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ditambah AD

AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD

Date Report : March 2018

IP → MR → WO
PO

Aircraft Registration : PK - PUZ
Aircraft Serial : 760820
Airframe Hours : 6,500.44
Airframe Cycles : 13,821

Aircraft Type : SIKORSKY S76 C++
Engine Type : Turbomeca Arriel 2S2 / Twin Engine
ENGINE 1 S/N/ TOT. TIME : 42040 / 8,747.11 / 8,178.2
ENGINE 2 S/N/ TOT. TIME : 21069 / 4,570.65 / 4,953.7

No	DGCA	TC Canada / EASA	FAA	SB Reference	Subjects	Date of Compliance	Airframe Total Time at Comp.	One Time / Recurring	Remarks	Inspector Verification
Airframe										
1			FAA AD 89-07-12 April 27, 1989	Life Limit	Tail Rotor (T/R) horn life limits - Time compliance until 12,000 Hours.	NCW			Due at 12,000 Hours	MB
2			FAA AD 98-17-15 February 1, 1999	SB 76-65-47	One-time visual inspection of the swashplate assembly uniball bearing retainer (retainer) to determine whether any retainer bolts are missing. Applicability: Model S-76A, B, and C helicopters, serial numbers prior to 760488, with swashplate assembly, part numbers 76104-08000-044, -045, -046, or 76104-08500-041 or -043, installed, certificated in any category.	N/A 29-Aug-2011		One Time	N/A by Aircraft Effectivity. Effectivity Check Carried out by Sikorsky	MB
3			FAA AD 99-01-09 March 3, 1999	SB 76-28-4	Revise the Rotorcraft Flight Manual (RFM) by inserting this AD in the Operating Limitations section of the RFM. Applicability: Model S-76C helicopters, serial numbers 760477, 760479, 760481 through 760487, 760490, 760491 and 760493, certificated in any category	N/A 29-Aug-2011		One Time	N/A by Component Effectivity. Effectivity Check Carried out by Sikorsky	MB
4			FAA AD 99-07-01 April 7, 1999	RFM SA 4047-76C-10	Insert the following into the Operating Limitations section and Performance Data section, as appropriate, of Rotorcraft Flight Manual (RFM) SA 4047-76C-10. Applicability: Model S-76C helicopters, with Turbomeca Arriel 2S1 engines with an engine Hydromechanical Unit (HMU), part number 0.292.92.822.0, 0.292.92.808.0, 0.292.92.813.0, or 0.292.92.828.0, installed, certificated in any category	N/A 29-Aug-2011		One Time	N/A by Component Effectivity. Effectivity Check Carried out by Sikorsky	MB
5			FAA AD 99-19-30 October 4, 1999	SB 76-55-16	Conduct a visual inspection of the attachment angle on each fairing. Applicability: Model S-76A, B, and C helicopters, certificated in any category.	16-Aug-17	Hrs. 5,993.46 Cyc. 12,704	1500 Hrs	Visual Inspection has been carried out. Carried out by Sikorsky Aircraft. Then ref: CAMP Sheet No: 121/ENG/S76C-E15F. Item: 4(E8) Nex Insp.: 7,493.46 Hrs	MB
6			FAA AD 2000-11-52 August 31, 2000	SB 76-65-50	To prevent main rotor blade root end pocket separation, impact with main rotor or tail rotor blades, and subsequent loss of control of the helicopter. Applicability: Model S-76 series helicopters, certificated in any category.	N/A 29-Aug-2011		One Time	N/A by Component Effectivity. Carried out by Sikorsky	MB

Prepared by

Sign.....

Checked by

Sign.....



AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD

Date Report : Desember 2017

Aircraft Registration : PK-PDA
Aircraft Serial Number : 36440
Airframe Hours : 2698.68
Airframe Cycles : 8637

Aircraft Type : BELL 412 EP
Engine Type : PT6T - 3DF / Twin Engine
ENGINE 1 S/N/ TOT. TIME : TH1161 / 905.01 / 942
ENGINE 2 S/N/ TOT. TIME : TH1160 / 905.01 / 941

date issued

due date & Remaining

No	DGCA	FAA / EASA	SB Reference	Subjects	Date of Compliance	Airframe Total Time at Comp.	One Time / Recurring	Remarks	Inspector Verification
1		FAA AD 97-07-06	ASB 412-92-57/61	Swashplate support assy	24-Aug-07	0		Found N/A to P/N installed on BHT Production Report	
2		FAA AD 97-11-04	ASB 412-94-81 Rev B	Retirement Index Number (RIN) on the Mast and Spline plate component history card(s)	24-Aug-07	0	C/W	C/W at production by BHT	
3		FAA AD 99-23-23	ASB 412-98-93	Main rotor yoke	24-Aug-07	0		Found N/A to P/N installed on BHT Production Report	
4	DGCA 00-10-006	FAA AD 2000-18-09	ASB 412-00-100	Tailboom attachment upper LH Cap angle	24-Feb-17	Hrs. 2605.4	every 100 hrs	C/W by AALP Next due at 2705.4	
5		FAA AD 2001-08-04	ASB 41105950-67A-01	Applicable for Model 412EP helicopters with an actuator, P/N 41000470, serial numbers with an "HR" prefix up to and including 10010, installed, certificated in any category.	24-Aug-07	0	C/W	C/W at production by BHT	
6	DGCA 01-05-30	FAA AD 2001-09-11	ASB 412-96-89	Tail rotor yoke fatigue failure	25-Sep-17	Hrs. 2679.29	25 FH	Ref FML R220141 Next Due at 2703.29	
7	DGCA 05-04-027	FAA AD 2001-13-01	ASB 412-00-102RA	Tail rotor counterweight bellcrank retention nut	24-Feb-17	Hrs. 2605.4	every 100 hrs	C/W by AALP Next due at 2705.4	
8		FAA AD 2001-22-14	OAS SB 22-09-97	Fire extinguishing system bottle cartridges installed to walter kidde firex bottles	24-Aug-07	0		Found N/A to P/N installed on BHT Production Report	
9		FAA AD 2005-20-38	AAI ASB AA-05005, Rev A, Dated June 27, 20015	Applicable for Model 412EP helicopters modified with Aeronautical Accessories, Inc. (AAI), Supplemental Type Certificate (STC) SH2820SO; or with AAI Parts Manufacturer Approval (PMA) reservoir assembly, part number (P/N) 212-372-050; or with adapter, P/N 212-371-002, installed, certificated in any category.	24-Aug-07	0	N/A	Found N/A to P/N Installed on BHT Production Report Reff Document Delivery	
10		FAA AD 2007-22-02	412-00-106 412CR-00-13 Rev A	Tail rotor blade (blade) forward tip weight retention block (tip block) and the aft tip closure (tip closure) for adhesive void and removing any blade with an excessive void from service. This AD also requires modifying certain blades by installing shear pins and tip closure rivets.	22-Jun-15	Hrs 2460.9		Found N/A to P/N Installed, C/W by EHI	
		FAA AD 2008-10-03 6 May 2008	ASB 412-13-155 4 February 2013	Inspection of specific Tail Rotor Blade P/N 212-010-750-ALL Superseded by FAA AD 2013-15-02	18-Apr-17	Hrs 2609.20 Cyc. 8412	25 Hrs	C/W by PAS on FML.220103 Repetitive Next Due at 2634.20 Hrs. Reff. Document Delivery Superseded by FAA AD 2013-15-02	?

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