

MTRN3500

Computing Applications in Mechatronics Systems

Developing a Flight Controller for a Quad-copter

T3 - 2020

Aim

- To logically analyze a complex system and develop a sophisticated software solution.
- The chosen system is an ordinary quad-copter operating in X configuration.
- Generally, these systems have non-linearities. However, in stable horizontal flying the non-linearities are minimal.

Approach

- Develop a complete schematic diagram incorporating all elements required.
- Breakdown the schematic to logical software units
- Develop classes for the logical units
- Develop the flight controller class

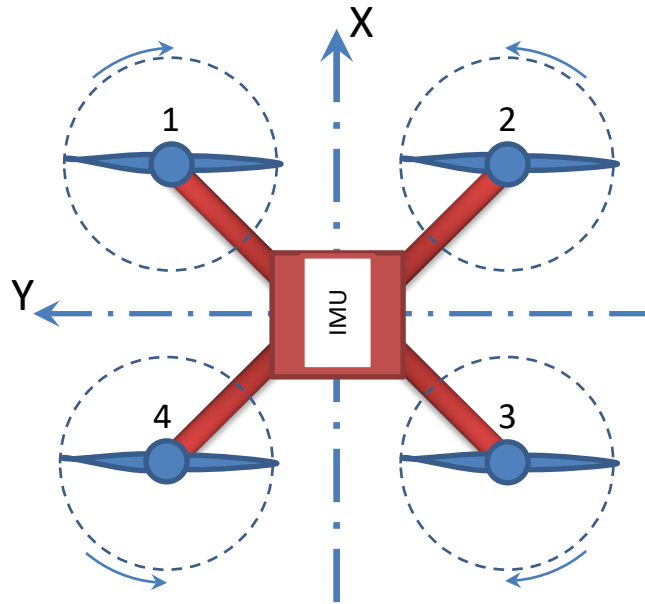
System Specifications

- System – quadcopter
- Controls required (attitude control only)
 - Pitch control
 - Roll control
 - Yaw control
- Controls required (position)
 - Heave rate control -> vertical position
- Sensing required
 - Gyro rates (roll, pitch and yaw rates in rad/sec).
 - Attitude sensing (roll, pitch and yaw angles in rad).
- Actuation
 - Speed control of four rotors
- Commands
 - From a remote transmitter

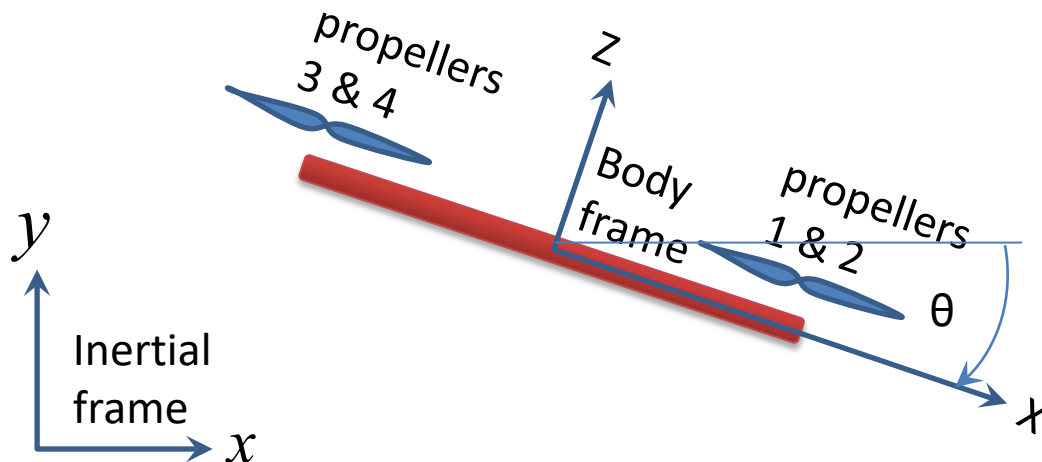


Red – Roll Command
Green – Pitch Command
Yellow – Yaw rate command
Purple – Heave rate command

