## Extending the Kalman Filter to Accelerometers with a DC Offset

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## 1 FILTER FORMULATION

The problem is to modify the Kalman filtered from *Bock et al.* (2011) to a filter that can ingest accelerometer data with a constant DC offset. That is the measured or observed acceleration  $a^{obs}$  is related to the true acceleration,  $a^{true}$  by

$$a_k^{true} = a_k^{obs} - \Omega_k + \epsilon_a$$

where  $\Omega_k$  is the DC offset at epoch k and  $\epsilon_a$  is the accelerometer noise. If we define our system states as the displacement d, the velocity v and the DC offset  $\Omega$  then following *Lewis et al.* (2008) section 2.4 we can write the continuous difference equation for this system:

$$\frac{d}{dt} \begin{bmatrix} d(t) \\ v(t) \\ \Omega(t) \end{bmatrix} = \frac{d}{dt} \mathbf{x}(t) = \mathbf{A}(t)\mathbf{x}(t) + \mathbf{B}(t)u(t) + \epsilon(t)$$

where

$$\mathbf{A} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & -1 \\ 0 & 0 & 0 \end{bmatrix} \; ; \; \mathbf{B} = \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} \; ; \; u = a^{obs} \; ; \; \boldsymbol{\epsilon} = \begin{bmatrix} 0 \\ \epsilon_a \\ \epsilon_{\Omega} \end{bmatrix} \; ,$$

where  $\epsilon_a$  is the accelerometer noise and  $\epsilon_\Omega$  is the DC offset noise. A small value of  $\epsilon_\Omega$  will allow the DC offset to vary slowly through time. This means that the noise vector  $\epsilon$  is Gaussian, such that  $\epsilon \sim (0, \mathbf{Q})$  where the covariance  $\mathbf{Q}$  depends on the accelerometer and DC offset noise

variances  $\sigma_a$  and  $\sigma_{\Omega}$  like

$$Q = \begin{bmatrix} 0 & 0 & 0 \\ 0 & \sigma_a & 0 \\ 0 & 0 & \sigma_{\Omega} \end{bmatrix}$$

You can expand the last two equations to verify the system:

$$\frac{d}{dt} \begin{bmatrix} d(t) \\ v(t) \\ \Omega(t) \end{bmatrix} = \begin{bmatrix} \dot{d} \\ \dot{v} \\ \dot{\Omega} \end{bmatrix} = \begin{bmatrix} v \\ -\Omega + a^{obs} + \epsilon_a \\ \epsilon_{\Omega} \end{bmatrix}.$$

If we are measuring noisy displacements we define the measurement process:

$$z(t) = d^{obs}(t) = \mathbf{H}(t)\mathbf{x}(t) + \eta_d = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix} \mathbf{x}(t) + \eta_d$$

where  $\eta_d$  is the displacement observation noise, which when assumed gaussian has a distribution  $\eta_d \sim (0, \mathbf{R})$  with  $\mathbf{R} = \sigma_d$  the displacement noise variance. The continuous system can now be discretized. Again following *Lewis et al.*(2008), we write the discrete system:

$$\mathbf{x}_{k+1} = \mathbf{A}^s \mathbf{x}_k + \mathbf{B}^s a_k^{obs} + \epsilon_k$$

where the discretized noise vector is Gaussian and distributed like  $\epsilon_k \sim (0, \mathbf{Q}^s)$ . If we discretize the continuous system at the sampling rate  $\tau_a$  of the accelerometer we must redefine the system transition matrices. These are obtained (*Lewis et al.*, 2008) from the MacLaurin series expansion of the integral forms as:

$$\mathbf{A}^{s} = \mathbf{I} + \mathbf{A}\tau_{a} + \frac{\mathbf{A}^{2}\tau^{2}}{2} = \begin{bmatrix} 1 & \tau_{a} & -\tau_{a}^{2}/2 \\ 0 & 1 & -\tau_{a} \\ 0 & 0 & 1 \end{bmatrix};$$

$$\mathbf{B}^{s} = \mathbf{B}\boldsymbol{\tau}_{a} + \frac{\mathbf{A}\mathbf{B}\boldsymbol{\tau}_{a}^{2}}{2} = \begin{bmatrix} \boldsymbol{\tau}_{a}^{2}/2 \\ \boldsymbol{\tau}_{a} \\ 0 \end{bmatrix};$$

$$\mathbf{Q}^s = \mathbf{Q}\boldsymbol{\tau}_a + \frac{1}{2}(\mathbf{A}\mathbf{Q} + \mathbf{Q}\mathbf{A}^T)\boldsymbol{\tau}_a^2 + \frac{1}{3}\mathbf{A}\mathbf{Q}\mathbf{A}^T\boldsymbol{\tau}_a^3 = \begin{bmatrix} \sigma_a\boldsymbol{\tau}_a^3/3 & \sigma_a\boldsymbol{\tau}_a^2/2 & 0\\ \sigma_a\boldsymbol{\tau}_a^2/2 & \sigma_a\boldsymbol{\tau}_a + \sigma_\Omega\boldsymbol{\tau}_a^3/3 & -\sigma_\Omega\boldsymbol{\tau}_a^2/2\\ 0 & -\sigma_\Omega\boldsymbol{\tau}_a^2/2 & \sigma_\Omega\boldsymbol{\tau}_a \end{bmatrix}.$$

The sampled version of the continuous measurement is written as:

$$z_k = d_k^{obs} = \mathbf{H}^s \mathbf{x}_k + \eta_d ,$$

where the noise is is white and Gaussian such that  $\eta_d \sim (0, R^s)$ . Sampling at the rate of the GPS,  $\tau_d$  the discretized matrices are simply:

$$\mathbf{H}^s = \mathbf{H} = \begin{bmatrix} 1 & 0 & 0 \end{bmatrix}$$

$$R^s = \sigma_d/\tau_d$$
.

With these definitions the traditional system update and measurement update stages of the Kalman filter can be carried out.