

CE 3305 Fluid Mechanics
Exam 2
Spring 2014

1. A 6-in. diameter cylinder falls at a rate of 4 ft/s in an 8-in. diameter tube containing an incompressible liquid. What is the mean velocity of the liquid (with respect to the tube) in the space between the falling cylinder and the tube wall?¹

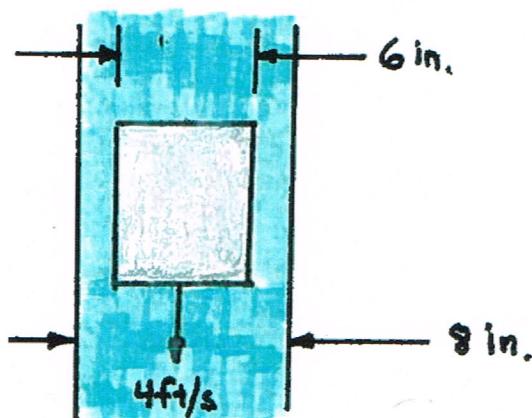


Figure 1: Falling cylinder in a tube.

C.F. ATTACHED TO AND MOVING WITH CYLINDER

C.F. MOVES DOWN AT 4 ft/s; V. RELATIVE TO C.F. IS $4A/s$

SKETCH OF AREAS, FLOWS/VELOCITY STATE C.F. MOVES WITH CYLINDER +5

$A_1 = \frac{\pi}{4} \left(\frac{8}{12}\right)^2 = 0.349 \text{ ft}^2$ (+2) #, UNITS

$A_2 = \frac{\pi}{4} \left(\frac{8}{12}\right)^2 - \frac{\pi}{4} \left(\frac{6}{12}\right)^2 = 0.153 \text{ ft}^2$ (+2) #, UNITS

¹Continuity

CONTINUED



20 POINTS TOTAL
THIS PROBLEM

Problem 1 (Continued)

GOVERNING EQUATION(s)CONTINUITY $(+1)$

$$\partial = \frac{d}{dt} \int \rho dV + \int \rho \vec{V} \cdot d\vec{A} \quad (+1)$$

$\cancel{\downarrow}$
C.S.

$$\partial; \frac{dV}{dt} = 0$$

$$\partial = - \rho A_1 V_1 + \rho A_2 V_2 \quad (+1) \quad \text{WATER, } \rho = \text{CONSTANT} \quad (+1)$$

 \therefore

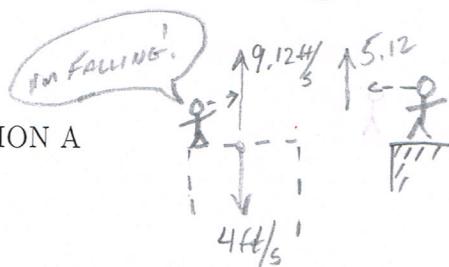
$$A_1 V_1 = A_2 V_2$$

$$V_2 = \frac{A_1 V_1}{A_2} = \frac{(0.349 \text{ ft}^2)(4 \text{ ft/s})}{(0.153 \text{ ft}^2)} = 9.12 \text{ ft/s} \quad \begin{matrix} (+1) \\ (+2) \end{matrix} \quad \#_{\text{units}}$$

HOWEVER, C.S. ② IS MOVING DOWNWARD
AT 4 ft/s.

$$\therefore V = V_2 - V_{\text{CYLINDER}} = 9.12 \text{ ft/s} - 4 \text{ ft/s} = 5.12 \text{ ft/s} \quad \begin{matrix} (+2) \\ (+2) \end{matrix} \quad \#_{\text{units}}$$

REVISION A



2. How long will it take the water surface to drain from 9 ft. to 1 ft. for the 12 ft. diameter tank in Figure 2. Assume the outlet pipe is 2 in. in diameter and the pressure at the outlet is atmospheric.²

