

Specifications for the Attitude Control and Determination System-XCubeSat

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Introduction

The objective of this document is to summarize in one place the specifications related to the software of the Attitude Control and Determination System, henceforth ADCS of the nanosatellite X-CubeSat developed by the students of cole Polytechnique, Paris. We have divided the document into the following sections

- Definition of the architecture of the ADCS
- External Interfaces available on the ADCS
- Sensors and peripherals on the ADCS
- Data formats used for computing
- Protocols for testing

This document shall be useful for all people currently concerned with the conception of the ADCS and eventually for students who will take over the project in the months to come. This document is however, far from being complete and additions, corrections and deletions will be made till the day of delivery.

Architecture of the ADCS

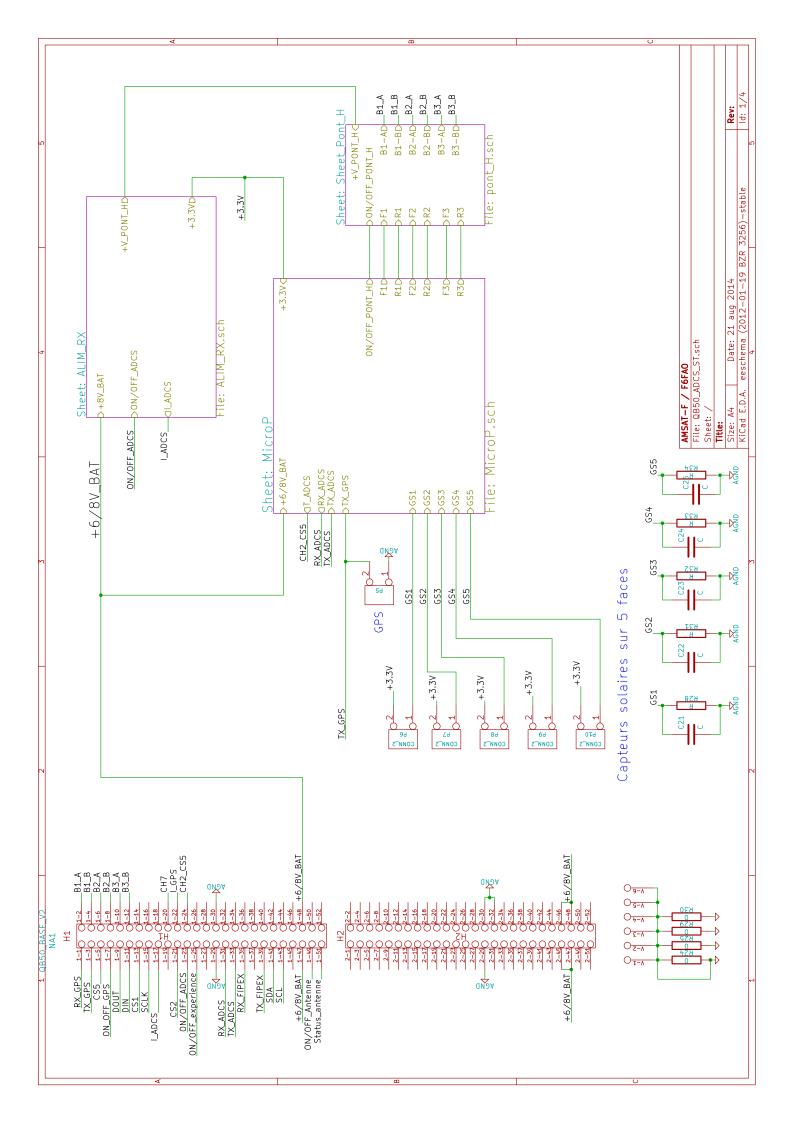
The ADCS uses a microprocessor manufactured by ST Microelectronics based on the ARM Cortex-M4 Architecture. The model used is STM 32F405 RGT6

