

OPERATIONAL CHARACTERISTICS OF A PROTOTYPE
SPACEBORNE HYDROGEN MASER*

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ABSTRACT

The operational characteristics are described of an advanced development model of a spaceborne hydrogen maser developed for the Naval Research Laboratory for possible use in their NTS-3 spacecraft. The NTS-3 is a technology satellite built as part of the Navy's support for the NAVSTAR/Global Positioning System. Size reduction is shown to not necessarily degrade maser performance.

INTRODUCTION

This paper describes our experiences in developing and operating an advanced development model (ADM) hydrogen maser testbed for the Naval Research Laboratory (NRL). The design of a space-qualifiable version based on these experiences has been described in a companion paper and will not be repeated here.

The advanced development model maser, shown in Figure 1 and termed "HYMNS III" (Hydrogen Maser for Navigational Satellites, IIIrd version) was delivered to the NRL in October 1977 and was immediately operational. Prior to delivery, some weeks of operating experience were obtained at Hughes Research Laboratories on the completed maser, and prior to this, considerable operating experience was obtained with an earlier version, HYMNS II, and a demountable testbed, HYMNS I. Separate tests were also made on various subsystems as well. For convenience and ease of reference, the operational characteristics of HYMNS III are described under the headings of its major subsystems.

Vacuum System and Pumps

For this prototype spaceborne maser, no particular attempt was made to fabricate a completely sealed, all-metallic system. Several viton O-rings as well as an indium wire seal and copper gaskets were employed.

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Figure 1. HYMNS III advanced development model testbed maser.

Typical residual pressure in the maser was about 3×10^{-7} Torr. A Varian Hi-Q ion pump with a rated speed of 30 liter/sec for hydrogen and weighing about 20 lb was selected early in the program for its compactness and relatively small stray magnetic field. With an operational hydrogen through-put of about 0.05 μ -liter/sec, the pump dissipates only about 7 W of electrical power, thus eliminating the need for cooling water. With a claimed 8000 Torr-liter hydrogen capacity, the Hi-Q pump would have an estimated life of 7.6 yr. Unfortunately, the pump requires a 7.5 kV operating voltage which is considered undesirable for use in space. Laboratory life tests at only slightly accelerated hydrogen flow rates also indicated that premature pump failure is likely to occur due to short circuits caused by detached flakes of hydrogen-saturated cathode material. As a result, alternative vacuum pumps were considered; active-metal chemical getter pumps were evaluated in several life tests and were found to be satisfactory for both the ADM and the proposed spaceborne design. As a result of these tests, a SAES getter cartridge pump was added to the HYMNS III maser. After an initial activation at high temperature, this zirconium-aluminum alloy cartridge pump hydrogen at room temperature with no power consumption. Its specified end-of-life capacity of about 20,000 Torr-liter has been confirmed by accelerated life tests giving it an estimated life of about 12.7 yr. under normal maser operation without an ion pump. The singular disadvantage of the sorption pump is that at room temperature its pumping speed for gases other than hydrogen is very small. Thus a small supplementary ion pump will be required to scavenge impurities. For this reason the Varian Hi-Q pump was retained on HYMNS III. Even if the efficiency of the beam optical system were to remain at the level demonstrated in HYMNS III, we can confidently project a total pump subsystem weighing no more than 20 lb and consuming less than 3 W for a spaceborne maser with a five-year design life.

Magnetic Field and Shielding

HYMNS III employs four layers of Mu-80 cylindrical magnetic shields with domed end caps. During the development process we evaluated shields fabricated from three different materials (molypermalloys, hypernom and Mu-80) from three different suppliers. As far as can be judged from the maser performance, we have no reason to prefer one over the others. The response to degaussing is also very similar for the three materials.

To generate the small quantization field required to operate HYMNS III, a solenoid with a second-order correction is employed. Since the current through the correction coil is a constant fraction of that in the main coil, no separate optimization was required. The homogeneity of the magnetic field averaged over the maser storage bulb is such that strong maser oscillation was obtained at applied magnetic fields as low as 50 μ G. In fact, steady maser oscillation was observed at applied

fields as low as 20 μ G, at which point residual fields from the shields and stray fields from the ac heaters of the temperature control system introduce significant perturbations.

Thermal Control System

Temperature-control resistive heater windings in HYMNS III employ alternating currents. Since perfect bifilar windings are difficult to obtain, perturbations to the maser oscillation frequency caused by magnetic field fluctuations arising from direct-current heater current variations are substantially reduced by using ac. This makes a significant contribution to the excellent operational characteristics of the maser at very low applied magnetic fields. Four separate heaters and control circuits are employed. Glass bead thermistors with nominal resistance of 50 k Ω at 25°C are used as sensors.

