j+1/2}^{k2} f_{j-1/2}^2 + l_{j+1/2}^{k3} f_{j-1/2}^3) \\ & + b(l_{j+1/2}^{k1} f_{j+1/2}^1 + l_{j+1/2}^{k2} f_{j+1/2}^2 + l_{j+1/2}^{k3} f_{j+1/2}^3) = \tilde{a}\alpha_{j-1}^k + \tilde{b}\alpha_j^k + \tilde{c}\alpha_{j+1}^k \\ & + c(l_{j+1/2}^{k1} f_{j+3/2}^1 + l_{j+1/2}^{k2} f_{j+3/2}^2 + l_{j+1/2}^{k3} f_{j+3/2}^3) \end{aligned}$$

Results in a **block tri-diagonal linear system** along each dimension (as compared to tri-diagonal system for conserved/primitive reconstruction)

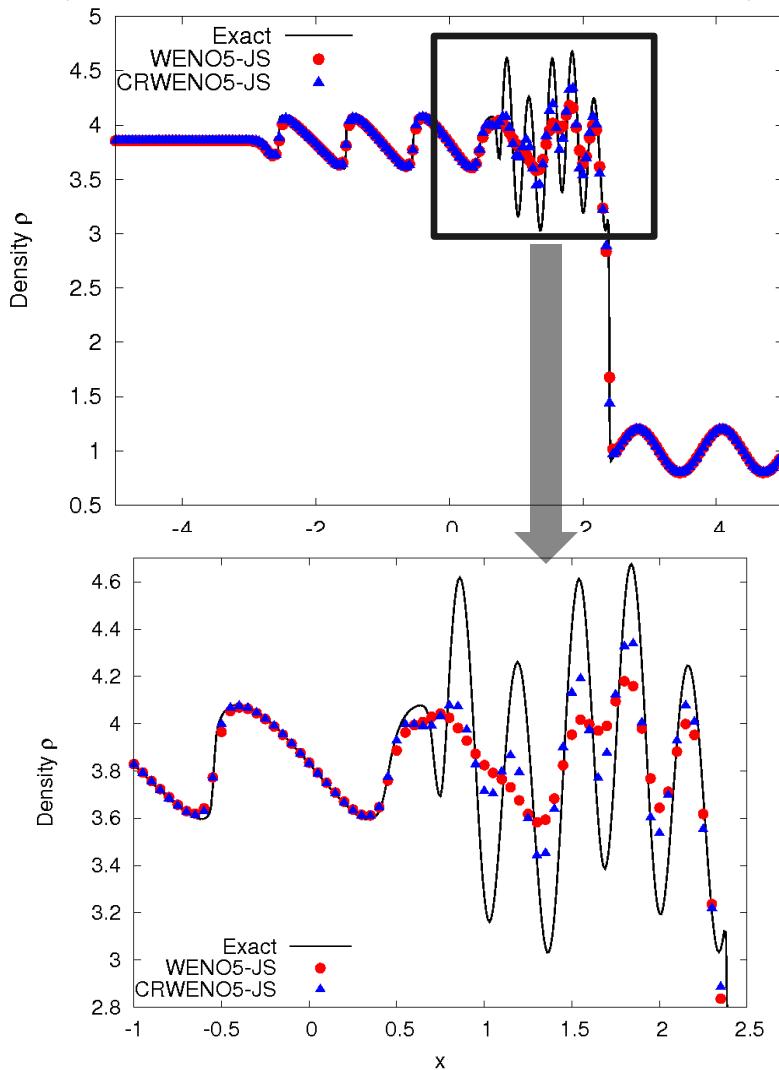
Inviscid Euler Equations (1D)

Entropy Wave Advection



- CRWENO5 errors **1/10th** that of WENO5 for smooth solution
- **Improved resolution of small length-scale** waves and discontinuities

Shu-Osher Problem (Characteristic-based reconstruction)

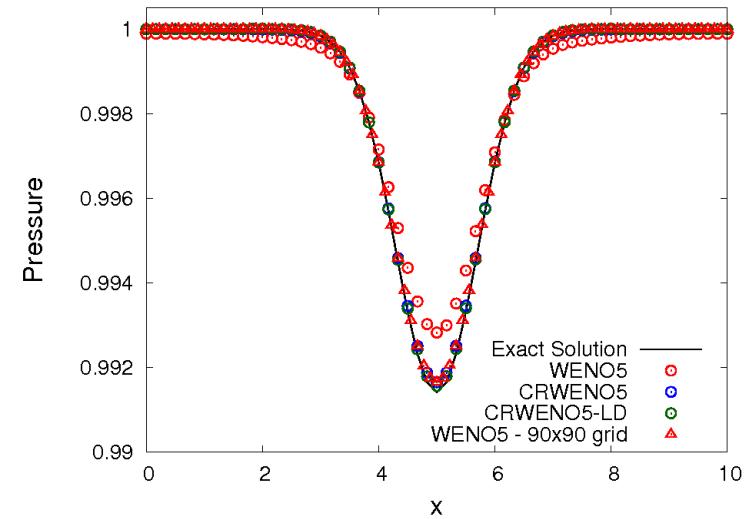
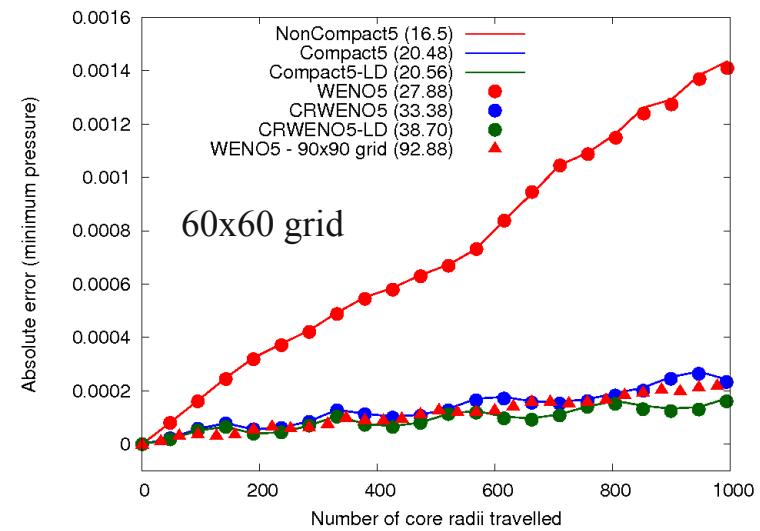
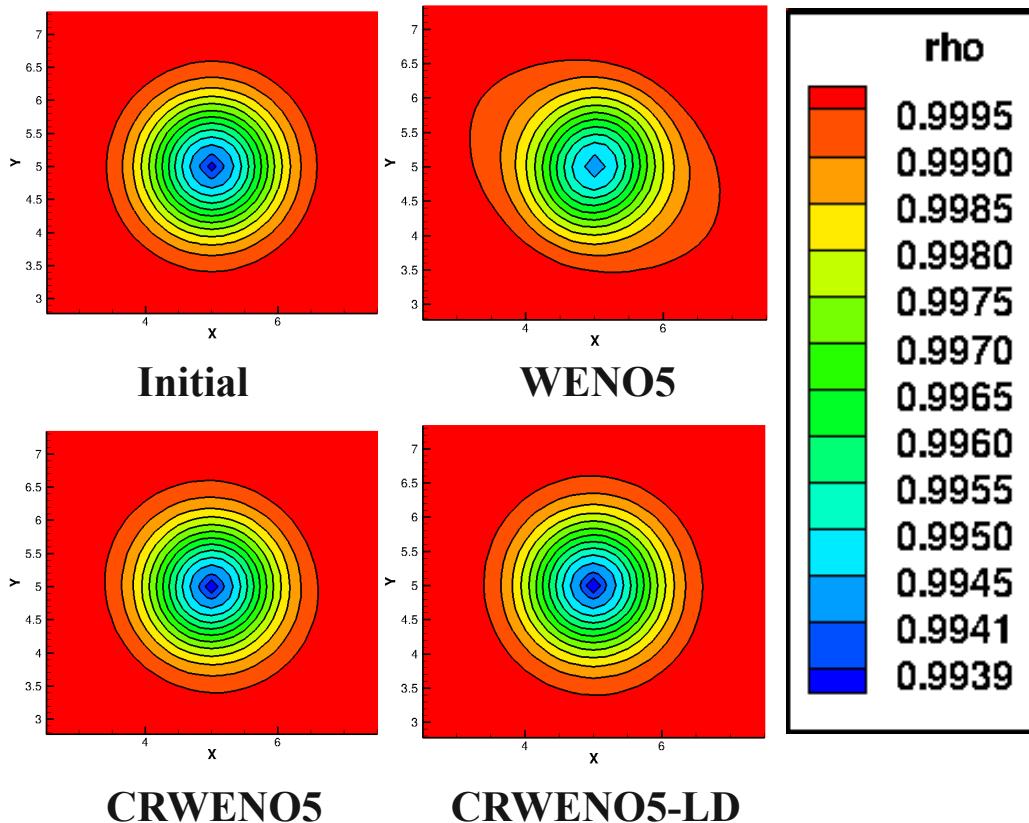


Inviscid Euler Equations (2D)

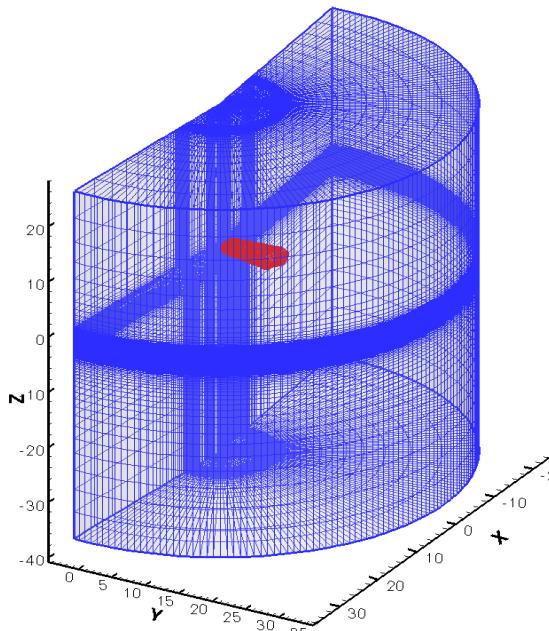
Isentropic Vortex Convection: Solution

after travelling 1000 core radii

Compact schemes show better shape and strength preservation for long term convection



