

SMJE 3153 PID Controllers Design



Chapter Outline

- 2.1 Introduction and Re-visit
- 2.2 Design with Gain Adjustment
- 2.3 Improving the Steady State Error
 - 2.3.1 Proportional Integral Controller (PI)
- 2.4 Improving the Transient Response
 - 2.4.1 Proportional Derivative Controller (PD)
- 2.5 Improving Transient Response and Steady-State Error
 - 2.5.1 Proportional-Integral-Derivative Controller (PID)
- 2.6 Tuning the PID using Ziegler-Nichols Technique
- 2.7 Controller design using MATLAB



Introduction

- Root locus (RL) is a powerful tool for design of control systems.
- We will study four design techniques:
 - Gain adjustment
 - Proportional-Integral (PI) controller
 - Proportional-Derivative (PD) controller
 - Proportional-Integral-Derivative (PID) controller



Re-visit: Input Signals

Input	Function	Description	Sketch	Use
Impulse	$\delta(t)$	$\delta(t) = \infty \text{ for } 0 - < t < 0 +$ $= 0 \text{ elsewhere}$ $\int_{0}^{0+} \delta(t) dt = 1$	f(t)	Transient response Modeling
Step	u(t)	u(t) = 1 for t > 0 $= 0 for t < 0$	f(t)	Transient response Steady-state error
Ramp	tu(t)	$tu(t) = t \text{ for } t \ge 0$ = 0 elsewhere	f(t)	Steady-state error
Parabola	$\frac{1}{2}t^2u(t)$	$\frac{1}{2}t^2u(t) = \frac{1}{2}t^2 \text{ for } t \ge 0$ = 0 elsewhere	f(t)	Steady-state error
Sinusoid	sin ωt		f(t)	Transient response Modeling Steady-state error
			t	



Re-visit: Poles and Zeros

- Poles: roots of the denominator of a transfer function.
- Zeros: roots of the numerator of a transfer function.
- Poles and zeros can be mapped on an s-plane (pole: x, zero: o).
- Important in control systems.

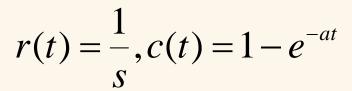


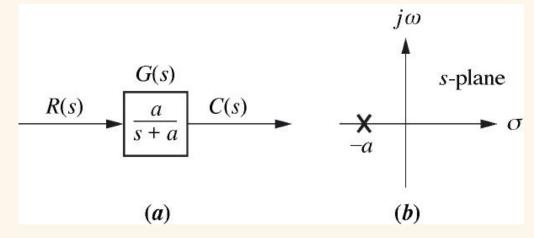
Re-visit: First Order System

Transfer function:

$$\frac{C(s)}{R(s)} = \frac{a}{(s+a)}$$

• With a unit step input,

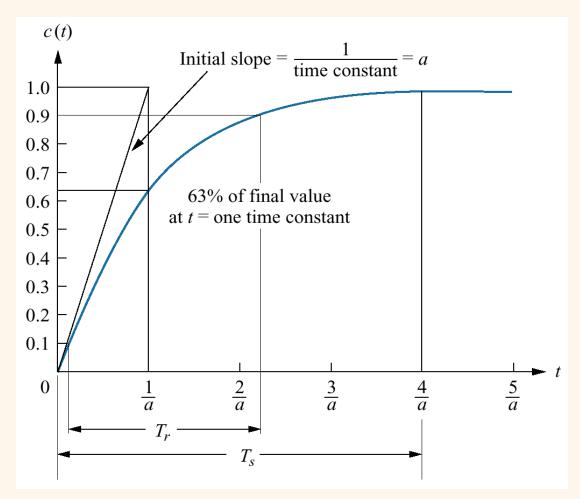






Re-visit: First Order System

- Time response
- Specifications:
 time constant,
 rise time, settling
 time.





Revisit: Second Order System

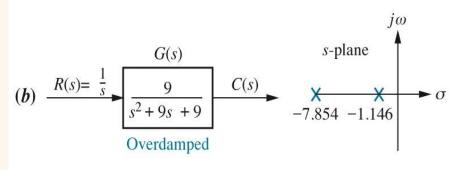
Transfer function:

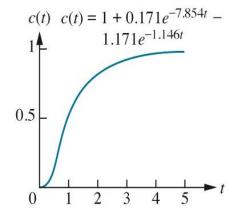
$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

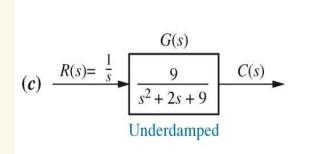
- The second order system responses are determined by damping ratio, ζ and natural frequency, ω_{n}
- Time responses can be categorised into overdamped, underdamped, undamped and critically damped based on the damping ratio and pole locations.

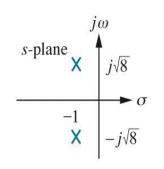


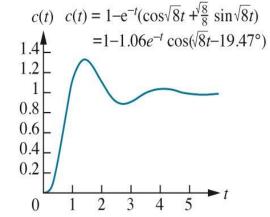
(a)
$$R(s) = \frac{1}{s} \longrightarrow \begin{bmatrix} G(s) \\ \frac{b}{s^2 + as + b} \end{bmatrix}$$
 General













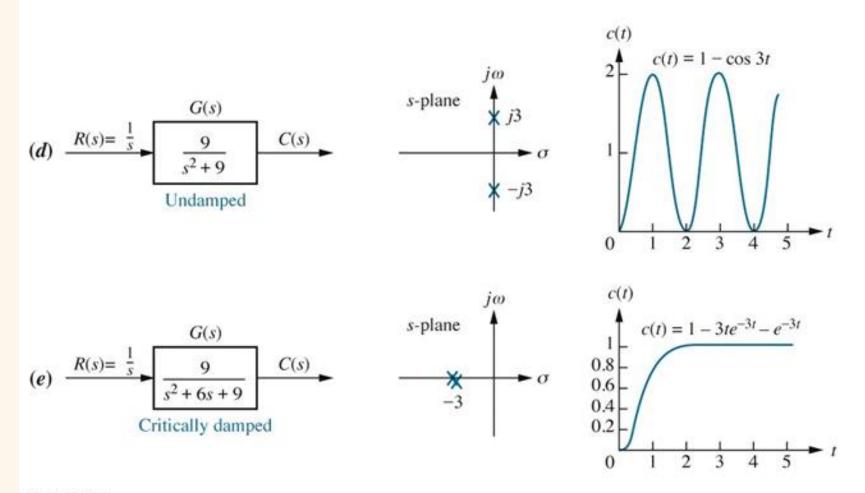
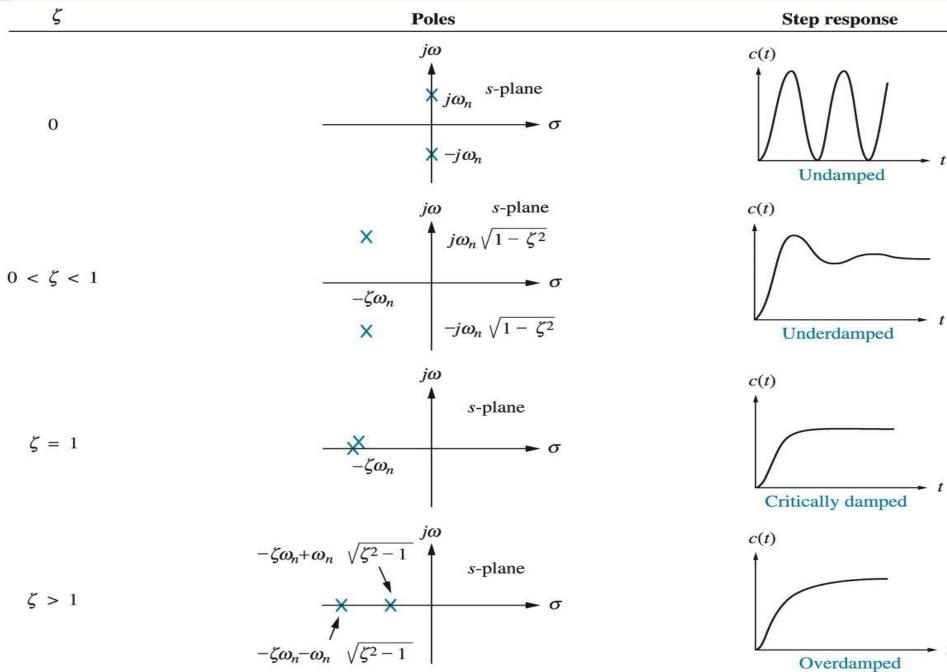