Atomic Hydrogen Source and Beam Optics

Atomic hydrogen is obtained by rf discharge in a 2-in. diam pyrex bulb. Delays in material delivery prevented the incorporation of a novel metal-ceramic dissociator in the delivered HYMNS III maser. However, a prototype metal-ceramic dissociator had been bench-tested separately and was found to be at least as efficient in hydrogen atom production. The metal-ceramic structure provides significantly improved mechanical strength and eliminates any possibility of breakage in the feed lines supplying hydrogen. Dissociator pressure regulation is obtained by a thermally regulated palladium-silver alloy valve in the form of a thin-wall tubing which formed part of the electrical circuit. Typical power consumption to heat this valve was about 0.5 W with a control constant of about 30 sec. The rf power supply utilized a linear power amplifier driven by a crystal-controlled oscillator at about 150 MHz. For normal maser operation, a sustaining rf power of about 3 to 5 W is sufficient. It is necessary to momentarily raise the rf level to about 10 W to ignite the discharge. The low sustaining drive level is desirable to extend the life of the dissociator. Moreover, higher rf levels do not give correspondingly higher atom production, as proved by actual measurement using thermal wire-bridge detectors and optical photometers in a separate test-stand evaluation of dissociator performance.

A hexapole magnet (FTS model HM-2) is used for state selection. Since the spacing between the hexapole magnet and the storage bulb aperture is only 8 in., a beam stop is necessary to block the unselected atoms traveling along the axis of the magnet. However, the efficiency of the system is such that under normal operation conditions, a total hydrogen throughput of less than 0.05 micron-liter/sec is sufficient.

For spin-exchange tuning, the flux is modulated by an electrically operated beam shutter with a beam attenuation of about 50% rather than by changing the flow or rf drive to the dissociator bulb. Since the

shutter operates independently of the dissociator, the response is essentially instantaneous. More important, the dissociator operating parameters remain constant, and the chances for irreversible damage to the dissociator by pressure or drive modulation are greatly reduced.

Microwave Cavity

The TE₀₁₁ microwave cavity is fabricated from a precision-machined satin-fused silica cylinder with aluminum end plates. The inside dimensions are 12 in. diam by 7.4 in. long, giving a length reduction of about 3.5 in. compared to the right cylindrical cavity. It is equipped with a linear varactor tuning loop and a single 50 Ω output coupling line. A high-conductivity silver coating gives the cavity a measured loaded Q of 65,000 with a coupling coefficient of 0.21. This is about 98% of the theoretical maximum. The resonant frequency of the cavity when loaded with the 6-in. maser storage bulb has a measured temperature coefficient of 1.3 kHz/°C.

To prevent the long-term drift in the cavity resonant frequency from limiting the maser stability performance, an electronic cavity frequency monitoring and control system was designed. The system operated by coherently detecting switched test signals at frequencies symmetrically situated with respect to the spin-exchange-tuned cavity frequency, and offset from it by about half the cavity resonant width. The error voltage thus derived is used to bias the varactor tuning loop. A prototype tuning system was tested on the HYMNS II maser, however, the use of a single-line cavity coupling system resulted in poor signal-to-noise ratio and the system has not been implemented on HYMNS III. In the future, by using a two-port cavity coupling system, we expect significant improvement in signal-to-noise, and the system should prove to be a valuable diagnostic tool in addition to the electronically stabilizing the cavity resonant frequency.

Maser Storage Bulb

The storage bulb is a 6-in. diam blown fused quartz sphere. It is coated with FEP-120 teflon by standard techniques. Although the spherical bulb geometry was selected for ease of fabrication, the filling factor in the D/L = 1.6 microwave cavity is very nearly the same as the optimal value of 0.37 for an ellipsoidal bulb. The geometrical storage time is designed to be 1.25 sec. Although no systematic measurement and analysis of linewidth contributions have been made yet, the full maser transition linewidth under normal operating conditions was measured as 0.75 Hz, corresponding to a line Q of 1.9×10^9 . This value compares very favorably with those for full-size laboratory masers.

Maser Receiver

The triple-conversion maser receiver provides standard outputs at 5 and 10 MHz, phase-locked to the maser. It is equipped with a synthesizer at 5751.xxxx Hz and is adjustable in steps of 1 part in 10^{14} of the maser frequency. For tuning and diagnostic purposes, the receiver can be operated open-loop with an external 5 MHz reference signal. Down-converted 5.75 kHz maser signal as well as low-frequency beat notes are available for measurements by a counter.

