

# Engineering Notes

## Symplectic Approaches for Solving Two-Point Boundary-Value Problems

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### I. Introduction

THE two-point boundary-value problem (TPBVP) plays a fundamental role in optimal control problems of aerospace engineering, including the problem of spacecraft orbit transfer [1], the optimal reconfiguration of spacecraft formations [2], and continuous thrust rendezvous problems [3]. Therefore, many techniques and methods for solving TPBVP have been proposed and developed [4–7].

The optimal control problem is as follows. The dynamic system is

$$\dot{\mathbf{x}}(t) = f(\mathbf{x}(t), \mathbf{u}(t), t) \quad (1)$$

and the goal is to minimize the following cost function:

$$J = h(\mathbf{x}(t_f), t_f) + \int_0^{t_f} \Phi(\mathbf{x}(t), \mathbf{u}(t), t) dt \quad (2)$$

in which  $t$  denotes time, the dot represents the derivative with respect to time,  $\mathbf{x}$  is a  $d$ -dimensional state vector,  $\mathbf{u}$  is a  $p$ -dimensional control input vector, and  $J$  is a cost function. In this Note, we mainly focus on the optimal control problem with a fixed terminal time, i.e.,  $t_f$  is given.

Assume that the control input is unconstrained. By introducing the Lagrangian multiplier  $\lambda$  and the Hamiltonian function

$$\bar{H}(\mathbf{x}, \mathbf{u}, \lambda) = \Phi(\mathbf{x}(t), \mathbf{u}(t), t) + \lambda^T f(\mathbf{x}(t), \mathbf{u}(t), t)$$

the necessary condition for the optimal control input can be derived by the standard variational approach [8] as follows:

$$\frac{\partial \bar{H}(\mathbf{x}, \mathbf{u}, \lambda, t)}{\partial \mathbf{u}} = 0 \quad (3)$$

Without loss of generality, assume that the control input  $\mathbf{u}$  can be expressed by the state variable  $\mathbf{x}$  and the costate variable  $\lambda$ , i.e.,  $\mathbf{u}(t) = g(\mathbf{x}(t), \lambda(t))$ . Then, a new Hamiltonian function  $H(\mathbf{x}, \lambda)$ , which depends only on the state and costate variables, can be given as follows:

$$H(\mathbf{x}, \lambda) = \Phi(\mathbf{x}(t), g(\mathbf{x}, \lambda), t) + \lambda^T f(\mathbf{x}(t), g(\mathbf{x}, \lambda), t) \quad (4)$$

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The action  $\bar{S}$  in a time interval  $(0, \eta)$  is defined [9] as the following:

$$\bar{S} = \int_0^\eta (\lambda^T \dot{\mathbf{x}} - H) dt \quad (5)$$

Because the terminal time is fixed, the variation for the action  $\bar{S}$  yields the following:

$$\begin{aligned} \delta \bar{S} = & \int_0^\eta (\delta \mathbf{x})^T \left( -\dot{\lambda} - \frac{\partial H}{\partial \mathbf{x}} \right) dt + \int_0^\eta (\delta \lambda)^T \left( \dot{\mathbf{x}} - \frac{\partial H}{\partial \lambda} \right) dt \\ & + \lambda^T \delta \mathbf{x} \Big|_0^\eta = 0 \end{aligned} \quad (6)$$

First, the variational principle equation (6) can be used to derive the nonlinear TPBVP according to the initial and terminal state conditions. For the optimal control problem, the initial state is given. In this Note, two different types of terminal state conditions are considered, i.e., the free terminal state conditions or the fixed terminal state conditions. If the terminal state is free, then the boundary conditions  $\mathbf{x}(0) = \mathbf{x}_0$  and  $\lambda(t_f) = \partial h(\mathbf{x}(t_f), t_f) / \partial \mathbf{x}(t_f)$  can be given by the variational principle [8]. If the terminal state is fixed, then the boundary conditions  $\mathbf{x}(0) = \mathbf{x}_0$  and  $\mathbf{x}(t_f) = \mathbf{x}_f$  can be given by the variational principle [8]. The symbols  $\mathbf{x}_0$  and  $\mathbf{x}_f$  are the given states at the initial and terminal times, respectively. According to Eq. (6), the solutions of the optimal control problem should satisfy the following Hamiltonian canonical equation:

$$\dot{\mathbf{x}} = \frac{\partial H}{\partial \lambda} = f(\mathbf{x}, g(\mathbf{x}, \lambda), t), \quad \dot{\lambda} = -\frac{\partial H}{\partial \mathbf{x}} \quad (7)$$

So far, the nonlinear optimal control problem has been transferred into the nonlinear TPBVP of Hamiltonian systems with different boundary conditions.

The shooting and multiple shooting methods [4] can be employed to solve this TPBVP. For these methods, unknown initial states or costates must be guessed until the transversality conditions are satisfied. They can reach effective convergence but may cause ill-conditioning for a problem with a long time interval. A generating function method [3] is proposed to solve the TPBVP of Hamiltonian systems. The generating function method has an advantage over the conventional numerical shooting method in the sense that it does not require a guess of initial or terminal costates. However, complicated series expansions are needed for the generating function method.

