Output Feedback Control of Underwater Vehicles Using Nonlinear State Observer

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Fourth Semi-Annual Meeting-MURI
Nonlinear Active Control of Dynamical Systems
Oct.6 1998





Outline

- Summary of Previous Work
- Problem Formulation
- Output Feedback Control
- Dive Plane Equation of Motion
- Observer Design
- Controller Design
- Simulation Results
- Conclusion





mmary of previous work

Effects of actuator dynamics

- Actuators in UUVs such as thrusters, rudders have time lag and this tends to degrade system performance and make quick and precise control difficult.
- Thruster dynamics has significant influence on the dynamics of vehicle.
- Investigation of the influence of the rudder dynamics on the system response. However, since UUVs are operated in low speed, rudder dynamics does not influence the UUV dynamics.
- Rudder dynamics is substantial only at high speed such as in torpedo application.





Problem Formulation

- Sliding Mode Controller
 - Advantages
 - Capable of handling nonlinear systems
 - Robust to uncertainties/parameter variation
 - Disadvantages
 - All states measurement required
 - Occurrence of undesirable chattering phenomena
 - Chattering can be avoided by including boundary layer
 - Design of nonlinear observer for state estimation





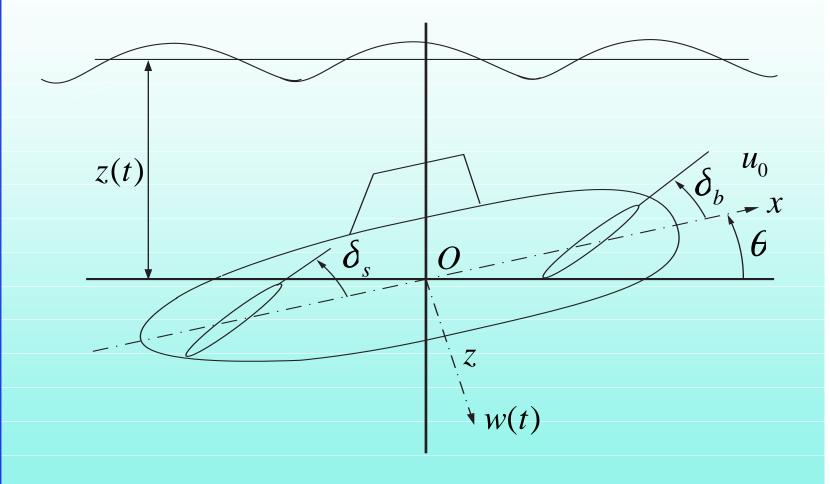
Output Feedback Control

- Healey(NPS) et.al.
 - Design of Robust Observer
 - Sliding Mode Control with estimated states
- Fossen(NIT) et.al.
 - Nonlinear Observer from the kinematic relationship with additional acceleration measurements
 - Adaptive Control with estimated states
- Nonlinear State observer using Sliding Mode





Dive Plane Equation of Motion







Plane Equation of Motion(Cont'd)

$$m[\dot{w} - uq - x_G \dot{q} - z_G q^2]$$

$$= z_{\dot{q}} \dot{q} + z_{\dot{w}} \dot{w} + z_q uq + z_w w + u^2 (z_b \delta_b + z_s \delta_s)$$

$$I_{y}\dot{q} - m[x_{G}(\dot{w} - uq) - z_{G}(\dot{u} + wq)$$

$$= M_{\dot{q}}\dot{q} + M_{\dot{w}}\dot{w} + M_{q}uq + M_{w}uw + u^{2}(M_{b}\delta_{b} + M_{s}\delta_{s})$$

$$-(x_{G}mg - x_{B}B)\cos\theta - (z_{G}mg - z_{B}B)\sin\theta$$

$$\theta = q$$







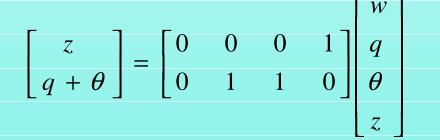
Plane Equation of Motion(Cont'd)

