

Homework Assignment 2

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I PROBLEM 1

A.

We can relate the energy of the system with the semi-major axis a using the following formula (since energy is conserved for the orbit):

$$E = \frac{1}{2}v^2 - \frac{1}{r} = -\frac{1}{2a}$$

Hence, we can find a with \mathbf{v} and \mathbf{r} as follows:

$$a = \frac{2}{r} - v^2 \quad (1)$$

Using the given vectors \mathbf{v} and \mathbf{r} , we find that v^2 and r are the following:

$$v^2 = 0^2 + \left(\frac{1}{10}\right)^2 = \frac{1}{100} \quad (2)$$

$$r = \sqrt{10^2 + 0^2} = 10 \quad (3)$$

Therefore, we get the following value for the semi-major axis a :

$$\therefore a = \frac{19}{100} \quad (4)$$

To find \mathcal{P} , we simply find the angular momentum \mathbf{L} and use the relation $\mathcal{P} = L^2$ to find \mathcal{P} (since angular momentum is conserved, we can use the initial conditions).

$$\mathbf{L} = \mathbf{r} \times \mathbf{v} \quad (5)$$

$$= (r_1 v_2 - v_1 r_2) \hat{\mathbf{z}} \quad (6)$$

$$= \left(10 \cdot \frac{1}{10} - 0 \cdot 0\right) \hat{\mathbf{z}} \quad (7)$$

$$\therefore \mathbf{L} = \hat{\mathbf{z}} \quad (8)$$

\mathcal{P} is then:

$$\therefore \mathcal{P} = 1 \quad (9)$$

We can then use $\mathcal{P} = L^2$ to find the eccentricity (e) of the orbit as follows:

$$e = \sqrt{1 - \frac{L^2}{GMa}} \quad (10)$$

$$= \quad (11)$$

Finally, we can find the period (in reduced units) using the following a -dependent equation:

$$T^2 = 4\pi^2 a^3 \quad (12)$$

$$\Rightarrow T = 2\pi a^{3/2} \quad (13)$$

$$\therefore T = 2\pi * \left(\frac{19}{100}\right)^{3/2} \quad (14)$$

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