MDO (Multi-disciplina	ry design & optimisation)
-> Syllabus (Geodient opti	obust, Multi-disciplinary, Calculation pti ophi: gradient )
-> Grading Mult	eb ,
Intro	Quiz 2(20%) Endsen (40%)
Szinivasan - OR	Internal [ - (40%)
Deepayali - OR Chaitarya - OR	
Honhar - OR Mathemat	onstraints/cost/
Raykumar - OR Design Dharmik - OR Pratik (Structure) OK	
Mital (Struct) -	
Min. If of marriner require	ed to produce a product.
-> time of madining	>/ov ball bearing
5 Cost of 1	
<i>f</i> <sub>1</sub> <i>f</i> <sub>2</sub>	3
	# of marchine
· · · · · · · · · · · · · · · · · · ·	cost of production of unit product time of machining
Constraint	
· · · · · · · · · · · · · · · · · · ·	
Aircraft	
O Objectives ( Am. Dray	ofton.
3) Range B Payload	
X & Man altitud	<u>le.</u>
	, production, maintenance, operations)
D. Size	
-CRA Military	VONETTI · · · · · · · · · · · · · · · · · ·
ISRO MAINTEN	PROPITION D
P	

Hacrooffedrign problem 1) Objective function.
2) Constraints

3 Design variables.

Objective functions <==> Constraints

These two are sometimes interchangeable.

- 1. Objective function 2. Constraints
- 3. Design variables



Constraint 
$$g(T, \alpha, a, b) = \delta \leqslant \delta_{max}$$

geometry)

Objective: 
$$f(a,b) = b \cdot a^2 \cdot s_{max}$$

const.

28/01/2021

Objective functions are dependent on the perspective (the end client). For e.g., I can optimise an aircraft for:

- manufacturer (low manufacturing cost)
  user (maximise safety and comfort)
  Airlines (minimise operating costs) etc.

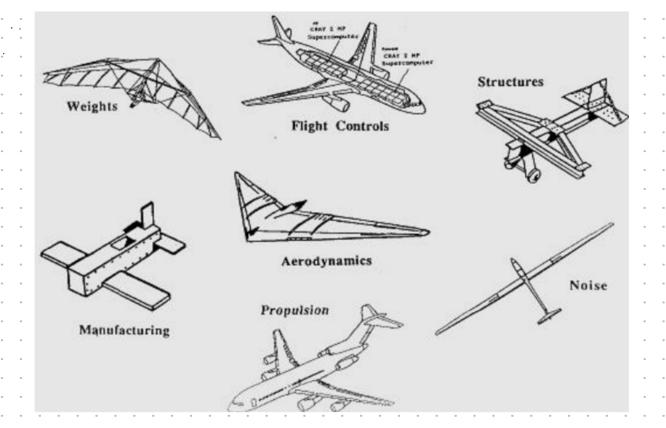
We note that there are multiple disciplines

But this course is about only the engineering aspect of the design and optimisation process.

So what are the various disciplines (engineering streams) required to design an aircraft?

Aerospace Aerodynamics/ APM Controls

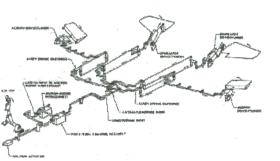
/ Structures/Avionius/Propulsion



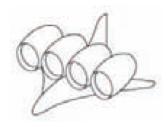


Aerodynamics group

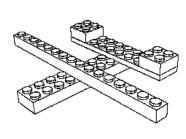
FIGURE 3 - FLIGHT CONTROL SYSTEM -- LATERAL CONTROLS



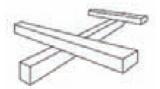
Flight control systems group



Propulsion group



Manufacturing group



Structure group

Stealth group

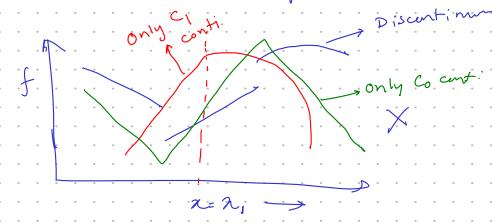
We begin our journey with 1D Unconstrained Optimisation problem.

min.  $f(\bar{x})$  $\bar{x} \in \mathbb{R}^N$  (N=1)

What about  $\bar{\pi} \in \mathbb{Z}^N$  (In legers) E.g. How many engines should I have?

Σ C B (Boolean) E.g. Should I use canard configuration?

