

## 1.1 FEED-FORWARD AND FEEDBACK CONTROL

Feedforward control has no feedback from the output. To solve this, the controller is set using the *inversion control strategy*.

$$C(s) = \frac{1}{P(s)}$$

If the resulting  $C(s)$  is improper  $C(s)$  can be modified to the following:

$$C(s) = \frac{1}{P(s) \cdot (\tau s + 1)}$$

For good tracking up to  $2\pi$ , set  $\frac{1}{\tau} \geq 10 * 2\pi \Rightarrow \tau \leq 0.016$

Feedback control feeds the output back to the reference. High-gain control in particular frequencies with a unity-feedback control loop is a good control strategy. At those frequencies, we achieve good tracking, disturbance rejection, and sensitivity.

### 1.1 INTRO TERMS

Open-loop poles refer to the poles of  $P(s)C(s)$ .

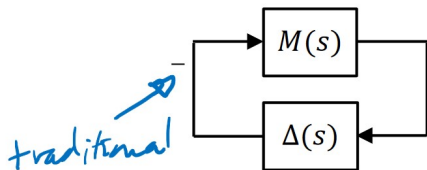
Closed-loop poles are the roots of  $N_C N_P + D_C D_P$ .

Open left-half plane (OLHP) refer to  $Re(s) < 0$ .

Close left-half plane (CLHP) refer to  $Re(s) \leq 0$ .

Open right-half plane (ORHP) and Closed right-half plane (CRHP) are similarly defined.

## 2.8 SMALL GAIN THEOREM



$M(s)$  is known and stable

$\Delta(s)$  is uncertain but stable with  $|\Delta(j\omega)| < 1$  for all  $\omega$

### 3.1 YOULA PARAMETERIZATION

For a stable plant. The set of all stabilizing controllers equals the following set:

$$\left\{ C(s) : C(s) = \frac{Q(s)}{1 - P(s)Q(s)} \right\}$$

where  $Q(s)$  is proper and stable

## COPRIME FACTORIZATION

Let  $P(s)$  be any proper TF. Then there exists stable proper TFs  $N_P(s)$ ,  $X_P(s)$ ,  $M_P(s)$ , and  $Y_P(s)$  such that the following relationships hold for all  $s$ :

$$P(s) = \frac{N_P(s)}{M_P(s)}$$

$$N_P(s)X_P(s) + M_P(s)Y_P(s) = 1$$

### YOULA PARAMETERIZATION FOR GENERAL $P(s)$

For an unstable plant. Assume there are no CRHP pole-zero cancellations in  $P(s)$ . Perform a coprime factorization on  $P(s)$ . Then the set of all stabilizing controllers equals the following set:

$$\left\{ C(s) : C(s) = \frac{X_P(s) + M_P(s)Q(s)}{Y_P(s) - N_P(s)Q(s)} \right\}$$

where  $Q(s)$  is proper and stable

### 3.2 PERFORMANCE LIMITATIONS

At frequencies where  $|S(j\omega)| \ll 1$ , good tracking, good disturbance rejection, and good sensitivity. Exact opposite when  $|S(j\omega)| \gg 1$ .

$$S(s) = \frac{1}{1 + L(s)}$$

Ideally:

- $|S(j\omega)|$  is small when  $|L(j\omega)|$  is large
- $|S(j\omega)|$  is near one (0 dB) when  $|L(j\omega)|$  is small

The complementary sensitivity,  $T(s)$ , tells us that robust stability is achieved iff there is nominal stability and:

$$|W(j\omega)T_0(j\omega)| < 1 \text{ for all } \omega$$

$$\Leftrightarrow |T_0(j\omega)| < \frac{1}{|W(j\omega)|} \text{ for all } \omega$$

It also tells us how the closed-loop system responds to sensor noise.

$$T(s) = \frac{L(s)}{1 + L(s)}$$

Ideally:

- $|T(j\omega)|$  is small (in fact,  $|T(j\omega)| \approx |L(j\omega)|$ ) at high frequencies when sensor noise is significant, and little feedback effort is used
- $|S(j\omega)|$  is near one (0 dB) at low frequencies, where lots of feedback effort is used