State Space Form

$$\begin{bmatrix} \dot{w} \\ \dot{q} \\ \dot{\theta} \\ \dot{z} \end{bmatrix} = \begin{bmatrix} a_{11} u & a_{12} u & a_{13} & 0 \\ a_{21} u & a_{22} u & a_{23} & 0 \\ 0 & 1 & 0 & 0 \\ 1 & 0 & -u & 0 \end{bmatrix} \begin{bmatrix} w \\ q \\ \theta \\ z \end{bmatrix} + \begin{bmatrix} b_{11} u^2 b_{12} u^2 \\ b_{21} u^2 b_{22} u^2 \\ 0 & 0 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \delta_b \\ \delta_s \end{bmatrix}$$

$$+\begin{bmatrix} F_{d} \cos \theta \\ M_{d} \cos \theta \\ 0 \\ 0 \end{bmatrix} + \begin{bmatrix} e_{11} q^{2} + e_{12} qw \\ e_{21} q^{2} + e_{22} qw \\ 0 \\ 0 \end{bmatrix}$$

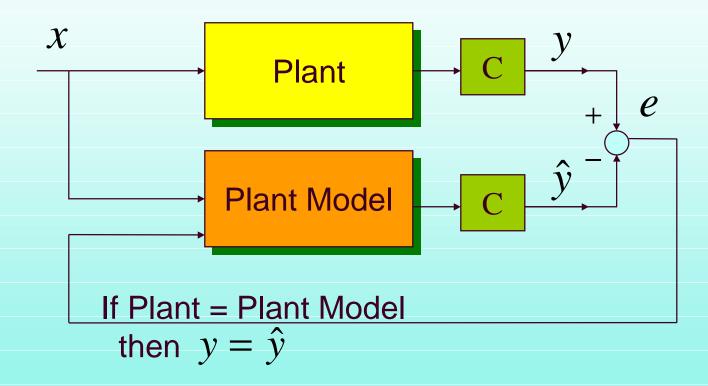
Output Equation







Observer Design







Server Design(Cont'd)

$$\dot{x} = Ax + f(t, x) + B[u(t) + v(t)]$$
$$y = Cx$$

Assumptions

- A1) The system is detectable so that we can find G. The spectrum of A-GC=0 is in the LHP
- A2) There exists $Q \in R^{n \times m}$, symmetric and positive definite, and function h and w such that:

$$f(t,x) = P^{-1}C^{T}h(t,x)$$

$$Bv(t) = P^{-1}C^{T}w(t)$$
where $A_0^{T}P + PA_0 = -Q$



A3) Let $\xi(t,x) = h(t,x) + w(t,x)$. Then, there exists a positive scalar

Such that:
$$\|\xi(t,x)\| < \rho$$



Server Design(Cont'd)

Sliding Mode Observer

$$\dot{\hat{x}} = A_o \hat{x} + Gy + S(\hat{x}, y) + Bu$$
 where
$$S(\hat{x}, y) = \begin{cases} -\frac{P^{-1}C^TCe}{\|Ce\|} \rho & \text{for } e \neq 0 \\ 0 & \text{for } e = 0 \end{cases}$$

- We measure the depth and the pitch rate.
- The pitch angle can be obtained by integrating pitch rate.





Controller Design

PD Controller

$$\begin{bmatrix} \delta_{b} \\ \delta_{s} \end{bmatrix} = \begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} \\ K_{21} & K_{22} & K_{23} & K_{24} \end{bmatrix} \begin{bmatrix} w - w_{com} \\ q - q_{com} \\ \theta - \theta_{com} \\ z - z_{com} \end{bmatrix}$$

Sliding Mode Controller

Switching surface

$$\sigma(t,x) = [\sigma_1(t,x) \dots \sigma_m(t,x)]^T = 0$$

Reaching condition

$$\frac{d}{dt}(\frac{1}{2}\boldsymbol{\sigma}^T\boldsymbol{\sigma}) = \boldsymbol{\sigma}^T\dot{\boldsymbol{\sigma}} < 0$$





Controller Design(Cont'd)

Gradient of **O**

$$\frac{\partial \sigma(t, x)}{\partial x} = S(t, x)$$

Control input

$$u = u_{eq_{nom}} + \overline{u}$$

$$\begin{cases} u_{eq_{nom}} = -[S(t, x)B]^{-1}(S(t, x)Ax + \frac{\partial \sigma}{\partial t}) \\ \overline{u} = -[S(t, x)B]^{-1} \eta \operatorname{sign}(\sigma) \end{cases}$$





