



Robotic Perception

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Content



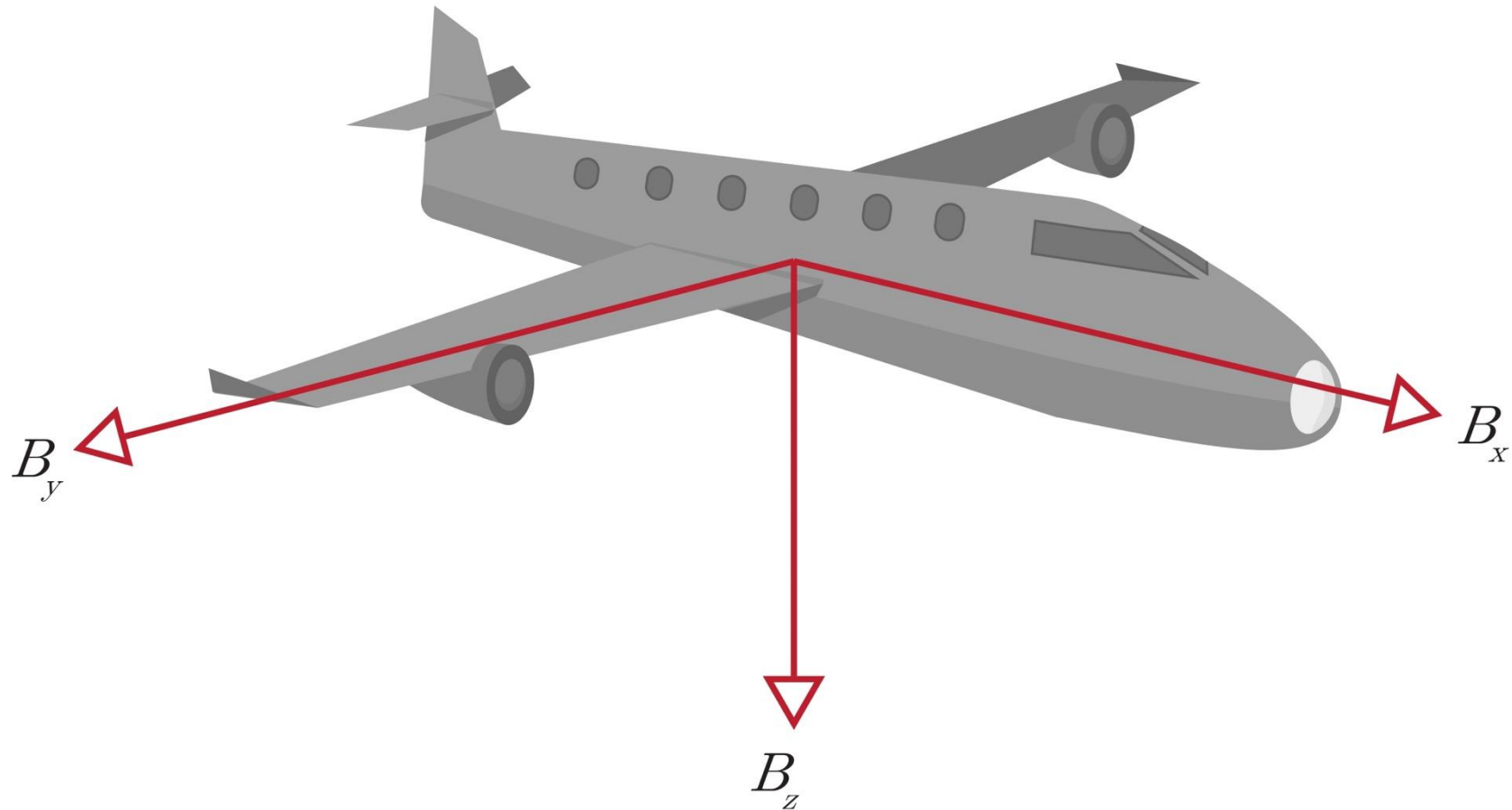
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Position and Orientation
Reference Frames
Attitude Representations
Transformations