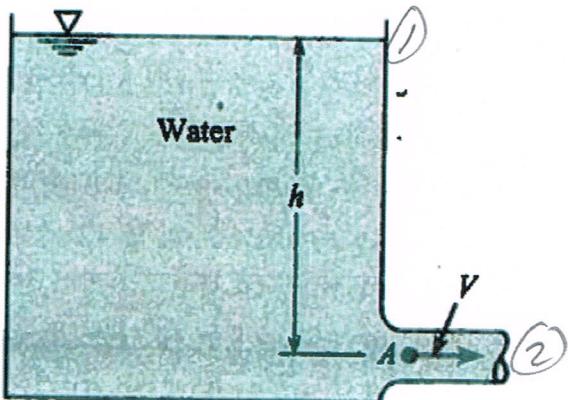
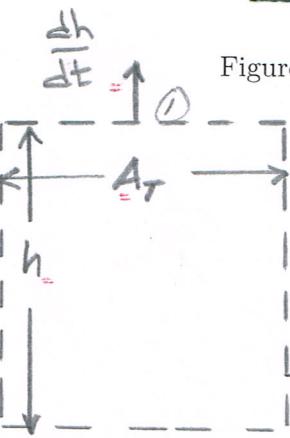


Figure 2: Reservoir draining through a pipe.

C.F. IS
FREE SURFACE
TO JET.
TOP MOVES
 $\frac{dh}{dt}$



SKETCH OF CV
WITH FLOWS, AREAS
AND NOTE CV MOVES
WITH FREE

SURFACE $\frac{dh}{dt}$ (+5)

KNOWN

$$A_T = \frac{\pi(12)^2}{4} = 113.09 \text{ ft}^2 \quad (+2) \text{ #, UNITS}$$

$$A_j = \frac{\pi(\frac{2}{12})^2}{4} = 0.022 \text{ ft}^2 \quad (+2) \text{ #, UNITS}$$

BERNOULLI From FREE SURFACE TO JET (+1)

$$\frac{P_1}{\rho} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\rho} + \frac{V_2^2}{2g} + z_2 \quad (+1)$$

Assume $V_1 = 0$ $z_1 = h$ $P_1 = 0$ gage $P_2 = 0$ gage $z_2 = 0$ datum

$$V(t) = \sqrt{2gh(t)} \quad (+1)$$

²Bernoulli, Continuity, Related Rates

Problem 2 (Continued)

CONTINUITY (1)

$$\partial = \frac{d}{dt} \int_C \rho dV + \int_S \rho V \cdot d\bar{A} = (1)$$

$$\partial = \cancel{\lambda} \cdot \frac{dh}{dt} \cdot A_T + \cancel{\lambda} V_j A_j = (1) \quad \text{WATER } \rho = \text{CONSTANT}$$

$$\frac{dh}{dt} = - \frac{V_j A_j}{A_T} = - \frac{A_j}{A_T} \sqrt{2gh}$$

SEPARATE & INTEGRATE

$$\frac{1}{\sqrt{h}} dh = - \frac{A_j}{A_T} \sqrt{2g} dt \quad (1)$$

$$\int_{t_0}^t \frac{dh}{\sqrt{h}} = - \frac{A_j \sqrt{2g}}{A_T} \int dt$$

CONSTANT OF INTEGRATION

$$2\sqrt{h} = - \frac{A_j \sqrt{2g}}{A_T} t + C \quad (1) \quad (\text{DEFINITE INTEGRAL}) \\ \text{OK TOO!}$$

Problem 2 (Continued)

APPLY BOUNDARY (INITIAL) CONDITIONS

$$t=0, h = 9 \text{ ft}$$

(+1) (IF DEFINITE
INTEGRAL THEN)