Flow around Harrington 2-Bladed Rotor

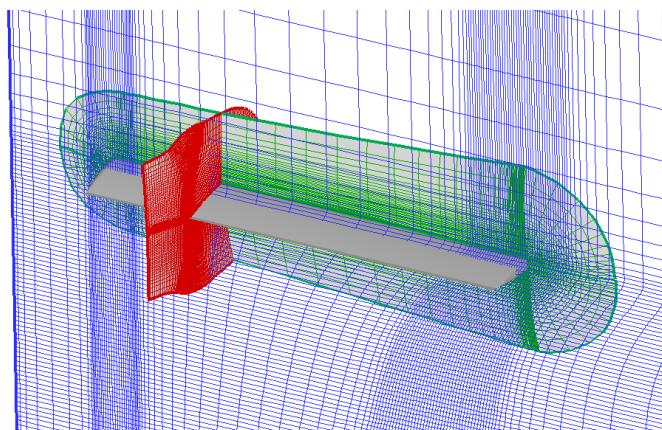


Cylindrical Background Mesh

127 x 116 x 118 points

Rotor Geometry:

- Aspect Ratio – 8.33
 - Airfoil section – NACA
- (t/c: 27.5% @ 0.2R, 15% @ R)

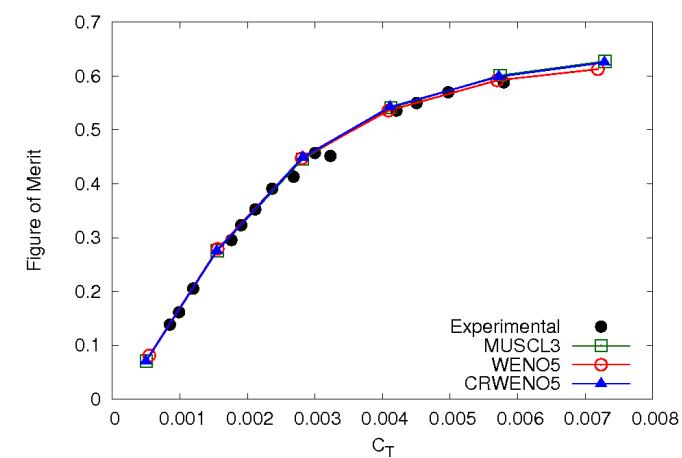
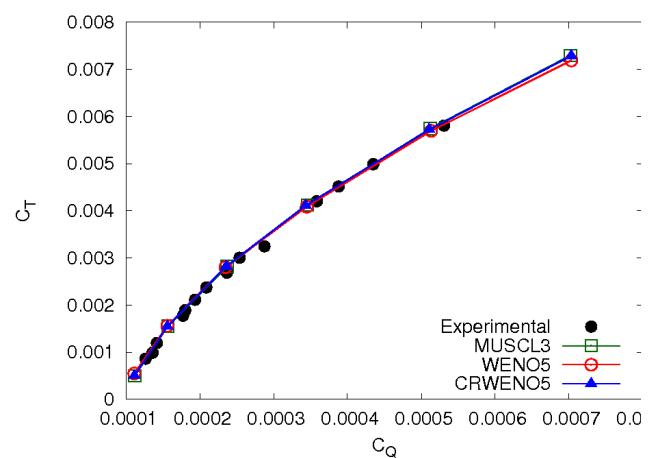


Blade Mesh –
C-O type, 267x78x56 points

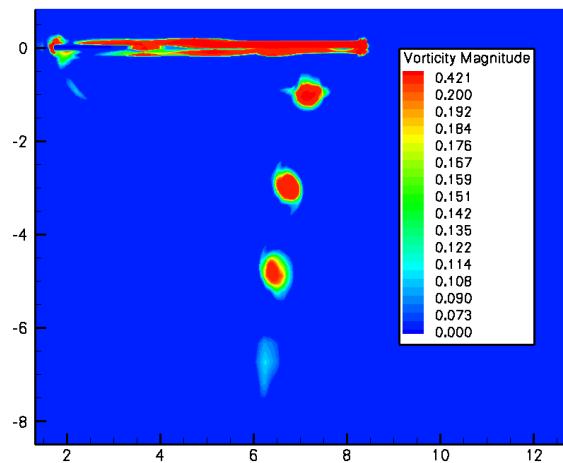
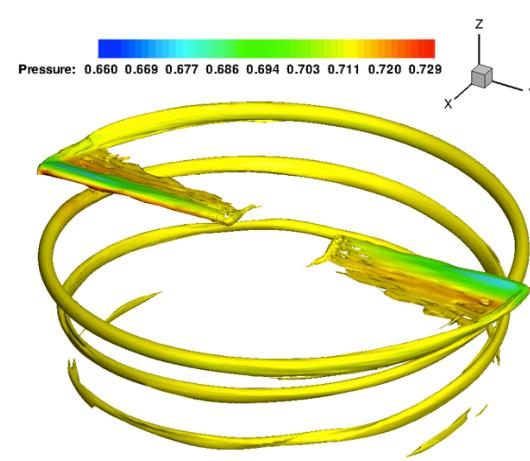
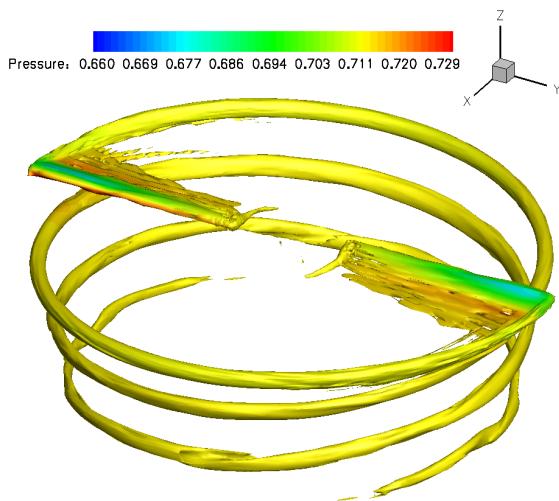
Flow Conditions:

- M_{tip} : 0.352
- Re_{tip} : 3.5 million

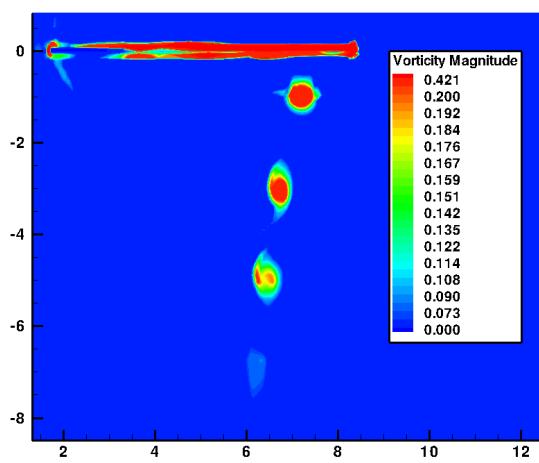
Validation of thrust & power coefficients and figure of merit



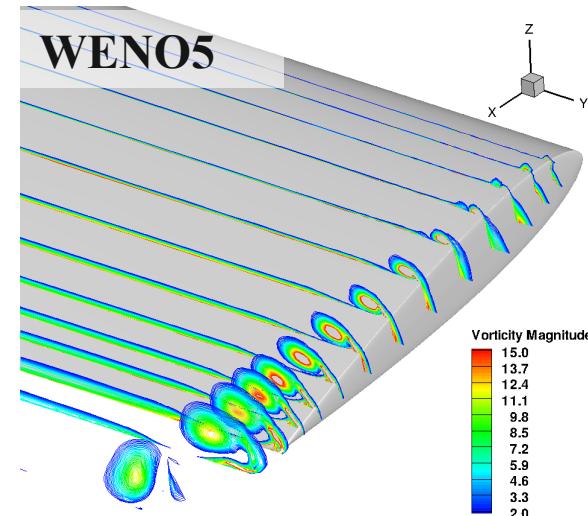
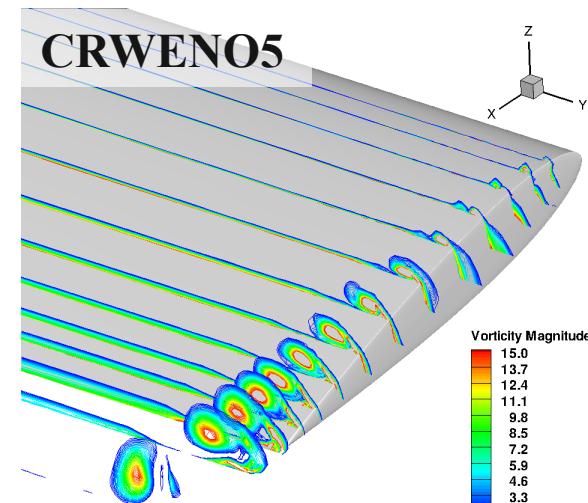
Harrington 2-Bladed Rotor (Near-Blade and Wake Flowfield)



