

CS 3630!

Lecture 25:
Drone Actions

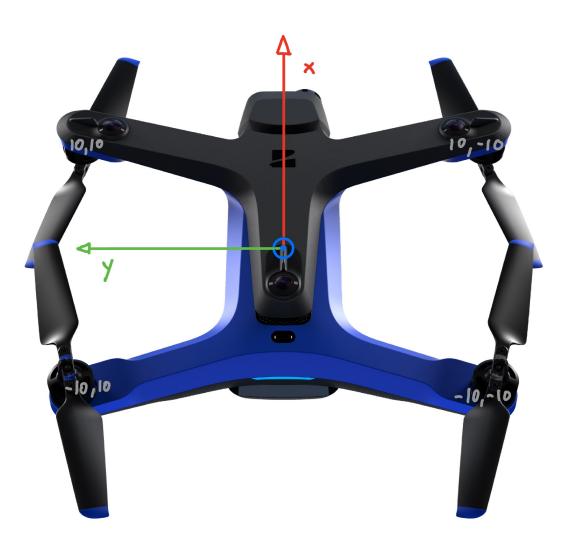


Actions for Quadrotor drones

- 1. Definitions
- 2. Hover
- 3. Forward Flight
- 4. Maximum Thrust
- 5. Drag

- 6. Kinematics
- 7. Simulation
- 8. Code Example
- 9. Dynamics
- 10. Gyroscopic effects

Definitions



- Body frame B: FLU = Forward-Left-Up
- Navigation Frame N: ENU = East-North-Up

- \cdot the vehicle's position $r^n \doteq [x,y,z]^T$,
- $oldsymbol{\cdot}$ its linear velocity $v^n=\dot{r^n}\doteq [u,v,w]^T$,
- $oldsymbol{\cdot}$ the attitude $R^n_b\doteq [i^b,j^b,k^b]\in SO(3)$, a 3 imes 3 rotation matrix the navigation frame \mathcal{N} ,
- ullet the body angular velocity $\omega^b \doteq [p,q,r]^T$.

Hover

$$F_z^b = \sum_{i=1}^4 f_i.$$

- Assume weight = 1kg
- $g = 10 \text{ m/s}^2$
- Need to provide 10N of thrust!

- $f_i=0N$ for $i\in 1..4$: downwards acceleration at $-10rac{m}{s^2}$.
- $f_i=2.5N$ for $i\in 1..4$: stable hover $0rac{m}{s^2}$.
- $f_i=5N$ for $i\in 1..4$: upwards acceleration at $10\frac{m}{s^2}$.

Forward Flight

$$F^n = R^n_b egin{bmatrix} 0 \ 0 \ F^b_z \end{bmatrix} = \hat{z}^n_b F^b_z$$

$$F^n = egin{bmatrix} 0 \ \sin heta \ \cos heta \end{bmatrix} F_z^b.$$

- Force is always up in body frame
- Need to rotate to navigation frame
- Thrust is always aligned with body zaxis expressed in navigation frame
- Maintain altitude:

$$\cos \theta F_z^b == 10N.$$

That means:

$$F_y^n = \sin heta F_z^b = \sin heta rac{10N}{\cos heta} = an heta \cdot 10N$$

Maximum Thrust

- Assume maximum thrust is 5N per rotor, i.e., 20N total!
- That means:

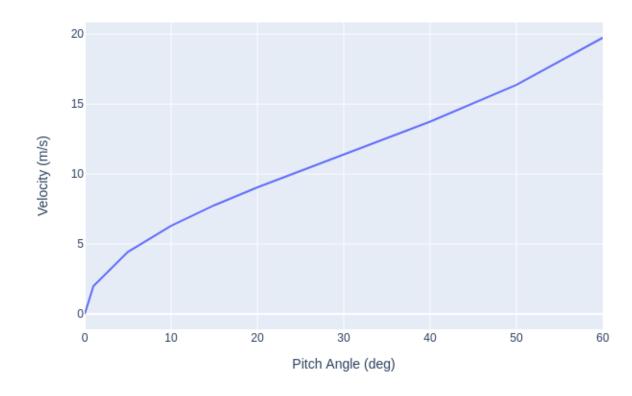
$$F_z^b = rac{10N}{\cos heta} \leq 20N \quad o \cos heta \geq 0.5 o -60^\circ \leq heta \leq 60^\circ.$$

Drag

- Air resistance increases quadratically with velocity
- Max velocity 20 m/s
- In book: calculate velocity while maintaining level flight:

 $v \, pprox 15 \sqrt{ an heta}$

Velocity vs. Pitch Angle



Kinematics (position)

• Easy kinematics: derivative of position is velocity

$$\dot{r}^n=v^n.$$

Angular velocity

- First, let us define angular velocity
- A three-vector defined in the body frame
- Axis-angle interpretation: velocity $\|\omega^b\|$ around axis ω^b
- Example: 10 degrees/sec around Y-axis (pitch down):

$$\omega^b = egin{bmatrix} 0 \ 1 \ 0 \end{bmatrix} 10 rac{\pi}{180}.$$

Kinematics (attitude)

• Not so easy: derivative of attitude is...

$$\dot{R}^n_b = R^n_b \hat{\omega}^b.$$

$$\hat{\omega}^b \doteq egin{bmatrix} -\omega_z^b & \omega_y^b \ \omega_z^b & -\omega_x^b \ -\omega_y^b & \omega_x^b \end{bmatrix}.$$