Figure 4.7cde

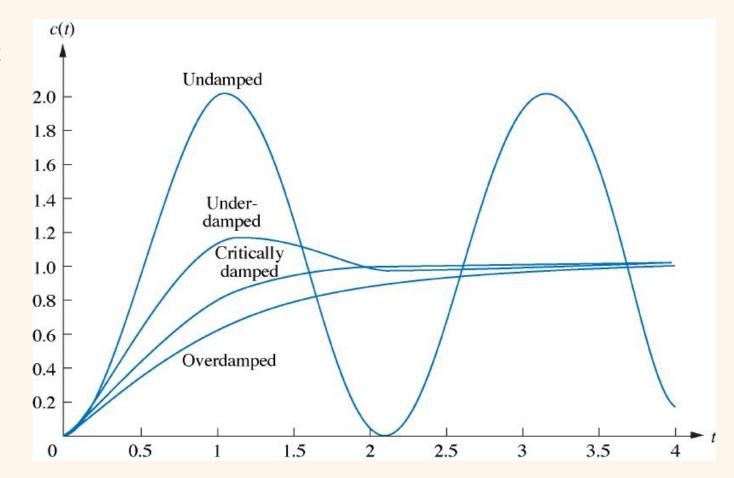
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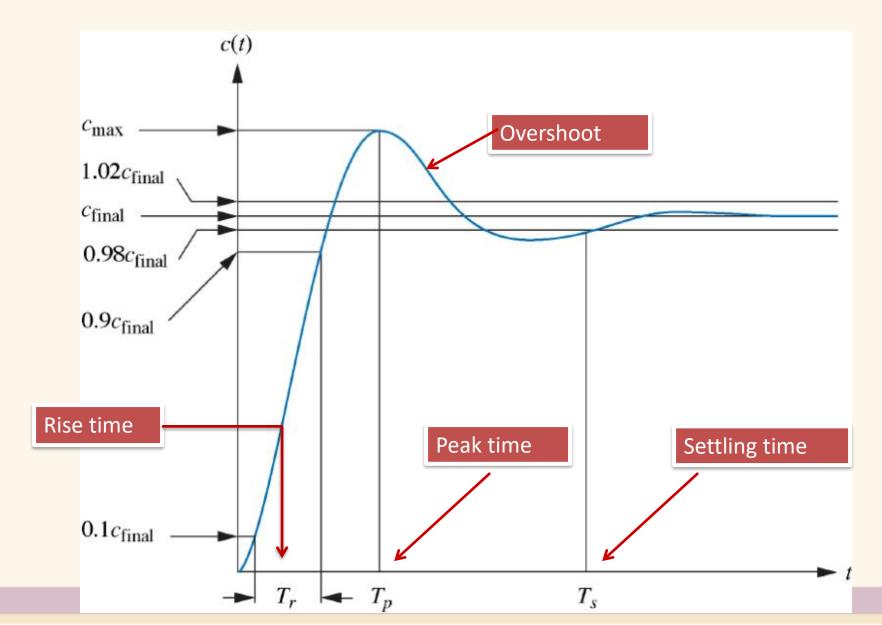
Re-visit: 2nd Order System Response

- Responses:
- Example





Re-visit: Underdamped Response





Re-visit: Performance Specifications

• Peak time,

$$T_p = \frac{\pi}{\omega_d} = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}}$$

• Percent overshoot,

$$OS = e^{-\frac{\zeta \omega_n}{\sqrt{1-\zeta^2}}} \times 100$$

• Settling time,

$$T_{s} = \frac{4}{\zeta \omega_{n}}$$



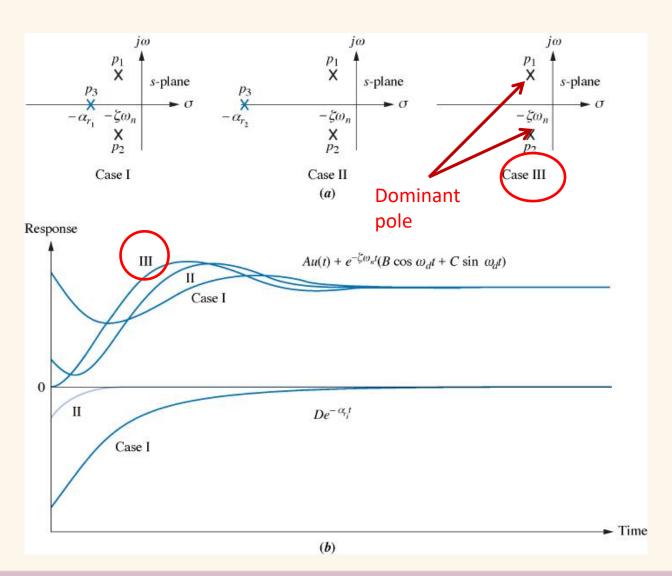
Re-visit: Additional Poles

- The formulae were derived based on a purely second order system. Therefore, the formulae valid only for second order system without zero.
- However, under certain conditions, a system with more that 2 poles can be approximated as a 2nd order system.
- In this case, the complex poles are known as dominant poles.



Re-visit: Additional Poles

- Consider 3 cases:
- Case III behaves as a 2nd order response
- Approximation is valid if the third pole more than 5 times farther to the left
- p1 and p2 are dominant poles.