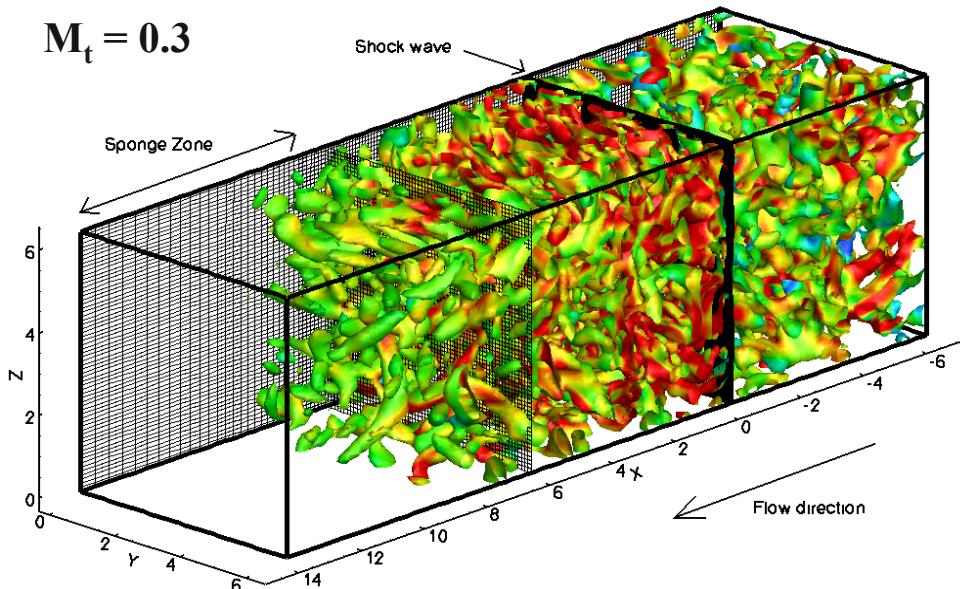
CRWENO5



WENO5

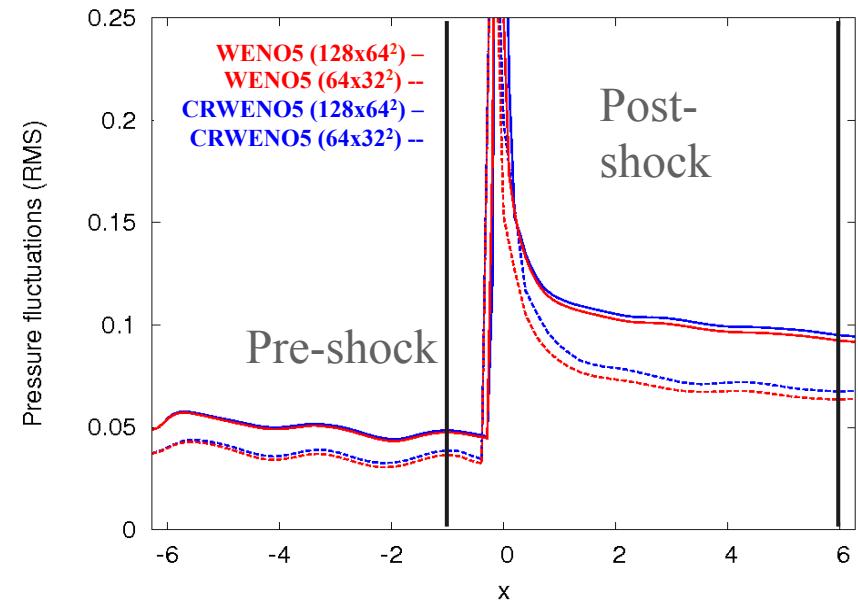


Shock – Turbulence Interaction



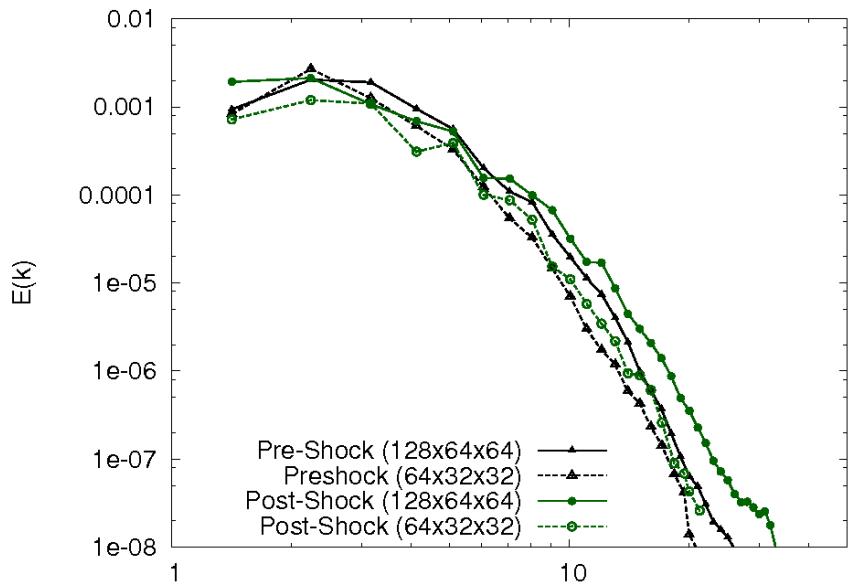
Iso-surfaces of 2nd invariant of velocity tensor,
colored by vorticity magnitude

- **Inflow:** Fluctuations from isotropic turbulence decay added to mean flow at Mach 2
- Interaction with a shock wave **magnifies the turbulent fluctuations**
- Problem solved on two grids: 64x32x32 and 128x64x64 points (uniform)
- CRWENO5 → Lower dissipation → **Predicts higher levels of fluctuations on both grids**

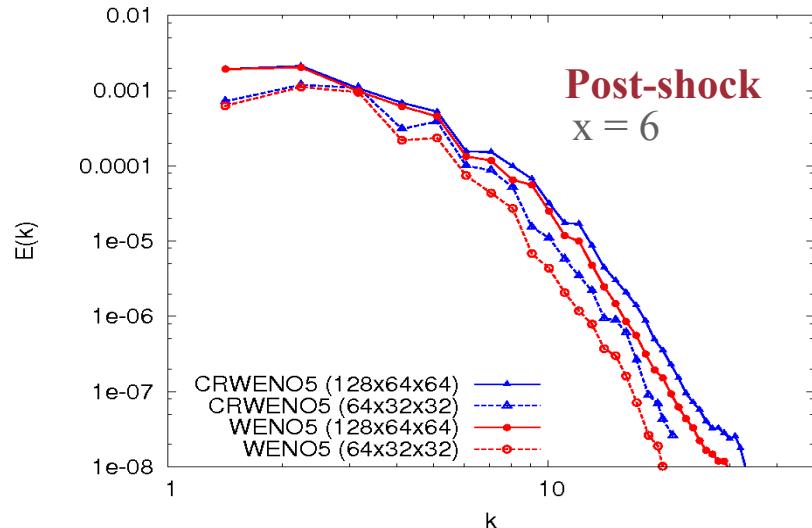
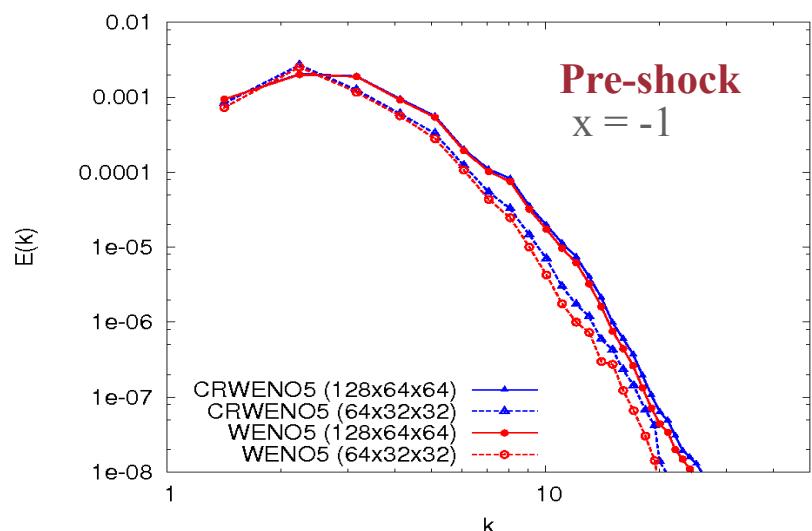


Stream-wise pressure fluctuations (RMS)

Shock – Turbulence Interaction



- Interaction with a shock wave amplifies intermediate and higher wavenumbers
- CRWENO5 shows improved resolution of the smaller length scales (both grids)



Implementation

Computational Efficiency

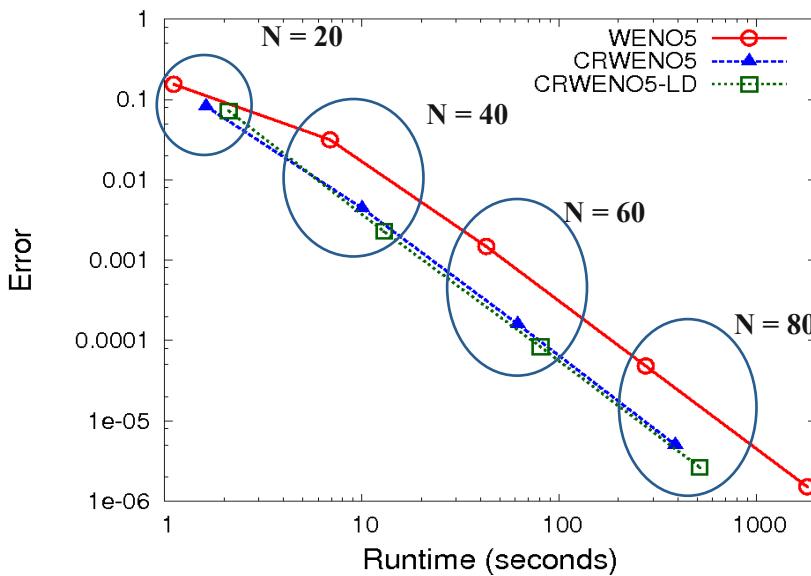
Parallel Implementation

Scalability

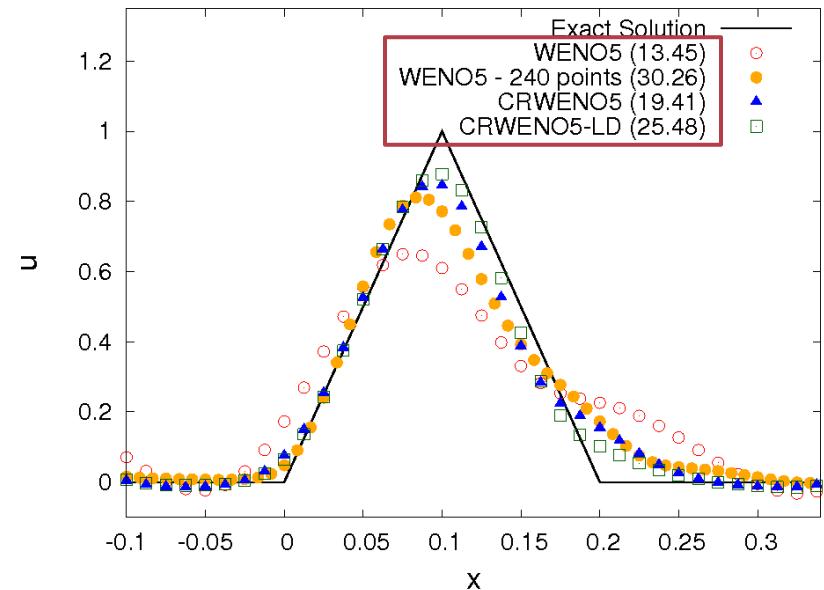


Computational Efficiency on Single Processor (Linear Advection Equation)

- CRWENO schemes **require solution to tridiagonal system** at each time-integration step/stage (solution-dependent weights) → **Higher numerical cost** for same number of points
- Higher accuracy (error is 1/10th that of WENO) → **Comparable solution obtained on coarser mesh** → Computationally more efficient