The most fundamental property of Hamiltonian systems is that the phase flow is a symplectic transformation [9]. Numerical methods preserving the symplectic structure are more effective for solving Hamiltonian systems. For example, the symplectic method exhibits excellent energy behavior and accurately reflects the qualitative behavior of the solution [10]. Therefore, many symplectic numerical methods have been proposed and applied in solving initial-value problem of Hamiltonian systems.

The experiences for initial-value problem show that symplectic-preserving is important for designing numerical methods of Hamiltonian systems. Reference [11] proposed a symplectic method called discrete mechanics and optimal control (DMOC) for the solving optimal control problem, which shows that the application of the symplectic method to the optimal control problem has similar advantages as the initial-value problem. For example, the symplectic numerical method can preserve the energy better than the non-symplectic method, and it leads to a reasonable approximation to the continuous solution for a small number of discretization points [11,12].

From a mathematical point of view, the generating function method and the DMOC method are both symplectic. For the

generating function method, symbolic computation and solving of large-scale ordinary differential equations may lead to significant computational efforts. For the DMOC method, the symplectic-preserving idea is only applied to discrete the dynamic equations in the optimal control system and not to the whole optimal control system. In this Note, a systematic procedure is developed to construct three symplectic approaches for solving the Hamiltonian TPBVP derived from the optimal control problem. These symplectic approaches have been used to solve an optimal rendezvous problem with fixed time and without constraints.

## II. Actions Based on the Variational Principle

Since the symplectic structure plays a fundamental role in Hamiltonian systems, numerical methods that preserve the symplectic structure are desired. For a numerical approach, the continuous time domain is divided into a series of discrete time intervals, and the solutions can only be obtained on discrete time points. Therefore, assume that the relationship of a solution vector between two adjacent time points can be given by

$$\mathbf{v}_j = \Psi(\mathbf{v}_{j-1}) \quad (8)$$

in which  $\mathbf{v}_j = \{\mathbf{x}_j^T, \boldsymbol{\lambda}_j^T\}^T$  and  $\Psi$  is a mapping function in state space.

A numerical approach is symplectic if and only if the Jacobi matrix of the mapping function  $\Psi$  is a symplectic matrix [10]. The Jacobi matrix of the mapping function  $\Psi$  can be defined by

$$\mathbf{S} = \frac{\partial \mathbf{v}_j}{\partial \mathbf{v}_{j-1}} = \begin{bmatrix} \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} & \frac{\partial \mathbf{x}_j}{\partial \boldsymbol{\lambda}_{j-1}} \\ \frac{\partial \boldsymbol{\lambda}_j}{\partial \mathbf{x}_{j-1}} & \frac{\partial \boldsymbol{\lambda}_j}{\partial \boldsymbol{\lambda}_{j-1}} \end{bmatrix} \quad (9)$$

Therefore, if the matrix  $\mathbf{S}$  satisfies the following equation,

$$\mathbf{S}^T \mathbf{J} \mathbf{S} = \mathbf{J}, \quad \mathbf{J} = \begin{bmatrix} \mathbf{0} & \mathbf{I} \\ -\mathbf{I} & \mathbf{0} \end{bmatrix} \quad (10)$$

then the corresponding numerical approach is symplectic.

The variational principle given in Eq. (6) not only gives the Hamiltonian canonical equation but also can be used as the foundation for constructing a numerical approach. According to Eq. (6), if the Hamiltonian canonical equation is satisfied within the time interval  $(0, \eta)$ , then the first type of relationship between the action  $\bar{S}$  and the states and costates at the two ends can be given as follows:

$$d\bar{S} = \boldsymbol{\lambda}_\eta^T d\mathbf{x}_\eta - \boldsymbol{\lambda}_0^T d\mathbf{x}_0 \quad (11)$$

where  $\mathbf{x}_0$  and  $\boldsymbol{\lambda}_0$  are the state and costate variables at the initial time, respectively, and  $\mathbf{x}_\eta$  and  $\boldsymbol{\lambda}_\eta$  are the state and costate variables at time  $\eta$ , respectively. Equation (11) implies that if the Hamiltonian canonical equation is satisfied within the time interval, then the action  $\bar{S}$  is a function of only  $\mathbf{x}_0$  and  $\mathbf{x}_\eta$ .

The second type of relationship between the action and the state as well as the costate at two ends can also be given from Eq. (11). Because

$$d(\boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta) = \boldsymbol{\lambda}_\eta^T d\mathbf{x}_\eta + \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (12)$$

Eq. (11) can be written as

$$d(\boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta) - d\bar{S}(\mathbf{x}_0, \mathbf{x}_\eta) = \boldsymbol{\lambda}_0^T d\mathbf{x}_0 + \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (13)$$

If we define  $U$  by

$$U = \boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta - \bar{S}(\mathbf{x}_0, \mathbf{x}_\eta) \quad (14)$$

then Eq. (13) gives

$$dU = \boldsymbol{\lambda}_0^T d\mathbf{x}_0 + \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (15)$$

Equation (15) implies that, if the Hamiltonian canonical equation is satisfied within the time interval  $(0, \eta)$ , and the state  $\mathbf{x}_0$  at the left

end and the costate  $\boldsymbol{\lambda}_\eta$  at the right end of the time interval are taken as the independent variables, then  $U$  must be a function of only  $\mathbf{x}_0$  and  $\boldsymbol{\lambda}_\eta$ .

