# SR22T

WITH CIRRUS PERSPECTIVE+ AVIONICS



### **Quick Reference Checklist**

for

SR22T Aircraft Serials 1460, 1471, 1473 and Subsequent with Cirrus Perspective+ Avionics Includes Serials w/ and w/o Flight Into Known Ice System (FIKI)



The procedures in this publication are abbreviated and derived from procedures in the FAA Approved Airplane Flight Manual and Pilot's Operating Handbook (POH) P/N 13772-007, Reissue A. These procedures do not supersede the procedures in the POH. In the event of conflict, the POH shall take precedence.

**PERFORMANCE** 

**EMERGENCY** 

### **Normal Procedures**

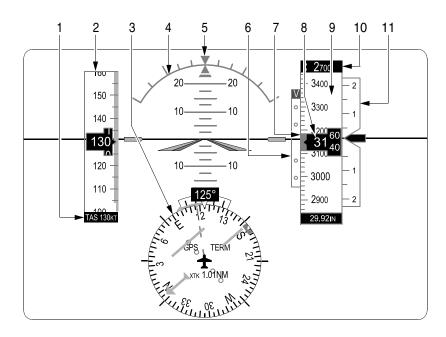
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### CIRRUS PILOT'S CHECKLIST

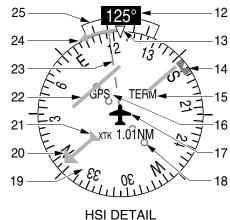
MODEL SR22T

#### **Primary Flight Display**



#### **LEGEND**

- 1. True Airspeed
- 2. Airspeed Indicator
- 3. Horizontal Situation Indicator (HSI)
- 4. Attitude Indicator
- 5. Slip/Skid Indicator
- 6. Vertical Deviation Indicator (VDI)
- 7. Selected Altitude Bug
- 8. Current Altitude
- 9. Altimeter
- 10. Selected Altitude
- 11. Vertical Speed Indicator (VSI)
- 12. Current Heading
- 13. Lubber Line
- 14. Selected Heading Bug
- 15. Flight Phase
- 16. Navigation Source
- 17. Aircraft Symbol
- 18. Course Deviation Scale
- 19. Rotating Compass Rose
- 20. Course Pointer



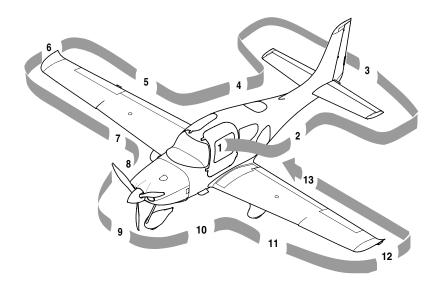
- 21. To/From Indicator
- 22. Course Deviation Indicator
- 23. Current Track Indicator
- 24. Turn Rate/Heading Trend Vector
- 25. Turn Rate Indicator

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## **Airspeeds for Normal Operation**

Takeoff Rotation:	
Normal, Flaps 50%	77 KIAS
Obstacle Clearance, Flaps 50%	85 KIAS
Enroute Climb, Flaps Up:	
Normal	120 KIAS
Best Rate of Climb, SL	103 KIAS
Best Rate of Climb, 10,000	102 KIAS
Best Angle of Climb, SL	88 KIAS
Best Angle of Climb, 10,000	88 KIAS
Landing Approach:	
Normal Approach, Flaps Up	90-95 KIAS
Normal Approach, Flaps 50%	85-90 KIAS
Normal Approach, Flaps 100%	80-85 KIAS
Short Field, Flaps 100% (V <sub>REF</sub> )	79 KIAS
Go-Around, Flaps 50%:	
Full Power	80 KIAS
Maximum Recommended Turbulent Air Pen	etration:
• 3600 lb	140 KIAS
• 2900 lb	123 KIAS
Maximum Demonstrated Crosswind Velocit	y:
Takeoff or Landing	21 Knots
Anti-Ice System (Serials w/ FIKI):	
<ul> <li>Minimum Airspeed For FIKI Conditions</li> </ul>	95 KIAS*
*Includes all phases of flight, including required for takeoff and landing.	ng approach, except as
Max Airspeed Anti-Ice System Ops 1	77 KIAS and 204 KTAS
Recommended Holding Airspeed	120 KIAS

MODEL SR22T



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## Preflight Inspection (All Serials)

1.	Cal	bin
	a.	Required Documents ON BOARD
	b.	AVIONICS SwitchOFF
	C.	BAT 2 SwitchON
	d.	PFDVERIFY ON
	e.	Essential Bus Voltage23-25 VOLTS
	f.	Flap Position LightOUT
	g.	BAT 1 SwitchON
	h.	Avionics Cooling FanAUDIBLE
	i.	Oxygen Masks/Cannulas and Hoses (if available)CHECK CONDITION
	j.	AVIONICS SwitchON
	k.	Fuel QuantityCHECK
	I.	Fuel Selector SELECT FULLER TANK
	m.	Flaps100%, CHECK LIGHT ON
		(Continued on following page)

	n.	LightsCHECK OPERATION
	0.	Stall Warning System InletUNOBSTRUCTED
	p.	Stall WarningTEST
	q.	Oxygen System (if available)ON
		(1) Quantity VERIFY ADEQUATE SUPPLY FOR FLIGHT WITH RESERVE
		(2) FlowCHECK FLOWMETER ON ALL MASKS
		(3) Oxygen System OFF
	r.	Pitot HeatON
		(1) Verify probe is hot.
	S.	Pitot HeatOFF
	t.	AVIONICS SwitchOFF
	u.	BAT 1 and BAT 2 SwitchesOFF
	V.	Alternate Static SourceNORMAL
	W.	Circuit BreakersIN
	Х.	Fire Extinguisher CHARGED AND AVAILABLE
	V.	Emergency Egress HammerAVAILABLE
	, Z.	CAPS Handle PIN REMOVED
2.	Lef	t Fuselage
	a.	Door LockUNLOCK
	b.	COM 1 Antenna (top)CONDITION AND ATTACHMENT
	C.	Transponder Antenna (underside)CONDITION AND ATTACHMENT
	d.	COM 2 Antenna (underside)CONDITION AND ATTACHMENT
	e.	Wing/Fuselage FairingCHECK
	f.	Baggage DoorCLOSED AND SECURE
	g.	Static Button
	h.	Parachute CoverSEALED AND SECURE
3.	Em	pennage
	a.	Tiedown RopeREMOVE
	b.	Horizontal and Vertical StabilizersCONDITION
	C.	Elevator and TabCONDITION AND MOVEMENT
		(Continued on following page)

	d.	RudderFREEDOM OF MOVEMENT
	e.	Rudder Trim Tab CONDITION AND SECURITY
	f.	Attachment Hinges, Bolts, and Cotter Pins SECURE
4.	Rig	ght Fuselage
	a.	Static Button
	b.	Wing/Fuselage FairingsCHECK
	c.	Door LockUNLOCK
5.	Rig	ght Wing Trailing Edge
	a.	Flap and Rub Strips (if installed)CONDITION AND SECURITY
	b.	Aileron and TabCONDITION AND MOVEMENT
	c.	Aileron Gap Seal SECURITY
	d.	Hinges, Actuation Arm, Bolts, and Cotter pins SECURE
6.	Rig	ght Wing Tip
	a.	TipATTACHMENT
	b.	Wing Tip Light and Lens CONDITION AND SECURITY
	c.	Fuel Vent (underside)UNOBSTRUCTED
7.	Rig	ht Wing Forward and Main Gear
	a.	Leading Edge and Stall StripsCONDITION
	b.	Fuel CapCHECK QUANTITY AND SECURE
	c.	Fuel Drains (2 underside)DRAIN AND SAMPLE
	d.	Wheel Fairings SECURITY, ACCUMULATION OF DEBRIS
	e.	TireCONDITION, INFLATION, AND WEAR
	f.	Wheel and BrakesFLUID LEAKS, EVIDENCE OF OVERHEATING, GENERAL CONDITION, AND SECURITY
	g.	Chocks and Tiedown RopesREMOVE
8.	No	se, Right Side
	a.	Vortex GeneratorCONDITION
	b.	Ice Inspection Light CONDITION AND SECURITY
	c.	Cowling ATTACHMENTS SECURE
	d.	Exhaust Pipe CONDITION, SECURITY, AND CLEARANCE
		(Continued on following page)

9.	Nos	se Gear, Propeller, and Spinner	
	a.	Tow BarREMOVE AND STOW	
	b.	Landing LightCONDITION	
	c.	Strut	
	d.	Wheel Fairing SECURITY, ACCUMULATION OF DEBRIS	
	e.	Wheel and Tire CONDITION, INFLATION, AND WEAR	
	f.	PropellerCONDITION (INDENTATIONS, NICKS, ETC.)	I
	g.	SpinnerCONDITION, SECURITY, AND OIL LEAKS	
	h.	Air InletsUNOBSTRUCTED	
	i.	Alternator CONDITION	
10.	Nos	se, Left Side	
	a.	Engine OilCHECK 6-8 QUARTS, LEAKS, CAP AND DOOR SECURE	
	b.	Ice Inspection Light CONDITION AND SECURITY	
	c.	Cowling ATTACHMENTS SECURE	
	d.	External PowerDOOR SECURE	
	e.	Gascolator (underside) DRAIN FOR 3 SECONDS, SAMPLE	
	f.	Vortex GeneratorCONDITION	
	g.	Exhaust PipeCONDITION, SECURITY, AND CLEARANCE	
11.	Lef	t Main Gear and Forward Wing	
	a.	Wheel Fairings SECURITY, ACCUMULATION OF DEBRIS	
	b.	TireCONDITION, INFLATION, AND WEAR	
	C.	Wheel and BrakesFLUID LEAKS, EVIDENCE OF OVERHEATING, GENERAL CONDITION, AND SECURITY	
	d.	Chocks and Tiedown RopesREMOVE	
	e.	Fuel Drains (2 underside)DRAIN AND SAMPLE	
	f.	Fuel Cap CHECK QUANTITY AND SECURE	
	g.	Leading Edge and Stall StripsCONDITION	
		(Continued on following page)	

12.	Left	Wing Tip
	a.	Fuel Vent (underside)UNOBSTRUCTED
	b.	Pitot ProbeCOVER REMOVED, UNOBSTRUCTED
	C.	Wing Tip Light and Lens CONDITION AND SECURITY
	d.	TipATTACHMEN
13.	Left	Wing Trailing Edge
	a.	Hinges, Actuation Arm, Bolts, and Cotter Pins SECURE
	b.	Aileron Gap Seal SECURITY
	c.	AileronFREEDOM OF MOVEMEN
	d.	Flap and Rub Strips (If installed)CONDITION AND SECURITY
Su	pp	lemental Preflight Inspection (Serials w/ FIKI
14.	Cab	oin
	a.	Circuit Breakers
	b.	BAT 1 SwitchON
	C.	Flaps100%
	d.	AVIONICS SwitchON
	e.	Cabin SpeakerON
	f.	Cabin DoorsCLOSE
	g.	WIND SHLD Push-Button PRESS
		(1) Verify evidence of deicing fluid from spray nozzles.
	h.	PUMP BKUP SwitchON
		(1) Metering Pump Duty CycleVERIFY CONTINUOUSLY ON
		(2) Deicing Fluid and Endurance IndicationsCHECH
	i.	PUMP BKUP SwitchOFF
	j.	ICE PROTECT System SwitchON
	k.	ICE PROTECT Mode SwitchNORM
		(1) Metering Pump Duty Cycle VERIFY 30 S ON, 90 S OFF
		(2) Deicing Fluid and Endurance IndicationsCHECH

(Continued on following page)

	I.	ICF PF	ROTECT Mode Switch	IIGH
	••		etering Pump Duty Cycle VERIFY CONTINUOU	
		_	v eicing Fluid and Endurance IndicationsCH	FCK
	m.	• •	spection Lights Switch	
	111.		rify LH and RH Operation.	Οι
		` '	,	IDC/
	n.		HEAT Switch . ON, CHECK, OFF (MAX 45 SECON	NDS)
15.	Εm	pennag		
	a.		zers Porous Panels CONDITION AND SECUI	
		` '	rify evidence of deicing fluid along length of panels evator horns.	and
	b.	Vertica	al Porous Panel CONDITION AND SECUI	RITY
		(1) Ve	rify evidence of deicing fluid along length of panel.	
16.	Rig	nt Wing	Forward and Main Gear	
	a.	Fluid T	ankVERIFY DESIRED QUAN	TITY
		(1) Fill	ler Cap CONDITION AND SECUI	RITY
		(2) Flu	uid Vent (underside wing)UNOBSTRUC	TED
	b.	Porous	Panels CONDITION AND SECUI	RITY
		(1) Ve	rify evidence of deicing fluid along length of panels	
	C.	Lift Tra	nsducer FaceplatePERCEPTIBLY	НОТ
	d.	Lift Tra	nsducer VaneVERY	НОТ
		` '	rify Stall Warning audio alert after lifting stall vane oden tooth pick or tongue depressor.	with
17.	Nos	e, Righ	nt Side	
	a.	Ice Ins	pection Light CONDITION AND SECUI	RITY
18.	Nos	lose Gear, Propeller, Spinner		
	a.	Slinge	r RingEVIDENCE OF DEICING FL	LUID
19.	Nos	e, Left	-	
	a.	Ice Ins	pection Light CONDITION AND SECUI	RITY
	b.		hield Spray Nozzles CONDITION AND SECUI	
			(Continued on following page)	, ,