$\int_1^9 = -\int_T^0$ MAY SWITCH
DEPENDING
ON SIGN
LOCATION

$$2\sqrt{9 \text{ ft}} = C$$

$$2\sqrt{h} = -\frac{A_j \sqrt{2g}}{A_T} t + 2\sqrt{9 \text{ ft}}$$

SET $h = 1 \text{ ft}$, SOLVE FOR t

$$-(2\sqrt{1} - 2\sqrt{9}) \frac{A_T}{A_j \sqrt{2g}} = t$$

$$-(2 - 6) \left(\frac{113.09}{0.022} \right) \left(\frac{1}{\sqrt{2(32.2)}} \right) = 2562.23 \text{ seconds}$$

(+2) #, UNITS

3. A turbofan engine (Figure 3) on a commercial jet takes in air, a portion of which passes through the compressor, combustion chamber where fuel is added and ignited, and the turbine, the remainder of the air bypasses the compressor and is accelerated by the fans. The mass flow rate of bypass air to the mass flow rate through the compressor-combustor-turbine path is called the "bypass ratio." The total flow rate of air entering the turbofan engine is 300 kg/s with a velocity relative to the engine of 300 m/s. The engine has a bypass ratio of 2.5. The bypass air exits at 600 m/s, whereas the air through the compressor-combustor-turbine path exits at 1000 m/s. What is the thrust generated by the turbofan engine? Clearly show the control volume and application of the linear momentum equation.³

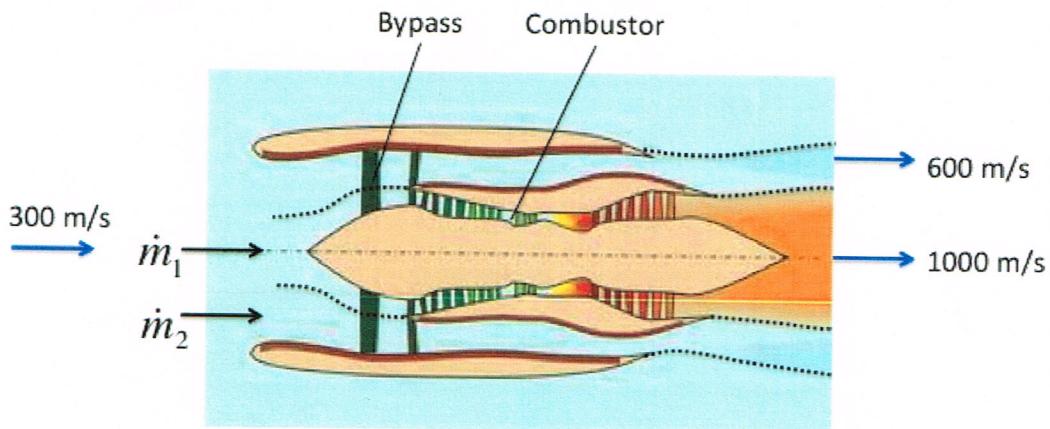
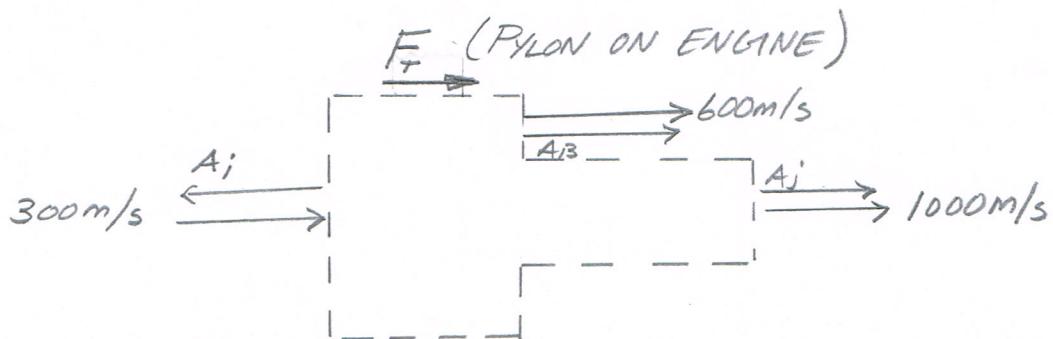


Figure 3: Typical turbofan engine



CV ENCLOSSES JET ENGINE
THRUST IS EQUAL/OPPOSITE OF
PYLON FORCE ON ENGINE

³Continuity, Linear Momentum

SHOWN AS F_T

REVISION A

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CV SKETCH NEEDS TO SHOW AREAS, FLOWS. OK IF \dot{m}_1, \dot{m}_2 +5 PTS
REPLACE INLET MASS FLOW.

