

Gravitation

2 Marks Answers and Questions

Q1. Explain gravitation.

Ans: Any two bodies will be attracted to one another by the force of gravity, also known as gravity. There is an attraction between everything in the cosmos, however most of the time the force is too faint to be seen due to the extreme distance between the objects.

Q2. Name two factors that influence whether or not a planet has an atmosphere.

Ans: The existence of an atmosphere on a planet is determined by two factors.

1. The temperature on the planet's surface
2. The presence of gravitational acceleration on the planet's surface

Q3. At the centre of the Earth, a body is weightless. Why?

Ans: $G = 0$ in the centre of the Earth.

It is determined by:

$$w = mg$$

Q3. At the centre of the Earth, a body is weightless. Why?

Ans: $G = 0$ in the centre of the Earth.

It is determined by:

$$w = mg$$

$$\Rightarrow w = 0$$

The body is therefore weightless at the centre of the Earth.

Q4.What is kepler's law of periods? Show it mathematically?

Ans.It states that the square of the period of revolution of a planet around the sun is proportional of a planet to the cube of the semi-major axis of the elliptical orbit.

$$\text{i.e. } T^2 \propto R^3$$

$$T^2 = KR^3$$

where T is time period of evolution

R is the length of semi major axis

K is constant for all planets

Q5.With two characteristics of gravitational force?

Ans.(1) It is a central force

(2) It is a conservation force

(3) It obeys inverse square law.

(4) It is a universal force and is always attractive in nature.

Q6. Assuming earth to be a uniform sphere finds an expression for density of earth in terms of g and G ?



Ans. Since $g = \frac{GM}{R^2}$

If earth is uniform sphere of mean density P

$$g = \frac{G}{R^2} \left(\frac{4}{3} \pi R^3 P \right)$$

$$g = \frac{4}{3} \pi G R P$$

$$\Rightarrow P = \frac{3g}{4\pi GR}$$

Q7. If radius of earth is 6400km, what will be the weight of 1 quintal body if taken to the height of 1600 km above the sea level?

Ans. $R = 6400\text{km} = 6400 \times 10^3\text{m}$

$h = 1600\text{km}$

$w = mg = 1 \text{ quintal} = 100 \text{ kg} = 100 \times 9.8 \text{ N}$

weight (w) = mgh

$$w = mg \left(\frac{R}{R+h} \right)^2$$

$$w = 100 \times 9.8 \left(\frac{6400}{1600 + 6400} \right)^2$$

$w = 64 \times 9.8\text{N} = 64\text{kg}$

Q8. A man can jump 1.5 m high on earth. Calculate the height he may be able to jump on a planet whose density is one fourth that of the earth and Whose radius is one third of the earth.

Ans. $g = \frac{4}{3} \pi G R \rho$

$$g' = \frac{4}{3} \pi G R' \rho'$$

The gain in PE at the highest point will be same in both cases. Hence $mg'h' = mgh$

$$h' = \frac{mgh}{mg'} = \frac{m \times \frac{4}{3} \pi G R \rho h}{m \frac{4}{3} \pi G R' \rho'}$$

$$= \frac{R \rho h}{R' \rho'} = \frac{3R' \times 4 \rho \times 1.5}{R' \times \rho'}$$

$= 18 \text{ m}$

4 Marks Answers and Questions

Q1. Assume that there are 2.5×10^{11} stars in our galaxy, each with a solar mass. How long will it take a star 50,000 light years from the galactic centre to make one revolution? Assume that the Milky Way is 105ly in diameter.