**PREFLIGHT** 

20.	Lef	: Wing Forward and Main Gear
	a.	Fluid Tank VERIFY DESIRED QUANTITY
		(1) Filler Cap CONDITION AND SECURITY
		(2) Fluid Vent (underside wing) UNOBSTRUCTE
	b.	Porous Panels CONDITION AND SECURITY
		(1) Verify evidence of deicing fluid along length of panels.
21.	Lef	: Wing Tip
	a.	Pitot Probe (underside) UNOBSTRUCTE
	b.	Pitot ProbeVERY HO
22.	Cal	Din
	a.	Fluid QuantityVERIFY 5 GALLONS MINIMUM
	b.	ICE PROTECT System Switch OF
	c.	Flaps
	d.	Cabin Speaker OF
	e.	AVIONICS SwitchOF
	f	DAT 1 Switch

### **Before Engine Start**

1.	Preflight InspectionCOMPLETED
2.	Weight and BalanceVERIFY WITHIN LIMITS
3.	Emergency Equipment ON BOARD
4.	PassengersBRIEFED
5.	Seats, Seat Belts, and HarnessesADJUST AND SECURE
En	gine Start
1.	External Power (If applicable)CONNECT
2.	BrakesHOLD
3.	BAT 1 and BAT 2 SwitchesON (CHECK VOLTS)
4.	Strobe LightsON
5.	MixtureFULL RICH
6.	Power LeverFULL FORWARD
7.	Fuel PumpBOOST
8.	Propeller AreaCLEAR
9.	Power Lever OPEN 1/4 INCH
10.	Ignition Switch START (RELEASE AFTER ENGINE STARTS)
11.	Power LeverRETARD (TO MAINTAIN 1000 RPM)
12.	Oil PressureCHECK
13.	MixtureLEAN UNTIL RPM RISES TO A MAXIMUM VALUE
14.	ALT 1 and ALT 2 SwitchesON
15.	AVIONICS SwitchON
16.	Engine ParametersMONITOR

17. External Power (If applicable)......DISCONNECT

## **Cold Weather Operation: Starting**

1.	Ignition Switch	OFF
2.	PropellerHAN	D TURN SEVERAL ROTATIONS
3.	External Power (If applicable)	CONNECT
4.	Brakes	HOLD
5.	BAT 1 and BAT 2 Switches	ON (CHECK VOLTAGE)
3.	Mixture	FULL RICH
7.	Power Lever	FULL FORWARD
3.	Fuel PumpHIGH	H BOOST/PRIME, THEN BOOST
9.	Propeller Area	CLEAR
10.	Power Lever	OPEN ¼ INCH
11.	Ignition Switch START (RELI	EASE AFTER ENGINE STARTS)
12.	Power LeverRE	TARD (TO MAINTAIN 1000 RPM)
13.	Oil Pressure	CHECK
14.	ALT 1 and ALT 2 Switches	ON
15.	AVIONICS Switch	ON
16.	Engine Parameters	MONITOR
17.	External Power (If applicable)	DISCONNECT
18.	Amp Meter/Indication	CHECK
19.	Strobe Lights	ON
Вє	efore Taxiing	
1.	Flaps	UP (0%)
2.	Radios/Avionics	
3.	Cabin Heat/Defrost	AS REQUIRED
4.	Fuel Selector	SWITCH TANK
Та	xiing	
1.	Parking Brake	DISENGAGE
2.	Brakes	
 3.	HSI Orientation	
۶. 4.	Attitude Gyro	
5.	•	CHECK

#### **Before Takeoff**

1.	Doors	LATCHED
2.	CAPS Handle	VERIFY PIN REMOVED
3.	Seat Belts and Shoulder Harness	SECURE
4.	Cabin Heat/Defrost	AS REQUIRED
5.	Air Conditioner	RECIRC DISABLED
6.	Fuel Quantity	CONFIRM
7.	Fuel Selector	FULLER TANK
8.	Fuel Pump	BOOST
9.	Mixture	FULL RICH
10.	Flaps	SET 50% AND CHECK
11.	Autopilot	CHECK AND DISCONNECT
12.	Transponder	SET
13.	COM and NAV/GPS	SET
14.	Brakes	HOLD
15.	Power Lever	1700 RPM
16.	Alternator	CHECK
	a. Pitot Heat	ON
	b. Navigation Lights	ON
	c. Landing Light	ON
17.	Voltage	CHECK
18.	Pitot Heat	AS REQUIRED
19.	Navigation Lights	AS REQUIRED
20.	Landing Light	AS REQUIRED
21.	MagnetosRPM drop must not exceed 150 F differential must not exceed 75 RPM	RPM for either magneto. RPM
	a. Ignition Switch	R, NOTE RPM, THEN BOTH
	b. Ignition Switch	L, NOTE RPM, THEN BOTH
22.	Engine Parameters	CHECK
23.	Power LeverI	DLE CHECK THEN 1000 RPM
24.	Flight Instruments, HSI, and Altime	ter CHECK AND SET
	(Continued on follo	wing page)

25.	. Flight Controls	FREE AND CORRECT
26.	. Trim	SET TAKEOFF
27.	. CAS Messages	CHECK
	If icing conditions are anticipated imm w/ FIKI):	ediately after takeoff (Serials
28.	. ICE PROTECT System Switch	ON
29.	. ICE PROTECT Mode Switch	HIGH
30.	. PITOT HEAT Switch	ON
31.	. Cabin Heat	НОТ
32.	. Windshield Defrost	ON
33.	. Ice Inspection Lights	AS REQUIRED
34.	. Verify airframe is free of contamination	n immediately before takeoff.
35.	. FlapsRETRACT	ΓAS SOON AS PRACTICAL

### **Normal Takeoff**

1.	Brakes	RELEASE (STEER WITH RUDDER ONLY)
2.	Power Lever	FULL FORWARD
3.	Engine Parameters.	CHECK
4.	Elevator Control	ROTATE SMOOTHLY AT 77-80 KIAS
5.	At 90 KIAS, Flaps	UP
Sł	nort Field Take	off
1.	Flaps	50%
2.	Brakes	HOLD
3.	Power Lever	FULL FORWARD
4.	Engine Parameters.	CHECK
5.	Brakes	RELEASE (STEER WITH RUDDER ONLY)
6.	Elevator Control	ROTATE SMOOTHLY AT 77 KIAS
7.	Airspeed at Obstacle	e85 KIAS
Fι	ıll Power Climb	o: Rich of Peak Technique
1.	Oxygen	AS REQUIRED
2.	Power Lever	FULL FORWARD
3.	Mixture	MAINTAIN FUEL FLOW IN GREEN ARC
4.	Flaps	VERIFY UP
5.	Airspeed	120 KIAS
6.	Fuel Pump	BOOST
7.	Fuel Flow	MONITOR
8.	Engine Parameters	MONITOR
Cı	ruise Climb: Le	ean of Peak Technique
1.	Power Lever	REDUCE TO 30.5 IN. HG
2.	Mixture	LEAN TO CYAN TARGET OR LESS
3.	Minimum Airspeed	120 KIAS
4.	Fuel Pump	BOOST
5.	Oxygen (if available)	AS REQUIRED
	(Cor	ntinued on following page)

	a.	Oxygen Masks/Cannulas	OON
	b.	Oxygen System	ON
	c.	Flow Rate ADJUST FOR PLANNED CRUISE ALTIT	UDE
	d.	Flowmeters and Quantity MONI	TOR
6.	Cyl	rlinder Head Temperatures MONI	TOR
Ici	ing	Conditions: In Flight (Serials w/ FIKI)	
		vertent Icing Encounter OR If Icing Condition	ons
Ex	ist:	:	
1.	PIT	TOT HEAT SwitchVERIFY	ON
2.	ICE	E PROTECT System Switch	ON
3.	ICE	E PROTECT Mode SwitchNO	ORM
4.	WII	IND SHLD Push-ButtonPRESS AS REQUI	RED
5.	Мо	onitor ice accumulation.	
		ice accretions persist on protected surfaces following cle:	each
	a.	ICE PROTECT Mode H	lIGH
	If ic	ce continues accumulating on protected surfaces:	
	b.	ICE PROTECT Mode Push-Button	MAX
	If ic	ce accretions do not shed from protected surfaces:	
	C.	PUMP BKUP Switch	ON
	d.	Perform Anti-Ice System Failure Checklist.	
	e.	WIND SHLD Push-Button PRESS AS REQUI	RED
	f.	AirspeedMAINTAIN 95-177 H	<ias< td=""></ias<>
		AND LESS THAN 204 K	CTAS

#### While in Icing Conditions:

1.	Flaps	UP
2.	Ice Inspection Lights	AS REQUIRED
3.	Cabin Heat	НОТ
4.	Windshield Defrost	ON
5.	Deicing Fluid Quantity	MONITOR
	a. Ensure adequate qua	ntity to complete flight.
Αf	ter Leaving Icing Cor	nditions:
1.	Anti-Ice System	OFF
2.	Airspeed	AS FLIGHT CONDITIONS DICTATE
3.	Ice Inspection Lights	AS REQUIRED
4.	Cabin Heat	AS REQUIRED

5. Windshield Defrost ......AS REQUIRED 6. WIND SHLD Push-Button...... PRESS AS REQUIRED

### Cruise

1.	Oxygen AS REQUIRED
2.	Cruise Altitude ESTABLISHED
3.	Power LeverREDUCE TO 30.5 IN. HG OR LESS
4.	Fuel PumpAS REQUIRED
5.	Mixture ADJUST
6.	Engine Parameters MONITOR
7.	Fuel Flow and Balance MONITOR
	If any CHTs exceed 420 °F:
8.	MixtureLEAN 0.5 GPH AND MONITOR
	If icing conditions are encountered during cruise (Serials w/ FIKI):
9.	Perform Icing Conditions: In Flight Checklist.
10.	Engine PowerINCREASE TO MAINTAIN CRUISE SPEED
11.	AutopilotAS REQUIRED
	Disconnect every 30 minutes to detect any out-of-trim conditions. When disconnecting the autopilot with ice accretions on the airplane, the pilot should be alert for out-of-trim forces.

#### **Descent**

	•	40.050.005
1.	Oxygen	
2.	Altimeter	SET
3.	Cabin Heat/Defrost	AS REQUIRED
4.	Landing Light	ON
5.	Fuel System	CHECK
6.	Power Lever	AS REQUIRED
	For Rapid Descent:	
	a. Power Lever SMOOTHLY REDUC	E MAP TO 18 - 20 IN. HG
7.	Mixture	AS REQUIRED
	For Rapid Descent:	
	a. MixtureMAINTA	AIN CHTS ABOVE 240 °F
8.	Brake Pressure	CHECK
lf I	lcing Conditions Exist <i>(Serials w</i>	/ FIKI):
1.	ICE PROTECT System Switch	ON
2.	ICE PROTECT Mode Switch	
3.	Monitor ice accumulation.	
	If ice continues to accumulate:	
	a. ICE PROTECT Mode Push-Button	MAX
	If ice does not shed:	
	b. PUMP BKUP Switch	ON
	c. Perform Anti-Ice System Failure Che	ecklist.
4.	WIND SHLD Push-Button	PRESS AS REQUIRED
5.	Ice Inspection Lights	AS REQUIRED
6.	Flaps	
7.	Airspeed	
8.	Airspeed on Short Final	88 KIAS

<b>Before Landii</b>	ng
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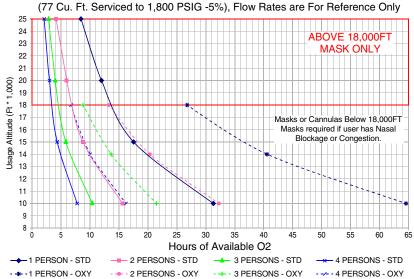
	J	
1.	Seat Belt and Shoulder Harness	SECURE
2.	Fuel Pump	BOOST
3.	Mixture	FULL RICH
4.	Flaps	AS REQUIRED
5.	Autopilot	AS REQUIRED
N	ormal Landing	
1.	Flaps	100%
2.	Airspeed	80-85 KIAS
	If Icing Conditions Exist:	
	a. Airspeed on Short Final	88 KIAS
3.	Power Lever	AS REQUIRED
	After touchdown:	
4.	Brakes	AS REQUIRED
Sł	nort Field Landing	
1.	Flaps	100%
2.	Airspeed	79 KIAS
3.	Power Lever	AS REQUIRED
	After clear of obstacles:	
4.	Power Lever	REDUCE TO IDLE
	After touchdown:	
5.	Brakes	MAXIMUM
Ва	alked Landing/Go-Around	
1.	Autopilot	DISENGAGE
2.	Power Lever	FULL FORWARD
3.	Flaps	50%
4.	Airspeed	80-85 KIAS
	After clear of obstacles:	
5.	Flaps	UP