The relationship between the two equations:

$$S(j\omega) + T(j\omega) = \frac{1}{1 + L(j\omega)} + \frac{L(j\omega)}{1 + L(j\omega)} = 1 \quad (1)$$

$$|S(j\omega)| + |T(j\omega)| \geq 1 \text{ for each } \omega \quad (2)$$

Notes:

- tradeoff between the reasons for wanting  $|S(j\omega)|$  to be small and the reason for wanting  $|T(j\omega)|$
- both cannot both be "good" (i.e., very small) at the same frequency, but it is possible for both to be "bad" (i.e., very large) at the same frequency
- We never want the magnitude bode plots of  $S(s)$  or  $T(s)$  to have large peaks

### 3.3 INTERPOLATION CONSTRAINTS

Assume that the closed-loop system is stable.

- If  $P(s)C(s)$  has a CRHP zero at  $s = z$ , then  $S(z) = 1$  and  $T(z) = 0$
- If  $P(s)C(s)$  has a CRHP pole at  $s = p$ , then  $S(p) = 0$  and  $T(p) = 1$

### 3.4 PERFORMANCE LIMITATIONS DUE TO ORHP

Bandwidth rule of thumb is

$$\text{bandwidth} > 2p$$

LEMMA

If:

- The closed-loop system is stable
- The plant has an open-loop unstable real pole at  $s = p > 0$
- The reference signal is a unit step

Then the tracking error,  $e(t) = r(t) - y(t)$  necessarily satisfies:

$$\int_0^\infty e(t)e^{-pt} dt = 0$$

An ORHP pole in the plant leads to overshoot in the closed-loop step response

$$y_{OS} \geq (1 - 0.9y_\infty)(e^{pt_r} - 1) > 0$$

$$t_r \leq \frac{1}{p} \ln \left( 1 + \frac{y_{OS}}{1 - 0.9y_\infty} \right)$$

### BODE SENSITIVITY INTEGRAL

Assume that:

- The controller stabilizes the closed-loop system
- The loop gain,  $L(s) = P(s)C(s)$ , has a **relative degree** of at least two (i.e., it has at least two more poles than zeros).

Let  $N_p \geq 0$  denote the number of ORHP poles of  $L(s)$  and denote the ORHP poles by  $p_1, \dots, p_{N_p}$ . Then the sensitivity function must satisfy

$$\int_0^\infty \ln |S(j\omega)| d\omega = \sum_{i=1}^{N_p} \text{Re}(p_i) \geq 0$$

If there are no unstable poles, then the equation reduces to

$$\int_0^\infty \ln |S(j\omega)| d\omega = 0$$

Where the negative area equals the positive area through a *waterbed effect*.

If there are an unstable poles, then the area of sensitivity increase must be *greater than* the area of sensitivity decrease.

### 3.5 PERFORMANCE LIMITATIONS DUE TO ORHP ZEROS

If the plant has an ORHP real zero at  $s = z$ , then good performance can be achieved if the closed-loop bandwidth is significantly smaller than  $z$ . ORHP Zeroes:

- Zeroes near the origin are worse than ORHP zeroes far from the origin
- ORHP zeroes lead to *undershoot* in the step responses
- You cannot get rid of CRHP zeroes, since  $P(s)C(s)$  is the numerator of  $T_{ry}(s)$

LEMMA

$$\int_0^\infty y(t)e^{-zt} dt = 0$$

$$y_{US} \geq \frac{0.98y_\infty}{e^{zt_s} - 1} > 0$$

$$t_r \leq \frac{1}{z} \ln \left( 1 + \frac{0.98y_\infty}{y_{US}} \right)$$

Bandwidth rule of thumb given a ORHP zero:

$$\text{bandwidth} < z/2$$

## POISSON INTEGRAL

$$\int_0^\infty \ln|S(j\omega)|W(\omega) d\omega = \pi \sum_{i=1}^{N_p} \ln \left| \frac{p_i + z}{p_i^* - z} \right| \geq 0$$

$$W(\omega) = \frac{2z}{z^2 + \omega^2}$$

### 4.1 BASIC MIMO CONCEPTS

$$y = (I + PCH)^{-1}PCr$$

$$L_o = PC$$

$$S_o = (I + L_o)^{-1}$$

$$T_o = L_o(I + L_o)^{-1}$$

$$S_o(s) + T_o(s) = I$$

The *minor* of a matrix is the determinant of any square submatrix obtained by deleting certain row's and/or columns of the matrix

The *rank* of a matrix is the max number of linearly independent column/row vectors.

