

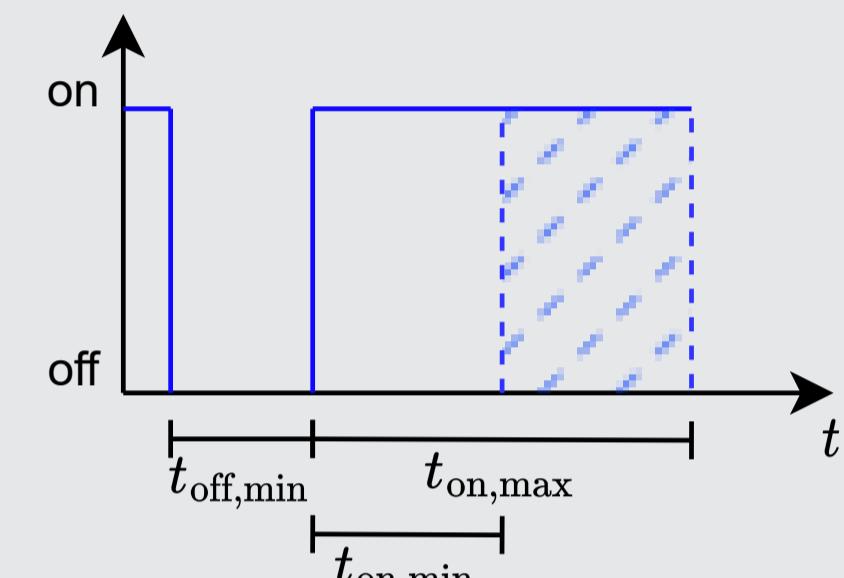
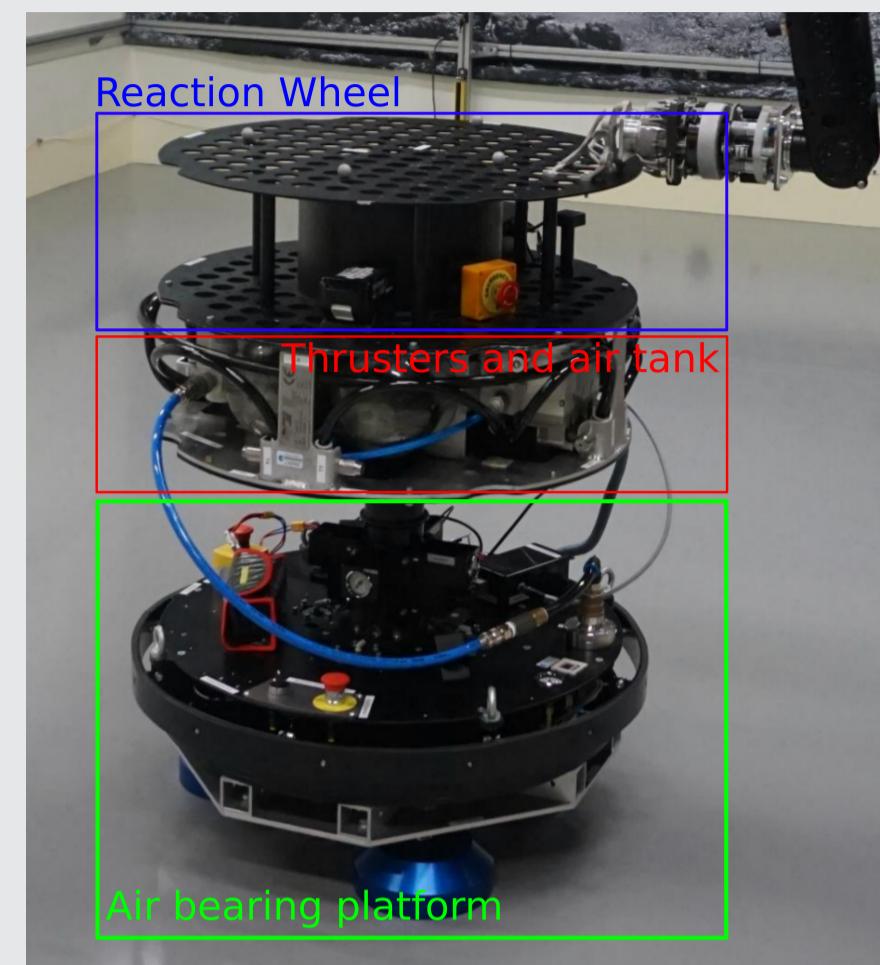
# Mixed Integer MPC for a Free-Floating Satellite Testbed