## **After Landing**

1.	Power Lever	1000 RPM
2.	Fuel Pump	OFF OR BOOST
3.	Mixture LEAN TO OB	TAIN MAXIMUM IDLE RPM
4.	Flaps	UP
5.	Lights	AS REQUIRED
6.	Pitot Heat	OFF
	Serials w/ FIKI:	
7.	ICE PROTECT System Switch	OFF
8.	PUMP BKUP Switch	OFF
9.	Ice Inspection Lights	OFF
Sł	nutdown	
1.	Fuel Pump (if used)	OFF
2.	Power Lever	IDLE
3.	Ignition Switch	CYCLE
4.	Mixture	CUTOFF
5.	All Switches	OFF
6.	Magnetos	OFF
7.	ELT	TRANSMIT LIGHT OUT
Ω	Chocks Tie-downs Pitot Covers	AS REQUIRED

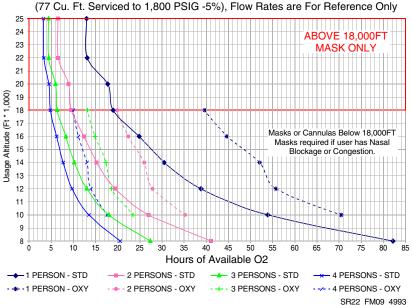
### Oxygen Duration (Precise Flight System)

Oxygen System Usage Duration - A4 Flowmeter (STD) Standard Cannula/Mask - (OXY) Oxymiser Cannual Tested Values at Altitude for flow rates STPD



Oxygen System Usage Duration - A5 Flowmeter
(STD) Standard Cannula/Mask - (OXY) Oxymiser Cannual
Tested Values at Altitude for flow rates STPD

Et. Serviced to 1 800 PSIG -5%). Flow Rates are For Reference C



#### Performance

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#### • Note •

Aircraft with optional Air Conditioning System; Brake Horsepower is reduced by approximately 6 BHP.

## CIRRUS PILOT'S CHECKLIST

MODEL SR22T

### Takeoff Distance: 3600 LB (1633 kg)

Weight: 3600 LB (1633 kg)

Approx. Speed at Liftoff: 80 KIAS

Speed over 50 ft Obstacle: 85 KIAS

Flaps: 50% Power: Full Throttle Mixture: Full Rich

Runway: Dry, Paved, Level

Headwind: Subtract 10% for each 12 knots

headwind.

Tailwind: Add 10% for each 2 knots tailwind up

to 10 knots.

Dry Grass: Add 15% of ground roll to

distances.

Wet Grass: Add 30% to ground roll. Runway Slope: Refer to POH.

Air Conditioner: Add 100 ft to ground roll and 150 ft to distance over 50 ft obstacle if Air Conditioner if ON during takeoff.

PRESS	DISTANCE		TEMPERATURE ~°C					
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1352	1461	1574	1692	1814	1941	1517
	Total	1865	2007	2154	2307	2465	2629	2080
1000	Grnd Roll	1443	1559	1680	1805	1936	2071	1595
	Total	1980	2131	2288	2450	2618	2792	2178
2000	Grnd Roll	1540	1664	1793	1927	2066	2210	1677
	Total	2104	2264	2431	2603	2782	2967	2281
3000	Grnd Roll	1645	1777	1914	2058	2206	2361	1764
	Total	2236	2407	2584	2767	2958	3154	2390
4000	Grnd Roll	1757	1898	2045	2198	2357	2522	1856
	Total	2378	2559	2748	2943	3146	3355	2505
5000	Grnd Roll	1878	2029	2186	2350	2520	2696	1954
	Total	2530	2723	2924	3132	3347	3570	2627
6000	Grnd Roll	2008	2170	2338	2513	2694	2883	2058
	Total	2693	2899	3113	3334	3564	3802	2756
7000	Grnd Roll	2149	2322	2501	2688	2883	3084	2168
	Total	2868	3088	3315	3552	3796	4050	2892
8000	Grnd Roll	2300	2485	2678	2878	3086	3302	2284
	Total	3056	3290	3533	3785	4046	4316	3036
9000	Grnd Roll	2463	2661	2868	3082	3305	3536	2408
	Total	3258	3508	3767	4036	4314	4603	3188
10000	Grnd Roll	2640	2852	3073	3303	3541	3789	2540
	Total	3476	3742	4019	4306	4603	4911	3350

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#### Takeoff Distance: 2900 LB (1315 kg)

Weight: 2900 LB (1315 kg) Approx. Speed at Liftoff: 67 KIAS Speed over 50 ft Obstacle: 72 KIAS

**Flaps:** 50%

Power: Full Throttle Mixture: Full Rich Runway: Dry, Paved, Level Headwind: Subtract 10% for each 12 knots

Tailwind: Add 10% for each 2 knots tailwind up

to 10 knots.

Dry Grass: Add 15% of ground roll to

distances.

Wet Grass: Add 30% to ground roll. Runway Slope: Refer to POH.

Air Conditioner: Add 100 ft to ground roll and 150 ft to distance over 50 ft obstacle if Air Conditioner if ON during takeoff.

PRESS	PRESS DISTANCE TEMPERATURE ~°C							
ALT				. –				
FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	485	524	564	606	650	695	544
	Total	766	823	882	944	1007	1073	852
1000	Grnd Roll	517	559	602	647	694	742	571
	Total	812	872	935	1000	1068	1138	891
2000	Grnd Roll	552	596	642	690	740	792	601
	Total	861	925	992	1061	1133	1207	932
3000	Grnd Roll	589	637	686	737	791	846	632
	Total	914	982	1053	1126	1202	1281	975
4000	Grnd Roll	630	680	733	788	845	904	665
	Total	970	1043	1118	1196	1277	1360	1021
5000	Grnd Roll	673	727	783	842	903	966	700
	Total	1030	1108	1188	1271	1357	1446	1069
6000	Grnd Roll	720	778	838	900	965	1033	737
	Total	1095	1177	1262	1351	1442	1537	1120
7000	Grnd Roll	770	832	896	963	1033	1105	777
	Total	1164	1252	1343	1437	1534	1634	1174
8000	Grnd Roll	824	890	959	1031	1106	1183	819
	Total	1239	1332	1428	1529	1632	1739	1231
9000	Grnd Roll	883	954	1028	1104	1184	1267	863
	Total	1318	1418	1521	1627	1738	1852	1291
10000	Grnd Roll	946	1022	1101	1183	1269	1358	910
	Total	1404	1510	1620	1733	1851	1973	1354

### CIRRUS PILOT'S CHECKLIST

MODEL SR22T

#### **Cruise Performance**

#### Conditions:

• Note •

Subtract 10 KTAS if nose wheel pant and fairing removed. Lower KTAS by 10% if nose and main wheel pants and fairings are removed.

Serials w/ Air Conditioning System: Cruise performance is reduced by 2 knots. For maximum performance, turn air conditioner off.

Serials w/ Enhanced Vision System: Cruise performance is reduced by up to 1 knot.

To use the chart below, set the desired % power with the throttle. Lean the engine by use of the mixture lever until the fuel flow matches that of the target fuel flow indication. Once the mixture is leaned to the target fuel flow, it may be necessary to adjust the power lever (MP) to achieve the desired % Power. Engine temperature and ambient air temperature may require the use of a leaner than target fuel flow for engine temperature management reasons. If the selected lean-of-peak fuel flow is too lean, the engine may run rough due to lean misfire. If a lean misfire is experienced, use a lower % Power (lower MP), and lean the engine to the new indicated target fuel flow. The values shown in gray may not be achievable for engine temperature management reasons.

CRUISE	PERFORMA	ANCE	ISA -	30 °C	ISA		ISA +30 °C		
Altitude (ft MSL)	Power (% of 315)	FF (GPH)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	
2000	85%	18.3	164	9.0	170	9.3	176	9.6	
	75%	16.4	157	9.6	162	9.9	167	10.2	
	65%	14.6	148	10.2	154	10.5	158	10.8	
	55%	12.7	138	10.9	143	11.2	147	11.5	
4000	85%	18.3	168	9.1	174	9.5	179	9.8	
	75%	16.4	160	9.7	165	10.1	170	10.4	
	65%	14.6	151	10.3	156	10.7	161	11.0	
	55%	12.7	140	11.0	145	11.4	149	11.8	
6000	85%	18.3	171	9.3	177	9.7	183	10.0	
	75%	16.4	163	9.9	168	10.2	174	10.6	
	65%	14.6	153	10.5	159	10.9	163	11.2	
	55%	12.7	143	11.2	147	11.6	152	11.9	
8000	85%	18.3	174	9.5	180	9.8	186	10.2	
	75%	16.4	166	10.1	171	10.4	177	10.8	
	65%	14.6	156	10.7	161	11.1	166	11.4	
	55%	12.7	145	11.4	150	11.8	154	12.1	
10000	85%	18.3	177	9.7	184	10.0	190	10.4	
	75%	16.4	169	10.3	175	10.6	180	11.0	
	65%	14.6	159	10.9	164	11.3	169	11.6	
	55%	12.7	148	11.6	152	12.0	157	12.3	

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CRUISE	PERFORM	ANCE	ISA -	30 °C	IS	SA .	ISA +	ISA +30 °C	
Altitude (ft MSL)	Power (% of 315)	FF (GPH)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	
12000	85%	18.3	181	9.9	187	10.2	193	10.6	
	75%	16.4	172	10.4	178	10.8	183	11.2	
	65%	14.6	162	11.1	167	11.5	172	11.8	
	55%	12.7	150	11.8	155	12.2	159	12.5	
14000	85%	18.3	184	10.0	191	10.4	197	10.8	
	75%	16.4	175	10.6	181	11.0	187	11.4	
	65%	14.6	165	11.3	170	11.7	175	12.0	
	55%	12.7	153	12.0	157	12.4	162	12.7	
16000	85%	18.3	187	10.2	194	10.6	201	11.0	
	75%	16.4	178	10.8	185	11.2	191	11.6	
	65%	14.6	167	11.5	173	11.9	179	12.2	
	55%	12.7	155	12.2	160	12.6	164	12.9	
18000	85%	18.3	191	10.4	198	10.8	205	11.0	
	75%	16.4	181	11.0	188	11.4	194	11.8	
	65%	14.6	171	11.7	176	12.1	182	12.5	
	55%	12.7	158	12.4	162	12.8	167	13.1	
20000	85%	18.3	195	10.6	202	11.0	209	11.4	
	80%	17.4	190	10.9	197	11.3	204	11.7	
	75%	16.4	185	11.2	192	11.7	198	12.0	
	65%	14.6	174	11.9	180	12.3	185	12.7	
	55%	12.7	160	12.6	165	13.0	169	13.3	
22000	85%	18.3	199	10.8	206	11.3	213	11.6	
	80%	17.4	194	11.1	201	11.6	208	12.0	
	75%	16.4	188	11.5	195	11.9	202	12.3	
	65%	14.6	177	12.1	183	12.5	188	12.9	
	55%	12.7	163	12.8	168	13.2	172	13.5	
24000	85%	18.3	202	11.1	210	11.5	218	11.9	
	80%	17.4	197	11.4	205	11.8	212	12.2	
	75%	16.4	192	11.7	199	12.1	206	12.5	
	65%	14.6	180	12.3	186	12.8	191	13.1	
	55%	12.7	165	13.0	170	13.4	174	13.7	
25000	85%	18.3	204	11.2	213	11.6	220	12.0	
	80%	17.4	199	11.5	207	11.9	214	12.3	
	75%	16.4	194	11.8	201	12.2	208	12.6	
	65%	14.6	181	12.4	188	12.9	193	13.2	
	55%	12.7	166	13.1	171	13.5	176	13.8	

### CIRRUS PILOT'S CHECKLIST

MODEL SR22T

### **Landing Distance - Flaps 100%**

Weight: 3600 LB (1633 kg)

Speed over 50 ft Obstacle: 79 KIAS

Flaps: 100% Power: Idle

Runway: Dry, Paved, Level

Headwind: Subtract 10% for each 13

knots headwind.

Tailwind: Add 10% for each 2 knots

tailwind up to 10 knots.