Ans: Mass of the Milky Way galaxy, $M = 2.5 \times 10^{11}$ solar mass

Solar Mass = Mass of the Sun = 2.0×10^{36} kg

Mass of our galaxy, $M = 2.5 \times 10^{11} \times 2 \times 10^{36}$

$M = 5 \times 10^{41}$ kg

Diameter of the Milky Way, $d = 105$ ly

Radius of the Milky Way, $r = 5 \times 10^4$ ly

$1 \text{ ly} = 9.46 \times 10^{15} \text{ m}$

$r = 5 \times 10^4 \times 9.46 \times 10^{15}$

$r = 4.73 \times 10^{20} \text{ m}$

Since a star rotates around the galactic centre of the Milky Way, its time period is given by

$$T = \frac{2\pi r^3}{GM} = \frac{4 \times 3.14 \times (4.73 \times 10^{20})^3}{6.67 \times 10^{-11} \times 5 \times 10^{41}}$$

$$T = 125.27 \times 10^3 \text{ s}$$

$$T = 1.12 \times 10^{16} \text{ s}$$

As we know,

$$1 \text{ year} = 365 \times 24 \times 60 \times 60 \text{ s}$$

We get,

$$1 \text{ s} = \frac{1}{365 \times 24 \times 60 \times 60} \text{ years}$$

$$1.12 \times 10^{16} \text{ s} = 1.12 \times 10^{16} \times \frac{1}{365 \times 24 \times 60 \times 60}$$

$$\therefore 1.12 \times 10^{16} \text{ s} = 3.55 \times 10^8 \text{ years}$$

Thus, the star will take 3.55×10^8 years to complete one revolution.

Q2. A body weighs 63N while it is on the Earth's surface. What is the magnitude of the Earth's gravitational pull on it at a height equal to one-half the radius of the Earth?

Ans: Weight of the body, $W = 63\text{N}$

Acceleration due to gravity at height h from the Earth's surface

R_e = Radius of the Earth

For $h=R_e/2$, gravity at h is given by

$$g_h = g_1 + \frac{h}{R_e} g_1$$

$$g_h = g_1 + \frac{1}{2} g_1$$

$$g_h = \frac{3}{2} g_1$$

Weight of a body of mass m at height h is given as"

$$W' = m g_h$$

$$W' = \frac{3}{2} m g_1$$

$$W' = \frac{3}{2} W$$

$$W' = \frac{3}{2} \times 63$$

$$W' = 94.5\text{N}$$

Therefore, at a height equal to half the Earth's radius, the gravitational force exerted on the body by the Earth is 94.5N.

Q3. A projectile can escape from the Earth's surface at a speed of 11.2 km/s. With three times this speed, a body is thrust out. What is the speed of a body travelling far from Earth? Ignore the sun's and other planets' presence.

Ans: The escape velocity of the projectile from the Earth, $v_{esc}=11.2\text{kms}^{-1}$

Projectile velocity of the projectile, $v_p=3v_{esc}$

Mass of the projectile = m

Projectile's velocity far from the Earth = v_f

$$\text{Projectile's total energy on the Earth} = \frac{1}{2} m v_p^2 - \frac{1}{2} m v_{esc}^2$$

The projectile's gravitational potential energy far from the Earth is 0.

$$\text{Total energy of the projectile far from the Earth} = \frac{1}{2} m v_f^2$$

From the law of energy conservation, we have

$$\frac{1}{2} m v_p^2 - \frac{1}{2} m v_{esc}^2 = \frac{1}{2} m v_f^2$$

We get,

$$v_f = v_p^2 - v_{esc}^2$$

$$v_f = 3v_{esc}^2 - v_{esc}^2$$

$$v_f = 8v_{esc}$$

$$v_f = 8 \times 11.2$$

$$v_f = 31.68 \text{ km s}^{-1}$$

Therefore, the speed of the body far from the Earth's surface is 31.68 km s⁻¹

Q4. A satellite orbits the earth at a height of 400 km above the surface. How much energy must be expended to rocket the satellite out of the earth's gravitational influence? Mass of the satellite = 200 kg; mass of the earth = 6.0×10^{24} kg; radius of the earth = 6.4×10^6 m; $G = 6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$.