To find  $\phi(s)$ , find all the minors. 1st order is just the element, while second order is as described above. Find the lowest common denominator of all non-zero minors.

A MIMO system is stable iff all the individual elements of the system are stable, and if all MIMO system poles are in the OLHP

The zero polynomial  $z(s)$  is the greatest common divisor of all the r-th order minors. Steps:

1. Find the minors for the MIMO system
2. Determine  $\phi(s)$
3. Make each minor as a fraction over  $\phi(s)$
4. Find the GCD of the r-th order minors.

### 4.2 DECENTRALIZED CONTROL

- Assume plant is square
- Ignore cross-channel interactions in the plant, resulting in a MIMO controller that is diagonal
- If there is too much coupling, then the resulting system is unstable

### 4.3 DECOUPLING CONTROL

- A decoupling block is added:  $P_{aug} = PW$
- Design a decentralized controller for the augmented plant:  $C = W\bar{C}$

Steps:

1. Decouple the plant at dc. i.e. Solve  $P(0)$
2. Solve  $W(s) = P_1^{-1}(0)$ , to get  $P_{aug} = P * W$
3. Solve for the controllers, ignoring the cross-channel interactions
4.  $C(s) = W\bar{C}(s)$

### 4.4 SEQUENTIAL LOOP CLOSURE

- Accounts for cross-channel interactions
- Feedback loops are closed sequentially instead of in parallel
- **This approach only applies if the plant has a triangular structure.**
- Ex:

– Given Plant:

$$P(s) = \begin{bmatrix} P_{11} & P_{12} \\ 0 & P_{22} \end{bmatrix}$$

$$y_1 = P_{11}u_1 + P_{12}u_2 \text{ and } y_2 = P_{22}u_2$$

- first design  $C_{22}$  to stabilize  $P_{22}$  since it is independent of  $P_{11}$  and  $P_{12}$
- Then design  $C_{22}$

### 5.1 BASIC STATE-SPACE FRAMEWORK AND MANIPULATIONS

State-space models are not unique. Infinite number of state-space realizations that correspond to the same transfer functions They follow the form:

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

Manipulation 1: Convert from SS to TF

$$Y(s) = [D + C(sI - A)^{-1}B] \cdot U(s)$$

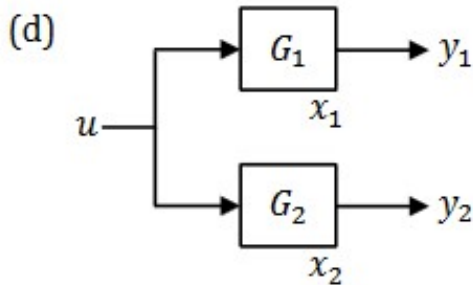
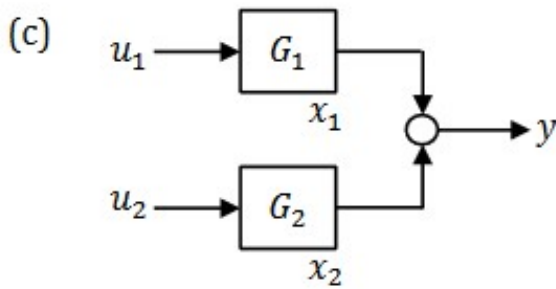
Manipulation 2: Poles and zeros of a SS system

- Poles of the SS system are  $\text{eig}(A) == \text{roots of } \det(sI - A)$
- Zeroes are the values of  $s$  where the rank of the following matrix falls below its normal rank:

$$R(s) = \begin{bmatrix} (sI - A) & -B \\ C & D \end{bmatrix}$$

Which can be computed by finding the the roots of  $\det(R(s))$

Manipulation 3: Interconnecting state-space models



(c)

$$\begin{aligned}\dot{x}_1 &= A_1 x_1 + B_1 u_1 \\ \dot{x}_2 &= A_2 x_2 + B_2 u_2 \\ y &= (C_1 x_1 + D_1 u_1) + (C_2 x_2 + D_2 u_2)\end{aligned}$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} A_1 & 0 \\ 0 & A_2 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} + \begin{pmatrix} B_1 & 0 \\ 0 & B_2 \end{pmatrix} \begin{pmatrix} u_1 \\ u_2 \end{pmatrix}$$

$$y = (C_1 \ C_2) \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} + (D_1 \ D_2) \begin{pmatrix} u_1 \\ u_2 \end{pmatrix}$$