RER - Continuou oph



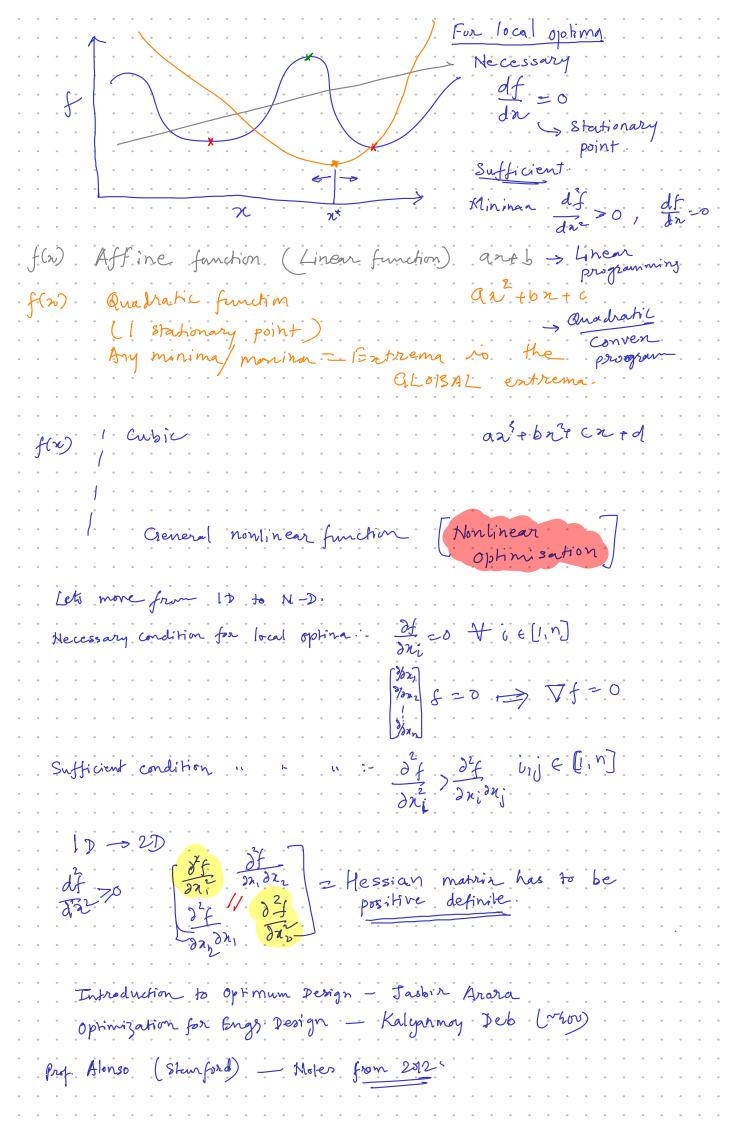
Co continuous - f is continuous

df is continuous - C, continuous

de 15 continuous - C, continuous

de 27 io 11 - C2 conti

An derivatives are continuous. Analytic ex, sincro, cor(x) function



Design a bucket of circular cross-section that can store at least 10 litres of water.

Assumption h- height of boncket 1) Bruket is cylindrical.
(1) Cost of bruket & surface of -> Design Variables r - gradius "  $f(h,r) = (\pi r^2 + 2\pi r h)$ obj function 3 All other object ignored. V(K,n) 7/10 Constraint V(h,r) = 722h. 九を2~710  $\begin{cases} (2h) \Rightarrow (1,2) \\ (d,h) \Rightarrow (212) \end{cases}$ りくれられっ  $x_1 \leq x \leq x_2$ (42, 711) => (212, 111) (r,h) f(h, x) = Same ) (ht2, 2t1) H=h+2, R=2t1 => h=H-2, 2= X, 2 da, t B  $(\lambda_1, \lambda_2)$ All linear combinations X 2= Y N21 & are also possible

Mattab "fmincon".

Also read, the documentation of fmincon

List names of of algos in fruincon.

List stopping for "fmincon".

By Friday. 12 pm.

Lec 4:

$$\frac{df}{dx} = 0$$
,  $\frac{d^2f}{dx^2} > 0$ 

 $\frac{df}{dn} \left| \frac{d^2f}{dn^2} \right| = \frac{d^2f}{dn^2} \left| \frac{(dn)^2 + HOT}{n} \right|$  $f(n^*+\Delta x) = f(n^*) +$ 

Local minima.