Body Frame



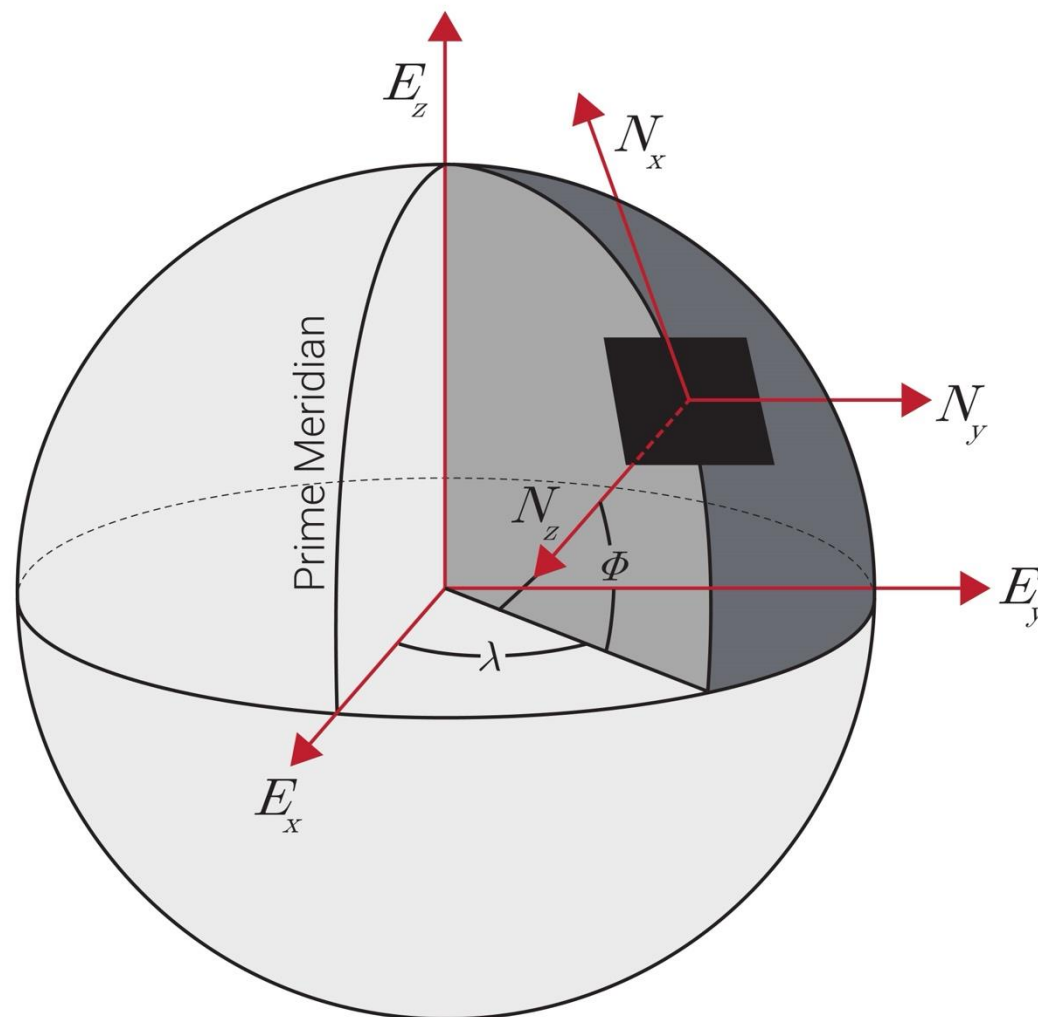
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ECEF and NED



