

1. Write a MATLAB function having

As inputs

- The stiffness matrix of a laminate ($E = \begin{bmatrix} A & B \\ B & D \end{bmatrix}$)
- The in-plane resultants, the bending and torsion moments ($[N_x \ N_y \ N_{xy} \ M_x \ M_y \ M_{xy}]^T$)
- Elastic Properties
- Stacking Sequence

As outputs

- The inverted stiffness matrix
- The midplane strains and curvatures
- The strains in the principal material directions in each layer
- The stresses in the principal material directions in each layer

2. Apply your code to the laminate with the given properties, stacking sequence and loading conditions

- Elastic Properties: $E_1 = 138 \text{ GPa}$ $E_2 = 9 \text{ GPa}$ $G_{12} = 6.9 \text{ GPa}$ $\nu_{12} = 0.32$
- Stacking sequence: $[30/-60/-30/60]$
- Lamina thickness $t = 1 \text{ mm}$
- Loadings:

$$\begin{array}{lll} N_x = 2.5 \text{ kN/mm} & N_y = 1.5 \text{ kN/mm} & N_{xy} = 1 \text{ kN/mm} \\ M_x = 20 \text{ Nm/mm} & M_y = 15 \text{ Nm/mm} & M_{xy} = 10 \text{ Nm/mm} \end{array}$$

- To find the laminate stiffness matrix, use the code you developed for HW#9.