Problem 3 (Continued)

MOMENTUM (+1)

$$\sum F = \frac{d}{dt} \int_{CV} \rho V dV + \int_{CS} \rho V (\bar{V} \cdot d\bar{A}) \quad (+1)$$

\downarrow

$$0 \frac{dV}{dt} = 0$$

$$\left\{ \begin{array}{l} F_T = -\rho A_{INLET} V_{INLET} \cdot V_{INLET} + \rho A_j V_j V_j + \rho A_B V_B V_B \\ \dot{m}_1 + \dot{m}_2 = \dot{m}_1 + \dot{m}_2 \end{array} \right.$$

(+6) pts
NEED TO RECOGNIZE THAT
 $\rho A V^2 = \dot{m} V$
OTHERWISE PROBLEM REALLY HARD TO SOLVE

KNOWN

$$\dot{m}_1 + \dot{m}_2 = 300 \text{ kg/s} \quad (\text{GIVEN}) \quad (+1)$$

$$\frac{\dot{m}_2}{\dot{m}_1} = 2.5 \quad (\text{GIVEN BY PASS RATIO}) \quad (+1)$$

$$V_{INLET} = 300 \text{ m/s}$$

$$V_j = 1000 \text{ m/s}$$

$$V_B = 600 \text{ m/s}$$

Problem 3 (Continued)

$$F_T = -(\dot{m}_1 + \dot{m}_2)V_{INLET} + \dot{m}_1V_j + \dot{m}_2V_B$$

CONTINUITY

$$0 = \frac{d}{dt} \int_{CV} \rho dV + \int_{CS} \rho V \cdot dA \quad (+1)$$

$$0 = -(\dot{m}_1 + \dot{m}_2) + \dot{m}_1 + \dot{m}_2 = -(\dot{m}_1 + \dot{m}_2) + \dot{m}_1 + 2.5\dot{m}_1$$

SOLVE FOR \dot{m}_1

$$3.5\dot{m}_1 = \dot{m}_1 + \dot{m}_2 = 300 \text{ kg/s}$$

$$\dot{m}_1 = \frac{300 \text{ kg/s}}{3.5} = 85.71 \text{ kg/s} \quad (+1)$$

$$\dot{m}_2 = 300 \text{ kg/s} - 85.71 \text{ kg/s} = 214.28 \text{ kg/s} \quad (+1)$$

SUBSTITUTE INTO MOMENTUM AND SOLVE FOR F_T

$$\begin{aligned} F_T &= -(300 \text{ kg/s})(300 \text{ m/s}) + (85.71 \text{ kg/s})(1000 \text{ m/s}) + (214.28 \text{ kg/s})(600 \text{ m/s}) \\ &= -90,000 \frac{\text{kg m}}{\text{s}^2} + 85,710 \frac{\text{kg m}}{\text{s}^2} + 128,568 \frac{\text{kg m}}{\text{s}^2} \\ &= 124,278 \frac{\text{kg m}}{\text{s}^2} = 124.2 \text{ kN} \quad (+2) \# \text{, UNITS} \end{aligned}$$

F_T IS FORCE OF PYLON ON ENGINE,
HENCE ENGINE "PUSHES" ON
PYLON 124.2 kN ←

4. Hydraulic mining uses high speed jets of water to demolish hills and the economically valuable material is then separated from the resulting slurry. Water flows from a reservoir through a pipe and discharges through a hydraulic monitor (a nozzle) as shown in Figure 4. The head loss in the pipe itself is $h_L = 0.025 \frac{L}{D} \frac{V^2}{2g}$ where L and D are the length and diameter of the pipe and V is the velocity in the pipe. What is the discharge of the water. (Assume $\alpha=1.0$) ⁴

