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A GENERALIZED IMU MODELING FRAMEWORK FOR VARYING INERTIAL SENSING **TECHNOLOGIES AND** PERFORMANCE GRADES

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Joint Navigation Conference 2023, San Diego California

15 June 2023





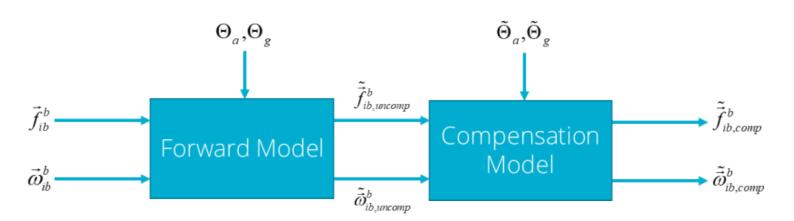
OVERVIEW

- The IMU Modeling Problem
- Basic IMU Modeling
- IEEE Standards for Inertial Sensing
- The Need for a Generalized IMU Modeling Framework
- Generalized Inertial Sensing Model Applications
- Future IMU Model Extensions



THE IMU MODELING PROBLEM

- The term "IMU Model" often has different meanings depending on the context, including:
 - Simulation of an IMU
 - IMU Calibration and Compensation
 - IMU Error Properties for Kalman Filtering Applications



IMU Modeling Questions

- What model should I use?
- What parameters belong in my model?

Forward Model

$$ilde{ec{f}}_{ib,uncomp}^{b} = f_a \Big(ec{f}_{ib}^{\,b}; \; oldsymbol{\Theta}_a \Big)$$

$$ilde{\vec{o}}^b_{ib,\mathit{uncomp}} = f_g \left(\vec{o}^b_{ib}; \; \Theta_g \right)$$

Compensation Model

$$ilde{\vec{f}}_{ib,comp}^b = g_a \Big(\vec{f}_{ib,uncomp}^b; \, \tilde{\Theta}_a \Big) \qquad \Delta \vec{f}_{ib}^b = \tilde{\vec{f}}_{ib,comp}^b - \vec{f}_{ib}^b$$

$$ilde{\vec{\omega}}^b_{ib,comp} = extbf{g}_g \left(ec{\omega}^b_{ib,\mathit{loncomp}}; \; ilde{\Theta}_g
ight)$$

Error Model

$$\Delta \vec{f}_{ib}^{\ b} = \tilde{\vec{f}}_{ib,comp}^{\ b} - \vec{f}_{ib}^{\ b}$$

$$\Delta ec{ec{lpha}}_{ib}^b = oldsymbol{ ilde{\hat{lpha}}}_{ib,comp}^b - ec{lpha}_{ib}^b$$



BASIC IMU MODELING

- A basic IMU Model typical consists of:
 - Fixed Bias
 - Scale Factor Error
 - Triad Misalignment
- These basic models are often sufficient for consumergrade MEMS devices

Basic IMU Modeling Questions

- Is this model still sufficient for highperformance IMU's?
- Is this model still applicable for all inertial sensing technologies?
- Where do we look for more detailed models?

Basic Accelerometer Triad Model

$$f_{ib,comp}^b = g_a \left(f_{ib,uncomp}^b; \Theta_a \right) = \left(I_3 + M_a \right)^{-1} \left(f_{ib,uncomp}^b - b_{a,FB} \right)$$

$$\Theta_a \in \overset{\smile}{b_{a,FB}}, M_a$$

Basic Gyroscope Triad Model

$$\overrightarrow{\phi_{ib,comp}} = g_g \left(\overrightarrow{\phi_{ib,uncomp}}; \, \Theta_g \right) = (I_3 + M_g)^{-1} \left(\overrightarrow{\phi_{ib,uncomp}} - \overrightarrow{b_{g,FB}} \right)$$

$$\Theta_g \in \overset{\smile}{b_{g,FB}}, M_g$$



IEEE STANDARDS FOR INERTIAL SENSING

The IEEE Gyroscope and Accelerometer (GAP) Panel maintains an international standard of specifying and testing inertial sensing components.