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Position, Orientation



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Position

x, y, z

Velocity

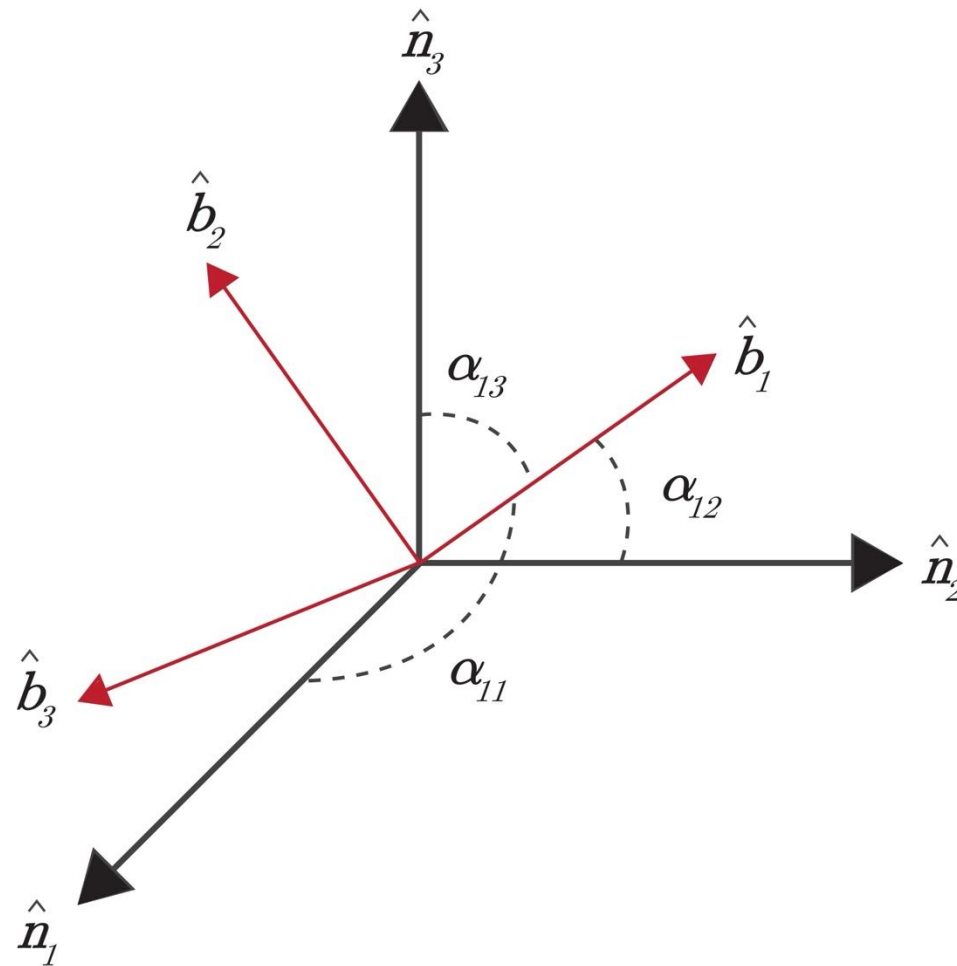
v_x, v_y, v_z

Orientations?