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## ESA's free-floating platform REACSA [1]

- 220 kg air-bearing platform
- floats on 9m × 5m flat-floor
- Reaction Wheel for precise torque control
  - RW speed limits lead to saturation
- 8 thrusters apply linear and angular acceleration
  - on/off (**binary** actuated) **thrusters**
- thrusters have activation **time constraints**
  - minimum off time:  $t_{\text{off,min}}$
  - minimum on time:  $t_{\text{on,min}}$
  - maximum on time:  $t_{\text{on,max}}$



## Enforcing binary inputs ( $\mathbf{u}_{\text{bin}} \in \mathbb{U}_{\text{bin}}$ )

- Mixed Integer Program (MIP)
- $$u_{i,t+k|t} \in \{0, 1\}, \forall u_i \in \mathbf{u}_{\text{bin}}, \forall k \in [0, N] \quad (1)$$

Requires special solver

- Penalty Term

$$\sum_{j=1}^3 4\beta (u_{i,t+k|t} - u_{i,t+k|t}^2), \beta > 0, \forall u_i \in \mathbf{u}_{\text{bin}} \quad (2)$$

Becomes non-convex Quadratic Program (QP)

- Linear Complementarity Constraints (LCC)

$$0 \leq (1 - u_{i,t+k|t}) \perp u_{i,t+k|t} \geq 0, \forall u_i \in \mathbf{u}_{\text{bin}} \quad (3)$$

Becomes Mathematical Program with Complementarity Constraints (MPCC)

## Activation time constraints ( $\mathbf{u}_{\text{bin}} \in \mathbb{U}_{\text{time}}$ )

- minimum off time:  $t_{\text{off,min}} = 0.1 \text{ s} = \Delta t$   
Enforced naturally by zero-order hold
- minimum on time:  $t_{\text{on,min}} = 0.2 \text{ s} = 2\Delta t$   
 $+u_{i,t+k-1|t} - u_{i,t+k|t} + u_{i,t+k+1|t} \leq 1, \forall k \in [-2, N-1], \forall i \in \mathbf{u}_{\text{bin}}$
- maximum on time:  $t_{\text{on,max}} = 0.3 \text{ s} = 3\Delta t$   
 $\sum_{j=k}^{k+3} u_{i,t+j|t} \leq 3, \forall k \in [-3, N-3], \forall i \in \mathbf{u}_{\text{bin}}$

$t_7$	$t_8$	$t_9$
1	0	1
$\oplus$	$\ominus$	$\oplus$
= 2	$\leq 1$	
$t_7$	$t_8$	$t_9$
0	0	0
$\oplus$	$\ominus$	$\oplus$
= 0	$\leq 1$	
$t_7$	$t_8$	$t_9$
1	0	0
$\oplus$	$\ominus$	$\oplus$
= 1	$\leq 1$	
$t_7$	$t_8$	$t_9$
0	1	0
$\oplus$	$\ominus$	$\oplus$
= -1	$\leq 1$	

