

# ADCS Homework 2

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## Problem 2

A matrix is given by

$$A = \begin{bmatrix} 0.8138 & 0.4698 & -0.3420 \\ -0.5630 & 0.4915 & -0.6645 \\ -0.1441 & 0.7333 & 0.6645 \end{bmatrix}$$

$$A = \begin{matrix} 3 \times 3 \\ \begin{matrix} 0.8138 & 0.4698 & -0.3420 \\ -0.5630 & 0.4915 & -0.6645 \\ -0.1441 & 0.7333 & 0.6645 \end{matrix} \end{matrix}$$

### Part A

Show that this matrix satisfies the six conditions for the a DCM (to numerical accuracy).

#### 1. Unit Vector Columns

$$u1 = A(1,:)$$

$$u1 = \begin{matrix} 1 \times 3 \\ \begin{matrix} 0.8138 & 0.4698 & -0.3420 \end{matrix} \end{matrix}$$

$$u2 = A(2,:)$$

$$u2 = \begin{matrix} 1 \times 3 \\ \begin{matrix} -0.5630 & 0.4915 & -0.6645 \end{matrix} \end{matrix}$$

$$u3 = A(3,:)$$

$$u3 = \begin{matrix} 1 \times 3 \\ \begin{matrix} -0.1441 & 0.7333 & 0.6645 \end{matrix} \end{matrix}$$

To show they are orthogonal, take the dot product between each unit vector

$$u1u2 = \text{dot}(u1,u2)$$

$$u1u2 = -3.7000e-06$$

$$u1u3 = \text{dot}(u1,u3)$$

$$u1u3 = -2.3240e-05$$

$$u2u3 = \text{dot}(u2,u3)$$

$$u2u3 = -1.5000e-05$$

$$u1u1 = \text{dot}(u1,u1)$$

$$u1u1 = 0.9999$$

$$u2u2 = \text{dot}(u2,u2)$$

$$u2u2 = 1.0001$$

```
u3u3 = dot(u3,u3)
```

```
u3u3 = 1.0001
```

2. Inversion is the same as the transpose

```
Ainv = inv(A)
```

```
Ainv = 3x3
```

```
0.8138 -0.5629 -0.1441
0.4698 0.4915 0.7333
-0.3420 -0.6644 0.6644
```

```
Atranspose = transpose(A)
```

```
Atranspose = 3x3
```

```
0.8138 -0.5630 -0.1441
0.4698 0.4915 0.7333
-0.3420 -0.6645 0.6645
```

```
abs(Ainv - Atranspose) < 5e-4 * ones(3,3)
```

```
ans = 3x3 logical array
```

```
1 1 1
1 1 1
1 1 1
```

Both of these matrices are equal

3. The determinant is +1

```
detA = det(A)
```

```
detA = 1.0001
```

4. The only real eigenvalue is +1

```
[V,D] = eig(A)
```

```
V = 3x3 complex
```

```
-0.7991 + 0.0000i 0.0644 - 0.4202i 0.0644 + 0.4202i
0.1131 + 0.0000i 0.7026 + 0.0000i 0.7026 + 0.0000i
0.5905 + 0.0000i -0.0476 - 0.5687i -0.0476 + 0.5687i
```

```
D = 3x3 complex
```

```
1.0000 + 0.0000i 0.0000 + 0.0000i 0.0000 + 0.0000i
0.0000 + 0.0000i 0.4849 + 0.8746i 0.0000 + 0.0000i
0.0000 + 0.0000i 0.0000 + 0.0000i 0.4849 - 0.8746i
```

Where the only real eigenvalue is

```
D(1,1)
```

ans = 1.0000

## Part B

1. If this represents a 3-2-1 sequence, what are the euler angles, yaw, pitch and roll?

Following the equations from lecture for a 3-2-1 rotation, we get the following

$$\theta = -\sin(A(1,3)) \cdot 180/\pi$$

$$\theta = 19.9988$$

$$\psi = \tan(A(1,2)/A(1,1)) \cdot 180/\pi$$

$$\psi = 29.9975$$

$$\phi = \tan(A(2,3)/A(3,3)) \cdot 180/\pi$$

$$\phi = -45$$

So, the yaw ( $\psi$ ) was approximately 30 degrees, the pitch ( $\theta$ ) was approximately 20 degrees, and the roll ( $\phi$ ) was approximately -45 degrees.

2. If this represents a 3-1-3 DCM for Keplerian orbit elements, what are  $\Omega$ ,  $i$ , and  $f$ ?

Using the equations I derived in problem 1, the values are

$$\Omega = -\tan(A(3,1)/A(3,2)) \cdot 180/\pi$$

$$\Omega = 11.1175$$

$$i = \cos(A(3,3)) \cdot 180/\pi$$

$$i = 48.3560$$

$$f = \tan(A(1,3)/A(2,3)) \cdot 180/\pi$$

$$f = 27.2336$$

3. What is the equivalent Quaternion for this DCM?

The formulas I will use are the following:

$$\mathbf{e} = [e_1, e_2, e_3]^T$$

Where  $\mathbf{e}$  is the Euler axis of Rotation, the eigenvector that corresponds to  $\lambda = 1$ .

$$\alpha = \cos^{-1} \left( \frac{\text{tr}[A] - 1}{2} \right)$$

$$q_i = e_i \sin\left(\frac{\alpha}{2}\right), \quad i = 1, 2, 3$$

$$q_4 = \cos\left(\frac{\alpha}{2}\right)$$

$$\mathbf{q} = [q_1, q_2, q_3]^T$$

$$[\mathbf{Q}] = \begin{pmatrix} 0 & -q_3 & q_2 \\ q_3 & 0 & -q_1 \\ -q_2 & q_1 & 0 \end{pmatrix}$$

$$[\mathbf{A}(\mathbf{q})] = (q_4^2 - \mathbf{q}^T \mathbf{q}) \mathbf{1} + 2\mathbf{q}\mathbf{q}^T - 2q_4[\mathbf{Q}]$$

```
e = V(:,1)
```

```
e = 3x1
    -0.7991
     0.1131
     0.5905
```

```
e1 = e(1,1)
```

```
e1 = -0.7991
```

```
e2 = e(2,1)
```

```
e2 = 0.1131
```

```
e3 = e(3,1)
```

```
e3 = 0.5905
```

```
alphaAngle = acos((trace(A) -1)/2)
```

```
alphaAngle = 1.0645
```

```
q1 = e1 * sin(alphaAngle/2)
```

```
q1 = -0.4055
```

```
q2 = e2 * sin(alphaAngle/2)
```

```
q2 = 0.0574
```

```
q3 = e3 * sin(alphaAngle/2)
```

```
q3 = 0.2997
```

```
q4 = cos(alphaAngle/2)
```

```
q4 = 0.8617
```

```
qVec = [q1 q2 q3]'
```

```
qVec = 3×1
    -0.4055
     0.0574
     0.2997
```

```
Qmat = [0 -q3 q2; q3 0 -q1; -q2 q1 0]
```

```
Qmat = 3×3
     0    -0.2997    0.0574
    0.2997     0    0.4055
   -0.0574   -0.4055     0
```

```
A_q = ((q4)^2 - dot(qVec, qVec))*eye(3) + 2*dot(qVec,qVec) - 2*q4*Qmat
```

```
A_q = 3×3
    1.0000    1.0315    0.4162
   -0.0013    1.0000   -0.1838
    0.6140    1.2140    1.0000
```