DCM



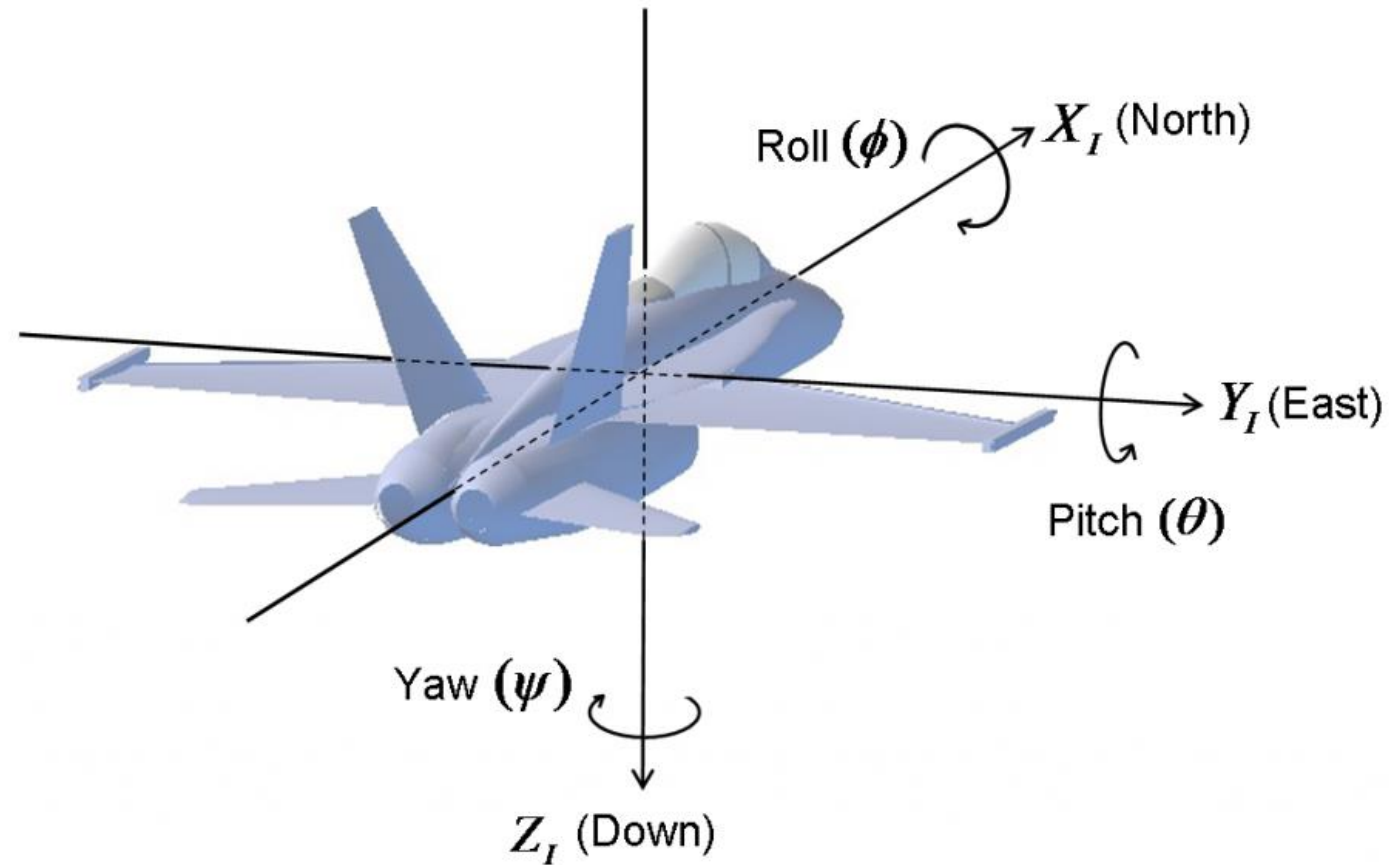
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Euler Angles (Yaw, Pitch, Roll)