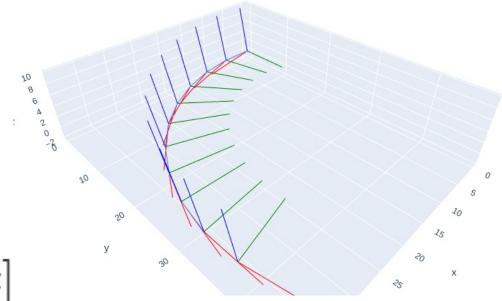
Simulation

- Forward integration
- Position:

$$r_{k+1}^n=r_k^n+d_{k+1}^k[v^n(t),\Delta t]$$

• Attitude:

$$R^n_{b,k+1}=R^n_{b,k}R^k_{k+1}[\omega^b(t),\Delta t]$$



Simulation (position)

- Approximation of exact integration = integration scheme
- Simplest: *Euler's method:*

$$egin{aligned} r^n_{k+1} &= r^n_k + d^k_{k+1}[v^n(t), \Delta t] \ & \ d^k_{k+1}[v^n(t), \Delta t] pprox v^n(t_k) \Delta t \end{aligned}$$

$$r_{k+1}^n = r_k^n + \, v^n(t_k) \Delta t$$

• Other schemes: backward Euler, trapezoidal method...

Simulation (attitude)

- A bit more complex
- Euler equivalent = Rodrigues' formula with constant angular velocity

$$egin{aligned} R^n_{b,k+1} &= R^n_{b,k} R^k_{k+1} [\omega^b(t), \Delta t] \ R^k_{k+1} [\omega^b(t), \Delta t] &pprox I + \sin heta K + (1-\cos heta) K^2 \ heta &= \|\omega^b_k\| \Delta t \ K &= \hat{\omega}^b_k/\|\omega^b_k\| \end{aligned}$$

Simulation (attitude, first order)

• For small rotation angles, we can approximate the Euler step:

$$R_{k+1}^k[\omega^b(t),\Delta t]pprox I+\sin heta Kpprox I+\hat{\omega}_k^b\Delta t=egin{bmatrix} 1&-\omega_z^b\Delta t&\omega_y^b\Delta t\ \omega_z^b\Delta t&1&-\omega_x^b\Delta t\ -\omega_y^b\Delta t&\omega_x^b\Delta t&1 \end{bmatrix}$$

• So, finally:

$$R^n_{b,k+1} = R^n_{b,k} R^k_{k+1} [\omega^b(t), \Delta t] pprox R^n_{b,k} egin{bmatrix} 1 & -\omega^b_z \Delta t & \omega^b_y \Delta t \ \omega^b_z \Delta t & 1 & -\omega^b_x \Delta t \ -\omega^b_y \Delta t & \omega^b_x \Delta t & 1 \end{bmatrix}$$

Code Example: terminal fwd. velocity + yaw

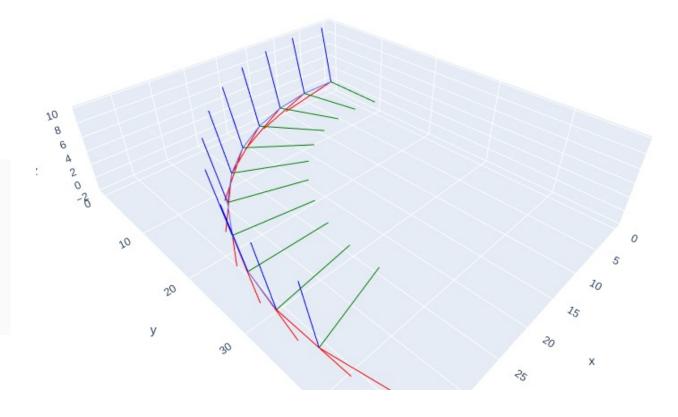
• Position:

$$r^n_{k+1}=r^n_k+d^k_{k+1}[v^n(t),\Delta t]$$

• Attitude:

$$R^n_{b,k+1}=R^n_{b,k}R^k_{k+1}[\omega^b(t),\Delta t]$$

```
# integrate forward
for k in range(K):
    vn = nRb[:,:,k] @ vb
    vn[2] = 0
    rn[:,k+1] = rn[:,k] + vn * delta_t
    delta_R = gtsam.Rot3.Expmap(wb * delta_t)
    nRb[:,:,k+1] = nRb[:,:,k] @ delta_R.matrix()
```



Dynamics

- Dynamics is harder
- Positional is easy:

$$F^n=m\dot{v}^n$$

• Attitude is harder:

$$au^bpprox I\dot{\omega}^b$$

- Mass: resists force
- Same resistance in all axes

- 3x3 Inertial matrix I
- How much do we resist torque?
- Typical: small-small-big



Creating forces and torques

• Linear force:

$$F_z^b = \sum_i f_i.$$

Angular torque:

$$au^b = egin{bmatrix} l(f_1 - f_2 - f_3 + f_4) \ l(f_1 + f_2 - f_3 - f_4) \ \kappa(f_1 - f_2 + f_3 - f_4) \end{bmatrix}$$



Gyroscopic effects*

• For large angular velocities:

$$au^b = I \dot{\omega}^b - \omega^b imes I \omega^b$$

Summary

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