Ans. Mass of the Earth, $M = 6.0 \times 10^{24} \text{ kg}$

Mass of the satellite, $m = 200 \text{ kg}$

Radius of the Earth, $R_e = 6.4 \times 10^6 \text{ m}$

Universal gravitational constant, $G = 6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$

Height of the satellite, $h = 400 \text{ km} = 4 \times 10^5 \text{ m} = 0.4 \times 10^6 \text{ m}$

Total energy of the satellite at height h
$$= \frac{1}{2}mv^2 + \left(\frac{-GM_em}{R_e + h} \right)$$

Orbital velocity of the satellite, $v = \sqrt{\frac{GM_e}{R_e + h}}$

Total energy of height, h
$$= \frac{1}{2}m + \left(\frac{GM_em}{R_e + h} \right) - \frac{GM_em}{R_e + h} = -\frac{1}{2} \left(\frac{GM_em}{R_e + h} \right)$$

The negative sign indicates that the satellite is bound to the Earth. This is called bound energy of the satellite.

Energy required to send the satellite out of its orbit = – (Bound energy)

$$= \frac{1}{2} \frac{GM_em}{(R_e + h)}$$

$$= \frac{1}{2} \times \frac{6.67 \times 10^{-11} \times 6.0 \times 10^{24} \times 200}{(6.4 \times 10^6 + 0.4 \times 10^6)}$$

$$= \frac{1}{2} \times \frac{6.67 \times 6 \times 2 \times 10}{6.4 \times 10^6} = 5.9 \times 10^9 \text{ J}$$

Q5. Choose the correct answer from among the given ones:

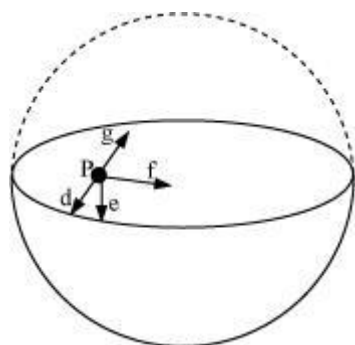
For the problem 8.10, the direction of the gravitational intensity at an arbitrary point P is indicated by the arrow (i) d, (ii) e, (iii) f, (iv) g.

Ans.(ii) Gravitational potential (V) is constant at all points in a spherical shell. Hence, the

gravitational potential gradient $\left(\frac{dV}{dr}\right)$ is zero everywhere inside the spherical shell. The gravitational potential gradient is equal to the negative of gravitational intensity. Hence, intensity is also zero at all points inside the spherical shell. This indicates that gravitational forces acting at a point in a spherical shell are symmetric.



If the upper half of a spherical shell is cut out (as shown in the given figure), then the net gravitational force acting on a particle at an arbitrary point P will be in the downward direction.



Since gravitational intensity at a point is defined as the gravitational force per unit mass at that point, it will also act in the downward direction. Thus, the gravitational intensity at an arbitrary point P of the hemispherical shell has the direction as indicated by arrow e.

Q6. A satellite is revolving in a circular path close to a planet of density P. find an expression for its period of revolution?

Ans. If satellite revolves around the earth of radius r

$$T = \frac{2\pi r}{v}$$

where v is orbital velocity

$$\text{where } v = \sqrt{\frac{Gm}{r}}$$

$$T = \frac{2\pi r}{v} = \frac{2\pi r}{\sqrt{\frac{Gm}{r}}} = 2\pi \sqrt{\frac{r^3}{Gm}}$$

If a satellite is revolving near the planet's surface then $r = R$ radius of planet and

$$M = \frac{4}{3} \pi R^3 P$$

$$T = \frac{2\pi}{\sqrt{G \cdot \frac{4}{3} \pi R^3 P}}$$

$$T = 2\pi \sqrt{\frac{3}{4\pi GP}}$$

$$T = \sqrt{\frac{3\pi}{GP}}$$

7 Marks Answers and Questions

Q1. One of Jupiter's satellites, IO, has an orbital period of 1.769 days and an orbital radius of 4.22 10⁸ metres. Prove that Jupiter's mass is around one thousandth that of the sun.