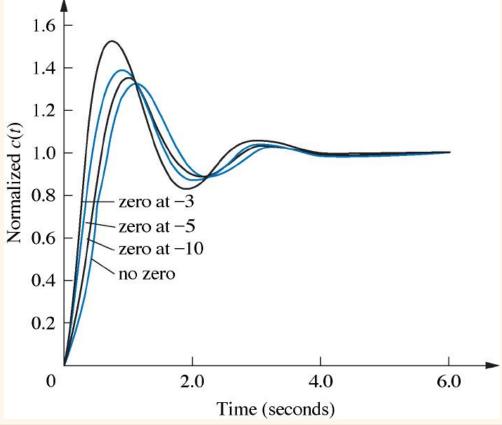


Additional Zeros

- The zeros affect the amplitude of response but do not affect the nature of the response.
- Consider a 2nd order system with poles, $s_{1,2} = -1 \pm j2.828$.

Adding zero at -3, -5 and -10.

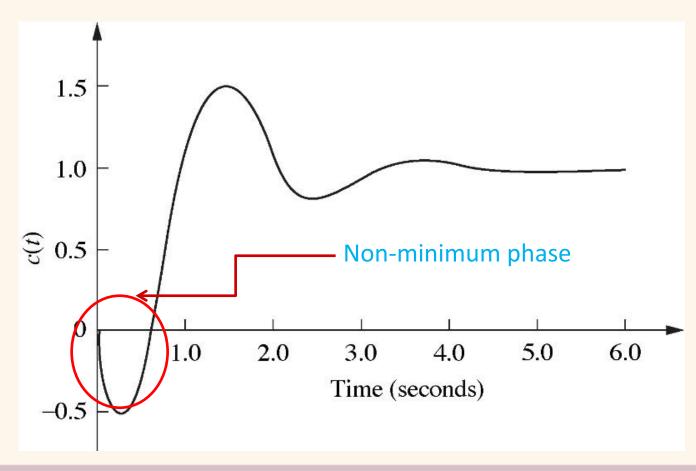
 The closer the zero to the dominant poles, the greater the effect on the transient response





Additional Zeros

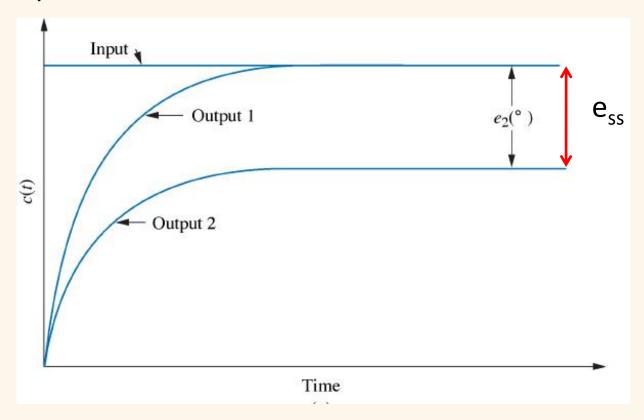
- Adding a positive zero,
- Example





Re-visit: Steady-State Errors

• Steady-state error (e_{ss}) is the difference between the input and output for a prescribed test input as $t \rightarrow \infty$.





Steady-State Errors

e_{ss} can be calculated using the final value theorem.

$$e_{ss} = \lim_{t \to \infty} e(t) = \lim_{s \to 0} sE(s)$$

- where E(s) = R(s) C(s)
- e_{ss} depends on the input and system type.
- Static error constants: position constant (K_p) , velocity constant (K_v) and acceleration constant (K_a) .

$$e(\infty) = \lim_{s \to 0} \frac{sR(s)}{1 + G(s)}$$



2.2 Design with Gain Adjustment



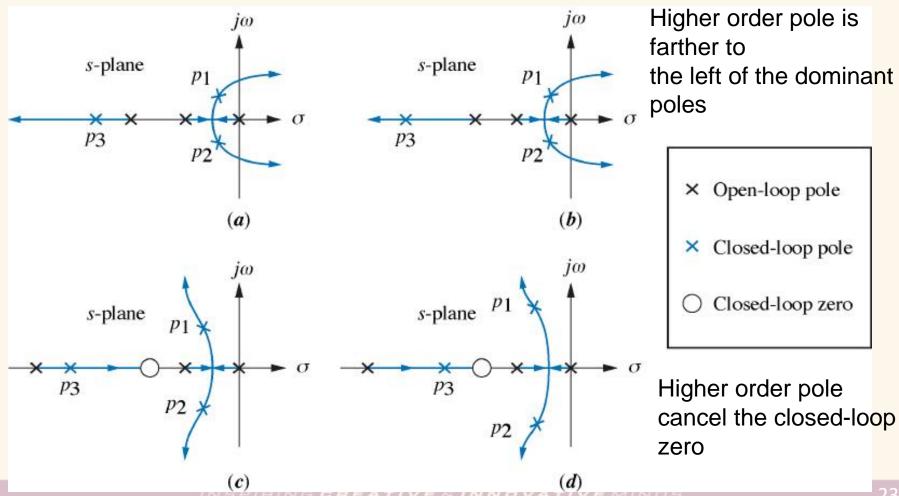
Second-order Approximation

- The formula for the time response specifications only valid for a pure second order system without zero.
- Conditions for second order approximation:
 - Higher order poles are five time farther into the left that the dominant second order poles.
 - Closed-loop zeros are nearly cancelled by the close proximity of higher order poles.



Second-order Approximation

• Figures (b) and (d) yield better approximation.





Design with gain adjustment

- Design procedure for higher order systems :
 - Sketch the root locus.
 - Assume the system is a second order system without zero and find the gain to meet the specifications.
 - Justify the second-order approximation by finding higher order poles.



Example 1

For a unity feedback system that has the forward-path transfer function

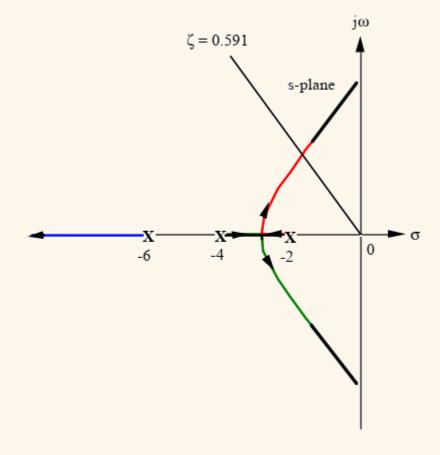
$$G(s) = \frac{K}{(s+2)(s+4)(s+6)}$$
 a) Sketch the root locus

- b) Design the value of K to yield 10% OS
- Estimate the settling time, peak time and steady-state error for the value of K in (b).
- d) Determine the validity of your second-order approximation.



