

# Modeling and Control of Mechatronics Systems

## Spring 2018-2019

### HW I Solution

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## Contents

1	Problem # 1	1
2	Problem # 2	4
3	Problem # 3	11
4	Problem # 4	13
5	Problem # 5	17

# 1 Problem # 1

- Section -a

Using Lagrange's formula

$$\mathcal{T} = \frac{1}{2}m\dot{z}^2$$

$$\mathcal{V} = \frac{1}{2}k_1z^2 + \frac{1}{2}k_2(z-x)^2$$

$$\mathcal{D} = \frac{1}{2}b_1\dot{z}^2 + \frac{1}{2}b_2(\dot{y}-\dot{x})^2$$

$$\mathcal{L} = \mathcal{T} - \mathcal{V} = \frac{1}{2}[m\dot{z}^2 - k_1z^2 - k_2(z-x)^2]$$

where T,V and D stands for the kinetic, potential and dissipated energy of the system, respectively. There are 3 generalized coordinates of the system which are  $q_1 = z$ ,  $q_2 = x$  and  $q_3 = y$ .

$$\frac{d}{dt}\left(\frac{\partial\mathcal{L}}{\partial\dot{q}}\right) - \frac{\partial\mathcal{L}}{\partial q} + \frac{\partial\mathcal{D}}{\partial\dot{q}} = Q_{ext}$$

$$\frac{\partial\mathcal{L}}{\partial z} = -k_1z - k_2(z-x)$$

$$\frac{\partial\mathcal{L}}{\partial\dot{z}} = m\dot{z} \quad \text{and} \quad \frac{\partial\mathcal{D}}{\partial\dot{z}} = b_1\dot{z}$$

$$\frac{d}{dt}\left(\frac{\partial\mathcal{L}}{\partial\dot{z}}\right) - \frac{\partial\mathcal{L}}{\partial z} + \frac{\partial\mathcal{D}}{\partial\dot{z}} = m\ddot{z} + (k_1 + k_2)z - k_2x + b_1\dot{z} = 0$$

For the 2<sup>nd</sup> coordinate x;

$$\frac{\partial\mathcal{L}}{\partial x} = k_2(z-x)$$

$$\frac{\partial\mathcal{L}}{\partial\dot{x}} = 0 \quad \text{and} \quad \frac{\partial\mathcal{D}}{\partial\dot{x}} = -b_2(\dot{y}-\dot{x})$$

$$\frac{d}{dt}\left(\frac{\partial\mathcal{L}}{\partial\dot{x}}\right) - \frac{\partial\mathcal{L}}{\partial x} + \frac{\partial\mathcal{D}}{\partial\dot{x}} = k_2(x-z) - b_2(\dot{y}-\dot{x}) = 0$$

For the 3<sup>rd</sup> coordinate y;

$$\frac{\partial \mathcal{L}}{\partial y} = 0$$

$$\frac{\partial \mathcal{L}}{\partial \dot{y}} = 0 \quad \text{and} \quad \frac{\partial \mathcal{D}}{\partial \dot{y}} = b_2(\dot{y} - \dot{x})$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{y}} \right) - \frac{\partial \mathcal{L}}{\partial y} + \frac{\partial \mathcal{D}}{\partial \dot{y}} = b_2(\dot{y} - \dot{x}) = F_{y_{ext}}$$

Using these equations, transfer function can be derived as follows taking the Laplace Transforms (under the assumption of zero initial conditions);

$$Z(s)[ms^2 + b_1s + k_1 + k_2] = X(s)k_2$$

$$X(s)[sb_2 + k_2] = Z(s)k_2 + b_2sY(s)$$

$$b_2sY(s) = b_2sX(s) + \mathbf{L}(F_{y_{ext}}) \implies b_2sY(s) = b_2sX(s) \quad (\text{Assuming Zero force exerted to the system})$$

$$\frac{Z(s)}{Y(s)} = \frac{Z(s)}{X(s)} \frac{X(s)}{Y(s)} = \frac{k_2}{ms^2 + b_1s + k_1 + k_2} * \frac{1}{1} = \frac{k_2}{ms^2 + b_1s + k_1 + k_2}$$

for m=2, b1=b2=16, k1=k2=8

$$\frac{Z(s)}{Y(s)} = \frac{8}{2s^2 + 16s + 16} = \frac{4}{s^2 + 8s + 8}$$

I definitely have a mistake here as this system must be in the order of 3 as there is a damper at the free end. So I will continue with the right answer for the poles and zeros and step response of the system.

$$\frac{Z(s)}{Y(s)} = \frac{4s}{s^3 + 8.5s^2 + 12s + 2}$$

- Section -b

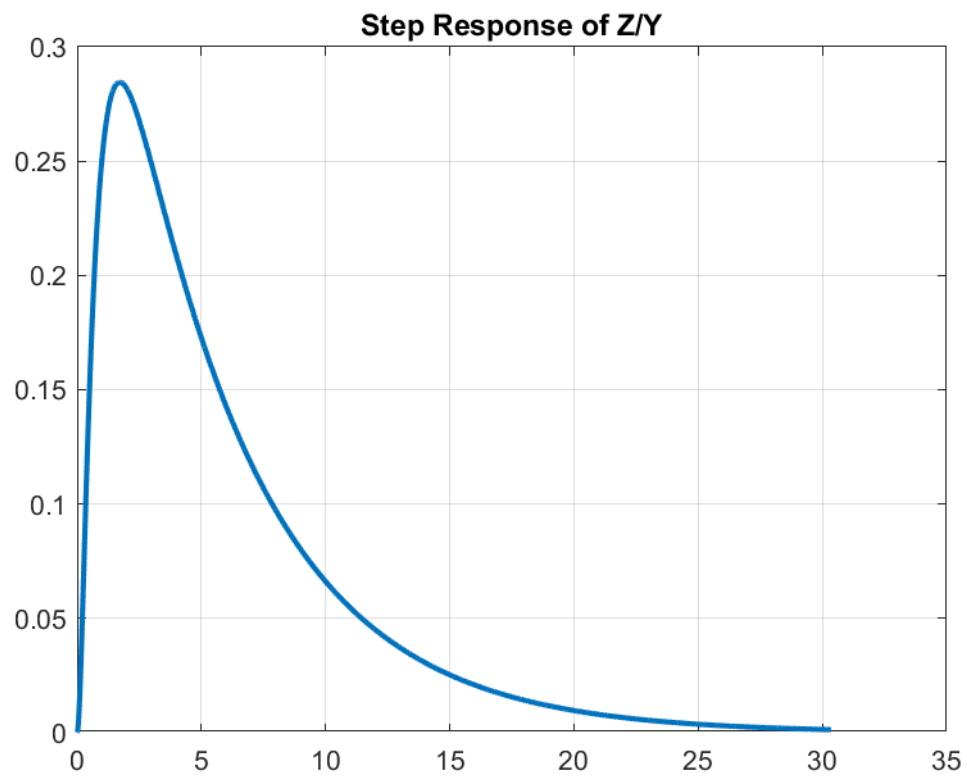


Figure 1: Step Response

- Section -c

Pole-Zero table for the system is given below

Pole Zero Table For Problem 1		
	Poles	Zeros
$\frac{Z}{Y}$	-6.7715	0
	-1.5363	
	-0.1923	