Configuration

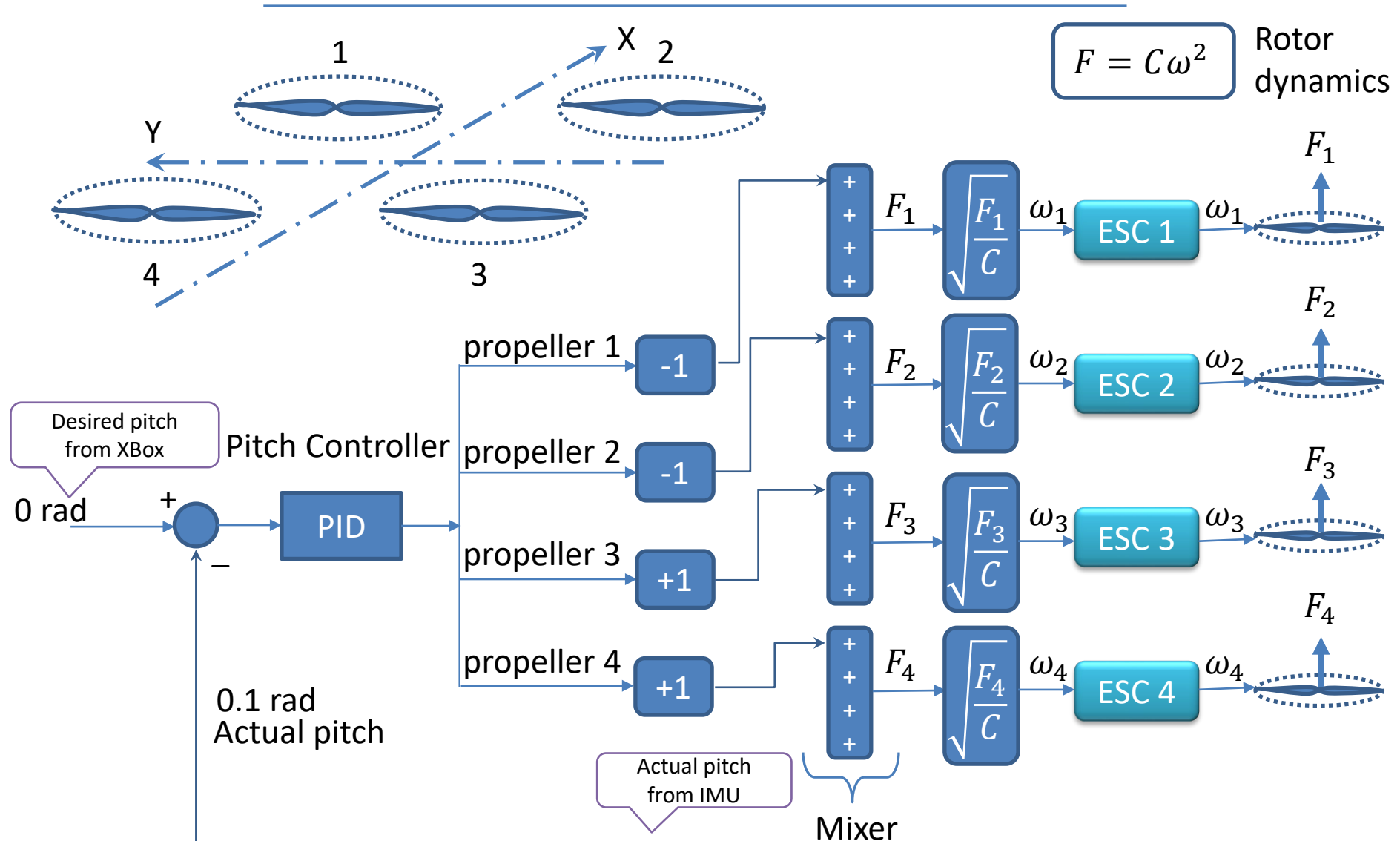


- X- configuration
- Counter rotating propellers
- X is the forward direction
- Z is vertically up
- Propellers are numbered from 1 – 4
- An IMU at the centre of the copter measures the attitude and gyro rates