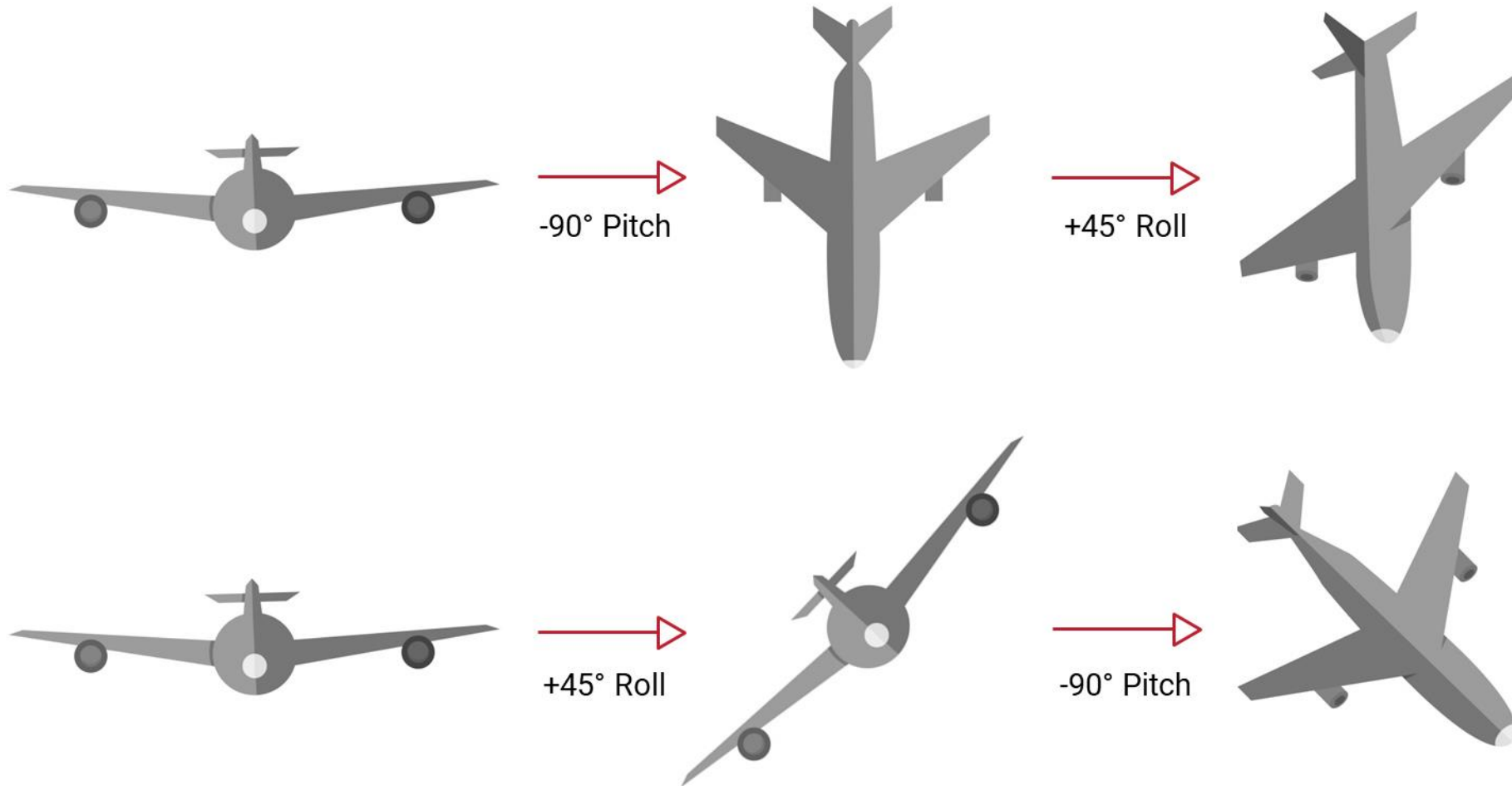
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Order of Angles



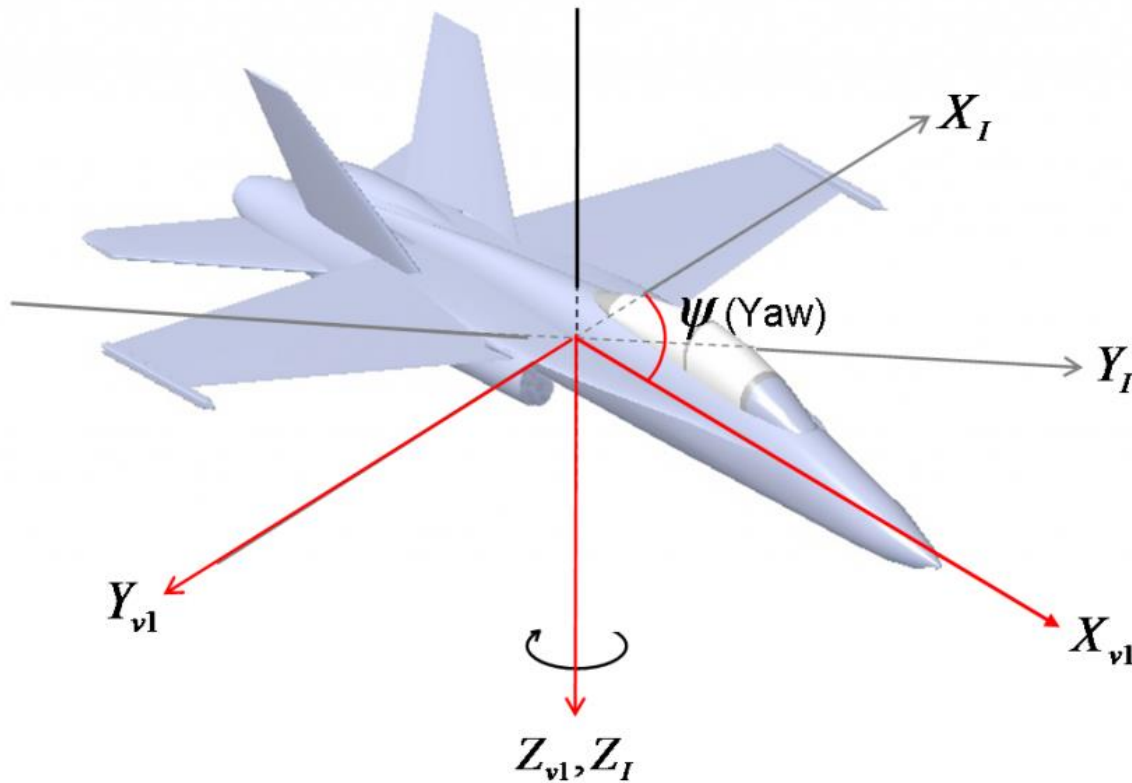
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Yaw



