Rotation Matrices Transformations

$$R_{0}^{\circ} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} \qquad R_{1}^{\circ} = \begin{bmatrix} C_{1} & 0 & S_{1} \\ S_{1} & 0 & -C_{1} \\ 0 & 1 & 0 \end{bmatrix} \qquad R_{2}^{\circ} = \begin{bmatrix} C_{1}C_{2} & -C_{1}S_{2} & S_{1} \\ S_{1}C_{2} & -S_{1}S_{2} & -C_{1} \\ S_{2} & C_{2} & 0 \end{bmatrix}$$

$$R_{3}^{\circ} = \begin{bmatrix} C_{1}C_{23} & -C_{1}S_{23} & S_{1} \\ S_{1}C_{23} & -S_{1}S_{2}S_{3} & -C_{1} \\ S_{23} & C_{23} & O \end{bmatrix}$$

Translation vectors

To end of joint To center of muss
$$d_{0j}^{\circ} = 0$$
 0
 0

$$d_{1}^{\circ} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

$$d_{1}^{\circ} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

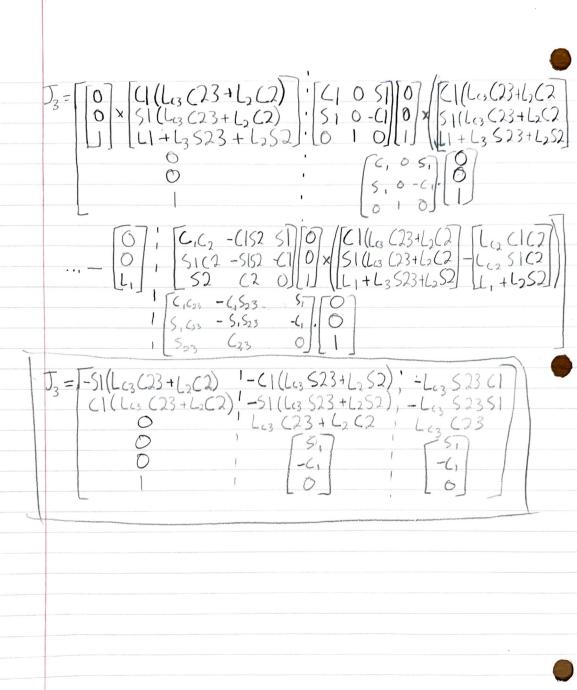
$$d_{2j}^{0} = \begin{bmatrix} c_{2} & c_{1} & c_{2} \\ c_{2} & s_{1} & c_{2} \\ c_{1} & s_{1} & c_{2} \end{bmatrix}$$

$$d_{2j}^{0} = \begin{bmatrix} c_{2} & c_{1} & c_{2} \\ c_{2} & s_{1} & c_{2} \\ c_{1} & s_{1} & c_{2} \end{bmatrix}$$

$$d_{2j}^{0} = \begin{bmatrix} c_{2} & c_{1} & c_{2} \\ c_{2} & s_{1} & c_{2} \\ c_{1} & s_{1} & c_{2} \end{bmatrix}$$

$$d_{3j}^{\circ} = \begin{bmatrix} C_{1}(L_{13}C_{23}+L_{2}C_{2}) & d_{3}^{\circ} c = \begin{bmatrix} C_{1}\cdot(L_{13}C_{23}+L_{2}C_{2}) \\ S_{1}\cdot(L_{13}C_{23}+L_{2}C_{2}) & S_{1}\cdot(L_{13}C_{23}+L_{2}C_{2}) \\ L_{1}+L_{3}S_{23}+L_{2}S_{2} & L_{1}+L_{2}S_{23}+L_{2}S_{2} \end{bmatrix}$$

Jacobians J, = -La (25) $J_{un} = J_n[1:3,:]$ Jun = Jn [4:6]



Jacobian C
Rotation Metrices

Inertia

$$J_{u_1} = \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 6 & 0 \end{bmatrix} \quad J_{u_2} = \begin{bmatrix} -L_0(25) & -L_0(152 & 0) \\ -L_0(162 & -L_05152 & 0) \\ 0 & -L_0(262 & -L_05152 & 0) \end{bmatrix}$$

$$J_{u_3} = \begin{bmatrix} -5| (L_{u_3}(23+L_{u_3}(2)) & -(1|(L_{u_3}523+L_{u_3}52) & -L_{u_3}52351] \\ -(1|(L_{u_3}(23+L_{u_3}(2)) & -5| ((L_{u_3}523+L_{u_3}52) & -L_{u_3}52351] \\ -(1|(L_{u_3}(23+L_{u_3}(2)) & -1|(L_{u_3}523+L_{u_3}52) & -L_{u_3}52351] \\ -(1|(L_{u_3}523+L_{u_3}(2)) & -1|(L_{u_3}523+L_{u_3}(2)) & -L_{u_3}52351] \\ -(1|(L_{u_$$