Accelerometer Components

 IEEE Std 1293-2018 – Linear Single Axis Nongyroscopic Accelerometers

Gyroscope Components

- IEEE Std 292-1969 Single DOF Spring-Restrained Gyroscopes
- IEEE Std 517-1974 Single DOF Rate-Integrating Gyroscopes
- IEEE Std 813-1988 Two DOF Dynamically Tuned Gyroscopes
- IEEE Std 1431-2004 Coriolis Vibratory Gyroscopes
- IEEE Std 647-2006 Single Axis Laser Gyroscope
- IEEE Std 952-2020 Single Axis Interferometric Fiber Optic Gyro

Inertial Measurement Units

• IEEE Std 1780-2022 – Specifying Inertial Measurement Units

Accelerometer Component Model from IEEE Std 1293-2018

$$\begin{split} E &= K_{1}\{K_{0} + \frac{K_{0'}}{2}\mathrm{sign}(\mathbf{a}_{i}) + \left(1 + \frac{K_{1'}}{2}\mathrm{sign}(a_{i})\right)a_{i} + K_{oq}a_{i}\left|a_{i}\right| + K_{2}a_{i}^{2} + K_{3}a_{i}^{3} + \sum_{n\geq4}K_{n}a_{i}^{4} \\ &+ \delta_{o}a_{p} - \delta_{p}a_{o} + K_{ip}a_{i}a_{p} + K_{io}a_{i}a_{o} + K_{po}a_{p}a_{p} + K_{pp}a_{p}^{2} + K_{oo}a_{o}^{2} \\ &+ K_{\mathrm{spin}}\omega_{i}\omega_{p} + K_{\mathrm{ang,accel}}\omega_{o} + \varepsilon\} \end{split}$$

Gyroscope Component Model from IEEE Std 952-2020

$$S_0 \left(\Delta N / \Delta t \right) = \frac{I + E + D}{1 + 10^{-6} \varepsilon_K}$$

$$I = \omega_{IRA} \sqrt{1 - \left[\left(\frac{\sin(\alpha)}{\alpha} \right) \alpha_{x} \right]^{2} - \left[\left(\frac{\sin(\alpha)}{\alpha} \right) \alpha_{y} \right]^{2}} + \omega_{XRA} \left[\left(\frac{\sin(\alpha)}{\alpha} \right) \alpha_{y} \right] - \omega_{XRA} \left[\left(\frac{\sin(\alpha)}{\alpha} \right) \alpha_{x} \right]$$

$$E = D_{\overline{t}} \Delta T + D_{\overline{t}} \left(\frac{\mathrm{d}T}{\mathrm{d}t} \right) + \overline{D}_{\nabla \overline{t}} \times \frac{\mathrm{d}\nabla \overline{T}}{\mathrm{d}t}$$

$$D = D_F + D_R + D_O$$



THE NEED FOR A GENERALIZED FRAMEWORK

A review of all IEEE Standards published by the GAP provide the following common inertial sensing component errors:

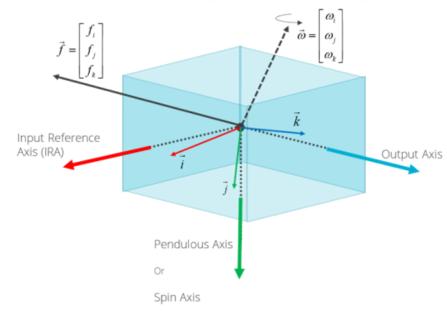
- Bias, Scale Factor, and Higher Order Terms
- Asymmetries of Bias, Scale Factor, and Higher Order Terms
- Component-Level Misalignment
- Cross Coupling and Cross Axis Nonlinearities
- Sensitivity to the Opposite Phenomenon