$t_7$	$t_8$	$t_9$
0	0	0
$\oplus$	$\ominus$	$\oplus$
= 0	$\leq 1$	
$t_7$	$t_8$	$t_9$
1	1	0
$\oplus$	$\ominus$	$\oplus$
= 0	$\leq 1$	
$t_7$	$t_8$	$t_9$
1	1	1
$\oplus$	$\ominus$	$\oplus$
= 4	$\leq 3$	
$t_7$	$t_8$	$t_9$
0	1	0
$\oplus$	$\ominus$	$\oplus$
= 1	$\leq 1$	
$t_7$	$t_8$	$t_9$
1	1	1
$\oplus$	$\ominus$	$\oplus$
= 3	$\leq 3$	

## System model

$$\dot{\mathbf{x}} = \underbrace{\begin{bmatrix} \mathbf{0}^{3 \times 4} & \mathbf{I}^{3 \times 3} \\ \mathbf{0}^{4 \times 7} \end{bmatrix}}_{\mathbf{A}} \underbrace{\begin{bmatrix} x & y & \theta & \dot{x} & \dot{y} & \dot{\theta} & \omega_{\text{RW}} \end{bmatrix}^T}_{\mathbf{x}} \quad (4)$$

$$+ \underbrace{\begin{bmatrix} 0 & -s\theta F_n & s\theta F_n & -c\theta F_n & c\theta F_n & -s\theta F_n & c\theta F_n \\ 0 & c\theta F_n & -c\theta F_n & -s\theta F_n & s\theta F_n & -c\theta F_n & s\theta F_n \\ -1 & \frac{F_n r}{I_s} & -\frac{F_n r}{I_s} & \frac{F_n r}{I_s} & -\frac{F_n r}{I_s} & \frac{F_n r}{I_s} & -\frac{F_n r}{I_s} \end{bmatrix}}_{\mathbf{B}(\theta)} \underbrace{\begin{bmatrix} \tau \\ \mathbf{u}_{\text{bin}} \end{bmatrix}}_{\mathbf{u}} \quad (5)$$

With  $\tau \in \mathbb{R}$  and  $\mathbf{u}_{\text{bin}} = [u_0 \ u_1 \ u_2 \ u_3 \ u_4 \ u_5 \ u_6 \ u_7]^T \in \{0, 1\}^8$

## Model Predictive Control formulation

Finite horizon optimal control problem:

$$J^*(\mathbf{x}_t) = \min_{\mathbf{u}_t, \mathbf{x}_t} \mathcal{L}_f(\mathbf{x}_{t+N|t}) + \sum_{k=0}^{N-1} \mathcal{L}(\mathbf{x}_{t+k|t}, \mathbf{u}_{t+k|t}) \quad (6a)$$

$$\text{s.t. } \mathbf{x}_{t+k+1|t} = \mathbf{x}_{t+k|t} + \Delta t \mathbf{A} \mathbf{x}_{t+k|t} + \Delta t \mathbf{B} \mathbf{u}_{t+k|t}, \forall k \in [0, N] \quad (6b)$$

$$-\tau_{\max} \leq \mathbf{u}_{0,t+k|t} \leq \tau_{\max}, \forall k \in [0, N] \quad (6c)$$

$$\mathbf{x}_{lb} \leq \mathbf{x}_{t+k|t} \leq \mathbf{x}_{ub}, \forall k \in [0, N] \quad (6d)$$

$$\mathbf{x}_{f,lb} \leq \mathbf{x}_{t+N|t} \leq \mathbf{x}_{f,ub} \quad (6e)$$

$$\mathbf{u}_{bin,t+k|t} \in \mathbb{U}_{bin}, \forall k \in [0, N] \quad (6f)$$

$$\mathbf{u}_{bin,t+k|t} \in \mathbb{U}_{time}, \forall k \in [0, N] \quad (6g)$$

$$\mathbf{x}_{t|t} = \mathbf{x}_t \quad (6h)$$

- Discretization  $\Delta t = 0.1 \text{ s}$
- Closed loop control cycle with 100 ms

## Feasibility analysis

On a simplified model (4 thrusters, no reaction wheel) the feasibility of all three binary input formulations is compared:

Linear Mixed Integer:

- Feasible solutions within 100 s
- For short prediction horizons solutions are optimal enough