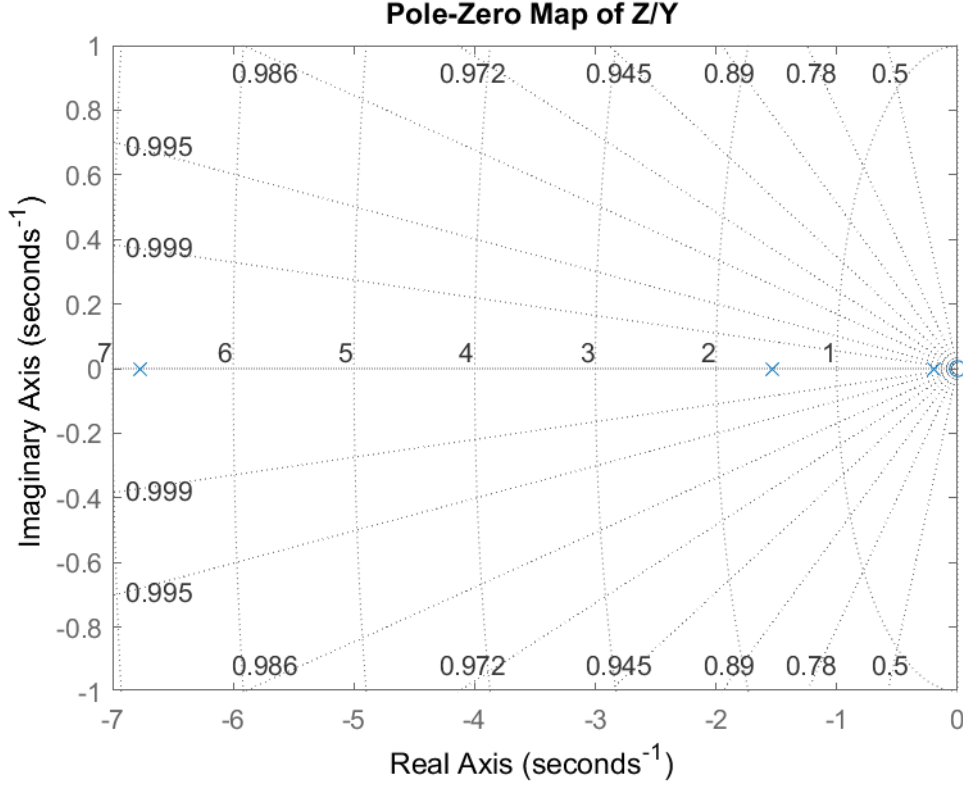


Figure 2: Pole-Zero Map of the system

All of the poles lie in the left hand side but there is a zero at the origin. system is stable but non-minimum phase.

## 2 Problem # 2

- **Section -a** Using Lagrange's formula

$$\mathcal{T} = \frac{1}{2}m_2\dot{y}^2 + \frac{1}{2}m_1\dot{y}^2$$

$$\mathcal{V} = \frac{1}{2}k_1(y_1 - y_0)^2 + \frac{1}{2}k_2(y_2 - y_1)^2$$

$$\mathcal{D} = \frac{1}{2}b(\dot{y}_2 - \dot{y}_1)^2$$

$$\mathcal{L} = \mathcal{T} - \mathcal{V} = \frac{1}{2}[m_2\dot{y}_2^2 + m_1\dot{y}_1^2 - k_1(y_1 - y_0)^2 - k_2(y_2 - y_1)^2]$$

where T,V and D stands for the kinetic, potential and dissipated energy of the system, respectively.

$$\frac{d}{dt}\left(\frac{\partial \mathcal{L}}{\partial \dot{q}}\right) - \frac{\partial \mathcal{L}}{\partial q} + \frac{\partial \mathcal{D}}{\partial \dot{q}} = Q_{ext}$$

Selecting  $q_1$  as  $y_1$  and  $q_2$  as  $y_2$  gives the following formulations,

$$\frac{\partial \mathcal{L}}{\partial q_1} = -k_1(y_1 - y_0) + k_2(y_2 - y_1)$$

$$\frac{\partial \mathcal{L}}{\partial \dot{q}_1} = m_1 \dot{y}_1 \quad \text{and} \quad \frac{\partial \mathcal{D}}{\partial \dot{q}_1} = -b(\dot{y}_2 - \dot{y}_1)$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_1} \right) - \frac{\partial \mathcal{L}}{\partial q_1} + \frac{\partial \mathcal{D}}{\partial \dot{q}_1} = m_1 \ddot{y}_1 + (k_1 + k_2)y_1 - k_2 y_2 - k_1 y_0 - b(\dot{y}_2 - \dot{y}_1) = 0$$

and for the second generalized coordinate  $q_2$

$$\frac{\partial \mathcal{L}}{\partial q_2} = -k_2(y_2 - y_1)$$

$$\frac{\partial \mathcal{L}}{\partial \dot{q}_2} = m_2 \dot{y}_2 \quad \text{and} \quad \frac{\partial \mathcal{D}}{\partial \dot{q}_2} = b(\dot{y}_2 - \dot{y}_1)$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_2} \right) - \frac{\partial \mathcal{L}}{\partial q_2} + \frac{\partial \mathcal{D}}{\partial \dot{q}_2} = m_2 \ddot{y}_2 + k_2(y_2 - y_1) + b(\dot{y}_2 - \dot{y}_1) = 0$$

Selecting the states such that  $x_1 = y_1$   $x_2 = \dot{y}_1$   $x_3 = y_2$   $x_4 = \dot{y}_2$  yields

$$\begin{aligned} m_1 \dot{x}_2 + (k_1 + k_2)x_1 - k_2 x_3 - k_1 u_0 - b(x_4 - x_2) &= 0 \\ m_2 \dot{x}_4 + k_2(x_3 - x_1) + b(x_4 - x_2) &= 0 \end{aligned}$$

In state space form

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{pmatrix} 0 & 1 & 0 & 0 \\ -\frac{k_1+k_2}{m_1} & -\frac{b}{m_1} & \frac{k_2}{m_1} & \frac{b}{m_1} \\ 0 & 0 & 0 & 1 \\ \frac{k_2}{m_2} & \frac{b}{m_2} & -\frac{k_2}{m_2} & -\frac{b}{m_2} \end{pmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{k_1}{m_1} \\ 0 \\ 0 \end{bmatrix} u(t)$$

$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{pmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{pmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \end{bmatrix} u(t)$$

- Section -b

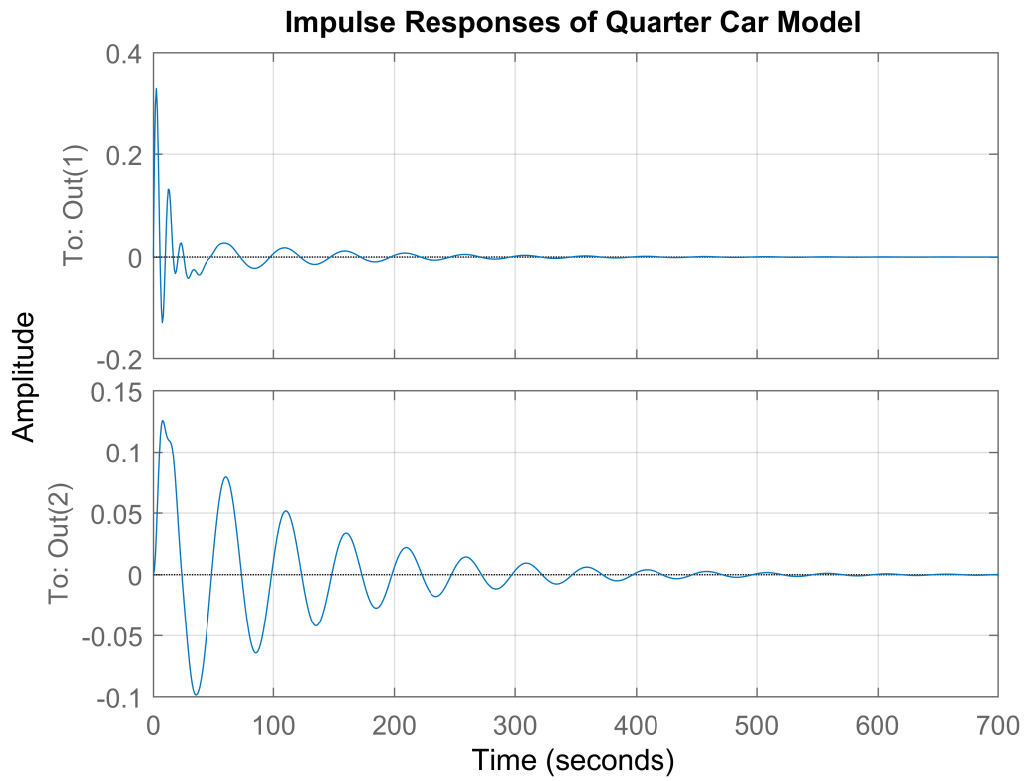


Figure 3: Impulse Responses (top- $y_1$  bottom- $y_2$ )