Runway Slope: Refer to POH. Dry Grass: Add 20% to ground roll. Wet Grass: Add 60% to ground roll.

PRESS	DISTANCE				ERATUF		ground	
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1117	1158	1198	1239	1280	1321	1178
	Total	2447	2505	2565	2625	2685	2747	2535
1000	Grnd Roll	1158	1200	1243	1285	1327	1370	1213
	Total	2506	2567	2630	2693	2757	2821	2585
2000	Grnd Roll	1201	1245	1289	1333	1377	1421	1250
	Total	2568	2633	2699	2765	2832	2900	2636
3000	Grnd Roll	1246	1292	1337	1383	1428	1474	1287
	Total	2635	2702	2771	2841	2911	2983	2691
4000	Grnd Roll	1293	1340	1388	1435	1482	1530	1326
	Total	2705	2776	2848	2922	2996	3070	2748
5000	Grnd Roll	1342	1391	1440	1489	1539	1588	1367
	Total	2779	2854	2930	3007	3085	3163	2808
6000	Grnd Roll	1393	1444	1495	1546	1598	1649	1409
	Total	2857	2936	3016	3097	3179	3261	2871
7000	Grnd Roll	1447	1500	1553	1606	1659	1712	1453
	Total	2941	3024	3108	3193	3279	3365	2937
8000	Grnd Roll	1503	1558	1613	1668	1724	1779	1499
	Total	3029	3116	3205	3294	3384	3475	3006
9000	Grnd Roll	1562	1619	1677	1734	1791	1848	1546
	Total	3122	3214	3307	3401	3496	3592	3079
10000	Grnd Roll	1624	1683	1743	1802	1862	1921	1595
	Total	3221	3318	3416	3515	3614	3715	3155

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### **Landing Distance - Flaps 50%**

Weight: 3600 LB (1633 kg)

Speed over 50 ft Obstacle: 87 KIAS

**Flaps:** 50% Power: Idle

Runway: Dry, Paved, Level

Headwind: Subtract 10% for each 13

knots headwind.

Tailwind: Add 10% for each 2 knots

tailwind up to 10 knots.

Runway Slope: Refer to POH. Dry Grass: Add 20% to ground roll. Wet Grass: Add 60% to ground roll.

PRESS	DISTANCE			TEMP	ERATUF	RE ~°C		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1166	1209	1251	1294	1337	1379	1230
	Total	2681	2745	2810	2875	2942	3010	2777
1000	Grnd Roll	1209	1253	1298	1342	1386	1430	1267
	Total	2745	2813	2881	2950	3020	3091	2833
2000	Grnd Roll	1254	1300	1346	1392	1438	1484	1305
	Total	2814	2885	2957	3029	3103	3178	2892
3000	Grnd Roll	1301	1349	1396	1444	1491	1539	1344
	Total	2886	2961	3037	3113	3191	3269	2954
4000	Grnd Roll	1350	1399	1449	1498	1548	1597	1385
	Total	2963	3042	3121	3202	3283	3366	3019
5000	Grnd Roll	1401	1453	1504	1555	1607	1658	1427
	Total	3045	3127	3211	3296	3382	3468	3087
6000	Grnd Roll	1455	1508	1561	1615	1668	1721	1472
	Total	3131	3218	3306	3395	3485	3576	3158
7000	Grnd Roll	1511	1566	1622	1677	1732	1788	1517
	Total	3223	3314	3407	3501	3595	3691	3233
8000	Grnd Roll	1570	1627	1685	1742	1800	1857	1565
	Total	3320	3416	3514	3612	3712	3812	3312
9000	Grnd Roll	1631	1691	1751	1810	1870	1930	1614
	Total	3423	3524	3627	3731	3835	3941	3395
10000	Grnd Roll	1695	1758	1820	1882	1944	2006	1666
	Total	3532	3639	3747	3856	3966	4077	3481

### CIRRUS PILOT'S CHECKLIST

MODEL SR22T

### **Landing Distance - Flaps 0%**

Weight: 3600 LB (1633 kg)

Speed over 50 ft Obstacle: 94 KIAS

Flaps: 0% Power: Idle

Runway: Dry, Paved, Level

Headwind: Subtract 10% for each 13

knots headwind.

Tailwind: Add 10% for each 2 knots

tailwind up to 10 knots.

Runway Slope: Refer to POH. Dry Grass: Add 20% to ground roll. Wet Grass: Add 60% to ground roll.

PRESS				ERATUF		ground		
ALT FT	FT	0	10	20	30	40	50	ISA
SL	Grnd Roll	1365	1415	1465	1515	1565	1615	1440
	Total	3165	3241	3319	3398	3478	3558	3280
1000	Grnd Roll	1415	1467	1519	1571	1623	1675	1483
	Total	3242	3323	3404	3487	3571	3656	3347
2000	Grnd Roll	1468	1522	1576	1629	1683	1737	1527
	Total	3324	3409	3495	3582	3670	3759	3418
3000	Grnd Roll	1523	1579	1635	1690	1746	1802	1574
	Total	3411	3500	3590	3682	3775	3868	3491
4000	Grnd Roll	1581	1638	1696	1754	1812	1870	1621
	Total	3503	3597	3692	3788	3885	3984	3569
5000	Grnd Roll	1641	1701	1761	1821	1881	1941	1671
	Total	3600	3699	3799	3900	4003	4106	3650
6000	Grnd Roll	1703	1766	1828	1890	1953	2015	1723
	Total	3703	3807	3913	4019	4127	4236	3736
7000	Grnd Roll	1769	1834	1899	1963	2028	2093	1776
	Total	3813	3922	4033	4145	4258	4373	3825
8000	Grnd Roll	1838	1905	1972	2040	2107	2174	1832
	Total	3929	4044	4161	4279	4398	4518	3919
9000	Grnd Roll	1910	1980	2049	2119	2189	2259	1890
	Total	4052	4173	4296	4420	4545	4671	4018
10000	Grnd Roll	1985	2058	2130	2203	2276	2348	1950
	Total	4183	4310	4439	4569	4701	4833	4122

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Performance of the airplane with the TCM TSIO-550-K turbocharged engine installed is equal to or better than the performance described in the following tables.

### **Stall Speeds with Ice Accumulation** (Serials w/ FIKI)

#### Conditions:

• Power......Idle Bank Angle .......Noted

• Note •

Altitude loss during wings level stall may be 600 feet or more.

KIAS values may not be accurate at stall.

Weight	Bank	STALL SPEEDS					
	Angle	Flap	s 0%				
		Ful	l Up	Flaps	50%		
			1				
LB	Deg	KIAS	KCAS	KIAS	KCAS		
3600	0	77	76	72	69		
	15	79	77	73	70		
Most	30	83	82	75	74		
FWD	45	91	90	82	82		
CG	60	107	107	95	98		
3600	0	77	76	72	69		
	15	79	77	73	70		
Most	30	83	82	75	74		
AFT	45	91	90	82	82		
CG	60	107	107	95	98		

### **Cruise Performance with Ice Accumulation** (Serials w/ FIKI)

#### **Conditions:**

• Note •

Aircraft with optional Air Conditioning System: Cruise performance is reduced by 2 knots. For maximum performance, the air conditioner should be off.

CRUISE	PERFORMA	ANCE	ISA -	30 °C	ISA		ISA +	ISA +30 °C	
Altitude (ft MSL)	Power (% of 315)	FF (GPH)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	
2000	85%	18.3	149	8.1					
	75%	16.4	138	8.4					
	65%	14.6	123	8.4					
	55%	12.7	104	8.2					
4000	85%	18.3	151	8.2					
	75%	16.4	139	8.4					
	65%	14.6	123	8.4					
	55%	12.7	104	8.2					
6000	85%	18.3	152	8.3					
	75%	16.4	140	8.5					
	65%	14.6	123	8.5					
	55%	12.7	103	8.1					
8000	85%	18.3	154	8.4	162	8.8			
	75%	16.4	141	8.5	148	9.0			
	65%	14.6	124	8.5	131	9.0			
10000	85%	18.3	156	8.5	164	8.9			
	75%	16.4	141	8.6	149	9.1			
	65%	14.6	124	8.5	131	9.0			
12000	85%	18.3	157	8.6	165	9.0			
	75%	16.4	142	8.7	150	9.1			
	65%	14.6	123	8.4	131	9.0			
14000	85%	18.3	158	8.6	167	9.1			
	75%	16.4	142	8.7	151	9.2			
	65%	14.6	123	8.4	131	9.0			

CRUISE PERFORMANCE			ISA -	30 °C	ISA		ISA +30 °C	
Altitude (ft MSL)	Power (% of 315)	FF (GPH)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)	TAS (KTAS)	Econ (nm/gal)
16000	85%	18.3	165	9.0	168	9.2		
	75%	16.4	150	9.1	151	9.2		
	65%	14.6	131	9.0	131	9.0		
18000	85%	18.3	167	9.1	169	9.2		
	75%	16.4	151	9.2	151	9.2		
	65%	14.6	131	9.0				
20000	85%	18.3	168	9.2	170	9.3		
	80%	17.4	160	9.2	161	9.3		
	75%	16.4	151	9.2	151	9.2		
	65%	14.6	131	8.9				
22000	85%	18.3	169	9.2	171	9.3		
	80%	17.4	161	9.3	162	9.3		
	75%	16.4	151	9.2	151	9.2		

### CIRRUS PILOT'S CHECKLIST

MODEL SR22T

### **Landing Distance with Ice Accumulation** (Serials w/ FIKI)

• Note •

Sloped Runway - Increase distances by 27% of ground roll distance for each 1% downslope. Decrease distances by 9% of ground roll distance for each 1% upslope.

WEIGHT: 3600 LB

Speed over 50 ft Obstacle: 88 KIAS

Flaps: 50%

Power: Smooth power reduction from obstacle to

idle at touchdown.

Runway: Dry, Paved, Level

Headwind: Subtract 10% for each 13

knots headwind.

Tailwind: Add 10% for each 2 knots tailwind up to 10 knots.

Runway Slope: Refer to POH. Dry Grass: Add 20% to ground roll.

	,,,	Wet Grass: Add 60% to ground roll.							
PRESS ALT	DISTANCE		TEMPE	RATURE ~°C					
FT	FT	-20	-10	0	5	ISA			
SL	Grnd Roll	1356	1409	1463	1489				
	Total	2833	2908	2984	3022				
1000	Grnd Roll	1406	1461	1517	1544				
	Total	2903	2981	3061	3101				
2000	Grnd Roll	1458	1516	1573	1602				
	Total	2977	3059	3143	3185				
3000	Grnd Roll	1513	1572	1632	1662				
	Total	3055	3142	3229	3274				
4000	Grnd Roll	1570	1632	1694	1725				
	Total	3138	3229	3321	3367				
5000	Grnd Roll	1629	1694	1758	1790				
	Total	3225	3321	3418	3466				
6000	Grnd Roll	1692	1758	1825	1859				
	Total	3318	3418	3520	3571				
7000	Grnd Roll	1757	1826	1896	1930				
	Total	3416	3522	3628	3682				
8000	Grnd Roll	1825	1897	1969	2005	1963			
	Total	3520	3631	3743	3800	3583			
9000	Grnd Roll	1896	1971	2046	2084	2025			
	Total	3630	3746	3864	3924	3656			
10000	Grnd Roll	1971	2049	2127	2166	2089			
	Total	3746	3869	3993	4055	3733			

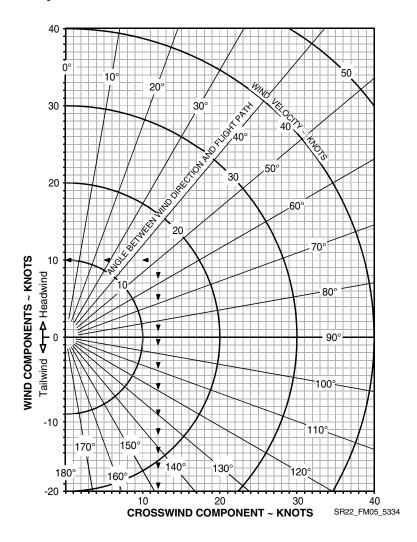
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## **Wind Components**

Conditions:	Example: (See Chart ▶ ▶ ▶)
Runway Heading 10°	Wind/Flight Path Angle50°
Wind Direction 60°	Crosswind Component 12 Knots
Wind Velocity15 Knots	Headwind Component 10 Knots

• Note •

The maximum demonstrated crosswind is 21 knots. Value not considered limiting.



## CIRRUS PILOT'S CHECKLIST

MODEL SR22T

## Weight and Balance

## **Loading Calculations**

For Moment/1000, refer to Loading Data table on following page.

