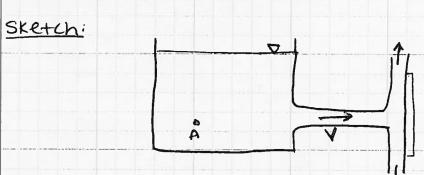
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#### CE 3305 Engineering Fluid Mechanics Exercise Set 15 Spring 2014

- 1. Problem 6.13, pg 239
- 2. Problem 6.64, pg 246
- 3. Problem 6.82, pg 248

COURSECE 3305 SHEET 1 OF

6.13) A horizontal water jet at 70°F issues from a circular orifice in a large tank. The jet strikes a vertical plate that is normal to the axis of the jet. A force of lead lot is needed to hold the plate in place against the action of the jet. If the pressure in the tank is 25 psig at point A, what is the diameter of the jet just downstream of the orifice?

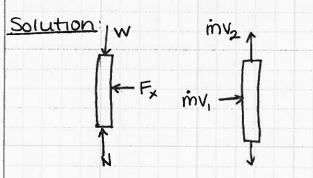


### known !

## unknown:

# Governing Equation

Bernoulli & momentum equation.



From the inside of tank to nozzle exit Bernoulli's Eqn.

$$V_1 = \int \frac{aP_A}{P} = \int \frac{a(a5 + bf/m^2)(144 m^2/ft^2)}{(b2.4 + bf/ft^3)(\frac{5lug}{32.2 + f/s^2})} = 60.9 ft/s$$

TEXAS TECH UNIVERSITY J.H. MURDOUGH ASCE STUDENT CHAPTER

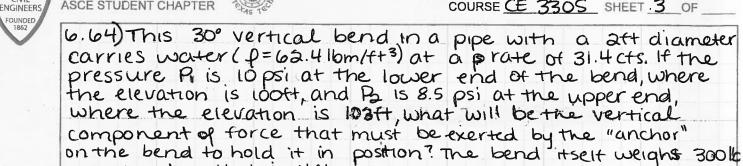


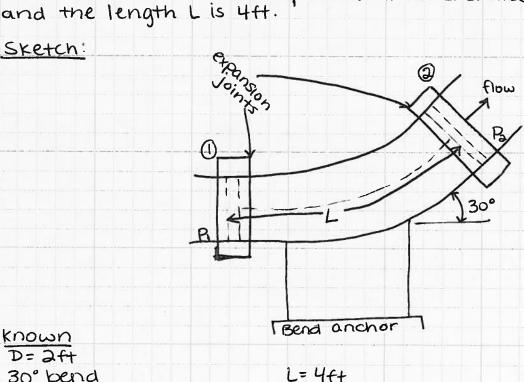
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ASCE STUDENT CHAPTER	COURSE CE 3305 SHEET JOF
6.13 continued)	
5Fx=-myx	
Fx = - (mv,) = PAV,2	
$A = \frac{F_{x}}{pV_{1}^{2}} = \frac{600}{60.416p/f+3 \times 510}$	016f 19/32.2ft/s2(60.9ft/s)2 71872[60016f ft4.52
A=0.083ft2	f+4 · S>
d= 4A = 4(0.083ft=)	= 0.326ft
A=0.324H	

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COURSE CE 330S SHEET 3 OF





30° bend D=62.416m/f+3 Q=31.4C+S P.=10psi == 100f+ Pa = 8.5 psi Za = 103ft WBena = 3001b

unknown!

Fanchar = ?

Governing Equation:

momentum Equation

Solution:

Fa - Wwater - Wbend - Pa Aasin (30°) = pa (vsin (30°) - vsin (0°))

TEXAS TECH UNIVERSITY J.H. MURDOUGH ASCE STUDENT CHAPTER



NAME Solution DATE 3/21/14

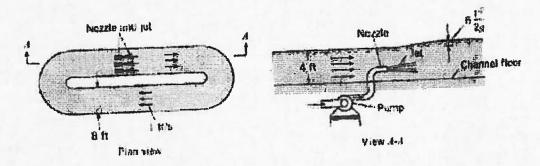
COURSE CE 3305 SHEET 4 OF

TAS 18	00010L <u>CC 3303</u> SILLI 1 01					
6.64 continued						
Fa= TI (2ft)2(4ft) (62.4 1bm/f+3)+	30016+					
$F_{a} = \frac{\pi}{4} (3ft)^{2} (4ft) (6a.4 lbm/ft^{3}) + 300 lb + \frac{144m^{3}}{4} 8.5 lb (\pi (3ft)^{2}) sin(30^{9}) + 1.94 lbf (31.4cfs) (10ft/s (0.5) ft^{4}s^{2}) ln^{2} (4 + 300 lb + 1922.lbf + 304.6 lbf = 3310.lbf$						
Fa = 3310 lbf						