Local minima.

$$\Delta f = f(n^2 + \Delta x) - f(n^2) = \frac{df}{dx} \Delta x + \frac{d^2f}{dx^2} \Delta x + \frac{d^2f}$$

$$\frac{df}{dx} = 0$$

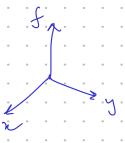
$$\frac{d^2f}{dx^2} = 0$$

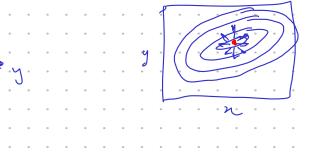
$$\frac{d^2f}{dx^2} = 0$$

$$\frac{d^2f}{dx^2} = 0$$

$$f(n) \geqslant f(n^{\prime})$$

$$8x = 10^{-10}$$





$$f(\pi) = \sum_{j=1}^{n} \sum_{j=1}^{n} p_{ij} n_i n_j$$

$$f(x_{1},x_{2}) = p_{11}x_{1}^{2} + p_{12}x_{1}x_{2} + p_{21}x_{2}x_{1} + p_{22}x_{2}^{2} = p_{11}x_{1}^{2} + (p_{12}+p_{21})x_{1}x_{2} + p_{22}x_{2}^{2}$$

$$= [x_{1} x_{2}][p_{11} p_{12}][x_{$$

If 
$$\bar{n} = \begin{bmatrix} n_1 \\ n_1 \end{bmatrix}$$
, then  $f(a_1, n_2) = \bar{n}^T p \bar{x}$ 

$$[xy, yxy, hx]$$

$$f(n_1,n_2) = \begin{bmatrix} n_1 & n_2 \end{bmatrix} \begin{bmatrix} p_1 & (p_1 + p_2)/2 \end{bmatrix} \begin{bmatrix} \chi_1 \\ n_2 \end{bmatrix} = \overline{n}^T A \overline{\lambda}.$$

$$(p_1 + p_2)/2 \quad p_2 = \overline{n}^T A \overline{\lambda}.$$

Taylor series exp. 
$$n^n fm$$

$$f(\vec{n} + o\vec{n}) = f(\vec{x}^*) + \nabla f(\vec{n}^*) \cdot \Delta \vec{x} + \frac{1}{2} \Delta \vec{n}^T H \Delta \vec{x} + 40T$$

A: The lowest non-zero derivative must be even ordered for a stationary point and positive for a local minimum point.

be a minima?

f(x)  $\frac{df}{dx}\Big|_{x^{2}} = 0$   $\frac{df}{dx^{2}}\Big|_{x^{2}} = 0$  $\frac{d^3f}{da^3}\Big|_{a^4} = 0 \qquad \frac{d^4f}{da^4}\Big|_{a^8} > 0$ - can I say at is not a minima. -> I cannot comment. Emplicit (b)  $y = Z^2$ ,  $Z = n^3$ ,  $\Rightarrow$ Dependent intermediate independent variable  $G \quad y = \sin(x) \quad = )$ CL(P) 1 Spride Implicit  $S C_L(\bar{x}) \equiv C_L(p(\bar{x}))$  $\begin{cases} GE(\bar{p}, \bar{x}) = 0 \end{cases}$ GE = Euler/N.S.  $\begin{cases} y^2 \neq 2 \\ \frac{d^2}{dn} + \sin(n) \cdot \cos(2) \end{cases}$ Black -1302 - Don't know if this function is C', C', or c' continuous. Creometry) in Thumbrules)

Cometry) in Thumbrules)

Cometry) in Thumbrules) CL(Z) independent

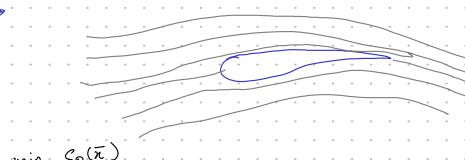
Lec 5. (oh/or/21)

Recap: Given the functional form of an objective function f(x), we know the necessary and sufficient condition to find a minima.

Q: Do we always know f(x) explicitly?

· (Emplicit, implicit & black-box functions)

Let us consider a standard Aerospace optimisation problem.



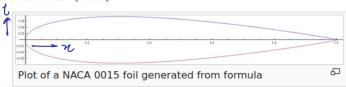
Design an airfoil that generates minimum drag at Cl=1.2, ( ) AOA=3 degrees, ( ) mach=0.6, ( M )

11 km ISA conditions.

min  $C_p(\bar{x})$  $\bar{x} \in \mathbb{R}$ subject to:  $C_1(\bar{x}, \alpha, M) = 1.2$ 

## Equation for a symmetrical 4-digit NACA airfoil [edit]

The formula for the shape of a NACA 00xx foil, with "xx" being replaced by the percentage of thickness to chord, is<sup>[4]</sup>



$$y_t = 5t \left[ 0.2969 \sqrt{x} - 0.1260 x - 0.3516 x^2 + 0.2843 x^3 - 0.1015 x^4 
ight],$$
 [5][6]

where:

x is the position along the chord from 0 to 1.00 (0 to 100%),

 $y_t$  is the half thickness at a given value of x (centerline to surface),

t is the maximum thickness as a fraction of the chord (so t gives the last two digits in the NACA 4-digit denomination divided by 100).