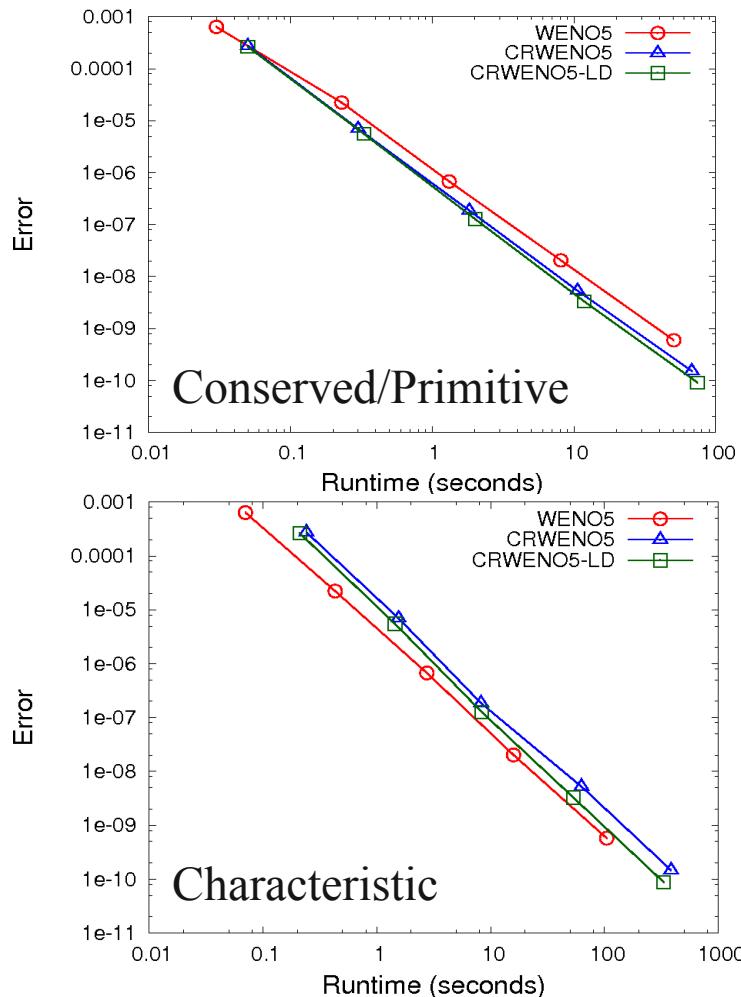
Error vs. runtime for periodic advection of a sine wave



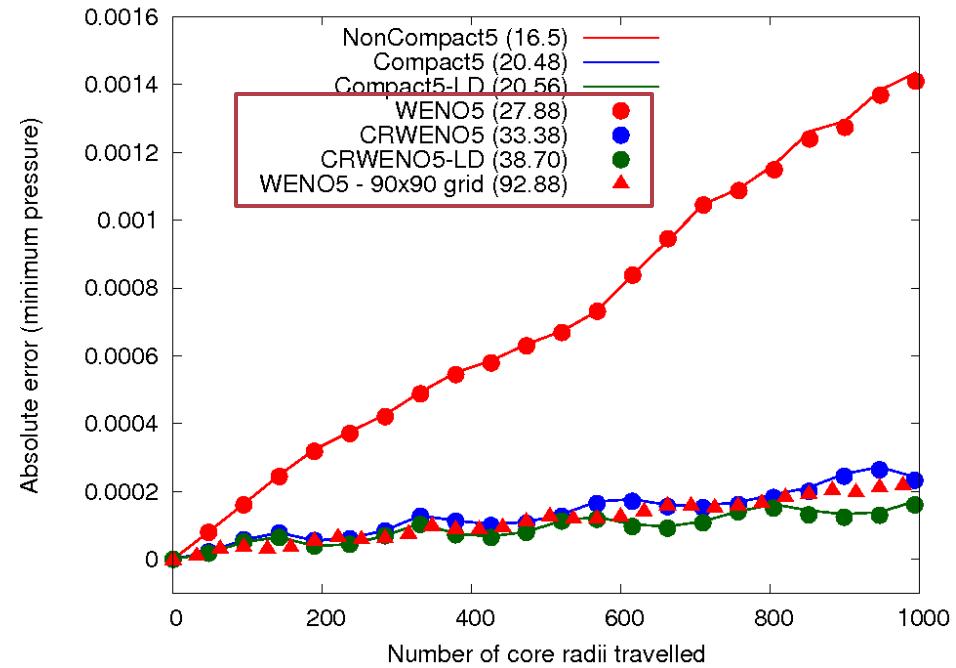
Periodic advection of a triangular wave after 100 cycles (160 point mesh)

Computational Efficiency on Single Processor (Inviscid Euler Equations)

Entropy Wave Advection



Isentropic Vortex Convection



CRWENO5 more efficient for component-wise reconstruction (scalar tridiagonal systems) but not for a characteristic-based reconstruction (block tridiagonal system)



Parallel Implementations of Non-Linear Compact Schemes (Distributed Memory)

Treat sub-domain boundary as physical boundary

- Decouple system of equations across processors
- *Chao, Haselbacher & Balachandar (JCP, 2009)*: Hybrid Compact-WENO scheme – Used the **5th order WENO scheme** at the MPI boundaries
- *Kim & Sandberg (Comp. & Fluids, 2012)*: Nonlinear compact scheme – Used **interior-biased compact schemes** at the MPI boundaries
- **Drawback:** Numerical properties of the scheme function of number of processors
- **Good for small number of processors, numerical errors grow or spectral resolution falls as number of processors increase for same problem size**

Parallel Implementation of the Thomas Algorithm

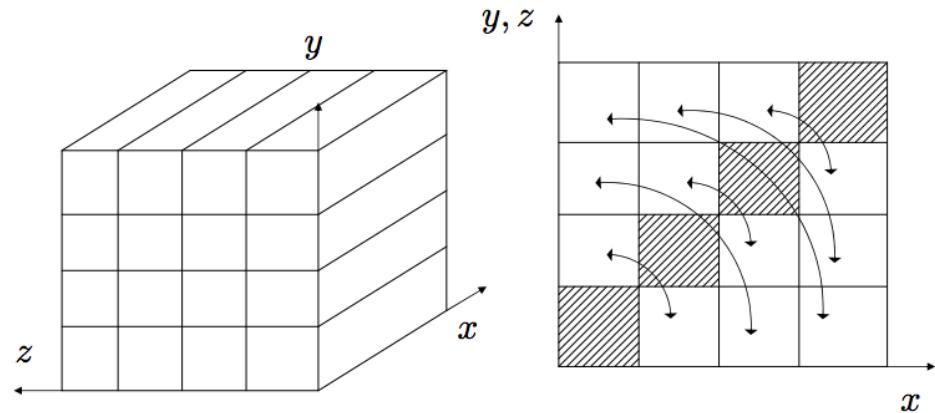
- **Pipelined Thomas Algorithm** (PTA) (*Povitsky & Morris, JCP, 2000*): Used a complicated static schedule to use idle times of processors to carry out computations – **Trade-off between computation & communication efficiencies**
- **Parallel Diagonally Dominant** (PDD) (*Sun & Moitra, NASA Tech. Rep., 1996*): Solve a perturbed linear system that introduces an error due to assumption of diagonal dominance
- Other implementations of tridiagonal solvers not applied to compact schemes
 - **Increased mathematical complexity compared to the serial Thomas algorithm**



Parallel Implementations of Non-Linear Compact Schemes (Distributed Memory)

Data Transposition

- Transpose pencils of data such that entire system of equations is collected on one processor.
- Huge communication cost



Existing approaches do not scale well!

Shankar Ghosh, Ph.D. Thesis, University of Minnesota, 2008

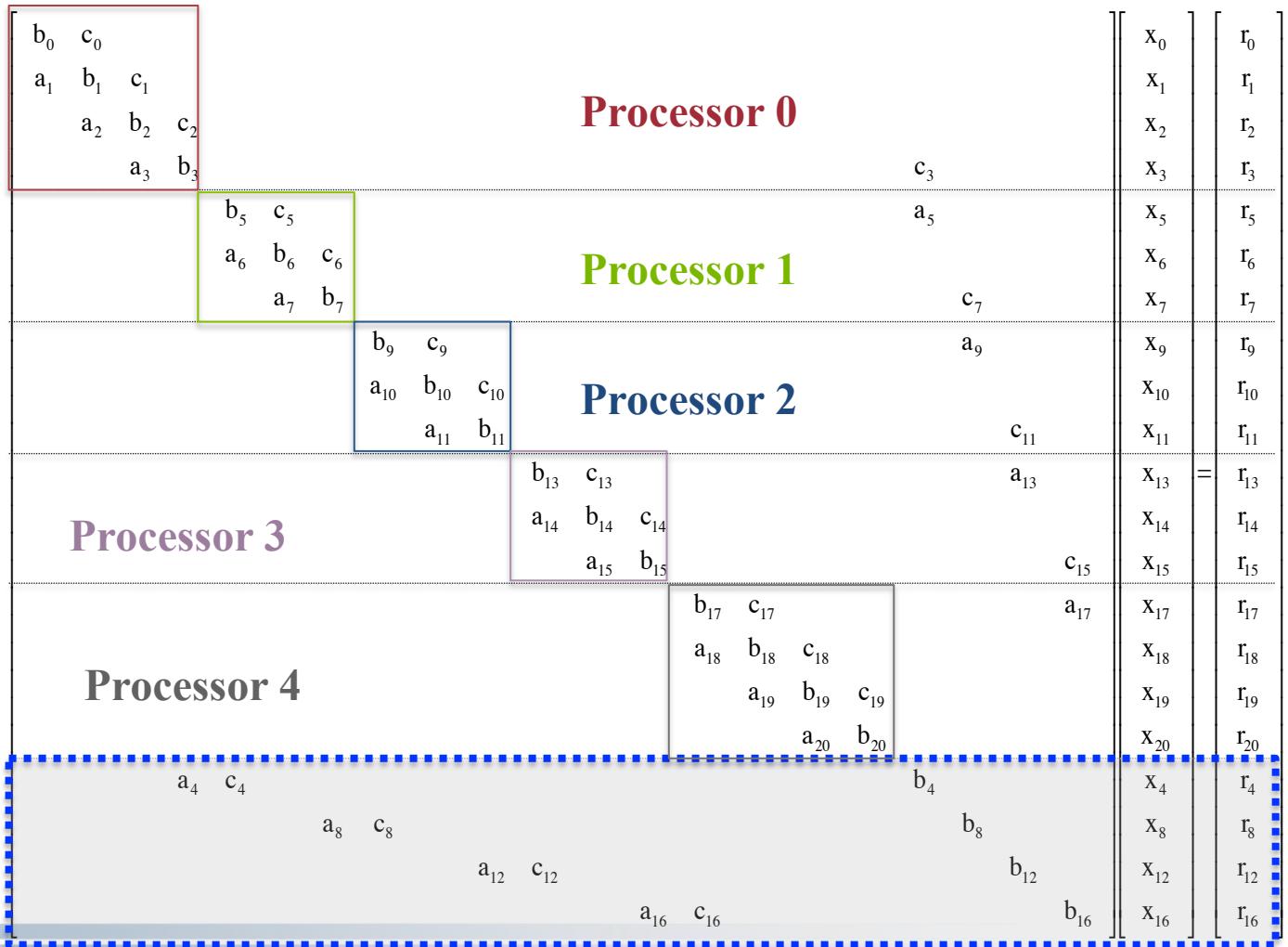
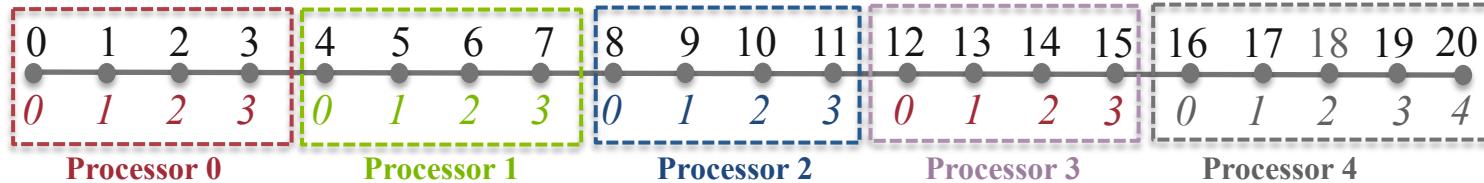
Our approach:

- Avoid parallelization-induced approximations or errors
- No complicated schedule of computations
- Mathematical complexity comparable to serial Thomas algorithm
- Overall scheme retains higher computational efficiency observed on single processors for practical subdomain sizes.

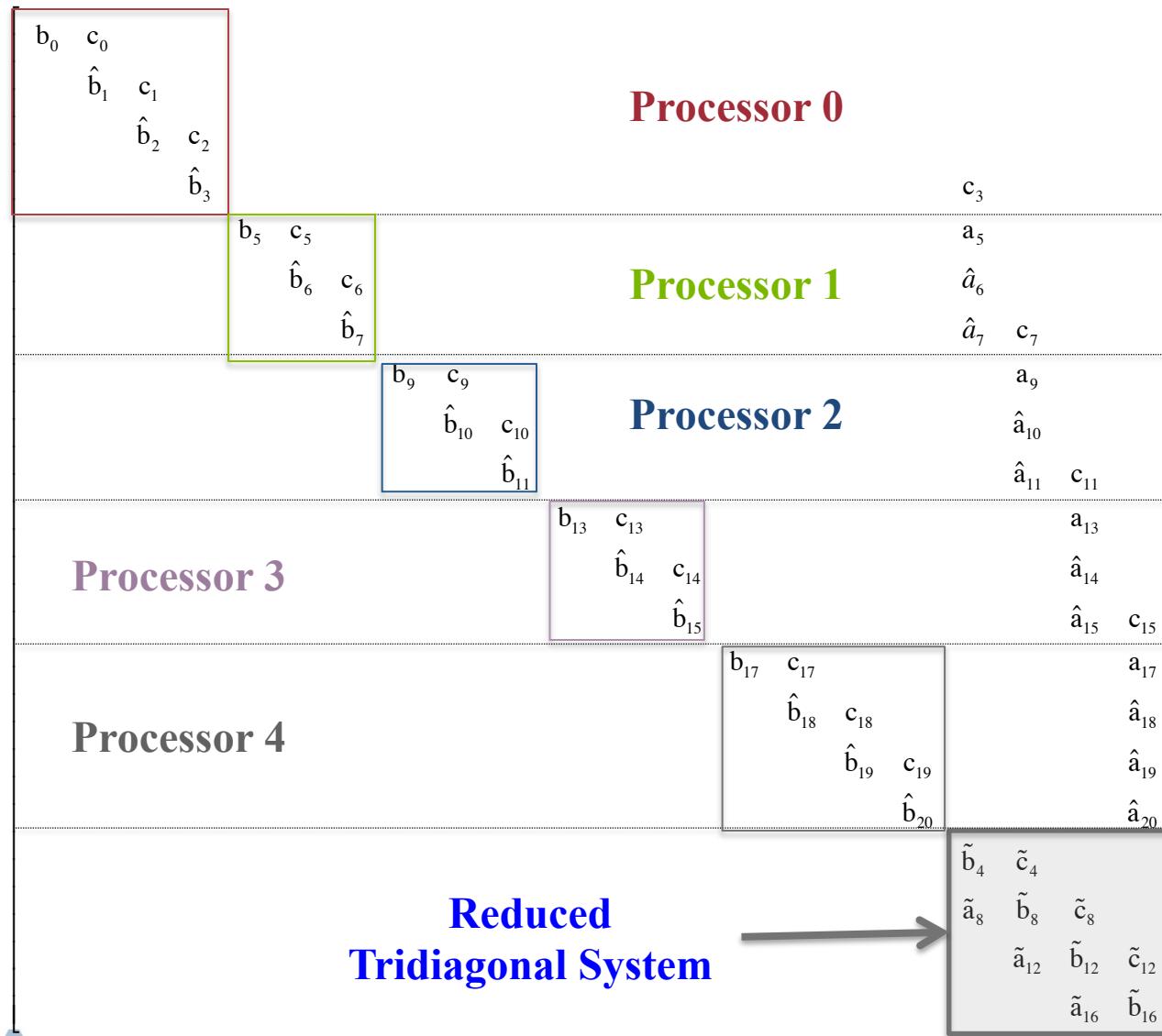


Parallel tridiagonal solver based on iterative sub-structuring

Parallel Tridiagonal Solver (1)



Parallel Tridiagonal Solver (2)



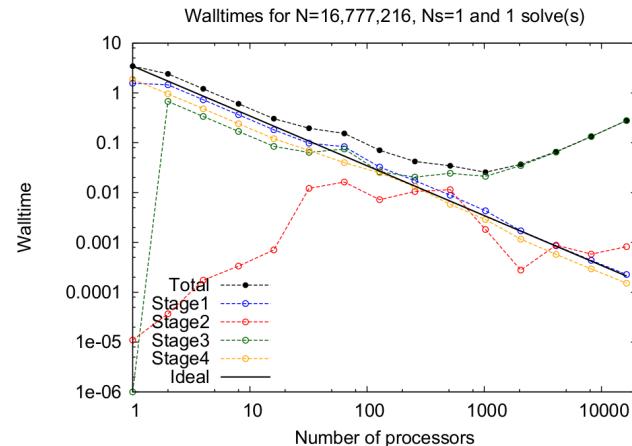
Elimination of interior rows on each process results in a reduced system of equations of size $p-1$ (p being the number of processors)

Each processor (except rank 0) has 1 row of the reduced system.

Parallel Tridiagonal Solver (3)

Solution of reduced system is critical to scalability

- One row on each processor → **Communication intensive**
- **Direct/Exact** solutions to the reduced system
 - **Cyclic Reduction / Recursive-Doubling Algorithm**
 - **Gather-and-Solve** – Gather the reduced system on one processor, solve, scatter the solution back (For multiple systems, each processor handles a set of systems)
- **Do not scale well!** (High communication cost)



Iterative Sub-structuring Method

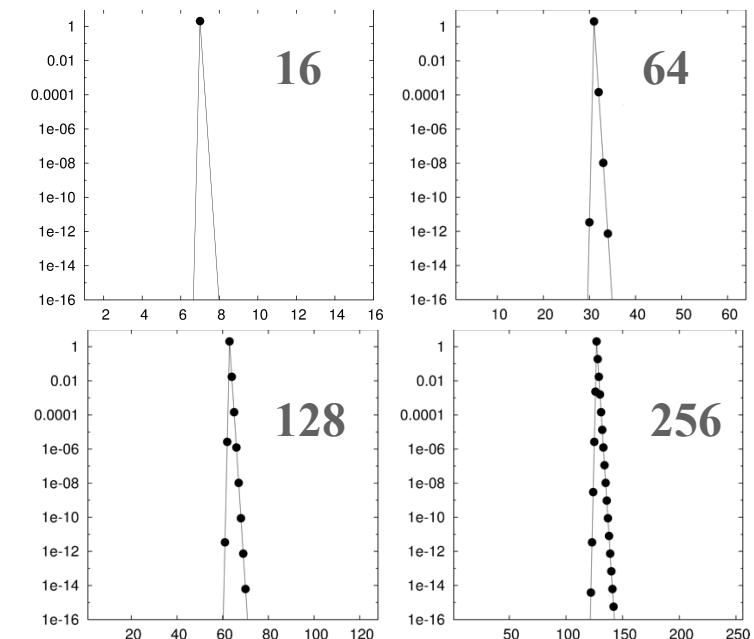
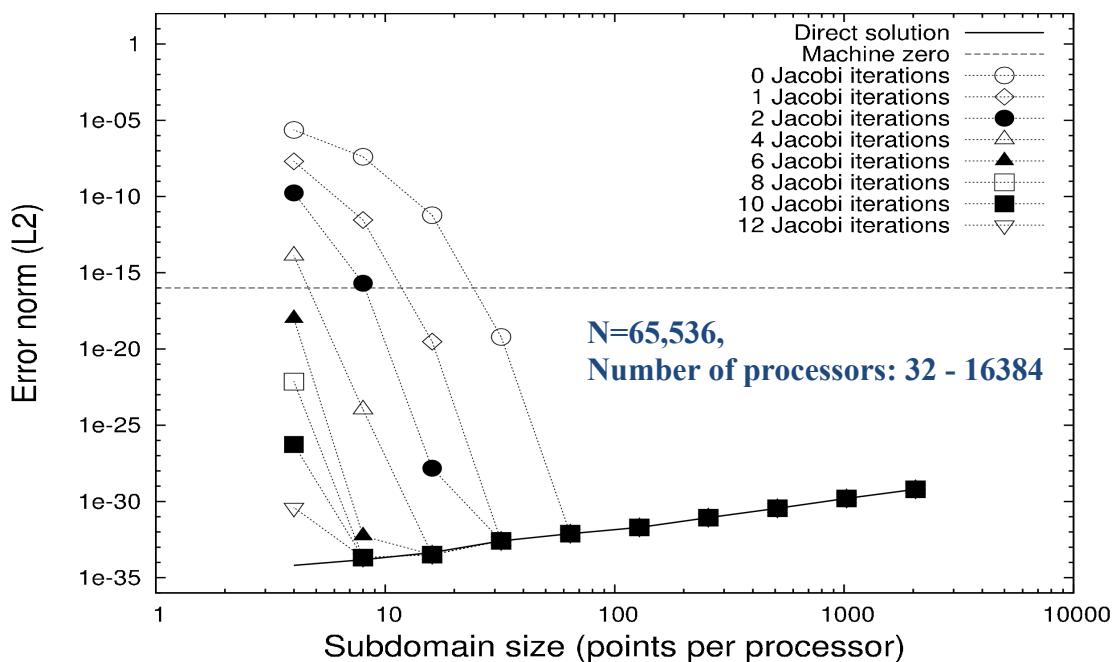
- Solve the reduced system using **Jacobi iterations**
- **But solve till machine zero convergence** → “Exact” solution (No parallelization-induced errors)
- Neighbor-to-neighbor communications required for each iteration
- Avoid collective communications by specifying *a priori* the number of iterations (instead of a norm-based exit criterion)



Iterative Solution of the Reduced System

Reduced system represents the **coupling between the first interface of every subdomain** through an approximation to a **hyperbolic flux**.