Solution 1

- K = 45.55
- $T_s = 1.97 \text{ s}, T_p = 1.13 \text{ s}, e_{ss} = 0.51$
- Second order approximation is *not valid*.





b. Searching along the $\zeta = 0.591$ (10% overshoot) line for the 180° point yields -2.028+j2.768 with K = 45.55.

c.
$$T_s = \frac{4}{|\text{Re}|} = \frac{4}{2.028} = 1.97 \text{ s}; \ T_p = \frac{\pi}{|\text{Im}|} = \frac{\pi}{2.768} = 1.13 \text{ s};$$

 $\omega_{\rm n}T_r=1.8346$ from the rise-time chart and graph in Chapter 4. Since $\omega_{\rm n}$ is the radial distance to the pole, $\omega_{\rm n}=\sqrt{2.028^2+2.768^2}=3.431$. Thus, $T_r=0.53$ s; since the system is Type 0, $K_p=\frac{K}{2*4*6}=\frac{45.55}{48}=0.949$. Thus,

$$e_{step}(\infty) = \frac{1}{1 + K_p} = 0.51.$$

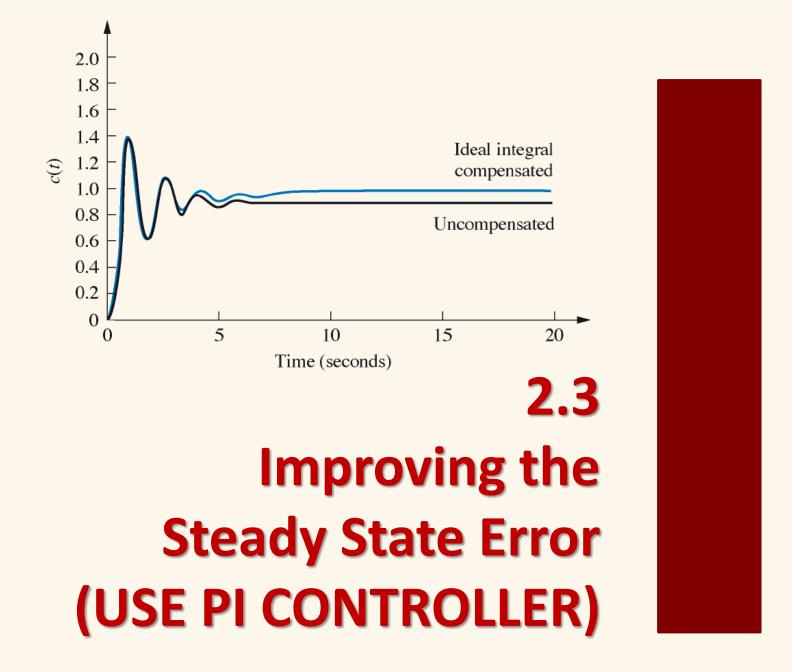
d. Searching the real axis to the left of –6 for the point whose gain is 45.55, we find –7.94. Comparing this value to the real part of the dominant pole, -2.028, we find that it is not five times further. The second-order approximation is not valid.



Design with gain adjustment

- The root locus allows us to choose a proper gain to meet a transient response specification.
- As the gain is varied, we move through different regions of response.
- However, we are limited to those responses that exists along the root locus only.

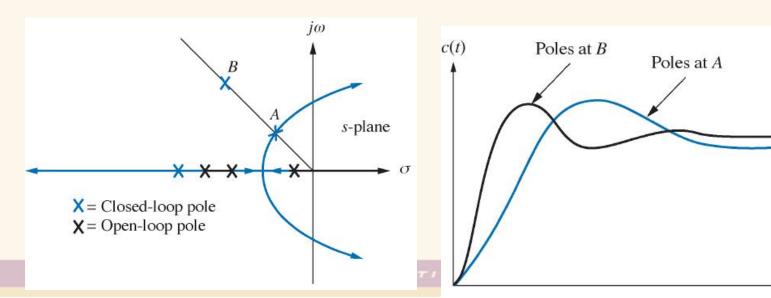






Improving the System Response

- Flexibility in the design can be increased if we can design for response that are not on the root locus.
- Consider the desired transient response defined by OS and settling time.
- With gain adjustment, we can only obtain settling time at A to satisfy the desired OS.
- Point B cannot be on the root locus with the gain adjustment.
- A controller/compensator has to be designed.





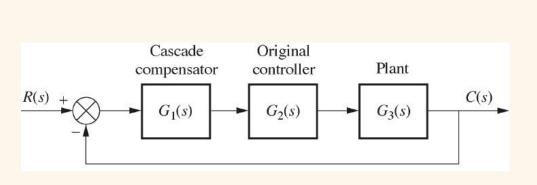
Improving Steady-state Error

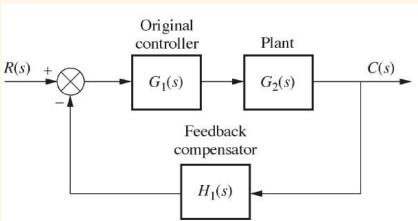
- Compensators can also be used to improve steady-state error.
- By increasing the gain, the ess reduces but OS increases.
- By reducing the gain, OS reduces but ess increases.
- Compensators can be designed to meet the transient response and steady-state error simultaneously.



Configurations

- Two configurations of compensation:
 - Cascade compensation: The compensation network is placed in cascade with the plant.
 - Feedback compensation: The compensator is placed in the feedback path
- Both methods change the OL poles and zeros, thus creating a new root locus that goes through the desired CL pole location.







Proportional-Integral (PI) Controller/Compensator

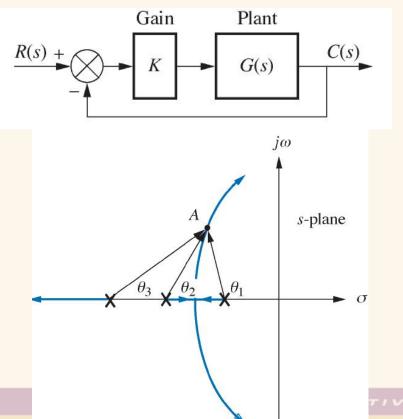
- PI controller/compensator is used to improve steady-state error.
- e_{ss} can be improved by placing an open-loop pole at the origin as this increases the system type by one.

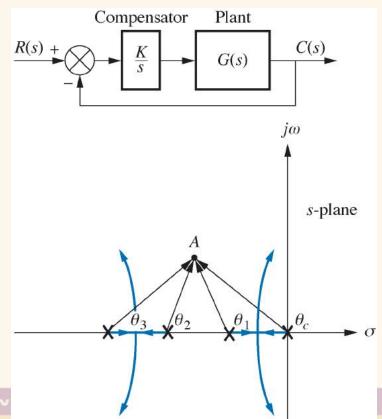
		Type 0		Type 1		Type 2	
Input	Steady-state error formula	Static error constant	Error	Static error constant	Error	Static error constant	Error
Step, $u(t)$	$\frac{1}{1+K_p}$	$K_p = \text{Constant}$	$\frac{1}{1+K_p}$	$K_p = \infty$	0	$K_p = \infty$	0
Ramp, tu(t)	$\frac{1}{K_{\nu}}$	$K_v = 0$	∞	$K_v = \text{Constant}$	$\frac{1}{K_{\nu}}$	$K_v = \infty$	0
Parabola, $\frac{1}{2}t^2u(t)$	$\frac{1}{K_a}$	$K_a = 0$	∞	$K_a=0$	∞	$K_a = \text{Constant}$	$\frac{1}{K_a}$



Proportional-Integral (PI) Controller/Compensator

- Consider a system operating at a desirable response with CL pole at A.
- By adding a pole to increase the system type, A is no longer a CL pole.