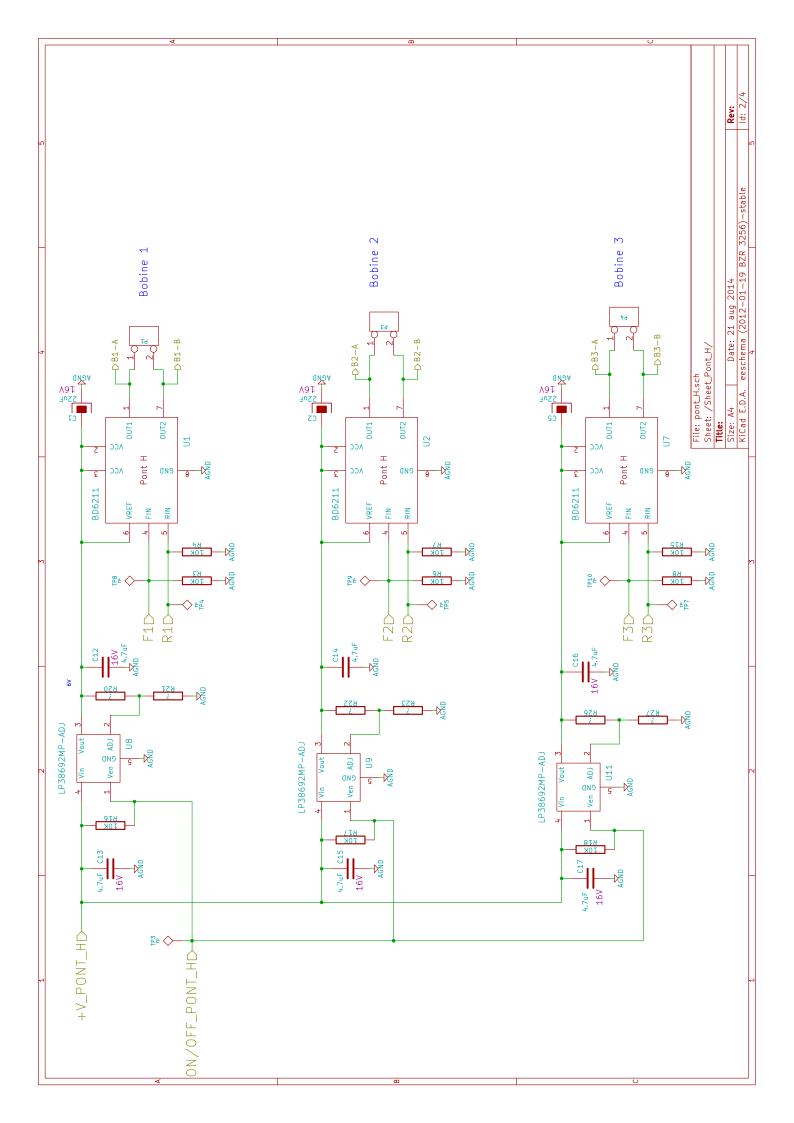
The available interfaces, sensors and other peripherals available on the ADCS are:

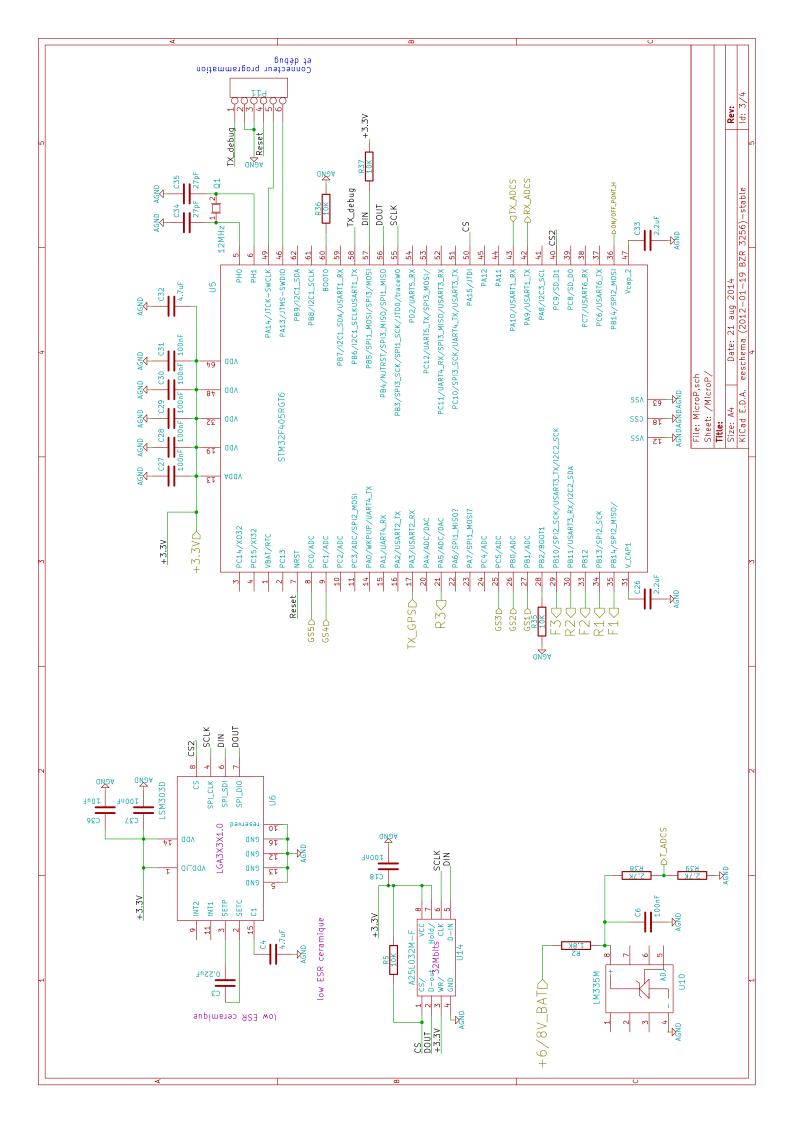
- 32 MB External Memory
- 7(or 9) sun sensors
- Magnetometer
- Gyrometer
- H-Bridges to modulate power to the magneto-torquers
- A Serial interface with the On-Board Computer (Ordinateur de Bord henceforth, ODB)