Sonræ. Wikipedia

Lec. 6:- 05/02/21. Descent methods. (Gradent descent methods) information it only using Po not use  $H(\bar{x}^{(\kappa)})$   $f(n_1, n_2) = 2n_1 n_2 + n_1^2$ Problem statement  $\min_{\bar{\mathbf{z}} \in \mathbb{R}^{N}} f(\bar{\mathbf{z}})$  $\rightarrow \nabla f(\bar{x}) = C$   $\rightarrow \bar{x}^{*} - minima for <math>f(\bar{x})$  $\nabla f = \begin{bmatrix} 2x_2 + 2x_1 \\ 2x_1 \end{bmatrix}$ d - direction vector 1) Initialise \(\frac{7}{\tau}\) (randomly/houristic/Prior experience)

(2) Calculate \(\frac{7}{\tau}\)(\(\tau^{(\dots)}\)) x= 1 & xi  $\vec{\sigma}^{(0)} = \vec{F} (\vec{x}^{(0)})$ G calculate slepsize of  $\vec{\chi}^{(i)} = \vec{\chi}^{(i)} + \propto \vec{d}$  $\|\nabla f(\bar{x}^{(k)})\|_2 \leq 10^6$   $\Rightarrow$  Tolerance 3 Stoppage criberia (A)  $J^{(k)} = -\nabla f(\bar{x}^{(k)}) \implies 81$  expest descent method. (B)  $J^{(k)} = -\nabla f(\bar{x}^{(k)}) \implies 81$ Choice of descent direction Stochastic

2 (x (x))

where i

is chosenly

rendomly.

(Uniform/Normal) D Slight variation (A)

Descent direction 2 (K1) = (K) + (K) = (K)  $\nabla f(\bar{x}^{(c)}) \cdot \bar{d}$  $f(\bar{\chi}^{(k)},\bar{\chi}^{(k)},\bar{\chi}^{(k)}) \propto f(\bar{\chi}^{(k)}) + \chi^{(k)}$  $C^{(k)}$  T  $T^{(k)}$ After every steps, soll has to improsve.  $f(\bar{x}^{(k_7)}) \leq f(\bar{x}^{(k_7)})$ from 0 80) (k) - J(k) < 0 \* If  $\tilde{c}^{(k)}$ ,  $J^{(k)} = 0$   $\Longrightarrow$  Easet stepsize calculation  $\nabla f(\bar{\mathbf{a}}^{(\kappa)})^{\top} \cdot \bar{\mathbf{d}} \leqslant 0$ en Branch Contract of the Cont Vf(x (1K)) 3,7 - Vf. 1 20 Tf d = | Ts | [al · Cos (Lafia) 1,2,8 → Vf.d>0 4, 5,6 - Vf-d <0 - Stoperst descent 1. Steeperst descent 2. Coordinate 3. Stochastic J(K) = - \{(xx)) - , 2 (x)

Let 
$$f''$$
 Stepsize calculation.

$$\frac{(k!)}{2} = \frac{(k)}{2} + \frac{(k)}{2} = f(k)$$

$$f(x^{(k!)}) = f(x^{(k)} + x^{(k)}, d^{(k)}) = f(k)$$

$$f(\bar{n}^{(\kappa')}) = f(\bar{n}^{(\kappa)}) = f(\bar{n}^{(\kappa)})$$

$$\frac{df}{dx} = 0$$
  $\Rightarrow$  Exact step-sije

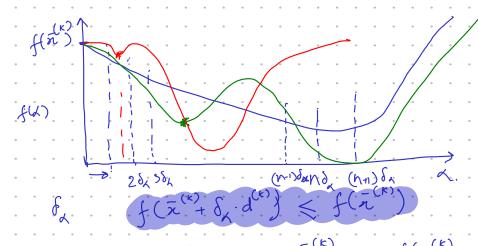
Eg. 
$$f(x) = 0.1x_1^2 + x_2^2 - 10$$
,  $n^{(1)} = (5,1)$ 