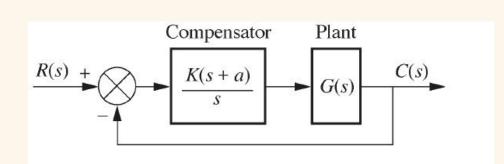


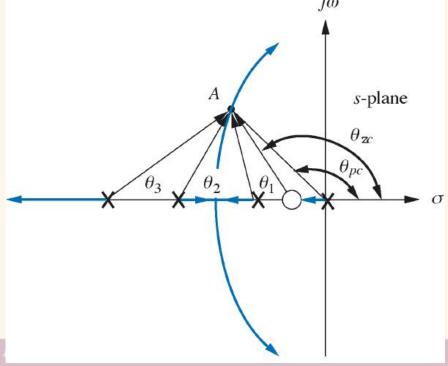




Proportional-Integral (PI) Controller/Compensator

- To solve, a zero close to the pole at the origin has to be added. Thus A is now a CL pole.
- Thus, we have improved the e_{ss} without affecting the transient response. This is known as *PI controller/compensator*.

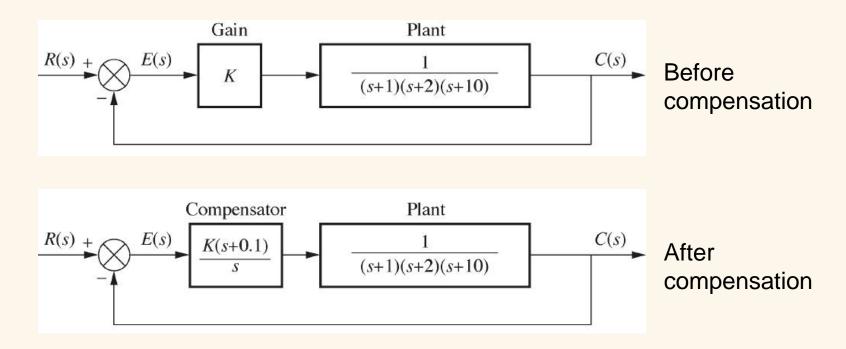




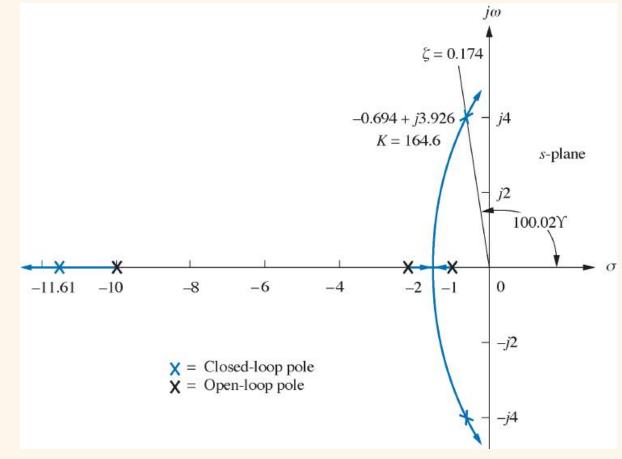


Example 2

• Given a system operating with a damping ratio of 0.174, show that the addition of the PI compensator reduces the e_{ss} to zero for a unit step input without affecting transient response.



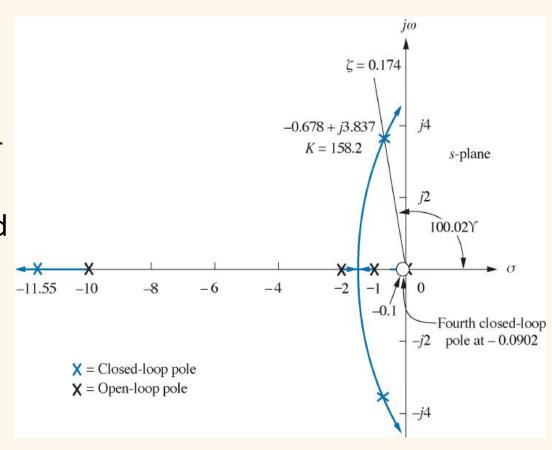




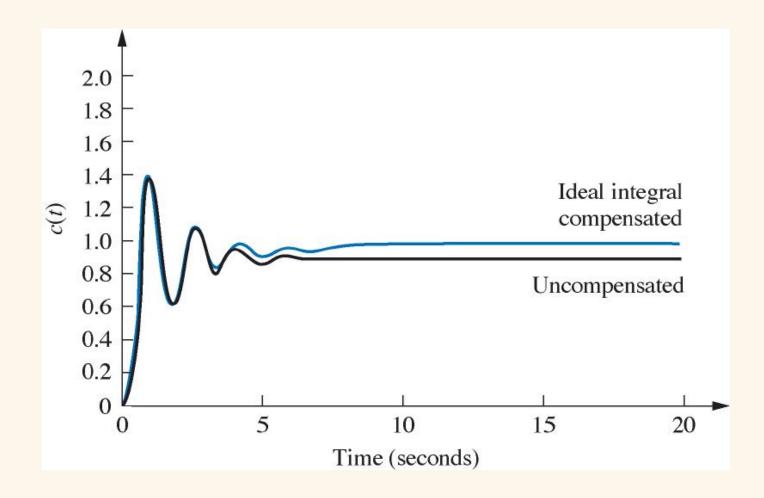
- For a damping ratio 0.174, the dominant poles: $s = -0.694 \pm j$ 3.962, K = 164.6.
- The third pole at -11.61 and second-order approximation is valid.
- $K_p = 8.23$, $e_{ss} = 0.108$.



- Adding a pole at the origin and zero at -0.1.
- With the same damping ratio, dominant poles, $s = -0.678 \pm j.837$, K = 158.2.
- Compensated CL poles and gain are approximately the same as the uncompensated system.
- Both gives the same transient response.
- However, with PI, system is type 1 and e_{ss} = 0.







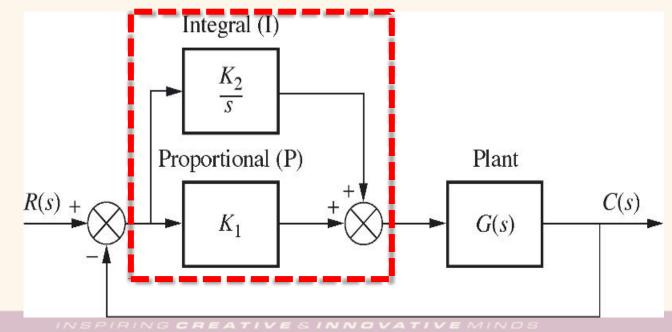


Proportional-Integral (PI) Controller/Compensator

In general, transfer function of PI controller

$$G_{PI}(s) = K_P + \frac{K_I}{s} = \frac{K_P(s + \frac{K_I}{K_P})}{s} = \frac{K(s+z)}{s}$$

Implementation





Proportional-Integral (PI) Controller/Compensator

- PI controller is designed by adding a pole at the origin and a zero at s = -z.
- The zero is chosen to satisfy transient response specifications.
- The value of the zero can be adjusted by varying K_I/K_P
- If the same transient response as the uncompensated system is required, choose the zero close to the origin. Example: s = -0.1, s = -0.01.

$$G_{PI}(s) = K_P + \frac{K_I}{s} = \frac{K_P(s + \frac{K_I}{K_P})}{s} = \frac{K(s + z)}{s}$$



Example 3

For the unity feedback system with

$$G(s) = \frac{K}{(2s+1)(0.5s+1)}$$

- design a PI controller to achieve a transient response as the following:
 - OS = 10 %
 - Settling time 16/3 s
 - Zero steady-state error to unit step input



- Sketch the root locus of the system without the PI Controller.
- Find the CL poles for the required transient response
- Desired CL poles: $s = -0.75 \pm j1$

$$T_s = \frac{4}{|\operatorname{Re}|}$$

- To achieve zero e_{ss}, add a pole at the origin.
- Determine the location of the zero and let $K_p = 1$.