Similarly, the third type of relationship between the action and the state as well as the costate at two ends can be given as follows:

$$d(\boldsymbol{\lambda}_0^T \mathbf{x}_0 - \boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta) = \boldsymbol{\lambda}_0^T d\mathbf{x}_0 + \mathbf{x}_0^T d\boldsymbol{\lambda}_0 - \boldsymbol{\lambda}_\eta^T d\mathbf{x}_\eta - \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (16)$$

which allows Eq. (11) to be written as

$$d(\boldsymbol{\lambda}_0^T \mathbf{x}_0 - \boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta + \bar{S}(\mathbf{x}_0, \mathbf{x}_\eta)) = \mathbf{x}_0^T d\boldsymbol{\lambda}_0 - \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (17)$$

If we define  $V$  by

$$V = \boldsymbol{\lambda}_0^T \mathbf{x}_0 - \boldsymbol{\lambda}_\eta^T \mathbf{x}_\eta + \bar{S}(\mathbf{x}_0, \mathbf{x}_\eta) \quad (18)$$

then Eq. (17) gives

$$dV = \mathbf{x}_0^T d\boldsymbol{\lambda}_0 - \mathbf{x}_\eta^T d\boldsymbol{\lambda}_\eta \quad (19)$$

Equation (19) implies that, if the Hamiltonian canonical equation is satisfied within the time interval  $(0, \eta)$ , and the costate variables  $\boldsymbol{\lambda}_0$  and  $\boldsymbol{\lambda}_\eta$  at two ends of the time interval are taken as the independent variables, then  $V$  must be the function of only  $\boldsymbol{\lambda}_0$  and  $\boldsymbol{\lambda}_\eta$ .

In this section, based on the variational principle and by choosing different types of independent variables at the two ends of the time step, three different actions [Eqs. (11), (15), and (19)] are given. In the next section, three symplectic numerical approaches for solving nonlinear TPBVP are proposed based on the three different actions.

## III. Symplectic Approaches Based on the Actions

For numerical integration, the time domain  $(0, t_f)$  is divided into  $L$  time intervals with equal time steps  $\eta = t_f/L$ , i.e.,  $t_0 = 0, t_1 = \eta, \dots, t_j = j\eta, \dots$ , and  $t_f = L\eta$ . According to Eq. (11), the first symplectic approach can be given as follows. Assume that, within the  $j$ th subinterval, the state variable  $\mathbf{x}(t)$  is approximated by Lagrange polynomials of degree  $m-1$  that interpolate  $m$  equidistant points, and the costate variable  $\boldsymbol{\lambda}(t)$  is approximated by Lagrange polynomials of degree  $n-1$  that interpolate  $n$  equidistant points; that is,

$$\mathbf{x}(t) = (M_1 \otimes \mathbf{I})\mathbf{x}_{j-1} + (\bar{\mathbf{M}} \otimes \mathbf{I})\bar{\mathbf{x}}_j + (M_m \otimes \mathbf{I})\mathbf{x}_j \quad (20)$$

$$\boldsymbol{\lambda}(t) = (\mathbf{N} \otimes \mathbf{I})\bar{\boldsymbol{\lambda}}_j \quad (21)$$

in which  $\mathbf{I}$  denotes the identity matrix. Vectors  $\mathbf{x}_{j-1}$  and  $\mathbf{x}_j$  denote the state variables at the left and right ends of the  $j$ th subinterval, vector  $\bar{\mathbf{x}}_j$  is composed of the state variables at the internal interpolation points within the  $j$ th subinterval (i.e.,  $\bar{\mathbf{x}}_j = \{\bar{\mathbf{x}}_j^1, \bar{\mathbf{x}}_j^2, \dots, \bar{\mathbf{x}}_j^{m-1}\}^T$ ), and  $\bar{\boldsymbol{\lambda}}_j$  is composed of the costate variables at all of the interpolation points within the  $j$ th subinterval (i.e.,  $\bar{\boldsymbol{\lambda}}_j = \{\bar{\boldsymbol{\lambda}}_j^1, \bar{\boldsymbol{\lambda}}_j^2, \dots, \bar{\boldsymbol{\lambda}}_j^n\}^T$ ). Other symbols in Eqs. (20) and (21) are defined as follows:

$$\bar{\mathbf{M}} = [M_2, M_3, \dots, M_{m-1}] \quad (22)$$

$$\mathbf{N} = [N_1, N_2, \dots, N_n] \quad (23)$$

$$M_i = \prod_{j=1, j \neq i}^m \frac{t - (j-1)\eta/(m-1)}{(i-j)\eta/(m-1)} \quad (24)$$

$$N_i = \prod_{j=1, j \neq i}^n \frac{t - (j-1)\eta/(n-1)}{(i-j)\eta/(n-1)} \quad (25)$$