A new generalized component model should provide:

- Consistent interpretations and usage of the above common error sources
- Expandable and adaptable terms for higher order models
- Express the model in a sum-of-products form for simple implementation

GENERALIZED INERTIAL SENSING COMPONENT MODELS

<u>Inertial Sensing Component Diagram</u>



- This inertial sensing diagram contains:
 - The internal inertial sensing axis (e.g., i, j, k)
 - The external reference cases
 - The specific forces and angular rates applied to the center of percussion

A detailed explanation of these equations is available in the back-up slides!

$$f_{IRA} = f_a\left(f, \omega, \widetilde{\omega}\right) = \sum_{n=0}^{N_g} b_{a,n} f_i^n + \sum_{n'=0}^{N_g} b_{a,n'} \operatorname{sign}(\phi_i) b_{a'}^{n'} + m_a^T f + f^T C_a f + \omega^T C_g \omega + f^T S_g f$$

$$\boldsymbol{\omega}_{RA} = f_g\left(\boldsymbol{\omega}, f\right) = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{r=0}^{N_{g'}} b_{g,n'} \operatorname{sign}(\boldsymbol{\omega}_i) \omega_i^{n'} + m_g^T \boldsymbol{\omega} + \boldsymbol{\omega}^T C_g \boldsymbol{\omega} + f^T S_f \boldsymbol{\omega}^T \boldsymbol{\omega}$$



GENERALIZED INERTIAL SENSING MODEL APPLICATION

Why use these new generalized models?

- These models provide coverage for:
 - Varying inertial sensing technologies
 - Varying performance grades of devices
- Rather than starting with a basic model and adding parameters, instead consider all parameters and eliminate the terms that are not relevant to your use case

How can I apply these models to my use case?

- In general, model the parameters which you can observe!
- Include the terms you can reasonably measure or estimate, and remove the terms you can not

Accelerometer Parameters	Symbol(s)	IEEE 1293 Req. Clause	IEEE 1293 Test Clause
Bias			
Absolute Value	K_0	5.3.7.1	12.3.4
Asymmetry	_	5.3.7.2	12.3.15
Scale Factor			
Absolute Value	K_1	5.3.5.1	12.3.4
Asymmetry	_	5.3.5.2	12.3.15
Nonlinearity	K_2, K_3	5.3.6	12.3.15
IA Misalignment	$\delta_{\rm p},\delta_{\rm o}$	5.3.8	12.3.4
Cross-axis Nonlinearity	$K_{\rm pp}$, $K_{\rm oo}$	5.3.10	12.3.15
Cross Coupling	$K_{ m ip}$, $K_{ m io}$, $K_{ m po}$	5.3.11	12.3.15

Gyroscope Parameters	Symbol(s)	IEEE 952 Req. Clause	IEEE 952 Test Clause
Bias	$D_{\mathtt{F}}$	5.3.3.1.1	12.11
Scale Factor			
Absolute Value	S	5.3.2	12.9
Asymmetry	_	5.3.2.1.2	12.9
IA Misalignment	α	5.3.4.1	12.12

$$\tilde{f}_{IRA} = f_a \left(\vec{f}, \vec{\omega}, \dot{\vec{\omega}} \right) = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_{a'}} b_{a,n'} \operatorname{sign}(f_i) f_i^{n'} + \vec{m}_a^T \vec{f} + \vec{f}^T C_a \vec{f} + \vec{\omega}^T S_{\vec{\omega}} \vec{\omega} + \vec{S}_{\dot{\vec{\omega}}}^T \dot{\vec{\omega}}$$

$$\tilde{\omega}_{IRA} = f_g\left(\vec{\omega}, \vec{f}\right) = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{n'=0}^{N_g} b_{g,n'} \operatorname{sign}(\omega_i) \omega_i^{n'} + \vec{m}_g^T \vec{\omega} + \vec{\omega}^T C_g \vec{\omega} + \vec{f}^T S_{\vec{f}} \vec{f}$$