	Description	Weight	Moment/1000
1.	Basic Empty Weight Includes unusable fuel and full oil		
2.	Front Seats Occupants Pilot and Passenger (total)		
3.	Rear Seats Occupants		
4.	Baggage Area 130 lb maximum		
5.	Zero Fuel Condition Weight Subtotal items 1 thru 4 3400 lb maximum		
6.	Fuel Loading 92 Gallon @ 6.0 lb/gal. maximum		
7.	Ramp Condition Weight Subtotal items 5 and 6		
8.	Fuel for start, taxi, and runup Normally 9 lb at avg. moment of 1394 (1.4)	-	-
9.	Takeoff Condition Weight Subtract Item 8 from item 7		

#### Calculation Instructions

- Enter the current basic empty weight and moment from the aircraft's Weight and Balance Record.
- 2. Enter the total weight and moment/1000 for the front seat occupants from the adjacent Loading Data Table.
- 3. Enter the total weight and moment/1000 for the rear seat occupants from the adjacent Loading Data Table.
- 4. Enter the total weight and moment/1000 for the baggage from the adjacent Loading Data Table.
- 5. If desired, subtotal the weight and moment/1000 entries from steps 1 4.
- 6. Enter the weight and moment/1000 of usable fuel loaded on the airplane.
- 7. Subtotal the weight and moment/1000.
- 8. Enter values for typical start, taxi, and run-up operations of 9 pounds at an average moment/1000 of 1.394.
- Subtract step 8 weight and moment/1000 from the Ramp Condition Weight to determine the Takeoff Condition Weight and moment/1000.
  - a. Verify Takeoff Weight does not exceed the 3600 pounds.
  - b. Verify Moment/1000 falls between the interpolated minimum and maximum values listed on the adjacent Moment Limits Table.

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## **Loading Data**

Use this table to determine the Moment/1000.

Weight	Fwd Pass	Aft Pass	Baggage	Fuel	Weight	Fwd Pass	Aft Pass	Fuel
LB	FS 143.5	FS 180.0	FS 208.0	FS 154.9	LB	FS 143.5	FS 180.0	FS 154.9
20	2.9	3.6	4.2	3.1	300	43.1	54.0	46.5
40	5.7	7.2	8.3	6.2	320	45.9	57.6	49.6
60	8.6	10.8	12.5	9.3	340	48.8	61.2	52.7
80	11.5	14.4	16.6	12.4	360	51.7	64.8	55.8
100	14.4	18.0	20.8	15.5	380	54.5	68.4	58.9
120	17.2	21.6	25.0	18.6	400	57.4	72.0	62.0
140	20.1	25.2	27.04*	21.7	420	60.3	75.6	65.1
160	23.0	28.8		24.8	440	63.1	79.2	68.2
180	25.8	32.4		27.9	460			71.3
200	28.7	36.0		31.0	480			74.4
220	31.6	39.6		34.1	500			77.5
240	34.4	43.2		37.2	520			80.5
260	37.3	46.8	·	40.3	552**			85.5
280	40.2	50.4		43.4	*130 lb Maximum **92 U.S. Gallons Usable			

### **Moment Limits**

Use this table to determine if Loading Calculations are within limits.

Weight	Moment/1000		Weight	Momen	t/1000
LB	Minimum	Maximum	LB	Minimum	Maximum
2200	304	326	2950	414	437
2250	311	333	3000	422	444
2300	318	341	3050	430	452
2350	325	348	3100	438	459
2400	332	356	3150	445	467
2450	340	363	3200	453	474
2500	347	370	3250	461	481
2550	354	378	3300	469	489
2600	361	385	3350	477	496
2650	368	393	*3400	484	504
2700	375	400	3450	494	511
2750	383	407	3500	501	519
2800	391	415	3550	508	526
2850	399	422	3600	515	533
2900	407	430			

\*NOTE: Maximum zero fuel weight

## CIRRUS PILOT'S CHECKLIST

MODEL SR22T

## **Temperature Conversion**

To convert from Celsius (°C) to Fahrenheit (°F), find, in the shaded columns, the number representing the temperature value (°C) to be converted. The equivalent Fahrenheit temperature is read to the right.

► EXAMPLE: 38 °C = 100 °F.

To convert from Fahrenheit (°F) to Celsius (°C), find in the shaded columns area, the number representing the temperature value (°F) to be converted. The equivalent Celsius temperature is read to the left.

► EXAMPLE: 38 °F = 3 °C.

Temp to Convert			Temp to Convert		Ter	np to Con	vert	
	°C or °F			°C or °F			°C or °F	
°C	<u> </u>	°F	°C	<b>▲ ト</b>	°F	°C	<b>◆ </b>	°F
-50	-58	-72	-17	2	36	17	62	144
-49	-56	-69	-16	4	39	18	64	147
-48	-54	-65	-14	6	43	19	66	151
-47	-52	-62	-13	8	46	20	68	154
-46	-50	-58	-12	10	50	21	70	158
-44	-48	-54	-11	12	54	22	72	162
-43	-46	-51	-10	14	57	23	74	165
-42	-44	-47	-9	16	61	24	76	169
-41	-42	-44	-8	18	64	26	78	172
-40	-40	-40	-7	20	68	27	80	176
-39	-38	-36	-6	22	72	28	82	180
-38	-36	-33	-4	24	75	29	84	183
-37	-34	-29	-3	26	79	30	86	187
-36	-32	-26	-2	28	82	31	88	190
-34	-30	-22	-1	30	86	32	90	194
-33	-28	-18	0	32	90	33	92	198
-32	-26	-15	1	34	93	34	94	201
-31	-24	-11	2	36	97	36	96	205
-30	-22	-8	3	38	100	37	98	208
-29	-20	-4	4	40	104	38	100	212
-28	-18	0	6	42	108	39	102	216
-27	-16	3	7	44	111	40	104	219
-26	-14	7	8	46	115	41	106	223
-24	-12	10	9	48	118	42	108	226
-23	-10	14	10	50	122	43	110	230
-22	-8	18	11	52	126	44	112	234
-21	-6	21	12	54	129	46	114	237
-20	-4	25	13	56	133	47	116	241
-19	-2	28	14	58	136	48	118	244
-18	0	32	16	60	140	49	120	248

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## **Flight Environment**

Inadvertent Icing	g Encounter	(Serials w/o	<b>FIKI</b>
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1.	Pitot HeatON
2.	Exit icing conditions. Turn back or change altitude.
3.	Cabin HeatMAXIMUM
4.	Windshield Defrost FULL OPEN
lna	advertent IMC Encounter
1.	Airplane Control ESTABLISH STRAIGHT AND LEVEL FLIGHT
2.	Autopilot ENGAGE TO HOLD HEADING AND ALTITUDE
3.	HeadingRESET TO INITIATE 180° TURN
Do	oor Open In Flight
1	Airplana Control MAINITAIN

## **Abnormal Landings**

## **Landing With Failed Brakes**

#### One brake inoperative

- Land on the side of runway corresponding to the inoperative brake.
- 2. Maintain directional control using rudder and working brake.

#### Both brakes inoperative

- Divert to the longest, widest runway with the most direct headwind.
- 2. Land on downwind side of the runway.
- 3. Use the rudder for obstacle avoidance.
- 4. Perform Emergency Engine Shutdown on Ground Checklist.

## **Landing With Flat Tire**

#### Main Gear

- 1. Land on the side of the runway corresponding to the good tire.
- 2. Maintain directional control with the brakes and rudder.
- Do not taxi. Stop the airplane and perform a normal engine 3. shutdown.

#### Nose Gear

- 1. Land in the center of the runway.
- 2. Hold the nosewheel off the ground as long as possible.
- 3. Do not taxi. Stop the airplane and perform a normal engine shutdown.

## **Engine System**

### Low Idle Oil Pressure

**OIL PRESS Caution** 

If In-Flight.....LAND AS SOON AS PRACTICABLE

### **High Oil Temperature**

**OIL TEMP Caution** 

#### **OIL TEMP**

- 1. Power......REDUCE AS MUCH AS PRACTICAL
- 2. Airspeed ......INCREASE
- 3. Mixture..... ADJUST FUEL FLOW TO TOP OF GREEN ARC
- 4. Oil Temperature Gauge ...... MONITOR

### **Manifold Pressure High**

MAN PRESSURE Caution

#### MAN PRESSURE

- 1. Power Lever......REDUCE TO LESS THAN 36.5"
- Flight......CONTINUE If noticeable surging is present:
- 3. Perform Overboost / Pressure Relief Valve Emergency Checklist.

## **Start Engaged**

### START ENGAGED Caution

#### START ENGAGED

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1.	Ignition Switch DISENGAGE PRIOR TO 10 SECONDS
2.	Wait 20 seconds before next start attempt.
	If starter does not disengage (relay or solenoid failure):
3.	BAT 1 Switch OFF
4.	Engine SHUTDOWN
5.	STARTER Circuit BreakerPULL
In-	Flight
1.	Ignition Switch ENSURE NOT STUCK IN START
2.	STARTER Circuit BreakerPULL
3.	FlightCONTINUE

## **Alternate Air Door Open**

#### **ALT AIR OPEN Caution**

#### **ALT AIR OPEN**

1.	Manifold PressureCHECK
	If environment suspect as cause (icing or visible debris):
2.	Flight Conditions
3.	Power REDUCE TO 30.5 IN.HG WHEN PRACTICAL
4.	FlightCONTINUE

## **Fuel System**

## Low Fuel Quantity

#### **FUEL LOW TOTAL Caution**

#### **FUEL LOW TOTAL**

	If totalized value differs significantly from sensed quantity:
2.	Totalized Fuel QuantityCHECK
1.	Fuel Quantity GaugesCHECK

- a. Initial Fuel Value......VERIFY AND CORRECT If message persists:
- a. Land as soon as practicable.

#### **Fuel Imbalance**

#### **FUEL IMBALANCE Caution**

#### **FUEL IMBALANCE**

- Fuel Quantity Gauges......CHECK
- 2. Fuel Pump ......BOOST If HIGH BOOST already in use for vapor suppression, pump should be left in this position for tank switch.
- Fuel Selector ...... SELECT FULLER TANK 3.
- 4. Fuel Pump ......AS REQUIRED

After switching tanks, message will remain until sensed imbalance is less than 10 gallons.

## FUEL IMBALANCE Advisory

#### **FUEL IMBALANCE**

- Fuel Quantity Gauges......CHECK
- 2. Fuel Pump ......BOOST If HIGH BOOST already in use for vapor suppression, pump should be left in this position for tank switch.
- 3. Fuel Selector ...... SELECT FULLER TANK
- Fuel Pump ......AS REQUIRED

After switching tanks, message will remain until sensed imbalance is less than 8 gallons.

## **Electrical System**

Low Voltage on Main Bus 1

M BUS 1 Caution

M BUS 1

1. Perform Alt 1 Caution (Failure) Checklist.

Low Voltage on Main Bus 2

M BUS 2 Caution

M BUS 2

1. Perform Alt 1 Caution (Failure) and Alt 2 Caution (Failure) Checklists.

## **Unexpected Discharge on Battery 1**

**BATT 1 Caution** 

BATT 1

- Main Bus 1, 2 and Non-Essential Bus Loads ...... REDUCE
- 2. Main Bus 1, 2 and Essential Bus Voltages ...... MONITOR
- Land as soon as practicable.

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## CIRRUS PILOT'S CHECKLIST MODEL SR22T

### **Low Alternator 1 Output**

#### ALT 1 Caution (Failure)

ALT 1
-------

- If alternator does not reset:
- ALT 1 Switch......OFF 3.
- Non-Essential Bus Loads ......REDUCE
  - If flight conditions permit, consider shedding the following to preserve Battery 1:
    - (1) Air Conditioning
    - (2) Landing Light
    - (3) Yaw Servo
    - (4) Convenience Power (aux items plugged into armrest jack)
    - (5) EVS Camera (if installed)
- Continue Flight, avoiding IMC or night flight as able (reduced power redundancy).

## **Low Alternator 2 Output**

#### ALT 2 Caution (Failure)

#### ALT 2

- 1. ALT 2 Circuit Breaker...... CHECK & SET
- If alternator does not reset:
- 3. ALT 2 Switch......OFF
- Continue Flight, avoiding IMC or night flight as able (reduced power redundancy).

## **Integrated Avionics System**

#### **Avionics Switch Off**

**AVIONICS OFF Caution** 

**AVIONICS OFF** 

AVIONICS Switch ...... ON, AS REQUIRED

### **PFD Cooling Fan Failure**

PFD FAN FAIL Advisory

PFD FAN FAIL

1. AVIONICS FAN 2 Circuit Breaker ...... CYCLE If annunciation does not extinguish:

- High cabin temperatures ...... LAND AS SOON AS PRACTICABLE
- Low cabin temperatures......CONTINUE, MONITOR

### MFD Cooling Fan Failure

MFD FAN FAIL Advisory

MFD FAN FAIL

- AVIONICS FAN 1 Circuit Breaker ...... CYCLE If annunciation does not extinguish:
  - High cabin temperatures ...... LAND AS SOON AS PRACTICABLE
  - Low cabin temperatures......CONTINUE, MONITOR

## Flight Displays Too Dim

INSTRUMENT dimmer knob ...... OFF (FULL COUNTER-CLOCKWISE)

If flight displays do not provide sufficient brightness:

Revert to standby instruments. 2.