(d)

$$\begin{aligned}\dot{x}_1 &= A_1 x_1 + B_1 u \\ \dot{x}_2 &= A_2 x_2 + B_2 u \\ y_1 &= (C_1 x_1 + D_1 u) \\ y_2 &= (C_2 x_2 + D_2 u)\end{aligned}$$

$$\begin{pmatrix} \dot{x}_1 \\ \dot{x}_2 \end{pmatrix} = \begin{pmatrix} A_1 & 0 \\ 0 & A_2 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} + \begin{pmatrix} B_1 \\ B_2 \end{pmatrix} u$$

$$\begin{pmatrix} y_1 \\ y_2 \end{pmatrix} = \begin{pmatrix} C_1 & 0 \\ 0 & C_2 \end{pmatrix} \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} + \begin{pmatrix} D_1 \\ D_2 \end{pmatrix} u$$

Manipulation 4: Converting from TF to SS

- SISO case, creates a SS with controllable canonical form (CCF):

$$\dot{x} = \begin{bmatrix} 0 & 1 & 0 & \cdots & 0 \\ 0 & 0 & 1 & & 0 \\ 0 & 0 & 0 & \ddots & \vdots \\ \vdots & \vdots & \vdots & & 1 \\ -a_0 & -a_1 & -a_2 & \cdots & -a_{n-1} \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} u$$

$$y = [c_0 \ c_1 \ c_2 \ \cdots \ c_{n-1}] x + [d] u$$

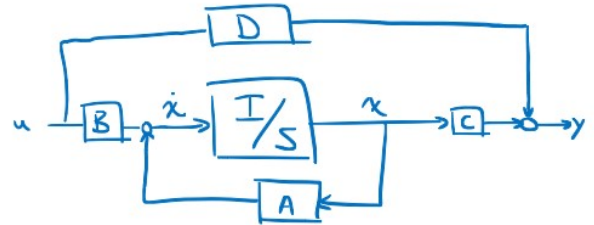
- SIMO case:

$$G(s) = \begin{bmatrix} A_1 & 0 & \cdots & B_1 \\ 0 & A_2 & & B_2 \\ \vdots & & \ddots & \vdots \\ C_1 & 0 & \cdots & D_1 \\ 0 & C_2 & & D_2 \\ \vdots & & \ddots & \vdots \end{bmatrix}$$

- MIMO case:

$$G(s) = \begin{bmatrix} A_1 & 0 & \cdots & B_1 & 0 & \cdots \\ 0 & A_2 & & 0 & B_2 & \\ \vdots & & \ddots & \vdots & & \ddots \\ C_1 & C_2 & \cdots & D_1 & D_2 & \cdots \end{bmatrix}$$

Manipulation 5: Block Diagram of SS Model



## 5.2 CONTROLLABILITY, OBSERVABILITY, STABILIZABILITY, AND DETECTABILITY

Modes that show up on the output  $y(t)$  are said to be *Observable*. If all modes are observable, the state-space realization is said to be *observable*. If not, it's *unobservable*.

Modes that are affected by the input are said to be *controllable*. If all modes are controllable, the state-space realization is said to be *controllable*. If not, it's *uncontrollable*.

A state-space realization where all unstable modes are controllable is said to be *stabilizable*.

controllability  $\Rightarrow$  stabilizability

A state-space realization where all unstable modes are observable is said to be *detectable*.

observability  $\Rightarrow$  detectability

A system is controllable and observable iff the corresponding TF matrix has no P-Z cancellations.

A system is stabilizable and detectable iff the corresponding TF matrix has no unstable P-Z cancellations.

### 5.2.2 CONTROLLABILITY AND STABILIZABILITY

$A, B$  is controllable iff the  $n \times (nm)$  *controllability matrix* has rank  $n$ .

$$M_c = [B \ AB \ A^2 B \ \cdots \ A^{n-1} B]$$

Using the PBH controllability test,  $A, B$  is controllable iff

$$\text{rank} [A - \lambda I \ B] = n \text{ for all } \lambda \in \text{eig}(A)$$

Recall:  $\text{eig}(A)$  is found as the solutions to  $\lambda$ , for  $\det(A - \lambda I)$ .