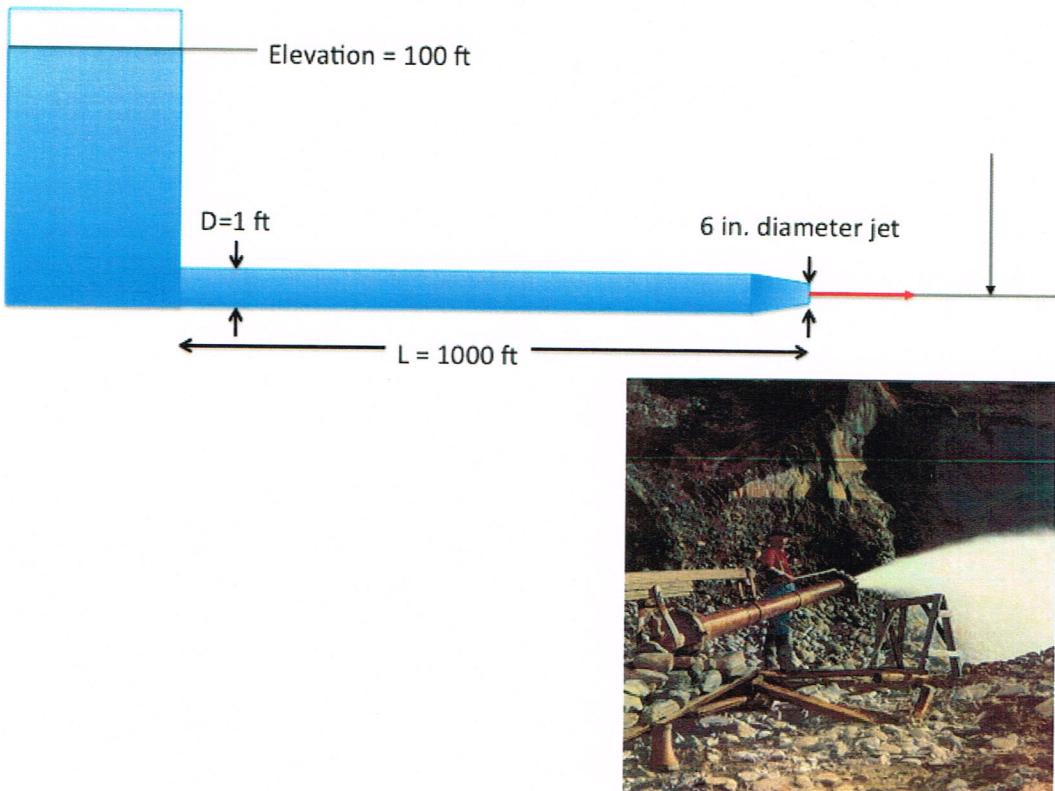
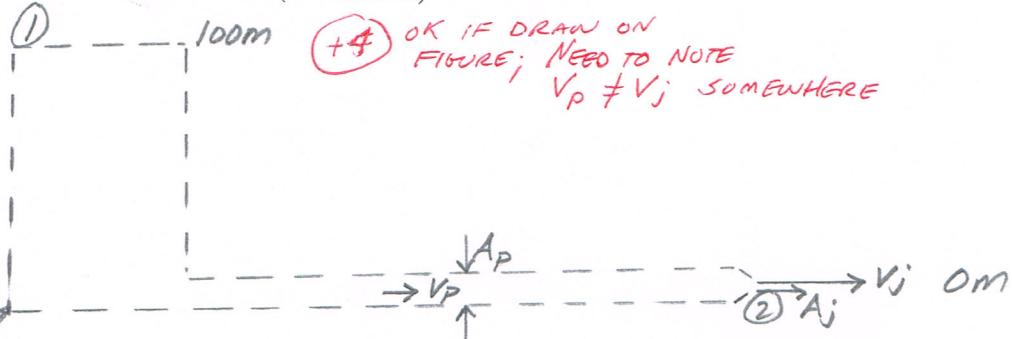


Figure 4: Schematic of monitor and supply line. Photograph of a monitor in use in California, circa late 1800s.