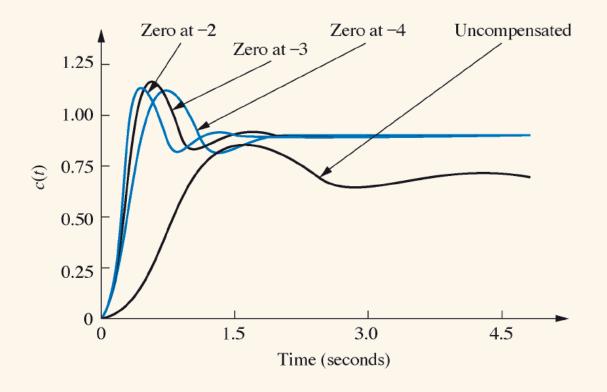
$$G_{PI}(s) = \frac{(s+0.01)}{s}$$

- Re-sketch the root locus of the system with the PI Controller
- Find the CL poles and the gain to achieve the transient and steady state responses. 2.6(s+0.7)
- z = 0.75, K = 2.6

$$G_{PI}(s) = \frac{2.6(s+0.75)}{s}$$

we want the overall system with the same transient response but zero steady state error.





2.4

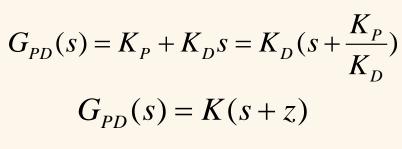
Improving the Transient Response (USE PD CONTROLLER)

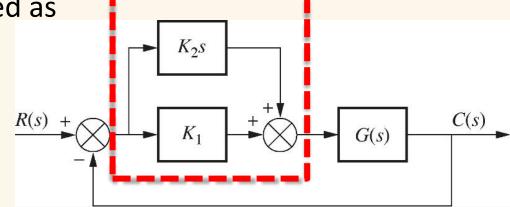


Proportional-Derivative (PD) Controller/Compensator

- PD controller is used to improve transient response and maintaining the steady-state error.
- Transient response can be improved by adding a single zero to the forward path of the feedback control system.

This zero can be represented as

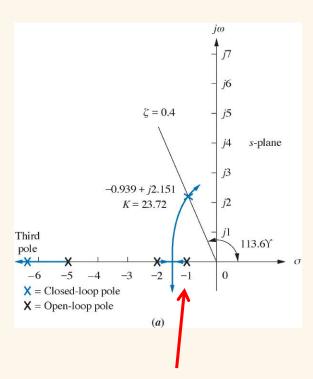




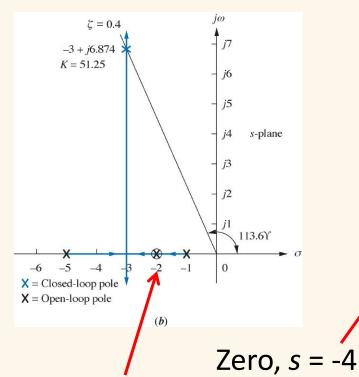
 A proper selection of the zero can improve the system's transient response.



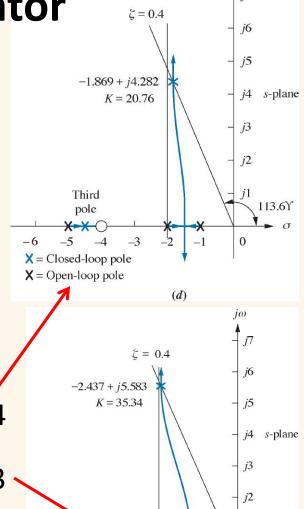
Proportional-Derivative (PD)
Controller/Compensator



Uncompensated



Zero, s = -2



(c)

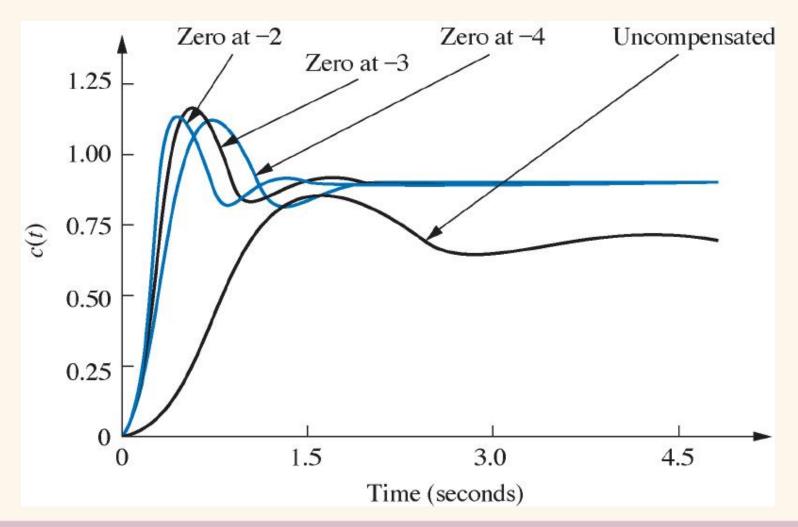


Proportional-Derivative (PD) Controller/Compensator

- All compensated systems operate at damping ratio, 0.4 as the uncompensated system.
- For all the compensated systems, the CL poles have more negative real and larger imaginary part as compared to the uncompensated system.
- Thus, the compensated systems operate with shorter settling time and peak time.
- System (b) gives the best transient response.
- In summary, adding a zero on the forward path improves the transient response.
- A proper selection of the value of the zero is required.



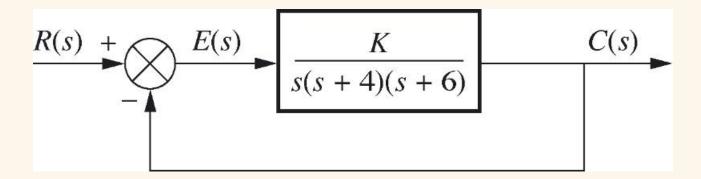
Proportional-Derivative (PD) Controller/Compensator





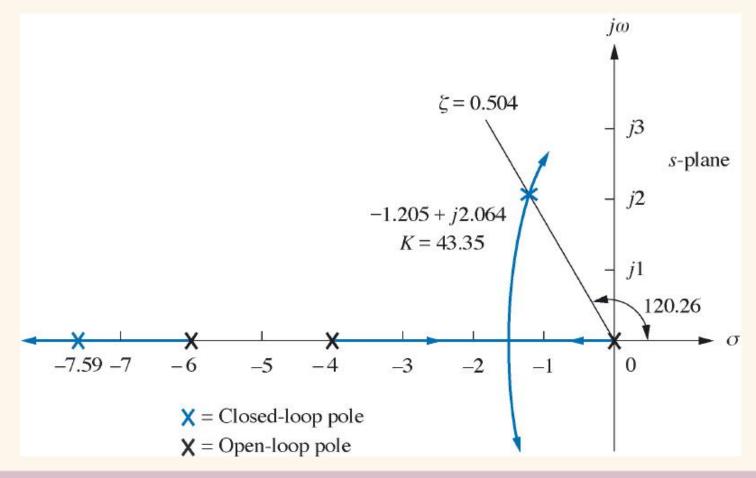
Example 4

• Given the system, design a PD controller to yield 16 % overshoot, with a threefold reduction in the settling time.