### 5.2.3 OBSERVABILITY AND DETECTABILITY

$A, C$  is observable iff the (pn) x n *observability matrix* has rank  $n$ .

$$M_o = \begin{bmatrix} C \\ CA \\ CA^2 \\ \vdots \\ CA^{n-1} \end{bmatrix}$$

Using the PBH observability test,  $A, C$  is controllable iff

$$\text{rank} \begin{bmatrix} A - \lambda I \\ C \end{bmatrix} = n \text{ for all } \lambda \in \text{eig}(A)$$

Recall:  $\text{eig}(A)$  is found as the solutions to  $\lambda$ , for  $\det(A - \lambda I)$

### 5.3 PLAIN STATE FEEDBACK CONTROL (POLE PLACEMENT)

The plain state-feedback controller  $u = -Kx + r$  can arbitrarily place poles. The plant **must be controllable**. If it is uncontrollable but stabilizable, we cannot place poles arbitrarily but we can find a  $K$  to stabilize  $A - BK$ .

$K$  is a row vector.

Steps:

$$\begin{aligned} A_{CLS} &= A - BK \\ \pi(s) &= \det(sI - A_{CLS}) \\ \pi_{des}(s) &= \pi(s) \end{aligned}$$

CLS poles are at  $\text{eig}(A_{CLS}) = \text{eig}(A - BK)$

### 5.4 OBSERVER

Consider the following Plant:

$$\begin{aligned} \dot{x} &= Ax + Bu \\ y &= Cx + Du \end{aligned}$$

and Observer

$$\begin{aligned} \dot{\hat{x}} &= A\hat{x} + Bu + H(y - \hat{y}) \\ \hat{y} &= C\hat{x} + Du \end{aligned}$$

The estimation error is given by  $e(t) = x(t) - \hat{x}(t)$ . Then:

- $\dot{e} = (A - HC)e$ , and  $e(t = \infty) = 0$  for any IC  $e(0)$  iff  $\text{eig}(A - HC) \in \text{OLHP}$ .
- The poles of  $\text{eig}(A - HC)$  can be placed arbitrarily iff  $(A, C)$  is observable.
- The poles of  $\text{eig}(A - HC)$  can be placed in OLHP iff  $(A, C)$  is only detectable.

$H$  is a column vector.

Steps:

$$\begin{aligned} \pi(s) &= \det(sI - (A - HC)) \\ \pi_{des}(s) &= \pi(s) \end{aligned}$$

Resulting state-space realization:

$$\begin{aligned} \begin{bmatrix} \dot{x} \\ \dot{\hat{x}} \end{bmatrix} &= \begin{bmatrix} A & 0 \\ HC & A - HC \end{bmatrix} \begin{bmatrix} x \\ \hat{x} \end{bmatrix} + \begin{bmatrix} B \\ B \end{bmatrix} u \\ \bar{y} &= I \cdot \begin{bmatrix} x \\ \hat{x} \end{bmatrix} \end{aligned}$$

### 5.4 OBSERVER-BASED STATE-FEEDBACK CONTROL

Mixture of solving for  $K$  and  $H$ , using the above steps. The CLS for the observer-based state-feedback controller:

$$\begin{aligned} \begin{bmatrix} \dot{x} \\ \dot{e} \end{bmatrix} &= \begin{bmatrix} A - BK & BK \\ 0 & A - HC \end{bmatrix} \begin{bmatrix} x \\ e \end{bmatrix} + \begin{bmatrix} B \\ 0 \end{bmatrix} r \\ y &= [C \ 0] \begin{bmatrix} x \\ e \end{bmatrix} \end{aligned}$$

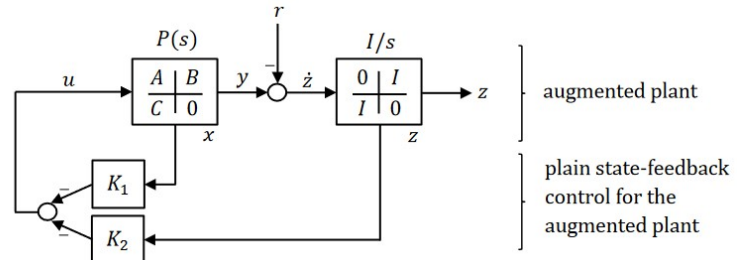
The closed loop poles are found at  $\text{eig}(A - HC) \cup \text{eig}(A - BK)$

General rule-of-thumb is to place  $\text{eig}(A - HC)$  about twice as fast (i.e., real parts twice as large) as  $\text{eig}(A - BK)$ .