Name:		

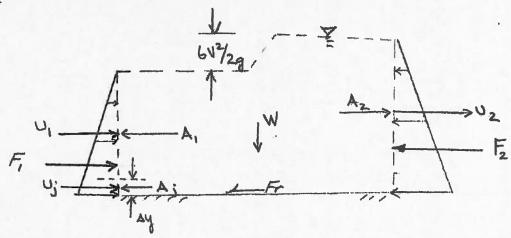
#### Problem 5 (Design Problem)

Jet pumps are sometimes used for special purposes such as pumping slurries or to circulate flow in basins where contact with mechanical impellers is not feasible, such as a fish-race in a hatchery. The figure below depicts the basic concept for this type of application. For the indicated fish-race, the jets would have to increase the water surface elevation downstream of the jets by an amount equal to  $6V^2/2g$ , where V is the average velocity in the basin (1 ft/sec in the drawing). **Design** a jet pump system (speed, nozzle diameter, and number of nozzles) to make such a re-circulating system work for a channel 8-feet wide and 4-feet deep. Your **design** should specify nominal diameters (i.e nozzles made from pipes commonly available 1-in, 2-in, etc.)



Hints: This problem involves "analysis" to determine the required momentum added by the jet(s), and the total jet area and speed. The design should be based on this analysis with the understanding that the total jet area should be small as compared to the total flow area (otherwise the fish will get stuck!).

MIXING ZONE



NEGLECT FRICTION (FINO)

F = FRESSURE FORCE APPROACH

F2 = PRESSURE FORCE EXIT

U, A, = FREE STREAM SPEED & AREA

U,A; = DET STREAM SPEED & AREA

UZ AZ = EVIT STREAM SPEED & ARGA

# MOMEHINM

F,-F2 = - U; QU; A; - U, QU, A, + U2 QU2 A2

A = 84 - 24)
A = 842
A = 844

F, = 694, 9, 8

F2 = 49 72 42 8

g (42-42) = - v3 by - v3 (y, - by) + v2 y2

Note: 12 15 Known!

NOW ARRANGE MOMENTUM IN TERMS U;

SUBSTITUTE U2 FROM CONTINUITY

IMPLICIT EQUATION IN ULAY

SOLVE BY

O TRIAL & FREDR OR

2 NEWTON'S METHOD

VERIFY WHAT'S KNOWN

$$y_2 = 4f + 4 \frac{6(1+1)^2}{Z(32.2+1/2)} = 4.0932 f + g = 32.2 f + 1/2$$

RECALL BY & JET AREA.

U = UNKNOWN

Ay = UNKNOWN

JETS ARE SMALL! (IN SIZE)

$$\frac{U_{i}^{2}\Delta y}{\Delta y} = -\frac{U_{i}^{2}(y_{i} - \Delta y)}{4} + \frac{U_{i}^{2}(y_{i} - \Delta y)^{2}}{4^{2}} - \frac{2U_{i}(y_{i} - \Delta y)U_{i}^{2}\Delta y}{4^{2}} + \frac{U_{i}^{2}\Delta y}{4^{2}} - \frac{e_{i}^{2}[y_{i}^{2} - y_{i}^{2}]^{\frac{1}{2}}}{4^{2}}$$

$$\frac{U_{i}^{2} - U_{i}^{2}\Delta y}{4^{2}} + \frac{2U_{i}(y_{i} - \Delta y)U_{i}}{4^{2}} = -\frac{U_{i}^{2}(y_{i} - \Delta y)}{4^{2}} + \frac{U_{i}^{2}(y_{i} - \Delta y)^{2}}{4^{2}} - \frac{g}{2\Delta y}[y_{i}^{2} - y_{i}^{2}]^{\frac{1}{2}}$$