The symbol  $\otimes$  in Eqs. (20) and (21) denotes the Kronecker product of two matrices. Substituting the approximate state and costate variables [i.e., Eqs. (20) and (21)] into Eq. (5) gives the following:

$$\bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j) = \int_0^\eta (\lambda^T \dot{\mathbf{x}} - H(\mathbf{x}, \lambda))_j dt$$

$$j = 1, 2, \dots, L \quad (26)$$

According to Eq. (11), if the state variables at the two ends of the  $j$ th subinterval are taken as the independent variables, and if the Hamiltonian canonical equation is satisfied within the  $j$ th subinterval,  $\bar{S}$  must be the function of only the state variables at the two ends of the  $j$ th subinterval. Hence, for satisfying the Hamiltonian canonical equation, the approximate action  $\bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)$  in the  $j$ th subinterval must be the function of only the state variables at the two ends of the  $j$ th subinterval. Therefore, the following equations must be satisfied:

$$\frac{\partial \bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\mathbf{x}}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (27)$$

$$\frac{\partial \bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\lambda}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (28)$$

Moreover, according to Eq. (11), we have

$$\frac{\partial \bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \mathbf{x}_{j-1}} + \lambda_{j-1} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (29)$$

$$\frac{\partial \bar{S}_j(\mathbf{x}_{j-1}, \mathbf{x}_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \mathbf{x}_j} - \lambda_j = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (30)$$

So far, based on Eq. (11), the nonlinear TPBVP equation (7) is transformed into a set of nonlinear equations [Eqs. (27–30)]. The nonlinear equations are derived by using the variational principle and the symplectic property; this proof is in the Appendix.

The second symplectic approach can be constructed according to Eq. (15). The state variable  $\mathbf{x}(t)$  and the costate variable  $\lambda(t)$  within the  $j$ th subinterval are approximated by using the Lagrange polynomials with orders  $m-1$  and  $n-1$ , respectively, as follows:

$$\mathbf{x}(t) = (M_1 \otimes \mathbf{I})\mathbf{x}_{j-1} + (\tilde{\mathbf{M}} \otimes \mathbf{I})\bar{\mathbf{x}}_j \quad (31)$$

$$\lambda(t) = (\tilde{\mathbf{N}} \otimes \mathbf{I})\bar{\lambda}_j + (N_n \otimes \mathbf{I})\lambda_j \quad (32)$$

in which vectors  $\mathbf{x}_{j-1}$  and  $\lambda_j$  denote the state variable at the left end and the costate variable at the right end of the  $j$ th subinterval, vector  $\bar{\mathbf{x}}_j$  is defined by  $\bar{\mathbf{x}}_j = \{\bar{\mathbf{x}}_j^1, \bar{\mathbf{x}}_j^2, \dots, \bar{\mathbf{x}}_j^m\}^T$ , and  $\bar{\lambda}_j$  is defined by  $\bar{\lambda}_j = \{\bar{\lambda}_j^1, \bar{\lambda}_j^2, \dots, \bar{\lambda}_j^{n-1}\}^T$ . Other symbols in Eqs. (31) and (32) are defined as follows:

$$\tilde{\mathbf{M}} = [M_2, M_3, \dots, M_m] \quad (33)$$

$$\tilde{\mathbf{N}} = [N_1, N_2, \dots, N_{n-1}] \quad (34)$$

The elements in Eqs. (33) and (34) are the same as those defined in Eqs. (24) and (25). Substituting approximate state variables (31) and costate variables (32) into Eq. (14) gives

$$U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j) = \lambda_j^T \bar{\mathbf{x}}_j - \int_0^\eta (\lambda^T \dot{\mathbf{x}} - H(\mathbf{x}, \lambda))_j dt$$

$$j = 1, 2, \dots, L \quad (35)$$

According to Eq. (15), the approximate action  $U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)$  in the  $j$ th subinterval must be the function of only the state variable at the left end and the costate variable at the right end of the  $j$ th subinterval. Therefore, the following equations must be satisfied:

$$\frac{\partial U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\mathbf{x}}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (36)$$

$$\frac{\partial U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\lambda}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (37)$$

In addition, according to Eq. (15), we have

$$\frac{\partial U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \mathbf{x}_{j-1}} - \lambda_{j-1} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (38)$$

$$\frac{\partial U_j(\mathbf{x}_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \lambda_j} - \mathbf{x}_j = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (39)$$

Based on Eq. (15), the nonlinear TPBVP equation (7) is transformed into a set of nonlinear equations [Eqs. (36–39)]. It is easy to prove that the approach defined by Eqs. (36–39) is symplectic. The proof is similar to the proof given in the Appendix for the first symplectic approach.

For the third symplectic approach, the approach can be constructed according to Eq. (19). Assume that the state variable  $\mathbf{x}(t)$  and the costate variable  $\lambda(t)$  within the  $j$ th subinterval are approximated by using the Lagrange polynomials with orders  $m-1$  and  $n-1$ , respectively, as follows:

$$\mathbf{x}(t) = (\mathbf{M} \otimes \mathbf{I})\bar{\mathbf{x}}_j \quad (40)$$

$$\lambda(t) = (N_1 \otimes \mathbf{I})\lambda_{j-1} + (\tilde{\mathbf{N}} \otimes \mathbf{I})\bar{\lambda}_j + (N_n \otimes \mathbf{I})\lambda_j \quad (41)$$

in which vectors  $\lambda_{j-1}$  and  $\lambda_j$  denote the costate variables at the left end and the right end of the  $j$ th subinterval, vector  $\bar{\mathbf{x}}_j$  is defined by  $\bar{\mathbf{x}}_j = \{\bar{\mathbf{x}}_j^1, \bar{\mathbf{x}}_j^2, \dots, \bar{\mathbf{x}}_j^m\}^T$ , and  $\bar{\lambda}_j$  is defined by  $\bar{\lambda}_j = \{\bar{\lambda}_j^1, \bar{\lambda}_j^2, \dots, \bar{\lambda}_j^{n-1}\}^T$ . Other symbols in Eqs. (40) and (41) are defined as follows:

$$\mathbf{M} = [M_1, M_2, \dots, M_m] \quad (42)$$

$$\tilde{\mathbf{N}} = [N_2, N_3, \dots, N_{n-1}] \quad (43)$$

The elements in Eqs. (42) and (43) are the same as those defined in Eqs. (24) and (25). Substituting approximate state variables (40) and costate variables (41) into Eq. (18) gives the following:

$$V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j) = \lambda_{j-1}^T \bar{\mathbf{x}}_j^1 - \lambda_j^T \bar{\mathbf{x}}_j^m$$

$$+ \int_0^\eta (\lambda^T \dot{\mathbf{x}} - H(\mathbf{x}, \lambda))_j dt, \quad j = 1, 2, \dots, L \quad (44)$$

According to Eq. (19), the approximate action  $V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)$  in the  $j$ th subinterval must be the function of only the costate variables at the two ends of the  $j$ th subinterval. Therefore, the following equations must be satisfied:

$$\frac{\partial V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\mathbf{x}}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (45)$$

$$\frac{\partial V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \bar{\lambda}_j} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (46)$$

Then, according to Eq. (19), we have

$$\frac{\partial V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \lambda_j} + \mathbf{x}_j = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (47)$$

$$\frac{\partial V_j(\lambda_{j-1}, \lambda_j, \bar{\mathbf{x}}_j, \bar{\lambda}_j)}{\partial \lambda_{j-1}} - \mathbf{x}_{j-1} = \mathbf{0}, \quad j = 1, 2, \dots, L \quad (48)$$

Based on Eq. (19), the nonlinear TPBVP equation (7) is transformed into a set of nonlinear equations [Eqs. (45–48)], which can also be proved to be symplectic.

#### IV. Application: Optimal Rendezvous Problem

Consider a spacecraft subject to a central gravity field. To establish the motion equations, we introduce a coordinate frame that is rotating along a circular orbit at a constant angular velocity. The non-dimensional dynamical equation can be given by [3]

$$\ddot{x} - 2\dot{y} + (1+x)\left(\frac{1}{r^3} - 1\right) = u_x \quad (49)$$

$$\ddot{y} + 2\dot{x} + y\left(\frac{1}{r^3} - 1\right) = u_y \quad (50)$$

$$\ddot{z} + \frac{1}{r^3} z = u_z \quad (51)$$

where  $r = \sqrt{(x+1)^2 + y^2 + z^2}$ . Symbols  $x$ ,  $y$ , and  $z$  are the position variables of the spacecraft, and they represent radial, tangential, and normal displacements from the origin of the rotating frame, respectively, and  $u_x$ ,  $u_y$ , and  $u_z$  are the control accelerations in the  $x$ ,  $y$ , and  $z$  directions of the spacecraft, respectively.

With the nondimensional equations of motion, the goal of the nonlinear optimal control is to minimize the following quadratic cost function; that is,

$$J = \frac{1}{2} \int_0^1 (u_x^2 + u_y^2 + u_z^2) dt \quad (52)$$

In the state space, the state vector is defined as  $\mathbf{x} = \{x, y, z, \dot{x}, \dot{y}, \dot{z}\}^T$ , and the goal is to transfer the spacecraft from an initial state  $\mathbf{x}(0) = \{0.2, 0.2, 0.2, 0.1, 0.1, 0.1\}^T$  to a final state  $\mathbf{x}(1) = \{0, 0, 0, 0, 0, 0\}^T$  in a fixed flying time  $t_f = 1$  while minimizing the cost function in Eq. (52).

A number of numerical tests shows that the combinations of  $m = n + 1$ ,  $m = n$ , and  $m = n - 1$  for the first, second, and third symplectic approaches are optimal. Therefore, the problem given above is solved by using the three symplectic approaches with  $m = 2$ ,  $n = 1$ ;  $m = n = 2$ ; and  $m = 1$ ,  $n = 2$ , respectively. Moreover, two different discretion schemes are used; that is, the whole time domain is divided into 8 or 64 time intervals. Figures 1–3 show the state, the costate, and the control history, respectively.

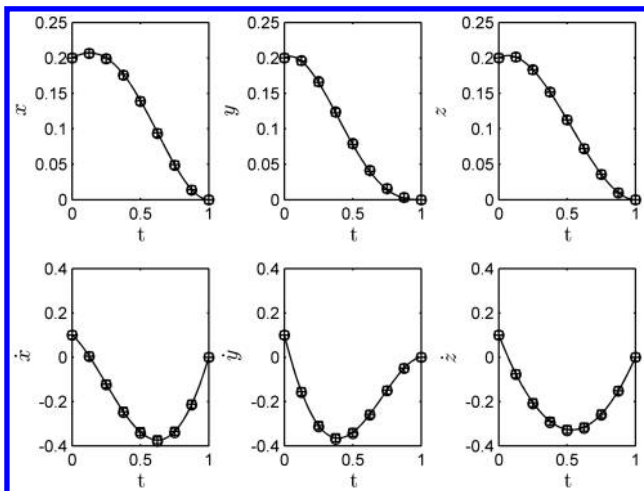


Fig. 1 State trajectory in normalized time.