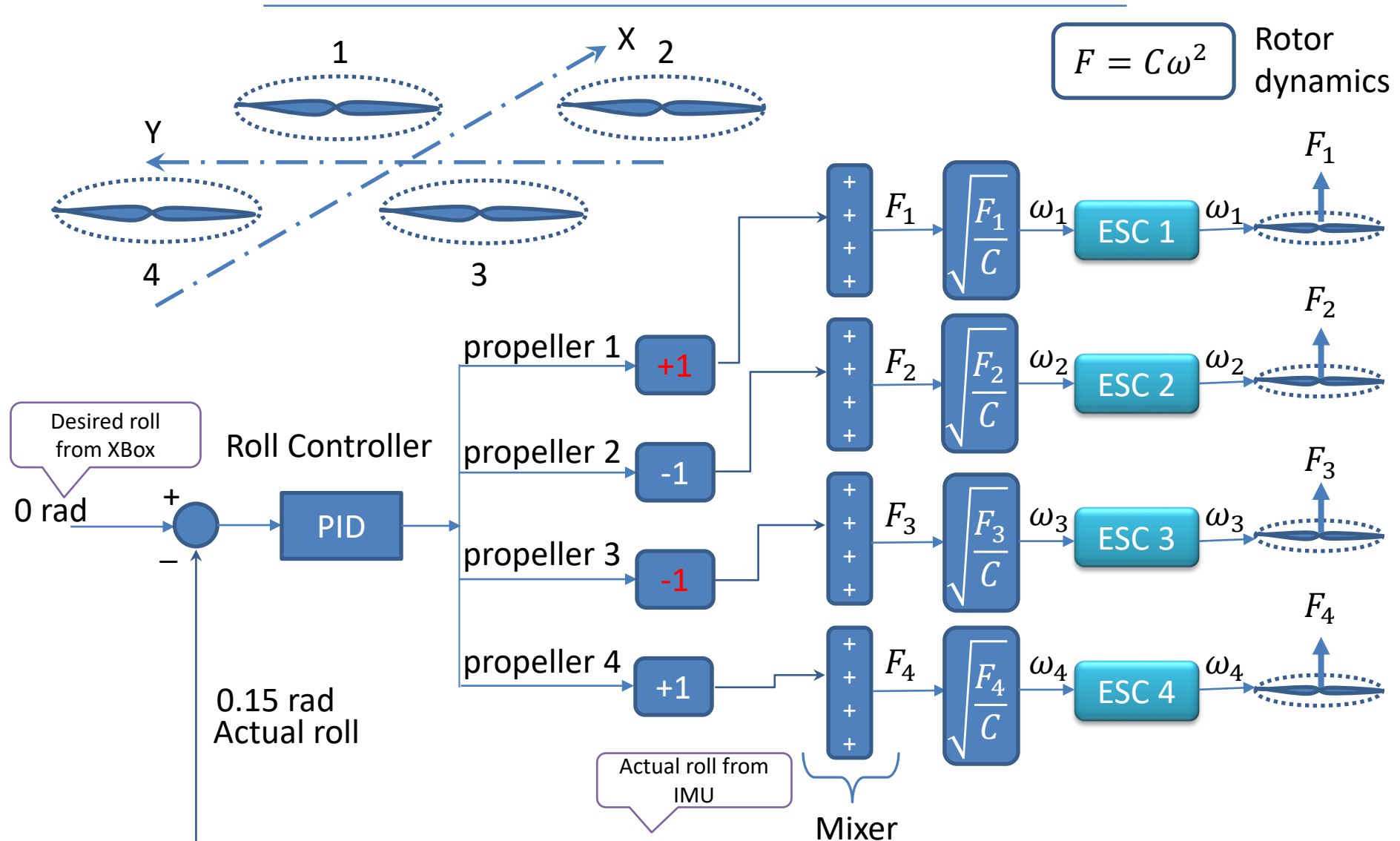


- A positive pitch angle θ will make the quadcopter move forward
- A positive roll angle ϕ will make the quadcopter move rightwards.
- A positive yaw angle ψ will rotate the quadcopter in counter clockwise direction.