$$f(\zeta) = f([5] + \alpha[-1]) = f(5-4, 1-2\zeta)$$

$$= 0.1(5-\zeta)^{2} + (1-2\zeta)^{2} - 10$$

$$\frac{df}{d\lambda} \leq 10^{-15}$$

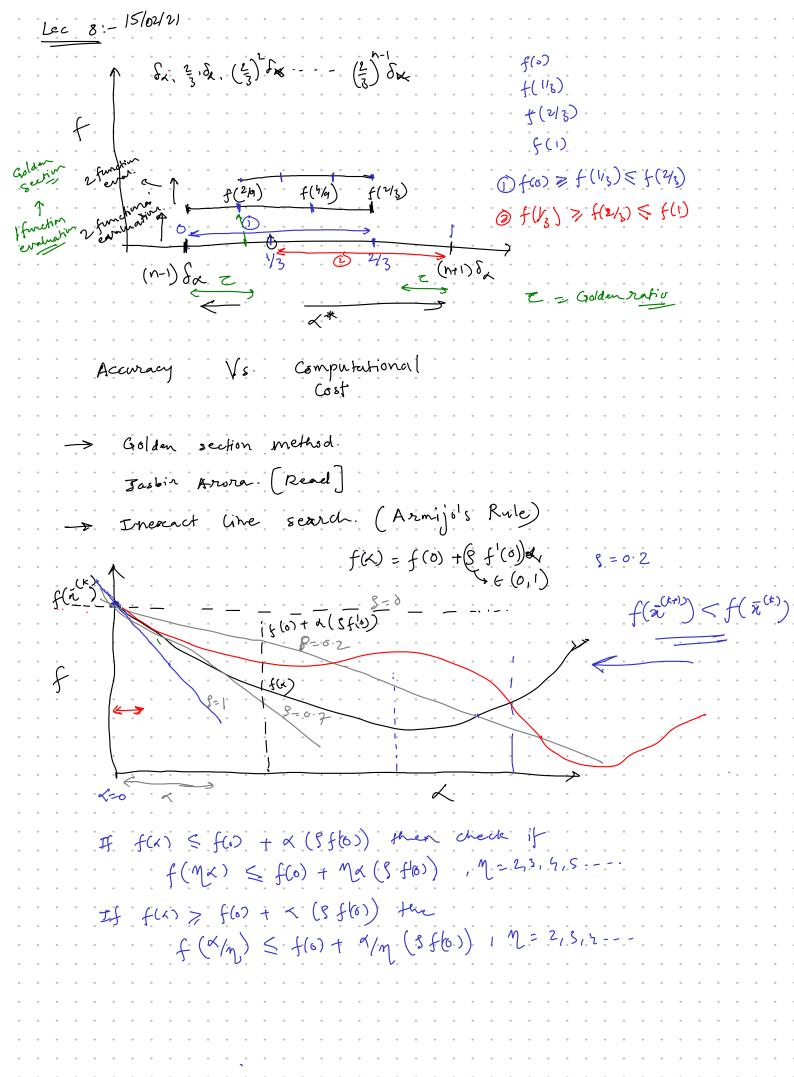
$$\chi = 0.64$$
 $\chi^{(1)} = \begin{bmatrix} 8 \\ 1 \end{bmatrix} + 0.64 \begin{bmatrix} -1 \\ -2 \end{bmatrix} =$ 