FUTURE IMU MODEL EXTENSIONS

- The generalized model presented today solely focuses on analytic parameters
- Future model extensions include:
 - Build upon IEEE Standards to develop standardized calibration procedures
 - Expanding model parameters as functions of the environment, (i.e., temperature)
 - Vectorizing the component level equations into a fully-realized IMU model
 - Identify and express stochastic error models available for characterization while the unit is under test



SUMMARY

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Input Reference Axis (IRA) Pendulous Axis Or Spin Axis

Inertial Sensing Component Diagram

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$$\tilde{f}_{\mathit{IRA}} = f_a\Big(\vec{f}, \vec{\omega}, \dot{\vec{\omega}}\Big) = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_g} b_{a,n'} \mathrm{sign}(f_i) f_i^{n'} + \vec{m}_a^T \vec{f} + \vec{f}^T C_a \vec{f} + \vec{\omega}^T S_{\vec{\omega}} \vec{\omega} + \vec{f}^T S_{\vec{f}} \vec{f}$$



Bias, Scale Factor, and Higher Order Terms

These terms follow a common polynomial expansion, for example:

$$f_{IRA} = b_{a,0} + b_{a,1}f_i + b_{a,2}f_i^2 + \dots$$

$$\omega_{IRA} = b_{g,0} + b_{g,1}\omega_i + b_{g,2}\omega_i^2 + \dots$$

This polynomial expansion can be re-expressed as the following summation:

$$f_{IRA} = \sum_{n=0}^{N_a} b_{a,n} f_i^n$$

$$\omega_{IRA} = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n$$

Asymmetries in Bias, Scale Factor, and Higher Order Terms

Accompanying each term to the left is the possibility of an asymmetry, which is depending on the *sign* of the input quantity

$$f_{IRA} = (b_{a,0} + b_{a,0} \operatorname{sign}(f_i)) + (b_{a,1} + b_{a,1} \operatorname{sign}(f_i)) f_i + (b_{a,2} + b_{a,2} \operatorname{sign}(f_i)) f_i^2 + \dots
\omega_{IRA} = (b_{g,0} + b_{g,0} \operatorname{sign}(\omega_i)) + (b_{g,1} + b_{g,1} \operatorname{sign}(\omega_i)) \omega_i + (b_{g,2} + b_{g,2} \operatorname{sign}(\omega_i)) \omega_i^2 + \dots$$

Re-stating this expansion into summations provide:

$$f_{IRA} = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_a'} b_{a,n'} \operatorname{sign}(f_i) f_i^n$$

$$\omega_{IRA} = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{n'=0}^{N_g'} b_{g,n'} \operatorname{sign}(\omega_i) \omega_i^n$$



Component Level Misalignment

- Component level misalignment differs from a traditional misalignment matrix used for a triad of sensors. Instead, component level misalignment is only concerned out input axis deflection in the pendulous/spin and output axes.
- Component level misalignment can be characterized will angles α_k and α_o , which describe the amount of angle deflection into its respective axis.
- To account for this misalignment in the IMU model, the deflection angles are re-expressed as:
- Then, the dot product of these misalignment terms and its respective quantity provides:

$$egin{aligned} egin{aligned} egin{aligned} & egin{aligned} & \cos\left(lpha_a
ight) \ & \sin\left(lpha_a
ight) \ & lpha_a \end{aligned} \end{aligned}, \quad egin{aligned} & egin{aligned} & \cos\left(lpha_g
ight) \ & \sin\left(lpha_g
ight) \ & lpha_g \end{aligned} \end{aligned}, \quad lpha_{a/g} = lpha_j^2 + lpha_k^2 \ & \frac{\sin\left(lpha_g
ight)}{lpha_g} lpha_{g,j} \end{aligned}$$

$$f_{IRA} = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_a'} b_{a,n'} \operatorname{sign}(f_i) f_i^n + m_a^T f$$

$$\omega_{IRA} = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{n'=0}^{N_g'} b_{g,n'} \operatorname{sign}(\omega_i) \omega_i^n + m_g^T \omega$$