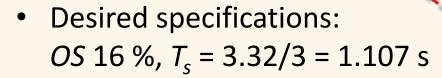
Draw the root locus without PD Controller





- To get 16% OS, the CL poles can be found at with $T_s = 3.32 \text{ s}$, $T_p = 1.52 \text{ s}$
- $e_{ss} = 1/K_v = 0.55$ (by calculation when K = 43.35)
- Third pole at s = -7.59. Thus second order approximation is

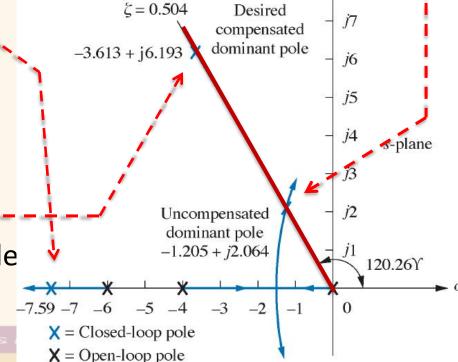
valid.



• Desired CL poles, (calculation)
$$T_s = \frac{4}{\sigma} \qquad s = -3.61 \pm \text{j}6.193 - - - -$$

How to calculate? - Use triangle

$$\omega = 3.613 \tan(180^{\circ} - 120.26^{\circ}) = 6.193$$





- To locate the zero, use angle property:
 - Total angles from poles to the root (275°) total angles from

zeros to the root = 180°

• The angle = 95.6° .

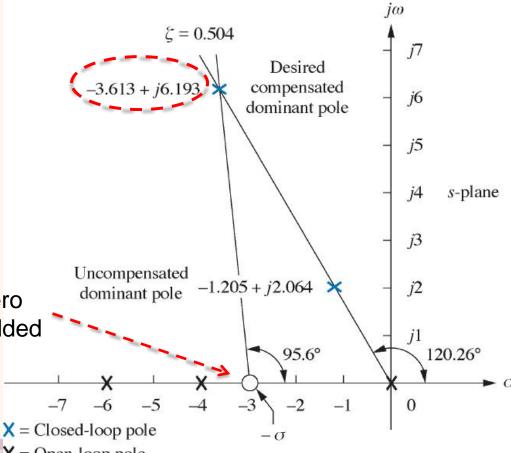
How?

• Thus, $z = 3 = \sigma$

How? Get the triangle.

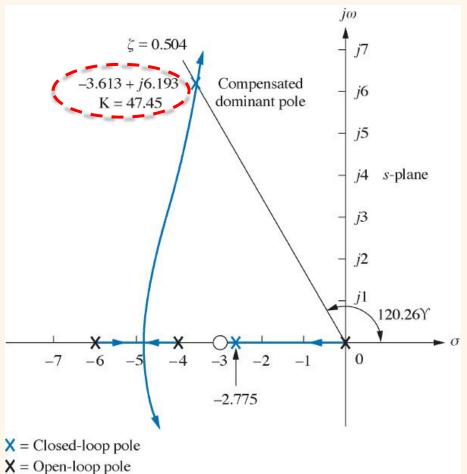
$$\frac{6.193}{3.613 - \sigma} = \tan(180^\circ - 95.6^\circ)$$

The zero to be added

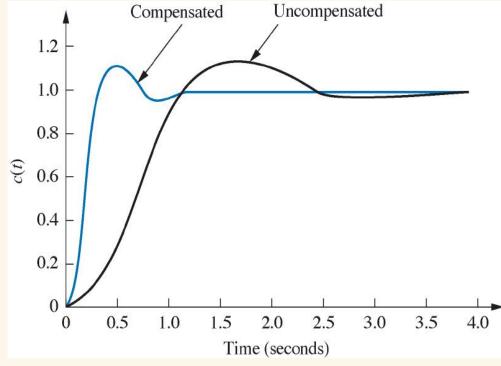




• The complete root locus.



Response





PD controller

$$G_{PD}(s) = K_P + K_D s = K_D (s + \frac{K_P}{K_D})$$

- $K_D = 47.45$, $K_P = 142.35$.
- A complete system:

$$G_{PD}(s)G(s) = \frac{47.45(s+3)}{s(s+4)(s+6)}$$



Example 5

For a unity feedback system with

$$G(s) = \frac{50}{s(s+5)}$$

- a) Determine percent OS and settling time of the system
- b) Design a controller to have 50 % reduction in the OS and four fold improvement in the settling time.



• CLTF
$$G(s) = \frac{50}{s^2 + 5s + 50} = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2}$$

- Do we need to draw the root locus of the system without controller? Can we find the specification needed?
 - No need to draw RL.
- OS = 30.45 %, $T_s = 1.6 \text{ s}$.
- Desired specifications: *OS* = 15.23 %, *Ts* = 0.4 s.
- Desired CL poles: $s = -10 \pm j16.69$
- Using angle property, z = 25.29
- $K = 15 = K_D$

$$G_{PD}(s)G(s) = \frac{15(s+25.29)}{s(s+5)}$$

Transfer function:



2.5 Improving Transient Response and Steady-State Error (USE PID CONTROLLER)



Proportional-Integral-Derivative (PID) Controller

- PID controller is used to improve steady-state error and transient response independently.
- PID controller is a combination of PI and PD controllers.
- We first design for transient response and then design for steady-state error.
- Transfer function:

$$G_{PID}(s) = K_P + \frac{K_I}{s} + K_D s$$

$$= \frac{K_{P}s + K_{I} + K_{D}s^{2}}{s} = \frac{K_{D}\left(s^{2} + \frac{K_{P}}{K_{D}}s + \frac{K_{I}}{K_{D}}\right)}{s}$$

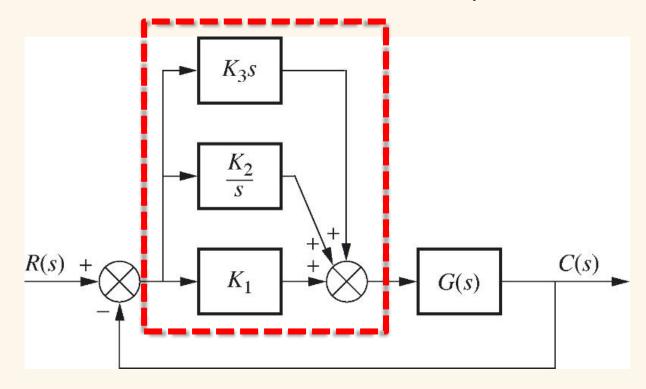
$$G_{PID}(s) = G_{PI}(s) * G_{PD}(s) \qquad G_{PID}(s) = K \frac{(s + z_{1})}{s} * (s + z_{2})$$

or
$$G_{PID}(s) = G_{PI}(s) * G_{PD}(s)$$
 $G_{PID}(s) = K \frac{(s + z_1)}{s} * (s + z_2)$



Proportional-Integral-Derivative (PID) Controller

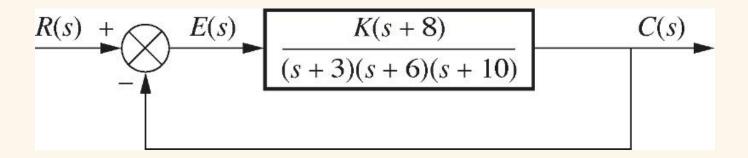
- Implementation:
- PID controller consists of two zeros and a pole at the origin.