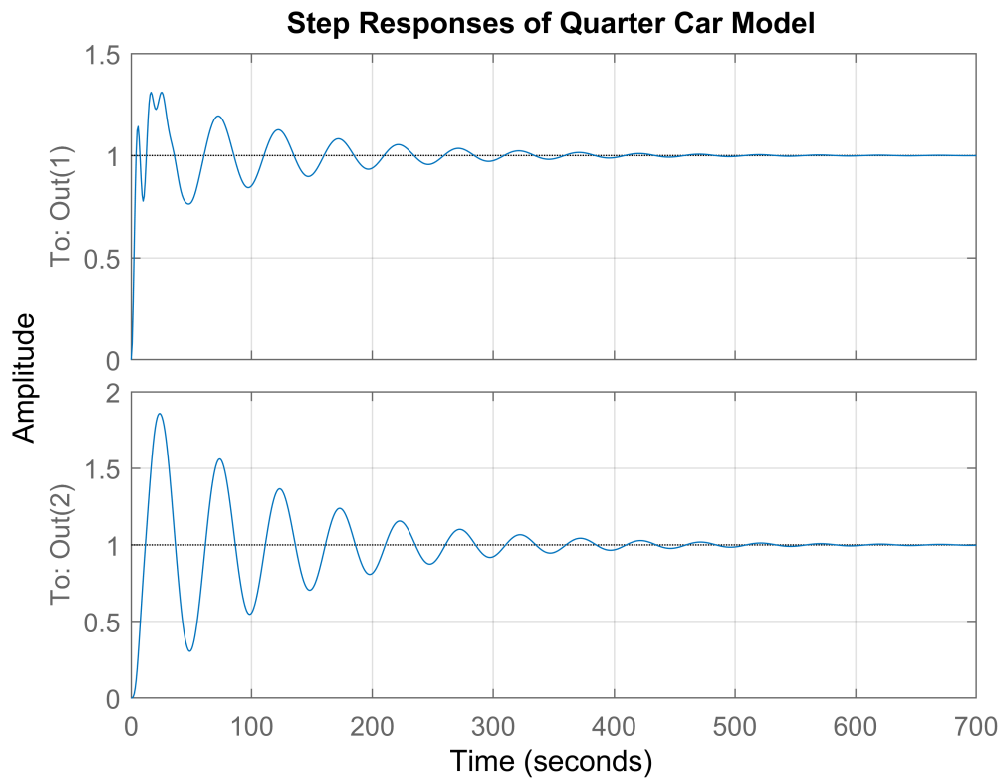


Figure 4: Step Responses (top- $y_1$  bottom- $y_2$ )

- **Section -c** By taking the Laplace Transform of the system equations with zero initial conditions

$$m_1 s^2 Y_1(s) + (k_1 + k_2) Y_1(s) - k_2 Y_2(s) - k_1 U(s) - bs(Y_2(s) - Y_1(s)) = 0$$

$$m_2 s^2 Y_2(s) + k_2(Y_2(s) - Y_1(s)) + bs(Y_2(s) - Y_1(s)) = 0$$

Rewriting these equation

$$Y_2(s)[m_2 s^2 + bs + k_2] = Y_1(s)[bs + k_2]$$

$$Y_1(s)[m_1 s^2 + bs + k_1 + k_2] = Y_2(s)[bs + k_2] + k_1 U(s)$$

$$\frac{Y_2(s)}{Y_1(s)} = \frac{bs + k_2}{m_2 s^2 + bs + k_2}$$

$$\frac{Y_1(s)}{U(s)} = \frac{k_1(m_2 s^2 + bs + k_2)}{m_1 m_2 s^4 + b(m_1 + m_2)s^3 + (k_1 m_2 + k_2 m_2 + k_2 m_1)s^2 + bk_1 s + k_1 k_2}$$

$$\frac{Y_2(s)}{U(s)} = \frac{k_1(k_2 + bs)}{m_1 m_2 s^4 + b(m_1 + m_2)s^3 + (k_1 m_2 + k_2 m_1 + k_2 m_2)s^2 + bk_1 s + k_1 k_2}$$

And finally, substituting the values of the parameters into these equations transfer functions of the system can be obtained as follows

$$\frac{Y_1(s)}{U(s)} = \frac{12500s^2 + 500s + 300}{50000s^4 + 12000s^3 + 19700s^2 + 500s + 300}$$

$$\frac{Y_2(s)}{U(s)} = \frac{500s + 300}{50000s^4 + 12000s^3 + 19700s^2 + 500s + 300}$$

With these transfer functions it is possible to separately recover the figures that obtained above through state space representation with Matlab