Spin-Exchange Tuning and Stability Measurement

The automated operating features of HYMNS III represent a significant advance in maser design. The procedures generally encountered in maser operations can be carried out simply by depressing appropriate control buttons. An operator with detailed knowledge of maser theory is not required to carry out what used to be a long, tedious and intricate spin-exchange tuning process. In fact, when the spin-exchange tuning program is executed, the process continues automatically until specified precision is achieved, or it can be terminated at any time desired. As mentioned earlier, since a beam shutter is employed for flux modulation, long waiting periods for the dissociator to stabilize are unnecessary, and a significant savings in tune-up time is obtained.

The system controller is a Hewlett-Packard Model 9825A programmable calculator. Programs for functions such as initial setup, spin-exchange tuning, stability measurement, adjustments of applied magnetic field, varactor bias and receiver synthesizer, etc. are stored in the calculator memory and are instantly accessible. As an illustration, a typical spin-exchange tuning run is shown in Figure 2. When the program is called, interactive messages are displayed. After the desired precision expressed by the varactor bias tolerance, the low-high flux pulling, and the number of periods averaged and counter readings to produce a frequency sample are specified, the program runs automatically. The data were taken with a VLG-10 maser as an external reference, and the tuning process was completed in less than 35 min. The specified 5×10^{-3} V varactor bias tolerance corresponds to a possible tuning error of 2 parts in 10^{13} .

Preliminary stability data taken by NRL which compares the testbed with a VLG-10 are shown in Figure 3. These data, taken with the exploratory development model receiver can be improved by a factor of 3 when the maser is used with our HYMNS II laboratory breadboard receiver. We plan to continue refinement of the receiver electronics as we proceed in our technology studies.

START SPIN EXCH TUNING		END SPIN EXCH TUNING	
AUTO TUNING		VARB:	5.43789
Period Avged		LoBm	
1e5			5751.672737
Readings/Set			5751.672539
5			5751.672407
Pull Tol			5751.673091
1.4e-5			5751.672638
VARB Tol		AvF:	5751.672683
5e-3		+/-	Ø.ØØØ259
VARB:	5.18292	HiBm	
LoBm			5751.672463
	5751.655968		5751.672929
	5751.655866		5751.672506
	5751.655833		5751.672741
	5751.655598		5751.672608
	5751.655925	AvF:	5751.672649
AvF:	5751.655838	+/-	Ø.ØØØ19Ø
+/-	Ø.ØØØ144	LoBm	
HiBm			5751.672774
	5751.653649		5751.672605
	5751.653583		5751.672509
	5751.653553		5751.672506
	5751.653676	AvF:	5751.672606
	5751.653653	+/-	Ø.ØØØ11Ø
AvF:	5751.653623	Pull=	Ø.ØØØØ5
+/-	Ø.ØØØØ52	+/-	Ø.ØØØ236
LoBm			
	5751.655866	SETVB	5.437327
	5751.655896	+/-	Ø.Ø26795
	5751.656031		
	5751.65579Ø	Wd Av	5.438Ø5
	5751.655833	+/-	Ø.ØØ474
AvF:	5751.655883	Cov SETd at VB =	
+/-	Ø.ØØØØ92		5.438Ø5
Full+	-Ø.ØØ2238		
+/-	Ø.ØØØ1ØØ		
VAR	5.6826Ø		

Figure 2. Frequency comparison of the Hughes testbed maser and the SAO VLG-10 maser using the XDM receiver in the Spin Exchange Tuning mode. Only the final 10 digits are shown.

CONCLUSIONS

The operational characteristics of HYMNS III along with extensive life-test data obtained during our technology program lead us to the conclusion that a reduced size, long-lived spaceborne maser can be space qualified using existing technology. We also conclude that the maser can be made compatible with the size, weight, and power constraints of the NTS-3 spacecraft without any sacrifice in the superior stability performance exhibited by laboratory devices.