Ans: Given,

Orbital period of IO = T_{IO} = 1.769 days = 1.769 × 24 × 60 × 60s

Orbital radius IO = R_{IO} = 4.22 × 10⁸m

Satellite IO is revolving around Jupiter.

Mass of Jupiter is given by

$$M_J = \frac{4R_{IO}^3}{GT_{IO}^2} \dots 1$$

Where,

M_J is the mass of Jupiter.

G is the Universal gravitational constant.

The orbital period of the Earth, T_e = 365.25 days = 365.25 × 25 × 60 × 60s

The orbital radius of the Earth, R_e = 1AU = 1.496 × 10¹¹m

Mass of the Sun is given as

$$M_s = \frac{4R_e^3}{GT_e^2} \dots 2$$

$$\frac{M_s}{M_J} = \frac{4R_e^3}{GT_e^2} \times \frac{GT_{IO}^2}{4R_{IO}^3} = \frac{R_e^3 R_{IO}^3 T_{IO}^2 T_e^2}{R_{IO}^3 T_{IO}^2 T_e^2}$$

$$\frac{M_s}{M_J} = \frac{1.769 \times 24 \times 60 \times 60 \times 365.25 \times 24 \times 60 \times 60 \times 1.496 \times 10^{11} \times 4.22 \times 10^8}{1.496 \times 10^{11} \times 365.25 \times 24 \times 60 \times 60 \times 1.496 \times 10^{11} \times 4.22 \times 10^8}$$

$$\frac{M_s}{M_J} = 1045.04$$

$$\therefore \frac{M_s}{M_J} \sim 1000$$

We get,

$$M_s \sim 1000 \times M_J$$

The mass of Jupiter is therefore estimated to be approximately one thousandth that of the Sun.

Q2. A rocket is launched vertically at a speed of 5 km/s-1 from the surface of the Earth. How far does the rocket travel before coming back to Earth? Earth's mass is 6.0×10^{24} kg; its mean radius is 6.4×10^6 m; and its gravitational constant is $6.67 \times 10^{-11} \text{ Nm}^2\text{kg}^{-2}$.

Ans: Distance from the Earth's centre = $8 \times 10^6 \text{ m}$

Velocity of the rocket, $v = 5 \text{ km/s} = 5 \times 10^3 \text{ m/s}$

Mass of the Earth, $M_e = 6.0 \times 10^{24} \text{ kg}$

Radius of the Earth, $R_e = 6.4 \times 10^6 \text{ m}$

Height reached by rocket mass m is h

At the surface of the Earth,

Total energy of the rocket = Kinetic energy + Potential energy

$$= \frac{1}{2}mv^2 - \frac{GMm}{R_e}$$

At highest point h , $v = 0$

Then, Potential energy = $-\frac{GMm}{R_e+h}$

Total energy of the rocket = $0 - \frac{GMm}{R_e+h} = -\frac{GMm}{R_e+h}$

From the law of conservation,

Total energy of the rocket at the Earth's surface = Total energy of rocket at height h

Then,

$$\frac{1}{2}mv^2 - \frac{GMm}{R_e} = -\frac{GMm}{R_e+h}$$

$$\frac{1}{2}v^2 = \frac{GM}{R_e} - \frac{GM}{R_e+h}$$

$$\frac{1}{2}v^2 = \frac{GM}{R_e} \left(\frac{R_e+h - R_e}{R_e(R_e+h)} \right)$$

$$\frac{1}{2}v^2 = \frac{GMh}{R_e(R_e+h)}$$

$$\frac{1}{2}v^2 = gh \frac{R_e}{R_e+h}$$

Where,

$g = \frac{GM}{R_e^2} = 9.8 \text{ m/s}^2$ is the acceleration due to gravity on the Earth's surface.