Simulation results

- Nominal forward speed 6 knot
- Bounds on nonlinear terms

 $|\theta| < 10^{\circ}$: pitch angle

|q| < 0.2 rad/sec : pitch rate

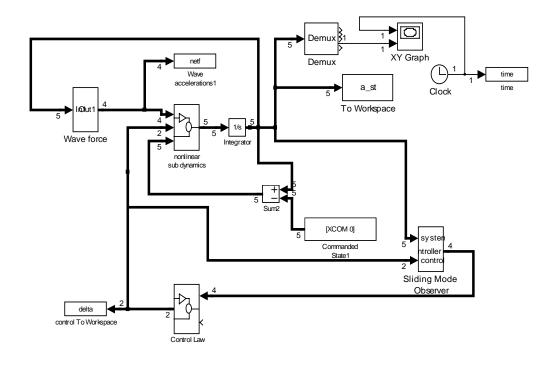
|w| < 1.5 ft/s : heave speed

Include bounds of wave force



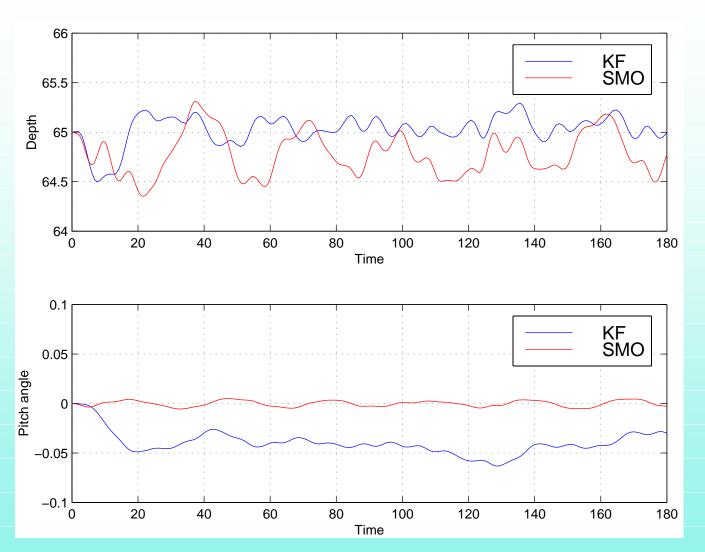


Simulink Model



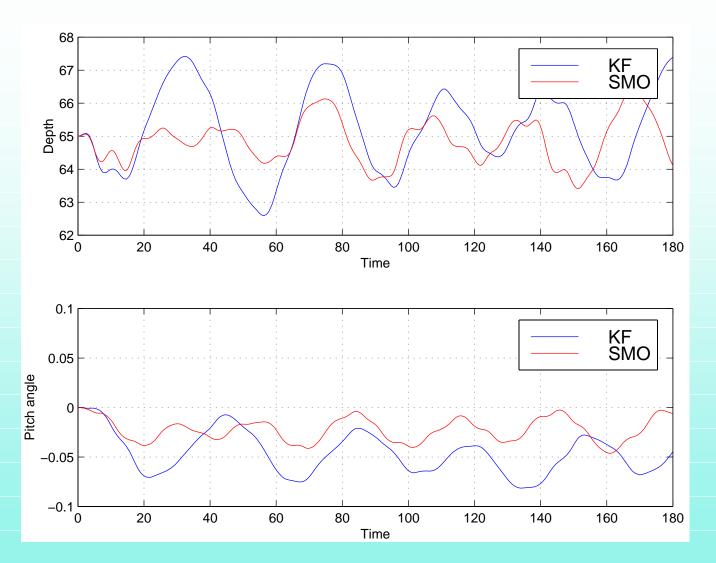






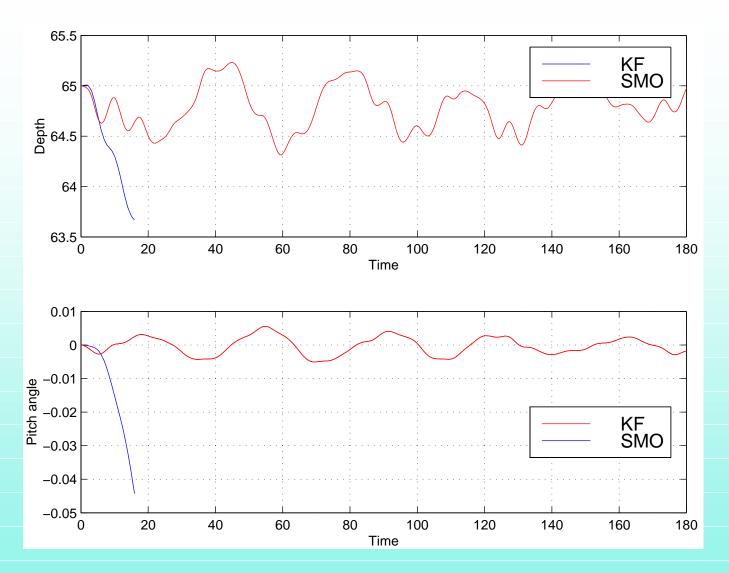


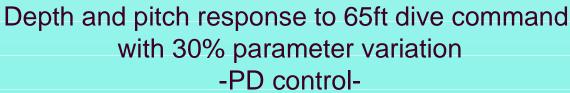
Depth and pitch response to 65ft dive command at sea state 3 -PD control-