## **Pitot Static System**

Statio	Source	Plac	kad
STATIC	Source	BIOC	Keo

1.	Pitot Heat	01
2.	Alternate Static Source	OPE
Pit	ot Tube Blocked	

1. Pitot Heat.....ON

### **Pitot Heat Current Sensor**

#### PITOT HEAT FAIL Caution

#### PITOT HEAT FAIL

						•••••••••••••••••••••••••••••••••••••••	
2.	Pit	tot Heat				CYCLE OF	F, ON
	If	inadvertent	icing	encountered,	perform	Inadvertent	Icing
	Er	ncounter Eme	rgency	Checklist and:			

- a. Airspeed...... EXPECT NO RELIABLE INDICATION
- b. Exit icing conditions using attitude, altitude, and power instruments.

## **Pitot Heat Required**

#### PITOT HEAT REQD Caution

PITOT HEAT REQD

1. Pitot Heat.....ON

## **Flight Control System**

1.	Airplane Control	MAINTAIN MANUALLY
2.	Autopilot (if engaged)	DISENGAGE
	If Problem Is Not Corrected:	
3.	Circuit Breakers	PULL AS REQUIRED
	PITCH TRIM	
	ROLL TRIM	
	YAW SERVO	
	AP SERVOS	
4.	Power Lever	AS REQUIRED
5.	Control Yoke	MANUALLY HOLD PRESSURE

## 6. Land as soon as practicable. Flap System Exceedance

### **FLAPS OVERSPEED Caution**

**FLAPS OVERSPEED** 

1.	Airspeed	REDUCE
	or	
1	Flans	RETRACT

1. Flaps......RET

## **Flaps Not In Takeoff Configuration**

**TAKEOFF FLAPS Caution** 

**TAKEOFF FLAPS** 

1.	Takeoff	ABORT
2.	Aircraft Control	MAINTAIN
3.	Aircraft Configuration	CHECK

## Flaps Not Set For Climb

FLAPS CLIMB Advisory

**FLAPS CLIMB** 

Flaps.....

## **Landing Gear System**

## **Brake Failure During Taxi**

1.	En	gine Power	AS REQUIRED
	a.	To Stop Airplane	REDUCE
	b.	If necessary for steering	INCREASE
2.	Dir	ectional Control	MAINTAIN WITH RUDDER
3.	Bra	ake Pedal(s)	PUMP
	If c	lirectional control can not be maint	ained:
4.	lgr	nition Switch	OFF

## **Left/Right Brake Over-Temperature**

#### **BRAKE TEMP Caution**

BRAKE TEMP

1. Stop aircraft and allow the brakes to cool.

## Oxygen System

## Oxygen Quantity Low

#### **OXYGEN QTY Caution**

#### **OXYGEN QTY**

<ol> <li>Oxvo</li> </ol>	en Pressure an	d Flow F	Rate	.CHECK
--------------------------	----------------	----------	------	--------

- 2. Oxygen Duration......CALCULATE
  - a. See Performance Tab Oxygen Duration: calculate duration based on remaining pressure, number of occupants and type of device (mask or cannula).
- Perform Normal Descent as necessary, dependent on duration calculation.
- 4. Annunciation indicated tank pressure is between 800 and 400 PSI. See Performance Tab - Oxygen Duration.

#### **OXYGEN QTY Advisory**

**OXYGEN QTY** 

#### On-Ground

1. Oxygen Supply ......REPLENISH IF USE OF OXYGEN IS ANTICIPATED

#### In-Flight

If use of oxygen is anticipated, verify adequate oxygen supply for flight duration. See Performance Tab - Oxygen Duration.

## Oxygen Required

#### **OXYGEN RQD Caution**

#### **OXYGEN RQD**

1.	Oxygen System	ON
2.	Oxygen Masks or Cannulas	DON
3.	Oxygen Flow Rate	SET AND MONITOR
4	Pulse Ovimeter	MONITOR O2 LEVELS

## Oxygen System Left On **OXYGEN LEFT ON Advisory**

**OXYGEN LEFT ON** 

1. Oxygen System ...... OFF

## **Check Oxygen System Status CHECK OXYGEN Advisory**

#### **CHECK OXYGEN**

1.	Pulse Oximeter	CHECK SATURATION LEVELS
	If O2 saturations are low:	
	a. Oxygen Mask/Cannula	DON
	b. Oxygen System	ON
	c. Oxygen Flow Rates	CHECK
2.	Oxygen Lines	VERIFY SECURITY AND ROUTING
3.	Oxvgen Quantity	CHECK

## **Windshield De-Ice System Malfunction**

1.	ICE	E PROTECT 1 Circuit BreakerCYCLE			
2.	Flu	id Quantity SWITCH TO FULLER TANK			
3.	WI	ND SHLD Push-ButtonPRESS AS REQUIRED			
		the forward field of view is overly restricted during landing oroach and taxiing:			
	a.	Cabin HeatHOT			
	b.	Windshield DefrostON			
	c.	Execute a forward slip as required for visibility.			
	d.	Avoid taxiing without adequate forward visibility.			
He	ate	ed Lift Transducer Malfunction			
lf i	f ice forms on lift transducer vane:				

- 2. PITOT HEAT Switch.......CYCLE OFF. ON

#### If ice remains on lift transducer vane:

- 1. Stall Warning System ....... EXPECT NO RELIABLE INDICATION This includes:
  - · Impending stall warning.
  - Stall speed indication.
- 2. Airspeed ...... MONITOR, DO NOT STALL
- Fly published V<sub>RFF</sub> speeds...MINIMUM 88 KIAS WITH 50% FLAP

## Low Temperature

#### ANTI ICE TEMP Caution

ANTI ICE TEMP

PFD Alerts Window: "Temperature is too low for ice protection (TKS)"

- 1. ICE PROTECT System Switch ...... OFF
- 2. Icing Conditions ......AVOID / EXIT

# Low Fluid Quantity

**ANTI ICE QTY Caution** 

**ANTI ICE QTY** 

PFD Alerts Window: "Fluid quantity is low (TKS)"

Icing Conditions .....AVOID / EXIT

### Fluid Quantity Imbalance Detected

Fluid Quantity Imbalance Caution

#### ANTI ICE QTY

PFD Alerts Window: "Fluid quantity imbalance has been detected" Imbalance between left and right sensed fluid quantity is greater than 1.0 gallon.

 Revert to AUTO control of the fluid source to control the fluid quantity.

If ANTI ICE PRESS annunciates:

- Revert to manual control of the fluid source to control the fluid level quantity.
  - (1) Fluid Quantity ...... SWITCH TO FULLER TANK
- b. WIND SHLD Push-Button ......PRESS
  - (1) Repeat operation of windshield pump to verify metering pumps are primed properly as evidenced by deicing fluid exiting windshield nozzles.

If Caution Annunciation extinguishes:

a. Anti-Ice System......MONITOR

If Caution Annunciation does not extinguish or intermittent:

- a. Fluid Quantity.....SWITCH TO OPPOSITE TANK
- b. WIND SHLD Push-Button ...... PRESS
  - Repeat operation of windshield pump to verify metering pumps are primed properly as evidenced by deicing fluid exiting windshield nozzles.
- c. lcing Conditions.....AVOID / EXIT

## Low Pressure

#### ANTI ICE PRESS Caution

#### **ANTI ICE PRESS**

PFD Alerts Window:	"Tail pressure	is low	(TKS)'
--------------------	----------------	--------	--------

1. ICE PROTECT 1 and 2 Circuit Breakers ...... SET

- 2. Anti-Ice TKS Quantity...... SWITCH TO FULLER TANK
- 3. WIND SHLD Push-Button.....PRESS
  - a. Repeat operation of windshield pump to verify metering pumps are primed properly as evidenced by deicing fluid exiting windshield nozzles.
- - a. Anti-Ice System......MONITOR

    If caution annunciation does not extinguish or is intermittent:
  - a. PUMP BKUP Switch ......ON
  - b. Icing Conditions.....AVOID / EXIT

## **High Pressure**

#### ANTI ICE PRESS Caution

#### **ANTI ICE PRESS**

### PFD Alerts Window: "Pressure is high (TKS)"

- 1. Evidence of Anti-Ice Flow ......MONITOR / VERIFY
- 2. Icing Conditions ......AVOID / EXIT

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## CIRRUS PILOT'S CHECKLIST MODEL SR22T

## Low/High Airspeed

#### **ANTI ICE SPEED Caution**

#### **ANTI ICE SPEED**

PFD Alerts Window: "Airspeed is too low/high for ice protection (TKS)"

1. Airspeed ...... MAINTAIN 95-177 KIAS AND LESS THAN 204 KTAS.

### Lift Transducer Heater Failure

ANTI ICE HEAT Caution

#### **ANTI ICE HEAT**

PFD Alerts Window: "Stall warning/AoA heater has failed"

- 1. STALL VANE HEAT Circuit Breaker.......CYCLE
- 2. PITOT HEAT Circuit Breaker ...... CYCLE
- Icing Conditions ......AVOID / EXIT
- Fly aircraft normally using airframe buffet as the stall warning. Ice accumulations on the stall warning vane may result in unreliable stall warning system operation.

Left/Right Anti-Ice Fluid Level - Serials 22T-1460, 22T-1471, 22T-1473 thru 22T-1709, 22T-1732

ANTI ICE LEVEL Caution

#### **ANTI ICE LEVEL**

PFD Alerts Window: "Right/Left tank fluid quantity is unreliable (TKS)"

Revert to manual control of the fluid source to control the fluid level quantity.

If ANTI ICE PRESS annunciates:

- Fluid Quantity ......SWITCH TO OPPOSITE TANK
- WIND SHLD Push-Button ...... PRESS
  - (1) Repeat operation of windshield pump to verify metering pumps are primed properly as evidenced by deicing fluid exiting windshield nozzles.

## Dynamic Stall Speed Band Unavailable Advisory AOA FAIL Advisory

**AOA FAIL** 

PFD Alerts Window: "Dynamic stall speed band is unavailable"

Angle of Attack signal has failed. This signal is used to calculate and display a dynamic stall speed awareness band (red band) on airspeed tape. With a failed AOA signal, the low speed red band extends to a fixed value of 61 knots.

## **Other Conditions**

### **Aborted Takeoff**

1.	Power Lever	IDLE
2.	Brakes	AS REQUIRED

## **Parking Brake Engaged**

#### PARK BRAKE Caution

#### PARK BRAKE

1.	Parking Brake	RELEASE	Ξ
----	---------------	---------	---

2. Monitor CAS for BRAKE TEMP Caution. Stop aircraft and allow the brakes to cool if necessary.

### **Communications Failure**

Switches, Controls	CHECK
Frequency	CHANGE
Circuit Breakers	SET
Headset	CHANGE
Hand Held Microphone (if equipped)	CONNECT
	Switches, Controls  Frequency  Circuit Breakers  Headset  Hand Held Microphone (if equipped)

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**EMERGENCY** 

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## **Airspeeds for Emergency Operations**

#### Maneuvering Speed:

3600 lb (1633 kg) ...... 140 KIAS

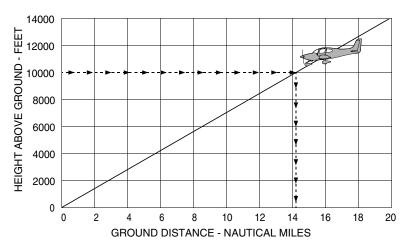
#### Best Glide:

#### Emergency Landing (Engine-out):

Flaps 100% ...... 80 KIAS

#### Maximum Glide

#### Maximum Glide Ratio ~ 8.6:1



SR22 FM03 3564

Note •

Maximum Glide Ratio with Ice Accumulation ~ 6.4: 1

## **Memory Items**

F - 1

Checklist steps emphasized by underlining such as the example below, should be memorized for accomplishment without reference to the procedure.