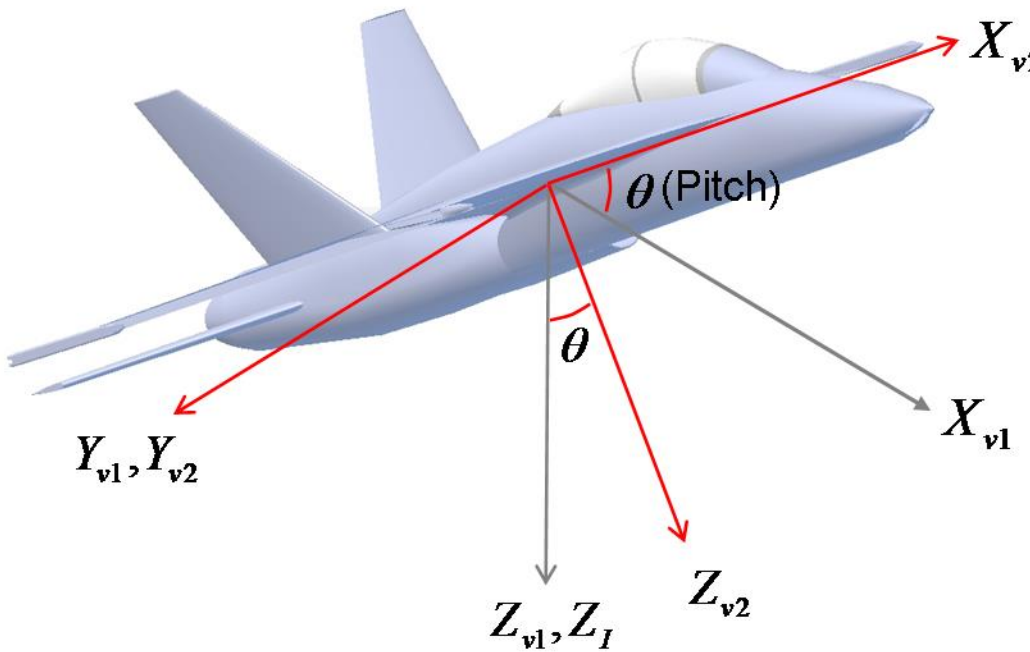
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$$R_I^{v1}(\psi) = \begin{pmatrix} \cos(\psi) & \sin(\psi) & 0 \\ -\sin(\psi) & \cos(\psi) & 0 \\ 0 & 0 & 1 \end{pmatrix}.$$