Clearly,

$$v^2(R_e+h) = 2gR_e h$$

$$v^2 R_E = h (2g R_E - v^2)$$

$$h = \frac{v^2 R_E}{2g R_E - v^2}$$

$$h = \frac{6.4 \times 10^6 \times 100.44 \times 10^6}{2 \times 9.8 \times 10^6 - 100.44^2 \times 10^6}$$

$$h = 1.6 \times 10^6 \text{ m}$$

Height achieved by the rocket with respect to the Earth's centre, H is given by: $H = R_E + h$

$$H = 6.4 \times 10^6 + 1.6 \times 10^6$$

We obtain,

$$H = 8.0 \times 10^6 \text{ m}$$

The distance the rocket travels before landing on Earth is 8.0×10^6 metres.

Q3. A head-on collision is imminent between two stars, each of which has a solar mass of one solar mass ($=2 \times 10^{30}$ kg). Their speeds are barely perceptible at a distance of 109 kilometres. What is the rate of their collision? Each star has a 104 km radius. Assume that up until their collision, the stars won't change. (Use the known value of G)

Ans: Mass of each star, $M = 2 \times 10^{30}$ kg

Radius of each star, $R = 104 \text{ km} = 1.04 \times 10^5 \text{ m}$

Distance between the stars, $r = 109 \text{ km} = 1.09 \times 10^5 \text{ m}$

For negligible speeds, $v = 0$

Total energy of two stars separated at distance r is given by

$$TE = -\frac{GMM}{r} + \frac{1}{2}mv^2$$

$$TE = -\frac{GMM}{r} + 0 \dots\dots(i)$$

Now, consider the case when the stars are about to collide:

Velocity of the stars = v

Distance between the centres of the stars = $2R$

Total kinetic energy of both stars = $\frac{1}{2}Mv^2 + \frac{1}{2}Mv^2 = Mv^2$

Total potential energy of both stars = $-\frac{GMM}{2R}$

Total energy of the two stars = $Mv^2 - \frac{GMM}{2R} \dots\dots(ii)$

From the law of conservation of energy,

$$Mv^2 - GMM/2R = -GMM/R$$

$$v^2 = 6.67 \times 10^{-10} \times 2 \times 10^{30} / 11012 + 12 \times 10^7$$

$$v^2 = 13.34 \times 10^{19} - 10^{-12} + 5 \times 10^{-8}$$

$$v^2 = 13.34 \times 10^{19} \times 5 \times 10^{-8}$$

$$v^2 = 6.67 \times 10^{12}$$

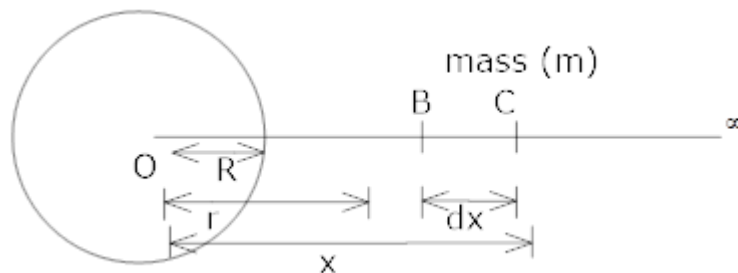
$$v = 6.67 \times 10^{12}$$

$$v = 2.58 \times 10^6 \text{ m/s}$$

The two stars collide at a speed of $2.58 \times 10^6 \text{ m/s}$.

Q4. Define Gravitational potential energy. Hence deduce an expression for gravitational potential energy of a body placed at a point near the surface of earth?

Ans. It is defined as the work done in bringing a body from infinity to that point.