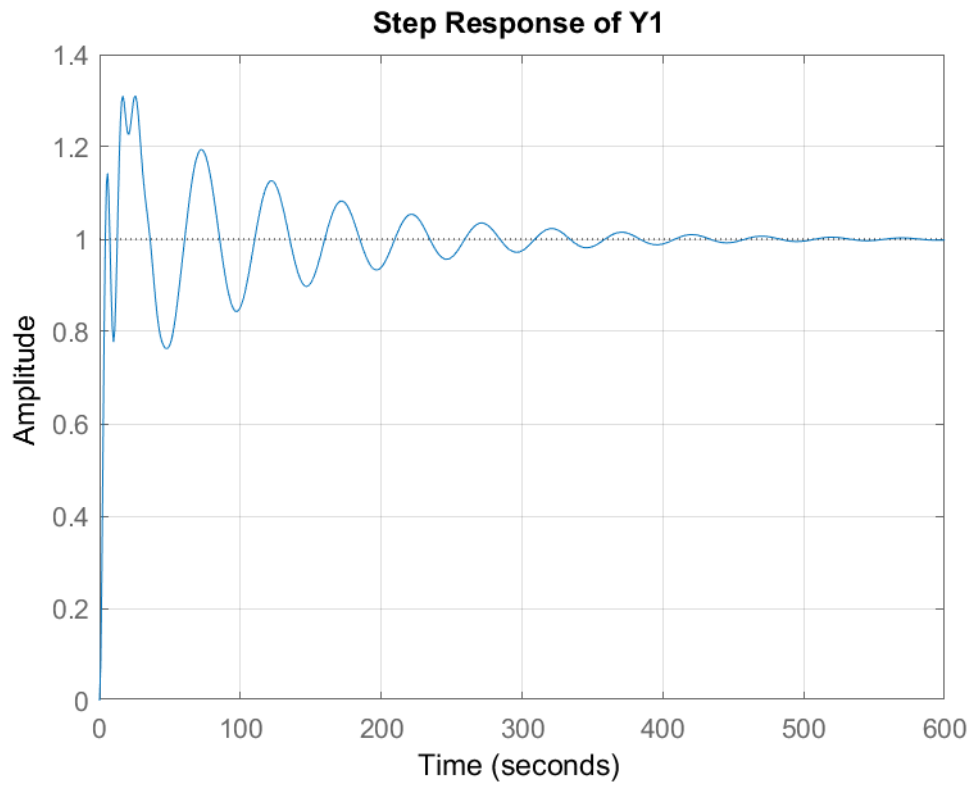


Figure 5: Step Response of the Output Y1

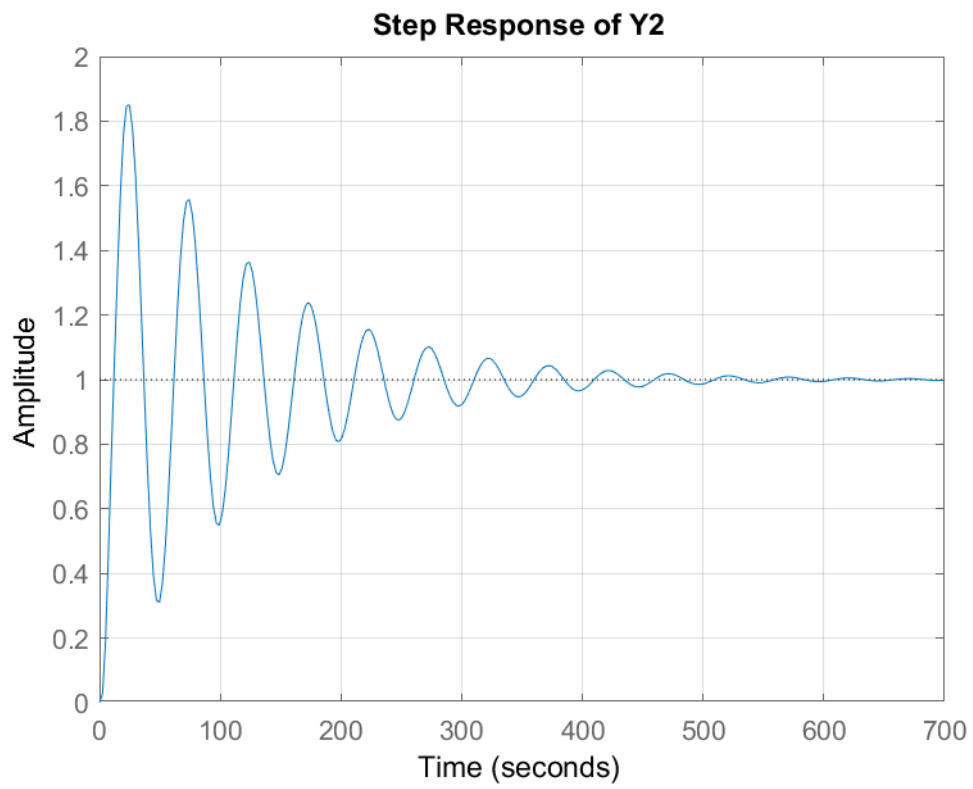
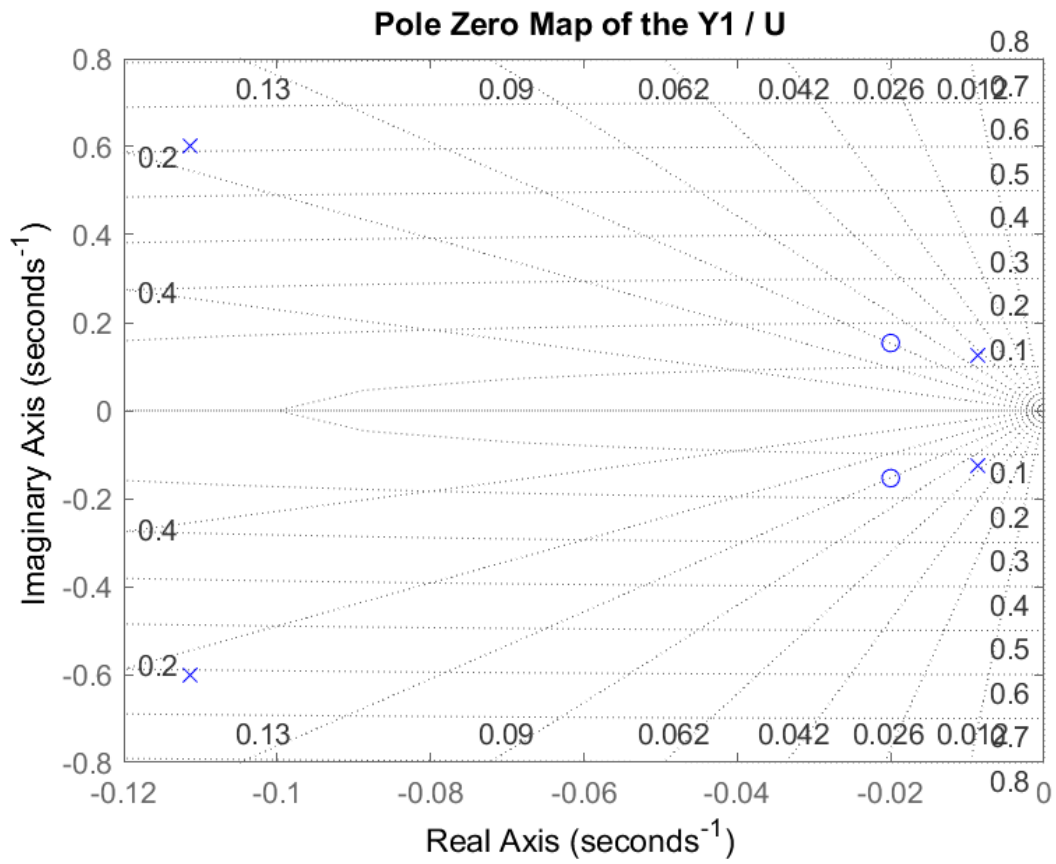
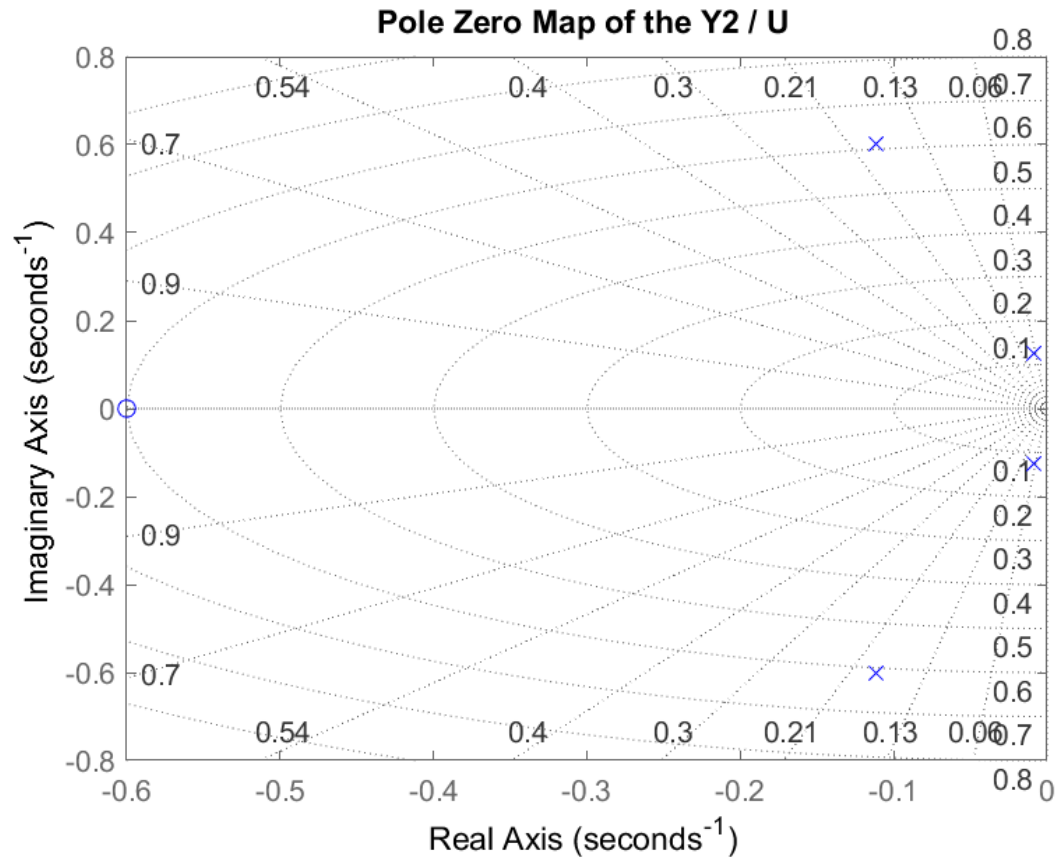


Figure 6: Step Response of the Output Y2

Pole-Zero table for the system is given below

Pole Zero Table For Quarter Car Model		
	Poles	Zeros
$\frac{y_1}{U}$	$-0.1114 + 0.6014i$	$-0.0200 + 0.1536i$
	$-0.1114 - 0.6014i$	$-0.0200 - 0.1536i$
	$-0.0086 + 0.1263i$	
	$-0.0086 - 0.1263i$	
$\frac{y_2}{U}$	$-0.1114 + 0.6014i$	$-0.6$
	$-0.1114 - 0.6014i$	
	$-0.0086 + 0.1263i$	
	$-0.0086 - 0.1263i$	





→ As all of the poles and zeros are in the Left Side of the Plane, system is stable and minimum phase.