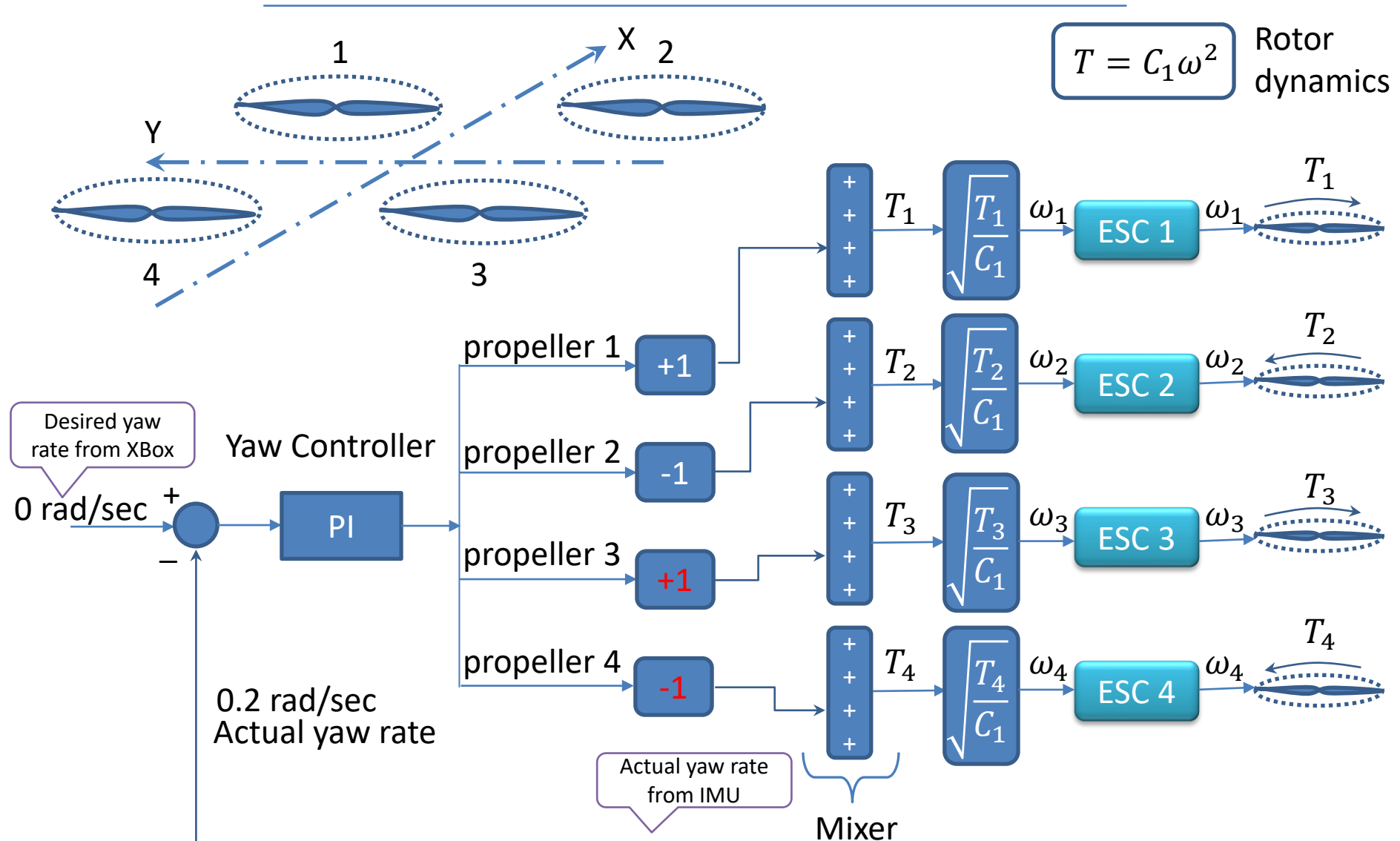
Pitch Control



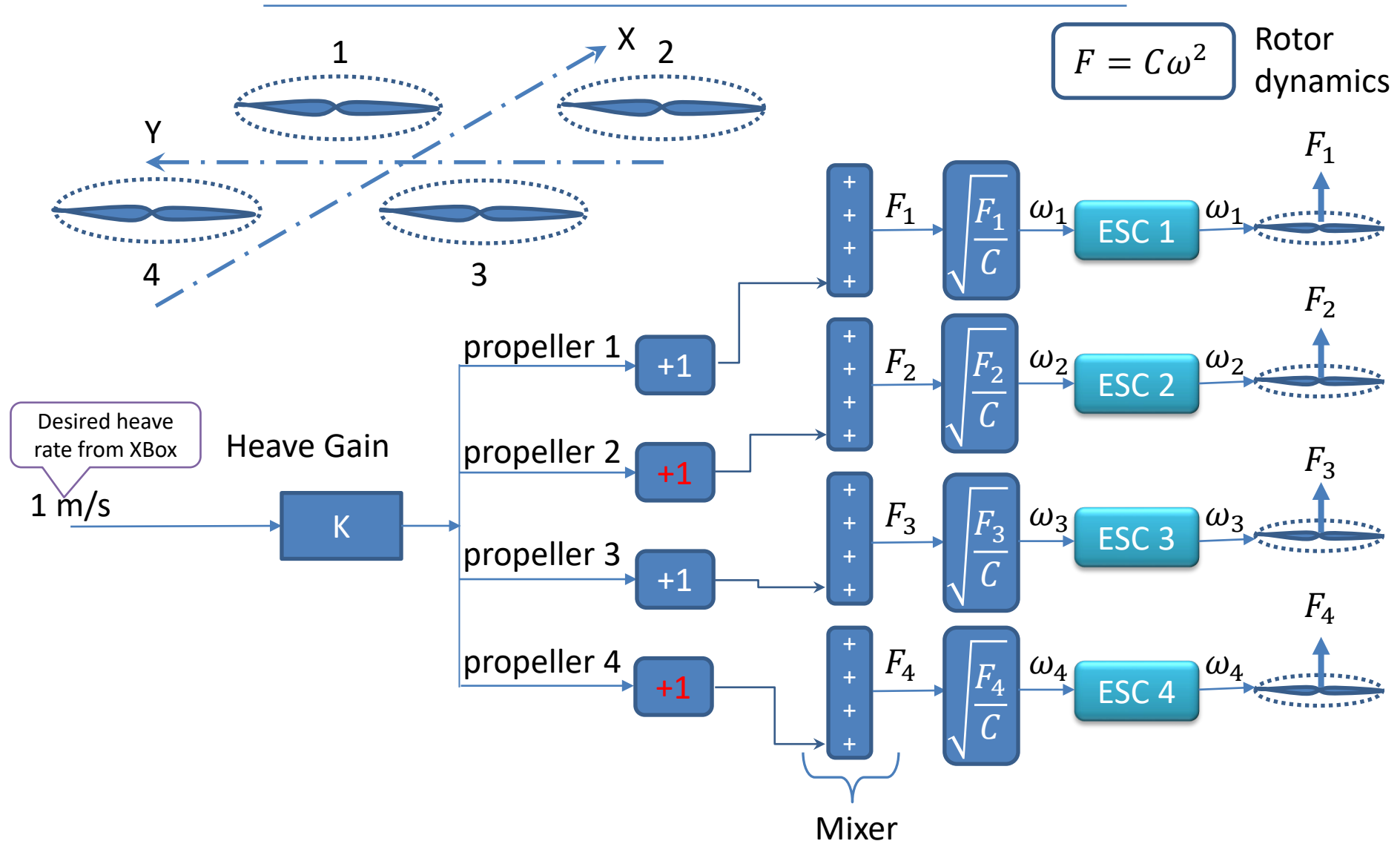