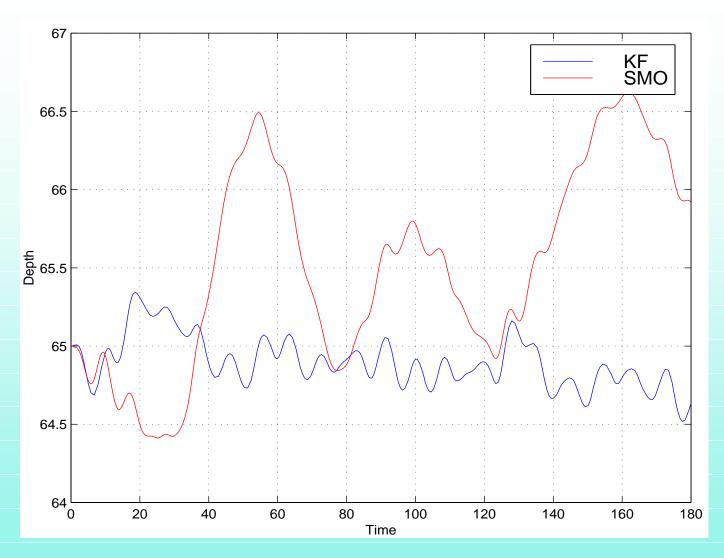
Depth and pitch response to 65ft dive command at sea state 4 -PD control-





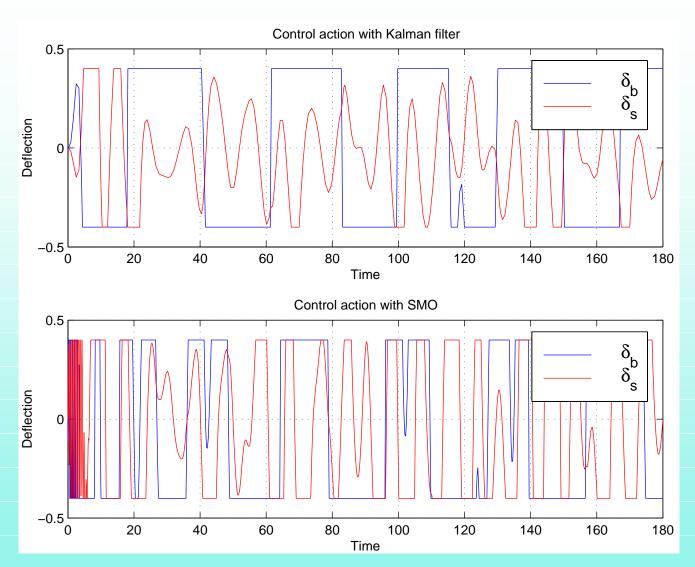


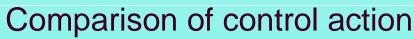






Depth and pitch response to 65ft dive command at sea state 3
-Sliding mode control-





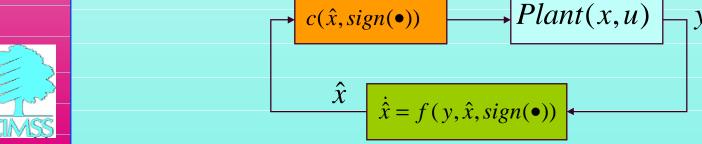




SMC and SMO Combination

- Unexpected results are observed when SMC(Sliding Mode Controller) and SMO(Sliding Mode Observer) are combined.
- Possible problem
 - nested discontinuities

$$f = sign(sign(u))$$







Conclusion

- Sliding mode observer is designed and applied in the dive plane of submarine in order to estimate linear velocity component(heave velocity).
- It is compared with the Kalman filter estimator which is based on linearized dynamics and showed better performance in terms of robustness to wave disturbance, model parameter uncertainties, and parameter variation.
- When,however, the SMC and SMO are combined, undesirable results are obtained. More investigation is required.
- This sliding mode observer can be applied to fault detection algorithm in replace of Kalman filter.
- It takes more computational time in estimating the states when sliding mode observer is used. More efficient computational algorithm needs to be investigated.