A body of mass (m) lying at a distance x from earth of mass (M)

$$F = \frac{GMm}{x^2}$$

If the body is displaced through a distance dx then

$$dw = Fdx = \frac{GMm}{x^2} dx$$

Total work done

$$W = \int_{\infty}^r \frac{GMm}{x^2} dx$$

$$w = GMm \int_{\infty}^r \frac{1}{x^2} dx$$

$$w = GMm \left[\frac{-1}{x} \right]_{\infty}^r$$

$$w = -GMm \left[\frac{1}{r} - \frac{1}{\infty} \right]$$

$$w = \frac{-GMm}{r}$$

This work done is equal to the gravitational potential energy

i.e.

$$w = U_g = \frac{-GMm}{r}$$

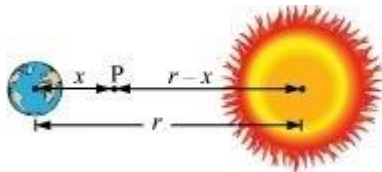
Q5. A rocket is fired from the earth towards the sun. At what distance from the earth's centre is the gravitational force on the rocket zero? Mass of the sun = 2×10^{30} kg, mass of the earth = 6×10^{24} kg. Neglect the effect of other planets etc. (orbital radius = 1.5×10^{11} m).

Ans. Mass of the Sun, $M_s = 2 \times 10^{30}$ kg

Mass of the Earth, $M_e = 6 \times 10^{24}$ kg

Orbital radius, $r = 1.5 \times 10^{11}$ m

Mass of the rocket = m



Let x be the distance from the centre of the Earth where the gravitational force acting on satellite P becomes zero.

From Newton's law of gravitation, we can equate gravitational forces acting on satellite P under the influence of the Sun and the Earth as:

$$\frac{GmM_s}{(r-x)^2} = Gm \frac{M_e}{x^2}$$

$$\left(\frac{r-x}{x} \right)^2 = \frac{M_s}{M_e}$$

$$\frac{r-x}{x} = \left(\frac{2 \times 10^{30}}{60 \times 10^{24}} \right)^{\frac{1}{2}} = 577.35$$

$$1.5 \times 10^{11} - x = 577.35x$$

$$577.35x = 1.5 \times 10^{11}$$

$$x = \frac{1.5 \times 10^{11}}{578.35} = 2.59 \times 10^8 \text{ m}$$

Q6. A rocket is fired vertically with a speed of 5 km s^{-1} from the earth's surface. How far from the earth does the rocket go before returning to the earth? Mass of the earth = $6.0 \times 10^{24} \text{ kg}$; mean radius of the earth = $6.4 \times 10^6 \text{ m}$; $G = 6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$.

Ans. $8 \times 10^6 \text{ m}$ from the centre of the Earth

Velocity of the rocket, $v = 5 \text{ km/s} = 5 \times 10^3 \text{ m/s}$

Mass of the Earth, $M_e = 6.0 \times 10^{24} \text{ kg}$

Radius of the Earth, $R_e = 6.4 \times 10^6 \text{ m}$

Height reached by rocket mass, $m = h$

At the surface of the Earth,

Total energy of the rocket = Kinetic energy + Potential energy

$$= \frac{1}{2}mv^2 + \left(\frac{-GM_em}{R_e} \right)$$

At highest point h , $V=0$

$$\text{And, Potential energy} = -\frac{GM_em}{R_e + h}$$

$$= 0 + \left(-\frac{GM_em}{R_e + h} \right) = -\frac{GM_em}{R_e + h}$$

Total energy of the rocket

From the law of conservation of energy, we have

Total energy of the rocket at the Earth's surface = Total energy at height h

$$\frac{1}{2}mv^2 + \left(-\frac{GM_em}{R_e} \right) = -\frac{GM_em}{R_e + h}$$

$$\frac{1}{2}v^2 = GM_e \left(\frac{1}{R_e} - \frac{1}{R_e + h} \right)$$

$$GM_e \left(\frac{R_e + h - R_e}{R_e(R_e + h)} \right)$$

$$\frac{1}{2}v^2 = \frac{GM_e h}{R_e(R_e + h)} \times \frac{R_e}{R_e}$$

$$\frac{1}{2}v^2 = \frac{gR_e h}{R_e + h}$$

$$= \frac{GM}{R_e^2} = 9.8 \text{ m/s}^2$$

Where $g =$

Acceleration due to gravity on the Earth's surface)