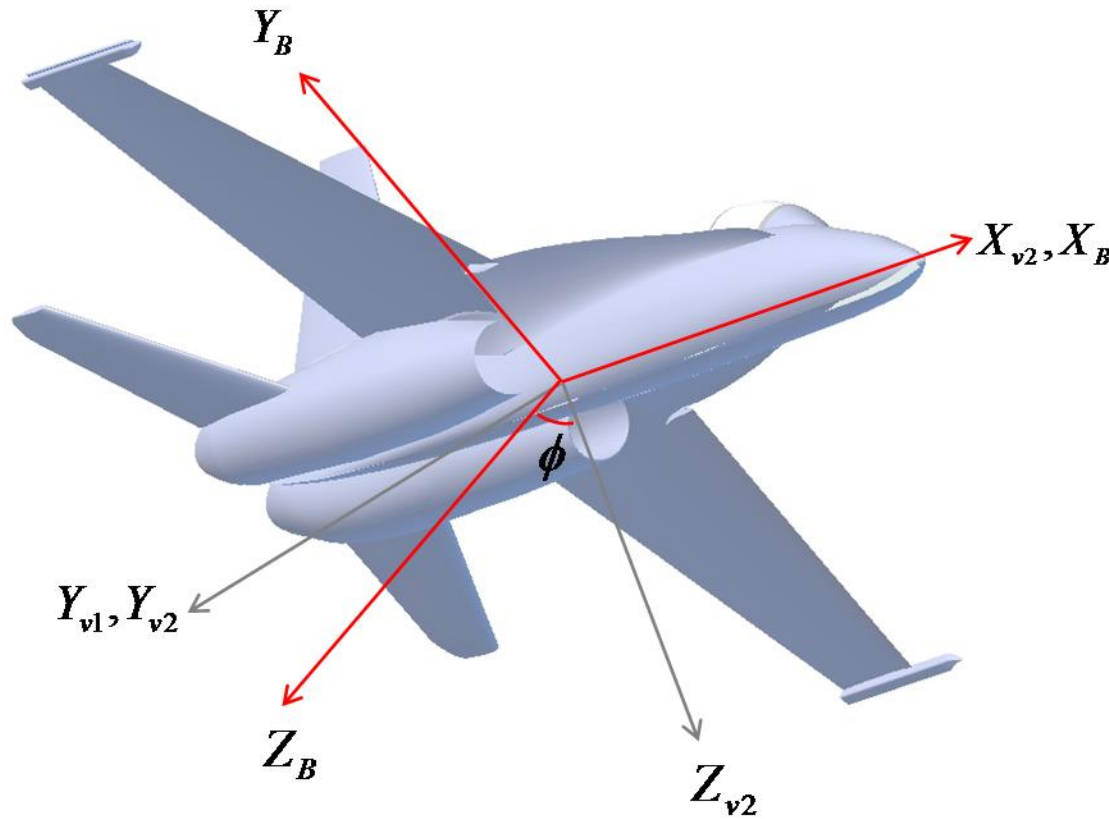
Pitch

$$R_{V1}^{V2}(\theta) = \begin{pmatrix} \cos(\theta) & 0 & -\sin(\theta) \\ 0 & 1 & 0 \\ \sin(\theta) & 0 & \cos(\theta) \end{pmatrix}.$$



$$R_I^{V2}(\theta, \psi) = R_{V1}^{V2}(\theta) R_I^{V1}(\psi).$$

Body Frame



$$R_{V2}^B(\phi) = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos(\phi) & \sin(\phi) \\ 0 & -\sin(\phi) & \cos(\phi) \end{pmatrix}.$$

$$R_I^B(\phi, \theta, \psi) = R_{V2}^B(\phi) R_{V1}^{V2}(\theta) R_I^{V1}(\psi).$$

$$R_I^B(\phi, \theta, \psi) = \begin{pmatrix} c(\psi)c(\theta) & c(\theta)s(\psi) & -s(\theta) \\ c(\psi)s(\phi)s(\theta) - c(\phi)s(\psi) & c(\phi)c(\psi) + s(\phi)s(\psi)s(\theta) & c(\theta)s(\phi) \\ s(\phi)s(\psi) + c(\phi)c(\psi)s(\theta) & c(\phi)s(\psi)s(\theta) - c(\psi)s(\phi) & c(\phi)c(\theta) \end{pmatrix}.$$