⁴Energy

20 POINTS TOTAL
THIS PROBLEM

Problem 4 (Continued)



C.V. FIXED TO
FREE SURFACE, ALL H₂O IN SYSTEM.

(+1)
ENERGY FROM RESERVOIR TO JET

$$\frac{P}{\rho} + \frac{V_1^2}{2g} + z_1 = \frac{P_2}{\rho} + \frac{V_2^2}{2g} + z_2 + h_L \quad (+1)$$

EVIDENCE OF REASON CANCELLED NEGLECT TERMS

Diagram showing levels: Reservoir Surface (z0), Ogove (z1), Odatum (z2), and Jet (zj). Velocity Vp is at Ogove, and Vj is at Jet. A curved arrow points from Vp to Vj.

$$z_1 = \frac{V_2^2}{2g} + 0.025 \frac{L}{D} \frac{V_p^2}{2g} \quad (+2)$$

NEED TO NOTE V_j & V_p ARE DIFFERENT

THIS IS AT JET V₂ = V_j

EVALUATE CONSTANTS

$$0.025 \frac{L}{D} = \frac{0.025(1000 \text{ ft})}{1 \text{ ft}} = 25$$

$$V_j = \text{VELOCITY JET}$$

$$V_p = \text{VELOCITY IN PIPE}$$

CONTINUITY (+1)

$$V_p A_p = V_j A_j$$

$$\frac{V_p A_p}{A_j} = V_j \quad (+1)$$

$$A_p = \frac{\pi (1A)^2}{4} = 0.785 \text{ ft}^2 \quad (+2)$$

$$A_j = \frac{\pi (0.5A)^2}{4} = 0.196 \text{ ft}^2 \quad (+2)$$

$$\therefore V_j = 3.99 V_p \quad (+1)$$

$$V_j \sim 4 V_p$$

Problem 4 (Continued)