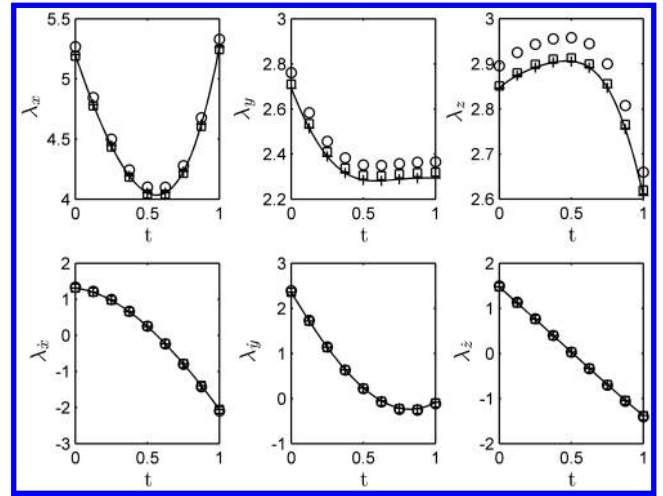


Fig. 2 Costate trajectory in normalized time.

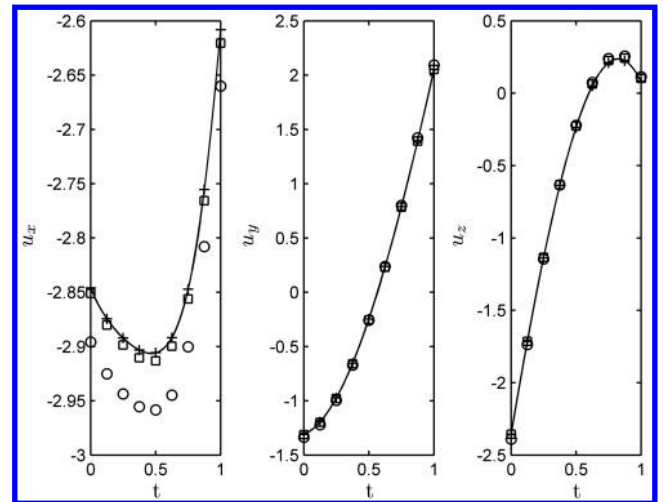


Fig. 3 Control input in normalized time.

When the number of the time intervals is eight, the results from the three different symplectic approaches are given by a circle, a plus, and a square in Figs. 1–3, while when the number of the time intervals is 64, the results from the three different symplectic approaches are given by the solid, dashed, and dotted lines in Figs. 1–3. Figures 1–3 show that the three symplectic approaches give different numerical results when the number of the time intervals is small, while all three approaches give almost the same results (mixed with each other) when the number of the time intervals is larger. Moreover, Fig. 1 shows that both the initial and terminal conditions are satisfied very well.

Table 1 Cost functions for different time intervals

Number of intervals	First symplectic approach	Second symplectic approach	Third symplectic approach
8	10.399283	10.037657	10.081721
16	10.158212	10.069116	10.080302
32	10.099259	10.077067	10.079874
64	10.084601	10.079058	10.079760
128	10.080941	10.079556	10.079731
256	10.080027	10.079680	10.079724
512	10.079798	10.079712	10.079722
1024	10.079741	10.079719	10.079722
2048	10.079727	10.079721	10.079722
4096	10.079723	10.079722	10.079722
8192	10.079722	10.079722	10.079722

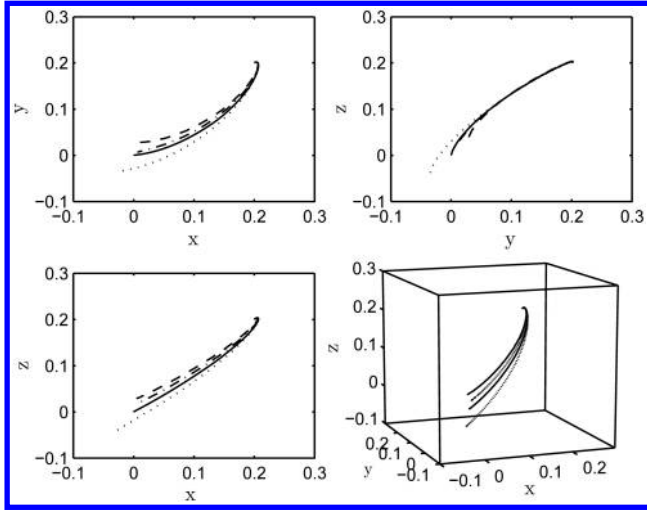


Fig. 4 Position trajectories.