$$\therefore v^2 (R_e + h) = 2gR_e h$$

$$v^2 R_e = h(2gR_e - v^2)$$

$$h = \frac{R_e v^2}{2gR_e - v^2}$$

$$= \frac{6.4 \times 10^6 \times (5 \times 10^3)^2}{2 \times 9.8 \times 6.4 \times 10^6 - (5 \times 10^3)^2}$$

$$h = \frac{6.4 \times 25 \times 10^{12}}{100.44 \times 10^6} = 1.6 \times 10^6 \text{ m}$$

Height achieved by the rocket with respect to the centre of the Earth

$$= R_e + h$$

$$= 6.4 \times 10^6 + 1.6 \times 10^6$$

$$= 8.0 \times 10^6 \text{ m}$$

Q7. Two heavy spheres each of mass 100 kg and radius 0.10 m are placed 1.0 m apart on a horizontal table. What is the gravitational force and potential at the mid point of the line joining the centers of the spheres? Is an object placed at that point in equilibrium? If so, is the equilibrium stable or unstable?

Ans. 0;

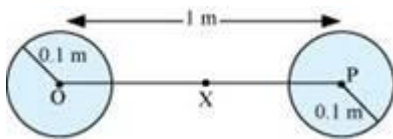
$-2.7 \times 10^{-8} \text{ J/kg}$;

Yes;

Unstable

Explanation:

The situation is represented in the given figure:



Mass of each sphere, $M = 100 \text{ kg}$

Separation between the spheres, $r = 1 \text{ m}$

X is the midpoint between the spheres. Gravitational force at point X will be zero. This is because gravitational force exerted by each sphere will act in opposite directions.

Gravitational potential at point X:

$$= \frac{-GM}{\left(\frac{r}{2}\right)} - \frac{GM}{\left(\frac{r}{2}\right)} = -4 \frac{GM}{r}$$

$$= \frac{4 \times 6.67 \times 10^{-11} \times 100}{r}$$

$$= -2.67 \times 10^{-8} \text{ J / kg}$$

Any object placed at point X will be in equilibrium state, but the equilibrium is unstable. This is because any change in the position of the object will change the effective force in that direction.

MCQs with Answers for Gravity

1.The value of gravitational acceleration on Earth is

- (a) 9.8 m/s
- (b) 9.8 m/s²
- (c) 8.9 m/s
- (d) 8.9 m/s²

Answer:(b) 9.8 m/s²

2.What would be the height of an artificial satellite so that it remains stationary with respect to earth's surface?

- (a) 36000 km above the earth surface
- (b) 40000 km above the earth surface
- (c) 26000 km above the earth surface
- (d) 63000 km above the earth surface

Answer:(a) 36000 km above the earth surface

3.The value of gravitational acceleration

- (a) increases as height increases from the earth.
- (b) decreases as height increases from the earth.
- (c) remains constant.
- (d) None of the above

Answer:(a) increases as height increases form the earth.

4.The centripetal force required by the artificial satellite to revolve around earth is provided by

- (a) fuel contained in the satellite
- (b) gravitational force due to sun

- (c) gravitational force due to earth
- (d) Thrust produced by burning fuel

Answer:(c) gravitational force due to earth

5.In which region of earth the weight of a body is slightly greater than the other regions?

- (a) At polar region
- (b) At equator
- (c) Tropic of Cancer
- (d) None of this

Answer:(a) At Polar region

6.What is the value of the escape velocity of earth?