EXPRESS EVERYTHING IN V_p

$$z_1 = \frac{(3.9V_p)^2}{2(32.2)} + \frac{25V_p^2}{2(32.2)}$$

$$z_1 = \frac{16V_p^2 + 25V_p^2}{64.4} = \frac{41V_p^2}{64.4} \quad (+1)$$

SUBSTITUTE IN NUMBERS AND SOLVE FOR V_p

$$100 = \frac{41V_p^2}{64.4} \quad V_p^2 = \frac{64.4 \cdot 100}{41} = 157.07$$

$$V_p = \sqrt{157.07} = 12.5 \text{ ft/s} \quad (+1)$$

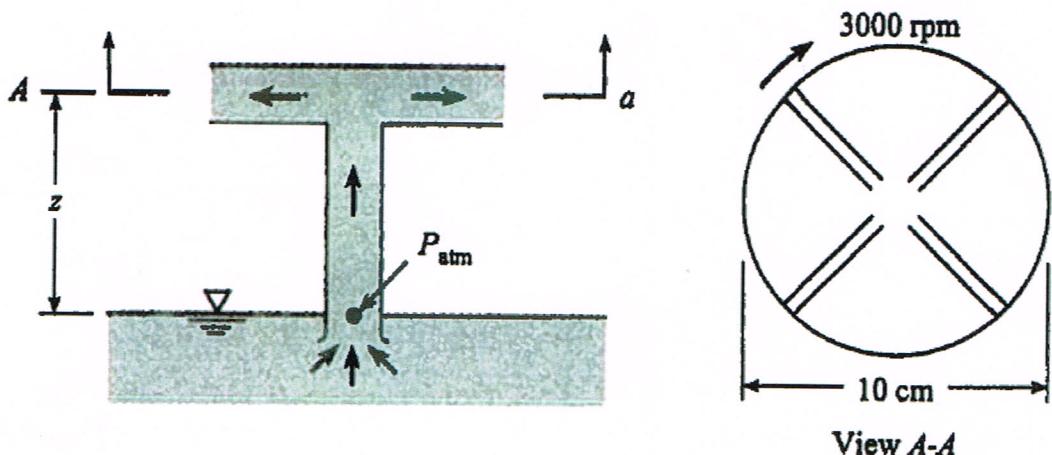
CONTINUITY

$$Q = A_p V_p$$

$$= (0.785 \text{ ft}^2) (12.5 \text{ ft/s}) = 9.81 \text{ ft}^3/\text{s}$$

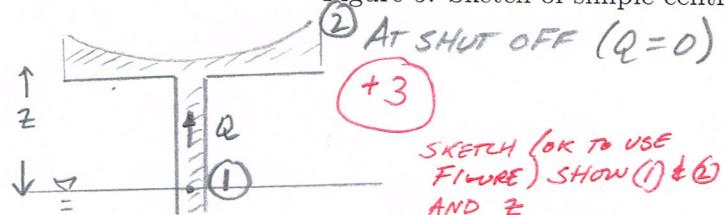
+2
ft, units

5. A simple centrifugal pump consists of a 10 cm disk with radial ports as shown. Water is pumped from a reservoir through a central tube on the axis. The wheel spins at 3000 rev/min, and the liquid discharges to atmospheric pressure. To establish the maximum height for operation of the pump (called the shutoff height), assume the flow rate is nearly zero and the pressure at the pump intake is atmospheric pressure. Calculate the maximum operational height z for the pump.⁵



View A-A

Figure 5: Sketch of simple centrifugal pump



IF $Q=0$, $V_r=0$
ONLY $V_t=r\omega$
NON ZERO
 ω , UNIT
 $r = \frac{0.1}{2} = 0.05 \text{ m}$
(GIVEN)

APPLY ROTATIONAL FORM EULER'S EQUATION

$$\frac{P_1}{\gamma} + \frac{Z_1}{\gamma} - \frac{\omega^2 r_1^2}{2g} = \frac{P_2}{\gamma} + \frac{Z_2}{\gamma} - \frac{\omega^2 r_2^2}{2g}$$

\checkmark Ogage \checkmark Odatum

$= 0$ r_1 IS AXIS OF ROTATION

⁵Euler and Bernoulli

REVISION A

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--- TEXTBOOK FORM IS

$$\frac{P_1}{\gamma} + \frac{\rho g z_1}{\gamma} - \frac{\rho \omega^2 r_1^2}{2} = C$$

WAY EASIER IF DIVIDE
EVERYTHING BY γg

20 POINTS TOTAL
THIS PROBLEM

Problem 5 (Continued)

$$\theta = z_2 - \frac{\omega^2 r^2}{2g} \quad z_2 \text{ IS SHUTOFF HEIGHT}$$

+2

$$z = \frac{\omega^2 r^2}{2g}$$

$$\omega = \frac{3000 \text{ rev}}{\text{min}} \cdot \frac{1 \text{ min}}{60 \text{ sec}} \cdot \frac{2\pi \text{ rads}}{\text{rev}}$$

$$= 314.15 \text{ rad/sec}$$

+1

$$z = \frac{(314.15 \frac{\text{rad}}{\text{s}})^2 (0.05 \text{ m})^2}{2(9.8 \text{ m/s}^2)} = 12.6 \text{ m}$$

UNITS
+2

Problem 5 (Continued)

$$\begin{array}{r} 20 \\ 20 \\ 20 \\ 20 \\ 20 \\ \hline 100 \end{array}$$