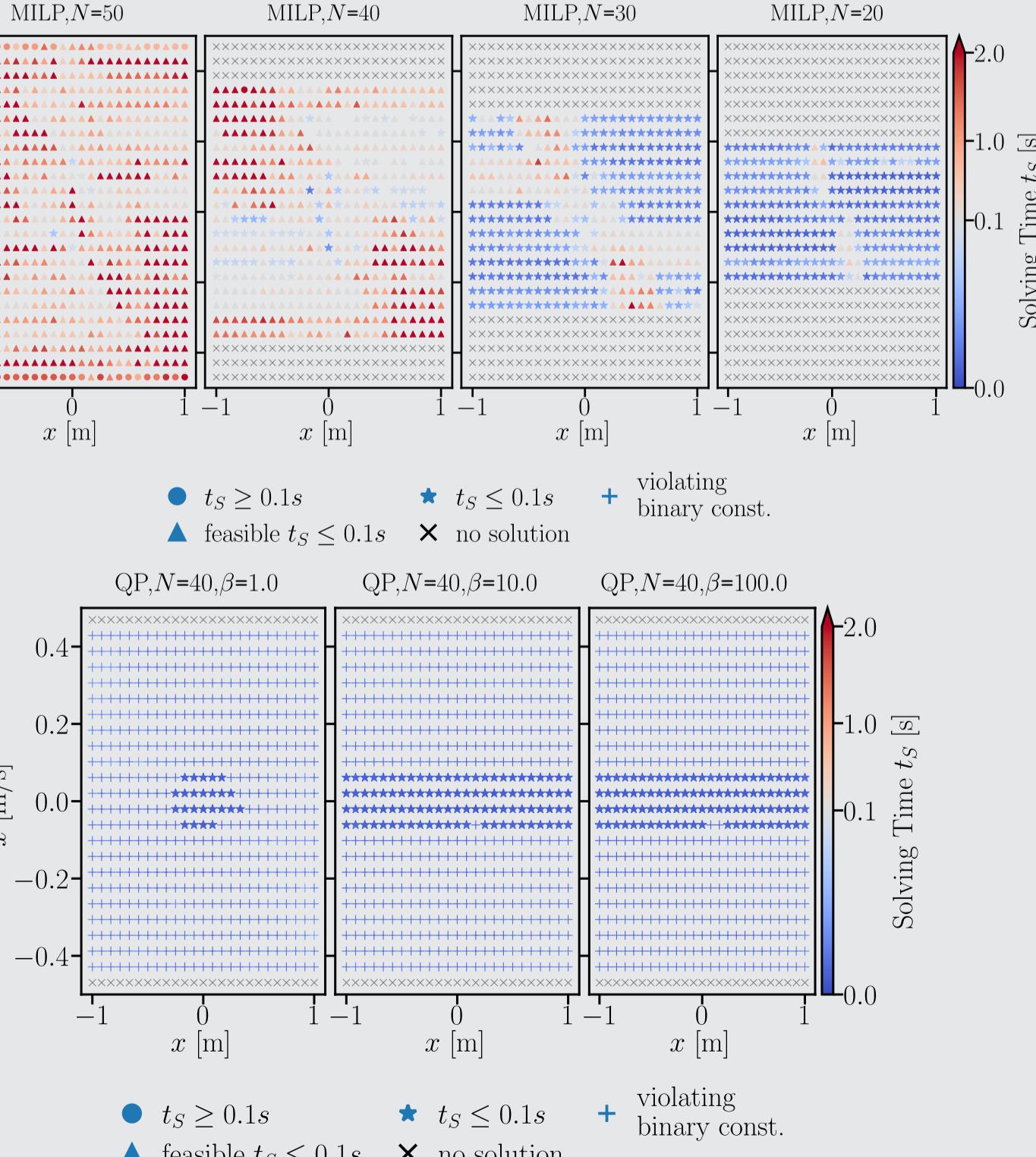
(Quadratic) Penalty-term:

- Penalty term not fully minimized
- Solutions have continuous values

Complementarity constraints:

- For this problem most of the time infeasible

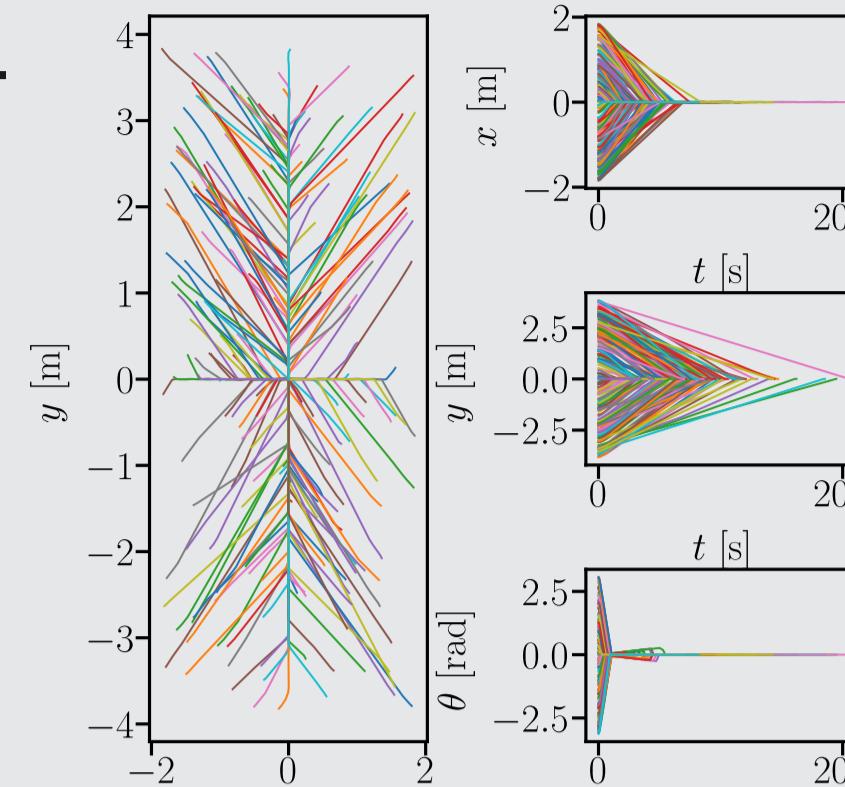
⇒ **Mixed Integer Linear MPC is used in this work**



## Simulation results

MILP was tested on 200 experiments with random initial states. Simulated using *drake toolbox*[2].

- Average solver time 56.08 ms with a standard deviation of 27.00 ms
- 27.45 % suboptimal solution
- In all experiments the system was steered towards and kept at the origin with an RMS error of 0.004 m and 0.097°



## Real world experiments on REACSA

Final MIMPC implemented in C++ using *SCIP Solver*[3], compared to existing TVLQR

Homing		MIMPC	TVLQR	1-meter		MIMPC	TVLQR	180 deg		MIMPC	TVLQR
To reach limit cycle				To reach limit cycle				To reach limit cycle			
Time		54.00 s	87.05 s	Time		27.00 s	47.02 s	Time		21.0 s	53.6 s
Thrust		7.10 s	9.10 s	Thrust		2.70 s	4.5 s	Thrust		1.90 s	4.3 s
RMS Error				RMS Error		0.011 m	0.031 m	RMS Error		0.008 m	0.027 m
In limit cycle				In limit cycle				In limit cycle			
RMS Error		0.0086 m, 0.0283 m, 0.584°	1.140°	RMS Error		0.015 m, 0.040 m, 0.739°	1.78°	RMS Error		0.013 m, 0.019 m, 0.221°	0.246°
Oscillation		0.0089 m, 0.030 m, 0.490°	0.837°	Oscillation		0.006 m, 0.015 m, 0.810°	0.76°	Oscillation		0.010 m, 0.02 m, 1.07°	1.67°
Thrust		0.083 s	0.104 s	Thrust		0.089 s	0.080 s	Thrust		0.050 s	0.060 s