KE

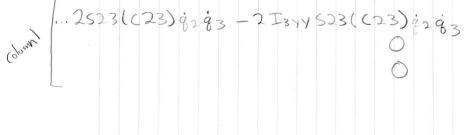
$$\int_{\omega_{1}}^{1} R_{\omega_{1}} \frac{1}{\lambda_{1}} R_{\omega_{1}} \frac{1}{\lambda_{2}} =
 \begin{bmatrix}
 2 & 0 & 0 & 0 \\
 0 & 0 & 0 & 0 \\
 0 & 0 & 0 & 0
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & -c_{1}c_{2} & S_{1} & 0 & 0 \\
 c_{1}c_{2} & S_{1}c_{2} & S_{2}c_{3} & -c_{1}c_{2} & S_{2}c_{3}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2} & S_{2}c_{3} & S_{2}c_{3}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2} & S_{2}c_{3} & S_{2}c_{3}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2} & S_{2}c_{3} & S_{1}c_{2}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2} & S_{2}c_{3}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2}
 \end{bmatrix}
 \begin{bmatrix}
 c_{1}c_{2} & S_{1}c_{2}$$

KE

m Juc Tuci = 0 m, Jves Jves = m, -Les C2S1 Les C1C2 La (2 -LC2 C152. -605152 -Lc2.m2.(52-1) -Lc2 C251 -La CIS2 0 -Lz 5152 Lc2 C1C2 La C2 M3 Just Jus = [-SI((c3 (23+6) (2)) (1((c3 (23+6) (2)) M3. -(1(LC3523+L252) -51(LC3523+L252) (C5C)3+L2 -Les 523C1 -Les 523S1 Les C23 -SI(Lc3 (23+6, C2) -CI(Lc3 523+6, 52 -Les 523Cl (1(Lc3 (23+L, (2) -51 (Lc3523+(252)

-6352351 L2 (23+6) (2 LC3 (23 m3 (LC3 (23+L2 (2)) 1) . M3(12+2C3-12-Lc3+(c3) . Lc3 m3 (Lc3+(2C3) ' La m3 (La+L2 (3) 1 La2.m3

$$D_{(1)} = -m_2 \cdot L_{c_3}^2 (52-1) + m_3 (L_{c_3} (23+L_2 (2)^2 + L_{1}yy + L_{2}yy + S_2^2 - L_{2}yy + S_2^2 - L_{3}yy + L_{3}yy + S_2^2 - L_{3}yy + S_2^2$$



C(q,q) term continued Column - 2 m 3 53 \(\delta_3\) ((q,q)= -Lc3 m3 L2 S3 &3 -LC3 M3 L253 93 G(q) term P=mg-c, + mag(L,+Lc25in(q2))+m3g(L,+L25in(q2)+Lc35in(q3)) 9,(P) = 0 92(P) = mag-les cos (q2) + m39 - 2 cos (q2) 93(P)= m39 cos(q3) (g) = | m2g. Le2 cos(q2) +m3g L2 cos(q2) m39 cos (93)

```
%% HW2: Custom Inverse Dynamics Function
% David Yackzan
% function takes in three 3x1 matrices: position(q), velocity(dq), and
% acceleration(ddq) with values for each of the 3 actuators.
% Outputs a 3x1 matrix for the required torques of the three joints
function [Tau] = custom inverse dynamics(q, dq, ddq)
    % create variables for the input values
    q1 = q(1);
    q2 = q(2);
    q3 = q(3);
    dq1 = dq(1);
    dq2 = dq(2);
    dq3 = dq(3);
    ddq1 = ddq(1);
    ddq2 = ddq(2);
    ddq3 = ddq(3);
    % Create variables for our known values
    q = 9.81;
    m1 = 1;
    m2 = 2;
    m3 = 3;
    L1 = .4;
    L2 = 1;
    L3 = 1;
    Lc1 = L1/2;
    Lc2 = L2/2;
    Lc3 = L3/2;
    % Solve for Dq
    Dq = [(3*L2^2*cos(2*q2))/2 - (67*cos(2*q2 + 2*q3))/200 -
(917*\cos(2*q2))/2000 + \text{Lc}2^2*\cos(2*q2) + (3*\text{Lc}3^2*\cos(2*q2 + 2*q3))/2 +
(3*L2^2)/2 + Lc2^2 + (3*Lc3^2)/2 + 3*L2*Lc3*cos(q3) + 3*L2*Lc3*cos(2*q2 + q3)
+ 2579/2000, 0, 0; ...
          0, 3*L2^2 + 6*cos(q3)*L2*Lc3 + 2*Lc2^2 + 3*Lc3^2 + 2, 3*Lc3^2 +
3*L2*cos(q3)*Lc3 + 1;...
          0, 3*Lc3^2 + 3*L2*cos(q3)*Lc3 + 1, 3*Lc3^2 + 1
    % Solve for C(q, dq)
    Cq dq = [-(3/2)*L2^2*sin(2*q2)*(2*dq2) + (67/200)*sin(2*q2 +
2*q3)*(2*dq2)*(2*dq3) + (917/2000)*sin(2*q2)*(2*dq2) -
Lc2^2*sin(2*q2)*(2*dq2) - (3/2)*Lc3^2*sin(2*q2 + 2*q3)*(2*dq2)*(2*dq3) -
3*L2*Lc3*sin(q3)*dq3 - 3*L2*Lc3*sin(2*q2 + q3)*(2*dq2)*(dq3), 0 0;...
            0, -Lc3*m3*L2*sin(q3)*dq3, -Lc3*m3*L2*sin(q3)*dq3;...
            0, -Lc3*m3*L2*sin(q3)*dq3, 0
    % Solve for G(q)
    Gq = [0; m2*q*Lc1*cos(q2)+m3*q*L2*cos(q2); m3*q*cos(q3)];
    % Solve for the output 3 variable array Tau
    Tau = Dq*ddq + Cq dq*dq + Gq;
```