- For large subdomain sizes, **very diagonally dominant**
- **Diagonal dominance decreases as sub-domain size grows smaller** (Number of processors increase for same problem size)



Non-machine-zero elements of an arbitrarily chosen column of the inverse of the reduced system ($N=1024$)

Number of Jacobi iterations required for an “exact” solution increases as sub-domain size grows smaller.

→ Cost of the tridiagonal solver increases

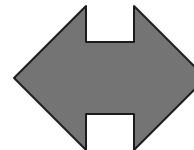
Performance Analysis

Governing Equations: Inviscid Euler Equations

Platform: ALCF Vesta (IBM BG/Q)

CRWENO5 scheme on N points per dimension

- On a single processor, CRWENO5 is more efficient.
- Cost of the tridiagonal solver increases as number of processors increases
- At a **critical sub-domain size**, the CRWENO5 becomes less efficient than the WENO5 scheme.



WENO5 scheme on fN points per dimension

- $f > 1$ ($f \sim 1.5$ for smooth solutions) – WENO5 yields solutions of comparable accuracy/resolution with f times more points
- Non-compact scheme, so almost ideal scalability expected.

Same number of processors for the CRWENO5 scheme (on N points) and the WENO5 scheme (on fN points)

Given p processors, is it faster to obtain a solution of given accuracy/resolution with the WENO5 or CRWENO5 scheme?

Performance Comparison for a Smooth Problem

Periodic Advection of a Sinusoidal Density Wave

1D

WENO5 yields solutions with comparable accuracy with **~1.5 times as many points**

2D

WENO5 yields solutions with comparable accuracy with **~ $1.5^2 = 2.25$ times as many points**

- **Time step size Δt is taken the same** for the CRWENO5 scheme on N points and the WENO5 scheme on fN points because of the linear stability limit.
- Solutions are obtained after one cycle over the periodic domain with the SSP-RK3 scheme
- It is verified that the **errors are exactly the same for the various number of processors** considered → There are no parallelization errors (Number of Jacobi iterations are specified *a priori* to ensure this)

Effect of Dimensionality

- The **factor by which the grid needs for WENO5 needs to be refined** to yield comparable solutions as CRWENO5 is **f^D** (f times per dimension)
- Several tridiagonal systems are solved for multi-dimensional problems → Higher arithmetic density → **Cost increases sub-linearly with number of systems**

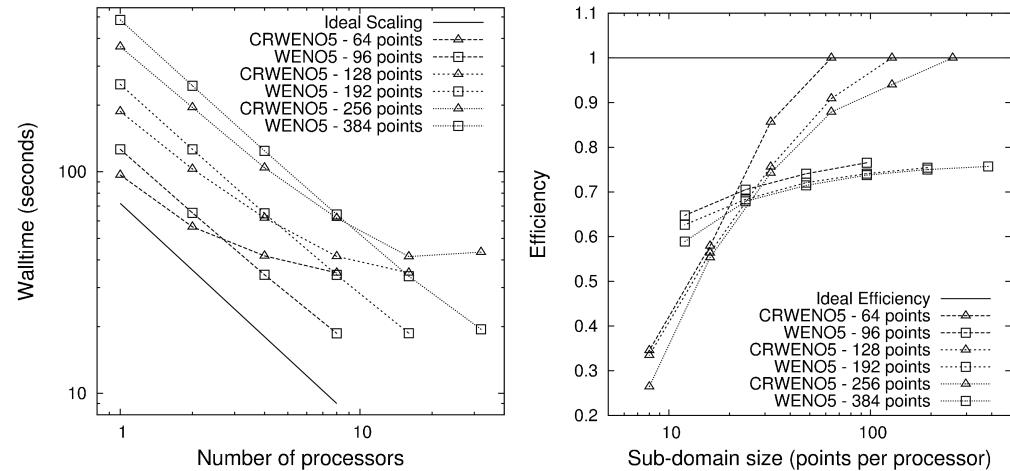


Performance Analysis for a Smooth Problem

1D

Cases: Number of grid points and number of processors considered

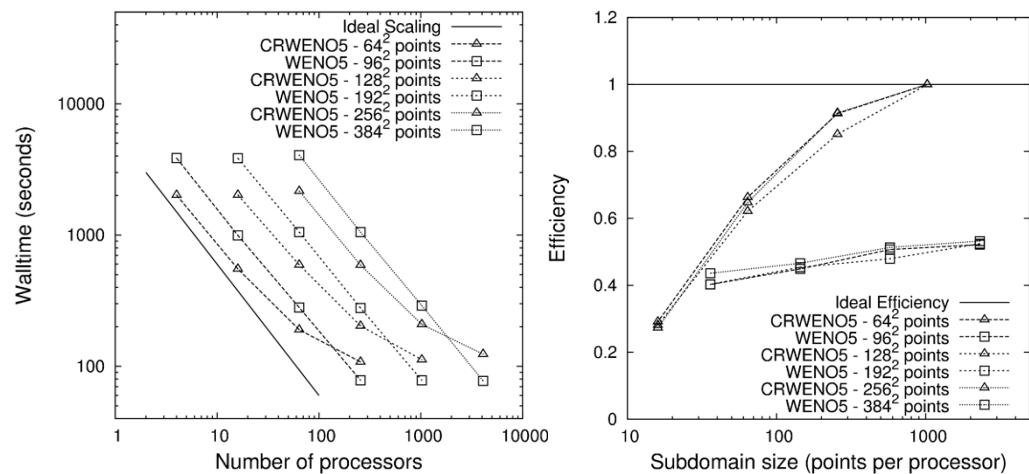
64 (96)	1	2	4	8		
128 (192)	1	2	4	8	16	
256 (384)	1	2	4	8	16	32



2D

Cases: Number of grid points and number of processors considered

64^2 (96 ²)	4	16	64	256
128^2 (192 ²)	16	64	256	1024
256^2 (384 ²)	64	256	1024	4096



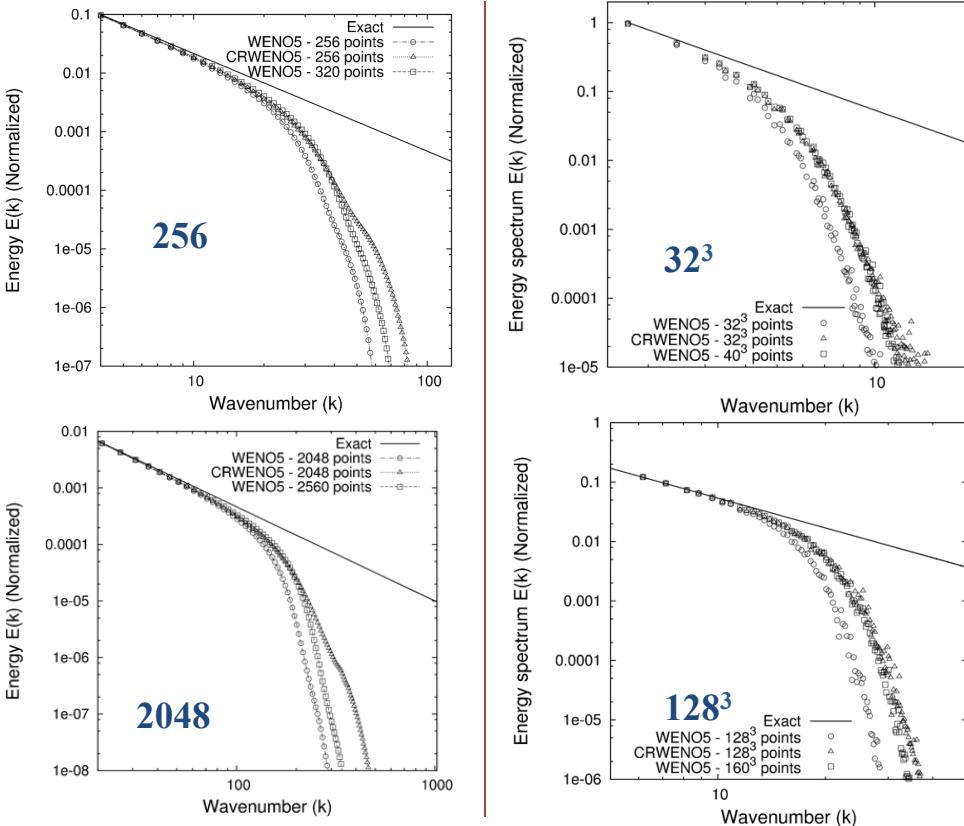
Note: Critical sub-domain size is insensitive to global problem size