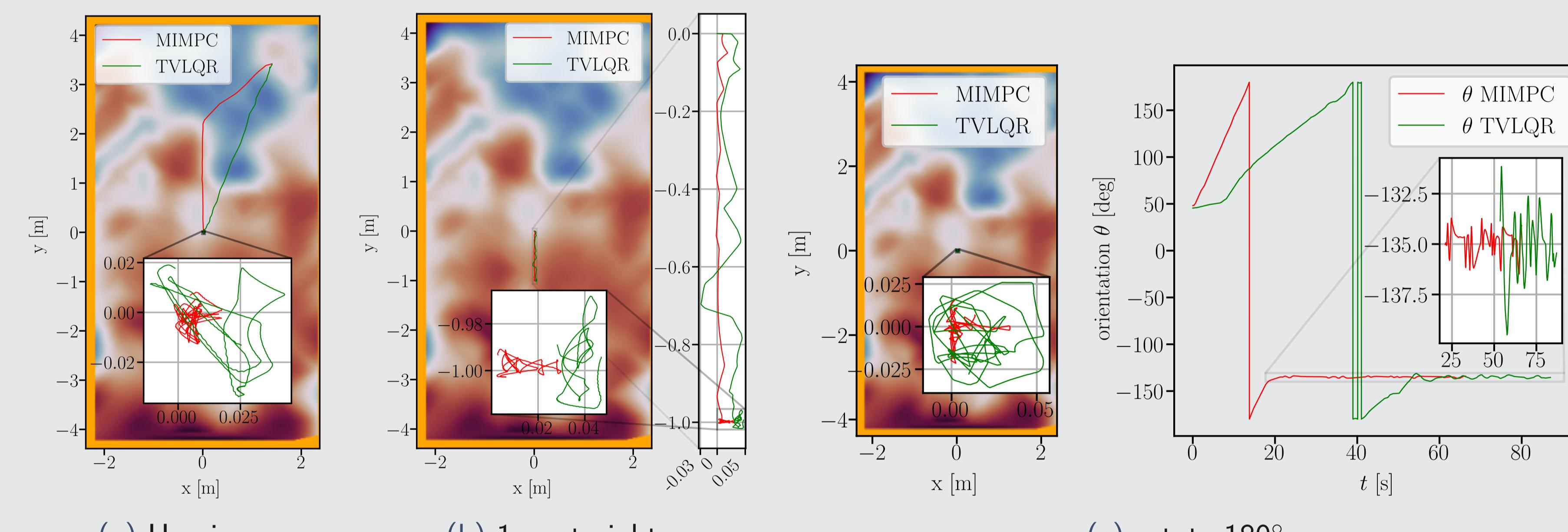


Figure: System trajectories on a height map of the not perfectly flat flat-floor

## Acknowledgment

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## References

- [1] Martin Zwick et al. "ORGL – ESA'S TEST FACILITY FOR APPROACH AND CONTACT OPERATIONS IN ORBITAL AND PLANETARY ENVIRONMENTS". en. In: *Proceedings of the International Symposium on Artificial Intelligence, Robotics and Automation in Space (i-SAIRAS)*. Vol. 6. Madrid, Spain, June 2018.
- [2] Russ Tedrake and Drake-Development-Team. *Drake: Model-based design and verification for robotics*. 2019. URL: <https://drake.mit.edu>.
- [3] Tobias Achterberg. "SCIP: solving constraint integer programs". en. In: *Mathematical Programming Computation* 1.1 (July 2009), pp. 1–41. ISSN: 1867-2957. DOI: 10.1007/s12532-008-0001-1. URL: <https://doi.org/10.1007/s12532-008-0001-1> (visited on 06/01/2023).