Body to Inertial

$$R_B^I(\phi, \theta, \psi) = R_{I1}^{V1}(-\psi) R_{V1}^{V2}(-\theta) R_{V2}^B(-\phi).$$

$$R_I^B(\phi, \theta, \psi) = \begin{pmatrix} c(\psi)c(\theta) & c(\psi)s(\phi)s(\theta) - c(\phi)s(\psi) & s(\phi)s(\psi) + c(\phi)c(\psi)s(\theta) \\ c(\theta)s(\psi) & c(\phi)c(\psi) + s(\phi)s(\psi)s(\theta) & c(\phi)s(\psi)s(\theta) - c(\psi)s(\phi) \\ -s(\theta) & c(\theta)s(\phi) & c(\phi)c(\theta) \end{pmatrix}.$$

Quaternions

$$\mathbf{q}_1 \mathbf{q}_2 = \begin{pmatrix} a_1 a_2 - b_1 b_2 - c_1 c_2 - d_1 d_2 \\ a_1 b_2 + b_1 a_2 + c_1 d_2 - d_1 c_2 \\ a_1 c_2 - b_1 d_2 + c_1 a_2 + d_1 b_2 \\ a_1 d_2 + b_1 c_2 - c_1 b_2 + d_1 a_2 \end{pmatrix}.$$

$$\mathbf{q}_i^b = (a \quad b \quad c \quad d)^T.$$

b.c.d are the vector about which rotations are performed and “a” specified the amount of rotation

If \theta is the rotation about (vx,vy,vz)

$$\begin{pmatrix} a \\ b \\ c \\ d \end{pmatrix} = \begin{pmatrix} \cos(0.5\theta) \\ v_x \sin(0.5\theta) \\ v_y \sin(0.5\theta) \\ v_z \sin(0.5\theta) \end{pmatrix}.$$

$$\mathbf{v}_B = \mathbf{q}_i^b \begin{pmatrix} 0 \\ \mathbf{v}_I \end{pmatrix} (\mathbf{q}_i^b)^{-1}.$$

Body to Inertial

$$R_i^b(\mathbf{q}_i^b) = \begin{pmatrix} a^2 + b^2 - c^2 - d^2 & 2bc - 2ad & 2bd + 2ac \\ 2bc + 2ad & a^2 - b^2 + c^2 - d^2 & 2cd - 2ab \\ 2bd - 2ac & 2cd + 2ab & a^2 - b^2 - c^2 + d^2 \end{pmatrix}.$$

$$\mathbf{v}_B = R_i^b(\mathbf{q}_i^b)\mathbf{v}_I.$$

$$\phi = \arctan\left(\frac{2(ab+cd)}{a^2-b^2-c^2+d^2}\right),$$

$$\theta = -\arcsin(2(bd - ac)), \text{ and}$$

$$\psi = \arctan\left(\frac{2(ad+bc)}{a^2+b^2-c^2-d^2}\right).$$

Gimbal Lock – Body Frame to Inertial



Pitch 90 degrees

$$\mathbf{v}_I = R_B^I(\phi, \theta, \psi) \mathbf{v}_B.$$

$$D(\phi, \theta, \psi) = \begin{pmatrix} 1 & \sin(\phi) \tan(\theta) & \cos(\phi) \tan(\theta) \\ 0 & \cos(\phi) & -\sin(\phi) \\ 0 & \sin(\phi)/\cos(\theta) & \cos(\phi)/\cos(\theta) \end{pmatrix}.$$

$$\begin{pmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{pmatrix} = \begin{pmatrix} p + q \sin(\phi) \tan(\theta) + r \cos(\phi) \tan(\theta) \\ q \cos(\phi) - r \sin(\phi) \\ q \sin(\phi)/\cos(\theta) + r \cos(\phi)/\cos(\theta) \end{pmatrix}.$$