Performance Comparison for a Non-Smooth Problem

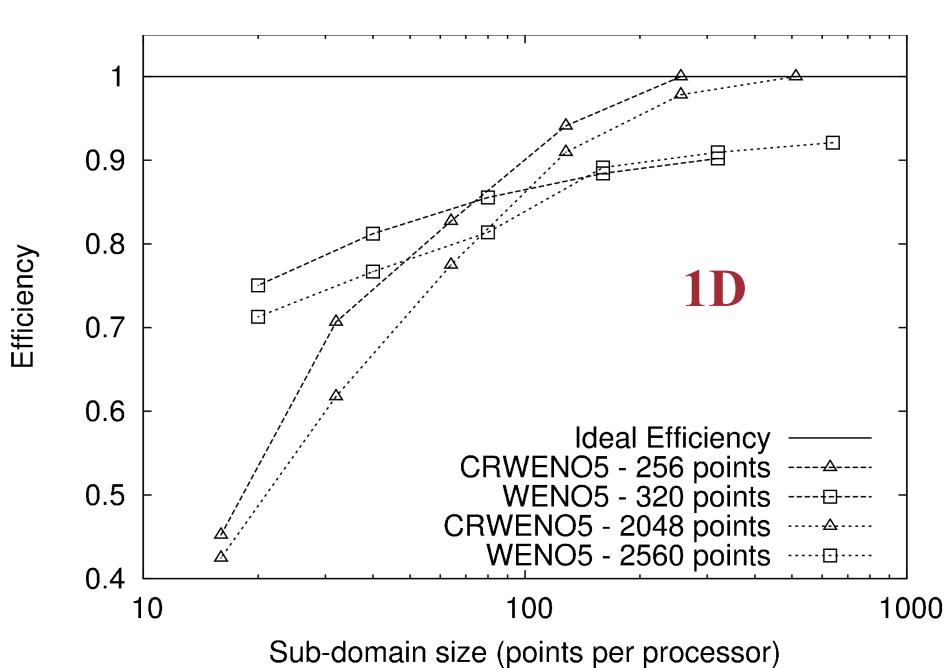
Periodic Advection of a Density Fluctuations (1D and 3D)

- Density is specified as the sum of sine waves with all grid-supported wavelengths
- Amplitude (as a function of wavelength) is representative of turbulent flows $A(k) \sim k^{-(5/3)}$
- Although solution is smooth, non-linear weights are not optimal

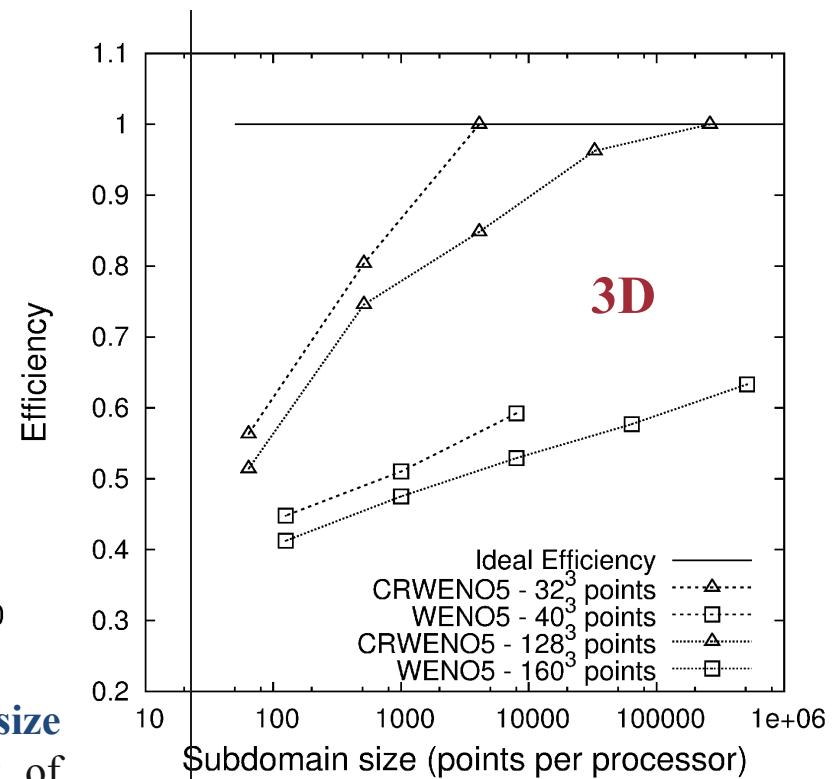


- Solutions obtained after 1 cycle over the periodic domain with the SSP-RK3 scheme
- Problems tried at **two different grid resolution** to demonstrate insensitivity of critical sub-domain size to global problem size.
- In 1D, WENO5 gives comparable resolution on grid with $\sim 1.25x$ more points
- In 2D, WENO5 gives comparable resolution on grid with $\sim 1.25^3x (\sim 2x)$ more points

Performance Comparison for a Non-Smooth Problem



1D



3D

- For both 1D and 3D, **efficiency vs. sub-domain size insensitive to global problem size and number of processors**
- In 1D, there exists a **critical sub-domain size** below which WENO5 is more efficient
- In 3D, **CRWENO5 is more efficient than the WENO5** even with sub-domain sizes close to the smallest practically possible.

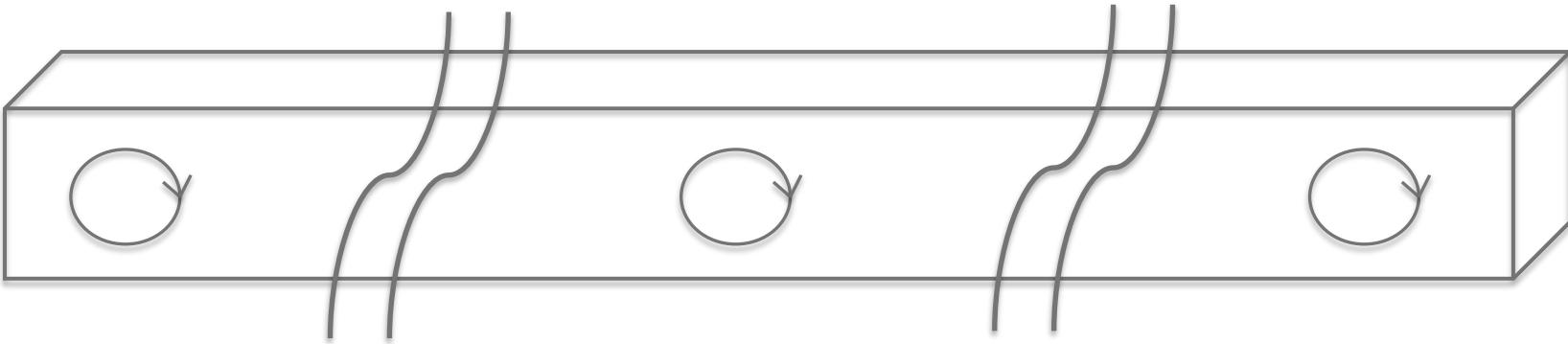


Smallest practical sub-domain size possible for 5th order schemes – 3 points per processor per dimension

Scalability Results for Benchmark Flow Problems

Isentropic Vortex Convection

- Long-term convection of a smooth flow feature over a large distance
- Convection over a long, 3D, cuboidal domain (instead of a smaller periodic domain)
- Periodic boundary conditions along z-direction to mimic 2D flow



Strong Scalability

Domain size: 1280 x 10 x 10 length units

CRWENO5 Grid: 8192 x 64 x 64 (~ 33 million points)

WENO5 Grid: 12288 x 96 x 96 (~ 113 million points)

Number of processors:

512 x 4 x 4 (8192) – Subdomain size 16^3 (24^3)

2048 x 16 x 16 (524288) – Subdomain size 4^3 (6^3)

Weak Scalability

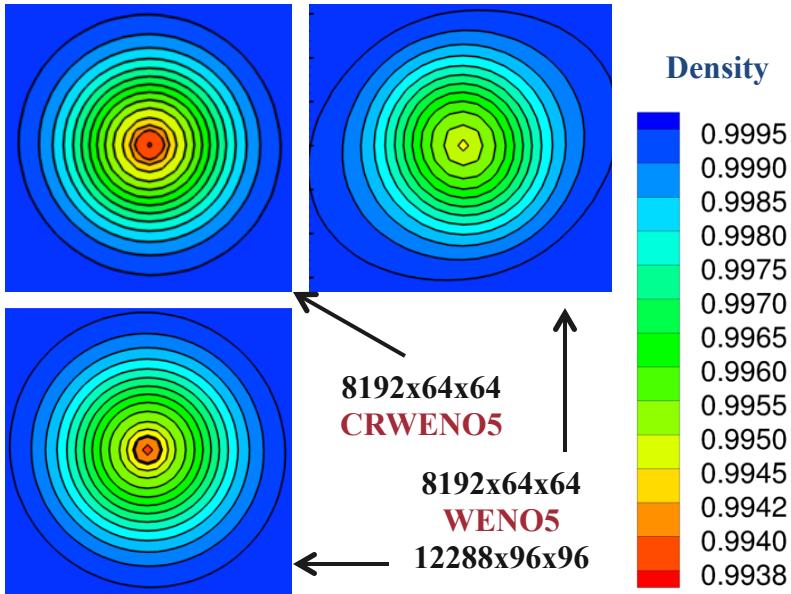
Subdomain size: 4^3 (CRWENO5), 6^3 (WENO5)

Domain:

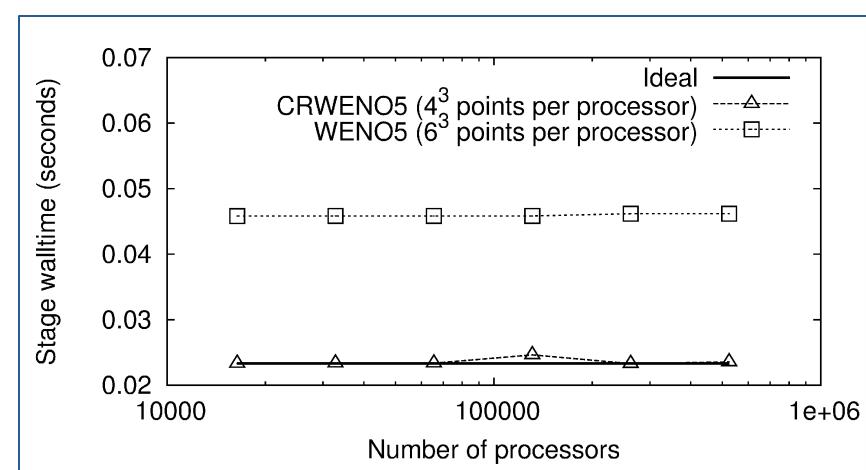
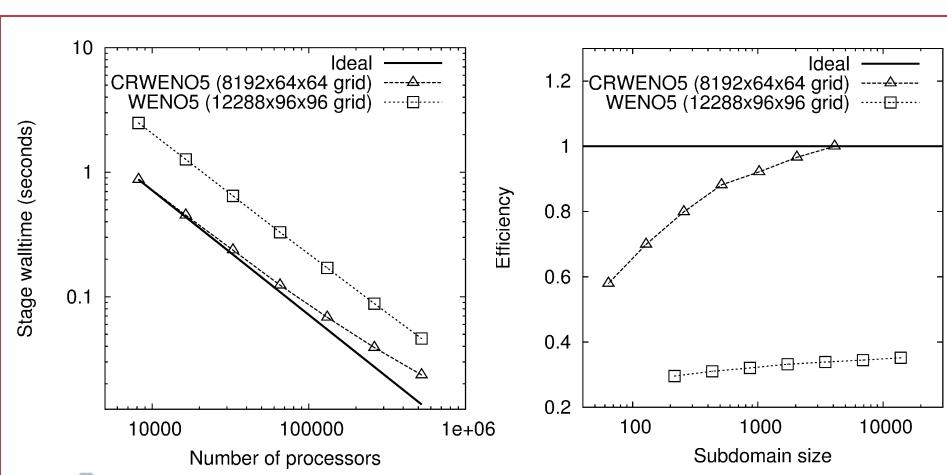
From 40 x 10 x 10 length units, 256 x 64 x 64 points
for CRWENO5, 64 x 16 x 16 processors