### 3 Problem # 3

- Section -a

$$J = ml_3^2 \quad \mathcal{T} = \frac{ml_3^2 \dot{\theta}^2}{2} \quad \mathcal{V} = \frac{1}{2}(x - l_1 \sin(\theta))^2 + mgl_3 \cos(\theta)$$

Assuming small displacements  $\sin(\theta) \Rightarrow \theta$  and  $\cos(\theta) \Rightarrow 1$

$$\mathcal{D} = \frac{1}{2}c(l_2 \sin(\dot{\theta}))^2 \Rightarrow \mathcal{D} = \frac{1}{2}c(l_2 \dot{\theta})^2 \quad \mathcal{L} = \mathcal{T} - \mathcal{V}$$

$$\mathcal{L} = \frac{ml_3^2 \dot{\theta}^2}{2} - \frac{1}{2}k(x - l_1 \theta)^2 - mgl_3$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}} \right) - \frac{\partial \mathcal{L}}{\partial q} + \frac{\partial \mathcal{D}}{\partial \dot{q}} = Q_{ext}$$

$$\frac{\partial \mathcal{L}}{\partial \theta} = k(x - \theta l_1) \quad \frac{\partial \mathcal{L}}{\partial \dot{\theta}} = ml_3^2 \dot{\theta} \quad \text{and} \quad \frac{\partial \mathcal{D}}{\partial \dot{\theta}} = cl_2 \dot{\theta}$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{\theta}} \right) - \frac{\partial \mathcal{L}}{\partial \theta} + \frac{\partial \mathcal{D}}{\partial \dot{\theta}} = 0$$

$$ml_3^2 \ddot{\theta} - kl_1(x - \theta l_1) + cl_2^2 \dot{\theta} = 0$$

$$ml_3^2 \ddot{\theta} + kl_1^2 \theta + cl_2^2 \dot{\theta} = kl_1 x$$

State space form is obtained as follows

$$x_1 = \theta \quad x_2 = \dot{\theta} \Rightarrow \dot{x}_1 = x_2 \quad \dot{x}_2 = \ddot{\theta}$$

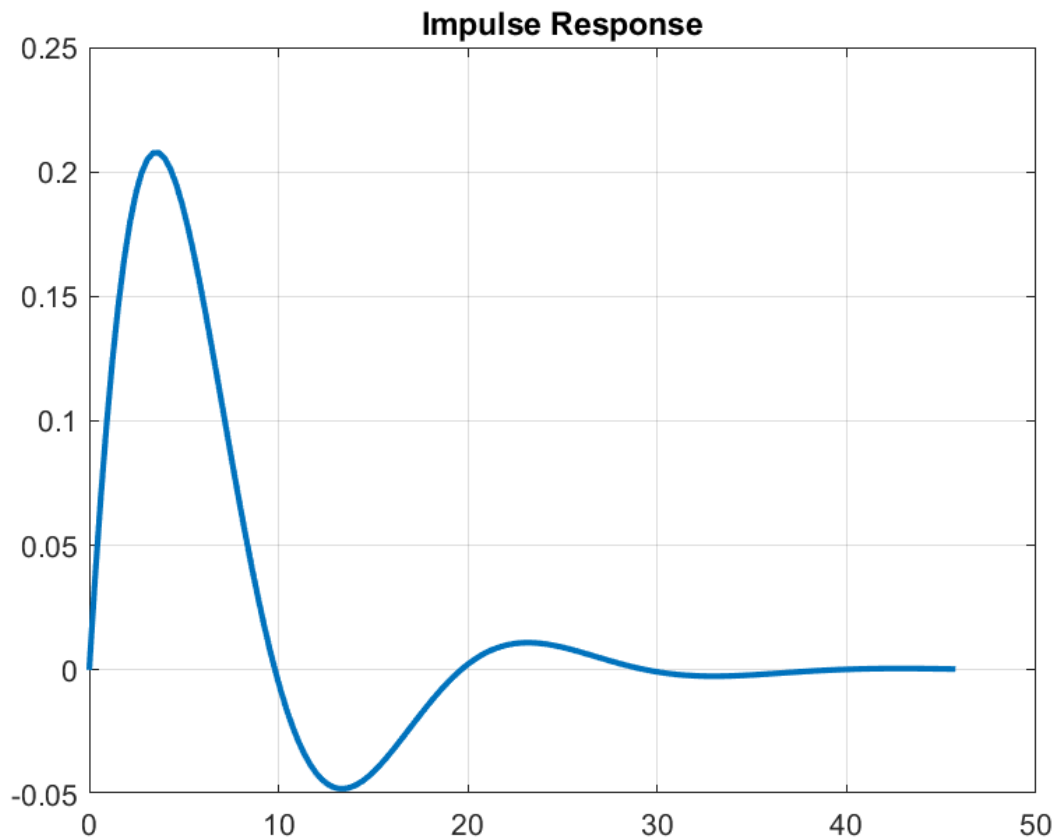
$$\ddot{\theta} = \dot{x}_2 = -\frac{kl_1^2}{ml_3^2}x_1 - \frac{cl_2^2}{ml_3^2}x_2 + \frac{kl_1}{ml_3^2}u$$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{pmatrix} 0 & 1 \\ -\frac{kl_1^2}{ml_3^2} & -\frac{cl_2^2}{ml_3^2} \end{pmatrix} \cdot \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ \frac{kl_1}{ml_3^2} \end{bmatrix} u$$

and if we are interested with only  $\theta$  we can construct the output as follows

$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \end{bmatrix} u$$

- Section -b



- Section -c

$$\theta(s)[s^2 m l_3^2 + s c l_2^2 + k l_1^2] = k l_1 U(s)$$

$$\frac{\theta(s)}{U(s)} = \frac{k l_1}{s^2 m l_3^2 + s c l_2^2 + k l_1^2}$$

$$\frac{\dot{\theta}(s)}{U(s)} = \frac{k l_1 s}{s^2 m l_3^2 + s c l_2^2 + k l_1^2}$$

Substituting the values into the transfer function gives

$$\frac{\theta(s)}{U(s)} = \frac{5}{40s^2 + 12s + 5}$$

*Poles are  $p_1 = -0.1500 + 0.3202i$ ,  $p_2 = -0.1500 - 0.3202i$   
and there are no zeros.*

All of the Poles are at the left hand side of the complex plane, so the system is stable and minimum phase.