Roll Control



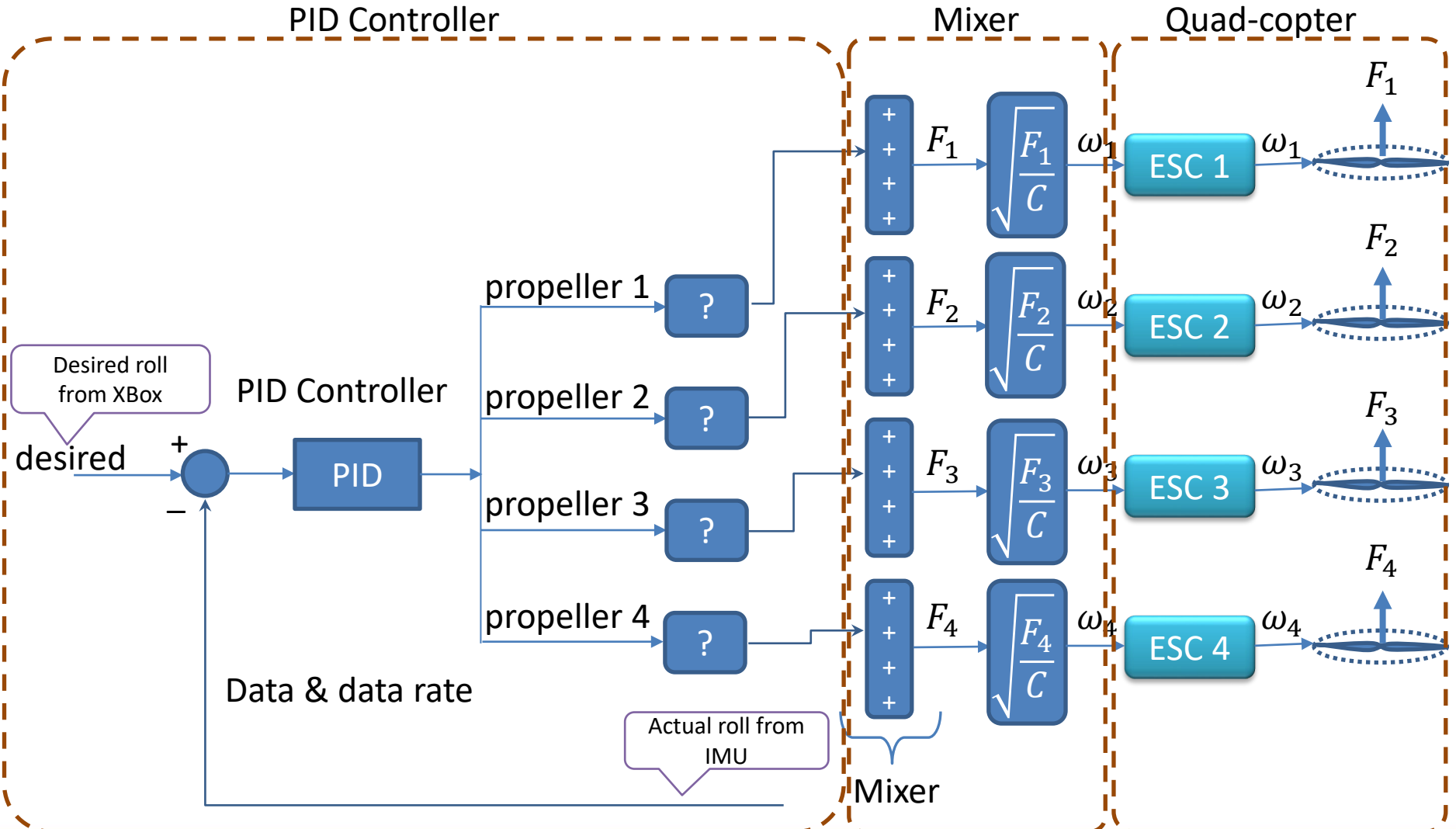
Yaw Rate Controller



Heave Rate Control



Flight Controller



Let us Look at Some Software