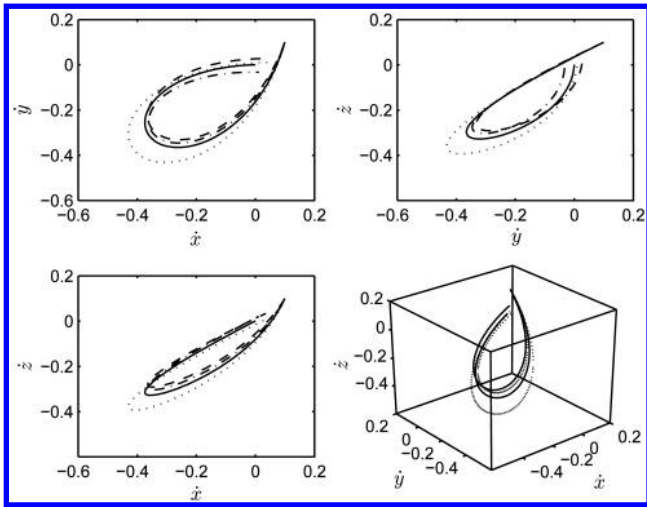


Fig. 5 Velocity trajectories.

To see more differences of the above three symplectic approaches, Table 1 shows the cost function changing with changes in the number of time intervals. When the number of time intervals is increased, all three symplectic approaches tend to convergence and converge to the same value.

For this example, some numerical comparisons between the proposed symplectic methods and the generating function method [3] are given in Figs. 4 and 5. The solid lines are the results given by the first symplectic method with 64 time intervals, and the dashed, dotted, and dashed-dotted lines are the results given by the first-, second-, and third-order generating function methods, respectively. Figures 4 and 5 show that the terminal values of positions and trajectories obtained by the proposed method satisfy the terminal boundary conditions. For the generating function method, terminal values of positions and trajectories with the increasing of the number of Taylor series expansions terms tend to satisfy the terminal boundary conditions. Although, terminal boundary conditions cannot be satisfied accurately by the generating function method, the generating function method can give a feedback control law, and so it is a closed-loop method.

## V. Conclusions

Three different symplectic numerical approaches are proposed based on the variational principle to solve TPBVP in optimal control. The three symplectic approaches are constructed by choosing different types of independent variables at the two ends of a time step. For each approach, the nonlinear TPBVP is transformed into a set of

nonlinear algebraic equations that can preserve the symplectic structure of the original Hamiltonian system. The proposed symplectic approaches are successfully applied to solve optimal orbital rendezvous problems in a central gravity field. The numerical results show that the three symplectic approaches give different numerical performances; however, with an increase in the number of time intervals, all three symplectic approaches give the same convergence results.

## Appendix: Symplectic Property of the First Approach

It is easy to verify that Eq. (10) is equivalent to the following three equations:

$$\left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \boldsymbol{\lambda}_j}{\partial \mathbf{x}_{j-1}} = \left[ \left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \boldsymbol{\lambda}_j}{\partial \mathbf{x}_{j-1}} \right]^T \quad (\text{A1})$$

$$\left( \frac{\partial \mathbf{x}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \frac{\partial \boldsymbol{\lambda}_j}{\partial \boldsymbol{\lambda}_{j-1}} = \left[ \left( \frac{\partial \mathbf{x}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \frac{\partial \boldsymbol{\lambda}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right]^T \quad (\text{A2})$$

$$\left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \boldsymbol{\lambda}_j}{\partial \boldsymbol{\lambda}_{j-1}} - \left( \frac{\partial \boldsymbol{\lambda}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \mathbf{x}_j}{\partial \boldsymbol{\lambda}_{j-1}} = \mathbf{I} \quad (\text{A3})$$

Next, we will prove that the numerical approach defined by Eqs. (27–30) satisfies Eqs. (A1–A3).

*Step 1:* Within an arbitrary interval  $j$ , assume that the variables  $\bar{\mathbf{x}}_j$ ,  $\bar{\boldsymbol{\lambda}}_j$ ,  $\mathbf{x}_j$ , and  $\boldsymbol{\lambda}_j$  are all functions of the variables  $\mathbf{x}_{j-1}$  and  $\boldsymbol{\lambda}_{j-1}$ . Then, based on the chain rule of derivatives, taking a derivative of Eqs. (27–30) with respect to variable  $\mathbf{x}_{j-1}$  gives the following:

$$\boldsymbol{\Omega}_1 \mathbf{X} + \boldsymbol{\Omega}_2^T = \left\{ \mathbf{0} \quad \mathbf{0} \quad \left( \frac{\partial \boldsymbol{\lambda}_j}{\partial \mathbf{x}_{j-1}} \right)^T \right\}^T \quad (\text{A4})$$

$$\boldsymbol{\Omega}_2 \mathbf{X} = - \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_{j-1} \partial \mathbf{x}_{j-1}} \quad (\text{A5})$$

in which

$$\boldsymbol{\Omega}_1 = \begin{bmatrix} \frac{\partial^2 \bar{S}_j}{\partial \bar{\mathbf{x}}_j \partial \bar{\mathbf{x}}_j} & \frac{\partial^2 \bar{S}_j}{\partial \bar{\mathbf{x}}_j \partial \boldsymbol{\lambda}_j} & \frac{\partial^2 \bar{S}_j}{\partial \bar{\mathbf{x}}_j \partial \mathbf{x}_j} \\ \frac{\partial^2 \bar{S}_j}{\partial \boldsymbol{\lambda}_j \partial \bar{\mathbf{x}}_j} & \frac{\partial^2 \bar{S}_j}{\partial \boldsymbol{\lambda}_j \partial \boldsymbol{\lambda}_j} & \frac{\partial^2 \bar{S}_j}{\partial \boldsymbol{\lambda}_j \partial \mathbf{x}_j} \\ \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_j \partial \bar{\mathbf{x}}_j} & \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_j \partial \boldsymbol{\lambda}_j} & \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_j \partial \mathbf{x}_j} \end{bmatrix} \quad (\text{A6})$$