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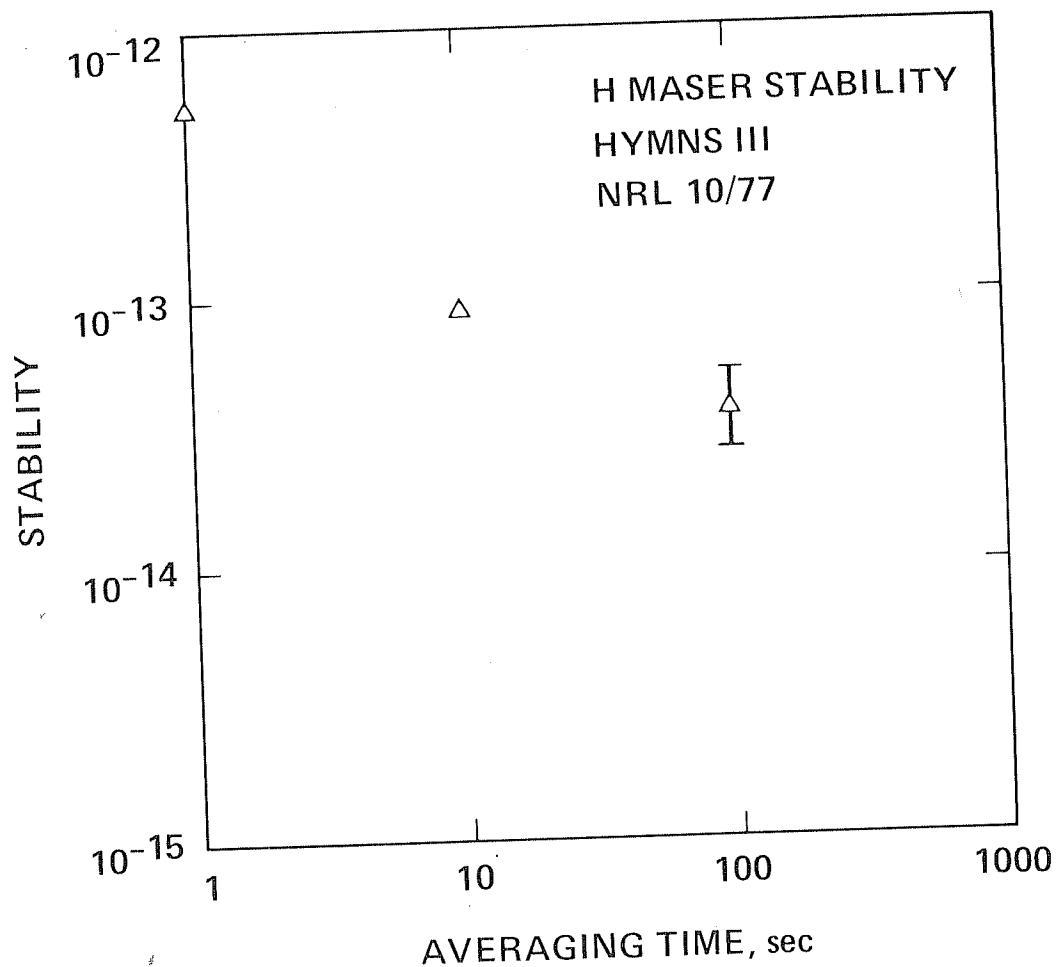


Figure 3. Preliminary stability data.

QUESTIONS AND ANSWERS

MR. WOLFGANG BAER, Ford Aerospace:

What is the final weight that you expect on the NTS-3 satellite? Where do you expect weight and size to go on the hydrogen maser designs?

MR. WANG:

I guess the weight and size, as given in the previous talk, depends on the spacecraft. Tentatively we are shooting for 100 pounds and consuming less than 100 watts with a 15-inch diameter and less than 30-inches long. But that is very tentative. Roger Easton or some people from NRL will be able to define those parameters better for you.

MR. BAER:

Do you have any feeling for where you might go in terms of weight and size in the maser development?

MR. WANG:

I feel that we can probably reduce the length and diameter by another inch or two on this active maser, but probably not much more. The weight, of course, depends on the polymer you finally select as well as the magnetic shield. Those are the heaviest components.

DR. ROBERT VESSOT, Smithsonian Astrophysical Observatory:

To give a little perspective on this, the one we flew weighed 88 pounds. It was 19-inches in diameter and about 24 inches tall. With a TE 111 cavity, allowing the same spacing between 4 layers of magnetic shields, you could make a maser fasten in a 12-inch diameter tube and be on the order of, I think, 20 inches in length. The cavity is a valid concept because it does oscillate and it will work.

In relation to the cartridge, we found that we could run the cartridge well over a year and it didn't saturate and it was a marvelous getter. But we just needed a very, very small ion pump to take care of the stuff that wasn't hydrogen, as I think you have also seen. But the function of the ion pump was also to take apart molecules that contained hydrogen as a dissociator.

I really think the future of hydrogen pumping for the maser is in the cartridge. I feel that we are contributing noise to our system by having a glow discharge. Even if it is in the millamps, that is just not a wholesome thing to have, plus the fact we have these monstrous magnets in the vac ions, the weight, the shield, and all the rest of it. I really feel that the way to go is in the cartridge.

MR. WOLF:

I agree.