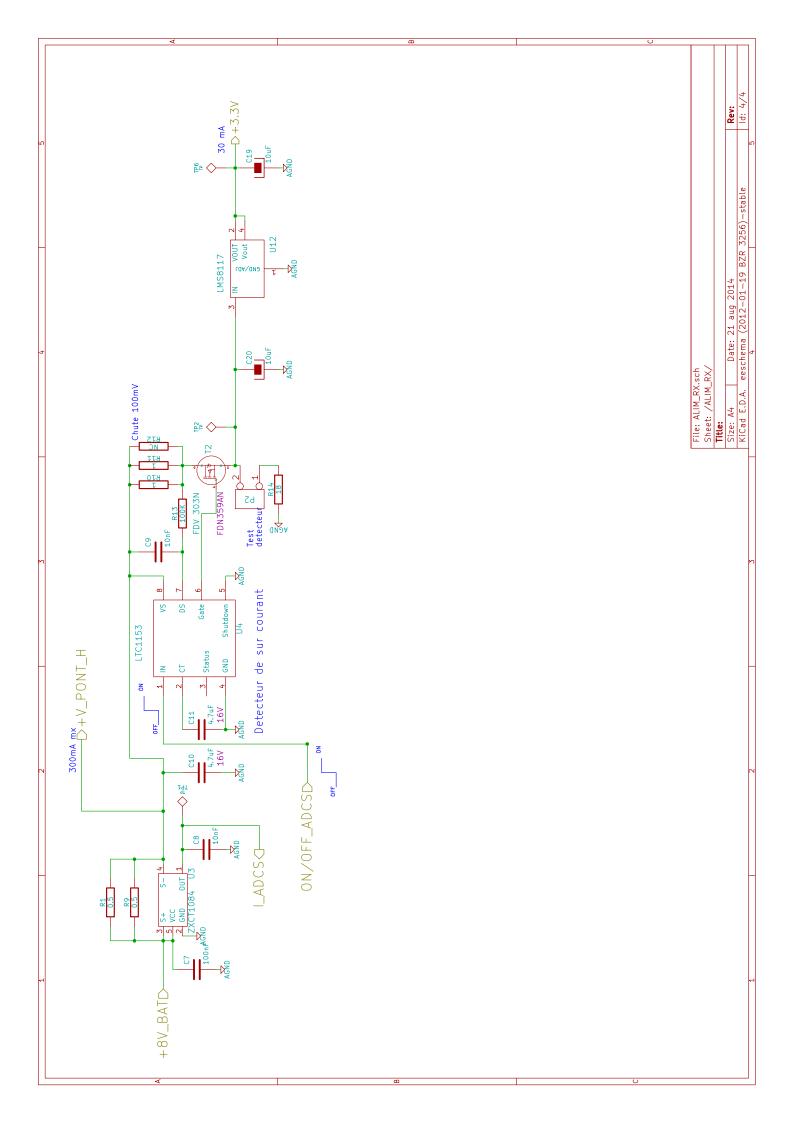
Following is the complete plan for the ADCS:











External interfaces



Sensors and peripherals



Data formats



Testing