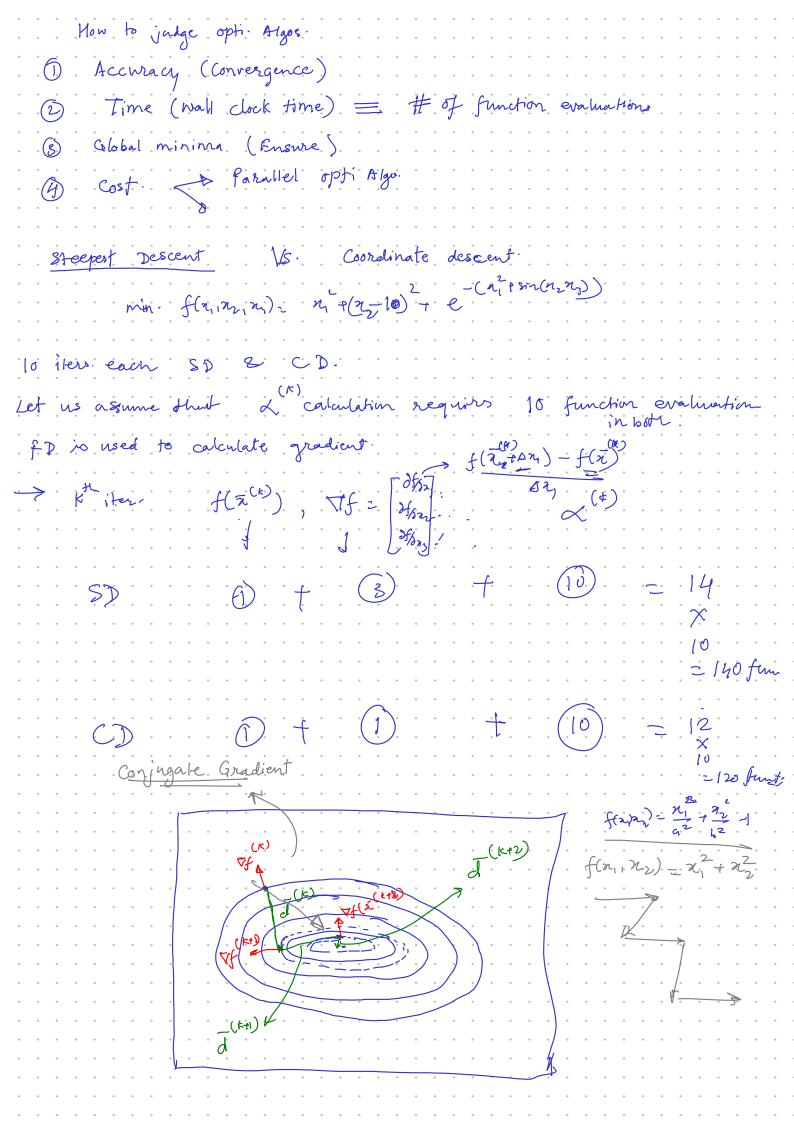


$$(1) \Rightarrow f(\bar{n}^{(k)} + (n+1) \delta_{k} \cdot \bar{d}^{(k)}) < f(\bar{n}^{(k)} + n \delta_{k} J^{(k)})$$

$$2) \Rightarrow f(\bar{\lambda}^{(k)} + (n_{1}) \delta_{\lambda} J^{(k)}) > f(\bar{\lambda}^{(k)} + n \delta_{\lambda} J^{(k)})$$

ICFD e IFEM soln.

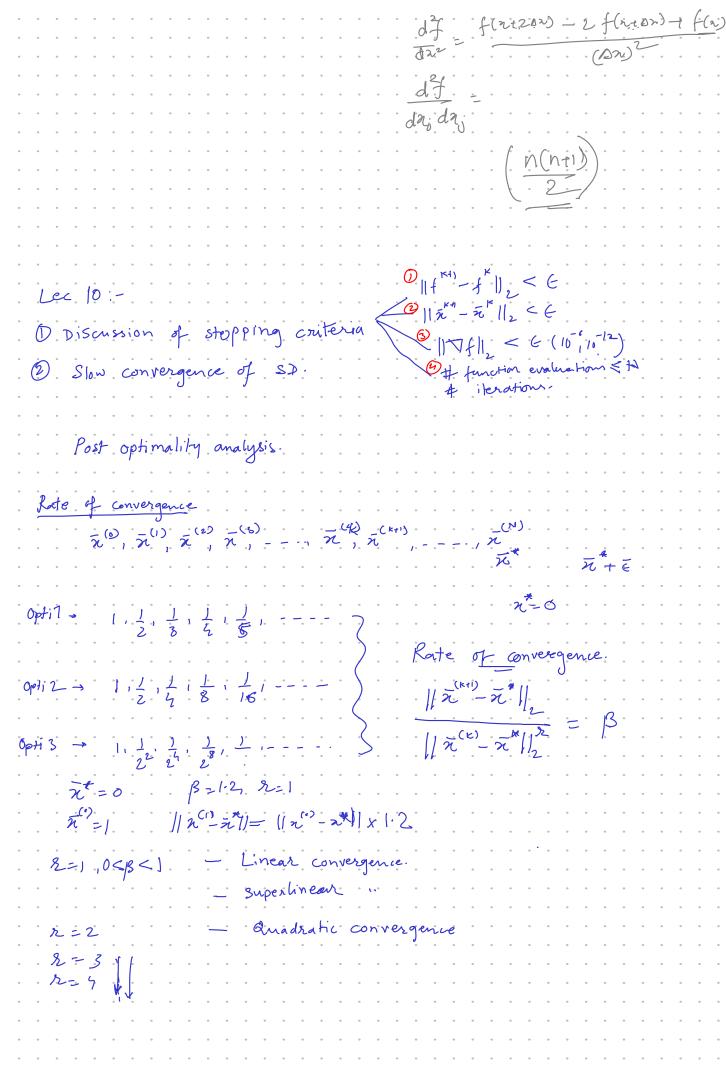




Meulation.