Example 6

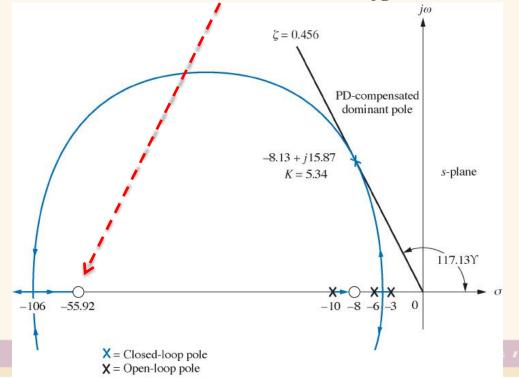
Given the system, design a PID controller so that the system can operate with a peak time that is two-thirds of the uncompensated system at 20 % overshoot, and with zero steady-state error for a step input.

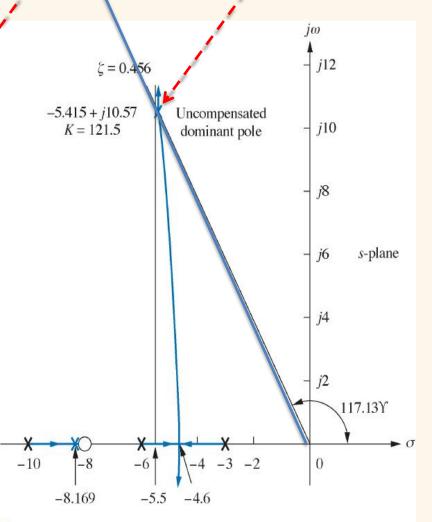




- Do we need to draw the original root locus?
- Peak time = 0.297 s.
- Desired pole: $s = -8.13 \pm j15.87'$
- Angle of the zero = 18.37° .

• Thus z = 55.92. $G_{PD}(s) = s + 55.92$





X = Closed-loop pole

X = Open-loop pole



• Design PI controller to reduce the error to zero. Choose

$$G_{PI}(s) = \frac{s + 0.5}{s}$$

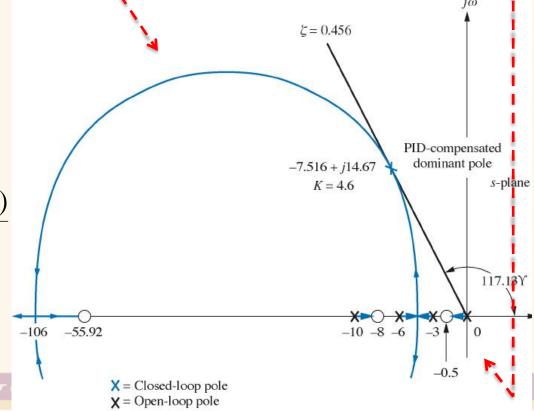
The root locus with PID controller:

$$G_{PID}(s) = \frac{K(s+55.92)(s+0.5)}{s}$$

$$= \frac{4.6(s+55.92)(s+0.5)}{s}$$

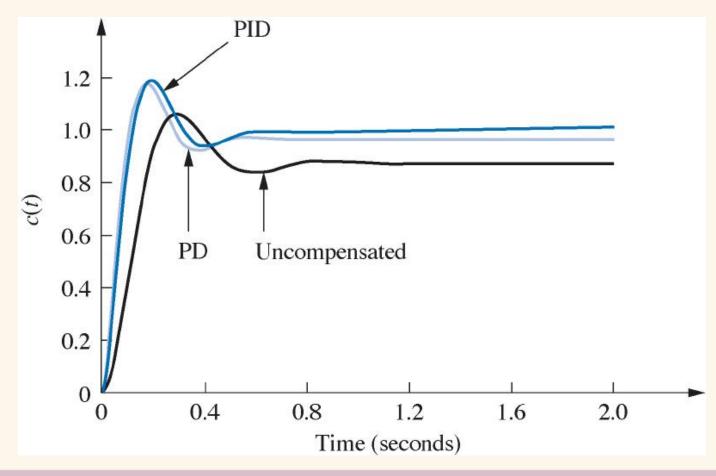
$$= \frac{4.6(s^2+56.42s+27.96)}{s}$$

• Thus, $K_D = 4.6$, $K_P = 259.5$, $K_I = 128.6$





The response





2.6 Tuning the PID using ZieglerNichols Technique



Ziegler-Nichols Method

- This method is used when the systems' models cannot or difficult to be obtained either in the form of differential equation or transfer function.
- The PID controller is widely used in the industries in which the K_p , K_i and K_d parameters often can be easily adjusted.
- Ziegler-Nichols has introduced an effective method for the parameters adjustment by using the *ultimate cycle method*.



Ziegler-Nichols steps

- The steps are as follows:
 - 1. Set $K_d = K_i = 0$ (to minimise the effect of derivation and integration)
 - 2. Increase the K_p gain until the system reach the critically stable and oscillate (i.e. when the closed-loop poles located at the imaginary axis). Obtain the gain, K_g and the oscillation frequency, ω_g on that time.
 - 3. Calculate the K_p gain that is supposedly needed using formula in the table.
 - 4. Based on the required type of controller, calculate the necessary gain using formulas in the table.



Ziegler-Nichols formulas

Controller	Optimum gain
a) Proportional: P	$K_p = 0.5 K_g$
b) PI	$K_p = 0.45K_g , K_i = \frac{0.54K_g}{\omega_g}$
c) PID	$K_{p} = 0.6K_{g}$, $K_{i} = \frac{1.2K_{g}}{\omega_{g}}$, $K_{d} = \frac{0.3K_{g}\omega_{g}}{4}$

$$G_{pid}(s) = K_p + K_d s + \frac{K_i}{s}$$



Example 7

 Design a PID controller for unity feedback system with the open loop system which is given as:

$$G(s) = \frac{K300}{s(s^2 + 30s + 100)}$$

$$[K_p = 6, K_i = 1.2, K_d = 7.5]$$



Conclusion

We have covered the design of

- ✓ Proportional (Gain) Controller
- ✓ PI Controller
- ✓ PD Controller
- ✓ PID Controller

THE END