## 4 Problem # 4

- **Section -a** Lagrange equations are derived as follows.  
For the first link of the elbow denoted by "1"

$$E_{k_1} = \frac{1}{2}m_1 l_{c_1}^2 \dot{q}_1^2 + \frac{1}{2}I_1 \dot{q}_1^2$$

$$E_{p_1} = m_1 g l_{c_1} \sin(q_1)$$

Denoting the center coordinates of the second link with  $G_{x_2}$  and  $G_{y_2}$

$$G_{x_2} = l_1 \cos(q_1) + l_{c_2} \cos(q_1 + q_2)$$

$$G_{y_2} = l_1 \sin(q_1) + l_{c_2} \sin(q_1 + q_2)$$

$$\dot{G}_{x_2} = -\dot{q}_1 l_1 \sin(q_1) - (\dot{q}_1 + \dot{q}_2) l_{c_2} \sin(q_1 + q_2) = A\dot{q}_1 + B\dot{q}_2$$

$$\dot{G}_{y_2} = \dot{q}_1 l_1 \cos(q_1) + (\dot{q}_1 + \dot{q}_2) l_{c_2} \cos(q_1 + q_2) = C\dot{q}_1 + D\dot{q}_2$$

$$A = -l_1 \sin(q_1) - l_{c_2} \sin(q_1 + q_2) \quad B = -l_{c_2} \sin(q_1 + q_2)$$

$$C = l_1 \cos(q_1) + l_{c_2} \cos(q_1 + q_2) \quad D = l_{c_2} \cos(q_1 + q_2)$$

$$\dot{G}_{x_2}^2 + \dot{G}_{y_2}^2 = (A^2 + C^2)\dot{q}_1^2 + (B^2 + D^2)\dot{q}_2^2 + 2(AB + CD)\dot{q}_1\dot{q}_2$$

$$A^2 + C^2 = l_1^2 + l_{c_2}^2 + 2l_1 l_{c_2} \cos(q_2) \quad B^2 + D^2 = l_{c_2}^2$$

$$AB = l_1 l_{c_2} \sin(q_1) \sin(q_1 + q_2) + l_{c_2}^2 \sin^2(q_1 + q_2)$$

$$CD = l_1 l_{c_2} \cos(q_1) \cos(q_1 + q_2) + l_{c_2}^2 \cos^2(q_1 + q_2)$$

$$AB + CD = l_1 l_{c_2} \cos(q_2) + l_{c_2}^2 \quad (\text{Using Trigonometric identities})$$

$$\Rightarrow \dot{G}_{x_2}^2 + \dot{G}_{y_2}^2 = \mathcal{V}_2^2 = \dot{q}_1^2 (l_1^2 + l_{c_2}^2 + 2l_1 l_{c_2} \cos(q_2)) + \dot{q}_2^2 l_{c_2}^2 + 2\dot{q}_1 \dot{q}_2 (l_1 l_{c_2} \cos(q_2) + l_{c_2}^2)$$

And finally  $E_{k_2}$  and  $E_{p_2}$  can be derived as follows

$$E_{k_2} = \frac{1}{2}m_2 \mathcal{V}_2^2 + \frac{1}{2}I_2 (\dot{q}_1 + \dot{q}_2)^2$$

$$E_{p_2} = m_2 g [l_1 \sin(q_1) + l_{c_2} \sin(q_1 + q_2)]$$

Total kinetic energy of the system is  $E_{k_1} + E_{k_2}$  and total potential energy of the system is  $E_{p_1} + E_{p_2}$  and  $\mathcal{L} = E_{k_1} + E_{k_2} - (E_{p_1} + E_{p_2})$

$$\mathcal{L} = \frac{1}{2}m_1l_{c_1}^2\dot{q}_1^2 + \frac{1}{2}I_1\dot{q}_1^2 + \frac{1}{2}m_2[\dot{q}_1^2(l_1^2 + l_{c_2}^2 + 2l_1l_{c_2}\cos(q_2)) + \dot{q}_2^2l_{c_2}^2 + 2\dot{q}_1\dot{q}_2(l_1l_{c_2}\cos(q_2) + l_{c_2}^2)] +$$

$$\frac{1}{2}I_2(\dot{q}_1 + \dot{q}_2)^2 - [m_1gl_{c_1}\sin(q_1) + m_2g(l_1\sin(q_1) + m_2gl_{c_2}\sin(q_1 + q_2))]$$

$$= \frac{1}{2}\dot{q}_1^2[m_1l_{c_1}^2 + I_1 + I_2 + m_2l_1^2 + m_2l_{c_2}^2 + 2m_2l_1l_{c_2}\cos(q_2)] + \frac{1}{2}\dot{q}_2^2[m_2l_{c_2}^2 + I_2] +$$

$$\dot{q}_1\dot{q}_2[I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] - g\sin(q_1)[m_1l_{c_1} + m_2l_1] - \sin(q_1 + q_2)m_2gl_{c_2}$$

For  $q_1$  and  $q_2$  there will be 2 equations:

$$\begin{aligned} \frac{\partial \mathcal{L}}{\partial \dot{q}_1} &= \dot{q}_1 [m_1l_{c_1}^2 + I_1 + I_2 + m_2l_1^2 + m_2l_{c_2}^2 + 2m_2l_1l_{c_2}\cos(q_2)] + \\ &\quad \dot{q}_2 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] \end{aligned}$$

$$\frac{\partial \mathcal{L}}{\partial \dot{q}_2} = \dot{q}_2 [m_2l_{c_2}^2 + I_2] + \dot{q}_1 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2]$$

$$\frac{\partial \mathcal{L}}{\partial q_1} = -g\cos(q_1)[m_1l_{c_1} + m_2l_1] - \cos(q_1 + q_2)m_2gl_{c_2}$$

$$\frac{\partial \mathcal{L}}{\partial q_2} = -\dot{q}_1^2m_2l_1l_{c_2}\sin(q_2) - m_2gl_{c_2}\cos(q_1 + q_2) - \dot{q}_1\dot{q}_2m_2l_1l_{c_2}\sin(q_2)$$

$$\begin{aligned} \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_1} \right) &= \ddot{q}_1 [m_1l_{c_1}^2 + I_1 + I_2 + m_2l_1^2 + m_2l_{c_2}^2 + 2m_2l_1l_{c_2}\cos(q_2)] - 2\dot{q}_1\dot{q}_2 [m_2l_1l_{c_2}\sin(q_2)] + \\ &\quad \ddot{q}_2 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] - \dot{q}_2^2 [m_2l_1l_{c_2}\sin(q_2)] \end{aligned}$$

$$\frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_2} \right) = \ddot{q}_2 [m_2l_{c_2}^2 + I_2] + \ddot{q}_1 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] - \dot{q}_1\dot{q}_2 [m_2l_1l_{c_2}\sin(q_2)]$$

$$\begin{aligned} \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_1} \right) - \frac{\partial \mathcal{L}}{\partial q_1} &= \ddot{q}_1 [m_1l_{c_1}^2 + I_1 + I_2 + m_2l_1^2 + m_2l_{c_2}^2 + 2m_2l_1l_{c_2}\cos(q_2)] - 2\dot{q}_1\dot{q}_2 [m_2l_1l_{c_2}\sin(q_2)] + \\ &\quad \ddot{q}_2 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] - \dot{q}_2^2 [m_2l_1l_{c_2}\sin(q_2)] + g\cos(q_1)[m_1l_{c_1} + m_2l_1] + \\ &\quad \cos(q_1 + q_2)m_2gl_{c_2} = Q_1 = 0 \end{aligned}$$

$$\begin{aligned} \frac{d}{dt} \left( \frac{\partial \mathcal{L}}{\partial \dot{q}_2} \right) - \frac{\partial \mathcal{L}}{\partial q_2} &= \ddot{q}_2 [m_2l_{c_2}^2 + I_2] + \ddot{q}_1 [I_2 + m_2l_1l_{c_2}\cos(q_2) + m_2l_{c_2}^2] + \dot{q}_1^2m_2l_1l_{c_2}\sin(q_2) + \\ &\quad m_2gl_{c_2}\cos(q_1 + q_2) = Q_2 = 0 \end{aligned}$$