(KY) T

(KY) = 0 Caradient at (kH) the iter is orthogonal to descent diretten at (k) iter. 7( k+1). 2 70 .+ X . J.  $f(\bar{a}^{(k+1)}) = f(\bar{a}^{(k)} + \sqrt{d}^{(k)})$  $\frac{df}{d\lambda} = 0 \implies \frac{df(\bar{n}^{(kn)})}{d\lambda}$ Jf ( in ( ke) ) - d = min f(zi) · (20100) Algo 1 . Aloja . 2..  $\rightarrow \sqrt{f}$ ,  $\sqrt{f}$ · Forward FD .. -> Forward FD. Zanore step-sije Cal > # of function evalutiation (A of derign variables) 1) function evaluation n functiona ...  $\frac{d}{dr} \left[ \begin{array}{c} f(2t\Delta n) - f(2) \\ \hline \Delta x \end{array} \right] = \frac{f(2t\Delta n) - f(2t\Delta n)}{(2t\Delta n) + f(2t\Delta n)} + f(2t\Delta n)$ 



```
Two ways to extimate the rate of convergence:
D Exporimental. (Assumer that you leave \bar{\lambda}^*). Usually, \bar{\chi}^{(k)} is assumed to be the \bar{\pi}^* and then the rate of convergence is found out.
2) Asymptotic theoretical analysis under simplifying assumptions
          (1) Oved. - but con
                             x'' = \beta \cdot (x'^{(0)})^2 = 0.9 \times 1 = 0.9
   x =0
                              x^{(\nu)} = |3 \cdot (x^{(\nu)})^2 = 0.9 \times (0.9)^2 = 0.7.9.9
   n = 1
                              \lambda^{(3)} = (0.9) \cdot (0.729)^2 =
                                                                         Qudratil
    2-2
                                                                   3) Good initial gueen
   B= 0.9
                                                                       x = 0.1
        2 Chadratic - good convergence const.
                                                                       B = 00)
                               201) = 0:1×(1) = 0:1
                                                                        50 = 6.1 × (0.1) = 10
   B = 0.1
                                2(0) = 0.1 × (0.1) = 0.001
                                                                         x(2)= 0.1×(10-3)2=10-7
                                 \chi^{(4)} = 0.1 \times (10^{-3})^2 = 10^{-2}
     Vf into- 1 storder info.
                                                    1 (linear conv.)
  Tf appront of - 1 order infort - history

Tf of infort - 1822 order infor-
                                              - 12822 (superlinear)
                                                      2 (Qualtratic)
                                                                     Newton interpolation
     11, 12, 174, 1781
                                                                         21. 221 23.
                                                                       f(n_1) f(n_2) f(n_3)
                                            f(x) - f(x_1)
        \frac{df}{dx^2} = \frac{f(n+2x) - f(n)}{3x}
                                                 22-2h
                                                               1 f(n)=f(n2)-f(1.)
        (x^{(k)}, x^{(k)}, \dots, x^{(n)}, \dots, x^{(n)})
(x^{(k)}, x^{(n)}, \dots, x^{(n)})
   x , x , x (1)
                                                       B, 2
                                                         x k=1,--, n-2
         (Least squares method)
```

$$-2 \le 2 \le 2 \implies 0 \le y \le 1$$

$$y = 2 + 2$$

$$y$$

$$f(n^{(k)}+\Delta n) = f(n^{(k)}) + c^{T}\Delta x + \frac{1}{2} \text{ on Hon} \qquad 0 \vec{n} = \alpha \vec{d} + \frac{6}{3} \vec{d} \qquad 0 = -\alpha \vec{d} \qquad$$

$$\frac{cT}{dn} \leq 0 \implies \frac{cT}{dn} \leq 0$$

$$\frac{dn}{dn} = \frac{d^{(n)}}{dn} = -H^{\frac{1}{2}} c \implies \frac{c^{(n)}}{c} = -H^{\frac{1}{2}} d$$

$$\frac{cT}{cT} \leq 0$$

In the neighbourhood of the minima, it is there definite

Hewton direction is a descent direction in the neighbourhood of minime.

Q:- But what happens away from the optima?