Cross Coupling and Cross Axis Nonlinearities

 Cross coupling effects are proportional to the product of the sensed quantity of each axis.

$$\Delta f_{CA} = f^T C_a f = \begin{bmatrix} f_i & f_j & f_k \end{bmatrix} \begin{bmatrix} 0 & c_{a,ij} & c_{a,ik} \\ 0 & c_{a,jj} & c_{a,jk} \\ 0 & 0 & c_{a,kk} \end{bmatrix} \begin{bmatrix} f_i \\ f_j \\ f_k \end{bmatrix}$$

 Cross axis nonlinearity effects are proportional to the square of the sensed quantity on the pendulous/spin and output axes.

$$\Delta \omega_{CA} = \omega^T C_g \omega = \begin{bmatrix} \omega_i & \omega_j & \omega_k \end{bmatrix} \begin{bmatrix} 0 & c_{g,ij} & c_{g,ik} \\ 0 & c_{g,jj} & c_{g,jk} \\ 0 & 0 & c_{g,kk} \end{bmatrix} \begin{bmatrix} \omega_i \\ \omega_j \\ \omega_k \end{bmatrix}$$

 Both sets of terms can be expressed within the same matrix such that:

$$f_{IRA} = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_a'} b_{a,n'} \operatorname{sign}(f_i) f_i^n + m_a^T f + f^T C_a f$$

$$\varpi_{IRA} = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{n'=0}^{N_g'} b_{g,n'} \operatorname{sign}(\omega_i) \omega_i^n + m_g^T \omega + \omega^T C_g \omega$$



Sensitivity to the Opposite Phenomenon

 It is not uncommon for accelerometers to be sensitive to angular velocities or angular accelerations.

$$\Delta f_{\omega,\varpi} = \omega^T S_{\omega} \omega + s_{\varpi} \omega = \begin{bmatrix} \omega_i & \omega_j & \omega_k \end{bmatrix} \begin{bmatrix} 0 & s_{\omega,ij} & s_{\omega,ik} \\ 0 & 0 & s_{\omega,jk} \\ 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \omega_i \\ \omega_j \\ \omega_k \end{bmatrix} + \begin{bmatrix} s_{\varpi,i} & s_{\varpi,j} & s_{\varpi,k} \end{bmatrix} \begin{bmatrix} \omega_i \\ \omega_j \\ \omega_k \end{bmatrix}$$

• It is also not uncommon for gyroscopes to be sensitive to specific force.

$$\Delta \omega_f^{\square} = f^T S_f^{\square} f = egin{bmatrix} f_i & f_j & f_k \end{bmatrix} egin{bmatrix} 0 & s_{f,ij} & s_{f,ik} \ 0 & 0 & s_{f,jk} \ 0 & 0 & 0 \end{bmatrix} egin{bmatrix} f_i \ f_j \ f_k \end{bmatrix}$$

 These sensitivities are more common in mechanical designs such as MEMS technologies others that contain a physical proof mass.

$$f_{IRA} = \sum_{n=0}^{N_a} b_{a,n} f_i^n + \sum_{n'=0}^{N_a'} b_{a,n'} \operatorname{sign}(f_i) f_i^n + m_a^T f + f^T C_a f + \omega^T S_{\omega} \omega + S_{\omega}^T \omega$$

These sensitivities are model such that:

$$\omega_{IRA} = \sum_{n=0}^{N_g} b_{g,n} \omega_i^n + \sum_{n'=0}^{N_g'} b_{g,n} \operatorname{sign}(\omega_i) \omega_i^n + m_g^T \omega + \omega^T C_g \omega + f^T S_f^{\Box} f$$