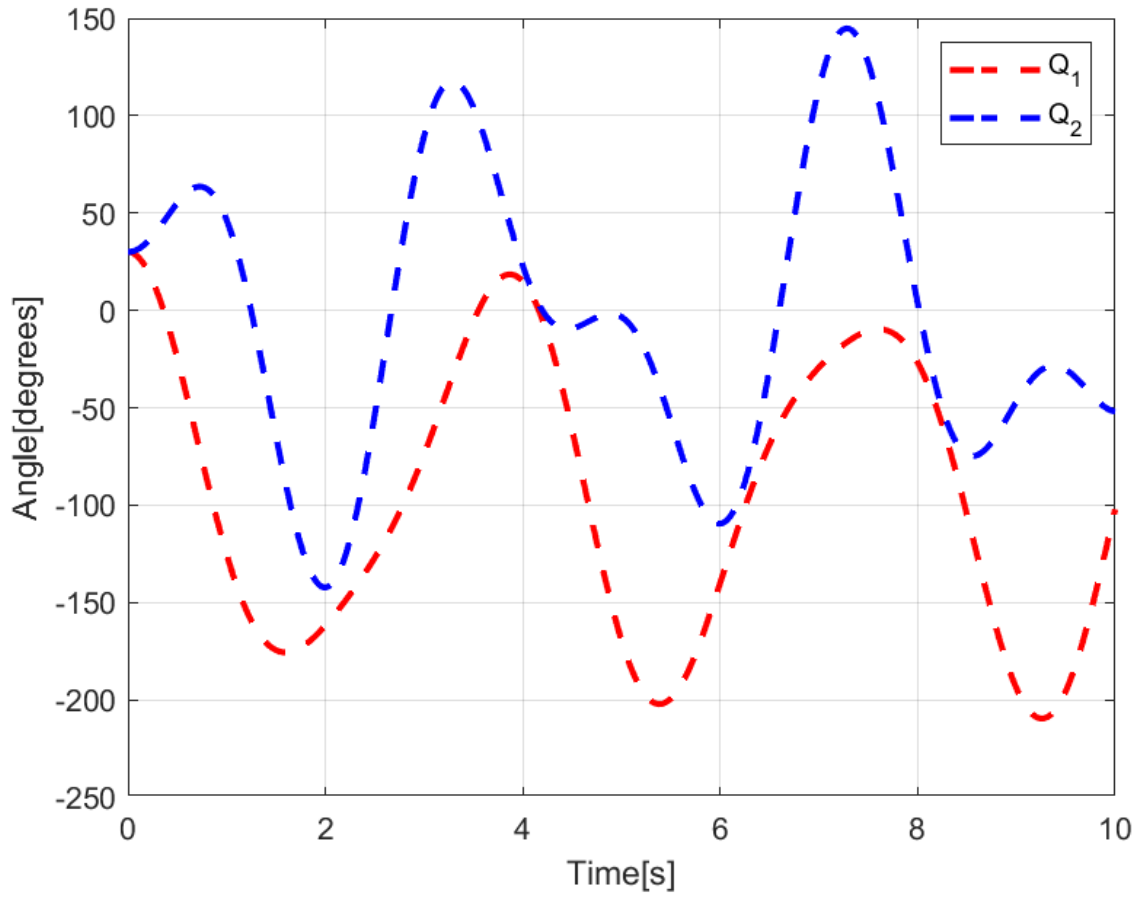
Assuming unit values and linearization around  $q_1 = \pi/2$  and  $q_2 = 0$  where the system is at the downward position and hence stable. Jacobians are derived in Matlab symbolic

toolbox and A and B matrices are calculated assuming  $\dot{q}_1 = 0$  and  $\dot{q}_2 = 0$  to be able to obtain a linear model.

$$\begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \\ \ddot{q}_1 \\ \ddot{q}_2 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -4.5510 & 1.8204 & 0 & 0 \\ 4.2476 & -7.5851 & 0 & 0 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ \dot{q}_1 \\ \dot{q}_2 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0.2165 & -0.4021 \\ -0.4021 & 1.1753 \end{bmatrix} \begin{bmatrix} C_1(t) \\ C_2(t) \end{bmatrix}$$

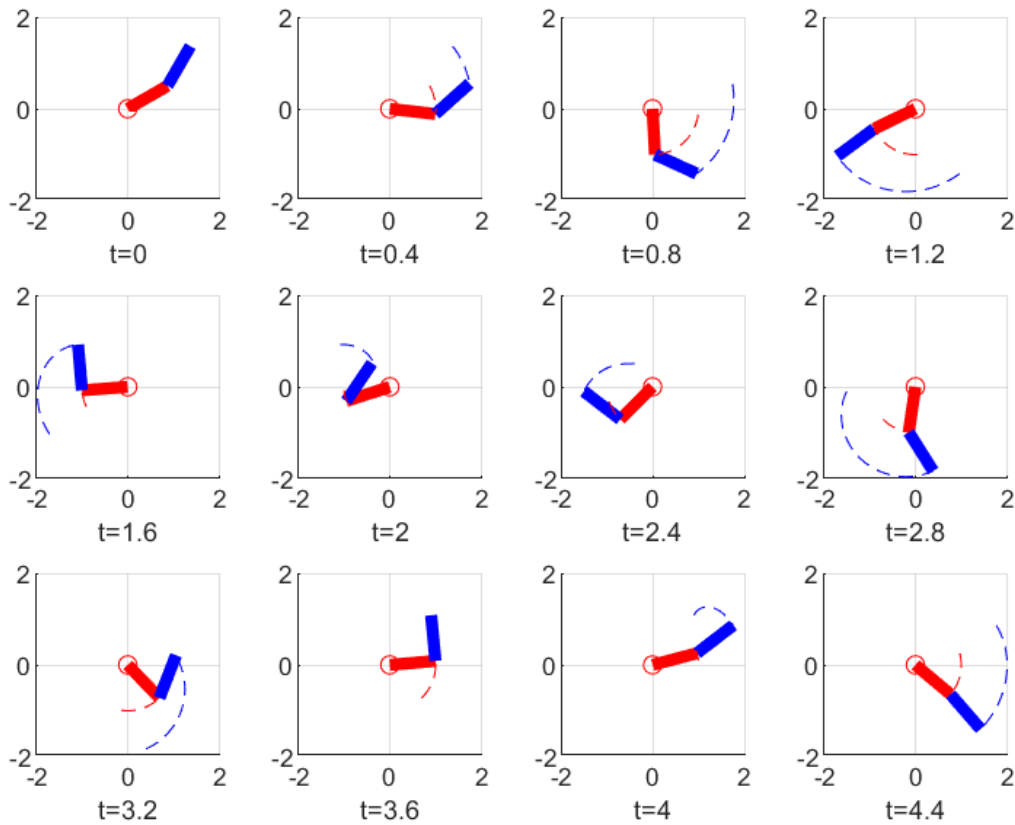
$$y = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix} \quad D = [0]$$

- Section -b





## Linearized 2-R Planar Manipulator Motion



- Section -c

$$\frac{\theta_1(s)}{C_1(s)} = \frac{0.2165s^2 + 0.9102}{s^4 + 12.14s^2 + 26.79}$$

$$\frac{\theta_2(s)}{C_1(s)} = \frac{-0.4021s^2 - 0.9102}{s^4 + 12.14s^2 + 26.79}$$

$$\frac{\theta_1(s)}{C_2(s)} = \frac{-0.4021s^2 - 0.9102}{s^4 + 12.14s^2 + 26.79}$$

$$\frac{\theta_2(s)}{C_2(s)} = \frac{1.175s^2 + 3.641}{s^4 + 12.14s^2 + 26.79}$$

Poles and zeros are not simply the poles and zeros of the transfer functions anymore. They are calculated as in the course notes<sup>1</sup> And the poles of the system are

$$p_1 = 3.0390i \quad p_2 = -3.0390i \quad p_3 = 1.7031i \quad p_4 = -1.7031i$$

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<sup>1</sup>Feedback Control Systems [https://ocw.mit.edu/courses/aeronautics-and-astronautics/16-30-feedback-control-systems-fall-2010/lecture-notes/MIT16\\_30F10\\_lec08.pdf](https://ocw.mit.edu/courses/aeronautics-and-astronautics/16-30-feedback-control-systems-fall-2010/lecture-notes/MIT16_30F10_lec08.pdf).

Or simply eigenvalues of the A matrix gives the poles. There are no zeros even it may seem like there are zeros by looking at the transfer functions.