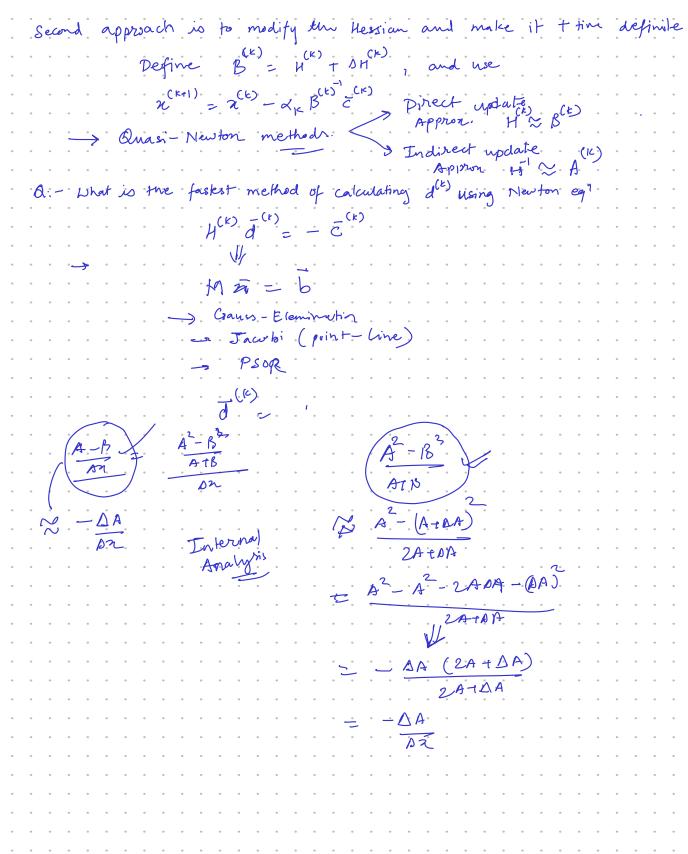
— we cannot guarantee that the Newton direction is a descent direction.

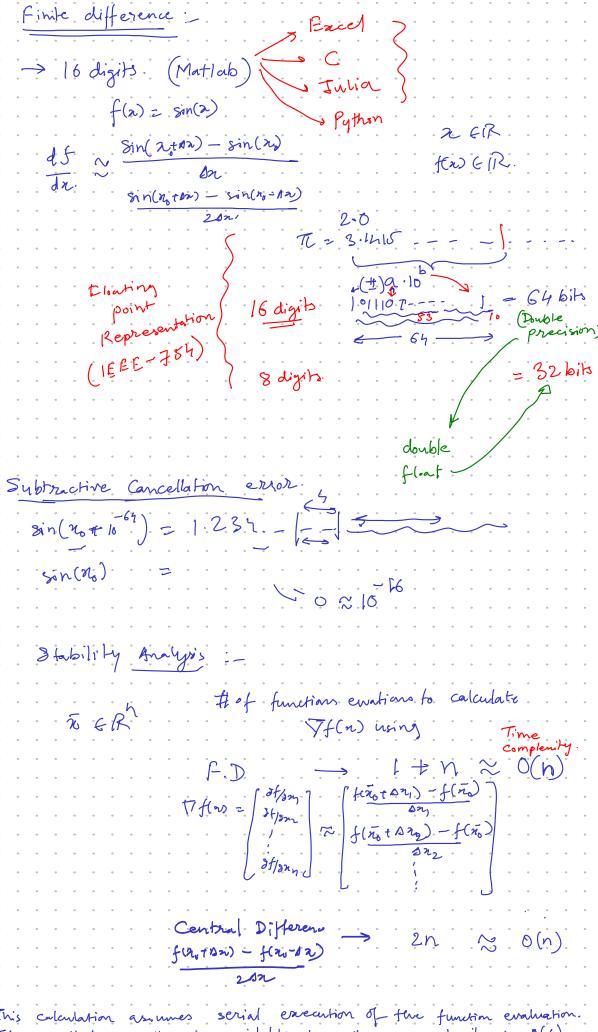
-> One option is instead of doing

$$\chi^{(k+1)} = \chi^{(k)} - H = C^{(k)}$$
 Newfor skp

where  $\alpha_{k}$  can be found from Armijo's rule.

If  $d_{k}=1$ ,  $\Rightarrow$  Pure Newton's method. If  $d_{k}\neq 1$ , modified Newton's method.





This culculation assumes serial execution of the function evaluation. If parallel computing is available then fine complexity is O(1), space complexity for serial execution -O(1) parallel 1 - O(n)

	Black-box objective function.	
	finite difference is the	only choice.
Step-size C the opti-	alculation is done prior. Algo!	- to starting
Sprie complexity	:- f(n) = L(n)	
	10° grid points.	
	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	5 x 10 number.
		double variable
	Memory requires	4 X 5 X 10 X 2-5 / Han sol ) [SBMB]
	CFD OK 1	EM sol ) SBMB
Serial execution.	- fix,), f(x, t sz,), f(x, tsz	
Parallel n.	- 50 MB + 50 MBt	> (1)
	(n)	MR
tosignment:	Spave fime	
	H(Tio)	seria)
		parallel.
using	F-D	

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