$K$ : Arbitrary pole placement is possible iff  $(A, B)$  is controllable, and stabilization is possible iff  $(A, B)$  is stabilizable.

$H$ : Arbitrary placement of poles of the observer error-dynamics is possible iff  $(A, C)$  is observable, and they can be placed (not necessarily arbitrarily) in the OLHP iff  $(A, C)$  is detectable.

### 5.5 PLAIN STATE-FEEDBACK CONTROLLER WITH INTEGRAL ACTION



$$\begin{aligned} \begin{bmatrix} \dot{x} \\ \dot{z} \end{bmatrix} &= \begin{bmatrix} A & 0 \\ C & 0 \end{bmatrix} \begin{bmatrix} x \\ z \end{bmatrix} + \begin{bmatrix} B \\ 0 \end{bmatrix} u + \begin{bmatrix} 0 \\ -I \end{bmatrix} r \\ \dot{\tilde{x}} &= \tilde{A}\tilde{x} + \tilde{B}u + \tilde{B}_r r \end{aligned}$$

The CLS poles ( $\text{eig}(\tilde{A} - \tilde{B}K)$ ) can be placed arbitrarily iff  $(\tilde{A}, \tilde{B})$  is controllable. Essentially, the same steps as in 5.3, however  $A$  is replaced with  $\tilde{A}$ . Still need to calculate the characteristic polynomial using  $\det(sI - (\tilde{A} - \tilde{B}K))$

## 5.5 OBSERVER-BASED STATE FEEDBACK CONTROLLER WITH INTEGRAL ACTION

Essentially, the same steps as in 5.4, however  $A$  is replaced with  $\tilde{A}$ .

CLS poles:  $\text{eig}(\tilde{A} - \tilde{B}K) \cup \text{eig}(A - HC)$

K: Arbitrary pole placement is possible iff  $(\tilde{A}, \tilde{B})$  is controllable. Stabilization is possible iff  $\tilde{A}, \tilde{B}$  is stabilizable.

H: Arbitrary placement of poles of the observer error-dynamics is possible iff  $(A, C)$  is observable. They can be placed (not necessarily arbitrarily) in the OLHP iff  $(A, C)$  is detectable.

## 5.6 OPTIMAL STATE-FEEDBACK CONTROL

Where  $Q \geq 0$  and  $R > 0$ , minimize:

$$J = \int_0^\infty x(t)^T Q x(t) + u(t)^T R u(t) dt$$

Assume that  $(A, B)$  is stabilizable and  $(A, Q)$  is detectable.

The resulting controller is the stabilizing linear state-feedback controller,  $u(t) = -Kx(t)$ , where  $K$  is computed by finding the unique symmetric PSD solution  $P$  to the algebraic Riccati equation (ARE):

$$A^T P + P A - P B R^{-1} B^T P + Q = 0$$

and setting  $K = R^{-1} B^T P$ .

$Q$  and  $R$  are tuning knobs.  $Q$  is commonly a diagonal matrix where  $q_{ii} \geq 0$ . This results in:

$$x(t)^T Q x(t) = q_{11} x_1^2(t) + q_{22} x_2^2(t) + \dots + q_{nn} x_n^2(t)$$

$Q$  is commonly chosen as  $Q = C^T C$ . This leads to:

$$\begin{aligned} y(t) &= Cx(t) \\ x(t)^T Q x(t) &= \|y(t)\|^2 \end{aligned}$$

$R$  is also commonly a diagonal matrix.

Decreasing  $R$  minimizes the control effort, and the poles go off to  $-\infty$ . Increasing  $R$  increases the control effort, and the poles approach the OL poles.

Changing  $R$  does the opposite effect.

Increasing  $Q$  and  $R$  by the same factor does not change  $K$  or the pole locations.