## 5 Problem # 5

- Section -a

$$\begin{aligned}x_a &= -\alpha a & x_b &= -\alpha a - l \sin(\theta) & y_b &= a + l \cos(\theta) \\ \dot{x}_a &= -\dot{\alpha} a & \dot{x}_b &= -\dot{\alpha} a - l \cos(\theta) \dot{\theta} & \dot{y}_b &= -l \sin(\theta) \dot{\theta} \\ \dot{x}_b^2 + \dot{y}_b^2 &= \dot{\alpha}^2 a^2 + l^2 \dot{\theta}^2 + 2la\dot{\alpha}\dot{\theta}\end{aligned}$$

Using Lagrange's formula

$$\begin{aligned}\mathcal{T} &= \frac{1}{2}m[\dot{\alpha}^2 a^2 + l^2 \dot{\theta}^2 + 2la\dot{\alpha}\dot{\theta}\cos(\theta)] + \frac{1}{2}M\dot{\alpha}^2 a^2 + \frac{1}{2}J_1\dot{\alpha}^2 + \frac{1}{2}J_2\dot{\theta}^2 \\ &= \frac{1}{2}[(M+m)a^2 + J_1]\dot{\alpha}^2 + \frac{1}{2}[ml^2 + J_2]\dot{\theta}^2 + mla\dot{\alpha}\dot{\theta}\cos(\theta) \\ \mathcal{V} &= mgl\cos(\theta)\end{aligned}$$

$$\mathcal{D} = 0$$

$$\mathcal{L} = \mathcal{T} - \mathcal{V} = \frac{1}{2}[(M+m)a^2 + J_1]\dot{\alpha}^2 + \frac{1}{2}[ml^2 + J_2]\dot{\theta}^2 + mla\dot{\alpha}\dot{\theta}\cos(\theta) - mgl\cos(\theta)$$

$$\frac{\partial \mathcal{L}}{\partial \theta} = [ml^2 + J_2]\dot{\theta} + mla\dot{\alpha}\cos(\theta) \quad \frac{\partial \mathcal{L}}{\partial \theta} = -mla\dot{\alpha}\sin(\theta) + mgl\sin(\theta)$$

$$\frac{\partial \mathcal{L}}{\partial \dot{\alpha}} = [(M+m)a^2 + J_1]\dot{\alpha} + mla\dot{\theta}\cos(\theta) \quad \frac{\partial \mathcal{L}}{\partial \alpha} = 0$$

$$\frac{d}{dt}\left(\frac{\partial \mathcal{L}}{\partial \dot{\alpha}}\right) - \frac{\partial \mathcal{L}}{\partial \alpha} = [(M+m)a^2 + J_1]\ddot{\alpha} + mla[\ddot{\theta}\cos(\theta) - \dot{\theta}^2\sin(\theta)] = u(t)$$

$$\frac{d}{dt}\left(\frac{\partial \mathcal{L}}{\partial \dot{\theta}}\right) - \frac{\partial \mathcal{L}}{\partial \theta} = [ml^2 + J_2]\ddot{\theta} + mla[\ddot{\alpha}\cos(\theta)] - mgl\sin(\theta) = -u(t)$$

$$k_1\ddot{\alpha} + k_3(\ddot{\theta}\cos(\theta) - \dot{\theta}^2\sin(\theta)) = u(t)$$

$$k_2\ddot{\theta} + k_3\ddot{\alpha}\cos(\theta) - k_4\sin(\theta) = -u(t)$$

Solving these 2 equations for  $\ddot{\alpha}$  and  $\ddot{\theta}$  gives

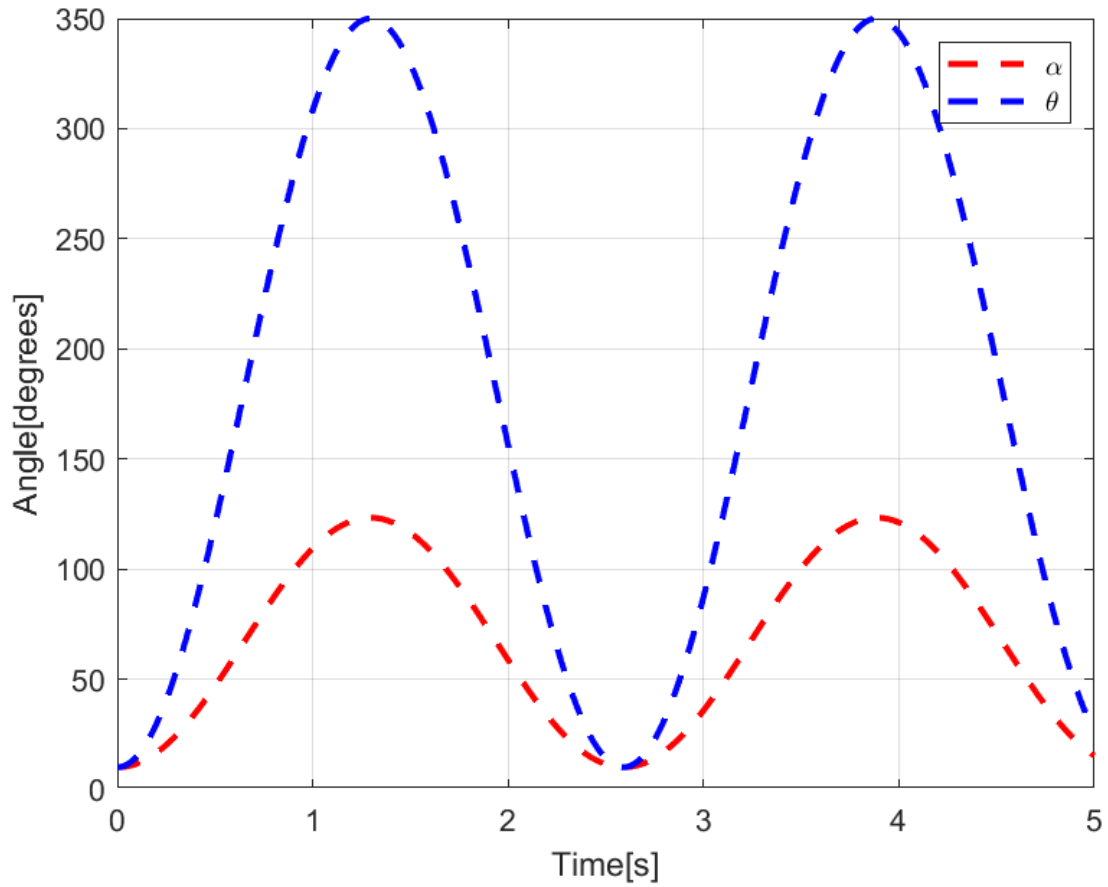
$$\ddot{\alpha} = \frac{k_2 k_3 \sin(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} \dot{\theta}^2 - \frac{k_3 k_4 \sin(\theta) \cos(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} + \frac{k_2 + k_3 \cos(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} u(t)$$

$$\ddot{\theta} = -\frac{k_3^2 \sin(\theta) \cos(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} \dot{\theta}^2 + \frac{k_1 k_4 \sin(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} - \frac{k_1 + k_3 \cos(\theta)}{k_1 k_2 - k_3^2 \cos^2(\theta)} u(t)$$

- **Section -b** Linearized around  $\theta = \pi$

$$\begin{bmatrix} \dot{\alpha} \\ \dot{\theta} \\ \ddot{\alpha} \\ \ddot{\theta} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & -1.962 & 0 & 0 \\ 0 & -5.886 & 0 & 0 \end{bmatrix} \begin{bmatrix} \alpha \\ \theta \\ \dot{\alpha} \\ \dot{\theta} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0.2 \\ -0.4 \end{bmatrix} u(t)$$

$$y = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix} \quad D = [0]$$



- Section -c

$$\frac{\alpha(s)}{U(s)} = \frac{0.2s^2 + 1.962}{s^2(s^2 + 5.886)}$$

$$\frac{\theta(s)}{U(s)} = \frac{-0.4}{s^2 + 5.886}$$

System poles are at  $p_1 = 0$      $p_2 = 0$      $p_3 = 0.4261i$      $p_4 = -0.4261i$

And there are no zeros in the system.

Poles are on the imaginary axis, which explains the oscillatory response of the system.