Now IF WE "FICK" BY ONLY UNKNOWN IS U; . SOLVABLE

SEE FTIAGHED WORKSHEET

FOR DY'S FROM ON TO 0,3

24	U'S	Ai	
0.1	10.1 ft/s	0.8445	Now DETERMINE JET AREAS
0.2	7.0 44   5	1.6 ft2	IN & FOOT CHANNEL
٤. ن	5.6 tt/s	2.4 ft <sup>2</sup>	A; = b4 8 t+

FOR FACH A; , DETERMINE NORRLE SPECS

SEE SECOND WORKSHEET

# INPUT	TOT GARITH	1, Problem #5	net masidu	
#_INIPUT		Annuary to the		
		Approach Spe	180	
y1		Approach Der	RIT .	
y2	4,0932	Exit Depth		
g # DEGIG	32,2	Gravitational (	Jonetant	
	VARIAB			
delta_y		Jet Depth (Are		
uj	10.13858	Jet Speed <	-	
	JTED_VAL			
y1-delta_	3.9			
	-0.75429			
	0.40932			-
u1/2	1			
너^2	102.7908			
# COMPL	JTED VAL			
	Term_1	102.7905	Term_1	-39
	Term 2	-2.51125	Term_2	37.15919
	Term 3	19.32004	Term_3	121,4401
	7.0			
	LHS	119.5993	RHS	119.5993
			ARREST LANGE TO SERVICE	
	DIFF (	9.65E-08 >	-	
#_INPUT				
u1	, 1	Approach Spe	ed	1
y1	4	Approach Dep		1
y2	4,0932	Exit Depth		
g	32.9	Gravitational C	onstant	
DESICH	VARIABI	F(8)	- Automit	
delta y	/02	Jet Depth (Are	al	
ul y	A CONTRACT	Jet Speed	a/	+
THE NAME OF THE OWNER, WHEN	TED VAL			
		TOTAL CONTRACTOR OF THE PARTY O		
y1-delta_y		.,		
	-0.75429			
	0.81884			
u1^2	1			
uj^2	48.77498			1000
			1255	
#_COMPL	TED VAL	JES		
	Term_1		Term_1	-19
	Term_2	-2.38322	Tem_2	17.63901
	Term_3	12.96729	Term 3	60.72004
	LHS	59.35905	RHS	59,35905
	-110	U-00000	RUS	סטייסטיים ס
	DIFF	-2.6E-07		1
i inici m				
# INPUT	- 4	Approach Spec	<b>J</b>	-
ri 2		Approach Dep		-
2		Exit Depth		41 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 -
DECICA	32.2	Gravitational C	onstant	
	VARIABL			
felta y		Jet Depth (Area	3)	
4(	5.608987			1
	TED_VALU	JES	ACCOUNT OF THE PARTY OF THE PAR	
1-delta_y	3.7			
142-4242	-0.75429			
2*delat_v	1.22796	-	www.day.com	
1142	1		100000	
f^2	31.48074			
	TED_VAL			
B-hira Friedrich der Berkerstein	Term_1	31.46074	Term_1	-12.3333
	Term 2	-2.30583	Term_2	11.14857
	Term 3	10.14038	Term_3	40.48003
	LHS	39.29527	RHS	39.29527
0	T			
	DIFF	-4.3E-07		

DEFLECT THIS VALUE -

BY CHANGING THIS CELL

- Use coller to Make this (- )