To 1280 x 10 x 10 length units, 8192 x 64 x 64 points
for CRWENO5, 2048 x 16 x 16 processors

Scalability Results for Benchmark Flow Problems



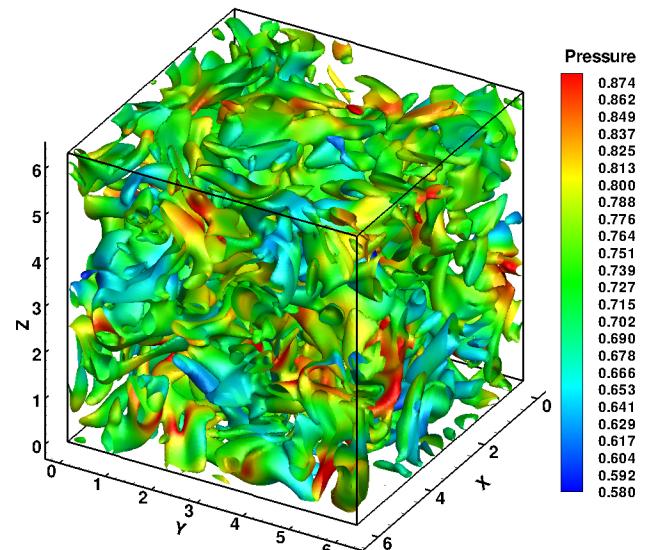
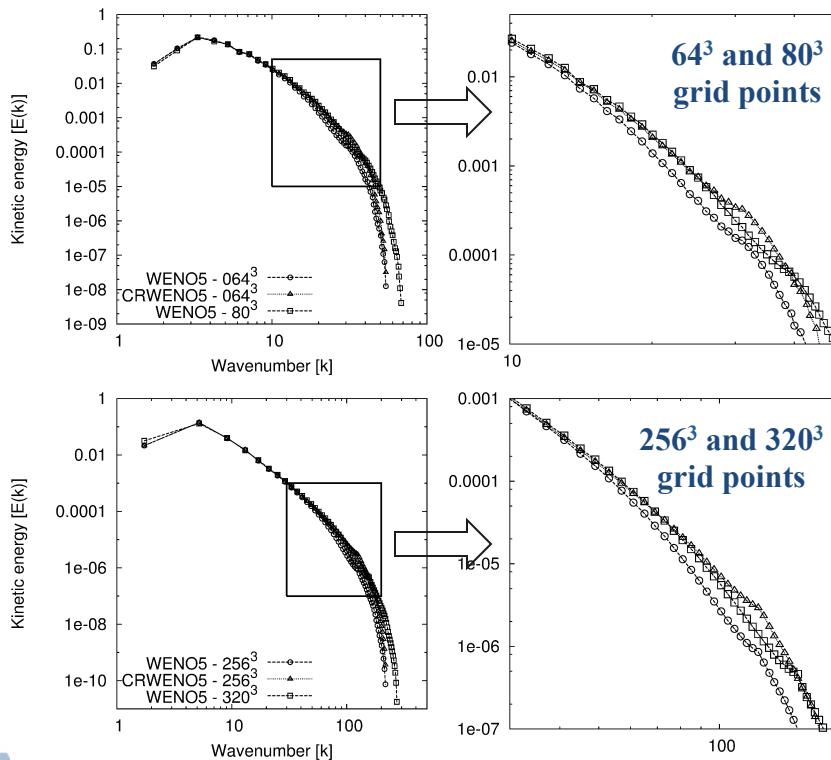
- Verified that WENO5 yields a solution of comparable accuracy on a grid with $\sim 1.53x$ ($\sim 3.4x$) more grid points
- ALCF/Mira (IBM BG/Q)
- **Strong Scaling:** At very small subdomain sizes, CRWENO5 does not scale as well, yet is more efficient / has lower absolute walltime
- **Weak Scaling:** CRWENO5 shows excellent weak scaling



Scalability Results for Benchmark Flow Problems

Implicit LES of Isotropic Turbulence Decay

- Benchmark turbulent flow problem characterized by energy transfer from large to small length scales
- Non-linear weights are **not optimal** and thus, the CRWENO5 and WENO5 schemes are not at their best accuracy



- Problem solved at two sets of grid resolutions – 64^3 & 80^3 , and 256^3 & 320^3
- Solutions obtained with the 4-stage, 4th order RK scheme
- WENO5 gives **comparable resolution** as CRWENO5 on grids with $\sim 1.25^3 \times$ ($\sim 2x$) more points.
- Scaling results obtained on ALCF/Mira (IBM BG/Q)

Scalability Results for Benchmark Flow Problems

Strong Scalability

Domain size: Cube of length 2π

CRWENO5 Grid: 256^3 (~ 16 million points)

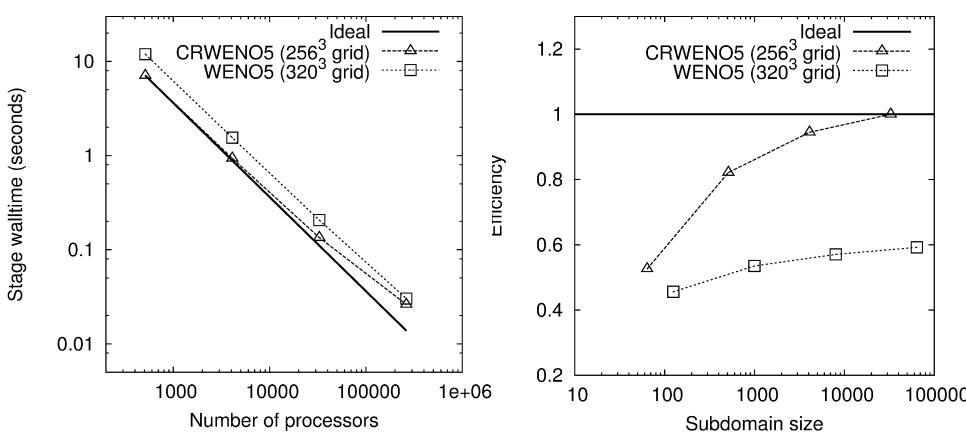
WENO5 Grid: 320^3 (~ 32 million points)

Number of processors:

8^3 (512) – Subdomain size 32^3 (40^3)

64^3 (262144) – Subdomain size 4^3 (5^3)

At very small subdomain sizes, CRWENO5 does not scale as well, yet absolute wall time (slightly) lower (higher efficiency)



Weak Scalability

Subdomain size: 4^3 (CRWENO5), 5^3 (WENO5)

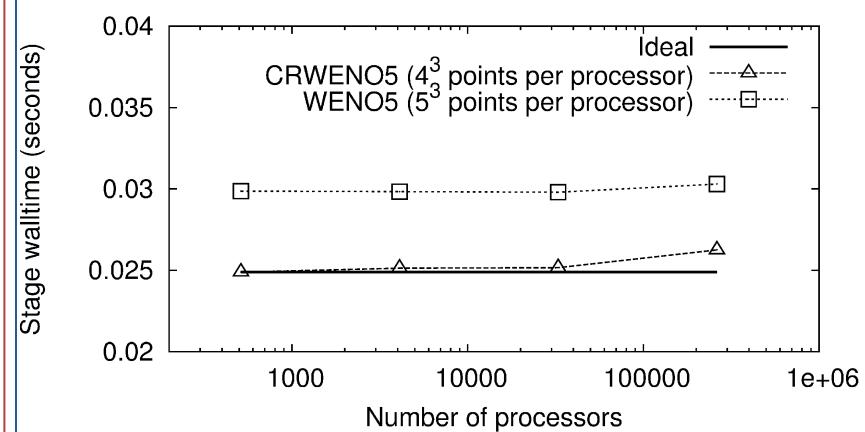
Domain: Cube of length 2π

CRWENO5 Grid (WENO5 grid is 1.25^3 x):

From 32^3 points, 8^3 (512) processors

To 256^3 points, 64^3 (262144) processors

CRWENO5 scales well as global domain size and number of processors increase



Conclusions and Observations



- **Compact-Reconstruction WENO Schemes**
 - Applies the WENO algorithm (solution-dependent interpolation) to compact schemes
 - **High spectral resolution** – Improved resolution of small length scales
 - **Non-oscillatory interpolation** across discontinuities and steep gradients
- **Parallel Implementation on Distributed Memory Platforms**
 - Based on an **iterative sub-structuring approach** to the tridiagonal system of equations
 - **No parallelization-induced errors** (however, need *a priori* estimate on the number of Jacobi iterations for the reduced system)
 - Avoids collective communication → **Excellent weak scaling**
 - Good strong scaling compared to a non-compact scheme; at very small subdomain sizes, **retains higher parallel efficiency** despite relatively poorer scaling

Future Work

- **Near Future (Practical and Interesting Applications)**
 - DNS of shock-turbulence interaction and shock-turbulent boundary layer interaction
 - Apply the CRWENO5 scheme to benchmark atmospheric flow problems and compare solution and scalability with spectral element methods (popular in that community)
 - Implement the CRWENO5 scheme with ARKIMEX time-integration schemes (PETSc) and study the overall scalability
 - Implement this implementation of the CRWENO5 scheme in an existing, validated Navier-Stokes solver and apply to practical flow problems
- **Distant Future**
 - Use threads for further speedup and exploit the hybrid architecture of most HPC platforms
 - Study the scalability on non-BG/Q platforms, with more noisy communication networks



Thank you!

Acknowledgements

- U.S. Army MAST CTA Center for Microsystem Mechanics (at UMD)
- U.S. Department of Energy, Office of Science, Advanced Scientific Computing Research (at Argonne National Laboratory)
- Argonne Leadership Computing Facility
- Dr. Paul Fischer (Argonne National Laboratory)