$$\boldsymbol{\Omega}_2 = \left\{ \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_{j-1} \partial \bar{\mathbf{x}}_j} \quad \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_{j-1} \partial \boldsymbol{\lambda}_j} \quad \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_{j-1} \partial \mathbf{x}_j} \right\} \quad (\text{A7})$$

$$\mathbf{X} = \left\{ \left( \frac{\partial \bar{\mathbf{x}}_j}{\partial \mathbf{x}_{j-1}} \right)^T \quad \left( \frac{\partial \bar{\boldsymbol{\lambda}}_j}{\partial \mathbf{x}_{j-1}} \right)^T \quad \left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \right\}^T \quad (\text{A8})$$

Similarly, taking the derivative of Eqs. (27–30) with respect to variable  $\boldsymbol{\lambda}_{j-1}$  gives

$$\boldsymbol{\Omega}_1 \mathbf{Y} = \left\{ \mathbf{0} \quad \mathbf{0} \quad \left( \frac{\partial \boldsymbol{\lambda}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \right\}^T \quad (\text{A9})$$

$$\boldsymbol{\Omega}_2 \mathbf{Y} = -\mathbf{I} \quad (\text{A10})$$

$$\mathbf{Y} = \left\{ \left( \frac{\partial \bar{\mathbf{x}}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \quad \left( \frac{\partial \bar{\boldsymbol{\lambda}}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \quad \left( \frac{\partial \mathbf{x}_j}{\partial \boldsymbol{\lambda}_{j-1}} \right)^T \right\}^T \quad (\text{A11})$$

Left multiplying on both sides of Eq. (A4) by the transpose of  $\mathbf{X}$  gives

$$\mathbf{X}^T \boldsymbol{\Omega}_1 \mathbf{X} + \mathbf{X}^T \boldsymbol{\Omega}_2^T = \left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \lambda_j}{\partial \mathbf{x}_{j-1}} \quad (\text{A12})$$

Substituting Eq. (A5) into Eq. (A12) gives

$$\mathbf{X}^T \boldsymbol{\Omega}_1 \mathbf{X} - \frac{\partial^2 \bar{S}_j}{\partial \mathbf{x}_{j-1} \partial \mathbf{x}_{j-1}} = \left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \lambda_j}{\partial \mathbf{x}_{j-1}} \quad (\text{A13})$$

The matrices  $\boldsymbol{\Omega}_1$  and  $\partial^2 \bar{S}_j / \partial \mathbf{x}_{j-1} \partial \mathbf{x}_{j-1}$  are both symmetric matrices, so the matrix  $(\partial \lambda_j / \partial \mathbf{x}_{j-1})^T (\partial \mathbf{x}_j / \partial \mathbf{x}_{j-1})$  in Eq. (A13) is also a symmetric matrix, which means that the approach satisfies Eq. (A1).

*Step 2:* Left multiplying on both sides of Eq. (A9) by the transpose of  $\mathbf{Y}$  gives

$$\mathbf{Y}^T \boldsymbol{\Omega}_1 \mathbf{Y} = \left( \frac{\partial \mathbf{x}_j}{\partial \lambda_{j-1}} \right)^T \frac{\partial \lambda_j}{\partial \lambda_{j-1}} \quad (\text{A14})$$

The matrix  $\boldsymbol{\Omega}_1$  is a symmetric matrix, so the matrix  $(\partial \mathbf{x}_j / \partial \lambda_{j-1})^T (\partial \lambda_j / \partial \lambda_{j-1})$  is also a symmetric matrix, which means that the approach satisfies Eq. (A2).

*Step 3:* Left multiplying on both sides of Eq. (A9) by the transpose of  $\mathbf{X}$  gives

$$\mathbf{X}^T \boldsymbol{\Omega}_1 \mathbf{Y} = \left( \frac{\partial \mathbf{x}_j}{\partial \lambda_{j-1}} \right)^T \frac{\partial \lambda_j}{\partial \lambda_{j-1}} \quad (\text{A15})$$

Transposing both sides of Eq. (A4) and then multiplying  $\mathbf{Y}$  gives

$$\mathbf{X}^T \boldsymbol{\Omega}_1 \mathbf{Y} + \boldsymbol{\Omega}_2 \mathbf{Y} = \left( \frac{\partial \lambda_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \mathbf{x}_j}{\partial \lambda_{j-1}} \quad (\text{A16})$$

Substituting Eq. (A10) into Eq. (A16) and then subtracting Eq. (A16) from Eq. (A15) gives

$$\left( \frac{\partial \mathbf{x}_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \lambda_j}{\partial \lambda_{j-1}} - \left( \frac{\partial \lambda_j}{\partial \mathbf{x}_{j-1}} \right)^T \frac{\partial \mathbf{x}_j}{\partial \lambda_{j-1}} = \mathbf{I} \quad (\text{A17})$$

Equation (A17) shows that the approach satisfies Eq. (A3).

We proved that the numerical approach defined by Eqs. (27–30) satisfies Eqs. (A1–A3), so it is a symplectic approach.

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