- (a) 9.8 km/sec
- (b) 10 km/sec
- (c) 11.2 km/sec
- (d) 12 km/sec

Answer:(c) 11.2 km/sec

7.How much time does a polar satellite take to complete one revolution around earth?

- (a) 1 hour 30 min
- (b) 2 hours
- (c) 2 hour 20 min
- (d) 3 hour

Answer:(b) 2 hours

8.In case of a planet's motion on its circular orbit

- (a) its velocity remains constant in its orbit
- (b) its angular velocity remains constant
- (c) its total angular momentum remains constant.
- (d) radius of orbit remains constant.

Answer:(b) its angular velocity remains constant

9.If the radius of earth is decreased keeping its mass constant, then the length of day will

- (a) decrease
- (b) Increase
- (c) remain same
- (d) cannot say

Answer:(a) decrease

10.If the earth would stop rotating then the weight of an object on north pole will

- (a) increase
- (b) decrease
- (c) remain same
- (d) be zero

Answer:(c) remain same

11.What is the weight of a body inside an artificial satellite of earth?

- (a) It depends on the mass of the body.
- (b) It depends on the velocity of satellite
- (c) Product of its mass and gravitational acceleration
- (d) Zero

Answer:(d) Zero

12.Weight of an object in a free fall is equal to

- (a) mass of the object \times gravitational acceleration
- (b) Zero
- (c) greater than when the object was at rest
- (d) less than when the object was at rest

Answer:(b) Zero

13.Which of the following is true for an artificial satellite revolving around earth in a circular orbit?

- (a) Its linear velocity is constant.
- (b) Its acceleration is constant.
- (c) Its acceleration keeps on changing.
- (d) Its angular velocity is constant.

Answer:(c) Its acceleration keeps on changing.

14.How does the weight of a body change with an increase in the speed of rotation of the earth?

- (a) Weight of the body will increase
- (b) Weight of the body will decrease
- (c) Weight of the body remains constant
- (d) Cannot be answered

15.What would be the escape velocity of a planet whose mass and radius are double from earth with the escape velocity of earth taken as V_e .

- (a) V_e
- (b) $2 V_e$
- (c) $4 V_e$
- (d) $16 V_e$

Answer:(b) $2 V_e$

FILL IN THE BLANKS

1. The gravitational force is a _____ force.

Answer: The gravitational force is a central force.

2. The acceleration due to earth's gravity decreases as _____ increases.

Answer: The acceleration due to earth's gravity decreases as altitude increases.

3. The energy of the satellite is _____.

Answer: The energy of the satellite is negative.

4.The gravitational potential energy is a _____ quantity.

Answer: The gravitational potential energy is a scalar quantity.

5.The escape speed is _____ of mass of the object.

Answer: The escape speed is independent of mass of the object.

6._____ proposed heliocentric theory.

Answer: Nicolaus copernicus proposed heliocentric theory.

7.Due to rotation of the earth, the acceleration due to gravity is maximum at _____.

Answer: Due to rotation of the earth, the acceleration due to gravity is maximum at poles.

8. _____ experimentally determined the value of gravitational constant 'G' using a torsion balance.

Answer: Henry cavendish experimentally determined the value of gravitational constant 'G' using a torsion balance.

9. According to Kepler's second law, the radius vector to a planet from the Sun sweeps out equal areas in equal intervals of time. This law is a consequence of the conservation of _____.

Answer: According to Kepler's second law, the radius vector to a planet from the Sun sweeps out equal areas in equal intervals of time. This law is a consequence of the conservation of angular momentum.

10. A geostationary satellite is orbiting the earth at a height of $6R$ above the surface of the earth, where R is the radius of the earth. The time period of another satellite at a height of $2.5 R$ from the surface of the earth is hours.

Answer: A geostationary satellite is orbiting the earth at a height of $6R$ above the surface of the earth, where R is the radius of the earth. The time period of another satellite at a height of $2.5 R$ from the surface of the earth is $6\sqrt{2}h$ hours.