1. Best Glide Speed..... ESTABLISH

## **Engine Failures**

## **Engine Failure On Takeoff (Low Altitude)**

<u>1.</u>	Best Glide or Landing Speed	ESTABLISH
<u>2.</u>	Mixture	CUTOFF
<u>3.</u>	Fuel Selector	OFF
<u>4.</u>	Ignition Switch	OFF
<u>5.</u>	Flaps	AS REQUIRED
	If time permits:	
	a. Power Lever	IDLE
	b. Fuel Pump	OFF
	c. Seat BeltsE	NSURE SECURED
	d. BAT 1, BAT 2, ALT 1, and ALT 2 Switches	OFF
Engine Failure In Flight		
<u>1.</u>	Best Glide Speed	ESTABLISH
<u>2.</u>	Fuel Selector	SWITCH TANKS
<u>3.</u>	Ignition Switch	CHECK, BOTH
<u>4.</u>	Fuel Pump	BOOST
<u>5.</u>	Power Lever	<u>½ OPEN</u>
<u>6.</u>	MixtureIDLE CUTOFF THEN SLOWLY ENGINE STARTS	Y ADVANCE UNTIL
	If engine does not start:	
<u>7.</u>	Perform Engine Airstart, CAPS Deployme Landing w/o Engine Power Checklist, as require	
	If engine starts:	
8.	CHTs and Oil TemperatureAT PARTIAL POWER IF REQUIRED	WARM ENGINE

## **Airstart**

## **Engine Airstart**

<u>1.</u>	BAT 1 and BAT 2 Switches	ON
<u>2.</u>	Power Lever	<u>OPEN ½ INCH</u>
<u>3.</u>	Mixture	RICH, AS REQ'D
<u>4.</u>	Fuel Selector	SWITCH TANKS
<u>5.</u>	Ignition Switch	<u>BOTH</u>
6.	Fuel Pump	BOOST
7.	ALT 1 and ALT 2 Switches	OFF
8.	Starter (Propeller not Windmilling)	ENGAGE
9.	Power Lever	SLOWLY INCREASE
10.	ALT 1 and ALT 2 Switches	ON
11.	CHTs and Oil TemperatureAT PARTIAL POWER IF REQUIRED FOR S	
12.	If engine will not start, perform <i>Emergence</i> Checklist.	ry Landing w/o Power

## **Smoke and Fire**

## **Cabin Fire In Flight**

<u>1.</u>	BAT 1, ALT 1, and ALT 2 SwitchesOFF, AS REQ'D
<u>2.</u>	Fire ExtinguisherACTIVATE
3.	AVIONICS Switch OFF
4.	All other switchesOFF
5.	Land as soon as possible.
	If setting master switches off eliminated source of fire or fumes and airplane is in night, weather, or IFR conditions:
6.	Airflow SelectorOFF
7.	BAT 1, BAT 2, ALT 1, and ALT 2 SwitchesON
8.	AVIONICS SwitchON
9.	Required SystemsACTIVATE ONE AT A TIME
10.	Temperature Selector
11.	Vent Selector FEET / PANEL / DEFROST POSITION
12.	Airflow SelectorSET AIRFLOW TO MAXIMUM
13.	Panel Eyeball OutletsOPEN
14.	Land as soon as possible.
En	gine Fire In Flight
<u>1.</u>	MixtureCUTOFF
<u>2.</u>	Fuel PumpOFF
<u>3.</u>	Fuel SelectorOFF
<u>4.</u>	Airflow SelectorOFF
<u>5.</u>	Power LeverIDLE
<u>6.</u>	Ignition SwitchOFF
7.	Land as soon as possible.

Wi	ing	Fire In Flight	
<u>1.</u>	Pito	ot Heat Switch	OFF
<u>2.</u>	Naν	rigation Light Switch	OFF
<u>3.</u>	Lan	ding Light	OFF
<u>4.</u>	Stro	bbe Light Switch	OFF
<u>5.</u>	<u>lf p</u>	ossible, side slip to keep flames away from fuel	tank and cabin.
<u>6.</u>	Lan	d as soon as possible.	
En	gin	e Fire During Start	
<u>1.</u>	Mix	ture	CUTOFF
<u>2.</u>	Fue	el Pump	OFF
<u>3.</u>	Fue	el Selector	OFF
<u>4.</u>	Pov	ver Lever	<u>FORWARD</u>
<u>5.</u>	Sta	rter	CRANK
6.		lames persist, perform <i>Emergency Engine</i> ound and <i>Emergency Ground Egress</i> Checklists	
Sn	nok	e and Fume Elimination	
1.	Air	Conditioner (if installed)	OFF
2.	Ten	nperature Selector	COLD
3.	Ver	t Selector FEET / PANEL / DEFRO	OST POSITION
4.	Airf	low SelectorSET AIRFLOW	TO MAXIMUM
	If s	ource of smoke and fume is firewall forward:	
	a.	Airflow Selector	OFF
5.	Par	nel Eyeball Outlets	OPEN
6.	Sup	pplemental Oxygen (if available)	
	a.	Oxygen Masks or Cannulas	DON
	b.	Oxygen System	ON
	C.	Oxygen Flow Rate	MAXIMUM

7. Land as soon as possible.

## **Emergency Descent**

## **Emergency Descent**

<u>1.</u>	Power Lever	<u>IDLE</u>
<u>2.</u>	MixtureA	S REQUIRED
<u>3.</u>	Airspeed	<u>V</u> NF

## **Forced Landing**

## **Emergency Landing w/o Engine Power**

<u>1.</u>	Best Glide Speed	<u>ESTABLISH</u>
		TRANSMIT (121.5 MHZ) MAYDAY
	GIVING LOCATION AND INTE	
<u>3.</u>	Transponder	SQUAWK 7700
<u>4.</u>	Power Lever	<u>IDLE</u>
<u>5.</u>	Mixture	CUTOFF
<u>6.</u>	Fuel Selector	OFF
<u>7.</u>	Ignition Switch	OFF
<u>8.</u>	Fuel Pump	OFF
9.	Flaps (when landing is assured	)100%
10.	Seat Belt(s)	SECURED
11.	BAT 1, BAT 2, ALT 1, and ALT 2	SwitchesOFF

## **Ditching**

<u> 1.</u>	<u> </u>		TRANSMIT	(121.5 MHZ)	<u> </u>
	<b>GIVING</b>	LOCATION AND INTE	<u>NTIONS</u>		
_	_			0011	

<u>2.</u>	Transponder	SQUAWK 7700
3	CAPS	ACTIVATE

<u>u.</u>	<u> </u>	 
4	Airnlane	EVACHATE

т.	Air plane	• • • • • • • • • • • • • • • • • • • •	LVAOOAIL
5.	Flotation Devices	INFLATE WHEN CLE	EAR OF AIRPLANE

## **Landing Without Elevator Control**

١.	Fiaps	50%
2.	Trim	80 KIAS
2	Power	AS DECLIDED FOR CLIDE ANGLE

## **Engine System**

## **Engine Partial Power Loss**

1.	. Air Conditioner (if installed)	OFF
2.	. Fuel PumpHIGH I	BOOST/PRIME
3.	. Fuel SelectorS	WITCH TANKS
4.	. MixtureCHECK APPROPRIATE FOR FLIGHT	CONDITIONS
5.	. Power Lever	SWEEP
6.	. Ignition SwitchBO	TH, L, THEN R
_		

### Land as soon as practicable.

### Oil Pressure Out of Range

#### OIL PRESS Warning

#### **OIL PRESS**

1. Oil Pressure Gauge ......CHECK If pressure low / high:

- a. Power ......REDUCE TO MINIMUM FOR SUSTAINED FLIGHT
- b. Land as soon as possible.
  - (1) Prepare for potential engine failure.

### Oil Temperature High

#### OIL TEMP Warning

#### **OIL TEMP**

1.	PowerREDUCE
2.	AirspeedINCREASE
3.	Mixture ADJUST FUEL FLOW TO TOP OF GREEN ARC
4.	Oil Temperature Gauge MONITOR
	If message persists:

Land as soon as possible.

# CIRRUS PILOT'S CHECKLIST

MODEL SR22T

## **High Cylinder Head Temperature**

**CHT Caution and Warning** 

СНТ

#### **On-Ground**

1.	Power LeverREDUCE
2.	Annunciations and Engine TemperaturesMONITOR
	If Caution or Warning annunciation is still illuminated:
3.	Power LeverMINIMUM REQUIRED
4.	FlightPROHIBITED
In-	Flight
1.	Power LeverREDUCE
2.	Mixture ADJUST FUEL FLOW TO TOP OF GREEN ARC
3.	AirspeedINCREASE
4.	Annunciations and Engine TemperaturesMONITOR
	If Caution or Warning annunciation is still illuminated:
5.	Power Lever MINIMUM REQUIRED
3.	Engine InstrumentsMONITOR
	If Caution annunciation only remains illuminated:
	a. Land as soon as practicable.

If Warning annunciation remains illuminated:

a. Land as soon as possible.

## **Turbocharger System Emergencies Unexpected Loss Of Manifold Pressure**

- Power ......ADJUST TO MINIMUM REQUIRED FOR SUSTAINED **FLIGHT**
- Mixture...... ADJUST FOR EGTS BETWEEN 1300 TO 1400 °F 2.
- Descend to MINIMUM SAFE ALTITUDE from which a landing may be safely accomplished.
- Divert to nearest suitable airfield. 4.
- Radio ......ADVISE ATC LANDING IS URGENT OR 5. TRANSMIT (121.5 MHZ) MAYDAY GIVING LOCATION AND INTENTIONS WHEN WORKLOAD PERMITS
- 6.
- 7. Land as soon as possible.

### **Manifold Pressure High**

#### MAN PRESSURE Warning

#### MAN PRESSURE

- Power Lever..... REDUCE MAP TO LESS THAN 36.5 IN.HG
- Flight......CONTINUE 2. If noticeable surging is present:
- 3. Perform Overboost / Pressure Relief Valve Checklist.

#### Overboost / Pressure Relief Valve

- Power Lever......REDUCE TO 30.5 IN.HG OR LESS
- Mixture ...... ADJUST FUEL FLOW TO TOP OF GREEN ARC 2. If continued surging is present:
- 3. Land as soon as practicable.

### **Turbine Inlet Temperature High**

### TIT Warning



1. Mixture...... ADJUST FUEL FLOW TO TOP OF GREEN ARC 2. Ignition Switch ......CHECK ON BOTH If TIT remains in excess of limits: a. Power ...... REDUCE b. Land as soon as practicable.

### **EGT, TIT or CHT Temperature Sensor Failure**

- Similar Gauges ......MONITOR
- Flight......CONTINUE USING REMAINING GAUGES AS 2. REPRESENTATIVE

## **Propeller System Emergencies**

### **Engine Speed High**

RPM Warning



- 1. Power Lever....... REDUCE BY 2 IN.HG MANIFOLD PRESSURE If governor is not in control (RPM reduces and remains lower after power adjustment):
- 2. Perform Propeller Governor Failure Checklist. If governor is in control (RPM remains high, but stable after power reduction):
- 3. Power Lever.... REDUCE BELOW 34 IN.HG FOR CLIMB, BELOW 30.5 IN.HG FOR CRUISE

If governed engine speed exceeds 2600 RPM:

- 4. Perform Propeller Governor Failure Checklist. If governed engine speed is 2600 RPM or less:
- Flight......CONTINUE

### **Propeller Governor Failure**

- Power Lever..... REDUCE TO MINIMUM NECESSARY FOR SUSTAINED FLIGHT
- 2. Airspeed ......REDUCE TO 85-90 KIAS
- 3.
- 4. Land as soon as practicable.

## **Fuel System Emergencies**

## Low Fuel Quantity in Left Tank

FUEL LOW LEFT Warning

		w		

1.	Fuel Quantity GaugesCHECK
2.	Fuel PumpBOOST
	If HIGH BOOST already in use for vapor suppression, pump should be left in this position for tank switching.
3.	Fuel SelectorRIGHT TANK

### Low Fuel Quantity in Right Tank

#### FUEL LOW RIGHT Warning

#### **FUEL LOW RIGHT**

١.	Fuel Quantity GaugesCHECK
2.	Fuel PumpBOOST
	If HIGH BOOST already in use for vapor suppression, pump should be left in this position for tank switching.

3. Fuel Selector ...... LEFT TANK

## Low Fuel Quantity

### FUEL LOW TOTAL Warning

#### **FUEL LOW TOTAL**

- 1. Fuel Quantity Gauges......CHECK 2. Totalized Fuel Quantity ......CHECK If totalized fuel quantity differs significantly from sensed quantity:
  - a. Initial Fuel Value......VERIFY AND CORRECT If message persists:

a. Land as soon as practicable.

#### **Fuel Imbalance**

#### FUEL IMBALANCE Warning

#### **FUEL IMBALANCE**

1.	Fuel Quantity GaugesCHECK
2.	Fuel PumpBOOST
	If HIGH BOOST already in use for vapor suppression, pump should be left in this position for tank switch.
3.	Fuel Selector SELECT FULLER TANK
4.	Fuel PumpAS REQUIRED
	After switching tanks, message will remain until sensed imbalance is less than 12 gallons.

### **High Fuel Flow**

### FUEL FLOW Warning

#### **FUEL FLOW**

#### **On-Ground**

If warning occurs during takeoff role:

- Takeoff Run......ABORT If on ground, not during takeoff run:
- 2. Power Lever......REDUCE
- Flight Operation ...... PROHIBITED Maintenance required to reduce fuel flow.