# CIVE 3434 E	vismil.			
# INPUT	Admir			
u1	1	Approach Speed		
y1	4	Approach Depth		
y2	4.0932	Exit Depth		
g	32.2	Gravitational Constan		
#_DESIGN_VA	RIABL			
delta y	0.1	Jet Depth (Area)		
uj #_COMPUTED	10.1385641236786	Jet Speed		
#_COMPUTED	VAL			
y1-delta_v	=B4-B8		-	
y1^2-y2^2	=84^2-85^2			
y2*delat_y	=B8*B5			
u1^2	=83^2			
u[^2	=89^2			
4 0011011700				
#_COMPUTED	the same of the sa			
	Term_1	=B15	Term_1	=-B14'B11/B8
	Tem_2	=-B15*B8/B5	Term_2	=B14'B11'B11/B13
	Tem_3	=(2*B3*B11*B9)/B6	Term_3	=-0.5*(B6*B12)/B8
	1140	-040,000,004		
Transcent Aven	LHS	=C19+C20+C21	RHS	=F19+F20+F21
	OIFF	=C23-F23		
	DIFF	-043-523	The second secon	
#_INPUT				
ul ul	1	Approach Speed		
y1		Approach Depth		
y2	4.0932	Exit Depth		
72	32.2	Gravitational Constan		
9 #_DESIGN_VAI	PIARI	Graymanchian Odnopari		
delta y	0.2	Jet Depth (Area)	**	
ut	6.98390894304089	Jet Speed		
COMPUTED_		det alaed		
y1-delta_y	=B29-B33			
y1^2-y2^2	=B29^2-B30^2			
y2°delat y	=B33*B30			
u1^2	=B2B^2		4.600	1700c Au
uj^2	=B34^2			
<del></del>				- consider the second s
COMPUTED	VALI			
	Term 1	=840	Term 1	=-B39*B36/B33
	Term 2	=-B40*B33/B30	Term 2	=B39*B36*B36/B38
	Term_3	=(2*B28*B36*B34)/B:	Term 3	=-0.5*(B31*B37)/B33
	LHS	=C44+C45+C46	RHS	=F44+F45+F46
	DIFF	=C48-F48		
#_INPUT				
11		Approach Speed		
1	4	Approach Depth	AND THE PERSON NAMED IN	
2	4.0932	Exit Depth		
u1 v1 v2 DESIGN_VAF	32.2	Gravitational Constan		
DESIGN_VAR	RABI			
della	0.8	Jet Depth (Area)		
	5.60898737282977	Jet Speed		
COMPUTED		Secretary of the secret		
y1-delta_y	=854-858	75607722		
11-2-42-2	=B54^2-B56^2			
2°delat y	=B58*B55			
11/2	=B53^2			
<u>//2</u>	=B59^2		2002	
· Alexandra	A STATE OF THE ABOVE THE ABOVE THE PARTY OF		process and the second	
COMPUTED				
	Term_1	≈865	Term_1	=-B64*B61/B58
	Term_2	=-B86*B58/B55	Term_2	=B64*B61*B61/B63
	Term_3	=(2°B53°B61°B59)/B(	Term_3	=-0.5*(B56*B82)/B58
	4.10			
	LHS	=C69+C70+C71	RHS	=F69+F70+F71
	DIFF	=C73-F73		

10 ft/sec desi	gn						
Diameter(in)			Specify	Total_Actual	_Jet_Area		
	0.005454	146.7	147	0.80			
2	0.021817	36.7	37	0.81			
	-0.049087	163	42	0 83			
4	0.087266		10	0.87			
	0.19635		5	0.98			
<b>8</b>	0 349066	2.3	<u> </u>	1.05			
7 ft/sec desig	1						an demons a the feet. He distributed
Diameter(in)				Total_Actual	_Jet_Area		
	0.005454		293	1.60		AND PROPERTY OF THE PROPERTY O	
2	0.021817	- 73.3	73			5 - M. W.	
3	0 049087	32.6	34	1.67			
4	0.087266	18.3	19	1.66			
6	0.19635	8.1	9	1.77			
	0 349066	8.1 4.6	5	1.75			
5.6 ft/sec des							4 * 1 * * * * * * * * * * * * * * * * *
Diameter(in)	Area(ft^2)	Number		Total_Actual	Jet_Area	Transport of the war	
	0.005454	440.0	441	241		UN MATE TO ME 6100	
2	0.021817	±5,1100	111	2 42			
3	0:049087	48.9	49	2.41	awasii	41	
	0.087266		20	2.44			
	0.19635	12.2	13	2.55			*****
8	0.349066	6.9	7	244		**************************************	
		THE SECTION SECTION SECTION SECTIONS					
All these spec				oon would be	hard to mainta	- حاله در بواف	and 4 0 2 !-
	na require a	f bood to -	acs, and the	uired jet spee	naru to mainta	ui — uius aisc	aru 1,2,3 IN
o-in. Would fe	quire a lot o	nead to g	erierale red	nited let spee	us		