#### In-Flight

- 1. Mixture Lever.. ADJUST FUEL FLOW TO TOP OF GREEEN ARC
- 2. Annunciations and Engine Temperatures ...... MONITOR

# CIRRUS PILOT'S CHECKLIST

MODEL SR22T

# **Electrical System Emergencies**

## High Voltage on Main Bus 1

M BUS 1 Warning

	M BUS 1	
1.	ALT 1 Switch	CYCLE
2.	M Bus 1 Voltage (M1)	CHECK
	If M Bus 1 Voltage is greater than 32 volts:	
3.	ALT 1 Switch	OFF
4.	Perform Alt 1 Caution (Failure) Checklist (do not r	eset alternator).
Hi	gh Voltage on Main Bus 2	
M	BUS 2 Warning	
	M BUS 2	
1.	Main Bus 1 Voltage (M1)	CHECK
	If M Bus 1 Voltage is greater than 32 volts:	
2.	If M Bus 1 Voltage is greater than 32 volts: Perform M Bus 1 Warning Checklist.	
2. 3.		CHECK
	Perform M Bus 1 Warning Checklist.	CHECK
	Perform <i>M Bus 1</i> Warning Checklist.  Main Bus 2 Voltage (M2)	
3.	Perform <i>M Bus 1</i> Warning Checklist.  Main Bus 2 Voltage (M2)  If <i>M Bus 2 Voltage is greater than 32 volts:</i>	CYCLE
3. 4.	Perform <i>M Bus 1</i> Warning Checklist.  Main Bus 2 Voltage (M2)  If <i>M Bus 2 Voltage is greater than 32 volts:</i> ALT 2 Switch	CYCLE

7. Perform Alt 2 Caution (Failure) Checklist (do not reset alternator).

## **High or Low Voltage on Essential Bus** ESS BUS Warning

ESS BUS
---------

- 1. Essential Bus Voltage (ESS) ......CHECK If Essential Bus Voltage is greater than 32 volts:
- Main Bus 1 and Main Bus 2 Voltages (M1 and M2)......CHECK
- Perform appropriate M Bus 1 Warning or M Bus 2 Warning 3. Checklist.

If Essential Bus Voltage is less than 24.5 volts:

Perform Alt 1 Caution (Failure) and Alt 2 Caution (Failure) Checklists.

If unable to restore at least one alternator:

- Non-Essential Loads ...... REDUCE
  - If flight conditions permit, consider shedding: Air Conditioning, Landing Light, Pitot Heat, Cabin Fan, Nav Lights, Strobe Lights, Audio Panel, COM 2, Yaw Damper
- 6. Land as soon as practicable (Battery reserve only).

# **Environmental System Emergencies**

## **Carbon Monoxide Level High**

CO LVL HIGH Warning

#### CO LVL HIGH

1.	Air	Conditioner (if installed)	NOT IN RECIRC MODE
2.	Ter	mperature Selector	COLD
3.	Ve	nt Selector FEET/PAN	EL/DEFROST POSITION
4.	Air	flow SelectorSET	AIRFLOW TO MAXIMUM
5.	Pa	nel Eyeball Outlets	OPEN
	If C	CO LVL HIGH does not extinguish:	
6.	Su	pplemental Oxygen (if available)	
	a.	Oxygen Masks or Cannulas	DON
	b.	Oxygen System	ON
	c.	Oxygen Flow Rate	MAXIMUM
7.	La	nd as soon as possible.	

## **Oxygen System Emergencies**

## | Oxygen System Fault - Above 12,500 Ft **OXYGEN FAULT Warning**

CIRRUS PILOT'S CHECKLIST

#### **OXYGEN FAULT**

	<u>1.</u>	<u>Ox</u>	ygen	Flow Rate	CHECK
		If n	o flo	w:	
Ì	<u>2.</u>	<u>Init</u>	iate E	Emergency Descent to below 12,500 ft:	
		a.	Pow	ver Lever	IDLE
		b.	Mixt	tureA	S REQUIRED
		c.	Airs	peed	V <sub>NE</sub>
			Belo	ow 12,500 ft:	
			(1)	Oxygen System	OFF
				FlightRemain below altitudes requiring supplement	
		If fi	low is	s normal:	
	3.	Ox	ygen	Flow Rate	MONITOR
	4.	Init	iate N	Normal Descent as soon as practical.	
		Be	low 1	2,500 ft:	
	5.	Ox	ygen	System	OFF
	6.	Flig	ght		CONTINUE
		Re	main	below altitudes requiring supplemental oxyg	jen.

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# CIRRUS PILOT'S CHECKLIST MODEL SR22T

# **Oxygen Quantity Low**

# **OXYGEN QTY Warning**

	OXYGEN QTY
1.	Oxygen Pressure and Flow RateCHECK
2.	Initiate Normal Descent (non-emergency) below 12,500 ft.
3.	Oxygen Flow RateMONITOR  Below 12,500 ft:
4.	FlightCONTINUE Remain below altitudes requiring supplemental oxygen.
O	kygen Required
OX	YGEN RQD Warning
	OXYGEN RQD
1.	Oxygen SystemON
2.	Oxygen Mask/Cannula
3.	Oxygen Flow RateSET AND MONITOR
4.	Pulse OximeterMONITOR O2 LEVELS

## Anti-Ice System (Serials w/ FIKI)

	Anti-Ice S	ystem Failure	Excessive	Ice A	ccumulation
--	------------	---------------	-----------	-------	-------------

- 1. ICE PROTECT 1 and 2 Circuit Breakers ...... SET
- Fluid Quantity ...... SWITCH TO FULLER TANK 2.
- 3. WIND SHLD Push-Button.....PRESS
  - Repeat operation of windshield pump to verify metering pumps are primed properly as evidenced by deicing fluid exiting windshield nozzles.
- 4. ICE PROTECT Mode Switch ...... VERIFY HIGH
- PUMP BKUP Switch.....ON 5 If determined windshield pump is not priming:
- 6. Exit Icing Conditions Immediately.
- 7. Airspeed ......95 KIAS OR GREATER Maintain a minimum airspeed of 95 KIAS or higher to stay above pre-stall buffet. If unable to maintain this airspeed, allow altitude to decrease in order to maintain 95 KIAS.
- 8. Minimum Approach Speed w/ Residual Ice (Flaps 50%)... 88 KIAS In severe icing conditions, it may not be possible to maintain altitude or proper glide path on approach; in this case, it is imperative that a safe airspeed be maintained, the stall warning system may not function and there may be little or no pre-stall buffet with heavy ice loads on the wing.
- 9. FLAPS ...... MINIMUM REQUIRED When landing is assured, select the minimum flap setting required, not to exceed 50%, and maintain extra airspeed consistent with available field length. Do not retract the flaps once they have been extended unless required for go-around.

## **Low Fluid Quantity** ANTI ICE QTY Warning

ANTI ICE QTY

PFD Alerts Window: "Fluid quantity is low (TKS)"

1. Icing Conditions ......AVOID / EXIT

## **AOA Probe Overheat** AOA OVERHEAT Warning

	AOA OVERHEAT		
PFD Alerts Wind	ow: "AOA probe is overheated"	-	
	·		
1. PITOT HEAT	Switch	OFF	
2. Icing Condition	ons	AVOID / EXIT	
Tank Control	Failure		
ANTI ICE CTL Warning			
	ANTI ICE CTL		
PFD Alerts Windo	ow: "Tank valves cannot be controlle	ed (closed) (TKS)"	
1. Icing Condition	ons	AVOID / EXIT	

## **Unreliable Left and Right Fluid Quantity** ANTI ICE QTY Warning

ANTI	ICE	OTV	
~!!!!	.~_	<b>Q</b> ::	

PFD Alerts Window: "Left and right fluid quantities unknown (TKS)" Both fluid quantities are unknown and both tanks are closed.

- ICE PROTECT System Switch ...... OFF
- Icing Conditions ......AVOID / EXIT 2.

## **Full Flaps Prohibited In Ice** FLAPS ICE Warning

**FLAPS ICE** 

PFD Alerts Window: "Full flaps prohibited in icing conditions"

Flaps......SET UP OR 50%

## **Integrated Avionics System Emergencies** Attitude & Heading Reference System (AHRS) Failure

- 1. Verify Avionics System has switched to functioning AHRS. If not, manually switch to functioning AHRS and attempt to bring failed AHRS back on-line:
- 2. Failed ADAHRS Circuit Breaker ...... SET If open, reset (close) circuit breaker. If circuit breaker opens again, do not reset.
- Be prepared to revert to Standby Instruments (Attitude, Heading).

### Air Data Computer (ADC) Failure

- 1. Failed ADAHRS Circuit Breaker ...... SET If open, reset (close) circuit breaker. If circuit breaker opens again, do not reset.
- 2. Revert to Standby Instruments (Altitude, Airspeed).
- 3. Land as soon as practicable.

### **PFD Display Failure**

- Display Backup ......ACTIVATE
- 2. Land as soon as practicable.

# **Unusual Attitude Emergencies**

## **Inadvertent Spin Entry**

1. CAPS..... ACTIVATE

### **SPIN SPIN SPIN Warning**

SPIN SPIN SPIN

CAPS.....ACTIVATE

### **Inadvertent Spiral Dive During IMC Flight**

- 1. Power Lever......IDLE
- 2. Stop the spiral dive by using coordinated aileron and rudder control while referring to the attitude indicator and turn coordinator to level the wings.
- 3. Cautiously apply elevator back pressure to bring airplane to level flight attitude.
- 4. Trim for level flight.
- Set power as required. 5.
- Use autopilot if functional otherwise maintain a constant heading through the coordinated aileron and rudder inputs.
- 7. Exit IMC conditions as soon as possible.

## **Other Emergencies**

## **Power Lever Linkage Failure**

1.	Power Lever Movement VERIFY
2.	PowerSET IF ABLE
3.	Flaps SET IF NEEDED
4.	Mixture AS REQUIRED (FULL RICH TO CUT-OFF)
5.	Land as soon as possible.
En	nergency Engine Shutdown On Ground
<u>1.</u>	Power LeverIDLE
<u>2.</u>	Fuel Pump (if used)OFF
3.	MixtureCUTOFF

4. Fuel Selector ......OFF 5. Ignition Switch ......OFF

## 6. BAT 1, BAT 2, ALT 1, and ALT 2 Switches...... OFF Left/Right Brake Over-Temperature

### **BRAKE TEMP Warning**

BRAKE TEMP

1. Stop aircraft and allow the brakes to cool.

## **Start Engaged** START ENGAGED Warning

#### START ENGAGED

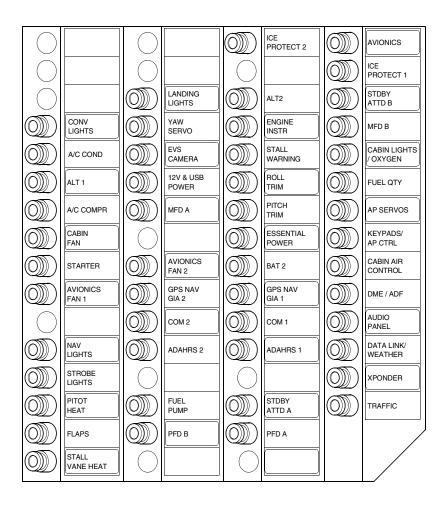
On	-Ground
1.	Ignition Switch
2.	Wait 1 minute before next start attempt.
	If starter does not disengage (relay or solenoid failure):
3.	BAT 1 Switch OFF
4.	EngineSHUTDOWN
5.	STARTER Circuit BreakerPULL
In-	Flight
1.	Ignition SwitchENSURE NOT STUCK IN START
2.	STARTER Circuit BreakerPULL
3.	FlightCONTINUE Engine start will not be available at destination.
En	nergency Ground Egress
<u>1.</u>	EngineSHUTDOWN
<u>2.</u>	Seat beltsRELEASE
<u>3.</u>	AirplaneEXIT

# **CAPS Deployment**

<u>1.</u>	Activation Handle CoverREMOVE
<u>2.</u>	Activation Handle (Both Hands)PULL STRAIGHT DOWN
After Deployment as time permits:	
3.	MixtureCUTOFF
4.	Fuel Selector OFF
5.	Fuel Pump OFF
6.	BAT 1, BAT 2, ALT 1, and ALT 2 SwitchesOFF
	Turn the master switches off after completing any necessary radio communications.
7.	Ignition Switch OFF
8.	ELTON
9.	Seat Belts and HarnessesTIGHTEN
10.	Loose Items SECURE
11.	Assume emergency landing body position.
12.	After the airplane comes to a complete stop, evacuate quickly and move upwind.

## CIRRUS PILOT'S CHECKLIST

## **Circuit Breaker Panel**



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